			Milesto	ne Review F	lysheet 2017-	-2018				
nstitution Purdue University				ity		Milestone		PDR		
Valida Buratin					Mater Preparties					
Vehicle Proper					Motor Brand	Motor Properties Motor Brand/Designation Aerotech L1520-T				
Total Length (in)			5.15			Max/Average Thrust (lb.)		352.5		
Diameter (in)						Total Impulse (lbf-s)		835.16		
Gross Lift Off Weigh (lb.)			35.4 FWFG		<u>-</u>	Mass Before/After Burn (lb.)		8.05/4.09		
Airframe Material(s)			0.1875 G10 FG			Liftoff Thrust (lb.)		340.1		
Fin Material and Thickness (in)		12/5			` '		eropack Retainer			
Coupler Length/Shoulder Length(s) (in)		12/5		Wotor Retern	WIOLOI RELEIILIOII WIELIIOO A		Eropack Netamer			
	St	ability Analy	/sis			Δ	scent Analy	sis		
Center o	of Pressure (in fr		94.79		Maxi	Maximum Velocity (ft/s)			2.27	
	of Gravity (in fro	,	77.38			Maximum Mach Number		0.62		
	itability Margin		3.38			Maximum Acceleration (ft/s^2)		323		
	ability Margin (a					Predicted Apogee (From Sim.) (ft)		5640		
	ust-to-Weight R		2.49 9.9		Fredicted	Apogee (11011	1 31111.) (11)	30	J40	
			-	144		Posovor	y System D	roportios		
Rail Size/Type and Length (in) Rail Exit Velocity (ft/s)			83.45				y System Pı ain Parachı			
110	II Exit velocity (I	14,3,	65.45		Ma	Manufacturer/Model Skyanele B2			iele B2	
	Pacava	ry System Di	onartics			Size/Diameter (in or ft)			100"	
		ry System Progue Parach				Altitude at Deployment (ft)			700	
M			Skyangle B2			Velocity at Deployment (ft/s)		86		
Manufacturer/Model Size/Diameter (in or ft)		24"			Terminal Velocity (ft/s)		13			
	•	<u> </u>	Apogee			Recovery Harness Material		Tubular Kevlar		
Altitude at Deployment (ft) Velocity at Deployment (ft/s)		2			Recovery Harness Size/Thickness (in)		1/2" Thick			
	minal Velocity (92		· · · · · ·	Recovery Harness Size/Trickness (iii) Recovery Harness Length (ft)			·0'	
		-	Tubular Kevlar		Recovery He	Ty Harriess Lei		, , ,		
Recovery Harness Material Recovery Harness Size/Thickness (in)		1/2" Thick		Harness/Airfra Interfaces		1/4" SS quick link through looped ends and 1/4" SS U-bolts throu bulkheads				
Kecov	ery Harness Len	gtn (rt)	4	0	Kinetic					
Harness/Airfra	ame Interfaces		nk through loop U-bolts through		Energy of Each Section (Ft-lbs)	Section 1 42.5	Section 2 23.43	Section 3 10.01	Section 4 N/A	
Kinetic	Section 1	Section 2	Section 3	Section 4						
Energy of Each Section	2315	1277	550	N/A		Reco	very Electr	onics		
(Ft-lbs)	2315	12//	550	N/A		Rocket Locators (Make/Model)		Altus Metrum Telemetrum		
	Rec	overy Electro	onics		Transmitting	Fraguancias				
Altimeter(s)/Timer(s) Altus		Altus M	1etrum Telemetrum,		(all - vehicle	Transmitting Frequencies (all - vehicle and payload)		***Required by CDR***		
(iviake)	iviouelj	Missile	works RRC3	+ Sport		ection System Energetics (ex.		i		
Podundani	cy Dlan and	Fully redundant and			Energetics Mass - Drogue Chute (grams)		Primary Backup	_	.2	
Redundancy Plan and Backup Deployment Settings		independent systems w		I		,			.2	
			l batteries, s		Energetics N		Primary		.2	
		wires, and ejection charges		charges	Chute (grams)		Backup	3.2		
Pad Stay Time (Launch Configuration)					Energetics Masses - Other (grams) - If Applicable		Primary		/A	
							Backup N/A			

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	Payload						
	Overview						
Payload 1 (official payload)							
	Our payload is programmed to identify three 40'x40' tarps with known RGB		y processing live video taken from an				
<u> </u>	onboard camera with a Rasberry Pi.						
	Overvie	W					
Payload 2 (non-scored payload)							
	N/A						
	1911						
	Test Plans, Status, and	Results					
Ejection Charge Tests							
	Will perform continuity checks using light bulbs to detect opens or shorts and ensure a complete circuit. Will also ground test using energetics prior to flight to ensure proper pressurization and recovery gear deployment.						
Sub-scale Test Flights							
	WIII fly an exact scale model of our full scale rocket on a smaller motor with that the design is sound and stable, the ejection system and recovery gear we shapes and contact the state of the state o	works as intended, and that the pa					
Full-scale Test Flights	Will fly a full scale rocket on a full scale motor as if it were the scored flight. Ti different colors will be staked to the ground. This will ensure that the design intended, and the payloac	n is sound and stable, our ejection					

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	Ad	dditional Comments	5			