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ENGINEERING MECHANICS

UNIT-II Chapter – 6
SIMPLE STRESS AND STRAIN
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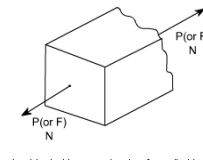
Properties of Materials

- some properties of materials which judge the strength of materials are given below:
- Elasticity: It is the property by virtue of which a material deforms under the applied load and it is enabled to return to its original position when the load is removed.
- Plasticity: It is the property of the material by which it undergoes inelastic strain beyond those at the elastic limit is known as the plasticity. It is also defined as unrecoverable strain.
- Ductility: It is the property of the material by which it undergoes considerable amount of deformation without rupture. A ductile material must possess a high degree of plasticity and strength.
- Brittleness: It is lack of ductility, a brittle material rupture with little or no plastic deformation.

Properties of Materials

- **Strength:** This is the maximum stress that a material can take. This is equal to maximum load divided by area.
- **Hardness:** It is the ability of material to resist against surface abrasion or indentation.
- **Toughness:** It is the property of material which enables it to absorb energy without fracture. It is represented by area under stress-strain curve upto fracture.
- **Fatigue:** Fatigue is the progressive and localized structural damage that occurs when a material is subjected to cyclic loading or repeated loading and unloading.

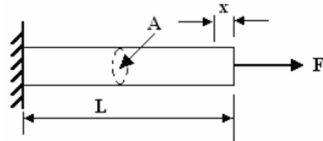
- **Stress:**
- when some external system of load acts on a body, the internal forces (equal & opposite) are set up at various section of the body, which resist the external force. This force per unit area at any section is known as stress.



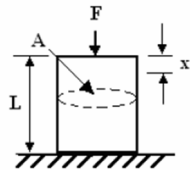
Types of Stresses & Strains

Direct Stress (σ)

When a force is applied to an elastic body, the body deforms. The way in which the body deforms depends upon the type of force applied to it.



Tensile Stress due to *tensile force*
A tensile force makes the body longer



Compressive Stress due to *compressive force*
A Compression force makes the body shorter.

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- **Resistance offered by the material per unit cross- sectional area is called STRESS**
- Tensile and compressive forces are called DIRECT FORCES
- Stress is the force per unit area upon which it acts.

$$\text{Stress} = \sigma = \frac{\text{Force}}{\text{Area}} = \frac{F}{A} \quad \text{N/m}^2 \quad \text{or Pascal (Pa)}$$

(σ is called as **Sigma**)

Note: Most of engineering fields used kPa, MPa, GPa.

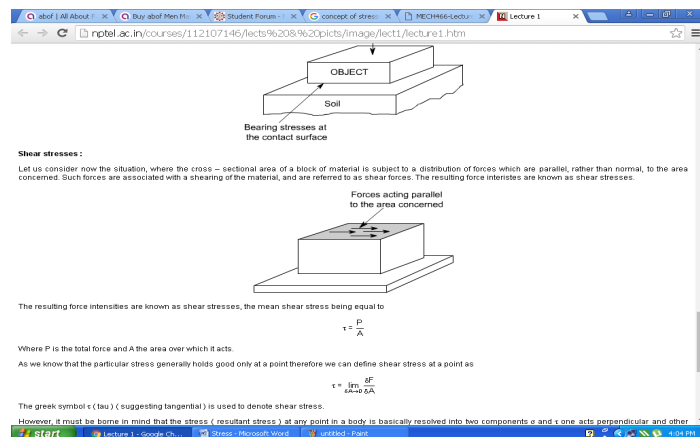
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TYPES OF STRESSES:

- Only two basic stresses exist: (1) normal stress and (2) shear stress. Other stresses either are similar to these basic stresses or are a combination of this e.g. bending stress is a combination tensile, compressive and shear stresses. Torsional stress, as encountered in twisting of a shaft is a shearing stress.

- Bearing Stress: When one object presses against another, it is referred to as bearing stress (They are in fact the compressive stresses).
- Here resulting force intensities are known as shear stresses, the mean shear stress being equal to
- Where P is the total force and A the area over which it acts

$$\tau = \frac{P}{A}$$



The screenshot shows a lecture slide with two diagrams. The top diagram illustrates bearing stress: a block labeled 'OBJECT' is shown pressing down on a surface labeled 'Soil'. Arrows point to the contact surface with the text 'Bearing stresses at the contact surface'. The bottom diagram illustrates shear stress: a block is shown with two horizontal arrows pointing in opposite directions, labeled 'Forces acting parallel to the area concerned'. Below this, the text states 'The resulting force intensities are known as shear stresses, the mean shear stress being equal to' followed by the equation $\tau = \frac{P}{A}$. Further text explains that P is the total force and A is the area, and that the Greek symbol τ (tau) is used to denote shear stress. A note at the bottom mentions that stress is resolved into components σ and τ .

Bearing stresses:

Let us consider now the situation, where the cross-sectional area of a block of material is subject to a distribution of forces which are parallel, rather than normal, to the area concerned. Such forces are associated with a shearing of the material, and are referred to as shear forces. The resulting force intensities are known as shear stresses.

Forces acting parallel to the area concerned

The resulting force intensities are known as shear stresses, the mean shear stress being equal to

$$\tau = \frac{P}{A}$$

Where P is the total force and A the area over which it acts.

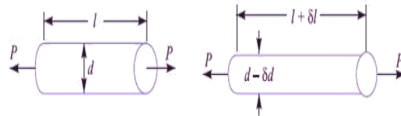
As we know that the particular stress generally holds good only at a point therefore we can define shear stress at a point as

$$\tau = \lim_{\Delta A \rightarrow 0} \frac{\delta F}{\Delta A}$$

The greek symbol τ (tau) (suggesting tangential) is used to denote shear stress.

However, it must be borne in mind that the stress (resultant stress) at any point in a body is basically resolved into two components σ and τ one acts perpendicular and other

- **Strain:** Strain is the deformation of a material from stress. It is simply a ratio of the change in length to the original length. Deformations that are applied perpendicular to the cross section are normal strains.
- **Linear Strain:** Linear strain of a deformed body is defined as the ratio of the change in length of the body due to the deformation to its original length in the direction of the force. If L is the original length and dL the change in length occurred due to the deformation, the linear strain e induced is given by $\epsilon = dL/L$.
- Linear strain may be a tensile strain, e_t or a compressive strain e_c according as dL refers to an increase in length or a decrease in length of the body. If we consider one of these as +ve then the other should be considered as -ve, as these are opposite in nature.
- **Lateral Strain:** Lateral strain of a deformed body is defined as the ratio of the change in length (breadth of a rectangular bar or diameter of a circular bar) of the body due to the deformation to its original length (breadth of a rectangular bar or diameter of a circular bar) in the direction perpendicular to the force.



Volumetric Strain

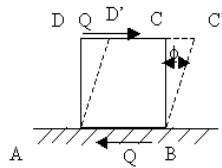
- **Volumetric Strain** is defined as the ratio of change in volume to the initial volume.

$$\text{Volumetric strain, } e_v = \frac{\text{Change in volume}}{\text{Initial volume}} = \frac{\Delta V}{V}$$

Where, ΔV = change in volume
 V = initial volume

Shear Strain:

- Shear strain is defined as the strain accompanying a shearing action. It is the angle in radian measure through which the body gets distorted when subjected to an external shearing action. It is denoted by ϵ
- Consider a cube ABCD subjected to equal and opposite forces Q across the top and bottom faces AB and CD. If the bottom face is taken fixed, the cube gets distorted through angle Φ to the shape ABC'D'. Now strain or deformation per unit length is
- Shear strain of cube = $CC' / CD = CC' / BC = \Phi$ radian



Direct Strain (ϵ) Also called as Longitudinal Strain

In each case, a force 'F' produces a deformation 'x'. In engineering, we usually change this force into stress and the deformation into strain and we define these as follows:

- **Strain is the deformation per unit of the original length.**



Strain, $\epsilon = \Delta L / L$ = Change in length / Original length
(ϵ is called as **Epsilon**)

- Strain has no unit's since it is a ratio of length to length

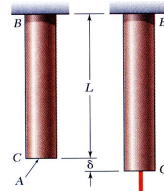


Fig. 2.1

$$\sigma = \frac{P}{A} = \text{stress}$$

$$\varepsilon = \frac{\delta}{L} = \text{normal strain}$$

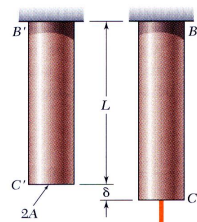


Fig. 2.3

$$\sigma = \frac{2P}{2A} = \frac{P}{A}$$

$$\varepsilon = \frac{\delta}{L}$$

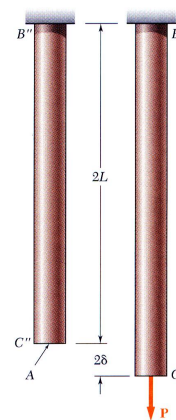


Fig. 2.4

$$\sigma = \frac{P}{A}$$

$$\varepsilon = \frac{2\delta}{2L} = \frac{\delta}{L}$$

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Hooks law

- Hooks law: Hooke's law stated that within elastic limit, The linear relationship between stress and strain for a bar in simple tension or compression is expressed by the equation, Stress \propto Strain, $= E\varepsilon$
- In which σ is the axial stress, ε is the axial strain, and E is a constant of proportionality known as the modulus of elasticity for the material. The modulus of elasticity is the slope of the stress-strain diagram in the linearly elastic region.
- The equation $\sigma = E\varepsilon$ is commonly known as Hooke's law

Hooke's law

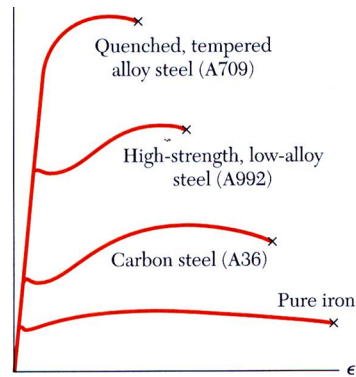


Fig. 2.16 Stress-strain diagrams for iron and different grades of steel.

Below the *yield stress*

Stress \propto Strain (ie) $\sigma \propto \epsilon$

$$\sigma = E \epsilon$$

Where **E** is a constant called as

Young's Modulus or **Modulus of Elasticity**

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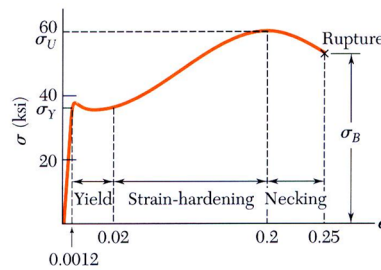
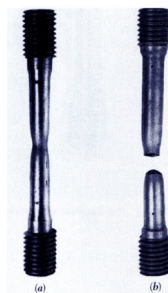
Elastic Constants

Elastic Constants:

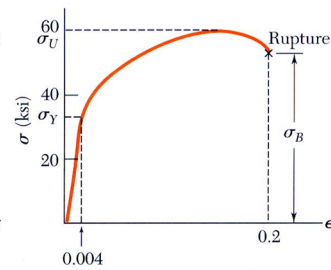
- The elastic constants mostly used in engineering mechanics E, G and K can be defined as follows
- **Modulus of elasticity** (Young's modulus) (E) describes tensile elasticity, or the tendency of an object to deform along an axis when opposing forces are applied along that axis; it is defined as the ratio of tensile stress to tensile strain. It is often referred to simply as the elastic modulus.

- **Modulus of Rigidity** or Shear Modulus (G): is the coefficient of elasticity for a shearing or torsion force. Modulus of Rigidity is the coefficient of elasticity for a shearing force. It is defined as "the ratio of shear stress to the displacement per unit sample length"
- Bulk modulus (K): When a body is subjected to like and equal direct stress along three mutually perpendicular directions, the ratio of this direct stress to corresponding volumetric strain is called Bulk modulus.
- The **Bulk Modulus Elasticity** or Volume Modulus is a material property characterizing the compressibility of a fluid, how easy a unit volume of a fluid can be changed when changing the pressure working upon it. It describes the elastic properties of a solid or fluid when it is under pressure on all surfaces.

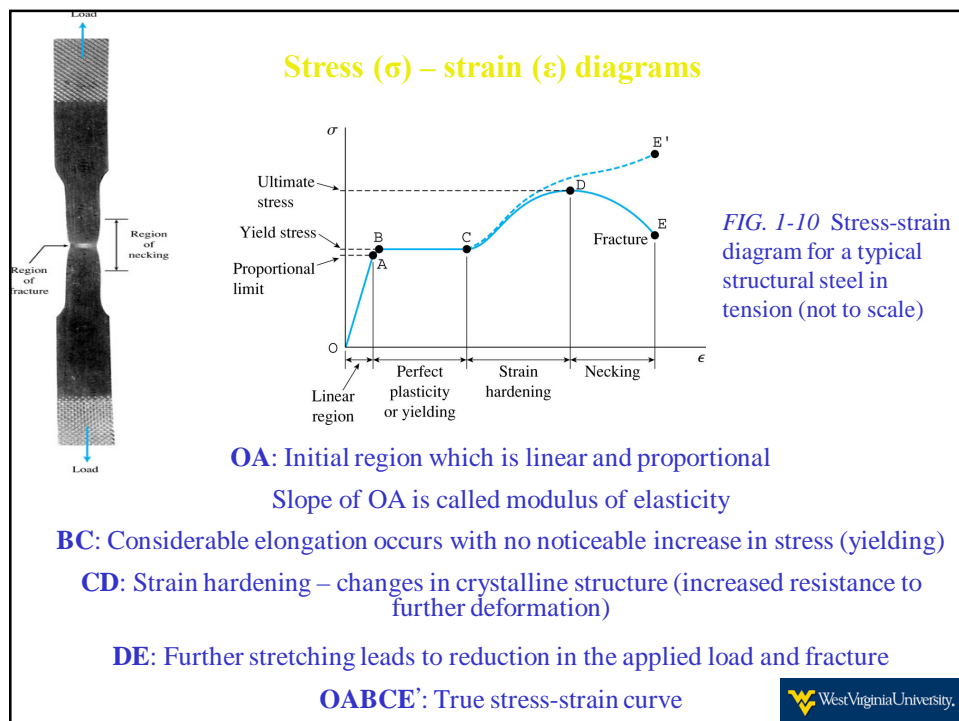
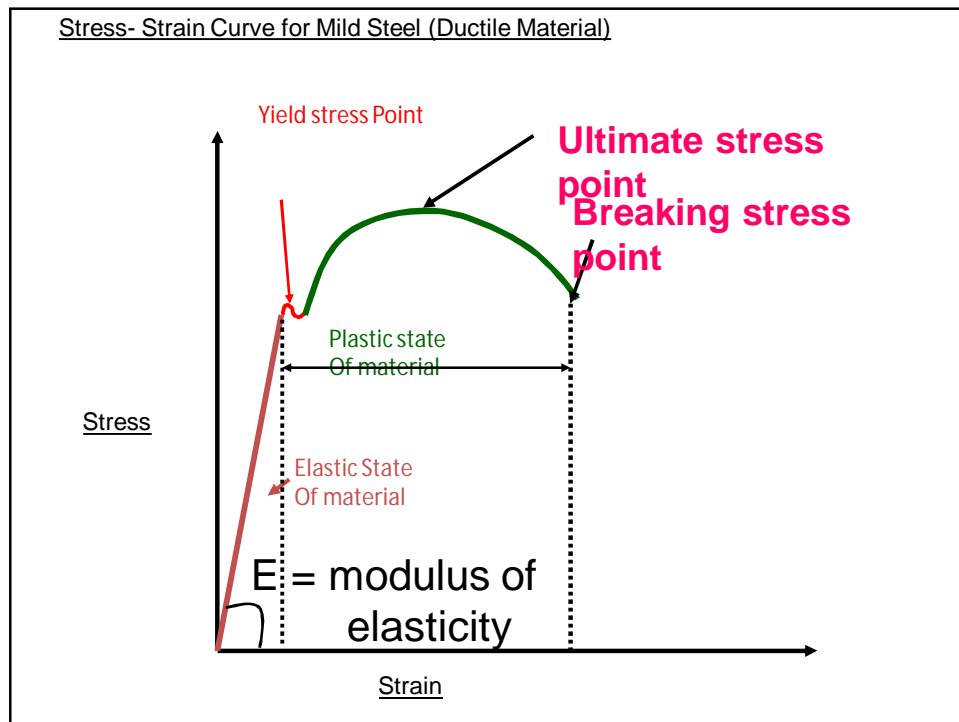
Stress-Strain diagram



(a) Low-carbon steel



(b) Aluminum alloy



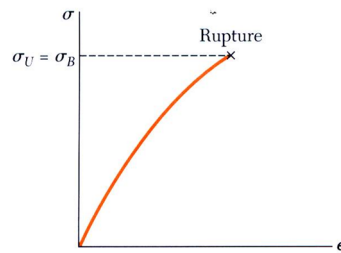


Fig. 2.11 Stress-strain diagram for a typical brittle material.

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Working stress

- The stress to which the material may be safely subjected in the course of ordinary use. Also called as **Allowable Load or Allowable stress**
- Max load that a structural member/machine component will be allowed to carry under normal conditions of utilisation is considerably smaller than the ultimate load
- This smaller load = Allowable load / Working load / Design load
- Only a fraction of ultimate load capacity of the member is utilised when allowable load is applied
- The remaining portion of the load-carrying capacity of the member is kept in reserve to assure its safe performance

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Factor of safety

- Factor of Safety (FoS) or safety factor (SF): is a term describing the load carrying capacity of a system beyond the expected or actual loads. Essentially, the factor of safety is how much stronger the system is than it usually needs to be for an intended load.
- It is a multiplier applied to the calculated maximum stress to which a component will be subjected. Typically, for components whose failure could result in substantial financial loss, or serious injury or death, a safety factor of at least four (4) is used. Non-critical components generally have a safety factor of two (2). Safety factors are needed to account for imperfections in materials, flaws in assembly, material degradation, and unexpected stresses.

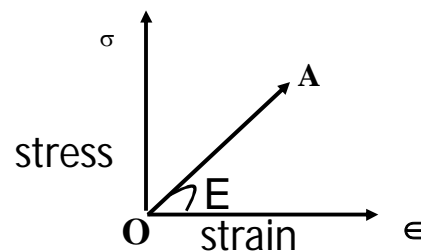
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Elastic moduli

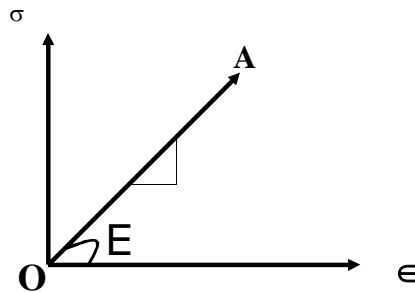
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Modulus of Elasticity:

- $\sigma = E \epsilon$
- Stress required to produce a strain of unity.
- Represents slope of stress-strain line OA.



Value of E is same in
Tension &
Compression.



• Hooke's Law:-

Up to elastic limit, Stress is proportional to strain

$$\sigma \propto \epsilon$$

$\sigma = E \epsilon$; where E=Young's modulus

$$\sigma = P/A \text{ and } \epsilon = \delta / L$$

$$P/A = E (\delta / L)$$

$$\delta = PL / AE$$

Volumetric Strain

Also we know that

$$e_1 = \frac{\sigma_1}{E} - \mu \frac{\sigma_2}{E} - \mu \frac{\sigma_3}{E}$$

$$e_2 = \frac{\sigma_2}{E} - \mu \frac{\sigma_1}{E} - \mu \frac{\sigma_3}{E}$$

$$e_3 = \frac{\sigma_3}{E} - \mu \frac{\sigma_1}{E} - \mu \frac{\sigma_2}{E}$$

$$\text{Further Volumetric strain} = e_1 + e_2 + e_3$$

$$= \frac{(\sigma_1 + \sigma_2 + \sigma_3)}{E} - \frac{2\mu(\sigma_1 + \sigma_2 + \sigma_3)}{E}$$

$$= \frac{(\sigma_1 + \sigma_2 + \sigma_3)(1 - 2\mu)}{E}$$

hence the

$$\boxed{\text{Volumetric strain} = \frac{(\sigma_1 + \sigma_2 + \sigma_3)(1 - 2\mu)}{E}}$$

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Problems

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