

1st Design Iteration

Team Name: GENEI RYODAN

Wing Information		
Airfoil	NACA2412	
Span	1417.9	
Root Chord	202.6	
Tip Chord	202.6	
Offset		
AR	7	
S_w	287.2	
Incidence Angle		
Twist		
Dihedral		

Vertical Tail Information			
Airfoil	NACAOOO9		
Semi Span (Total Length)	324.9		
Root Chord	108.3		
Tip Chord	108.3		
Offset			
S_v/S_w	122.5		
$\frac{S_v/S_w}{AR_v}$	3		
V_v	0.035		
Incidence Angle			
Tail Arm	405.1		
Shifted Length in Z-Direction			

Horizontal Tail Information		
Airfoil	NACADDO9	
Span	440.3	
Root Chard	146.8	
Tip Chord	146.8	
Offset		
S_H/S_W	0.225	
S_H/S_W AR_H	3	
V_H	0.45	
Incidence Angle	-2	
Tail Arm	405.1	
Shifted Length in Z-Direction	0	

Propulsion System Information					
Brushless Motor	Manufacturer	Model	Voltage	No Load	Resistance

			Constant (KV)	Current (A)	(ohm)
	Electrifly	Rimfire .15	1200	1.7	0.0415
Dattony	Manufacturer	No. Of Cells	Voltage	Capacity (mAh)	C-Rating
Battery	LiPo	3	3.7	5000	45/60C
Propeller Size (Diameter x Pitch)	10*4.7				
Speed Controller (Current Rating Value)	60A				

Flight Phases

Phase 1 (payload isn't deployed)		
MTOW	1.5	
X_{CG}	146.7	
Static margin (%)	10	
CL_{Cruise}	0.1236	
V_{Stall}	8	
V_{Cruise}	26	
$lpha_{Trim}$	1.86	
Required Static Thrust	0.609	
Required Dynamic Thrust (at V_{Cruise})	0.3672	

Phase 2 (payload is deployed)		
Mass	1.05	
X_{CG}	146.7	
Static margin (%)	10	
CL_{cruise}	0.1209	
V_{Stall}	8	
V_{Cruise}	22	
$lpha_{Trim}$	1.86	
Required Static Thrust	0.4734	
Required Dynamic Thrust (at V_{cruise})	0.2848	

ightharpoonup Note: lengths should be in (mm), angles in (deg), weights and thrust in (Kg).