

Part Number (Rev Level) and Part Name							
	KU661-SM19	Dilatant Failure					
Basic Information							
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	Date:	7-Mar-22					
	Software:	Patran/Nastran 2018					
Executive Summary							
	<i>The objective of this project is to use the knowledge gained from ME661 and apply it to real life scenarios. In this case, FEA is used to analyze some part that attach to a pilot's helmet using Patran/Nastran 2018. The model was made by a KU team to help with Dilatant LLC, to make a design that will help reduce whiplash that may occur during flight.</i>						
Problem Statement							
	<i>Two parts of the design were isolated and observed using Patran/Nastran2018. An analysis was required to see how the part's current geometry affects the stress concentrations in the part and possibly provide recommendations to changes in the geometry.</i>						
System Properties							
	Component	Weight (lbs)	Material				
	Cylinder	0.0028	Titanium				
	Total	0.0028	-				
	Component	Weight (lbs)	Material				
	Ball Joint	0.0107	Titanium				
	Total	0.0107	-				
	Material	E (ksi)	n	syield (ksi)	sultimate (ksi)	sallowable (ksi)	Density(lb/in ³)
	1018 Steel	1.68E+04	0.340	128.0	138	128	0.163
Model Geometry							

Both models were made through SolidWorks2018. The cylinder, however, ran into some problems with it's original shape, so the model was then made through Patran/Nastran2018 by making a surface made with seven points, connected them with curves, then made into a surface by using the vertices option. It then had the mesh swept 360° to make the cylinder shape.

Mesh (figures located in Mesh worksheet)

The Cylinder was a troublesome model to deal with. The surface was made with a Quad4 with an AutoMesh Paver. It was then swept 360° to form a cylinder. Elements 1-304 forms the surface which has the Quad4 mesh. From 305-18544, the Elements have a Hex8 mesh.

The ball joint was created after applying a significant number of mesh seeds to the part. The hole and curved edge connecting to the sphere used 32 element mesh seeds, the short edge connecting to the sphere and long plate edges used 16 element mesh seeds. The outermost face opposite of the sphere has an 8 element mesh seed on the short edge and 16 element mesh seed on the long edges. After creating the mesh seeds, a Tetrahedral mesh was applied to the solid.

Case	Element Type	AutoMesh	# of Elements	Total Elements	Nodes	Dof
Cylinder1	Quad4/Hex8	Paver	40	4223	5680	17040
Cylinder2	Quad8/Hex20	Paver	30	510	3720	11160
Cylinder3	Quad8/Hex20	Paver	20	340	2480	7440
Cylinder4	Quad8/Hex20	Paver	10	187	1240	3720
Ball Joint 1	Tet4	Isomesh	8228	8228	1871	5613
Ball Joint 2	Tet10	Isomesh	8228	8228	12854	38562
Ball Joint						
Ball Joint						

Loads and Restraints (mesh with boundary conditions shown in worksheets LR1, LR2, etc.)

Load Case 1: Cylinder

	Load Type	Location	Magnitude	Direction	Notes		
	Force	Inner Lip Top Face	1	Positive Y	Force applied on the edge of lip		
Restraint Case 1: Cylinder							

	Restraints	Location	Magnitude	Direction	Notes	
	Displacement	Bottom Face	0	All		
	Load Case 2: Ball Joint					
	Load Type	Location	Magnitude	Direction	Notes	
	Force	Inside Right Hole Node	1	Positive Z		
	Restraint Case 2: Ball Joint					
	Restraints	Location	Magnitude	Direction	Notes	
	Displacement	Sphere	<0,0,0>	All	Displacement conditions in rough circular shape where cylinder contacts.	
	MPCs	Right Half of Hole Nodes	Uz=Uz	N/A	Deforms hole evenly in Z-direction	
	Analysis of Results (plots shown in worksheets Results 1, Results 2, etc.)					
	<p>The effective stress concentration factor was determined by comparing the maximum stress to a reference stress, which is the stress the part would have had if the part had an axial load applied evenly over its smallest cross-sectional area. The reference area for the cylinder was taken at its base, with a cross-sectional area measured to be .043 sq in. The reference area for the ball-socket joint was taken at the thinnest cross-section of the beam where the hole is, determined to be .0116 sq in.</p> <p>A quirk in the FEA program when dealing with total load resulted in much larger loads being applied than the intended unit loads. For the purpose of this lab, which is to estimate stress concentration factors, the reference stress is adjusted to the actual loads that were applied in the OLOAD section, so this should not significantly effect the resulting concentration factors.</p>					
	Load Case	Max Stress	Reference Stress	Effective Stress Concentration Factor		
	Cylinder 1	12000.0000	920	13		
	Cylinder 2	18400.0000	690	27		
	Cylinder 3	13900.0000	460	30		
	Cylinder 4	8120.0000	460	18		
	Ball Joint 1	102.0000	86.2	1.18		
	Ball Joint 2	387.0000	86.2	4.49		

	Other Areas of Note						
	Load Case	Notes					
	Cylinder						
	Ball Joint						

Model Verification Checks							
Case1: Cylinder1							
Total Applied Loading vs Resultant							
Load Type	Fx	Fy	Fz	Mx	My	Mz	
OLOAD	0.00E+00	4.00E+01	0.00E+00	-2.24E-08	0.00E+00	9.31E-09	
SPCFORCE RESULTANT	3.75E-14	-4.00E+01	-1.80E-15	2.24E-08	1.59E-15	-9.31E-09	
Epsilon vs External Work							
Epsilon	-3.09E-15						
External Work	1.62E-03						
Residual Work	-4.99E-18						
Case2: Cylinder2							
Total Applied Loading vs Resultant							
Load Type	Fx	Fy	Fz	Mx	My	Mz	
OLOAD	0.00E+00	3.00E+01	0.00E+00	1.27E-07	0.00E+00	6.89E-08	
SPCFORCE RESULTANT	1.77E-13	-3.00E+01	-2.60E-14	-1.27E-07	-5.17E-15	-6.89E-08	
Epsilon vs External Work							
Epsilon	4.97E-15						
External Work	1.00E-03						
Residual Work	4.98E-18						

External Work	4.18E-04
Residual Work	3.10E-18

Model Verification Checks								
	Case1: Ball Joint (Tet4)							
	Total Applied Loading vs Resultant							
	Load Type	Fx	Fy	Fz	Mx	My	Mz	
	OLOAD	0.00E+00	0.00E+00	1.00E+00	4.87E-02	5.80E-04	0.00E+00	
	SPCFORCE RESULTANT	-2.44E-15	7.71E-16	-1.00E+00	-4.86E-02	-5.80E-04	4.39E-16	
	Epsilon vs External Work							
	Epsilon	2.16E-14						
	External Work	4.95E-07						
	Residual Work	1.07E-20						
	Case2: Ball Joint (Tet10)							
	Total Applied Loading vs Resultant							
	Load Type	Fx	Fy	Fz	Mx	My	Mz	
	OLOAD	0.00E+00	0.00E+00	1.00E+00	4.87E+02	5.80E-04	0.00E+00	
	SPCFORCE RESULTANT	-4.69E-15	3.94E-16	-1.00E+00	-4.87E-02	-5.11E-04	-4.18E-16	
	Epsilon vs External Work							
	Epsilon	-7.85E-14						

	External Work	7.25E-07						
	Residual Work	-5.68E-20						

Conclusions/Recommendations

Under such loads, the cylinder would definitely not be able to withstand a impulse force of 15 G, judging by the fact that under a total point load of 40 lbs, the material yielded. The main reason could perhaps because of the thickness of the lip was not thick enough and not enough support around the edges. Different alloys could also perhaps strengthen the model. As for the ball joint, it withstood better than the cylinder, but at the end, it will not be enough to withstand a 15G impulse force.

Both models were terribly small and the group did predict that they would not be able to resist the force well enough. However, it did show how the models deformed and where improvements can be made.

Both the cylinder and the ball joint results had results with a somewhat mediocre results. However, not enough mesh analysis were made for the ball joint to see any convergence, and the cylinder's data had a big range because of the inconsistency of the loading.

The loading was down on each FEM node on the model in Patran/Nastran2018, which resulted in the data showing the y load being larger than 1. Each of the Y loads are the sum of all the node loads, so in a 30 element mesh, a total of 30 lbf was applied.

For the ball joint, the geometry was very difficult to handle, as well as the meshing on the spherical part.

There are definitely recommendations for future cases like these models. Having the same number of elements and changing the topology would have brought out a more consistent data, and perhaps see the convergence. For the ball joint, saving multiple files with a basic model that will have the same loading but different mesh will save a lot more time and more efficient.