Title:- Thermal & Fallback Resilience of Autonomous DysonNode Power Units During Solar Flare Events.

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Scientific context and Problem statement:-

Dyson Node_00001 is a space based Autonomous Energy Absorbing System operating in Close Stellar Orbit. It's a part of Theoretical Dyson Swarm, Absorbing Solar flux for Conversion into Usable Power. The Key Environmental Hazard Simulated here is a Solar Surge--

a Short Term, High Energy Radiation spikes Caused by;

- \star Solar Flares (especially X- Class).
- ★ Coronal Mass Ejections (CMEs).
- ★ Fluctuating Plasma Ejection Zones
- ★ Starspot Oscillations

In the Context of Such Surges, Power Modules face Rapid Energy inflow, leading to;

- ★ Internal Thermal Build-up
- ★ Systematic Stress on Capacitors
- ★ High Risk of Burnout if fallback logic

Objective of the Scenarios:- (i) Solar radiation spikes over time using a Time-series Sinusoidal Fluctuation Model.

- (ii) Thermal Response of Internal Systems due to Excess Flux.
- (iii) Fallback Triggering System that protects the node when Safety Thresholds are Breached.
- (iv) Generates Logs that could later be Consumed by Diagnostic AI Modules.

Solar Irradiance and FLux Modeling:-

- (A) Solar constant
- → Earth Receives a constant Average Solar FLux of 1361 W/m^2.
- (B) Flare Induced Surge

 \Rightarrow Solar flares can Temporarily Increases Localized FLux by +2% to +50%, depending on Orbital Proximity.

F(t)=1361+ $A \cdot sin(i)$., $i = energy \ cycle \ ID \ (simulation \ step)$ $A = 2.0 \ represents \ amplitude \ of \ sinusoidal \ surge$

(~+0.14%)

This simulates a Non-Linear, Cyclic Solar Spike, similar to Electromagnetic Oscillation Pattern observed during Real Flare Event.

FallBack Condition:- When;

 $T > 78.5 \circ C \Rightarrow System Fallback Activated$

This simulates the thermal cutoff threshold built into real-world spacecraft power systems (similar to thermal fuses in satellites).

References:-

Kopp, G., & Lean, J. L. (2011). A new, lower value of total solar irradiance: Evidence and climate significance. Geophysical Research Letters, 38(1).

⇒ (Establishes 1361 W/m² as the modern solar constant)

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→ (Surge events and solar flux enhancement during flares)

Gilmore, D. G. (2002). Spacecraft Thermal Control Handbook, Volume I. The Aerospace Press.

→ (Standard temperature limits & fallback systems in real satellites)

ECSS-E-ST-31C (2008). Thermal Control – European Cooperation for Space Standardization.

» (Fallback thresholds and emergency logic)

J. Wertz & W. Larson. (1999). Space Mission Analysis and Design. Microcosm Press.

» (Structure for satellite systems analysis + scenario testing)

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Ogata, K. (2010). Modern Control Engineering. Prentice Hall.

» (Fallback triggers and stability logic)

Sidi, M. J. (1997). Spacecraft Dynamics and Control: A Practical Engineering Approach. Cambridge University Press.

» (Thermal and radiative damping modeling)

NASA Parker Solar Probe Mission Data Sheets https://www.nasa.gov/content/goddard/parker-solar-probe » (Thermal systems operating under 20x Earth flux)

James Webb Space Telescope Thermal Control Systems

https://webb.nasa.gov/content/observatory/thermalControl.html

» (Passive + active temp regulation + shielding from flares)

Authorship & Originality Note:- This simulation model, fallback logic, and code structure are original contributions by Heet Trivedi, independently developed and self-published under the Project SOLIS framework.

Referenced works are cited to support external constants, scientific models, and real-world analogs used for simulation accuracy.