$\ensuremath{\mathsf{TTK4190}}$ - Assignment 3

Group 27

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1 Part 2: Heading Autopilot

1.1 Problem 1 - Environmental Disturbances

1.1.1 Problem 1.a

With $\beta_{V_c} = 45^{\circ}$ and $V_c = 1$, and from using the following equations from [1]

$$u_c = V_c cos(\beta_{V_c} - \psi)$$
$$v_c = V_c sin(\beta_{V_c} - \psi)$$

```
for i=1:Ns+1
125
126
        t = (i-1) * h;
                                                % time (s)
127
        R = Rzyx(0,0,eta(3));
128
        % current (should be added here)
130
        nu_c(1) = Vc*cos(betaVc - eta(3));
                                               % surge current
131
        nu_c(2) = Vc*sin(betaVc - eta(3));
132
133
        nu_r = nu - nu_c;
```

Listing 1: MATLAB code for 2D irrotational ocean current in surge and sway

and the resulting plots become

1.1.2 Problem 1.b

With zero ocean currents (and wind) affecting the vessel, there will be no crab or sideslip angle. This will result in the vessel following a straight path.

1.1.3 Problem 1.c

The angle of attack between the wind and the boat is given as

$$\gamma_w = \psi - \beta_{V_w} - \pi \tag{1}$$

The wind-coefficients are approximated as in [1] in sway and yaw direction as

$$C_Y(\gamma_w) \approx c_u \sin(\gamma_w)$$
 (2)

$$C_N(\gamma_w) \approx c_n \sin(2\gamma_w) \tag{3}$$

The moment

$$\tau_{wind} = \begin{bmatrix} \tau_{Y_{wind}} \\ \tau_{N_{wind}} \end{bmatrix} = \begin{bmatrix} \frac{\rho_a V_{ra}^2}{2} C_Y(\gamma_{rw}) A_{L_w} \\ \frac{\rho_a V_{ra}^2}{2} C_N(\gamma_{rw}) A_{L_w} L_{oa} \end{bmatrix}$$
(4)

```
137
        if t > 200
            % UNSURE ABOUT RELATIVE VS ABSOLUTE WIND SPEED AND ANGLE-OF-ATTACK
138
            gamma_w = eta(3) - betaVw - pi;
139
            C_Ywind = cy*sin(gamma_w);
                                                      % sway wind coefficient
140
                                                      % yaw wind coefficient
            C_Nwind = cn*sin(2*gamma_w);
141
            Ywind = 1/2*rho_a*Vw^2*C_Ywind*A_Lw;
142
                                                     % expression for wind moment in ...
                sway should be added.
            Nwind = 1/2*rho_a*Vw^2*C_Nwind*A_Lw*L; % expression for wind moment in ...
143
                yaw should be added.
144
145
            Ywind = 0;
            Nwind = 0;
146
147
        tau_env = [0 Ywind Nwind]';
```

Listing 2: Wind movement in sway and yaw.

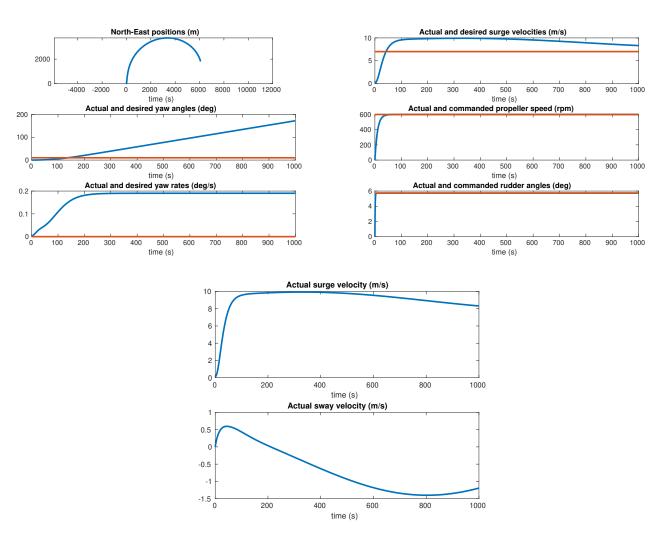


Figure 1: Boat behaviour with current disturbance

1.2 Problem 2 - Heading Autopilot

1.2.1 Problem 2.a

From the given matlab code we get that

$$\mathbf{C}_{A} = \begin{bmatrix} 0 & 0 & Y_{\dot{v}}v + Y_{\dot{r}}r \\ 0 & 0 & -X_{\dot{u}}u \\ -Y_{\dot{v}}v - Y_{\dot{r}}r & X_{\dot{u}}u & 0 \end{bmatrix}$$
 (5)

$$C_{RB} = \begin{bmatrix} 0 & -mr & -mx_g r \\ mr & 0 & 0 \\ mx_g r & 0 & 0 \end{bmatrix}$$
 (6)

and the nonlinear coriolis forces become

$$\mathbf{C}_{A}(\nu)\nu = \begin{bmatrix} Y_{\dot{v}}vr + Y_{\dot{r}}r^{2} \\ -X_{\dot{u}}ur \\ -Y_{\dot{v}}vu - Y_{\dot{r}}ru + X_{\dot{u}}uv \end{bmatrix}$$
(7)

$$C_{RB}(\nu)\nu = \begin{bmatrix} -mrv - mx_g r^2 \\ mru \\ mx_g ru \end{bmatrix}$$
 (8)

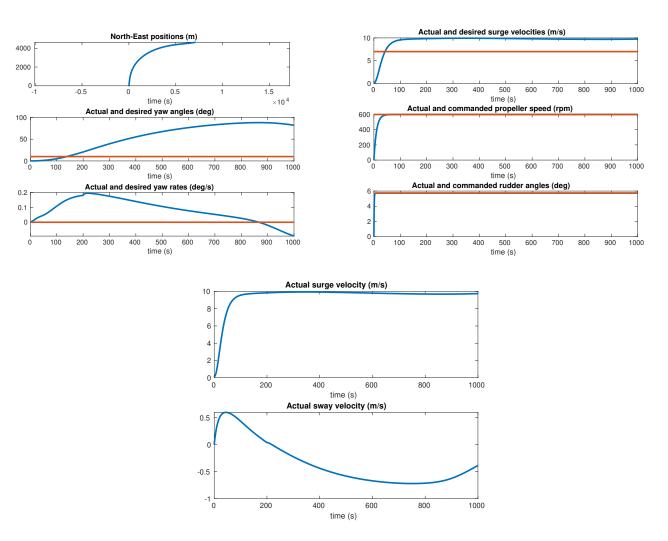


Figure 2: Boat behaviour with current disturbance and wind disturbance after 200 seconds

Linearized at $\nu = \nu^*$ we get

$$\mathbf{C}_{A}^{*} = \begin{bmatrix} 0 & 0 & 0 \\ 0 & 0 & -X_{\dot{u}}U_{d} \\ 0 & -Y_{\dot{v}}U_{d} + X_{\dot{u}}U_{d} & -Y_{\dot{r}}U_{d} \end{bmatrix}$$
(9)

$$C_{RB}^* = \begin{bmatrix} 0 & 0 & 0 \\ 0 & 0 & mU_d \\ 0 & 0 & mx_gU_d \end{bmatrix}$$
(10)

1.2.2 Problem 2.b

The model is given in [1] as

$$M\dot{\nu_r} + N\nu_r = b\delta \tag{11}$$

and by rearranging we get that

$$\dot{\nu_r} = -M^{-1}N\nu_r + M^{-1}b\delta \tag{12}$$

The maneuvering model can be expressed as a state-space model as

$$\dot{\mathbf{x}} = \mathbf{A}\mathbf{x} + \mathbf{B}\mathbf{u},$$

y = Cx + Du

where

$$\mathbf{A} = -\mathbf{M}^{-1}N$$
$$\mathbf{B} = \mathbf{M}^{-1}b$$
$$\mathbf{C} = \begin{bmatrix} 0 & 1 \end{bmatrix}$$

$$\mathbf{D} = 0$$

The resulting transfer function can be approximated by using ss2tf(.)

$$\frac{r}{\delta}(s) = \frac{8.04 \cdot 10^{-5} s + 6.15 \cdot 10^{-6}}{s^2 + 0.151 s + 8.20 \cdot 10^{-4}} = \frac{K(T_3 s + 1)}{(T_1 s + 1)(T_2 + 1)}$$
(13)

With the numerical values yielding $K = 7.50 \cdot 10^{-3}$, $T_3 = 14.054$, $T_2 = 117.70$ and $T_1 = 6.90$.

1.2.3 Problem 2.c

Going from Equation (13) to the Nomoto model

$$\frac{r}{\delta}(s) = \frac{K}{Ts+1} \tag{14}$$

where $T := T_1 + T_2 - T_3 = 169.55$ and $K = 7.50 \cdot 10^{-3}$

1.2.4 Problem 2.d

A PID controller for heading autopilot is created using pole placement. We can also see that the actual yaw angle and yaw rate is following the desired yaw angle and yaw rate. From the north east position we can see that the course angle is different from zero, however this is because we have created a heading controller, not a course controller. Because of the slow dynamics, the controller use some time to reach the desired angle. That is what causes the north east angle.

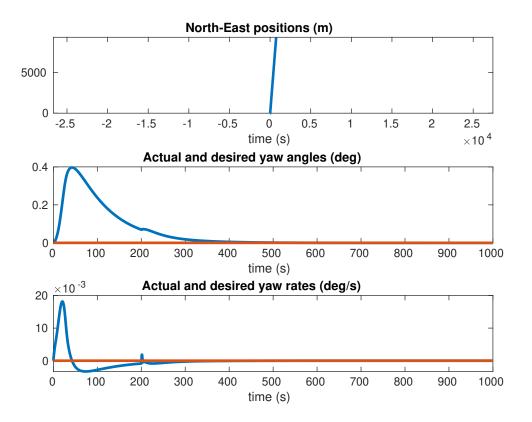


Figure 3: Boat behaviour with PID controller for heading autopilot.

1.2.5 Problem 2.e

The heading autopilot can compensate for the 10 deg to 20 deg maneuver, as can be seen in the North East position plot in fig. 4. Integral windup is not necessary in this case because we don't have saturation in the actuator. However it is good practice to implement.

```
M_{-}lin = MRB + MA;
110
    M_{-1in} = M_{-1in}(2:3,2:3);
111
112
    M_{-lin_{-inv}} = inv(M_{-lin});
113
    CRB_lin = [0]
                         m*U_d
114
115
                   0
                         m*xg*U_d
116
                                                -Xudot * U_d
117
    CA\_lin = [0]
                   -Yvdot * U_d + Xudot * U_d
                                                -Yrdot * U_d
118
119
120
    N_{-}lin = CRB_{-}lin + CA_{-}lin + D(2:3,2:3);
    b_{-}lin = [-2*U_{-}d*Y_{-}\Delta - 2*U_{-}d*N_{-}\Delta]';
121
122
    [num, den] = ss2tf(-M_lin_inv*N_lin, M_lin_inv*b_lin, [0 1], 0);
123
124
                                              % gain can be found by using the steady-state ...
125
    K_{lin} = num(3)/den(3);
          value (s=0) of the transfer function
126
127
    poles_lin = roots(den);
    zeros_lin = roots(num);
128
129
130
    T3_{lin} = -1/zeros_{lin};
    T1_{-1}in = -1/poles_{-1}in(1);
131
    T2\_lin = -1/poles\_lin(2);
132
```

```
134  T_lin = T1_lin+T2_lin-T3_lin;  % nomoto first—order time constant, eq. (7.24)
135
136  % controller gains (example 15.7)
137  Kp = wn^2*T_lin/K_lin;
138  Kd = (2*zeta*wn*T_lin-1)/K_lin;
139  Ki = wn/10*Kp;
```

Listing 3: Linearized C_a and C_{RB} and the gains for PID controller using pole placement

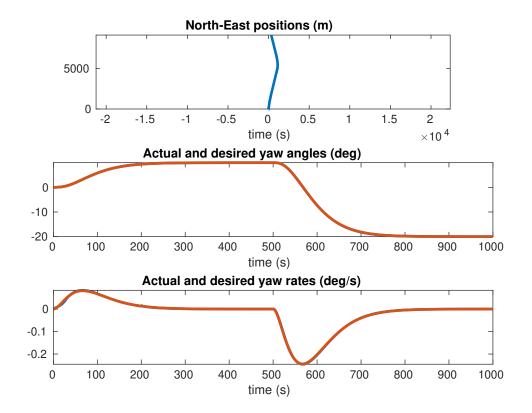


Figure 4: Boat behaviour when doing a 10-20 degree maneuver.

References

[1] T.I. Fossen. Handbook of Marine Craft Hydrodynamics and Motion Control. John Wiley & Sons, 2021.