

**Czech Technical University in Prague**

Department of Electromagnetic Field



**BACHELOR THESIS**

**Surveillance FMCW Radar**

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Study programme: Elektronika a komunikace

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## DECLARATION

Já, níže podepsaný

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prohlašuji, že jsem bakalářskou prací s názvem

Pohledový FMCW radar

vypracoval samostatně a uvedl veškeré použité informace zdroje v souladu s Metodickým pokynem o dodržování etických principů při přípravě vysokoškolských záloh na nichž prací a Rámcovými pravidly používání umělé inteligence na VUT pro studijní a pedagogické účely v Bc a NM studiu.

Prohlašuji, že jsem v průběhu přípravy a psaní zálohy na práce použil nástroje umělé inteligence. Vygenerovaný obsah jsem ověřil. Stvrdzuj, že jsem si v domě, že za obsah zálohy na práce plně zodpovídám.

V Praze dne 12.05.2025

Kryštof Havránek

.....  
podpis

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TODO

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# Abbreviations

Abbreviation	Meaning
CFAR	Constant False Alarm Rate
AGC	Automatic Gain Control
CW	Continuous Wave
devkit	Development Kit
DBSCAN	Density-Based Spatial Clustering of Applications with Noise
DFT	Discrete Fourier Transform
FMCW	Frequency Modulated Continuous Wave
FFT	Fast Fourier Transform
MIMO	Multiple Input Multiple Output
SNR	Signal to Noise Ratio
SISO	Single Input Single Output
TSV	Tab Separated Values
RCS	Radar Cross Section

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# Introduction

Following theses concerns itself with realization of a surveillance radar based on FMCW technology. Commonly surveillance radars based on FMCW technology have multiple receiving antennas are realized as MIMO systems. In case of MIMO systems electronic steering of the radar beam is possible, allowing for a larger field of view and target detection in two directions. For simpler systems with just multiple receiving antennas only azimuth estimation is possible based on the phase difference analysis [1]. These systems however usually require more processing power and aren't able to provide information about whole 3D space.

Therefore a simpler solution is proposed with radar system that has only one RX and TX antenna where mechanical steering used to control the radar beam direction. At first capabilities of SiRad Easy<sup>®</sup> evaluation board are analyzed and then a custom rotary platform is designed and constructed. This involves both hardware, in a form of two axis rotary platform, and software development be it for driving the rotary platform or processing radar data. Similar systems have been already realized but relied only one axis of rotation [2, 3].

FMCW is a radar technology that uses a continuous wave signal whose frequency is modulated over time. By mixing sent signal over received a harmonic components are created whose frequencies are proportional to the distance of the target [4]. Thus by applying a Fourier transform to the received signal, individual distances can be determined. As opposed to pulsed radar, FMCW is able provide accurate distance measurement with relatively low power consumption [5]. Speed estimations of the target is also possible by using the Doppler effect.

As already stated this thesis is focused on the SiRad Easy<sup>®</sup> radar system. It is a low-cost evaluation board designed to familiarize users with the FMCW technology offering both 24 GHz header and 122 GHz[6]. This allows a relatively larger range of applications with ranges from couple of meters to up to 400 m [7] under ideal condition. However as SiRad Easy<sup>®</sup> offers rather low sampling rate and uses relatively low speed microcontroller on board it is not suitable for any kind of speed measurement. Maximal speeds detectable are in the ranges of tens of mm per second and even then are not very accurate.

For the purpose of this thesis a custom rotary platform was designed and constructed. While such platforms already exists [8, 9] most of them are prohibitively expensive and offer unnecessary features. Thus a custom solution was designed using off-the-shelf and 3D printed parts. Whole system is then controlled with ESP32C6 microcontroller which interpreters G-code like commands and drives the stepper motors.

Data from radar and platform then are processed and visualized using MATLAB. While processing pipeline is relatively standard and uses common techniques such as FFT (Fast Fourier Transform) and CFAR (Constant False Alarm Rate) [10]. However a number of steps can be turned on and off, or their behavior modified by the user allowing a great deal of flexibility. The data are then stored in radar cubes, common structure used in such applications [10]. Processed data can then be visualized in a number of ways depending on user needs – from simple 2D range-azimuth plots to visualization of the whole 3D space.

Thesis is split into five main chapters. First provides theoretical background to FMCW radar workings and its advantages over other possible radar tech-

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nologies. Second chapter is focused on SiRad Easy<sup>®</sup> evaluation board and its capabilities specifically in relation to its application in surveillance radar. Third chapter describes the design requirements and the design process of custom rotary platform. Software needed to drive the rotary platform is also described in this chapter. Fourth chapter is a general overview of the MATLAB desktop application used to control the whole system and process the radar data. With fifth chapter focusing on the data processing pipeline and visualizations in depth.

# 1. FMCW Radar Fundamentals

While classical Continuous Wave (CW) radars broadcast signal only on a single frequency FMCW radars make a linear sweep across given bandwidth. This approach enables range estimation without requiring pulsed transmissions while still allowing speed measurements using the Doppler shift. However, velocity calculations in FMCW radars are more complex compared to single-frequency CW radars.

The "MW" suffix in FMCW radar denotes that the system operates in the microwave frequency range. These high frequencies enable use of compact antenna arrays, that sometimes can even be integrated directly on-chip. Additionally, the millimeter-wave (MMW) portion of the spectrum is typically license-free [11] and offers large bandwidths, reducing the risk of interference.

## 1.1 Comparison FMCW Radar to Pulse Radar

Distance measurement using radar predates FMCW technology by several decades. Early radar systems primarily relied on pulsed signals, where distance can be calculated from the time of flight  $t$  of the signal as

$$d = \frac{c_o \cdot t}{2}. \quad (1.1)$$

Speed calculation is also relatively as Doppler effect is much clearer

$$v = \frac{f_{\text{dop}} c_o}{2 f_{\text{rad}}}, \quad (1.2)$$

where  $f_{\text{dop}}$  is Doppler frequency,  $c_o$  is speed of light and  $f_{\text{rad}}$  is frequency of the radar signal [5].

While this approach is conceptually straightforward, it has several limitations, particularly in applications requiring high precision at close range. Achieving fine resolution in distance measurement necessitates very short pulses. However, to maintain sufficient signal-to-noise ratio (SNR), the transmitted pulse power must remain high, regardless of the number of pulses [5].

Maintaining high average transmission power poses legal and technical challenges. It increases the risk of interference with other devices and demands bulky, high-power circuitry – often requiring high voltages and even vacuum tubes. Consequently, pulsed radar is predominantly used in applications where fine range resolution is not essential, such as long-range target detection.

One key advantage of pulsed radar is its relatively simple data processing. In contrast, FMCW radar data processing is more complex due to the interdependence of distance and velocity measurements – both distance and velocity of a target contribute to frequency changes in the received signal.

## 1.2 Basic principles of ideal FMCW radar

Let us picture an ideal FMCW radar system sending a periodic chirp with periodically repeated frequency sweep from  $f_c$  to  $f_c + BW$ . Placed in in scene

## 1. FMCW Radar Fundamentals

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with single larger target that reflect all incoming signal. Aside from sawtooth waveform other modulations are also frequently used such as linear triangular modulation or segmented linear frequency modulation. These offer some advantages but the nature of the beat signal (which forms a sine wave with sawtooth modulation) is more complex. Especially in case of triangle modulations when multiple targets are present [5]. SiRad Easy<sup>®</sup> kit, used in this thesis, technically uses a segmented linear frequency modulation due to its limited computational power [12] but the mathematical principles are similar as for sawtooth.

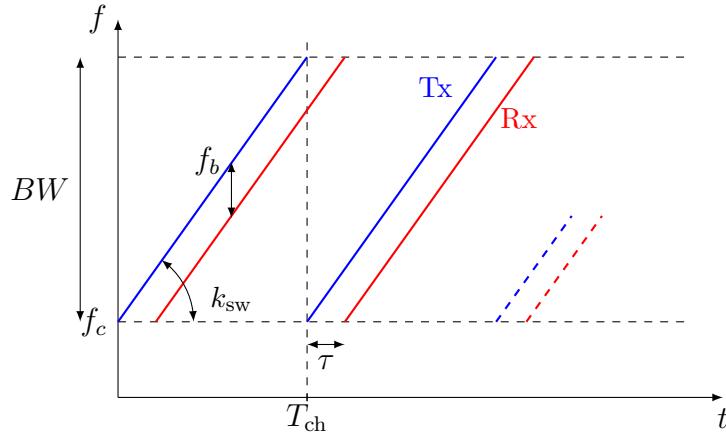


Figure 1.1: Ideal relation of frequency on time for received and sent signal [adler202]

We can clearly see that in given time  $t$  the frequency spread from sent signal to received signal is proportional to the time delay  $\tau$ . However doing some simple subtraction in spectrogram of both signals isn't really feasible – the calculation need to take a smarter approach.

Let us define the chirp slope  $k_{\text{sw}}$  with which we can describe the change in frequency of the broadcasted signal as

$$\Delta f_s(t) = k_{\text{sw}}t = \frac{BW}{T_{\text{ch}}}t, \quad (1.3)$$

where  $t$  is the time goes from 0 to chirp length  $T_{\text{ch}}$ . Standard equation of FM signal can be written as

$$s_t(t) = A \cos \left( \omega_c t + 2\pi \int_0^t f_s(t) dt \right), \quad (1.4)$$

where  $A$  is amplitude of the signal,  $\omega_c$  is carrier frequency and  $f(s)$  is frequency of the signal. Substituting (1.3) into (1.4) we get the signal broadcasted by the radar [13]

$$s_t(t) = A \cos(\omega_c t + \pi k_{\text{sw}} t^2). \quad (1.5)$$

Signal bounced back from the target will have the same equation with the only difference being the time delay  $\tau$ ,

$$s_r(t) = A \cos(\omega_c(t - \tau) + \pi k_{\text{sw}}(t - \tau)^2). \quad (1.6)$$

## 1. FMCW Radar Fundamentals

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Now we can calculate the product of the two signals, this can be done easily in the real world using a frequency mixer. The result of the multiplication is [13]

$$s(t) = s_r(t) \cdot s_t(t) = \frac{A^2}{2} \cos(2(\omega_c - 2\pi k_{sw} T_\tau)t + 2\pi k_{sw} t^2 + (\pi k_{sw} \tau^2 - \omega_c \tau)) + \\ + \frac{A^2}{2} \cos(2\pi k_{sw} \tau t + (\omega_c \tau - \pi k_{sw} \tau^2)). \quad (1.7)$$

First additive term will lead to a signal with very high frequency, well above  $2\omega_c$ , this term doesn't carry any useful information and is usually filtered out – either by low pass filter or the frequency mixer itself. Second term is so called beat signal whose frequency is directly proportional to the time delay  $\tau$ . Deriving the second argument by time we get the frequency of the beat signal [4]

$$f_b = \frac{1}{2\pi} \frac{\partial}{\partial t} (2\pi k_{sw} \tau t + (\omega_c \tau - \pi k_{sw} \tau^2)) = k_{sw} \tau. \quad (1.8)$$

Calculating the distance to the target is now trivial, delay  $\tau$  is equal to the time it takes for the signal to travel to the target and back [4]

$$R = \frac{c_0 \tau}{2}, \quad (1.9)$$

By substituting (1.8) into (1.9) we get the equation for distance [4]

$$R = \frac{c_0 f_b}{2k_{sw}} = \frac{c_0 f_b T_{ch}}{2BW}. \quad (1.10)$$

### 1.2.1 Limits of Range Measurement

Maximal distance is given by the time it takes for the signal to travel from the radar to the target and back. Would the distance be greater than a time of single chirp the signal would be interpreted as coming from a closer target. That gives us an maximal limit on beat frequency  $f_b = BW$ .

However in most case the limit will imposed not by  $T_{ch}$  respectively  $BW$  but by sampling frequency  $f_s$ . In order to avoid aliasing the Nyquist-Shannon theorem must be satisfied thus limiting the maximal beat frequency to  $f_s/2$  and resulting in maximal distance of [5]

$$R = \frac{c_0 f_s}{4k_{sw}}. \quad (1.11)$$

While sampling with frequency  $f_s$  we get  $N = f_s T_{ch}$  samples applying a DFT to the signal we get  $N$  samples in spectrum with frequency resolution of

$$\Delta f_b = \frac{f_s}{N} = \frac{1}{T_{ch}}. \quad (1.12)$$

We can see that the resolution of spectrum is only inversely proportional to the chirp length and doesn't have any relation to sampling frequency [5]. Now we can enter  $\Delta f_b$  into (1.10) to get the minimal distance that can be measured as

$$\Delta R = \frac{c_0}{2BW} \quad (1.13)$$

## 1. FMCW Radar Fundamentals

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Thus in order to increase resolution in range a wider bandwidth is needed.

There are of course other effect impeding the resolution of the radar system – such a phase noise around targets or sweep nonlinearity. Sweep nonlinearity can be either in the ramp itself – leading to decreasing resolution with range (with both linear and quadratic errors present) [4]. Or in sweep recovery (time to return to the start of the sweep) which leads to a constant decrease in resolution [14]. Both are however largely compensated in modern radar systems by using a closed feedback loop [4].

### 1.2.2 Speed Measurement

In order to demonstrate the effect of moving target on the beat frequency we can redefine the time delay  $\tau$  as

$$\tau = \frac{2(R_0 + vt)}{c_0} \quad (1.14)$$

where  $R_0$  is the initial distance to the target and  $v$  is the radial speed of the target. Within a single chirp there is no way to distinguish between the effects distance and speed of the target – thus multiple chirps are needed [13]. We can rewrite the equation (1.14) as

$$\tau = \frac{2(R_0 v(nT_{\text{ch}} + t_s))}{c_0} \quad (1.15)$$

where  $n$  is the number of chirps,  $T_{\text{ch}}$  is the chirp length (or in case of segmented signal whole cycle duration) and  $t_s$  denotes time within a single chirp ( $0 \leq t_s \leq T_{\text{ch}}$ ). Substituting (1.15) into low frequency part (1.7) leads to very complex equation, however according to [13] most of the terms can be neglected leading us to

$$s(t_s, n) = \frac{A^2}{2} \cos \left( \frac{4\pi k_{\text{sw}} R_0}{c_0} t_s + \frac{2\omega_c v n}{c_0} T_{\text{ch}} + \varphi_0 \right), \quad (1.16)$$

where  $\varphi_0$  is a phase shift given by the initial distance to the target. Its clear that first element describes predominantly the distance to the target and the second one the speed of the target. We can also see that speed will not affect a beat frequency in a single spectrum but will lead to a phase shift across multiple spectrums [13].

In order to calculate Doppler shift frequency

$$f_d = \frac{2f_c v}{c_0}, \quad (1.17)$$

where  $f_c$  is the center frequency, 2D Fourier transform can be used. At first FFT is run on each individual chirps and then another FFT is applied to these individual outputs in order to determine phase shift. Given that FFT spectrum is calculated from complex data both halves of the spectrum contain useful information – after applying `fftshift` we get spectrum from negative radial speeds (moving away from the radar) to positive (moving towards the radar) with zero in the middle. This will lead to a so called range-Doppler map which on one axis contains information about speed and on second distance of the target.

Speed resolution is derived from a number of chirps  $N$  we are analyzing and their length  $T_{\text{ch}}$  as [13]

$$\Delta v = \frac{c_0}{2f_c} \frac{1}{NT_{\text{ch}}}. \quad (1.18)$$

## 1. FMCW Radar Fundamentals

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Maximal speed is dictated by how large of a phase shift can we measure without ambiguity – that is  $\pm\pi$ . From (1.16) we can formulate the maximal speed measurable as

$$v_{\max} = \frac{c_0}{4f_c T_{\text{ch}}},$$

according to [15]. But given the nature of FFT this speed applies to negative side of the spectrum - that is for targets moving away from the radar. In case target is moving towards the radar the speed limit is smaller by a single speed bin width.

### 1.3 Radar equation

Fundamental to all radar systems is so called radar equation. This formula takes establishes a relation of transmitted/received power, RCS of the target and other system parameters to maximal range of the radar. It can be written as

$$R_{\max}^4 = \left( \frac{P_{\text{CW}} G_{\text{T}} G_{\text{R}} \lambda^2}{(4\pi)^3 L k T F_{\text{R}} B_{\text{Ro}} (S/N R_{\text{Ro}})} \sigma_T \right), \quad (1.19)$$

where  $P_{\text{CW}}$  is average system power in watts,  $G_{\text{T}}$  and  $G_{\text{R}}$  gain of transmitting and receiving antennas,  $\sigma_T$  is radar cross section of the target,  $\lambda$  is wavelength,  $L$  is system loss,  $k$  is Boltzmann's constant,  $T$  is effective temperature of the system in Kelvin,  $F_{\text{R}}$  is receiver noise figure,  $B_{\text{Ro}}$  is receiver bandwidth and  $S/N R_{\text{Ro}}$  is required signal to noise ratio [5].

Knowing all parameters, or calibrating the system based on known reference, we can calculate target's RCS. This value gives us much clear understanding of the target as opposed to just displaying power received or its logarithm. This is due to RCS being a constant value for a given target and its orientation, regardless of the distance [10].

### 1.4 Angle of Arrival/Spatial Information

Commonly in order to get spatial information radars rely on multiple RX antennas. This allows for estimation of the angle of arrival (AoA) of the target by taking into analyzing phase differences between the signals received by different antennas [13]. However usually only single linear series of RX antennas is employed, therefore only one angle can be estimated.

More complex MIMO systems employ not only multiple RX antennas but also multiple TX antennas. Multiple TX antennas enable beamforming, which is the process of steering the radar beam in a specific direction using positive interference. If antennas are still arranged in a linear array this only makes angle of estimation more accurate by effectively creating virtual RX antennas. However if 2D arrangement of antennas is used, the system can estimate both azimuth and elevation angles [1].

While these systems allow superb performance they are also very expensive and require a lot of processing power. Therefore this thesis works with a simpler and older solution – a system with single RX antenna and TX antenna where radar is pointed to given direction mechanically. This approach is not only cheaper but also allows to gather information about whole 3D space with just a single radar.

### 1.5 CFAR

CFAR is a simple algorithm that enables basic target detection in noisy radar data. For each range bin its neighborhood is analyzed and the average power level is calculated. This neighborhood is separated by a number of guard cells from the cell under test that don't partake in the calculation. In case bin under test is greater than the scaled average power level of the neighborhood, we can assume that there is a target present [5].

There are many different variations of CFAR, each with its own advantages and disadvantages. What was described is a basic cell averaging CFAR (CA-CFAR). This algorithm works best in homogeneous environments, where the noise level is relatively constant across the neighborhood. In case this cannot be assumed for example CFAR with greatest of the neighborhood (CAGO-CFAR) can be used. This algorithm works by taking the maximum value of the neighborhood and comparing it to the bin under test instead of mean [16].

## 2. SiRad Easy<sup>®</sup>

Indie Semiconductor's SiRad Easy<sup>®</sup> is an FMCW radar system development kit designed primarily for automotive applications. Out of the box, it offers two headers a 24 GHz and a 122 GHz both based on ICs from Indie Semiconductor (TRX-024-007 and TRA-120-001, respectively). Both are strictly SISO ICs with two antennas – one for receiving another for transmitting so azimuth estimation from phase shift is not possible.

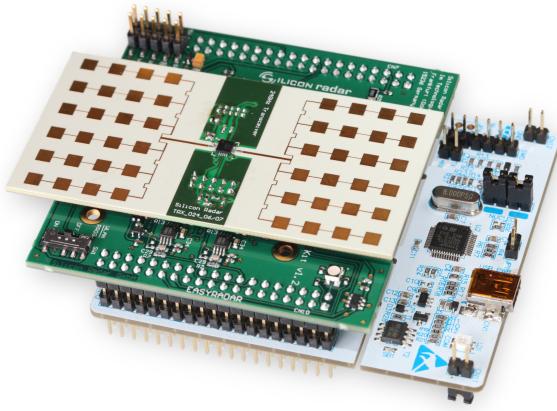


Figure 2.1: SiRad Easy<sup>®</sup> 24 GHz configuration

Direct communication with the radar header is not possible, or at least, the communication interface is undocumented. Instead there is always an intermediary in the form of STM32 Nucleo series microcontroller. To this microcontroller does the user connect either directly with UART over USB or with WiFi enabled by on-board ESP32. Both are relatively low bandwidth communication – the serial maxing at baudrate of 1 000 000 [6].

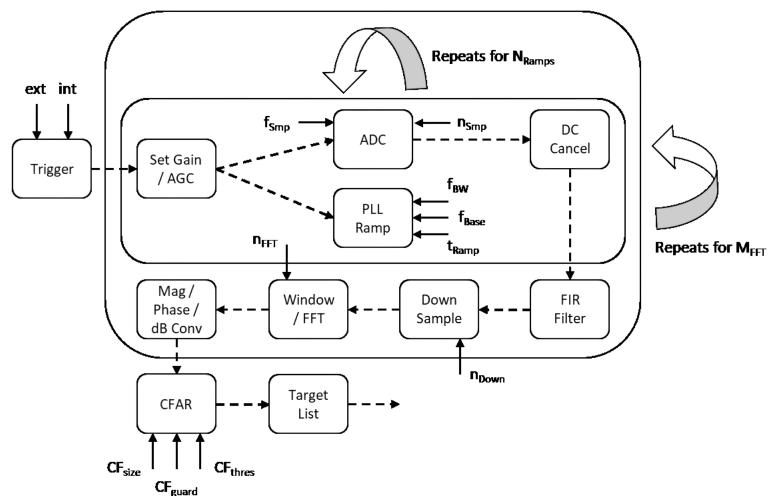


Figure 2.2: Flow of Radar Measurement on SiRad Easy<sup>®</sup>

The devkit is designed for easy integration into existing projects; however, this unfortunately poses limitations for this thesis. In normal operation, the radar system functions as a black box (processing schema shown in 2.2), implementing

## 2. SiRad Easy<sup>®</sup>

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its own filtering, FFT, and CFAR algorithms while only reporting a target list to the user.

This entire sequence can be triggered either internally (driven by a configurable oscillator) or externally (via a GPIO pin or UART input). Once trigger is received the device carries out a user configured number of chirps. The chirp length is determined by the number of sampling steps and the ADC sampling time as follows:

$$t_{\text{ramp}} = \frac{t_{\text{ADC}} \cdot (N_{\text{samples}} + 85)}{36 \text{ MHz}} \mu\text{s}, \quad (2.1)$$

with the manufacturer recommending an optimal time of 1 ms for good SNR [12].

SiRad is design to leverage coherent averaging to improve SNR of the output signal where multiple ramps are broadcasted and averaged. While this technique was found to significantly aid in measurement in a cluttered environments averaging effectively destroy the information about the Doppler shift. Additionally, due to the devkit's limited computational resources instead of sawtooth waveform radars broadcasts a segmented one with significant delays in between[12].

The devkit is also clearly designed primarily for range measurement with little to no regard payed for speed calculation. Not only are chirps spaced with large delays (at minimum some 20 ms) but the radar doesn't even maintain constant intervals between chirps. This is the case even when relying on manual triggering of the radar, when using radar's self trigger results are even worse.

When measuring reporting interval with 20 ms trigger the radar achieved average time between updates of 20.5 ms with standard deviation of 1.3 ms. However from time to time they were outliers when report came in only after 30 ms in the worst case or just 16 ms in the best case. This was on relatively small dataset of 5000 samples. In order to rule out that the problem isn't only in reporting output signal from the radar was captured by Rohde & Schwarz FSW26 Signal and Spectrum Analyzer. By analyzing some 40 samples, measured with the same trigger interval of 20 ms, average time between chirps was 20.7 ms with standard deviation of 0.7876 ms thus proving broadcast itself is irregular.

Unfortunately as the radar doesn't maintain constant intervals between chirps speed measurement precision is severely limited. This could be partially mitigated if we knew when the chirp was broadcasted, giving us ability to introduce a correction. However no such information is provided by the radar. Thus each chip is timestamp with it's time of arrival which is assumed to be the time of chirp broadcast. Respectively broadcast took time a trigger period before but this shift is irrelevant to the outcome. Using this timing information speed spectrum can still be estimated using non-uniform DFT.

Per radars manual [12] following equation was used to calculate width of a single range bin

$$\Delta R = \frac{c_0(N + 85)}{2 \cdot BW \cdot NFFT}, \quad (2.2)$$

where variable  $N$  denotes number of samples and  $NFFT$  number of DFT points used to calculate the spectrum. Except the corrective constant of 85 it closely follows the formula (1.10).

As for velocity the manufacturer doesn't recommend any formula so (1.18) was used with  $T_{\text{ch}}$  substituted by user-chosen trigger period. However as the minimal period of chirps is still rather high not to mention the radar doesn't maintain constant intervals between chirps, the velocity measurement is not very

## 2. SiRad Easy<sup>®</sup>

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practical. Were we to use 24 GHz header with 25 ms trigger we get maximal speed of 0.124 m/s and on low 8 NFFT steps the resolution is only  $0.031 \text{ ms}^{-1}$ .

### 2.1 Outline of the Chosen Configuration

For the purpose of this project, the devkit was configured to output raw data from the 12-bit ADC in the form of in-phase and quadrature components of the signal. This allows for a more complex processing than if spectrum was calculated on the radar itself. For these reasons, the default windowing and signal filtering were also disabled. Device was set up to be triggered by command sent over UART.

Another parameter to consider in relation to surveillance application is AGC. Using it adds two additional ramps that are used solely to set gain value leading to some 3 ms delay. However bigger problem is the fact that radar doesn't report the gain value. Using AGC could then lead to situations where weighting of the data is not constant during operation. Thus it was deemed safest to turn AGC off and set the gain manually depending on how the readings appear after processing.

It is also important to note that the radar system is not well-suited for on-the-fly configuration changes. Applying a new configuration takes some time and the radar does not provide any feedback to indicate when a new configuration was applied. As a result, commonly used techniques such as alternating chirp slope are not feasible.

In regards to the output format, the radar system supports two options: binary and TSV. Since the output speed does not differ significantly between the two, the primary deciding factor was ease and speed of parsing. Even when MATLAB is not particularly optimized for parsing binary data, it still processes binary output approximately 40% faster than human-readable TSV data.

Other parameters such as sampling frequency, number of samples and so on are left to the user to configure. Setting these strongly depends on what header is used and what environment is the radar used in. Thus making their configuration static wouldn't be a good idea.

### 2.2 24 GHz Header

Center of 24 GHz header is a SISO TRX-024-007 transceiver which integrates low noise amplifier, frequency mixer, filters and VCO into a single chip. It is primarily designed to operate in the ISM band (24.0–24.25GHz), with an additional ultra-wideband mode supporting 23–26 GHz [17]. On the SiRad Easy<sup>®</sup> there is no distinction made between those two modes and the user is free to set any bandwidth [12]. The transmitter output power ranges from 2.5 dBm to 6 dBm, depending on the configuration [17]. A maximum range of 400 m is advertised [7], though this is likely under ideal conditions when observing a large target. Also as previously stated the radar will operate on much lower sampling frequency than needed to achieve such results.

As shown in Figure 2.1, the chip is connected to two microstrip patch antennas. The patches are arranged in a relatively standard configuration, forming a  $6 \times 4$  array with spacing approximately equal to half the wavelength at 24 GHz.

Since the manufacturer did not provide any information about the radiation pattern of the array, a simulation was conducted using the CST simulation

suite. Additionally, the manufacturer did not disclose the substrate parameters for the radar board specifically. Fortunately, the TRX-024-007 datasheet includes a board stack-up for the chip's evaluation board. It was assumed that the same stack-up would be used—18  $\mu\text{m}$  copper for the traces and ground plane, with a 250  $\mu\text{m}$  thick Rogers RO4350B substrate [17].

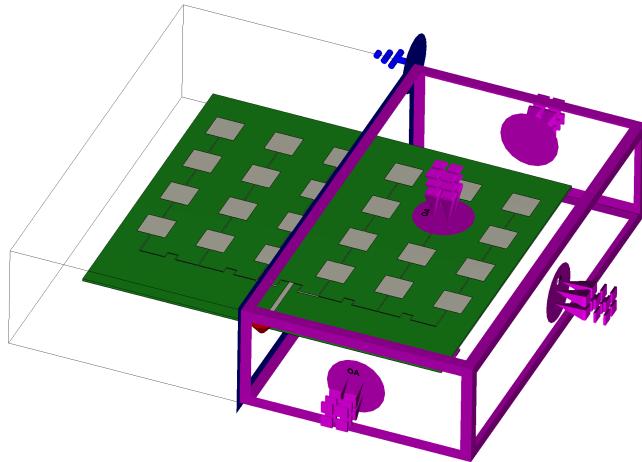


Figure 2.3: Simulated 24 GHz header with boundary conditions showed

After measuring the dimensions of the array using an optical microscope, the entire array was redrawn in CST Studio. Figure 2.3 shows the applied boundary conditions: the antenna was placed in open space, and an  $H = 0$  condition was set along the symmetry plane to speed up the simulation.

After performing a standard time-domain simulation with an excitation signal ranging from 0 to 26GHz, the antenna array exhibited a minimum reflection coefficient  $s_{11} \doteq -26.4$  dB at 23A.478 GHz with second minimum at 24.518 GHz (Figure 2.4). Lack of minimum at 24 GHz may be attributed to measurement difficulties (The etching quality of the copper traces was suboptimal.) and the neglecting variations of substrate parameter due to interactions of different layers.

At both frequencies with minimal reflection, a far-field radiation pattern was calculated. For clarity, only the 24.518 GHz pattern is shown in Figure 2.5. The main lobe width was measured at approximately 16 degrees (Figure 2.6) along the 180-degree norm (for orientation refer to the red cone on the PCB), with a peak gain of 18.6 dBi and side-lobe suppression of -13 dB. For the 90-degree norm, the main lobe width (Figure 2.7) was 30 degrees, with side-lobe suppression of -10 dB.

### 2.3 122 GHz Header

The 122 GHz header is based on the TRX-120-001 transceiver, which, in addition to the essential components required for RF transmission and reception, also incorporates two on-chip antennas. It's designed to operate in the 122-123 GHz band, with output power ranging from -7 dBm to 1 dBm [18]. The chip is capable of detecting large targets at distances of up to 40 m [7].

Out of the box performance of the system is quite bad (Figure 2.9) with width of the main lobe being roughly  $\pm 40^\circ$  in both E-plane and H-plane [18]. However,

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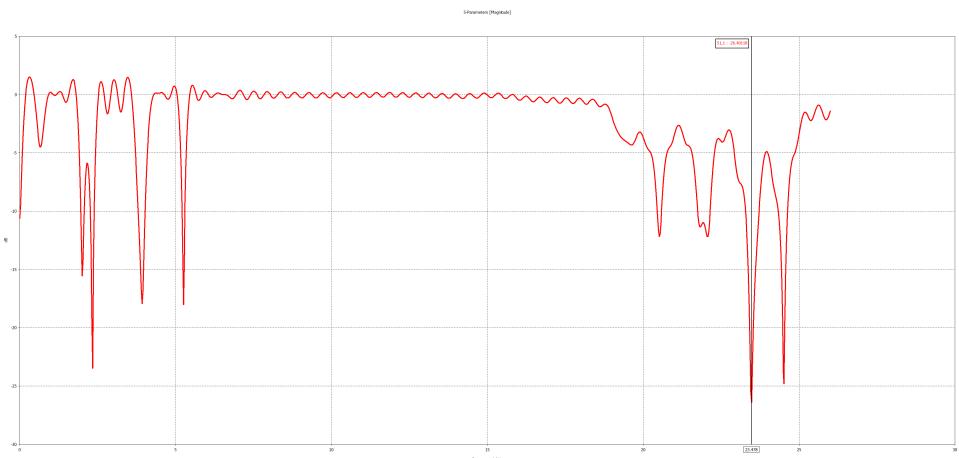


Figure 2.4:  $s_{11}$  parameter of the 24 GHz header

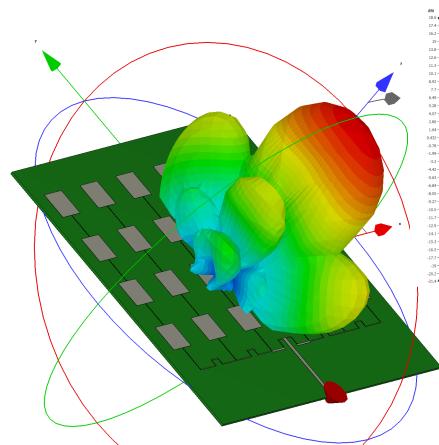


Figure 2.5: Radiation pattern of 24 GHz header – 3D view

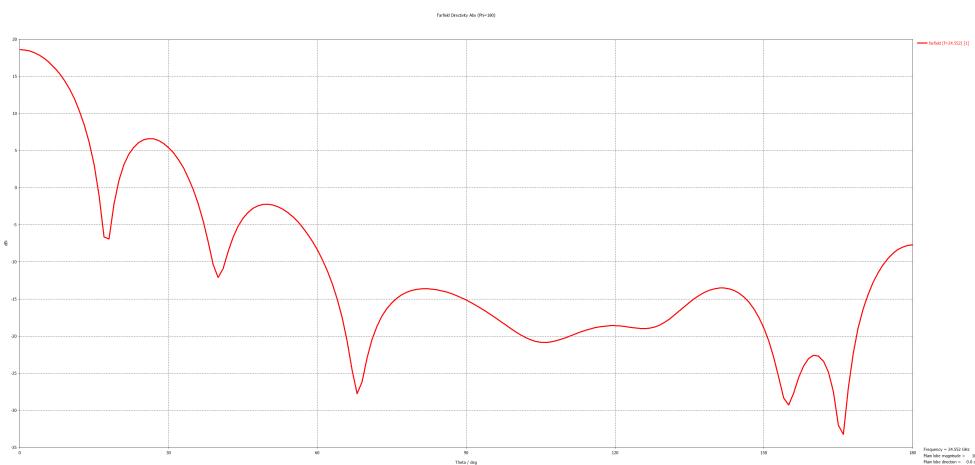


Figure 2.6: Radiation pattern of 24 GHz header – 180° norm

this can be significantly improved using the supplied collimator lens, reducing the main lobe width to  $\pm 4^\circ$  [19] (see Figure 2.9).

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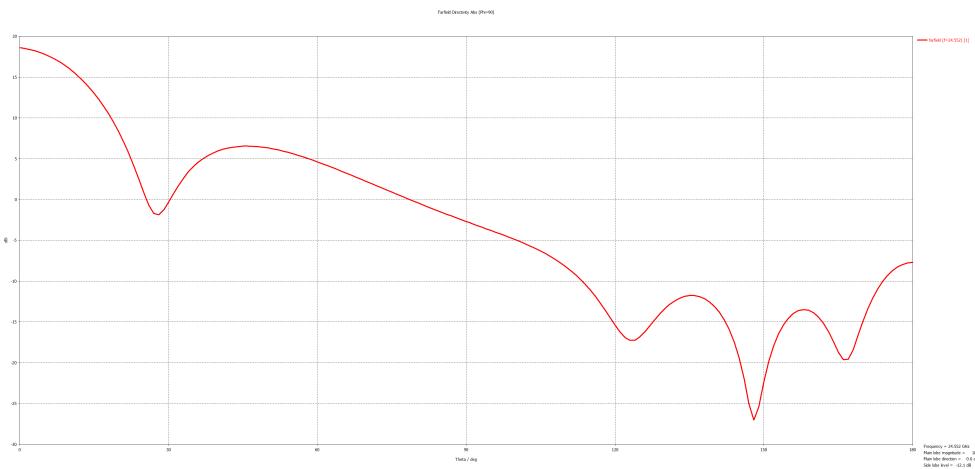


Figure 2.7: Radiation pattern of 24 GHz header – 90° norm

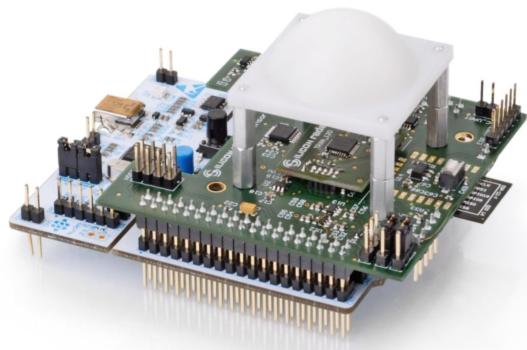


Figure 2.8: SiRad Easy<sup>®</sup> with 122 GHz header

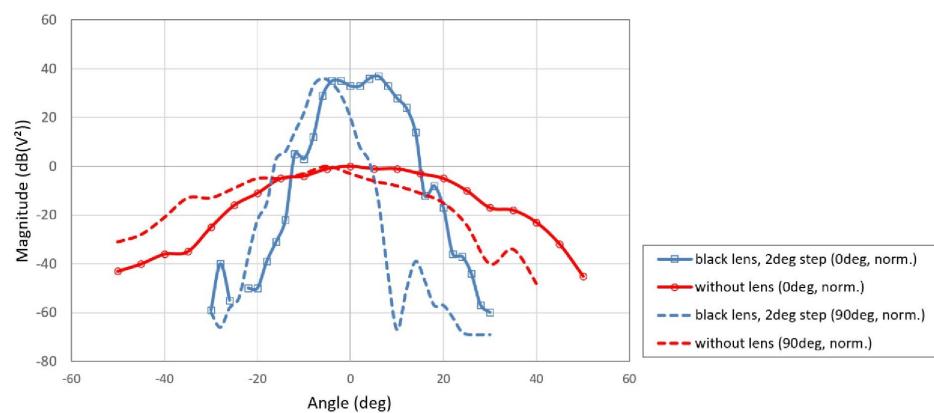


Figure 2.9: Radiation pattern of 122 GHz header comparison

## 3. Rotary Platform

Following chapter outlines design process and the operation of a rotary platform specifically designed for SiRad Easy<sup>®</sup> radar system.

### 3.1 Platform Construction

As the platform needs to transmit data from the rotating section to the stationary base, a slip ring is required. Due to the relatively low transmission speed of the radar and the absence of special requirements such as waterproofing, an affordable model, UH3899-01-0810 from Senring, was selected. It is a classical contact slip ring that features a dedicated USB 2.0 connection along with 8 additional signal wires, with an advertised insertion loss of less than 2 dB [20]. More problematic than loss is however cross talk between signal wires when stepper motor responsible for pitch control of the radar is running. This necessitated an addition of small capacitor to the endstop signal pin to filter out high frequency noise.

While USB 2.0 connection doesn't seem affected by the crosstalk it's connection still isn't ideal. Without signal conditioner connected directly to the slipring output signal quality is bad enough that direct connection to PC is needed. And even then depending on PC's wiring some ports might exhibit issues. In addition the manufacturer opted for a non-standard male-male USB 2.0 connection, requiring a female-female adapter to connect the radar to the slip ring.

Due to the relatively low angular resolution of the radar, high platform precision was not a requirement. A basic 200-step stepper motor with a step size of 1.8° was deemed sufficient. In case user would require more granular control over the position it can be facilitated by microstepping.

Similarly high-speed movement is unnecessary for this application. As was already outlined the radar is capable at best of 50 Hz update rate. With such update frequency even at low 8 RPM we travel a degree under 21 ms too right on the edge of what is radar capable. Also at these RPMs standard 40 mm NEMA17 stepper motors are powerful enough to turn the slipring.

The rest of the design is relatively simple. The fixed section mounts the slip ring with the stepper motor positioned underneath, directly driving a shaft connected to a rotating base. The connection is secured using long M4 set screws that pass through the base the slip ring and hold the shaft in place. A 3D-printed housing serves only as a centering guide and is not responsible for connecting shaft to the rotary base.

The rotating section features a simple A-frame design that elevates the radar, which is mounted on bearings, allowing it to tilt freely. To control the pitch, a second stepper motor is mounted on the rotating platform and linked to the radar via a 2:1 down-gearing ratio using a standard 8 mm belt. An optical endstop, used for homing of the platform, is mounted on the second support strut.

Unfortunately while the platform offers some ability to tension the belt in currently printed configuration the range is not sufficient to fully tighten the belt. While slippage is not a problem radar there is some slack in radar's tilt. However as the radar's capabilities in pitch direction aren't particularly high it

### 3. Rotary Platform

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was decided not to reprint the whole assembly, only to fix issue in supplied 3D models.

Given that mechanical stresses are minimal, most parts can be 3D-printed using standard PLA filament. The only non-3D-printed components are the screws, bearings, and stepper motors. The final assembly (Figure 3.1) measures approximately 33 cm in height and has a footprint of  $20 \times 20$  cm.

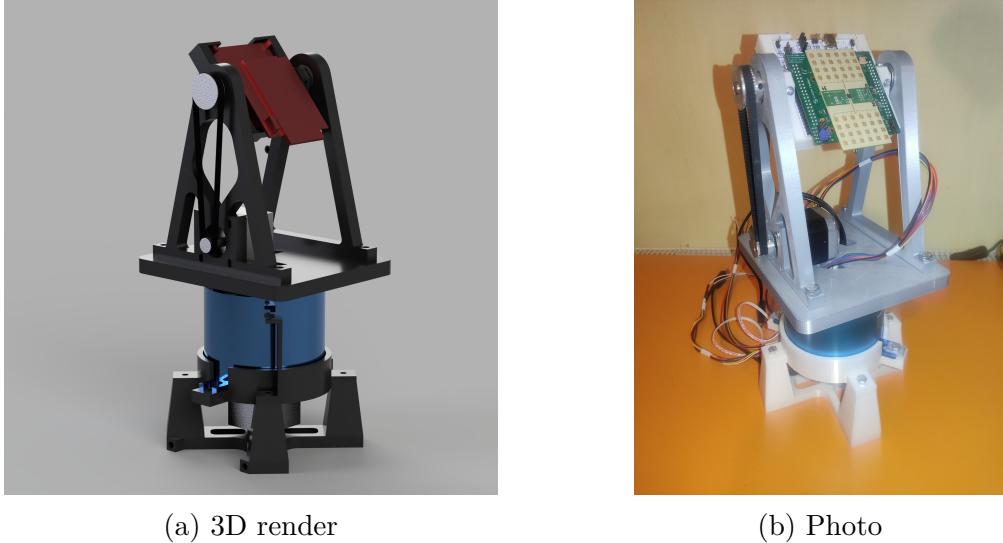


Figure 3.1: Form of the final assembly

#### 3.1.1 Platform Electronics

The electronic aspect of the platform is relatively simple. Only two main problems needed to be tackled – driving the stepper motors and implementing a homing mechanism.

Given the low load on the stepper motors and the platform's inability to accumulate significant momentum, a basic stepper driver without feedback control was deemed sufficient. For this purpose the A4988 stepper driver was chosen due to its low cost, microstepping capabilities and basic current control [21]. A minor drawback is the lack of feedback from the driver to the microcontroller, including the absence of stall detection. To simplify the design, A4988 development kits were used and soldered onto a prototyping board, eliminating the need for a custom PCB. In addition if the need arises A4988 development kit can be replaced by one with a different IC, such as TCM2209, as manufacturers keep the same pinout and dimensions on these complementary kits. This might be necessary in application where platform loudness would be a serious problem as A4988 produces rather noisy driving signal leading to even 20 dB higher operating volume [22].

For homing implementation, two potential solutions were considered: Hall effect sensors and optical gates. Hall effect sensors offer the advantage of angle sensing, allowing correction for positional drift during operation; however, they require precise alignment. If the orthogonal Hall effect sensor is not perfectly placed along the axis of rotation, calibration becomes necessary [23]. While feedback would be beneficial, the microcontroller already tracks each step taken by

### 3. Rotary Platform

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the motor during normal operation, making it possible to determine the platform's position purely in software. Thus for simplicity and ease of integration, optical gates were selected.

The system is controlled by an ESP32 microcontroller. The ESP32-C6 version was chosen due to the author's extensive experience with this particular model. Still since the system does not require specialized peripherals or high processing power most other microcontrollers could have been used instead.

## 3.2 Platform Software

Given its widespread adoption as an industry standard for controlling multi-axis machines, G-code over serial is a natural choice for the platform's communication format. Beyond the basic functionality typically offered by G-code interpreters, the platform must support additional features to reduce the user's manual control burden. These features include common features like defining movement limits, absolute/relative positioning mode, spindle mode or enabling automatic homing. Lastly the platform should support a programming interface that will allow application to send whole choreography of movements to the platform at once and from that point only monitor its status.

For uplink communication, the platform must provide real-time information about its current position and speed. This data enables the user to make any mathematical corrections and properly interpret radar's gathered data.

To maximize efficiency in processing commands and ensure accurate stepper motor control, the program workflow is divided into three distinct layers, as illustrated by figure 3.2. The commonly used two-component architecture—where one component handles communication/command parsing and the other manages execution was deemed unsuitable for this use case. Such approach would complicate integration of programming interface and require just-in-time processing of commands, which could lead to performance issues.

In the chosen architecture, the degree of abstraction decreases with each successive layer, simplifying processing at each step. This design allows the final layer to operate with maximum efficiency, where transition from one command to the next is primarily limited by the inertia of stepper motors and not by the software.

Also it is necessary to note that the platform doesn't store any permanent configuration, including microstepping settings. This approach was taken as platform doesn't offer any easy way to validate current setting and is expected to be only operated from computer application, which can handle this and provide user with feedback.

### 3.2.1 Communication Layer

The communication layer manages incoming data over the serial line. Whose efficient handling is facilitated with the aid of RTOS queues. Upon receiving command string is parsed and either pushed to a queue or added to programm declaration, in case we are currently declaring program. Some special commands are executed immediately, like those enabling emergency shut off. For complete overview of supported commands consult Appendix A.

### 3. Rotary Platform

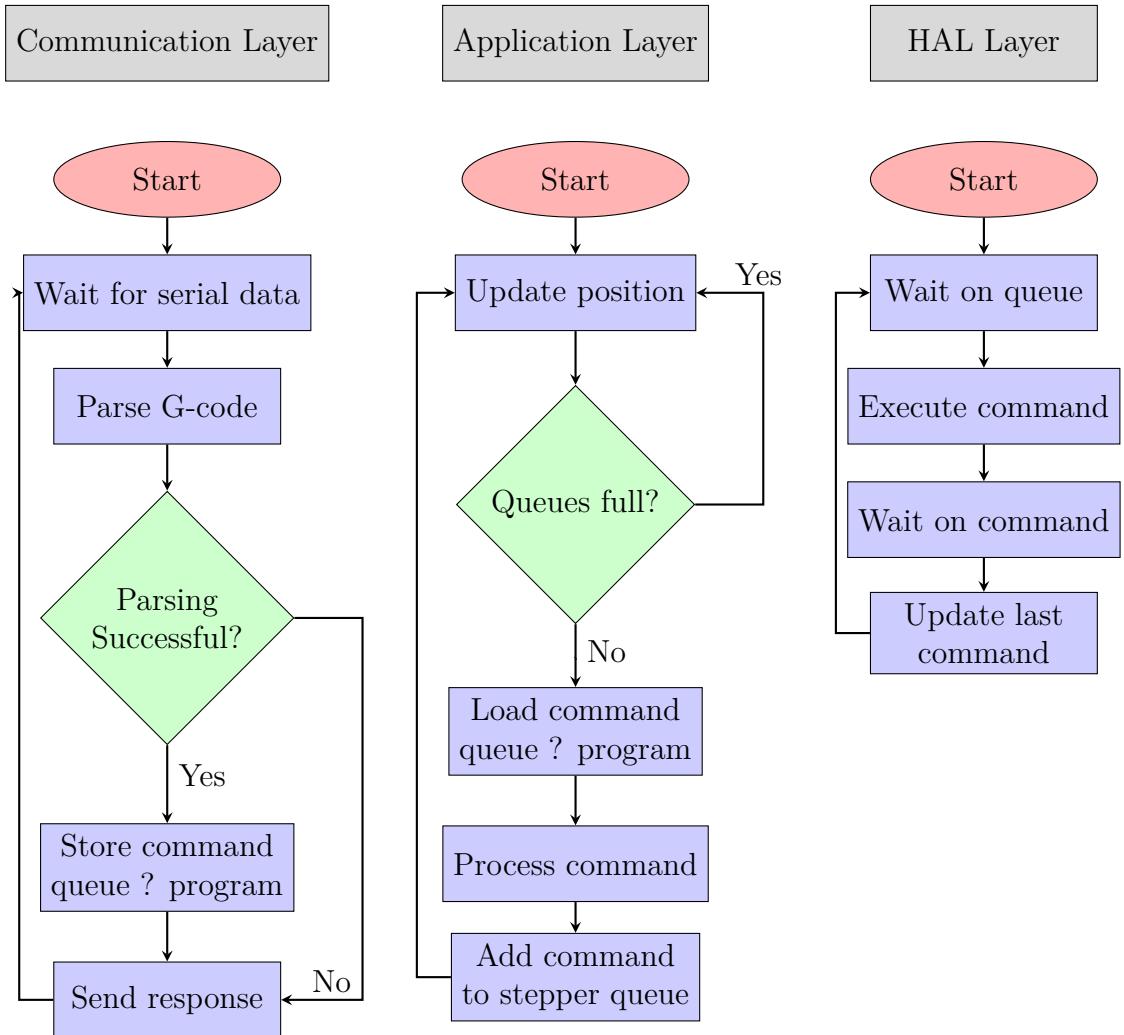


Figure 3.2: Program flow diagram

Immediately after parsing, a response is send to the user confirming whether the command was parsed correctly or not. However, as the communication layer does not a can not check command within context of all previous commands, it is possible that command will be parsed correctly but its execution will fail in the application layer.

#### 3.2.2 Application layer

The application layer performs two primary functions: tracking the current device position and scheduling commands to be sent to stepper motors. Aside from current position the program also keeps track of the end position of the last scheduled command. Thanks to this the application layer make necessary calculations to facilitate absolute positioning and enforce movement limits.

A key departure from standard G-code interpreters, like [24], is how the platform handles single-axis move commands. When a move command targets only one axis, the other axis remains free to read next command and begin its execution. If this behavior is undesirable, the user must issue commands for both axes. In relative positioning mode, a zero value results in no motion; in absolute positioning mode, the command must specify the current position to prevent

### 3. Rotary Platform

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movement.

This behavior is a necessary side effect of the spindle regime, which typically cannot be toggled on or off dynamically. Another consequence is the requirement for separate positioning modes for each axis. Continuous rotation prevents calculations of a move's end position, making it impossible to make calculation for absolute positioning commands – thus necessitating relative positioning. However it would be rather restrictive to force user to relative positioning on second axis, therefore the independent positioning settings.

In order to support or possible operating regimes a manual override mode was also implementing. This enables the user to manually push a move command directly to stepper queues totally skipping the application layer. Primary usecase of this mode is to allow tracking of targets or other application that require real time control of the platform. However in this regime no limits are enforced and the platform operates strictly in relative positioning mode.

#### 3.2.3 HAL Layer

The final layer manages stepper motor control and provides the application layer with essential data for position calculations. In its loop, the program waits for the next command in the stepper queue. Upon receiving a command, it sets up execution, waits for one or both steppers to complete their movement, and then proceeds to the next command. Since limit and absolute positioning calculations are handled in the application layer whole routine remains highly efficient.

The main challenge was in generating precise PWM signals (Used to control stepper motors drivers.) and stopping signal generation after a specific number of steps. As generating PWM signal in approach is unpractical dedicated ESP32C6 peripherals need to be leveraged. Ther are two main options for this task: Remote Controlled Transceiver (RMT) and Motor Control Pulse Width Modulation (MCPWM) combined with Pulse Counter (PCNT). While RMT allows smooth PWM frequency adjustments, it has several drawbacks. Such as the fact that generating a specific number of pulses is supported only on newer ESP32 models [25], synchronization is restricted to its proprietary API and there is no straightforward way to track progress during a move [26].

For these reasons, MCPWM and PCNT were chosen. MCPWM handles pulse generation, while PCNT counts steps, enabling easy control over the movement and providing robust API for step count tracking [27]. The only limitation is the PCNT's 15-bit counter, which caps the maximum steps per move at  $\pm 32,767$ .

#### Performance of the HAL Layer

Table 3.1 illustrates the stability of PWM generation by the MCPWM module at various speeds. Measurements were conducted using a Saleae Logic Pro 16 logic analyzer, with no microstepping enabled.

The results show that frequency deviation is minimal, though the generated speed is consistently marginally faster than the target, and the error increases slightly with higher speeds. Nevertheless, when measuring time of 24,000 steps at 120 RPM, the relative error in time duration (or speed) was only  $\epsilon = -0.004\%$ , demonstrating excellent accuracy.

An attempt was made to also measure the delay between switching commands,

### 3. Rotary Platform

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Table 3.1: Stability of PWM generation

RPM	$f_{\text{desired}}$ (Hz)	$f_{\text{low}}$ (Hz)	$f_{\text{high}}$ (Hz)	$f_{\text{avg}}$ (Hz)
10	33.334	33.334	33.334	33.334
30	100	100	100.003	100.002
60	200	200	200.01	200.004
120	400	400	400.02	400.007

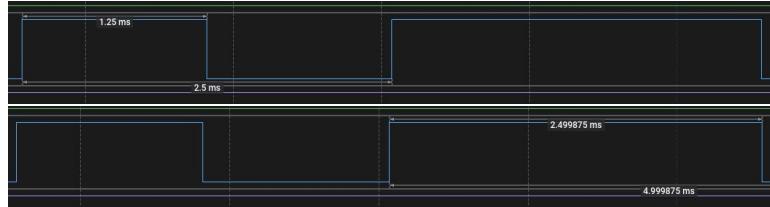


Figure 3.3: Moment of change between commands (120RPM  $\Rightarrow$  60RPM)

displayed in figure 3.3. The results indicate that the delay between commands is imperceptible. Similar outcomes were also observed for other command combinations.

This demonstrates the efficiency of the HAL layer in managing stepper motor control and transitioning seamlessly between commands. As long as stepper queues are supplied with commands in advance, the platform can operate without noticeable interruptions. Most importantly, the platform's timely and predictable behavior ensures that mathematical corrections to the radar data can be applied accurately.

# 4. Control application

As previously stated control application for the whole system is written in MATLAB. This application is split into two distinct parts - one is managing rotary platform and the other is processing radar data. Alongside these there are some shared components, such as preferences menu.

## 4.1 Main application interface

Upon starting the application user is presented with the main window (Figure 4.1). This window is primarily used to display radar data and control basic functionality of the application. On the sidebar user can enable serial connections to the radar and the platform, pause processing, save current visualization to jpg and in case of an emergency stop the platform. Lastly topbar is used to access preferences and platform control interface.

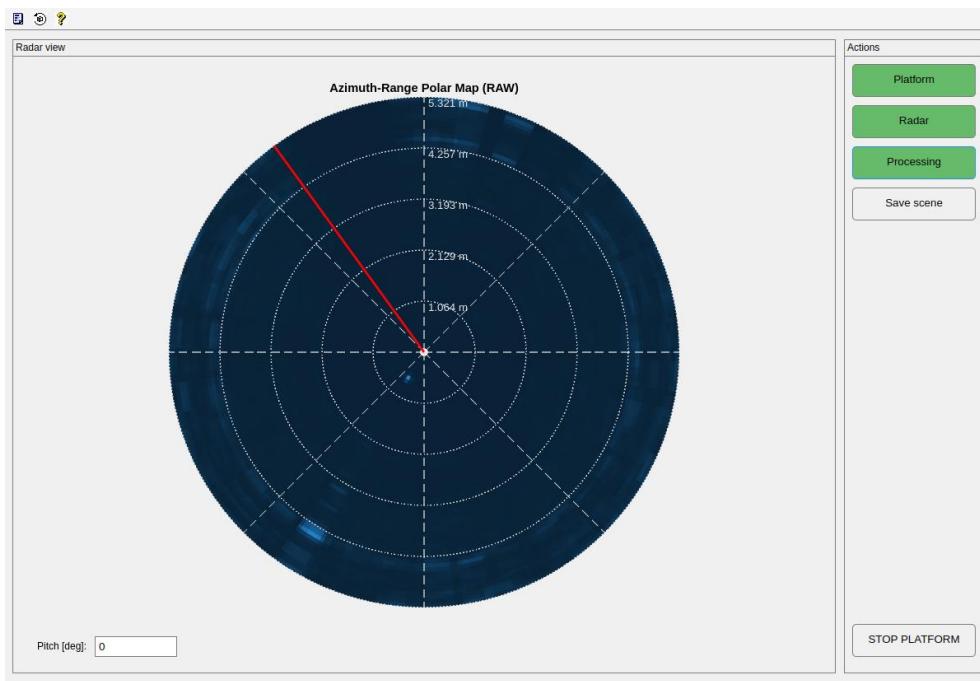


Figure 4.1: Main application window

### 4.1.1 Configuration options

As this thesis concerns itself with rather generic implementation of surveillance radar system the application is designed to be as flexible as possible. For this purpose GUI menu was designed to enable configuration of all important elements - configuring the platform, radar, data processing and data visualization. However it is necessary to point out that given extensive configuration options user is expected to have some knowledge about the subject to achieve satisfactory results in given application. Firstly – configuration options aren't checked against each other if they aren't mutually exclusive. For example speed calculation ramp averaging must be turned off, but this as things stand application will allow user

## 4. Control application

to continue. Secondly – configuration of components such a CFAR or DBSCAN strongly depended on bandwidth, samples, NFFT and other parameters. CFAR guard parameter of 3 might be too small on 5 GHz bandwidth where width of single range bin is only 50 mm, but if used on 24 GHz header with  $BW = 500$  MHz where range bin width is 50 cm it could be way too large. To aid with choosing correct parameters the application provides some basic information like dimensions of the bins however it is up to the user to make the final decision.

Figure 4.2 shows the preferences menu, as one can see there are four main sections. First configures serial port connections to the radar and the platform. However baudrates are usually fixed – 230400 or 1000000 for the radar, depending on the firmware used and 115200 for the platform. Second section handles configuration of the platform, followed by a section designated for radar configuration. Last section configures how the data are processed and via which method they are visualized.

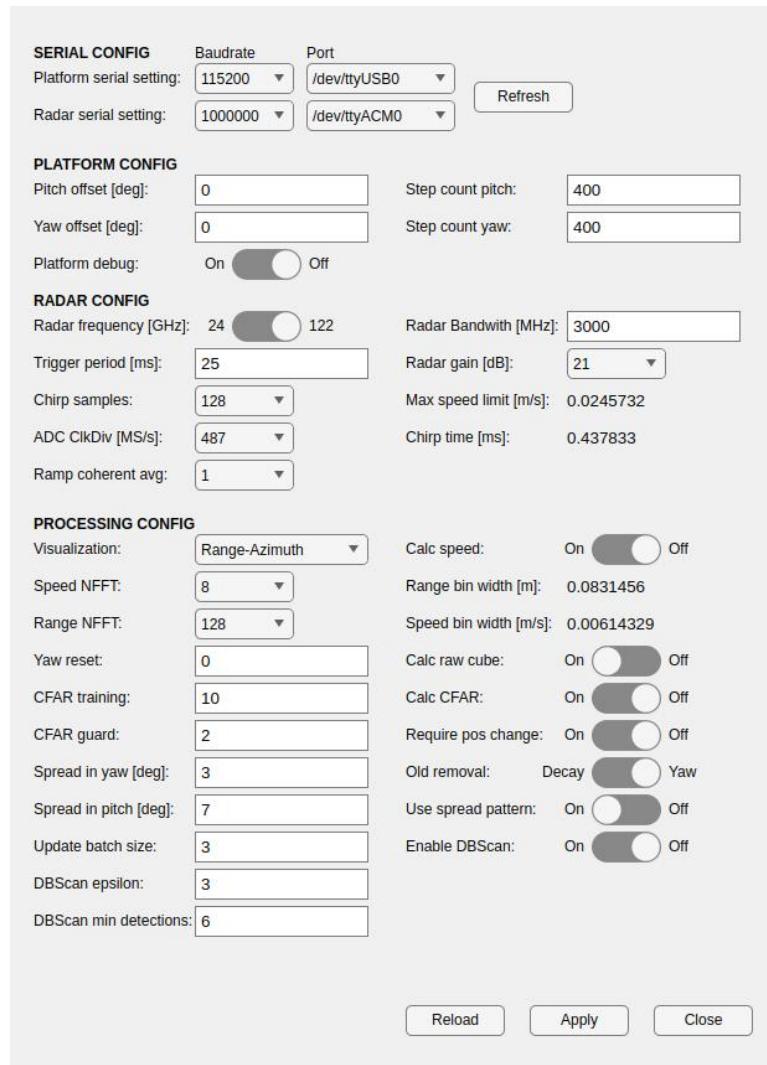


Figure 4.2: Preferences menu

User's preferences are stored in a INI file format in directory whose path depends on used operating system – %APPDATA%/Local/fmcw/fmcw.conf on Windows and \$HOME/.config/fmcw/fmcw.conf for unix based systems. As INI format is easily human readable and editable it is possible to edit the file directly.

## 4. Control application

### 4.2 Platform Control Interface

It is necessary to emphasize that the platform control is entirely independent of the radar operation. User loads up program to the platform, starts its execution and from then application is only processing current position reports coming from the platform. There is no feedback from the radar to the platform based on processed data, although as the platform enables direct access to the stepper queues it would be possible to implement such a feature. Nor does processing take into account from which movements is platform program comprised.

In preferences menu 4.2 user can configure static offsets to the platform with the **Pitch offset** and **Yaw offset** options. This is useful in case zero position of the platform is not aligned with what user want's to be zero in data. Aside from offset step count configuration is also available here with **Step count pitch** and **Step count yaw**. Step count takes into account both microstepping and physical down gearing ratio. It is important to note that as platform doesn't store step count in non volatile memory this setting is essential to maintain proper operation of the platform. The step count command is sent to the platform each time it is connected or config is changed. Lastly **Platform debug** toggle enables full debug output to be visible in platform control window.



Figure 4.3: GUI of the platform control

Managing platform programs is done via a separate window (Figure 4.3). This interface allows user to load, edit and send programs to the platform. Left sidebar allows user to load stored programs by clicking on their name. Middle text boxes are used to edit programs header (top part) and program body (middle part) with bottom text field being used to display platforms debug output. On the right side are then buttons that handle basic operations over program – load, save, store to file, upload and start.

Commands are written one per line in G-code like syntax. Control commands used to denote start of program, end of header and so on are automatically sent to the device when rest of the program is uploaded. To validate whether program was parsed correctly user can consult the text window at the bottom of the screen

## 4. Control application

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which forwards platforms debug output to the user. In case user clicks the store button the program is stored in same INI file as preferences.

Backend side of the platform control is rather straightforward. After connection is established MATLAB serial interface calls a callback each time terminator sequence is found in received data. Received message is then parsed with debug output being forwarded to the user and positions logged with timestamp in a circular buffer. Radar data processing part can then request a list of positions for given time interval.

Platform control can also emit an event when platform finds itself close to a predetermined position. This location is specified in option `Yaw reset` with tolerance zone fixed at two degress on either side. In orde to prevent multiple emission of the event the platform will only emit it once per 1 s. This event allows rest of the code to delete stored data and start gathering new one in next cycle.

### 4.3 Radar configuration

Given that the radar's output is displayed in the main window there isn't any specialized interface dedicated just to the radar. Still in preferences menu (figure 4.2) there are some paramters that can be configured.

`Radar frequency` toggle chooses between 24 GHz and 122 GHz header. However this only affect configuration sent to the radar, it doesn't reflash the firmware what is also necessary to do. After base frequency `Radar bandwidth` can also be configred, both positive and negative values are accepted however manual verification is required to verify that the radar is working as expected. While manufacturer states a 3 GHz limit on the 24 GHz header and 5 GHz on the 122 GHz header [12], in practice, especially with the 24 GHz header, achievable bandwidth was usually much lower.

User can also configure the number of samples per chirp via `Chirp samples` and ADC rate with `ADC ClkDiv`. These two paramters determine time of a single ramp which is calcualted with equation 2.1. Together with gain, ramp time plays a crutial role in maintaining correct signal to noise ratio.

Drop down menu `Ramp coherent avg` allows user to pick how many ramps will be averaged together. However use of this feature is generally not recommended unless surrounding environment is too cluttered (such as in a lab). Each ramps adds some 2 ms overhead to the radar's operation and makes speed calculation impossible.

Last paramters is `Trigger period` which dictates how often will the application send a trigger command to the radar. As previously stated the radar is at best capable of around 50 Hz update rate or 20 ms per chirp with slightly more stable performace at some 22 ms.

# 5. Radar Data Processing

Firstly it is necessary to outline what inputs and outputs of the whole processing pipeline are. Data to the applications comes from two sources - serial connections to the radar and to the platform. These two streams are then paired together based on timestamps logged when a data is received. Processing than does user chosen calculations – parts such as CFAR calculation, speed calculation and so on can be turned on or off. Output data are then stored in radar cube, common method to manage radar's data that enables easy integration of more complex algorithms in the future [10].

Given that both CFAR and raw data (That is individual range-Doppler maps.) calculation might need to be stored application has two cubes. One is a 4D structure with dimensions of fast time  $\times$  slow time  $\times$  yaw  $\times$  pitch, used to store complete range-Doppler map for each spatial degree. In this thesis this structure is only used for visualizing of raw data for given fixed pitch angle or showing range-Doppler map for given yaw and pitch. Any 3D visualization style of the raw data would be too cluttered and hard to interpret. Second cube is a 3D structure with dimensions of range  $\times$  yaw  $\times$  pitch, this one is used to store CFAR output.

In each case yaw resolution is fixed 1 degree thus the cube is  $360^\circ$  wide. Given quality of radar mounting bracket,  $1.8^\circ$  resolution of stepper and tightness of radiation pattern  $1^\circ$  degree was deemed sufficient. As for pitch same resolution was kept with angles ranging from  $-20^\circ$  to  $+80^\circ$  being covered. Lower bound is limited by the radar's mounting bracket and upper bound is more or less arbitrary, slightly guided by attempt to prevent radar from seeing behind itself.

Main complicating factor in the processing was the need to keep main thread responsive. Most time critical thing main thread does is timely readout of data at serial ports. If data can't be read right after its arrival assigned timestamps will be incorrect and visualization will be wrong. In addition GUI starts to be quite unresponsive if the resources of main thread are stretched too thin.

To solve this problem processing heavily leverages MATLAB's parallel computing toolbox, offloading as much as possible to different cores. Thorough whole processing pipeline there are mechanism of buffering data into larger batches and if needed discarding some in case software is unable to keep up. Unfortunately as MATLAB's parallelization capabilities with threads are rather limited – external language interface don't work and so is the case for certain file/memory mapping functions [28]. Thus the parallelization needs to be done with processes. This restrict the application to machines with at least 16 GB of RAM as each process (of which there are three) requires around 2 GB of RAM.

Processing pipeline can be divided into multiple distinct steps as follows:

1. Data acquisition: Data are read, simple FFT is calculated and data are buffered into a circular buffer.
2. Data processing: Data retrieved from the circular buffer paired with positions and processed.
3. Cube update: Processed data are buffered into a ping-pong buffer. In case buffer is full the cube update routine is launched.
4. Visualization: After cubes are updated data are visualized in user chosen format.

### 5.1 Data acquisition

Basic workflow of this step is shown on figure 5.1. Action starts as soon as MATLAB detects line terminator within incoming data stream. After whole frame from the radar is read, the data are timestamped and I and Q channels buffered. Observer `dataProcessor.m` is then notified and left to retrieves the data when main thread is free.

After retrieval inphase and quadrature channels are merged to form a complex singla which is multiplied by Hamming window. Basic 1D FFT is then calculated with  $NFFT$  equal to `Range NFFT`, in case fft order is larger than the number of samples zero padding is added. Resulting spectrum is again buffered, as spectrums will be subjected to further processing they are kept in original complex form. Following this step program has multiple FFT spectrums computed – those can be leveraged to enable range-Doppler map computation.

Program then verifies if the platform has moved. This check enables to save processing resources by not computing R-D map that would be immediately rewritten. After movement code then takes all spectrums (Except the last one when platform changed position.) and launches another processing step in a separate thread. In case user wishes for processing to always take place (e.g. they are verifying radar settings when platform is static) this check can be skipped by toggling `Require pos change`.

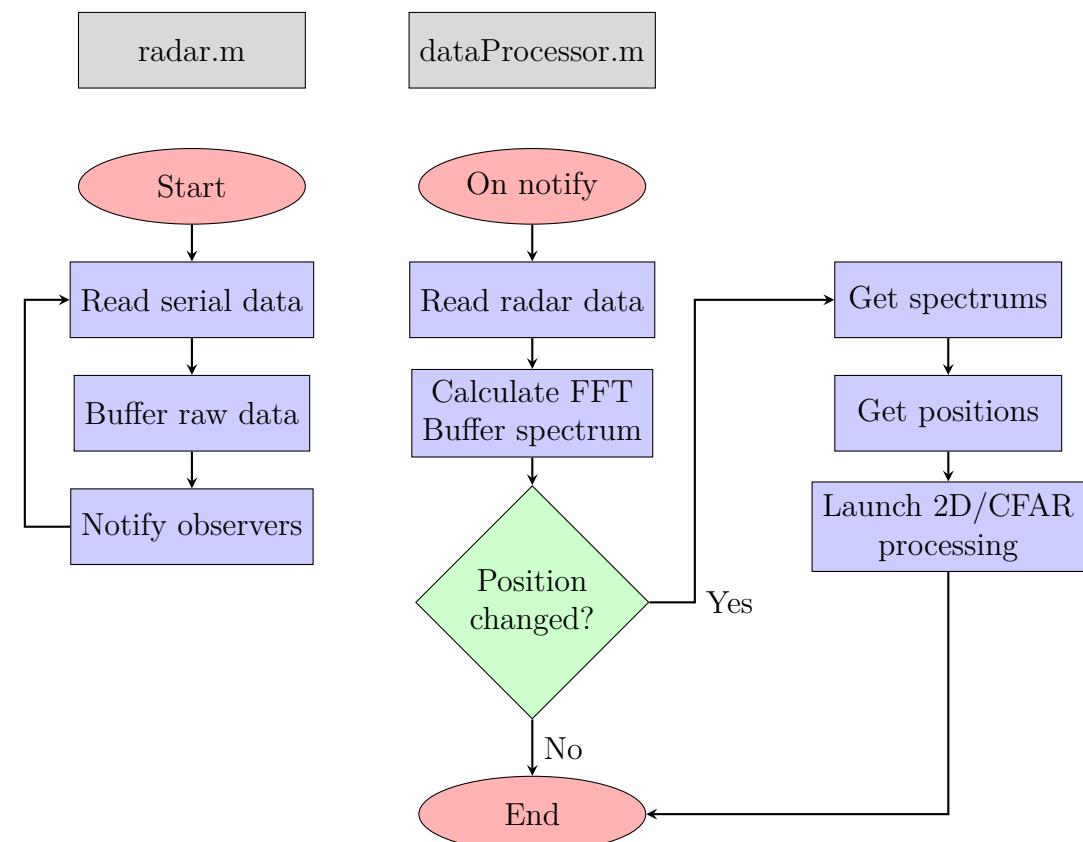


Figure 5.1: Processing flow - Data acquisition

### 5.2 Data processing

This parallel process takes calculated FFT spectrums and their accompanying positions and processes them into single range-Doppler map and CFAR array . Basic outline of the processing flow is shown in figure 5.2.

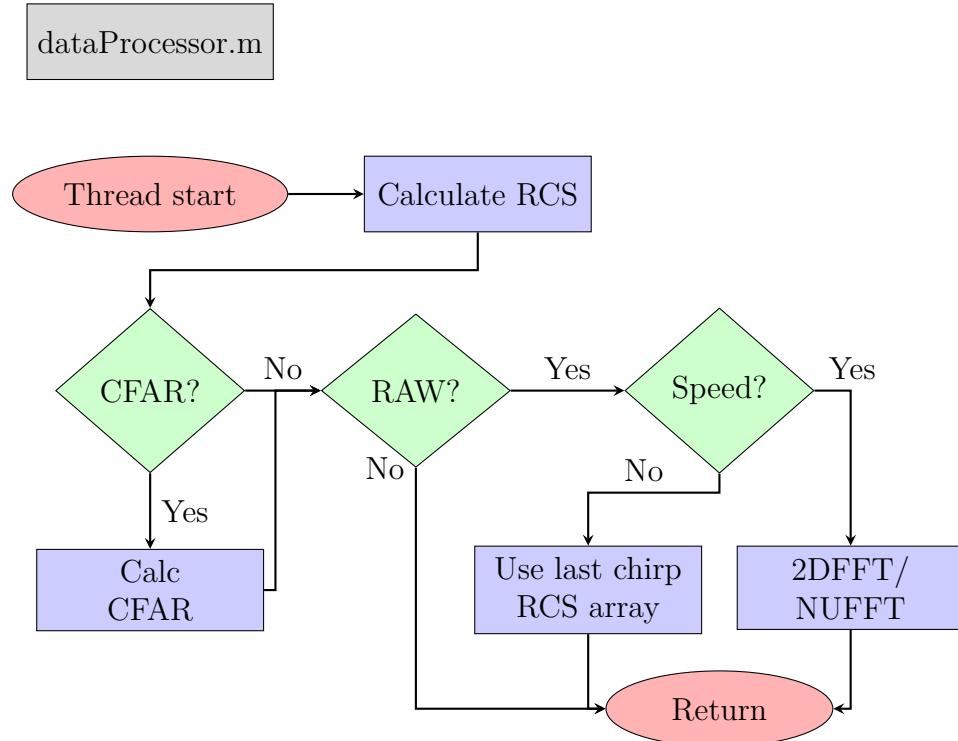


Figure 5.2: Processing flow - Data processing

In the first step last spectrum is converted to values proportional to the radar cross section (RCS) of the target. Starting with equation 1.19 we can see that the target's RCS is proportional to the received power divided by fourth power of the distance to the target. Range can be calculated from manufacturer's equation 2.2 and power is proportional to the square of values reported from the radar. Given that constants of the radar equation changed based on configuration calibration would be required in order to get real RCS values. However this calibration is not supported and thus onwards when referring to RCS we will be referring to the values proportional to the RCS and not real RCS of the object.

After RCS conversion if `Calc CFAR` is enabled CFAR will be runned on latest chirp in the batch. In this application simple 1D CA-CFAR from MATLAB's Phased Array System Toolbox is used which parametrized with two variables – number of guard bins (Those are excluded from the average and are right next to bin we are testing.) and number of training bins (Those are used to calculate the average.) [29]. Both these parameters are configurable from preferences menu with `CFAR training` and `CFAR guard`. As already stated, due to wide variation in bin widths for different bandwidth both parameters are left to be configured to the user.

If raw data calculation is enabled (Using `Calc raw cube` toggle.) the program takes one of two paths depending on whether speed calculation is toggled on or off with `Calc speed`. In case speed calculation is disabled the program just returns half of the original spectrum converted to RCS.

## 5. Radar Data Processing

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When speed calculation is on program firstly cuts of samples that are too far away from position of the latest chirp. After that timing of chirps is also analyzed choosing whether NUFFT or 2DFFT transformation should be used. Decision which transformation to use is based solely on the ratio of maximal deviation of the chirp intervals to the median of the chirp intervals – if this ratio is over 20 % NUFFT is used. For more accurate measurement more complex method should have been used but as the radar timing data is not very precise not to mention radar can track only very slow moving objects, this method was deemed sufficient. For similar reasons no correction for the rotation of the platform is done and the threshold for distance isn't user configurable. Output range-Doppler map is then shifted in order to get 2D array with  $\text{Range NFFT} \times \text{Speed NFFT}$  number of elements with speed going from  $-\frac{\text{Speed NFFT}}{2} \cdot w$  to  $(\frac{\text{Speed NFFT}}{2} - 1) \cdot w$ , where  $w$  is speed bin width calculated with (1.18).

### 5.3 Cube update

Last step of processing pipeline handles updates of radar cubes. Range-Doppler maps and CFAR data are firstly buffered in a ping-pong buffer – where one is being used to update the cube while the other is being filled with new data. Upon filling the buffer they switch roles and cube update routine is launched.

As matlab processes do not share same memory space by default in order not to have redundant copies of cube matlab's memory mapping is used. Using function `memmapfile` the cube is mapped to a file on the disk which can be shared between two processes [30]. This approach might lead to some performance degradation if the cube is too large and cannot be held in RAM but in testing it was found that for cubes of size 512MB (roughly cube for range-Doppler map of  $256 \times 16$ ) the program was still able to keep up with the radar.

Update routine for cubes containing either CFAR or raw data, are quite similar, in case spread pattern is disabled. New data are just added to the cube to their respective positions where they replace old ones. In case cube is set to decay whole cube is multiplied by decay factor that is derived from platforms speed of movement (The slower the platform moves the slower the decay). Using this option will lead to output more visually resembling radar output of traditional analogue radar systems where most recent data were brighter than the older ones. However user needs to keep in mind that color changes doesn't mean a change in RCS values at that position. In addition this operation is rather computationally expensive and is the primary reason updates to the cube must be buffered. Even on smaller cubes of around 256 MB this operations takes around 50 ms to complete on a laptop i7-10210U CPU. If change of value would be undesirable behaviour decay can be turned off in the settings in favour of hard reset of the cube based on trigger on specified yaw position.

Enabling spread pattern leads to degradation of quality of the data in exchange for better visualization quality – especially in range-azimuth visualization style. Instead of adding new data just to one position in the cube the range-Doppler map is spread over a larger area with a aid of 2D matrix, containing a 2D Gaussian function. As the data in series are usually close to each other in terms of position instead of directly adding these contributions to the full cube much smaller cube is created that covers only the area needed for the batch. This enables significant

## 5. Radar Data Processing

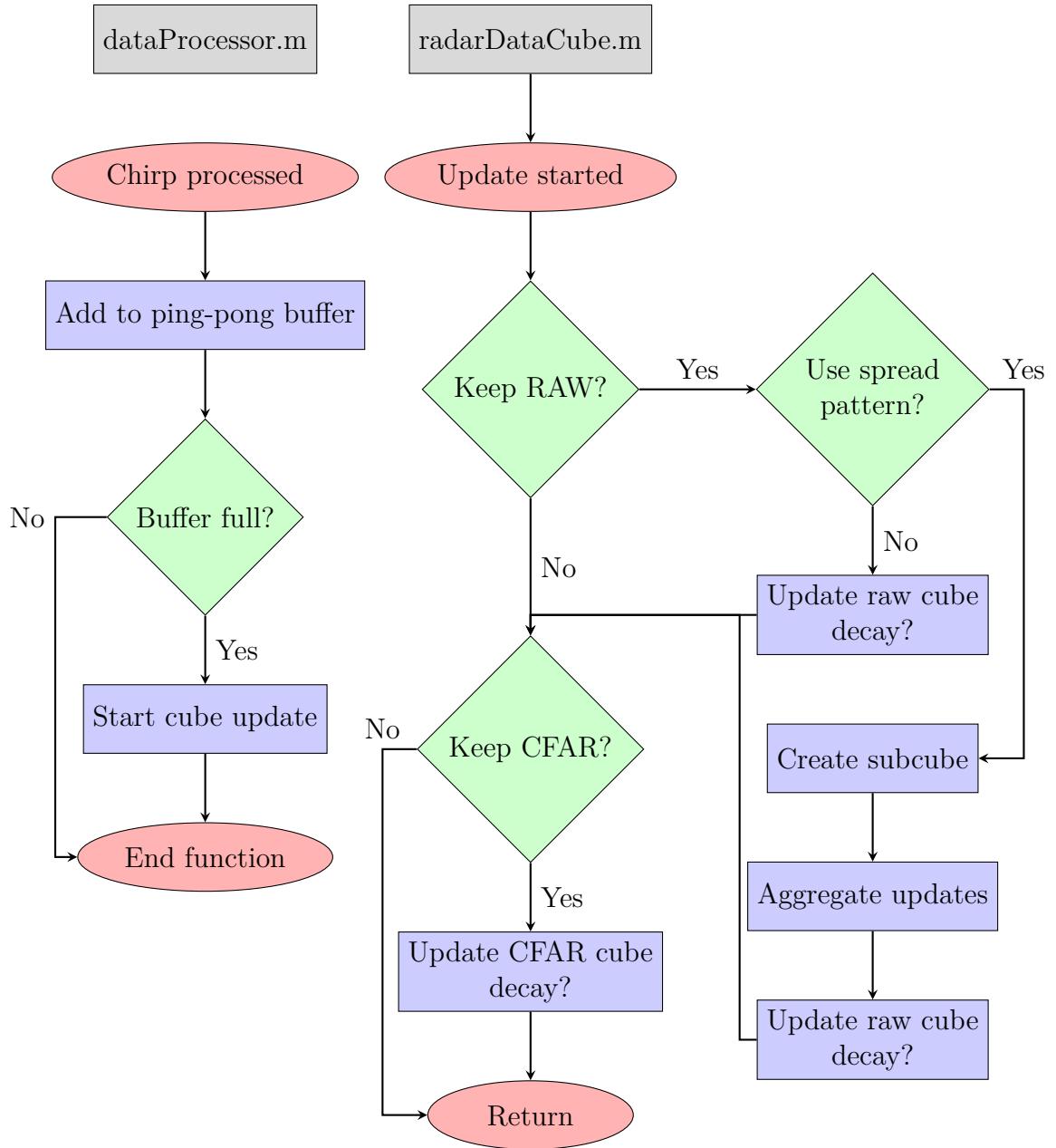


Figure 5.3: Processing flow - Cube update

speedup as smaller memory size enables better cacheing. After each update is aggregated to the subcube the original cube is update in one go.

Due to lacking performance of MATLAB on author's machine some parts of the cube update routine were implemented in C++ with the aid of MATLAB's MEX interface. This enabled the use of AVX2 x86 instruction set extension to speed up floating point operations. Chiefly decaying of the cube exhibited roughly two fold perforace increase thanks to SIMD instructions.

## 5.4 Visualization

After cube update finishes program returns to the main thread and the data are visualized. Other than graphical access to the data is not possible. However the application offers a number of different visualization styles to suit the user's

## 5. Radar Data Processing

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needs.

1. Range-Azimuth: Polar plot with range on the radius and azimuth on the angle.
2. Range-Doppler: Classical xy plot with range on x axis and Doppler on y axis, if speed is not calculated the y axis is replaced with RCS value.
3. Target-3D: CFAR data are displayed in 3D cartesian space

### 5.4.1 Range-Azimuth

This visualization style is used to display the data in polar plot for fixed pitch angle. Which pitch angle is used is chosen via the small textbox at the bottom left corner of the screen. The demonstration of Range-Azimuth map can be seen on figure 4.1, which captures testing with 122 GHz header with rather large bandwidth of 5 GHz giving only some 3 meters of range.

As for the form of the data instead of calculation projection into horizontal plane the data are displayed in a conical projection. That is no additional calculations are needed and the cone we get by fixing the pitch angle is simply stretched to a circle. To aid with orientation in a plot simple line markers are added to provide range and angle reference points.

It is necessary to point out that as 24 Ghz header radiation pattern in pitch is rather large at relatively low angles the radar provides information about a larger area of the scene. And if the user wishes using the spread patterns allows to range-azimuth map to include data even from different pitch angles.

### 5.4.2 Range-Doppler

Picking this visualization style from the preferences menu will display the data in a classical xy plot for a fixed yaw and pitch angle. Just as with Range-Azimuth map both angles are configurable via the textboxes at the bottom of the screen. In cases when speed calculation is disabled the y axis is replaced with RCS value.

Main purpose of this visualization style is to provide verification method for the radar data processing and setting. As the devkit doesn't report much of a useful errors and its technical documentation is sometimes incorrect (e.g. in allowed bandwidth values) it is useful to have a way to verify that the radar is working correctly.

On figure 5.4 we can see two different visualization styles both displaying the same static scene. As previously stated given rather poor sampling rate speed calculation is not very useful and thus no complex testing has taken.

## 5. Radar Data Processing

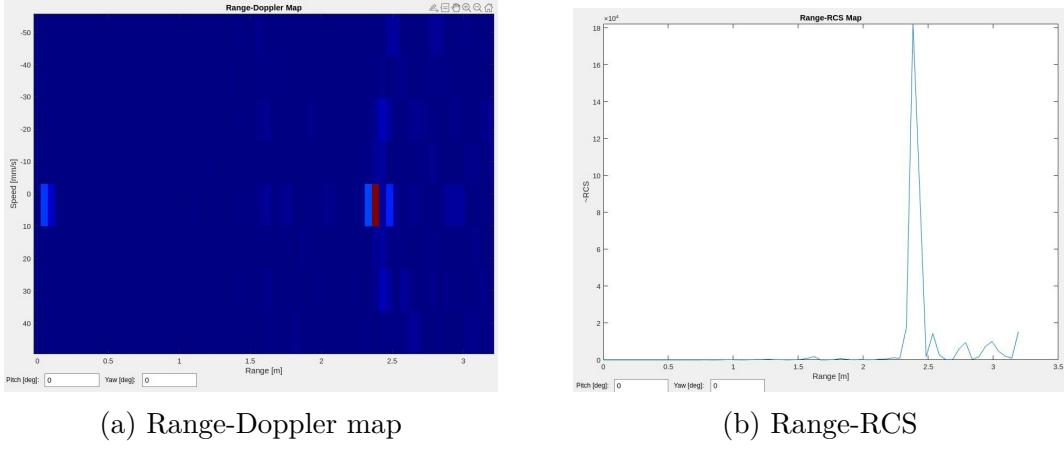


Figure 5.4: Range-Doppler and Range-RCS maps

### 5.4.3 Target-3D

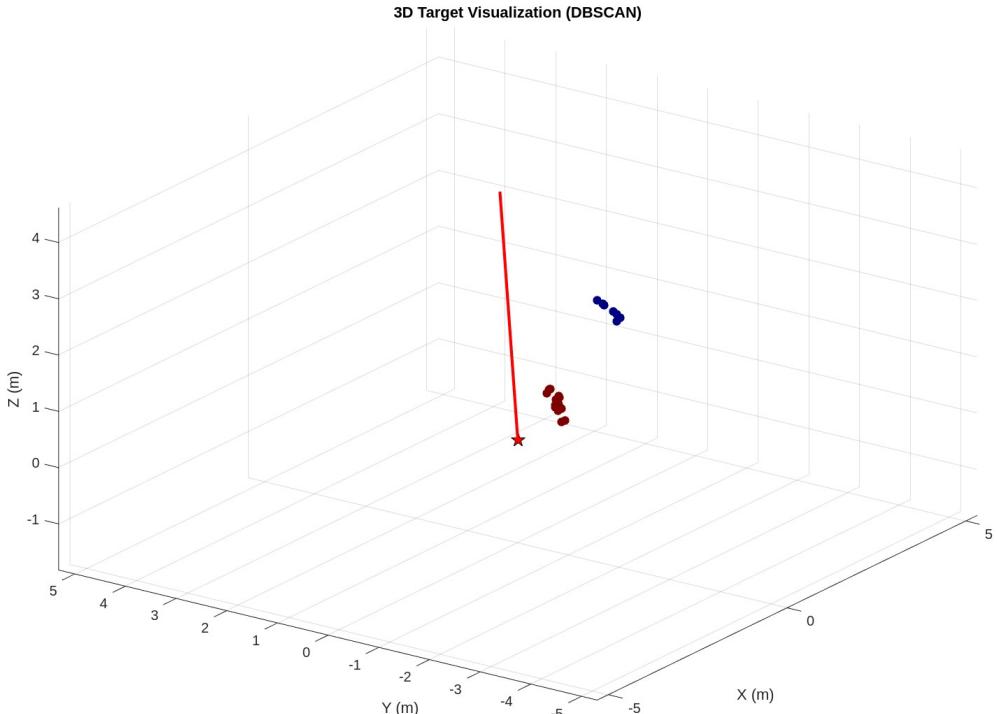


Figure 5.5: 3D Space visualization method

This visualization style is used to display CFAR data in 3D cartesian space (Figure 5.5). Visualization method is rather simple with just positions being recalculated from the range, azimuth and pitch angles to x, y and z coordinates using trigonometric functions. In case decay is enabled then small threshold is used to remove points whose values was sufficiently regraded. To aid with orientation within the plot current yaw and pitch of the platform at time of the update is also displayed as a red line;

If environment is too cluttered and CFAR generates too many points the user can enable basic clustering with matlab's built in DBSCAN function [31]. General idea of DBSCAN is that to a given point in the dataset distances to other points are calculated and compared with user defined threshold  $\varepsilon$ , in case

## 5. Radar Data Processing

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sufficient number  $N$  of points is found within the threshold the point is marked as a cluster centers [32]. In MATLAB's in algorithm distance threshold and minimal number of points are both configurable variables with other input being just the points list [31]. As original data are in polar coordinates system problem arises with calculating distances given that in such system distance between two points of given fixed yaw and pitch varies widely with range. Aside from that we also have significantly more points closer to the radar as space is much more densely scanned there compared to the far end of the range. Both these factors lead to problems with choosing both  $N$  and  $\varepsilon$  parameters.

For accurate results these problems can be overcome using many different methods like using adaptive thresholds in relation to radius [32] or relying on more clever axis systems than just polar or cartesian [33]. However as in this thesis as DBSCAN is more of an illustration of possible postprocessing that can be added to existing base no such complex methods were implemented. Both  $\varepsilon$  and  $N$  parameters are left to the user to configure. Distance between points is firstly converted to cartesian coordinates then metric is calculated using standard euclidean formula after which is later divided by average radius of the two points. Still it needs to be said that, while dividing by the mean of distances is sometimes used when working with spherical data [34], author is not aware of any literature that would suggest use in DBSCAN. Especially on longer distance this method will lead to significant errors.

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# Conclusion

Goal of this thesis was to realize a surveillance radar system based on FMCW technology. This technology should enable accurate distance measurements of targets with a relatively low power consumption. Instead of more conventional MIMO systems, a simpler solution was proposed with a single RX and TX antenna that relies on mechanical steering of the radar beam.

Using off-the-shelf components and 3D printed parts a custom rotary platform was designed and constructed. All be it minor issues in regards to the belt tension on pitch axes in enables controlling of the radar position in both yaw and pitch directions. Normally with  $1.8^\circ$  degree of accuracy that can be extended with the use of microstepping up to roughly  $0.1^\circ$ . Quality of life features such a automatic homing system or limits to the rotation were also implemented. Whole system is controlled by an ESP32C6 microcontroller which interprets G-code like commands and drives the stepper motors. Due to its similarities to other G-code base systems it should be readily adaptable to other systems. In addition the platform was designed to support capabilities that aren't strictly necessary for the purpose of this thesis.

Capabilities of SiRad Easy<sup>®</sup> evaluation board were analyzed and were found to be suitable for the purpose of this thesis. However not much room is left to improvement of the system as the board is limited by its low reporting rate of 50 Hz. In addition the radar module isn't able to maintain a constant frequency of updates. This limits the maximum detectable speed to tens of mm per second which effectively eliminates any possibility of tracking moving targets.

However it's ability to switch headers between 24 GHz and 122 GHz allows for a wide range of applications. Be it short range sensing at couple of meters or using 24 GHz header for up to hundreds of meters.

Control application for the surveillance radar was developed in MATLAB. It integrates both the rotary platform management and radar data processing into one combined package. The data processing pipeline is relatively standard relying on common techniques such as FFT, CFAR, DBSCAN and stores data in standard radar cubes. Also heavy parallelization was employed to ensure that the processing is done in a timely manner. This enables application to maintain speed with the radar even if costly operations, entailing millions of floating point operations, such as a decaying of the whole cube are performed.

Given rather generic design requirements large degree of customization is allowed to the processing pipeline. From simple things such as the number of FFT points, step count per rotation to more complex things such as the CFAR parametr can all be set to suit a specific application. One major downside of this approach is that the user is required to have a good understanding of the underlying algorithms and principles.

Only visual outputs were implemented in this thesis with both 3D and 2D visualization supported. Classical 2D Range-Azimuth map is probably the most useful and given the radar's radiation pattern provides information about sizable part of the environment. In case of 3D visualization only CFAR data are visualized in exchange for a more detailed view of the environment by providing pitch information. Were radar to be deployed in a more cluttered environment DBSCAN algorithm can be employed to filter out unwanted noise. However it's

## 5. CONCLUSION

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implementation in the thesis is rather illustrative and not optimized for performance or accuracy.

In addition the whole processing pipeline is written in a way that if radar module was exchanged for a faster one capabilities could be extended while keeping most of the codebase similar. More complex operations such as 2D FFT or cube updates are executed in parallel processes and are triggered by platform movement, not for each radar update.

In case number of FFT points both in speed or range would be significantly larger than what has been tested (Maximal cubes of around 500 MB were validated.) different approach to cube update would be required. First possibility would be leverages GPU acceleration to handle operations in the cube update. Specifically the decaying of the cube would be possible for much larger dimensions. In case of solely CPU based processing decay functions probably would be impractical for much larger cubes and the cube would need to be split into smaller chunks where only few ones would be loaded into RAM at one time – depending on the current position of the platform and direction of travel.

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# Appendix

## Appendix A – G-code glossary

Following sections acts as glossary of all G-code commands used in the platform control. More in-depth description of the commands can be found in the source code.

### Constants

- Parameters defined in Kconfig (require firmware recompilation):
- STEPPER\_Y\_STEP\_COUNT, STEPPER\_P\_STEP\_COUNT: Steps per rotation (modifiable via M92, non-persistent)
- STEPPER\_[Y/P]\_PIN\_DIR, STEPPER\_[Y/P]\_PIN\_STEP: GPIO pins for direction/step signals
- STEPPER\_[Y/P]\_PIN\_ENDSTOP: Endstop detection pins
- STEPPER\_MAX\_SPEED: Maximum RPM (hardware limit)
- STEPPER\_DEFAULT\_SPEED: Default motion RPM
- STEPPER\_MIN\_SPINDLE\_TIME: Minimum step interval of rotation in spindle mode
- STEPPER\_HAL\_TIMER\_PERIOD/RESOLUTION: PWM timer period/resolution, affects achievable RPM range

### Features

- **Units:** Switch between degrees and steps using G20/G21 respectively
- **Positioning Modes:**
  - Absolute (G90): Automatically normalizes angles to  $[0, 360^\circ]$  or  $[0, \text{STEP\_COUNT}]$
  - Relative (G91): Limited to  $\pm 32767$  steps per command, or respective angle counterpart
  - in absolute positioning devices moves in the shortest path to the target
- **Spindle:** continuous rotation mode
- **Limits:** Limits on rotation can be imposed on both axes
  - limits are applied only to one rotation – it is not possible to restrict to e.g. two full rotations in one direction and one in the other
  - `low < high`: Valid range  $[\text{low}, \text{high}]$
  - `low > high`: Valid range  $[\text{low}, 360^\circ] \cup [0, \text{high}]$
  - Out-of-range targets snap to nearest valid position
- **Programming:** G-code program can be uploaded to the device and executed
- **Synchronization:** If command is issued to both axes they will wait for each other, or not other axis will be free to execute next command

## 5. APPENDIX

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### Motion Control Commands

Axes are denoted as Y for yaw and P for pitch. The S parameter is used for speed for both axes, while SY and SP are used for speed of yaw and pitch respectively. In case of spindle Y or P are not followed by step count/angle but instead rotation direction.

**M80** Enable high-voltage power (enables drivers)

**M81** Disable high-voltage power (disables drivers)

**M82** Emergency stop (clears command queues)

**G20** Set units to degrees

**G21** Set units to steps

**G90 [Y/P]** Set absolute positioning Mode

- absolute positioning is not available in spindle mode, if you issue a M03 command to stepper in absolute positioning mode, it will be automatically switched to relative positioning and throw an error
- if no argument is provided both axis are switched to relative positioning

**G91 [Y/P]** Set relative positioning Mode

**G28 [Y/P]** Auto-home specified axis(es)

**G92 [Y/P]** Set current position as zero

**G0 Y<val> P<val> [SY/SP<rpm>]** Stepper-mode movement

**M03 Y/P<+/-> [SY/SP<rpm>]** Start spindle mode (continuous rotation)

**M05 [Y/P]** Stop spindle mode

**M201 LY/HY/LP/HP<angle>** Set angular limits (degrees/steps)

**M202 [Y/P]** Disable limits

### Special Commands

These commands bypass certain sections and safety checks of the firmware. Use with caution.

**G3 Y/P<steps> [SY/SP<rpm>]** Bypass scheduler and adds movement directly to the queue

- limits are not checked and cannot be checked
- absolute positioning is not available
- all values are interpreted as steps regardless of current unit setting
- *Warning:* Use M82 before G3 to maintain position integrity

**W3 T<ms>** Application-layer delay

- useful to wait for end of homing process

### Programming Commands

**P90 <id>** Start programming mode (overwrites existing program if any)

**P91** Transition from header to main program body

**P92** Finalize program (discarded if loops unclosed)

**P21 I<iterations>** For-loop declaration

**P22** End loop block

**P29** Infinite loop marker (header only) - main body will be executed infinitely

**W0 Y/P<sec>** Wait in seconds

**W1 Y/P<ms>** Wait in milliseconds

## 5. APPENDIX

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### Uplink Protocol

- !P <timestamp>, <yaw>, <pitch>: Position update (20ms interval)
- !R OK: Command acknowledgment
- !R ERR <code>: Command error

Code	Description
1	Malformed command syntax
2	Invalid arguments
3	Queue lock failure
4	Unsupported command
5	System busy (homing/program running)
6	Runtime exception
7	Unclosed loop in program
8	Invalid context

### Example Program

Command	Mode	Purpose
P90 prog	Header	Initialize program ID=1
G91	Header	Set relative positioning
P29	Header	Enable infinite looping
M03 SY6 Y+	Body	Start Yaw spindle (6 RPM)
G0 SP30 P100	Body	Pitch movement
G0 SP30 P-100	Body	Return pitch
P92	Body	Finalize program