CE394M: Linear Elasticity

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Overview

Linear Elasticity

Soil engineering properties

References

Isotropic linear elastic stress-strain relations

The relationship between the stress and strain tensor is a linear one. The stress component is a linear combination of the strain tensor. The most general form for *linear* stress-strain relations for a *Cauchy elastic* material is given by:

$$\sigma_{ij} = \sigma_{ij}^0 + D_{ijkl}\varepsilon_{kl}$$

Where σ_{ij}^0 is the components of initial stress tensor corresponding to the initial strain free (when all strain components $\varepsilon_{kl} = 0$). D_{ijkl} is the tensor of material *elastic constants*.

If it is assumed that the initial strain free state corresponds to an *initial* stress free state, that is $\sigma_{ij}^0 = 0$, the equations reduces to:

$$\sigma_{ij} = D_{ijkl} \varepsilon_{kl}$$

Observation on linear elasticity

- ① $\sigma_{ij} = D_{ijkl} \varepsilon_{kl}$ is a general expression relating stress to strains for a linear solid.
- ② D_{ijkl} is a 4th order tensor containing 81 terms (we trick using symmetry and reduce order).
- **1** D_{ijkl} material response functions having dimensions F/L^2 .
- \bigcirc Homogeneous: D_{ijkl} independent of position
- **1** Isotropic: D_{ijkl} independent of frame of reference.
- **3** Because the stress is symmetric: $\sigma_{ij} = \sigma_{ji}$, $D_{ijkl} = D_{jikl}$. Strain is symmetric $\varepsilon_{kl} = \varepsilon_{lk}$ and $D_{ijkl} = D_{ijlk}$. Hence the number of independent variables drop from 81 to 36.
- Both the stress and the strain tensor have only 6 independent values, therefor write them as vectors, then the stiffness tensor can be written as a matrix (compromise I can not rotate tensor).

Stress-strain relationship

$$\begin{bmatrix} \sigma_{11} \\ \sigma_{22} \\ \sigma_{33} \\ \sigma_{12} \\ \sigma_{23} \\ \sigma_{31} \end{bmatrix} = \begin{bmatrix} D_{11} & D_{12} & D_{13} & D_{14} & D_{15} & D_{16} \\ D_{21} & D_{22} & D_{23} & D_{24} & D_{25} & D_{26} \\ D_{31} & D_{32} & D_{33} & D_{34} & D_{35} & D_{36} \\ D_{41} & D_{42} & D_{43} & D_{44} & D_{45} & D_{46} \\ D_{51} & D_{52} & D_{53} & D_{54} & D_{55} & D_{56} \\ D_{61} & D_{62} & D_{63} & D_{64} & D_{65} & D_{66} \end{bmatrix} \begin{bmatrix} \varepsilon_{11} \\ \varepsilon_{22} \\ \varepsilon_{33} \\ \varepsilon_{12} \\ \varepsilon_{23} \\ \varepsilon_{31} \end{bmatrix}$$

 D_{ijkl} is a tensor of material *elastic constants*. However, the above [**D**] is not a tensor anymore. So we can not rotate the matrix to another frame of reference. This relationship is useful for isotropic materials, where **D** is independent of the frame of reference.

$$\{\sigma\} = [\mathbf{D}] \{\varepsilon\}$$

The inverse of the relationship (Compliance matrix):

$$\{\varepsilon\} = [\mathbf{C}] \{\sigma\} \qquad [\mathbf{C}] = [\mathbf{D}]^{-1}$$

Hooke's law

Empirical observation:

$$\Delta\varepsilon_{\textit{a}} = \Delta\sigma_{\textit{axial}} \cdot \frac{1}{\textit{E}} \rightarrow \Delta\varepsilon_{11} = \frac{\Delta\sigma_{11}}{\textit{E}}$$

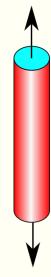
Where E is defined as the *Young's modulus*.

The lateral strains are defined as:

$$\Delta \varepsilon_{22} = -\nu \Delta \varepsilon_{11}$$
$$\Delta \varepsilon_{33} = -\nu \Delta \varepsilon_{11}$$

Using superposition for principal stresses:

$$\varepsilon_{11} = (1/E) \left[\sigma_{11} - \nu \sigma_{22} - \nu \sigma_{33} \right]
\varepsilon_{22} = (1/E) \left[-\nu \sigma_{11} + \sigma_{22} - \nu \sigma_{33} \right]
\varepsilon_{33} = (1/E) \left[-\nu \sigma_{11} - \nu \sigma_{22} + \sigma_{33} \right]$$



Hooke's law

$$\begin{cases} \varepsilon_{11} \\ \varepsilon_{22} \\ \varepsilon_{33} \end{cases} = \frac{1}{E} \begin{bmatrix} 1 & -\nu & -\nu \\ -\nu & 1 & -\nu \\ -\nu & -\nu & 1 \end{bmatrix} \begin{cases} \sigma_{11} \\ \sigma_{22} \\ \sigma_{33} \end{cases}$$

It is possible to invert the matrix to obtain the generalized Hooke's law: $[\sigma] = [\mathbf{D}][\varepsilon]$.

Where $\alpha = E/((1+\nu)(1-2\nu))$. Similarly, we can obtain the inverse matrix.

Hooke's law

The matrices [**C**] and [**D**] contains two indepdent variables E and μ , where E>0 and $-1 \le \nu \le 0.5$. The matrix can also be defined in terms of Lame's constants.

$$\begin{cases} \lambda = \frac{E\nu}{(1+\nu)(1-2\nu)} & \text{Lame's modulus (wave propagation)} \\ \mu = G = \frac{E}{2(1+\nu)} & \text{Shear modulus (shear behavior)} \\ K = \frac{E}{3(1-2\nu)} & \text{Bulk modulus (volumetric behavior)} \end{cases}$$

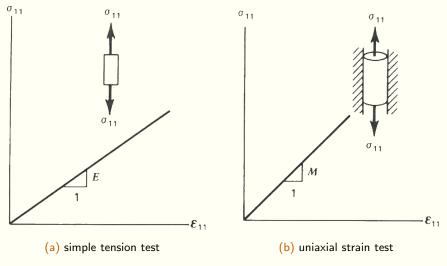


Fig. Behavior of isotropic linear elastic material in simple tests

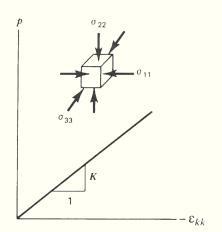
Simple tension test The only non-zero components of stress: $\sigma_{11} = \sigma$ The *Young's modulus, E,* and *Poisson's ratio,* ν as.

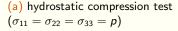
$$E = \frac{\sigma_{11}}{\varepsilon_{11}}$$
 $\nu = \frac{-\varepsilon_{22}}{\varepsilon_{11}} = \frac{-\varepsilon_{33}}{\varepsilon_{11}}$

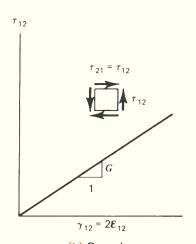
Uniaxial strain test The test is carried out by applying a uniaxial stress component σ_{11} in the axial direction of a cylindrical sample, whose lateral surface is restrained against lateral movement (Oedometer test). Axial strain ε_{11} is the only nonvanishing component. The constrained modulus M or as PLAXIS calls it E_{oed} is defined as the ratio between σ_{11} and ε_{11} .

$$\sigma_{11} = \frac{E}{(1+\nu)(1-2\nu)} \left[(1-\nu)\varepsilon_{11} + \nu \varepsilon_{22} + \nu \varepsilon_{33} \right]^{0} = \frac{E(1-\nu)\varepsilon_{11}}{(1+\nu)(1-2\nu)}$$

$$M = E_{oed} = \frac{\sigma_{11}}{\varepsilon_{11}} = \frac{(1-\nu)E}{(1+\nu)(1-2\nu)} = (\lambda + 2\mu)$$







(b) Pure shear test

Fig. Behavior of isotropic linear elastic material in simple tests

Hydrostatic compression test The non-zero components of stress:

$$\sigma_{11} = \sigma_{22} = \sigma_{33} = -p = \sigma_{kk}/3.$$

The Bulk modulus, K, is defined as the ratio between the hydrostatic pressure p nd the corresponding volume change $\delta \varepsilon_{v} = \varepsilon_{kk}$.

$$K = -\frac{p}{\varepsilon_{kk}} = \lambda + \frac{2}{3}\mu$$

Simple shear test

The non-zero components of stress: $\sigma_{12} = \sigma_{21} = \tau_{12} = \tau_{21} = \tau$. The *Shear modulus*, *G* or μ , is defined as:

$$G = \mu = \frac{\sigma_{12}}{\gamma_{12}} = \frac{\tau}{2\varepsilon_{12}}$$

Plane stress v Plane strain

For a frame with an axis perpendicular to the plane of interest, x_3 or z:

Plane stress $\sigma_{33} = \sigma_{zz} = \tau_{xz} = \tau_{yz} = 0$.

The strain in z is written as:

$$arepsilon_{
m zz} = rac{-
u}{E} (\sigma_{
m xx} + \sigma_{
m yy}) = rac{-
u}{1-
u} (arepsilon_{
m xx} + arepsilon_{
m yy})$$

The plane stress are commonly used for thin flat plates loaded in the plane of the plate.

Plane strain $\varepsilon_{33} = \varepsilon_{zz} = \gamma_{xz} = \gamma_{yz} = 0$.

The stress in z is written as:

$$\sigma_{zz} = \nu (\sigma_{xx} + \sigma_{yy})$$

The plane strains are commonly used for elongated bodies of uniform cross sections subjected to uniform loading along the longitudinal axis (tunnels, dams, retaining walls, soil slopes, etc.).

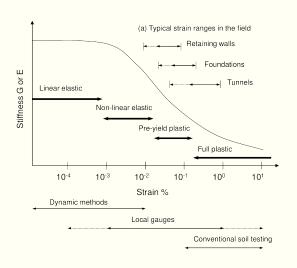
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Stiffness: small to large strains



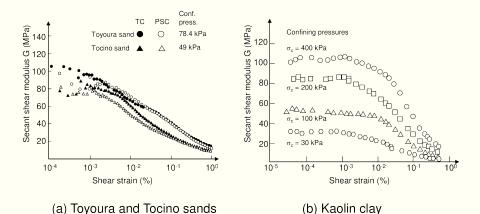


Local gauges





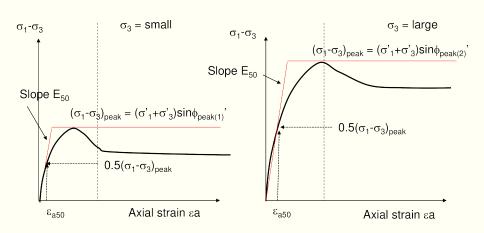
Stiffness: small to large strains



E or G is a function of:

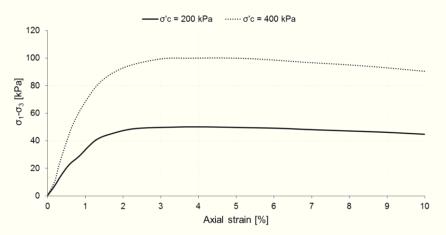
- confining stress and void ratio (relative density)
- strain level

Stiffness at intermediate strains



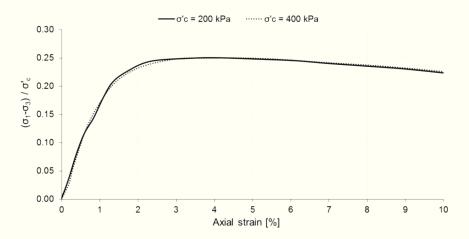
$$E_{50}=E$$
 at 50% of $(\sigma_1-\sigma_3)_{peak}$ Note: $\phi'_{peak1}<\phi'_{peak2}$ $E_{50ref}=(0.70\pm0.10)E_{oedref}$

Undrained strength



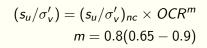
Triaxial compression test data of homogeneous clay (Ladd & Foott, 1974)

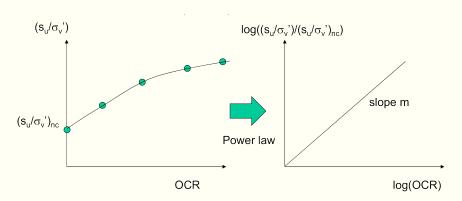
Normalized undrained strength



Normalised triaxial compression test data of homogeneous clay (Ladd & Foott, 1974)

Stress History and Normalized Soil Engineering Properties (SHANSEP) (Ladd and Foote, 1974)





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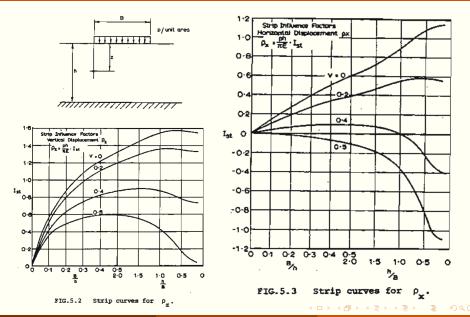
—Stress History and Normalized Soil Engineering Properties (SHANSEP) (Ladd and Foote, 1974)



Laboratory tests conducted at the Imperial College using remolded clays (Henkel (1960) and Parry (1960)) and at the Massachusetts Institute of Technology on a wide range of clays, give evidence that clay samples with the same over-consolidation ratio (OCR), but different consolidation stress σ_c' and therefore different pre-consolidation stress σ_p' , exhibit very similar strength and stress-strain characteristics when the results are normalized over the consolidation stress σ_c' .

In practice, normalized behaviour is not as perfect as shown, there is discrepancy in the normalized plots caused by different consolidation stresses, soil deposit heterogeneity or even the fact that the conditions from one soil test to another are not identical. However, this discrepancy is reported to be quite small.

Elastic solution for settlement under foundations



Elastic solutions

- **1** Ease of use Only two parameters, choose equivalent values representative of stran/stress levels.
- 2 Disadvantage: No failure criteria.
- Validate code with chart solutions ("Exact solutions")., e.g., Poulos and Davis (1974).
- Useful to get feeling of problem (low stress levels) not wide distribution of plastic zones.

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Isotropic Linear Elastic Stress-strain relationship

The isotropic tensor D_{ijkl} :

$$D_{ijkl} = \lambda \delta_{ij} \delta_{kl} + \mu (\delta_{ik} \delta_{jl} + \delta_{il} \delta_{jk}) + \alpha (\delta_{ik} \delta_{jl} - \delta_{il} \delta_{jk})$$

Where λ, μ , and α are scalar constants. Since D_{ijkl} must satisfy symmetry, $\alpha=0$.

$$D_{ijkl} = \lambda \delta_{ij} \delta_{kl} + \mu (\delta_{ik} \delta_{jl} + \delta_{il} \delta_{jk})$$

So the stress:

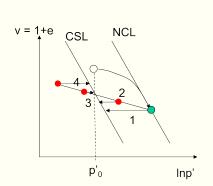
$$\sigma_{ij} = \lambda \delta_{ij} \delta_{kl} \varepsilon_{kl} + \mu (\delta_{ik} \delta_{jl} + \delta_{il} \delta_{jk}) \varepsilon_{kl}$$

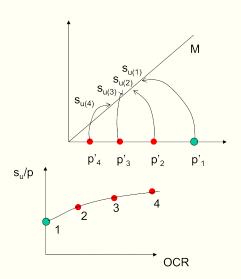
$$\sigma_{ij} = \lambda \varepsilon_{kk} \delta_{ij} + 2\mu \varepsilon_{ij}$$

Hence for an isotropic linear elastic material, there are only two independent material constants, λ and μ , which are called *Lame's constants*.

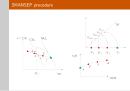
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SHANSEP procedure





SHANSEP procedure



- 1. obtain a sample of "clayey" soil
- 2. Reconsolidate to 1.5 to 2.0 of insitu effective mean pressure p_0' , or reconsolidate back to the virgin normal compression line
- 3. Rebound to desired OCR
- 4. Perform undrained shearing and find s_u .
- 5. Perform multiple tests at different OCRs.
- 6. Evaluate $(s_u/\sigma'_v) = (s_u/\sigma'_v)_{nc} \times OCR^m$