Milestone Review Flysheet 2017-2018

nstitution Purdue University

Milestone CDR

Vehicle Properties			
Total Length (in)	122		
Diameter (in)	5.15		
Gross Lift Off Weigh (lb.)	30		
Airframe Material(s)	FWFG		
Fin Material and Thickness (in)	3/16" G10 FG		
Coupler Length/Shoulder Length(s) (in)	12/5		

Motor Properties		
Motor Brand/Designation	Aerotech L1520-T	
Max/Average Thrust (lb.)	352.5	
Total Impulse (lbf-s)	835.16	
Mass Before/After Burn (lb.)	8.05/4.09	
Liftoff Thrust (lb.)	340.1	
Motor Retention Method	Aeropack Retainer	

Stability Analysis		
Center of Pressure (in from nose)	94.11	
Center of Gravity (in from nose)	77.75	
Static Stability Margin (on pad)	3.18	
Static Stability Margin (at rail exit)	2.25	
Thrust-to-Weight Ratio	9.4	
Rail Size/Type and Length (in)	1.5, 144	
Rail Exit Velocity (ft/s)	81.25	

Ascent Analysis			
Maximum Velocity (ft/s) 649.7			
Maximum Mach Number	0.58		
Maximum Acceleration (ft/s^2)	300.2		
Predicted Apogee (From Sim.) (ft)	5281		

Recovery System Properties					
	Dro	gue Parachu	te		
Manufacturer/Model			Skyangle B2		
S	ize/Diameter (in	or ft)	24"		
Alt	itude at Deploym	nent (ft)	Apogee		
Velo	ocity at Deploym	ent (ft/s)	2		
Т	erminal Velocity	(ft/s)	89.5		
Recovery Harness Material		Tubular Kevlar			
Recovery Harness Size/Thickness (in)		1/2" Thick			
Recovery Harness Length (ft)		40'			
Harness/Airframe ends and 1/		nk through looped tether /4" SS U-bolts through bulkheads			
Kinetic Section 1		Section 2	Section 3	Section 4	
Energy of Each Section (Ft-	2403	1207	585	N/A	

Recovery System Properties					
Main Parachute					
Manu	facturer/Mo	del	Skyanele B2		
Size/Di	ameter (in o	rft)	100"		
Altitude a	at Deployme	ent (ft)	70	700	
Velocity a	t Deploymer	nt (ft/s)	8	6	
Termin	al Velocity (ft/s)	13	13.5	
Recovery Harness Material		Tubular Kevlar			
Recovery Harness Size/Thickness (in)		1/2" Thick			
Recovery Harness Length (ft)		40'			
Harness/Airtrame Intertaces		k link through looped tether SS U-bolts through bulkheads			
Kinetic Energy of	Section 1	Section 2	Section 3	Section 4	
Each Section (Ft- lbs)	54.6	27.5	13.25	N/A	

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Recovery Electronics			
Altimeter(s)/Timer(s)	Altus Metrum Telemetrum,		
(Make/Model)	Missileworks RRC3+ Sport		
	Fully redundant and		
Redundancy Plan and	independent systems with		
Backup Deployment Settings	individual batteries, switches,		
Settings	wires, and ejection charges		
Pad Stay Time (Launch Configuration)			

Recovery Electronics				
Rocket Locators (Make/Model)	Altus Metrum Telemetrum			
Transmitting Frequencies (all - vehicle and payload)	70cm ham band			
Ejection System Energetics (ex	. Black Powder)	Black Powder		
Energetics Mass - Drogue	Primary	4		
Chute (grams)	Backup	4		
Energetics Mass - Main Chute	Primary	3.2		
(grams)	Backup	3.2		
Energetics Masses - Other	Primary	N/A		
(grams) - If Applicable	Backup	N/A		

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nstitution	Purdue University	Mileston€	CDR
	Pay	load	
	ray	Overview	
		- Creation	
Payload 1 (official payload)			
	Our payload is programmed to identify three 40'x40' tarps wit from an onboar	h known RGB values in real time. The camera with a Rasberry Pi.	nis is done by processing live video taken
		Overview	
Payload 2 (non- scored payload)			
		N/A	
		•	
	Test Plans, Sta	tus, and Results	
Ejection Charge Tests	Will perform continuity checks using light bulbs to detect o	pens or shorts and ensure a comple	ete circuit. Will also ground test using
	energetics prior to flight to ensure pro		
Sub-scale Test Flights			

Will fly a full scale rocket on a full scale motor as if it were the scored flight. The rocket will contain a working redundant camera system and tarps of different colors will be staked to the ground. This will ensure that the design is sound and stable, our ejection system and recovery gear works as intended, and the payload functions properly

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Full-scale Test Flights

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		Additional Comme	nts	
		N/A		