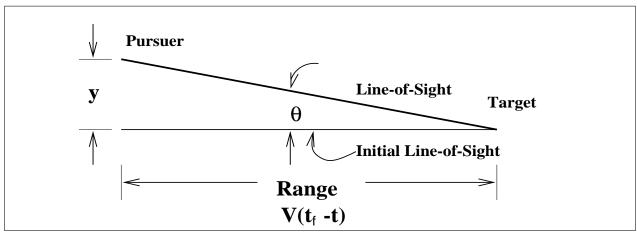
MAE 271B Project Missile State Estimation

Due Date: March 10, 2021

Consider estimating the relative state of a missile intercept (see Figure).



Missile Intercept Illustration

The dynamics of the problem are

$$\begin{split} \dot{y} &= v \\ \dot{v} &= a_p - a_{\scriptscriptstyle T}. \end{split}$$

where a_p , the missile acceleration, is known and assumed here to be zero. The input, a_T , is the target acceleration and is treated as a random forcing function with an exponential correlation,

$$E\left[a_{\scriptscriptstyle T}\right] = 0$$

$$E\Big[a_{\scriptscriptstyle T}(t)a_{\scriptscriptstyle T}(s)\Big] = E[a_{\scriptscriptstyle T}^2]e^{\frac{-|t-s|}{\tau}}.$$

The scalar, τ , is the correlation time. The initial lateral position, $y(t_0)$, is zero by definition. The initial lateral velocity, $v(t_0)$, is random and assumed to be the result of launching error:

$$E[y(t_0)] = 0$$

$$E[v(t_0)] = 0$$

$$E[y(t_0)^2] = 0$$

$$E[v(t_0)^2] = 0$$

$$E[v(t_0)^2] = 0$$

$$E[v(t_0)^2] = 0$$

The measurement, z, consists of a line-of-sight angle, θ . For $|\theta| \ll 1$,

$$\theta \approx \frac{y}{V_c(t_f - t)}.$$

It will also be assumed that z is corrupted by fading and scintillation noise so that

$$z = \theta + n$$

$$E[n(t)] = 0$$

$$E[n(t)n(\tau)] = V\delta(t - \tau) = \left[R_1 + \frac{R_2}{(t_f - t)^2}\right]\delta(t - \tau).$$

The process noise spectral density, W, is

$$W = GE \left[a_T^2 \right] G^T = \left[\begin{array}{ccc} 0 & 0 & 0 \\ 0 & 0 & 0 \\ 0 & 0 & E[a_T^2] \end{array} \right].$$

Consider the case where the parameters of the problem are

$$V_c = 300 \frac{\text{ft}}{\text{sec}}, E[a_T^2] = [100 \,\text{ft sec}^{-2}]^2, \qquad t_f = 10 \,\text{sec}, \qquad R_1 = 15 \times 10^{-6} \,\text{rad}^2 \,\text{sec},$$

 $R_2 = 1.67 \times 10^{-3} \,\text{rad}^2 \,\text{sec}^3, \qquad \tau = 2 \,\text{sec},$

The initial covariance is

$$P(0) = \begin{bmatrix} 0 & 0 & 0 \\ 0 & (200 \,\text{ft sec}^{-1})^2 & 0 \\ 0 & 0 & (100 \,\text{ft sec}^{-2})^2 \end{bmatrix}.$$

which may be the parameters for a Falcon or Sparrow guided missile.

- 1. Compute the error variance of the error in the state and the associate Kalman filter gains. See Figures 9.6 and 9.7 in the book.
- 2. Develop a simulation for the dynamics and the measurement.
- 3. Show by a Monte Carlo simulation that the actual error variance computed in the simulation matches the *a priori* error variance used in the Kalman filter gains.
- 4. Show that the residual process is uncorrelated in time, i.e. a white noise process.
- 5. The Gauss-Markov model is an approximation to the random telegraph signal model that is more realistic. In this model the value a_T changes sign at random times given by a Poisson probability. We assume that $a_T(0) = \pm a_T$ with probability .5 and $a_T(t)$ changes polarity at Poisson times. The probability of k sign changes in a time interval of length T, P(k(T)), is $P(k(T)) = \frac{(\lambda T)^k e^{-\lambda T}}{k!}$ where λ is the rate. The probability of an even number of sign changes in T, P(even # in T), is

$$P(even \# \text{ in } T) = \sum_{\substack{k=0 \ k \text{ even}}}^{\infty} \frac{(\lambda T)^k e^{-\lambda T}}{k!} = e^{-\lambda T} \sum_{k=0}^{\infty} \frac{(1 + (-1)^k)(\lambda T)^k}{2k!}$$

Since $e^{\lambda T} = \sum_{k=0}^{\infty} \frac{(\lambda T)^k}{k!}$, then

$$P(even \# \text{ in } T) = e^{-\lambda T} \left[\sum_{k=0}^{\infty} \frac{(\lambda T)^k}{2k!} + \sum_{k=0}^{\infty} \frac{(-\lambda T)^k}{2k!} \right] = \frac{1}{2} e^{-\lambda T} \left[e^{\lambda T} + e^{-\lambda T} \right] = \frac{1}{2} \left[1 + e^{-2\lambda T} \right]$$

By a similar process, $P(odd \# in T) = \frac{1}{2} [1 - e^{-2\lambda T}]$. Then the probability that $a_T(t)$ is positive is

$$P(a_T(t) = a_T) = P(a_T(t) = a_T | a_T(0) = a_T) P(a_T(0) = a_T)$$

$$+ P(a_T(t) = a_T | a_T(0) = -a_T) P(a_T(0) = -a_T)$$

$$= \frac{1}{2} P(even \# \text{ in } T = [0, t]) + \frac{1}{2} P(odd \# \text{ in } T = [0, t])$$

$$= \frac{1}{2} \left\{ \frac{1}{2} \left[1 + e^{-2\lambda t} \right] + \frac{1}{2} \left[1 - e^{-2\lambda t} \right] \right\} = \frac{1}{2}$$

Then, the mean acceleration, \bar{a}_T is

$$\bar{a}_T = a_T P(a_T(t) = a_T) - a_T P(a_T(t) = -a_T) = 0$$

The autocorrelation, $R_{a_T a_T}(t_1, t_2) = E[a_T(t_1)a_T(t_2)]$ is

$$R_{a_T a_T}(t_1, t_2) = a_T^2 P(a_T(t_2) = a_T(t_1)) - a_T^2 P(a_T(t_2) \neq a_T(t_1))$$

$$= a_T^2 \frac{1}{2} \left[1 + e^{-2\lambda|t_2 - t_1|} \right] - a_T^2 \frac{1}{2} \left[1 - e^{-2\lambda|t_2 - t_1|} \right] = a_T^2 e^{-2\lambda|t_2 - t_1|}$$

If $\frac{1}{\tau} = 2\lambda$ then the autocorrelation of the Gauss-Markov process is the same as the random telegraph signal $(t_2 - t_1) = (t - s)$ and the means of both are zero.

We need a method of generating the random switching times. Let $T = t_{n+1} - t_n$ be the time between two switch times. The probability that the switch occurred after t_{n+1} is

$$P(T' > T | t = t_n) = 1 - P(T' \le T | t = t_n)$$

Now the probability that no switch occurred in T, but occurred after t_{n+1} , is

$$P(T' > T | t = t_n) = P(\# \text{ of sign changes in T is zero}) = e^{-\lambda T}$$

Then,

$$P(T' \le T | t = t_n) = 1 - e^{-\lambda T}$$

which is the probability that at least one change occurred. To produce the random time t_{n+1} , set $P(T' \leq T | t = t_n)$ equal to U, the output of [0, 1] from a uniform density function. Then,

$$1 - e^{-\lambda T} = U \implies e^{-\lambda T} = 1 - U$$

$$\Rightarrow -\lambda T = \ln(1 - U)$$

$$\Rightarrow T = \frac{-1}{\lambda} \ln(1 - U)$$

$$\Rightarrow t_{n+1} = t_n - \frac{1}{\lambda} \ln(1 - U)$$

Since 1-U is also a uniform density function, then

$$t_{n+1} = t_n - \frac{1}{\lambda} \ln(U)$$

The objective of the above is to use the random telegraph signal in the simulation rather than the Gauss-Markov process that was used to ensure that the Kalman filter was implemented correctly. By running a Monte Carlo analysis using the random telegraph signal with $a_T = 100 \,\text{ft sec}^{-2}$ and $\lambda = .25/\text{sec}$, determine the actual error variance.