

Chapter 4

goal is to show the Kepler problem of planetary orbits can be solved w/ Lagrangian Mech
 ↳ first consider the general solution for motion in a 1D potential $V(q)$

Motion of "benign" one-dimensional system

^{1 deg of freedom, w/ L(q, q̇)}

any 1-Deg of free problem can be solved by expressing the answer as an integral

↳ $t(q)$, inverse of what one means by a sol'n ($q(t)$), $q(t)$ can't always be expressed analytically,

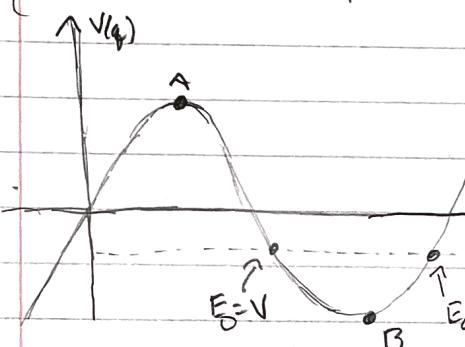
information about the soln to any 1D problem

can be deduced from graphing the potential ($V(q)$) w.r.t q)

then consider what happens as func of over, E, how are

$$E = T + V = \frac{1}{2}m\dot{q}^2 + V(q)$$

$$\dot{q} = \pm \sqrt{\frac{2T}{m}} = \pm \sqrt{\frac{2(E - V(q))}{m}}$$



* while $E \geq V$ T is positive and non-zero.

* graph of how the potential changes as a func of gen coord.

↳ In our potential $V(q)$ for small energies E_0 , system

behaves like $\ddot{q} \propto q$, when $E_0 = V$, $\dot{q} = 0$, $T = 0$

↳ called turning points (approximates parabola)

→ for higher energies it's no longer a parabola

(rearranging from above)

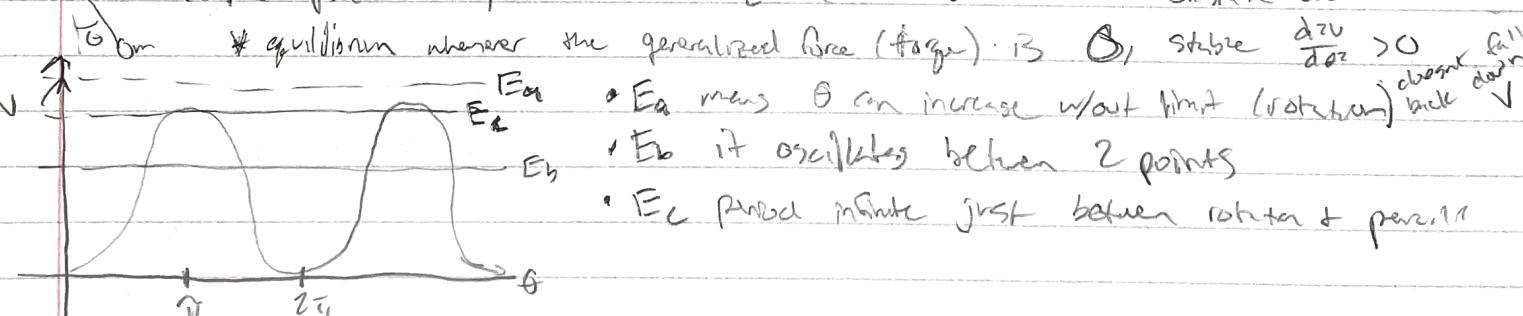
$$\frac{dq}{\pm \sqrt{2(E - V(q))}} = dt \Rightarrow t = \int_0^q \frac{dq}{\sqrt{2(E - V(q))}}$$

Grandfather clock

(scaled time)

$$V(\theta) = mgd(1 - \cos\theta) \rightarrow E = \frac{\dot{\theta}^2}{2} + (1 - \cos\theta)$$

$$\text{unstable } \frac{d^2V}{d\theta^2} < 0$$



[multivariable
taylor series]

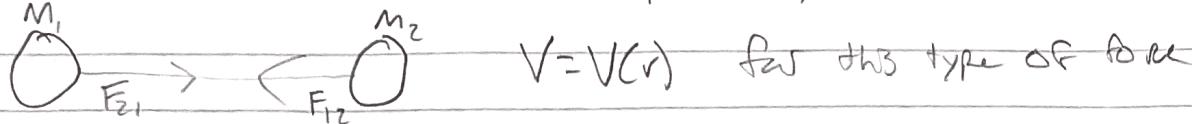
$$P_3(x) = f(x_0) + \frac{\partial f}{\partial x_1}(x_0) \cdot v_1 + \frac{\partial f}{\partial x_2}(x_0) v_2 + O(V^3)$$

$$\begin{aligned} x - x_0 &= V \\ v_1 &= v_2 \text{ for later} \end{aligned}$$

$$\begin{aligned} \vec{r}_1 &= \vec{r}_1(a) + \vec{e}_a \\ \vec{r}_2 &= \vec{r}_2(a) + \vec{e}_a \end{aligned}$$

Central Forces

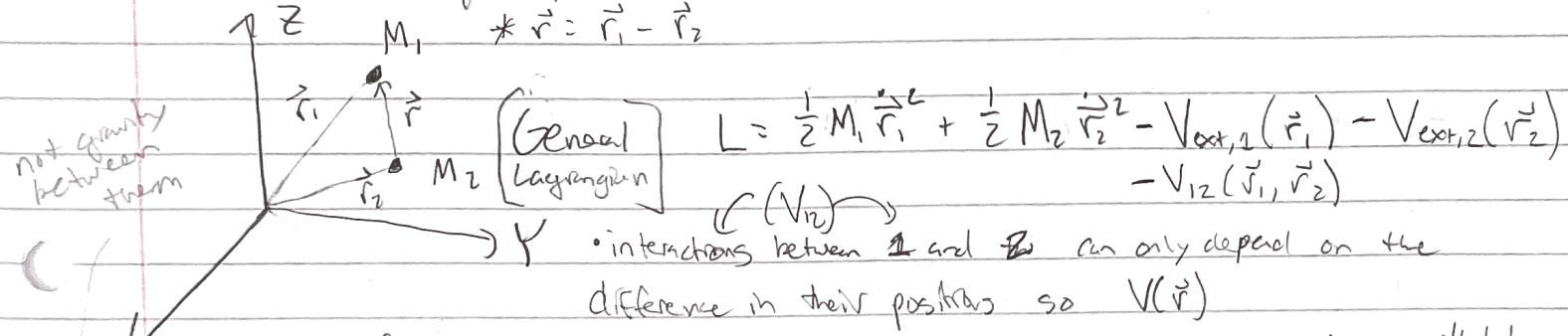
- applying tools earlier in the chapter to solve the motion due to gravity
- we want a central force that depends only on the distance "r" between them



$$V = V(r) \text{ for this type of force}$$

Motion of the center of mass

- you can predict the motion of the center of mass of an exploding artillery shell without knowing the internal forces released, center of mass of a system behaves as a single particle under the action of external gravitational force



- external forces can only depend on the particles absolute positions, think about falling in gravity (external) to both, it will depend on their absolute positions,

[No external forces] $\rightarrow V_{\text{ext}} = 0$, only force is between particle 1 and particle 2

\hookrightarrow follows that L doesn't change by translating the origin. So we can

$$\begin{aligned} \vec{r}_1 &\Rightarrow \vec{r}_1 + \vec{a} \\ \vec{r}_2 &\Rightarrow \vec{r}_2 + \vec{a} \end{aligned} \quad \left. \begin{array}{l} \text{can do this since} \\ \text{it doesn't change} \\ \text{the position} \end{array} \right\}$$

$$L = L(\vec{r}_1, \vec{r}_2, \dot{\vec{r}}_1, \dot{\vec{r}}_2) = L(\vec{r}_1 + \vec{a}, \vec{r}_2 + \vec{a}, \dot{\vec{r}}_1, \dot{\vec{r}}_2) \quad \left[\begin{array}{l} \text{translational} \\ \text{invariance} \end{array} \right]$$

[make an infinitesimal translation] $\rightarrow L(\vec{r}_1 + \vec{\epsilon}\vec{a}, \vec{r}_2 + \vec{\epsilon}\vec{a}, \dot{\vec{r}}_1, \dot{\vec{r}}_2)$
[$\vec{\epsilon}\vec{a}$, make ϵ arbitrarily small]

$$[\text{Taylor expand this in } \vec{\epsilon}] = L(\vec{r}_1, \vec{r}_2, \dot{\vec{r}}_1, \dot{\vec{r}}_2) + \vec{\epsilon}\vec{a} \cdot (\vec{\nabla}_1 + \vec{\nabla}_2)L + O[\vec{\epsilon}^2]$$

$$\begin{cases} \vec{r}_1 \rightarrow \vec{r}_1 + \vec{\epsilon}\vec{a} \\ x = x_0 + V \end{cases}$$

$\vec{\nabla}_1$ come from partial derivatives and factoring out L

[both variables are changed by $\vec{\epsilon}\vec{a}$ so can also pull that out]

$$\vec{\nabla}_1 = \left(\frac{\partial}{\partial x_1}, \frac{\partial}{\partial y_1}, \frac{\partial}{\partial z_1} \right), \vec{\nabla}_2 = \dots$$

$\nabla_1 L = -\nabla_1 V$ (since T depends only on \vec{q}_1)
 $\nabla_2 L = -\nabla_2 V$
 $-\nabla_1 V = \text{force on particle } 1,$ by definition a
 $-\nabla_2 V = \text{force on particle } 2,$ 3rd law pair so
 $\nabla_2 V = 0$ it has to be 0

Now we have $= L(\vec{r}_1, \vec{r}_2, \dot{\vec{r}}_1, \dot{\vec{r}}_2) + \vec{F}_1 \cdot (\vec{\dot{r}}_1 + \vec{\dot{r}}_2)L$

↳ let's look at $[\vec{\nabla}_1 L + \vec{\nabla}_2 L]$ the change in L for \vec{r}_1
 the change in L for \vec{r}_2

Since external forces are 0, L is invariant under coordinate transformations. We can choose an \vec{a} , such that the $\vec{\nabla}_1 L + \vec{\nabla}_2 L = 0$
 ↳ I think you bring origin to center \vec{a} here $\left[\begin{array}{ccc} 0 & \vec{r}_1 & \vec{r}_2 \\ & \vec{r}_1 & \vec{r}_2 \end{array} \right]$ so $\frac{\partial}{\partial x_1}$ and $\frac{\partial}{\partial x_2}$ are equal and opposite?

look at $\frac{\partial L}{\partial x_1} + \frac{\partial L}{\partial x_2} = 0$

$$\frac{d}{dt} \frac{\partial L}{\partial \dot{x}_1} = \frac{d P_{1x}}{dt} \quad \text{and} \quad \frac{d}{dt} \frac{\partial L}{\partial \dot{x}_2} = \frac{d P_{2x}}{dt}$$

$$\frac{d}{dt} (P_{1x} + P_{2x}) = 0 \rightarrow \boxed{\vec{P}_{\text{tot}} = \text{const} = \vec{P}_1 + \vec{P}_2}$$

$\frac{d \vec{P}_{\text{tot}}}{dt} = \frac{d}{dt} [M_1 \dot{\vec{r}}_1 + M_2 \dot{\vec{r}}_2] = 0$ let's redefine and look at this ex w/ regard to a R_{cm}

(quark)
 (center) $\vec{P}_{cm} = M \dot{\vec{R}}_{cm} = (m_1 + m_2) \dot{\vec{r}}_{cm}$ and $P_{cm} = M \dot{r}_{cm}$ $\left[\vec{R}_{cm} = \frac{M_1 \vec{r}_1 + M_2 \vec{r}_2}{M} \right]$

now make from $r = r_1 - r_2$
 $[r_2 = r_1 - r]$ solve for r_1 and r_2 $\left[\begin{array}{l} \vec{r}_1 = \vec{R}_{cm} + \frac{M_2}{M} \vec{r} \\ \vec{r}_2 = \vec{R}_{cm} - \frac{M_1}{M} \vec{r} \end{array} \right]$ $M = M_1 + M_2$

$$T = \frac{1}{2} M_1 \dot{\vec{r}}_1^2 + \frac{1}{2} M_2 \dot{\vec{r}}_2^2 = \frac{1}{2} M \dot{\vec{R}}_{cm}^2 + \frac{1}{2} \left(\frac{M_1 M_2}{M_1 + M_2} \right) \dot{\vec{r}}^2$$

$\mu = \text{reduced mass}$

$$L = \frac{1}{2} M \dot{\vec{R}}_{cm}^2 + \frac{1}{2} \mu \dot{\vec{r}}^2 - V(r)$$

$\underbrace{}$ motion about CM $\underbrace{}$ motion about M

We can treat the $\dot{\vec{R}}_{cm} = \text{constant} = 0$ think about gravity between Sun and Earth, $R_{cm} = 0$.

$$L = \frac{1}{2} \mu \dot{\vec{r}}^2 - V(r)$$

Angular Momentum is conserved for central forces

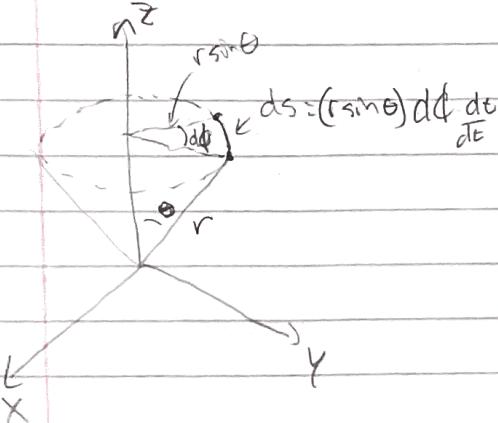
$$L = \frac{1}{2} \mu \dot{r}^2 - V(r)$$

no external forces $V(\vec{r}_1, \vec{r}_2) = V(\vec{r}_1 - \vec{r}_2)$
+ translational invariance

$$V(\vec{r}_1 - \vec{r}_2) = V(|\vec{r}_1 - \vec{r}_2|) = V(r)$$

only central force = rotational invariance

"central force" means that the interaction between the particles doesn't depend on their absolute spatial position or orientation but only dist between them



$$\left(\frac{ds}{dt}\right)^2 = \dot{r}^2 = \dot{x}^2 + \dot{y}^2 + \dot{z}^2, \text{ do inside Lagrangian}$$

but spherical

$$L = \frac{1}{2} \mu (r^2 \dot{\theta}^2 + r^2 \dot{\phi}^2 \sin^2 \theta) - V(r)$$

$$\frac{\partial L}{\partial \dot{\phi}} = 0 \quad \frac{\partial L}{\partial \phi} = \text{constant} = \mu r^2 \dot{\phi} \sin^2 \theta = l_z$$

l_z is the z -component of the angular momentum

just like l_z is a constant, you could reconst so \vec{l} has
 x, y, z components $(0, 0, l_z)$ like in my drawing

$$dr = r d\dot{\theta}$$

2 consequences of angular momentum conservation

- a) entire motion takes place in a plane
- b) "Equal Areas swept in Equal time"

a) $\vec{l} = \text{constant} = \vec{r} \times \vec{p}$, \vec{p} and \vec{r} must be in a plane $\perp \vec{l}$ for all time

$$\frac{d\vec{l}}{dt} = \frac{d}{dt}(\vec{r} \times \vec{p}) = \vec{r} \times \frac{d\vec{p}}{dt} + \vec{p} \times \frac{d\vec{r}}{dt}$$

$\frac{d\vec{p}}{dt}$ = total net force, for central forces
w/ no external forces = 0

$$\vec{l} = \vec{r} \times \vec{p} = \text{constant}$$

Using $\vec{r} \times \vec{p}$ are in same plane, all motion is in the same plane

so $\dot{\theta} = 0$, then $\theta = \text{constant}$, lets say its $\theta = \pi/2$, makes l along z

$$\Rightarrow l_z = \mu r^2 \dot{\phi} (\sin \theta = 1)$$

$$A = \frac{1}{2} (r \sin \theta)^2 d\phi = \frac{1}{2} (\sin \theta)^2 \dot{\phi} dt$$

$$\frac{dA}{dt} = \frac{l_z}{2\mu} = \text{constant} \quad \checkmark$$

a) and b) proven

Gravitational Attraction

$V(r) = -\frac{GM}{r}$

of Potential between 2 bodies

$\rightarrow V_{ext} = \frac{1}{2}r^2$

Look back at the Lagrangian

now we have, $\dot{\theta} = 0, \theta = \frac{\pi}{2}, \dot{\phi} = \frac{l}{mr^2}$

* dont substitute $\dot{\phi}$ first into the Lagrangian

↳ ELE are derived under the fact of independent variations in each coordinate, if you eliminate it first then you dont have that

$$H = E = 2T - L = \frac{1}{2}m(\dot{r}^2 + r^2\dot{\theta}^2 + r^2\sin^2\theta\dot{\phi}^2) + V(r)$$

$$E = \frac{1}{2}m\dot{r}^2 + \underbrace{\frac{\partial^2}{2mr^2} + V(r)}_{V_{ext}}, \quad \frac{\partial^2}{2mr^2}$$

comes from the kinetic energy
and is responsible for the
centrifugal force

ELE	$\ddot{r} = \frac{\partial^2}{mr^2} - \frac{dV}{dr}$
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Gravitational Attraction

Gravitational Attraction

[potential energy between 2 bodies] $V(r) = -\frac{GM_1 M_2}{r}$ define $\frac{l^2}{2\mu} = \beta$

$\Rightarrow V_{\text{eff}} = \frac{l^2}{2\mu r^2} + V(r) = -\frac{k}{r} + \frac{\beta}{r^2} = V_{\text{eff}}$ * derived this V_{eff} from central forces, and conservation laws

$L = T - V = \frac{1}{2}\mu r^2 - V_{\text{eff}} \xrightarrow{\text{EOM}} \ddot{r} = \frac{l^2}{\mu r^3} - \frac{k}{r^2}$, * inverse square law of gravity is opposed by the "centrifugal force"

Getting a better solution

[make the substitution] $u = \frac{1}{r}$ w/ our assumption about inverse $\ell = \mu r^2 \dot{\phi}$ $\frac{d}{dt} = \frac{d}{du} u^2 \frac{du}{dt}$

\Rightarrow rearranging $r = \frac{dr}{dt} = \frac{d}{dt}(\frac{1}{u}) = -\frac{1}{u^2} \frac{du}{dt} = -\frac{l}{\mu} \frac{du}{d\phi}$

$\Rightarrow [E] \rightarrow E = \frac{1}{2}\mu r^2 + \frac{l^2}{2\mu r^2} + V(r) \xrightarrow{-\frac{k}{r}}$

\Rightarrow turns into $E = \frac{l^2}{2\mu} \left[\left(\frac{du}{d\phi} \right)^2 + u^2 \right] - k u$ E is a constant $\Rightarrow \frac{dE}{d\phi} = 0$

\Rightarrow $\left[\text{put } \frac{dE}{d\phi} = 0 \right] \rightarrow \frac{d^2u}{d\phi^2} + u = \frac{\mu k}{l^2}$ \rightarrow "a SHO, solve by inspection" $u = \frac{1}{r} = \frac{\mu k}{l^2} + A \cos \phi$ $T = 2\pi$

put into here

rewriting the sol'n to a more familiar look.

$P \equiv \frac{l^2}{\mu k}$ and $\epsilon \equiv PA$ $\rightarrow P\dot{u} = \frac{P}{r} = 1 + \epsilon \cos \phi \Rightarrow P = r + \epsilon r \cos \phi$

\Rightarrow (switch to cartesian) $P = \sqrt{x^2 + y^2} + \epsilon x$ square and rearrange $(1 - \epsilon^2)x^2 + 2\epsilon Px + y^2 - P^2 = 0$ if $\epsilon = 0$ then it's a circle $r = P$

$\epsilon = 1$: parabola
 $\epsilon > 1$: hyperbola ϵ is the eccentricity of the ellipse get from

Total Energy: $E = \frac{\mu k^2}{2\lambda^2} (\epsilon^2 - 1) = \frac{k}{2P} (\epsilon^2 - 1) = \frac{-k}{2a} = E$ a_1 is the semi-major axis

start w/ eq 4.53: complete the square $(1 - \epsilon^2)(x + \frac{\epsilon P}{1 - \epsilon^2})^2 + y^2 = \frac{P^2}{1 - \epsilon^2} \Rightarrow \frac{(x - x_c)^2}{a^2} + \frac{y^2}{b^2}$

$x_c = \text{ellipse center}$ $a \pm -\frac{\epsilon P}{1 - \epsilon^2}$

$a = \frac{P}{1 - \epsilon^2}$, $b = \sqrt{1 - \epsilon^2}$, turning points when $v = 0$ $\frac{dr}{dt} = \frac{dr}{d\phi} \frac{d\phi}{dt} = 0 \rightarrow$ motion about sun $r_{\max} = \text{aphelion}$ $r_{\min} = \text{perihelion}$

MJRK ON BACK

$$\hookrightarrow \frac{dr}{dt} = \frac{dr}{d\phi} \frac{d\phi}{dt} = \frac{\ell^2}{\mu r^2} \frac{dr}{d\phi}, \quad r(\phi) = \rho (1 + \varepsilon \cos \phi)^{-1}$$

$$\frac{dr}{dt} = 0 = \frac{\rho \varepsilon \sin \phi}{(1 + \varepsilon \cos \phi)^2} \quad \Rightarrow \quad \boxed{\phi = 0, \pi} \quad r(\phi=0) = \frac{\rho}{1+\varepsilon} = a(1-\varepsilon) = r_{min}$$

$$r(\phi=\pi) = \frac{\rho}{1-\varepsilon} = a(1+\varepsilon) = r_{max}$$

Normality to find t
we will use our
integral definition

$$\Rightarrow \frac{x}{z} = \sqrt{\frac{\mu}{2}} \int_{r_{min}}^{r_{max}} \frac{dr}{\sqrt{K - V(r) - \frac{e^2}{4\pi r^2}}} \quad * \text{ easier way tho.}$$

1) $\frac{dA}{dt} = \frac{dt}{d\phi} \Rightarrow \int_0^{A=a b \pi} \frac{dA}{dA/dt} = \int_0^T dt \quad \Rightarrow \boxed{T: \frac{2\mu}{\ell} \pi a b = \frac{2\mu}{\ell} \frac{\pi \rho^2}{(1-\varepsilon^2)^{3/2}} = 2\pi \sqrt{\frac{\mu}{K}} \sqrt{a^3}}$

$\varepsilon > 1$ hyperbola: hyperbolic orbit only possible if the force is repulsive
 \hookrightarrow scattering of positively charged particles

