

Satellite Payloads for Telecommunications

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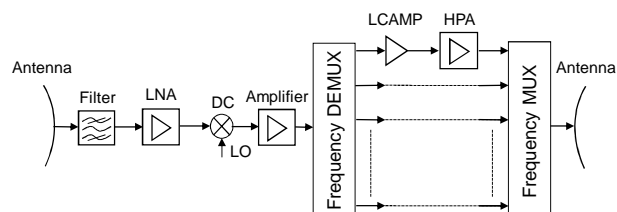
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Outline

- **Transparent payload**
 - Antennas
 - Transponders
 - Redundancy management
- **Flexible payload for High Throughput Satellite (HTS)**
 - Architecture for Single-Feed-Per-Beam (SFPB)
 - Architecture for Multiple-Feed-Per-Beam (MFPB)
 - On-Board Processing (OBP)
- **Examples of HTS Payloads**
 - Transparent payload
 - Flexible payload
 - Future flexible payload

Transparent Payload

- **Input section on the global bandwidth**
 - Wideband input filter to suppress the feed-back from the satellite transmission antenna
 - Low Noise Amplifier (LNA) to minimize the repeater noise temperature
 - Down Converter (D/C) to convert to the downlink frequency using a Local Oscillator (LO)
 - Wideband linear amplifier
- **Filtering and amplification per channel bandwidth**
 - Input Demultiplexer (IMUX or DEMUX) to split the input signal in several channels
 - Amplification based on a Linearized Channel Amplifier (LCAMP) and a High Power Amplifier (HPA)
 - Output Filter (OMUX) to removes thermal noise plus out-band harmonics generated by the HPA

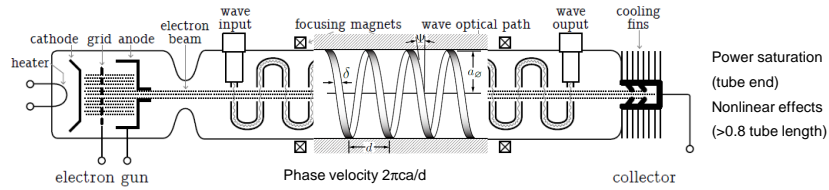


Antennas

- **Often mission-critical**
 - Risks during launch (often semi-detached)
 - Start of life (often complex deployment mechanisms)
 - Operational life (direct exposure to space environment)
- **Three complementary roles**
 - Transforming guided energy into radiation: Gain, efficiency, reflection coefficient, power handling, noise figure (antenna temperature, for reception)
 - Filtering in both frequency and space (angular) domain: Transfer characteristic (for different points in space), pattern shape (vs angle and frequency), isolation among input ports, polarization purity, group delay, out-of-band response (to help rejecting disturbances)
 - Transforming electric currents into radio waves: Circuit parameters (input impedance, S-parameters), Ohmic losses, spurious radiation (e.g. feeding lines), high-power effects (multipactor, Corona, passive inter-modulation)
- **Need of RF analysis and tests**
 - Antenna radiation highly dependent on spacecraft structure, antenna suite & mutual coupling effects

Travelling Wave Tube Amplifier (TWTAs)

- Slow wave structure to amplify wave when electron speed higher than wave phase velocity



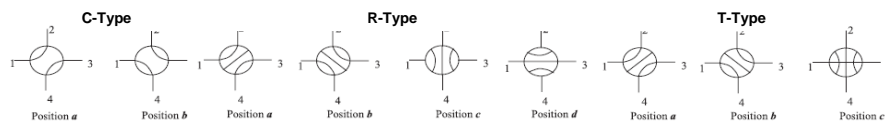
- Recent space TWTAs manufactured by Thales Electron Devices

Name	TL 4150	TH 4795	TH 4816	TH 4626	TH 4606C	THL40040CC
GHz	3.4-4.2	10.7-12.75	17.3-20.2	~ 26	32	37.5-42.5
P_{out}	150 W	150 W	160 W	50 W	35 W	40W
Gain	50 dB	> 50 dB	> 50 dB	> 50 dB	> 50 dB	48 dB
Eff.	73%	68%	63%	55%	54%	50%
Mass	1000 g	800 g	900 g	700 g	700 g	<890g

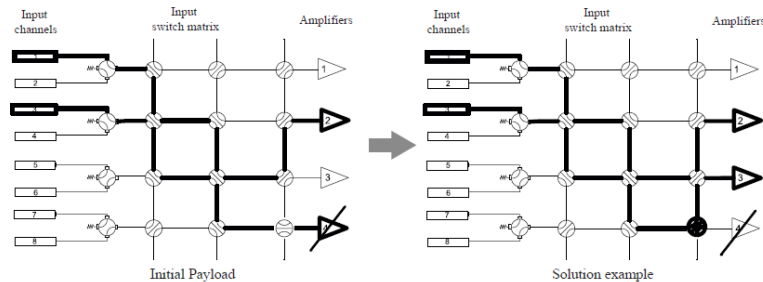
The Traveling-Wave Tube in the History of Telecommunication, D. Minenna & all, European Physical Journal, 2019

Redundancy Management

- C-type (4-port, 2-positions), R-type (4-port, 4-position), T-type (4-port, 3-position) switches



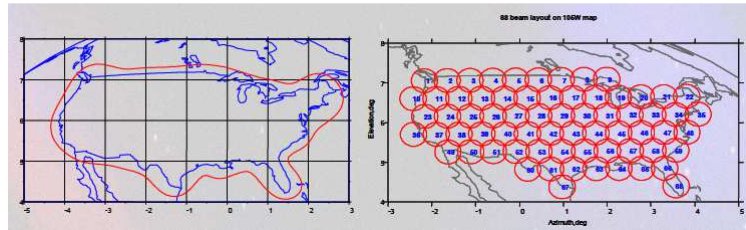
- Example of restoration in case of failure of a TWTAs



Optimising communication satellites payload conuguration with exact approaches, A. Stathakis & all, Engineering Optimization, 2015

Flexible Payload for High Throughput Satellite (HTS)

- From contoured beam to spot beams for feeder links and multibeam for user links



Global beam 31dBi (EOC) C/I=30dB

Multibeam with 4*17 Spots 0.6° 46dBi (EOC), C/I=12dB

- From RF flexibility to on-board processing
 - Downconverter (RF to IF or BB)
 - Analog to digital converter (sampling, quantization)
 - Switch (polyphase filters) or signal processing (demodulation, routing, modulation)
 - Digital to analog converter (sampling, quantization)
 - Upconverter (IF or BB to RF)

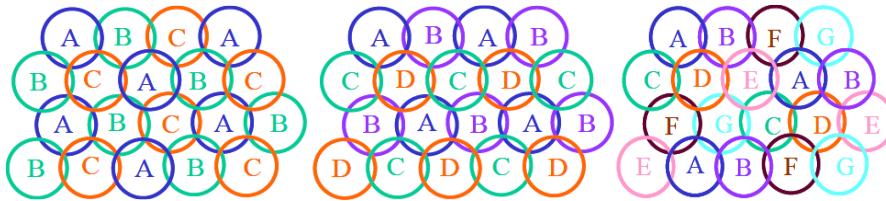
Approaches for Feeder Links

- Large RF gateways operating at Q/V or even W-band**
 - Dedicated on-board payload
 - Tenths of large RF gateways required operating in smart diversity
 - Terrestrial interconnection costly
- Small RF gateways sharing the Ka-band user link band**
 - Reuse of the Ka-band active antennas avoiding dedicated input section and offering full reconfigurability
 - Support of smart gateway diversity and progressive gateway deployment
 - Easier to install, lower connection cost but reusing user link precious bandwidth
- Optical gateways with very high rate optical links**
 - 3 on-board solutions: user link regeneration (complex and inflexible), signal sampling and quantizing (bandwidth expansion of 16 or so), RF over optical transmission (simplest payload but power inefficient)
 - Heavily affected by atmospheric impairments requesting spatial diversity and pre-correction techniques
 - N+P smart GW diversity to reduce bit rate by N (e.g. 3 active with 9 extra in diversity)

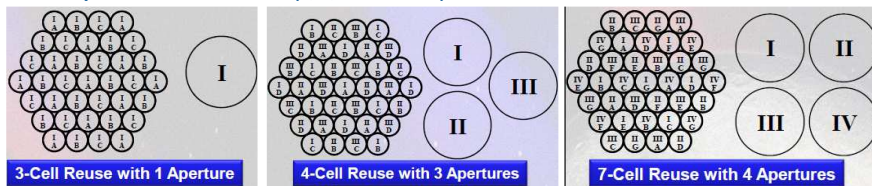
Challenges in future satellite communications, R. de Gaudenzi, IEEE Communication Theory Workshop, 2018

Single-Feed-Per-Beam (SFPB)

- Frequency reuse schemes (3, 4, 7)



- Aperture reuse schemes (1, 3, 4 reflectors)

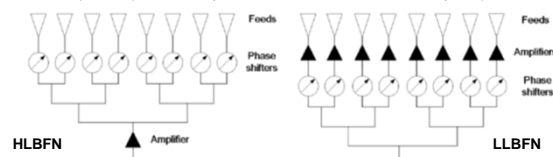


- Inmarsat-3 (L-Band, 1995), SkyTerra-1 (L-band, 2010), Eutelsat 172B (Ku-Band, 2017)...

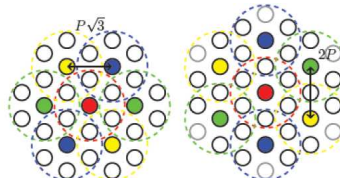
Development of a 45 GHz multiple-beam antenna for military satellite communications, IEEE TAP, S. Rao & all, 1995

Multiple-Feed-Per-Beam (MFPB)

- Each beam produced by a cluster of radiating elements fed by an appropriate BFN
 - Better C/I -> Reduced number of reflectors / Lower illuminating efficiency -> Lower gain
 - High Level BFN (HLBFN) between amplifiers and feeds -> More RF loss (Ka/Ku bands)
 - Low Level BFN (LLBFN) before amplifiers -> More efficient, but complex (S/L/X bands)



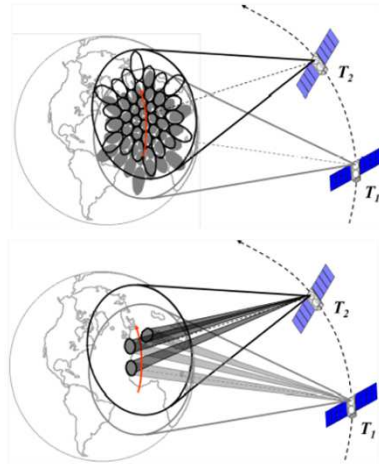
- Examples of MFPB using 7 feed horns, 4 colors frequency and polarization reuse



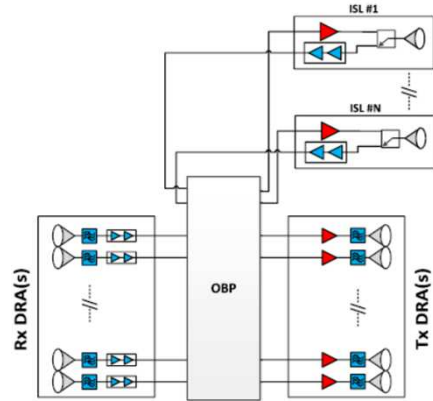
- Iridium (S-Band, 1997), Spaceway-3 (Ka-Band, 2007)...

Ka-Band multiple feed per beam focal array using interleaved couplers, IEEE TMTT, C. Leclerc & all, 2018

Multibeam Antenna in LEO



Satellite-fixed cell (LEO-MSS) vs. Earth-fixed cell (LEO-HTS)



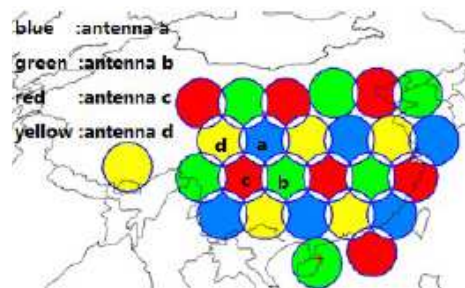
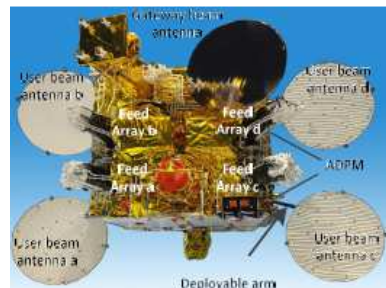
Payload with Intersatellite Links

Future technologies for very high throughput satellite systems, R. Gaudenzi & all, Wiley ISCN, 2019

Example of HTS Transparent Payload (Chinasat-16)

Antennas accommodation

- 1 reflector for gateway beam
- 4 reflectors (2 on the west panel, 2 on the east panel) mounted on a deployable arm
- RF tracking feed chain based on an RF signal transmitted from a ground beacon
- Angular pointing errors corrected at 0.05°



ADPM: Antenna Deployment and Pointing Mechanism

Ka-band multibeam antenna for Chinasat-16, Wei & all, IEEE, 2019

HTS Global Capacity

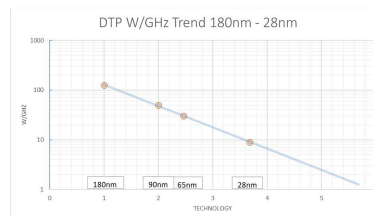
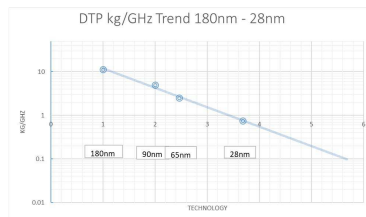
Total useful data rate for download and upload

$$C = \left(\sum_i (\eta_{FL_i} p_{FL_i}) B_{FL_i} + \sum_j (\eta_{RL_j} p_{RL_j}) B_{RL_j} \right) * N * P / K * \eta_{HL}$$

- p_{FL_i} & p_{DL_i} , uplink & downlink percentages of time minimizing link margins for a given link availability
- η_{FL_i} & η_{DL_i} , uplink & downlink low layer efficiencies (including guard bands) for p_{FL_i} & p_{DL_i}
- B_{FL_i} & B_{DL_i} , uplink & downlink bandwidths (not including double polarization)
- N, number of beams
- P, number of polarizations (1 or 2)
- K, number of colors (1/K frequency reuse factor)
- η_{HL} , high layers efficiency (throughput decrease due to overhead, signaling, latency)
- **Satellite throughput increase**
 - Use of Fade Mitigation Techniques (FMT) to limit link margins for a given link availability
 - Higher spectral efficiency (higher on-board power, better user terminal)
 - Use of higher frequency bands (Q/V or optical), especially for feeder links
 - Increase of number of beams (larger reflectors to have narrower beams)
 - Lower frequency reuse factor limiting interferences (beam hopping or non-orthogonal frequency reuse)
 - High layers protocols and on-board processing (lower overhead, signalling, latency)

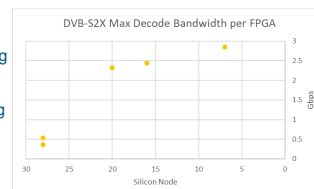
Future Digital Payloads

Mass and power per GHz of processed bandwidth



Components

- ASIC RF front-end for ADC/DAC and Digital Beamforming
- FPGA / SoC (> 250 MHz) for flexible MoDem
- Software packet switching / router for MAC layer handling
- High Speed Serial Link (HSSL) for interconnection
- Laser digital signal processing for ISL



Satellite radio-frequency payloads and instruments - Overview and challenges, ESA ESTEC, 2023