



UNIVERSITY COLLEGE LONDON

MENG MECHANICAL ENGINEERING

MECH0074 ENGINEERING IN EXTREME ENVIRONMENTS

EXTREME PRESSURE COURSEWORK

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Part I

Review of expansion loops in the energy sector

Part II

Technical assessment of water hammer for expansion loops

1 Frequency of the water hammer pressure fluctuation in a pipe

1.1 MATLAB script

1.1.1 Pressure fluctuation and Joukowski prediction

The maximum magnitude of the water hammer pulse is (1). This is also known as the Joukowski equation.

$$\Delta p = \rho a_0 \Delta v \text{ (Pa)} \quad (1)$$

where Δp is the change in pressure, ρ is the fluid density, a_0 is the sonic velocity in the pipe and Δv is the change in fluid velocity. The sonic velocity can be determined through (2)

$$\frac{1}{a^2} = \frac{1}{c^2} + \frac{\rho D}{\tau E} \quad (2)$$

where c is the speed of sound in the fluid, ρ is fluid density, D is pipe internal diameter, τ is wall thickness of pipe and E is Young's Modulus of pipe. Running the MATLAB script with the values shown in Table 1, Figure 1 was generated.

Parameter	Value
c	1100 m s ⁻¹
ρ	455 kg m ⁻³
D	793.94 mm
τ	19.06 mm
E	200 GPa
Δv	1.5 m s ⁻¹
a_0	1041.89 m s ⁻¹

Table 1: Values for Joukowski Prediction.

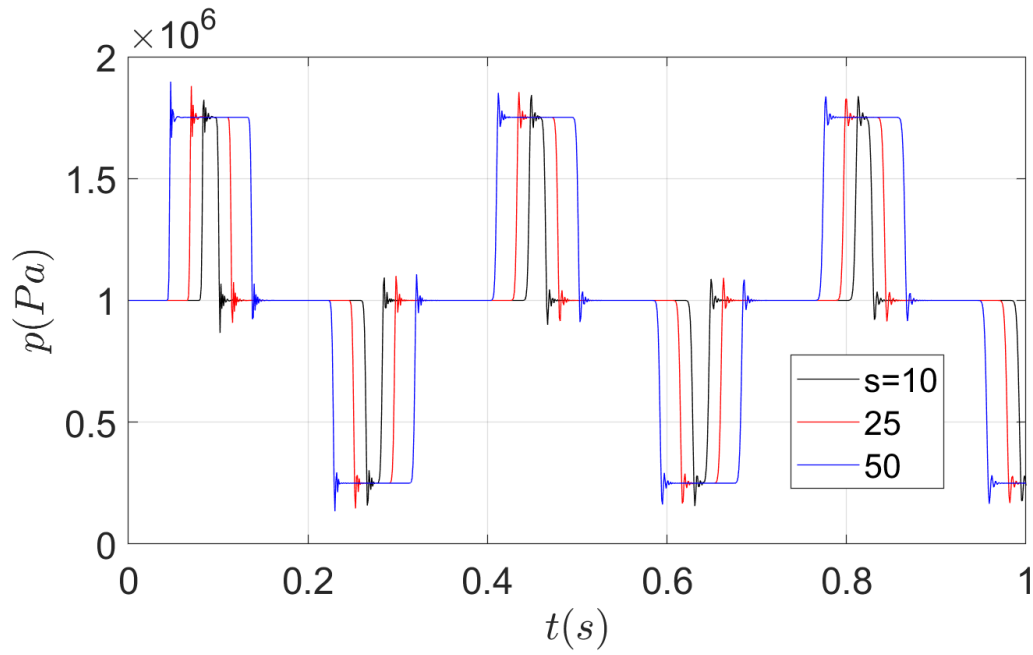


Figure 1: MATLAB script results showing pressure against time for $s = 10, 25, 50$ m.

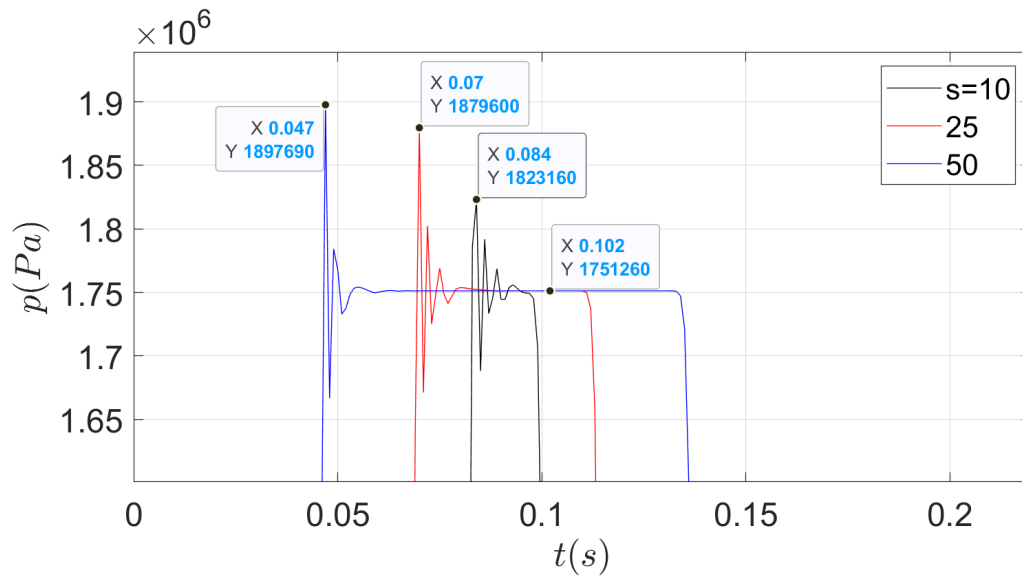


Figure 2: Index points for finding maximum pressure differential and settled pressure differential.

Using the Joukowsky equation, we attain a value for our pressure differential:

$$\Delta p_{joukowsky} = 711\,087 \text{ Pa} \quad (3)$$

Table 2 shows the maximum pressure differentials for each index point (s - value) and the pressure differential of the settled value.

<i>s</i> - value	Maximum pressure differential/Pa	Percentage difference compared to analytical
10	823160	15.76%
25	879600	23.70%
50	897690	26.24%
Settled value	751260	5.64%

Table 2: Pressure differentials for indexed *s*- values and settled values.

1.1.2 Frequencies associated with pressure fluctuation and analytical prediction

An analytical result for the frequencies of the pressure fluctuation can be found using (4).

$$T = \frac{1}{f} = \frac{2L}{c} = \frac{2 \cdot s}{1100} \quad (4)$$

Analysing the period of the pressure differentials, we can find that the frequencies of the pressure fluctuation in Table 3. The index points to find the periods of oscillation were selected as the first peak of the overshoot.

<i>s</i> - value	Frequency/Hz (MATLAB)	Frequency/Hz (Analytical)	Percentage difference
10	10.99	11	0.09%
25	21.74	22	1.18%
50	55.56	55	1.02%

Table 3: Frequency of pressure fluctuation.

1.2 Force analysis

1.2.1 Force acting on bend due to reflection of pressure wave

Using MATLAB, the forces acting on each bend of the loop were calculated. Figure 3 shows a plot of the forces from the first reflection of the pressure wave. Figure 4 shows the forces on each bend for the given time period from each reflection of the pressure wave.

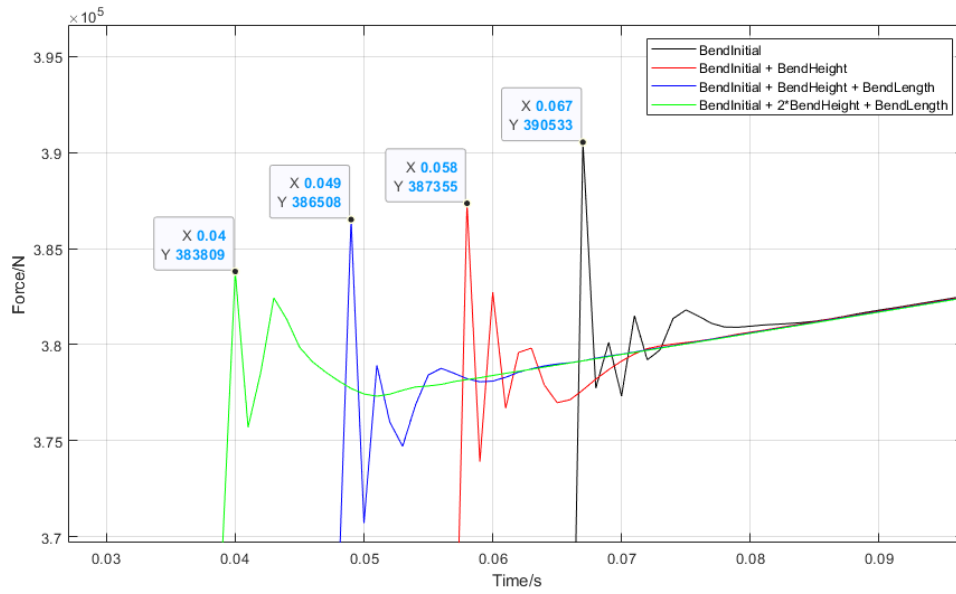


Figure 3: Forces on each bend from first reflection of pressure wave.

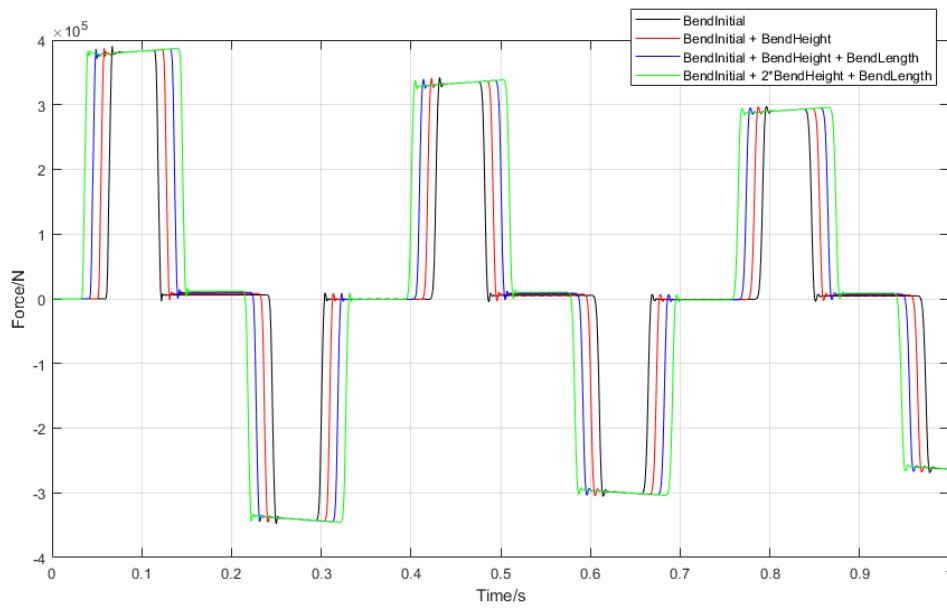


Figure 4: Forces on each bend from reflections of pressure wave over the time period.

1.2.2 Direction of total force on expansion loop

2 Modal analysis of expansion loop

2.1 Modal analysis of a cantilevered pipe

In order to conduct our ANSYS analysis, we need to find the effective pipe density, we can use (5).

$$\rho_{eff} = \rho_s + \rho_{LNG} \left(\frac{d_i^2}{d_e^2 - d_i^2} \right) \quad (5)$$

Table 4 shows the values used in the ANSYS simulations.

Parameter	Value
d_i	793.94 mm
d_e	813 mm
ρ_s	7850 kg m ⁻³
ρ_{LNG}	455 kg m ⁻³
ρ_{eff}	17 214.06 kg m ⁻³

Table 4: Values for cantilevered beam analysis in ANSYS.

2.1.1 List of unique frequencies & mesh sensitivity test

Table 5 shows values of the modal frequencies for a cantilevered pipe with lengths $L = 10, 20, 30$ m. For the theoretical values, we can calculate the bending moment of a cantilevered beam using (6).

$$f_n = \frac{\alpha_n^2}{2\pi} \left(\frac{EI}{AL^4} \right)^{\frac{1}{2}} \quad (6)$$

$$I = \frac{A}{16} (d_e^2 + d_i^2) \quad (7)$$

where coefficient α_n values were found in literature. The theoretical values were calculated using MATLAB (Appendix A.2)

Mode	Frequency/Hz					
	ANSYS Theory		ANSYS Theory		ANSYS Theory	
	$L = 4$ m		$L = 10$ m		$L = 20$ m	
1	31.65	33.86	5.35	5.42	1.35	1.35
2	149.93	212.23	31.49	33.96	8.37	8.49
3	332.21	594.32	80.99	95.09	22.69	23.77
4	521.06	1164.66	143.54	186.35	42.93	46.59
5	712.09	1925.06	214.08	308.01	68.06	77.00
6	857.22	2875.85	289.12	460.14	97.07	115.03

Table 5: Table to show values of modal frequencies for a cantilever beam of length $L = 10, 20, 30$ m, compared to theoretical values.

A mesh sensitivity test was conducted using the $L = 4$ m case. The element size was varied from 0.5 m to 0.01 m and the modal frequencies were found. By taking the frequency of a particular mode and element size and comparing it to the frequency found in the simulation with element size 0.01 m (for the same mode), a percentage difference was calculated. Figure 5 shows how frequency values converge. We see that the percentage difference in frequency for element size 0.125 m and smaller is negligible. Hence, for the simulations conducted, a mesh size of 0.125 m was used.

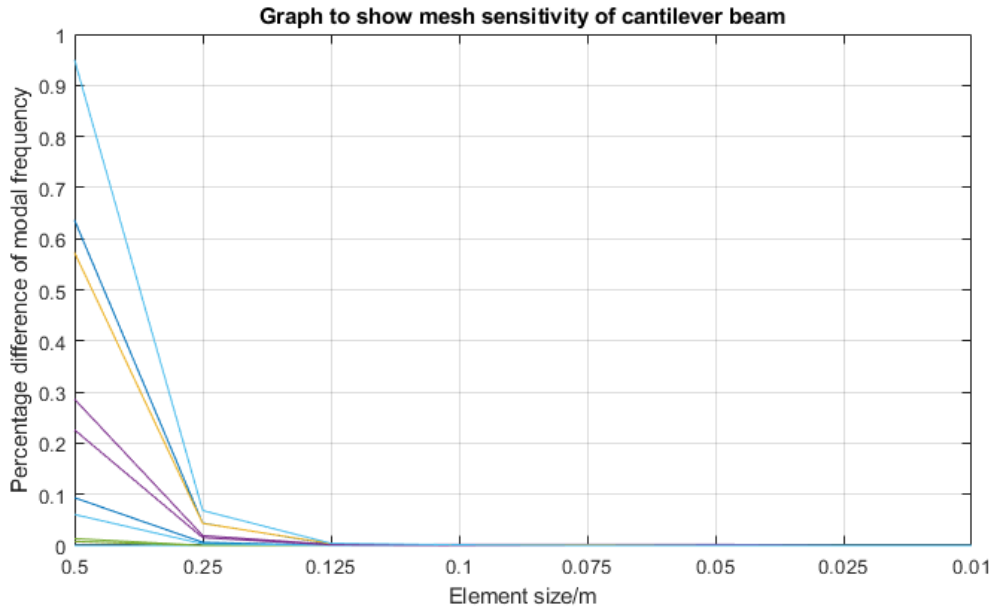


Figure 5: Mesh sensitivity test results. Conducted on a beam with length $L = 4$ m.

2.1.2 Differences and similarities between theoretical and calculated modes

Using MATLAB, the percentage difference between the ANSYS values and the theoretical values was found. This is shown in Table 6.

Mode	Percentage difference (%)		
	$L = 4$ m	$L = 10$ m	$L = 20$ m
1	7.00	1.27	0.34
2	41.56	7.84	1.43
3	78.90	17.41	4.77
4	123.52	29.82	8.52
5	170.33	43.88	13.14
6	235.49	59.15	18.51

Table 6: Table to show percentage difference between ANSYS modal frequencies and theoretical values.

We find that for lower L - values the modal frequencies are higher in our ANSYS simulation. This trend matches with the theoretical values, where modal frequency decreases as a function of L . For the first mode, we find that ANSYS and theory values are closely matched with single digit or decimal percentage difference. For higher modes, the percentage difference increases substantially. As L increases, our percentage difference between ANSYS and theoretical values decreases. As the mode number increases, the percentage difference increases.

As we are using the Euler-Bernoulli theorem to calculate our theory values, the assumptions made can potentially explain the discrepancy between our modal frequency values. One such assumption is that plane sections remain plane. We know that for higher modes, the beam undergoes a larger displacement, as more sections of the beam are displaced from the normal axis. This large displacement results in plane sections no longer remaining plane, making our assumption invalid for higher modes.

2.2 Modal analysis of an expansion loop

3 Discussion and context

A MATLAB code

A.1 Mesh sensitivity plots

```
%HD

clc
clear
close all

%import data
data = readmatrix('data.xlsx','Sheet','L4','Range','A4:P23');
elSize = [0.5;0.25;0.125;0.1;0.075;0.05;0.025;0.01];

%array initialise
A = zeros(numel(data(:,1)),numel(data(1,:))/2);

%calculate percentage difference of result compared to 0.01m el size
for i = 1:8
    for j = 1:numel(data(:,1))
        A(j,i) = 100*(data(j,i*2) - data(j,end))/data(j,end);
    end
end

%plot
for i = 1:numel(A(:,1))
    plot(A(i,:))
    hold on
end
xticklabels(elSize);
grid on
axis tickaligned
xlabel('Element size/m');
ylabel('Percentage difference of modal frequency');
title('Graph to show mesh sensitivity of cantilever beam')
hold off
```

A.2 Theoretical modal frequency values for cantilevered beam

```
%HD

clc
clear
close all

alpha = [1.875;4.694;7.855;10.996;14.137;17.279]; %modal bending coeffs
E = 200e9; %Youngs modulus GPa
L = [4;10;20]; %length of beams
de = 0.813; %external diameter m
di = 0.79394; %internal diameter m
rhoS = 7850; %density steel
rhoLNG = 455; %density gas
```

```
rhoEff = rhoS + rhoLNG*((di^2)/(de^2-di^2)); %density effective
A = (pi*(de/2)^2) - (pi*(di/2)^2); %cross-sectional area m^2
I = (A/16)*(de^2 + di^2); %second moment of inertia m^4

fn = zeros(3,6);
for i = 1:3
    for j= 1:6
        fn(i,j) =
            ((alpha(j)^2)/(2*pi))*((E*I)/(rhoEff*A*((L(i))^4)))^(0.5);
    end
end
```
