PD-Control

- PD-Control
- Feedback Linearization

- PD-Control
- Feedback Linearization
- Robust and Adaptive Motion Control

Given a mechanical system with n-degrees of freedom

$$D(q)\ddot{q}+C(q,\dot{q})\dot{q}+g(q)= au,\ q=\left[q_1,\ldots,q_n
ight]^{\mathrm{\scriptscriptstyle T}},\ oldsymbol{ au}=\left[oldsymbol{ au_1},\ldots,oldsymbol{ au_n}
ight]^{\mathrm{\scriptscriptstyle T}}$$

Given a mechanical system with n-degrees of freedom

$$D(q)\ddot{q}+C(q,\dot{q})\dot{q}+g(q)=oldsymbol{ au},\ q=\left[q_1,\ldots,q_n
ight]^{ \mathrm{\scriptscriptstyle T} },\ oldsymbol{ au}=\left[oldsymbol{ au_1},\ldots,oldsymbol{ au_n}
ight]^{ \mathrm{\scriptscriptstyle T} }$$

We suppose that each of degrees of freedom is controlled by a geared DC-motor

$$J_{m_k}\ddot{ heta}_k + B_k\dot{ heta}_k = rac{K_{m_k}}{R_k}V_k - rac{1}{r_k} au_k, \quad k=1,\,\ldots,\,n$$

where

- θ_k is the k^{th} -motor angle;
- r_k is a gear ration;
- J_{m_k} , B_k , K_{m_k} , R_k are parameters or computed from parameters of the k^{th} DC-motor

Given a mechanical system with n-degrees of freedom

$$D(q)\ddot{q}+C(q,\dot{q})\dot{q}+g(q)=oldsymbol{ au},\ q=\left[q_1,\ldots,q_n
ight]^{{\scriptscriptstyle T}},\ oldsymbol{ au}=\left[oldsymbol{ au_1},\ldots,oldsymbol{ au_n}
ight]^{{\scriptscriptstyle T}}$$

We suppose that each of degrees of freedom is controlled by a geared DC-motor

$$J_{m_k}\ddot{ heta}_k + B_k\dot{ heta}_k = rac{K_{m_k}}{R_k} V_{m k} - rac{1}{r_k} m{ au_k}, \quad k=1,\,\ldots,\,n$$

and that the motor and the link angles are related by

$$\theta_{m_k} = r_k q_k, \quad k = 1, \ldots, n$$

Given a mechanical system with n-degrees of freedom

$$D(q)\ddot{q}+C(q,\dot{q})\dot{q}+g(q)= au,\ q=\left[q_1,\ldots,q_n
ight]^{\mathrm{\scriptscriptstyle T}},\ oldsymbol{ au}=\left[oldsymbol{ au_1},\ldots,oldsymbol{ au_n}
ight]^{\mathrm{\scriptscriptstyle T}}$$

We suppose that each of degrees of freedom is controlled by a geared DC-motor

$$oldsymbol{-}J_{m_k}\ddot{ heta}_k+B_k\dot{ heta}_k=rac{K_{m_k}}{R_k}oldsymbol{V_k}-rac{1}{r_k}oldsymbol{ au_k},\quad k=1,\,\ldots,\,n$$

and that the motor and the link angles are related by

$$\theta_{m_k} = r_k q_k, \quad k = 1, \ldots, n$$

Then the actuator equations are

$$ightarrow r_k^2 J_{m_k} \ddot{q}_k + r_k^2 B_k \dot{q}_k = r_k rac{K_{m_k}}{R_k} V_{m k} - m{ au_k}, \quad k=1,\,\ldots,\,n$$

Improved Dynamical Model (Cont'd)

The dynamical systems

$$D(q)\ddot{q}+C(q,\dot{q})\dot{q}+g(q)= au,\ q=\left[q_1,\ldots,q_n
ight]^{\mathrm{\scriptscriptstyle T}},\ au=\left[au_1,\ldots, au_n
ight]^{\mathrm{\scriptscriptstyle T}}$$

$$r_{m{k}}^2 J_{m_k} \ddot{q}_k + r_{m{k}}^2 B_{m{k}} \dot{q}_k = r_{m{k}} rac{K_{m_k}}{R_{m{k}}} m{V_k} - m{ au_k}, \quad k = 1, \, \dots, \, n$$

can be rewritten as one, if we exclude τ !

Improved Dynamical Model (Cont'd)

The dynamical systems

$$D(q)\ddot{q}+C(q,\dot{q})\dot{q}+g(q)= au,\ q=\left[q_1,\ldots,q_n
ight]^{\mathrm{\scriptscriptstyle T}},\ oldsymbol{ au}=\left[oldsymbol{ au_1},\ldots,oldsymbol{ au_n}
ight]^{\mathrm{\scriptscriptstyle T}}$$

$$r_k^2 J_{m_k} \ddot{q}_k + r_k^2 B_k \dot{q}_k = r_k \frac{K_{m_k}}{R_k} V_k - \tau_k, \quad k = 1, \ldots, n$$

can be rewritten as one, if we exclude $\tau!$

Indeed, it is

$$M(q)\ddot{q} + C(q,\dot{q})\dot{q} + B\dot{q} + g(q) = \mathbf{u}$$

where

$$ullet$$
 $M(q)=D(q)+J$ with $J= ext{diag}\left\{r_1^2J_{m_1},\ldots,r_n^2J_{m_n}
ight\}$

•
$$B = [r_1^2 B_1, r_2^2 B_2, \dots, r_n^2 B_n]^{\mathrm{T}}$$

$$ullet \ oldsymbol{u} = [oldsymbol{u_1}, oldsymbol{u_2}, \dots, oldsymbol{u_n}]^{ \mathrm{\scriptscriptstyle T} }$$
 with $oldsymbol{u_k} = r_k rac{K_{m_k}}{R_k} oldsymbol{V_k}, \, k = 1, \dots, n$

PD-Controller Design

Given a mechanical system

$$M(q)\ddot{q} + C(q,\dot{q})\dot{q} + B\dot{q} + g(q) = \mathbf{u}$$

we are interested

- to design a controller to stabilize a particular configuration of the robot: $q=q_d$
- to analyze the closed-loop system

PD-Controller Design

Given a mechanical system

$$M(q)\ddot{q} + C(q,\dot{q})\dot{q} + B\dot{q} + g(q) = \mathbf{u}$$

we are interested

- to design a controller to stabilize a particular configuration of the robot: $q=q_d$
- to analyze the closed-loop system

Assume that B=0 and g(p)=0

PD-Controller Design

Given a mechanical system

$$M(q)\ddot{q} + C(q,\dot{q})\dot{q} + B\dot{q} + g(q) = \mathbf{u}$$

we are interested

- to design a controller to stabilize a particular configuration of the robot: $q = q_d$
- to analyze the closed-loop system

Assume that B=0 and g(p)=0

The first controller to analyze is

$$\mathbf{u} = -K_p \left(q - \mathbf{q_d} \right) - K_d \dot{q}$$

with K_p and K_d are diagonal matrices with positive elements.

To analyze the behavior of the closed loop system

$$M(q)\ddot{q} + C(q, \dot{q})\dot{q} = \mathbf{u} = -K_p (q - \mathbf{q_d}) - K_d \dot{q}$$

consider a scalar function

$$V(q,\dot{q}) = \frac{1}{2}\dot{q}^{\scriptscriptstyle T}M(q)\dot{q} + \frac{1}{2}\left(q - \mathbf{q_d}\right)^{\scriptscriptstyle T}K_p\left(q - \mathbf{q_d}\right)$$

To analyze the behavior of the closed loop system

$$M(q)\ddot{q} + C(q, \dot{q})\dot{q} = \mathbf{u} = -K_p (q - \mathbf{q_d}) - K_d \dot{q}$$

consider a scalar function

$$V(q,\dot{q}) = rac{1}{2}\dot{q}^{\scriptscriptstyle T}M(q)\dot{q} + rac{1}{2}\left(q - oldsymbol{q_d}
ight)^{\scriptscriptstyle T}K_p\left(q - oldsymbol{q_d}
ight)$$

$$\frac{d}{dt}V = \dot{q}^{\scriptscriptstyle T}M(q)\ddot{q} + \dot{q}^{\scriptscriptstyle T}\frac{d}{dt}\frac{1}{2}\left[M(q)\right]\dot{q} + \dot{q}^{\scriptscriptstyle T}K_p\left(q - \mathbf{q_d}\right)$$

To analyze the behavior of the closed loop system

$$M(q)\ddot{q} + C(q, \dot{q})\dot{q} = \mathbf{u} = -K_p (q - \mathbf{q_d}) - K_d \dot{q}$$

consider a scalar function

$$V(q,\dot{q}) = rac{1}{2}\dot{q}^{\scriptscriptstyle T}M(q)\dot{q} + rac{1}{2}\left(q - oldsymbol{q_d}
ight)^{\scriptscriptstyle T}K_p\left(q - oldsymbol{q_d}
ight)$$

$$\frac{d}{dt}V = \dot{q}^{T}M(q)\ddot{q} + \dot{q}^{T}\frac{d}{dt}\frac{1}{2}\left[M(q)\right]\dot{q} + \dot{q}^{T}K_{p}\left(q - \mathbf{q_{d}}\right)$$

$$= \dot{q}^{T}\left[\mathbf{u} - C(q, \dot{q})\dot{q}\right] + \dot{q}^{T}\frac{d}{dt}\frac{1}{2}\left[M(q)\right]\dot{q} + \dot{q}^{T}K_{p}\left(q - \mathbf{q_{d}}\right)$$

To analyze the behavior of the closed loop system

$$M(q)\ddot{q} + C(q, \dot{q})\dot{q} = \mathbf{u} = -K_p (q - \mathbf{q_d}) - K_d \dot{q}$$

consider a scalar function

$$V(q,\dot{q}) = rac{1}{2}\dot{q}^{\scriptscriptstyle T}M(q)\dot{q} + rac{1}{2}\left(q - oldsymbol{q_d}
ight)^{\scriptscriptstyle T}K_p\left(q - oldsymbol{q_d}
ight)$$

$$\frac{d}{dt}V = \dot{q}^{T}M(q)\ddot{q} + \dot{q}^{T}\frac{d}{dt}\frac{1}{2}\left[M(q)\right]\dot{q} + \dot{q}^{T}K_{p}\left(q - \mathbf{q_{d}}\right)$$

$$= \dot{q}^{T}\left[\mathbf{u} - C(q, \dot{q})\dot{q}\right] + \dot{q}^{T}\frac{d}{dt}\frac{1}{2}\left[M(q)\right]\dot{q} + \dot{q}^{T}K_{p}\left(q - \mathbf{q_{d}}\right)$$

$$= \dot{q}^{T}\left[\mathbf{u} + K_{p}\left(q - \mathbf{q_{d}}\right)\right] + \dot{q}^{T}\left\{\frac{d}{dt}\frac{1}{2}\left[M(q)\right] - C(q, \dot{q})\right\}\dot{q}$$

To analyze the behavior of the closed loop system

$$M(q)\ddot{q} + C(q, \dot{q})\dot{q} = \mathbf{u} = -K_p (q - \mathbf{q_d}) - K_d \dot{q}$$

consider a scalar function

$$V(q,\dot{q}) = rac{1}{2}\dot{q}^{\scriptscriptstyle T}M(q)\dot{q} + rac{1}{2}\left(q - oldsymbol{q_d}
ight)^{\scriptscriptstyle T}K_p\left(q - oldsymbol{q_d}
ight)$$

$$\frac{d}{dt}V = \dot{q}^{T}M(q)\ddot{q} + \dot{q}^{T}\frac{d}{dt}\frac{1}{2}\left[M(q)\right]\dot{q} + \dot{q}^{T}K_{p}\left(q - q_{d}\right)$$

$$= \dot{q}^{T}\left[\mathbf{u} - C(q,\dot{q})\dot{q}\right] + \dot{q}^{T}\frac{d}{dt}\frac{1}{2}\left[M(q)\right]\dot{q} + \dot{q}^{T}K_{p}\left(q - q_{d}\right)$$

$$= \dot{q}^{T}\left[\mathbf{u} + K_{p}\left(q - q_{d}\right)\right] + \dot{q}^{T}\left\{\frac{d}{dt}\frac{1}{2}\left[D(q) + J\right] - C(q,\dot{q})\right\}\dot{q}$$

$$= 0$$

To analyze the behavior of the closed loop system

$$M(q)\ddot{q} + C(q,\dot{q})\dot{q} = \mathbf{u} = -K_p (q - \mathbf{q_d}) - K_d \dot{q}$$

consider a scalar function

$$V(q,\dot{q}) = rac{1}{2}\dot{q}^{\scriptscriptstyle T}M(q)\dot{q} + rac{1}{2}\left(q - oldsymbol{q_d}
ight)^{\scriptscriptstyle T}K_p\left(q - oldsymbol{q_d}
ight)$$

$$\begin{split} \frac{d}{dt}V &= \dot{q}^{T}M(q)\ddot{q} + \dot{q}^{T}\frac{d}{dt}\frac{1}{2}\left[M(q)\right]\dot{q} + \dot{q}^{T}K_{p}\left(q - \mathbf{q_{d}}\right) \\ &= \dot{q}^{T}\left[\mathbf{u} - C(q, \dot{q})\dot{q}\right] + \dot{q}^{T}\frac{d}{dt}\frac{1}{2}\left[M(q)\right]\dot{q} + \dot{q}^{T}K_{p}\left(q - \mathbf{q_{d}}\right) \\ &= \dot{q}^{T}\left[\mathbf{u} + K_{p}\left(q - \mathbf{q_{d}}\right)\right] \\ &= \dot{q}^{T}\left[-K_{p}\left(q - \mathbf{q_{d}}\right) - K_{d}\dot{q} + K_{p}\left(q - \mathbf{q_{d}}\right)\right] = -\dot{q}^{T}K_{d}\dot{q} \end{split}$$

To analyze the behavior of the closed loop system

$$M(q)\ddot{q} + C(q,\dot{q})\dot{q} = \mathbf{u} = -K_p (q - \mathbf{q_d}) - K_d \dot{q}$$

consider a scalar function

$$V(q,\dot{q}) = rac{1}{2}\dot{q}^{\scriptscriptstyle T}M(q)\dot{q} + rac{1}{2}\left(q - oldsymbol{q_d}
ight)^{\scriptscriptstyle T}K_p\left(q - oldsymbol{q_d}
ight)$$

Its time-derivative along solutions of the closed-loop system is

$$rac{d}{dt}V = -\dot{q}^{\scriptscriptstyle T}K_d\dot{q} \, \leq \, 0$$

Therefore

- V is positive definite, $V(q,\dot{q})=0 \Rightarrow \{q=q_{d},\dot{q}=0\}$
- $V(q(t), \dot{q}(t))$ is monotonically decreasing!

$$\Rightarrow \exists \lim_{t \to +\infty} V(q(t), \dot{q}(t)) = V_{\infty}$$
 and $\exists \lim_{t \to +\infty} \dot{q}(t) = \dot{q}_{\infty}(t) = 0$

To analyze the behavior of the closed loop system

$$M(q)\ddot{q} + C(q, \dot{q})\dot{q} = \mathbf{u} = -K_p (q - \mathbf{q_d}) - K_d \dot{q}$$

consider a scalar function

$$V(q,\dot{q}) = rac{1}{2}\dot{q}^{\scriptscriptstyle T}M(q)\dot{q} + rac{1}{2}\left(q - oldsymbol{q_d}
ight)^{\scriptscriptstyle T}K_p\left(q - oldsymbol{q_d}
ight)$$

If we substitute this limit trajectory into the dynamics, we obtain

$$M(\mathbf{q}_{\infty}) \underbrace{\ddot{\mathbf{q}}_{\infty}}_{=0} + C(\mathbf{q}_{\infty}, \dot{\mathbf{q}}_{\infty}) \underbrace{\dot{\mathbf{q}}_{\infty}}_{=0} = -K_{p} (\mathbf{q}_{\infty} - \mathbf{q}_{d}) - K_{d} \underbrace{\dot{\mathbf{q}}_{\infty}}_{=0}$$

that is

$$0 = -K_p \left(q_{\infty} - q_d \right)$$

To analyze the behavior of the closed loop system

$$M(q)\ddot{q} + C(q, \dot{q})\dot{q} = \mathbf{u} = -K_p (q - \mathbf{q_d}) - K_d \dot{q}$$

consider a scalar function

$$V(q,\dot{q}) = rac{1}{2}\dot{q}^{\scriptscriptstyle T}M(q)\dot{q} + rac{1}{2}\left(q - oldsymbol{q_d}
ight)^{\scriptscriptstyle T}K_p\left(q - oldsymbol{q_d}
ight)$$

If we substitute this limit trajectory into the dynamics, we obtain

$$M(\mathbf{q}_{\infty})\underbrace{\ddot{\mathbf{q}}_{\infty}}_{=0} + C(\mathbf{q}_{\infty}, \dot{\mathbf{q}}_{\infty})\underbrace{\dot{\mathbf{q}}_{\infty}}_{=0} = -K_{p}(\mathbf{q}_{\infty} - \mathbf{q}_{d}) - K_{d}\underbrace{\dot{\mathbf{q}}_{\infty}}_{=0}$$

that is

$$0 = -K_p \left(q_{\infty} - q_d \right)$$

$$K_p = \operatorname{diag} \{K_{p1}, K_{p2}, \dots, K_{pn}\} > 0 \Rightarrow q_{\infty} = q_d$$

PD-Controller Design with Gravity Compensation

Given a mechanical system

$$M(q)\ddot{q} + C(q,\dot{q})\dot{q} + B\dot{q} + g(q) = \mathbf{u}$$

How to modify the controller

$$\mathbf{u} = -K_p \left(q - \mathbf{q_d} \right) - K_d \dot{q}$$

if
$$B \neq 0$$
 and $g(p) \neq 0$?

PD-Controller Design with Gravity Compensation

Given a mechanical system

$$M(q)\ddot{q} + C(q,\dot{q})\dot{q} + B\dot{q} + g(q) = \mathbf{u}$$

How to modify the controller

$$\mathbf{u} = -K_p \left(q - \mathbf{q_d} \right) - K_d \dot{q}$$

if $B \neq 0$ and $g(p) \neq 0$?

The controller

$$\mathbf{u} = -K_p \left(q - \mathbf{q_d} \right) - K_d \dot{q} + g(q)$$

is stabilizing, if K_p and K_d are diagonal matrices such that

$$K_p > 0$$
 $K_d - B > 0$

- PD-Control
- Feedback Linearization
- Robust and Adaptive Motion Control

Given a mechanical system

$$M(q)\ddot{q} + C(q,\dot{q})\dot{q} + B\dot{q} + g(q) = \mathbf{u}$$

How to define a control variable

$$\mathbf{u} = \alpha(q, \dot{q}) + \beta(q, \dot{q})\mathbf{v}$$

so that the closed loop system is:

• linear? I.e. is equivalent to

$$\dot{x} = Ax + B\mathbf{v}$$

• linear and stabilizable? I.e. the pair (A,B) is stabilizable

Given a mechanical system

$$M(q)\ddot{q} + C(q,\dot{q})\dot{q} + B\dot{q} + g(q) = \mathbf{u}$$

How to define a control variable

$$\mathbf{u} = \alpha(q, \dot{q}) + \beta(q, \dot{q})\mathbf{v}$$

so that the closed loop system is:

• linear? I.e. is equivalent to

$$\dot{x} = Ax + B\mathbf{v}$$

• linear and stabilizable? I.e. the pair (A,B) is stabilizable

What if

$$\mathbf{u} = M(q)\mathbf{v} + C(q, \dot{q})\dot{q} + B\dot{q} + g(q)$$
?

Given a mechanical system

$$M(q)\ddot{q} + C(q,\dot{q})\dot{q} + B\dot{q} + g(q) = \mathbf{u}$$

How to define a control variable

$$\mathbf{u} = \alpha(q, \dot{q}) + \beta(q, \dot{q})\mathbf{v}$$

so that the closed loop system is:

• linear? I.e. is equivalent to

$$\dot{x} = Ax + B\mathbf{v}$$

• linear and stabilizable? I.e. the pair (A, B) is stabilizable

Then

$$M(q)\ddot{q} = M(q)\mathbf{v} \quad \Rightarrow \quad \ddot{q} = \mathbf{v}$$

Given a mechanical system

$$M(q)\ddot{q} + C(q,\dot{q})\dot{q} + B\dot{q} + g(q) = \mathbf{u}$$

How to define a control variable

$$\mathbf{u} = \alpha(q, \dot{q}) + \beta(q, \dot{q})\mathbf{v}$$

so that the closed loop system is:

• linear? I.e. is equivalent to

$$\dot{x} = Ax + B\mathbf{v}$$

• linear and stabilizable? I.e. the pair (A,B) is stabilizable

$$\ddot{q} = oldsymbol{v} \qquad \Rightarrow \qquad \dot{x} = \left[egin{array}{ccc|c} 0_n & I_n \ 0_n & 0_n \end{array}
ight] x + \left[egin{array}{ccc|c} 0_n \ I_n \end{array}
ight] oldsymbol{v}, \quad x = [q^{\scriptscriptstyle T}, \dot{q}^{\scriptscriptstyle T}]^{\scriptscriptstyle T}$$

Given a mechanical system

$$M(q)\ddot{q} + C(q,\dot{q})\dot{q} + B\dot{q} + g(q) = \mathbf{u}$$

and the desired trajectory $q_d=q_d(t)$, introduce the controller

$$\mathbf{u} = M(q)\mathbf{v} + C(q,\dot{q})\dot{q} + B\dot{q} + g(q)$$

$$\mathbf{v} = \ddot{q}_d(t) - K_p(q - q_d(t)) - K_d(\dot{q} - \dot{q}_d(t))$$

Given a mechanical system

$$M(q)\ddot{q} + C(q,\dot{q})\dot{q} + B\dot{q} + g(q) = \mathbf{u}$$

and the desired trajectory $q_d = q_d(t)$, introduce the controller

$$\mathbf{u} = M(q)\mathbf{v} + C(q,\dot{q})\dot{q} + B\dot{q} + g(q)$$

$$\mathbf{v} = \ddot{q}_d(t) - K_p(q - q_d(t)) - K_d(\dot{q} - \dot{q}_d(t))$$

Then the closed loop system is

$$\ddot{q} = v = \ddot{q}_d(t) - K_p(q - q_d(t)) - K_d(\dot{q} - \dot{q}_d(t))$$

Given a mechanical system

$$M(q)\ddot{q} + C(q,\dot{q})\dot{q} + B\dot{q} + g(q) = \mathbf{u}$$

and the desired trajectory $q_d = q_d(t)$, introduce the controller

$$\mathbf{u} = M(q)\mathbf{v} + C(q,\dot{q})\dot{q} + B\dot{q} + g(q)$$

$$\mathbf{v} = \ddot{q}_d(t) - K_p(q - q_d(t)) - K_d(\dot{q} - \dot{q}_d(t))$$

Then the closed loop system is

$$\ddot{q} = \mathbf{v} = \ddot{q}_d(t) - K_p(q - q_d(t)) - K_d(\dot{q} - \dot{q}_d(t))$$

It can be rewritten in error-variables as

$$\ddot{e} + K_d \dot{e} + K_p e = 0, \qquad e = q - q_d(t)$$

- PD-Control
- Feedback Linearization
- Robust and Adaptive Motion Control

Given a mechanical system

$$M(q)\ddot{q} + C(q,\dot{q})\dot{q} + g(q) = \mathbf{u}$$

and the desired trajectory $q_d=q_d(t)$, the controller

$$\mathbf{u} = M(q)\mathbf{v} + C(q,\dot{q})\dot{q} + g(q)$$

$$\mathbf{v} = \ddot{q}_d(t) - K_p(q - q_d(t)) - K_d(\dot{q} - \dot{q}_d(t))$$

cannot be implemented!

Given a mechanical system

$$M(q)\ddot{q} + C(q,\dot{q})\dot{q} + g(q) = \mathbf{u}$$

and the desired trajectory $q_d = q_d(t)$, the controller

$$u = M(q)v + C(q,\dot{q})\dot{q} + g(q)$$

$$\mathbf{v} = \ddot{q}_d(t) - K_p(q - q_d(t)) - K_d(\dot{q} - \dot{q}_d(t))$$

cannot be implemented!

The approximation might be used

$$\mathbf{u} = [M(q) + \triangle M] \mathbf{v} + [C(q, \dot{q}) + \triangle C] \dot{q} + [g(q) + \triangle g]$$

$$\mathbf{v} = \ddot{q}_d(t) - K_p(q - q_d(t)) - K_d(\dot{q} - \dot{q}_d(t))$$

The uncertainty in parameters of controller

$$\mathbf{u} = [M(q) + \triangle M] \mathbf{v} + [C(q, \dot{q}) + \triangle C] \dot{q} + [g(q) + \triangle g]$$

$$\mathbf{v} = \ddot{q}_d(t) - K_p(q - q_d(t)) - K_d(\dot{q} - \dot{q}_d(t))$$

motivates two approaches

Robust Control:

Design K_p , K_d and $q_d(t)$ such that the error signal

$$e(t) = q(t) - q_d(t) \approx 0 \quad \forall \{ \triangle M, \triangle C, \triangle g \} \in \mathcal{W}$$

The uncertainty in parameters of controller

$$\mathbf{u} = [M(q) + \triangle M] \mathbf{v} + [C(q, \dot{q}) + \triangle C] \dot{q} + [g(q) + \triangle g]$$

$$\mathbf{v} = \ddot{q}_d(t) - K_p(q - q_d(t)) - K_d(\dot{q} - \dot{q}_d(t))$$

motivates two approaches

Robust Control:

Design K_p , K_d and $q_d(t)$ such that the error signal

$$e(t) = q(t) - q_d(t) \approx 0 \quad \forall \{ \triangle M, \triangle C, \triangle g \} \in \mathcal{W}$$

Adaptive Control: Improve estimates for

$$M(q), \quad C(q,\dot{q}), \quad g(q)$$

in the course of regulating the system

$$M(q)\ddot{q} + C(q,\dot{q})\dot{q} + g(q) = \mathbf{u}$$

$$\mathbf{u} = [M(q) + \triangle M] \mathbf{v} + [C(q, \dot{q}) + \triangle C] \dot{q} + [g(q) + \triangle g]$$

$$\mathbf{v} = \ddot{q}_d(t) - K_p(q - q_d(t)) - K_d(\dot{q} - \dot{q}_d(t))$$

$$M(q)\ddot{q} + C(q,\dot{q})\dot{q} + g(q) = \mathbf{u}$$
 $\mathbf{u} = [M(q) + \Delta M]\mathbf{v} + [C(q,\dot{q}) + \Delta C]\dot{q} + [g(q) + \Delta g]$
 $\mathbf{v} = \ddot{q}_d(t) - K_p(q - q_d(t)) - K_d(\dot{q} - \dot{q}_d(t))$

$$egin{aligned} M(q)\ddot{q}+C(q,\dot{q})\dot{q}+g(q) &= \\ &= \left[M(q)+\triangle M\right] oldsymbol{v} + \left[C(q,\dot{q})+\triangle C\right]\dot{q} + \left[g(q)+\triangle g\right] \end{aligned}$$

$$M(q)\ddot{q} + C(q,\dot{q})\dot{q} + g(q) = \mathbf{u}$$
 $\mathbf{u} = [M(q) + \triangle M]\mathbf{v} + [C(q,\dot{q}) + \triangle C]\dot{q} + [g(q) + \triangle g]$
 $\mathbf{v} = \ddot{q}_d(t) - K_p(q - q_d(t)) - K_d(\dot{q} - \dot{q}_d(t))$

$$M(q)\ddot{q} + C(q,\dot{q})\dot{q} + g(q) =$$

$$= [M(q) + \Delta M] \mathbf{v} + [C(q,\dot{q}) + \Delta C] \dot{q} + [g(q) + \Delta g]$$

$$\Rightarrow M(q)\ddot{q} = [M(q) + \Delta M] \mathbf{v} + \Delta C \dot{q} + \Delta g$$

$$M(q)\ddot{q} + C(q,\dot{q})\dot{q} + g(q) = \mathbf{u}$$
 $\mathbf{u} = [M(q) + \triangle M]\mathbf{v} + [C(q,\dot{q}) + \triangle C]\dot{q} + [g(q) + \triangle g]$
 $\mathbf{v} = \ddot{q}_d(t) - K_p(q - q_d(t)) - K_d(\dot{q} - \dot{q}_d(t))$

$$M(q)\ddot{q} + C(q, \dot{q})\dot{q} + g(q) =$$

$$= [M(q) + \triangle M] \mathbf{v} + [C(q, \dot{q}) + \triangle C] \dot{q} + [g(q) + \triangle g]$$

$$\Rightarrow M(q)\ddot{q} = [M(q) + \triangle M] \mathbf{v} + \triangle C\dot{q} + \triangle g$$

$$\Rightarrow \ddot{q} = M(q)^{-1} [M(q) + \triangle M] \mathbf{v} + M(q)^{-1} [\triangle C\dot{q} + \triangle g]$$

$$M(q)\ddot{q} + C(q,\dot{q})\dot{q} + g(q) = \mathbf{u}$$
 $\mathbf{u} = [M(q) + \triangle M]\mathbf{v} + [C(q,\dot{q}) + \triangle C]\dot{q} + [g(q) + \triangle g]$
 $\mathbf{v} = \ddot{q}_d(t) - K_p(q - q_d(t)) - K_d(\dot{q} - \dot{q}_d(t))$

$$M(q)\ddot{q} + C(q, \dot{q})\dot{q} + g(q) =$$

$$= [M(q) + \triangle M] \mathbf{v} + [C(q, \dot{q}) + \triangle C] \dot{q} + [g(q) + \triangle g]$$

$$\Rightarrow M(q)\ddot{q} = [M(q) + \triangle M] \mathbf{v} + \triangle C \dot{q} + \triangle g$$

$$\ddot{q} = M^{-1} [M + \triangle M] \mathbf{v} + M^{-1} [\triangle C \dot{q} + \triangle g] \pm M^{-1} M \mathbf{v}$$

$$M(q)\ddot{q} + C(q,\dot{q})\dot{q} + g(q) = \mathbf{u}$$
 $\mathbf{u} = [M(q) + \Delta M]\mathbf{v} + [C(q,\dot{q}) + \Delta C]\dot{q} + [g(q) + \Delta g]$
 $\mathbf{v} = \ddot{q}_d(t) - K_p(q - q_d(t)) - K_d(\dot{q} - \dot{q}_d(t))$

$$M(q)\ddot{q} + C(q, \dot{q})\dot{q} + g(q) =$$

$$= [M(q) + \Delta M] \mathbf{v} + [C(q, \dot{q}) + \Delta C] \dot{q} + [g(q) + \Delta g]$$

$$\Rightarrow M(q)\ddot{q} = [M(q) + \Delta M] \mathbf{v} + \Delta C \dot{q} + \Delta g$$

$$\ddot{q} = M^{-1} [M + \Delta M] \mathbf{v} + M^{-1} [\Delta C \dot{q} + \Delta g] \pm M^{-1} M \mathbf{v}$$

$$\Rightarrow \ddot{q} = \mathbf{v} + M^{-1}(q) \left[\triangle M(q) \mathbf{v} + \triangle C(q, \dot{q}) \dot{q} + \triangle g(q) \right]$$

$$M(q)\ddot{q} + C(q,\dot{q})\dot{q} + g(q) = \mathbf{u}$$
 $\mathbf{u} = [M(q) + \triangle M]\mathbf{v} + [C(q,\dot{q}) + \triangle C]\dot{q} + [g(q) + \triangle g]$
 $\mathbf{v} = \ddot{q}_d(t) - K_p(q - q_d(t)) - K_d(\dot{q} - \dot{q}_d(t))$

$$M(q)\ddot{q} + C(q, \dot{q})\dot{q} + g(q) =$$

$$= [M(q) + \triangle M] \mathbf{v} + [C(q, \dot{q}) + \triangle C] \dot{q} + [g(q) + \triangle g]$$

$$\Rightarrow M(q)\ddot{q} = [M(q) + \triangle M] \mathbf{v} + \triangle C \dot{q} + \triangle g$$

$$\ddot{q} = M^{-1} [M + \triangle M] \mathbf{v} + M^{-1} [\triangle C \dot{q} + \triangle g] \pm M^{-1} M \mathbf{v}$$

$$\ddot{q} = \mathbf{v} + \eta(q, \dot{q}, \mathbf{v})$$

$$M(q)\ddot{q} + C(q,\dot{q})\dot{q} + g(q) = m{u}$$
 $m{u} = [M(q) + \Delta M] \, m{v} + [C(q,\dot{q}) + \Delta C] \, \dot{q} + [g(q) + \Delta g]$
 $m{-v} = \ddot{q}_d(t) - K_p(q - q_d(t)) - K_d(\dot{q} - \dot{q}_d(t))$

$$M(q)\ddot{q} + C(q,\dot{q})\dot{q} + g(q) =$$

$$= [M(q) + \triangle M] \mathbf{v} + [C(q,\dot{q}) + \triangle C] \dot{q} + [g(q) + \triangle g]$$

$$\Rightarrow M(q)\ddot{q} = [M(q) + \triangle M] \mathbf{v} + \triangle C\dot{q} + \triangle g$$

$$\ddot{q} = M^{-1} [M + \triangle M] \mathbf{v} + M^{-1} [\triangle C\dot{q} + \triangle g] \pm M^{-1} M \mathbf{v}$$

$$\ddot{a} = \ddot{\mathbf{v}} + n(q,\dot{q},\dot{q},\dot{\mathbf{v}})$$

$$M(q)\ddot{q} + C(q,\dot{q})\dot{q} + g(q) = \mathbf{u}$$

$$\mathbf{u} = [M(q) + \triangle M] \mathbf{v} + [C(q, \dot{q}) + \triangle C] \dot{q} + [g(q) + \triangle g]$$

$$\mathbf{v} = \ddot{q}_d(t) - K_p(q - q_d(t)) - K_d(\dot{q} - \dot{q}_d(t)) + \mathbf{w}$$

Given a trajectory $q = q_d(t)$, consider the closed loop system

$$M(q)\ddot{q} + C(q,\dot{q})\dot{q} + g(q) = \mathbf{u}$$

$$\mathbf{u} = [M(q) + \triangle M] \mathbf{v} + [C(q, \dot{q}) + \triangle C] \dot{q} + [g(q) + \triangle g]$$

$$\mathbf{v} = \ddot{q}_d(t) - K_p(q - q_d(t)) - K_d(\dot{q} - \dot{q}_d(t)) + \mathbf{w}$$

It can be rewritten as

$$\ddot{q} = \mathbf{v} + \eta(q, \dot{q}, \mathbf{v})$$

or

$$(\ddot{q} - \ddot{q}_d) + K_d (\dot{q} - \dot{q}_d) + K_p (q - q_d) = \mathbf{w} + \eta(q, \dot{q}, \mathbf{v})$$

Given a trajectory $q = q_d(t)$, consider the closed loop system

$$M(q)\ddot{q} + C(q,\dot{q})\dot{q} + g(q) = \mathbf{u}$$
 $\mathbf{u} = [M(q) + \Delta M]\mathbf{v} + [C(q,\dot{q}) + \Delta C]\dot{q} + [g(q) + \Delta g]$
 $\mathbf{v} = \ddot{q}_d(t) - K_p(q - q_d(t)) - K_d(\dot{q} - \dot{q}_d(t)) + \mathbf{w}$

It can be rewritten as

$$\ddot{q} = \mathbf{v} + \eta(q, \dot{q}, \mathbf{v})$$

or

$$(\ddot{q} - \ddot{q}_d) + K_d (\dot{q} - \dot{q}_d) + K_p (q - q_d) = \mathbf{w} + \eta(q, \dot{q}, \mathbf{v})$$

or

$$rac{d}{dt}e = \left[egin{array}{cc} 0_n & I_n \ -K_p & -K_d \end{array}
ight]e + \left[egin{array}{cc} 0_n \ I_n \end{array}
ight](oldsymbol{w} + \eta)\,,\quad e = \left[egin{array}{cc} (q-q_d) \ (\dot{q}-\dot{q}_d) \end{array}
ight]$$

Given a trajectory $q = q_d(t)$, consider the closed loop system

$$M(q)\ddot{q} + C(q,\dot{q})\dot{q} + g(q) = \mathbf{u}$$
 $\mathbf{u} = [M(q) + \Delta M]\mathbf{v} + [C(q,\dot{q}) + \Delta C]\dot{q} + [g(q) + \Delta g]$
 $\mathbf{v} = \ddot{q}_d(t) - K_p(q - q_d(t)) - K_d(\dot{q} - \dot{q}_d(t)) + \mathbf{w}$

It can be rewritten as

$$\ddot{q} = \mathbf{v} + \eta(q, \dot{q}, \mathbf{v})$$

or

$$(\ddot{q} - \ddot{q}_d) + K_d (\dot{q} - \dot{q}_d) + K_p (q - q_d) = \mathbf{w} + \eta(q, \dot{q}, \mathbf{v})$$

or

$$rac{d}{dt}e = Ae + B\left(oldsymbol{w} + \eta
ight), \quad e = \left[egin{array}{c} (q - q_d) \ (\dot{q} - \dot{q}_d) \end{array}
ight]$$

To continue with design of \boldsymbol{w} for the system

$$rac{d}{dt}e = Ae + B\left[oldsymbol{w} + \eta(e, oldsymbol{w})
ight]$$

we need to impose some assumptions on $\eta(\cdot)$, namely

$$\|\eta(e, \mathbf{w})\| \le \alpha \|\mathbf{w}\| + \gamma_1 \|e\| + \gamma_2 \|e\|^2 + \gamma_3, \quad \alpha < 1$$

To continue with design of \boldsymbol{w} for the system

$$rac{d}{dt}e = Ae + B\left[oldsymbol{w} + \eta(e, oldsymbol{w})
ight]$$

we need to impose some assumptions on $\eta(\cdot)$, namely

$$\|\eta(e, \mathbf{w})\| \le \alpha \|\mathbf{w}\| + \gamma_1 \|e\| + \gamma_2 \|e\|^2 + \gamma_3, \quad \alpha < 1$$

Matrix A is stable, therefore $\forall\,Q>0,\,\exists\,P=P^{\scriptscriptstyle T}>0$ such that

$$A^{\scriptscriptstyle T}P + PA = -Q$$

To continue with design of \boldsymbol{w} for the system

$$rac{d}{dt}e = Ae + B\left[oldsymbol{w} + \eta(e, oldsymbol{w})
ight]$$

we need to impose some assumptions on $\eta(\cdot)$, namely

$$\|\eta(e, \mathbf{w})\| \le \alpha \|\mathbf{w}\| + \gamma_1 \|e\| + \gamma_2 \|e\|^2 + \gamma_3, \quad \alpha < 1$$

Matrix A is stable, therefore $\forall\, Q>0,\,\exists\, P=P^{\scriptscriptstyle T}>0$ such that

$$A^{\scriptscriptstyle T}P + PA = -Q$$

Consider a Lyapunov function candidate as $V = e^{T}Pe$, then

$$egin{array}{ll} rac{d}{dt}V &= rac{d}{dt}e^{\mathrm{T}}Pe + e^{\mathrm{T}}Prac{d}{dt}e \ &= e^{\mathrm{T}}(A^{\mathrm{T}}P + PA)e + 2e^{\mathrm{T}}PB\left[oldsymbol{w} + \eta(e, oldsymbol{w})
ight] \end{array}$$

To continue with design of \boldsymbol{w} for the system

$$rac{d}{dt}e = Ae + B\left[oldsymbol{w} + \eta(e, oldsymbol{w})
ight]$$

we need to impose some assumptions on $\eta(\cdot)$, namely

$$\|\eta(e, \mathbf{w})\| \le \alpha \|\mathbf{w}\| + \gamma_1 \|e\| + \gamma_2 \|e\|^2 + \gamma_3, \quad \alpha < 1$$

Matrix A is stable, therefore $\forall\,Q>0,\,\exists\,P=P^{\scriptscriptstyle T}>0$ such that

$$A^{\scriptscriptstyle T}P + PA = -Q$$

Consider a Lyapunov function candidate as $V = e^{T}Pe$, then

$$\frac{d}{dt}V = -e^{\scriptscriptstyle T}Qe + 2e^{\scriptscriptstyle T}PB\left[{\color{red} {\color{blue} {w}}} + \eta(e, {\color{red} {\color{blue} {w}}})
ight]$$
 $\leq 0 \leftarrow \text{How to achieve this by choosing } {\color{red} {\color{blue} {w}}}?$

To continue with design of \boldsymbol{w} for the system

$$rac{d}{dt}e = Ae + B\left[\mathbf{w} + \eta(e, \mathbf{w})
ight]$$

we need to impose some assumptions on $\eta(\cdot)$, namely

$$\|\eta(e, \mathbf{w})\| \le \alpha \|\mathbf{w}\| + \gamma_1 \|e\| + \gamma_2 \|e\|^2 + \gamma_3, \quad \alpha < 1$$

Let us look at the second term

$$\frac{d}{dt}V = -e^{T}Qe + 2e^{T}PB\left[\mathbf{w} + \eta(e, \mathbf{w})\right]$$

when \boldsymbol{w} has the form

$$oldsymbol{w} = - oldsymbol{
ho}(\cdot) rac{z}{\sqrt{z^{\scriptscriptstyle T} z}}, \qquad z = B^{\scriptscriptstyle T} P e \ , \qquad oldsymbol{
ho} \ \ ext{is a function to choose}$$

To continue with design of \boldsymbol{w} for the system

$$rac{d}{dt}e = Ae + B\left[oldsymbol{w} + \eta(e, oldsymbol{w})
ight]$$

we need to impose some assumptions on $\eta(\cdot)$, namely

$$\|\eta(e, \mathbf{w})\| \le \alpha \|\mathbf{w}\| + \gamma_1 \|e\| + \gamma_2 \|e\|^2 + \gamma_3, \quad \alpha < 1$$

Let us look at the second term

$$\frac{d}{dt}V = -e^{T}Qe + 2e^{T}PB\left[\mathbf{w} + \eta(e, \mathbf{w})\right]$$

when \boldsymbol{w} has the form

$$oldsymbol{w} = - oldsymbol{
ho}(\cdot) rac{z}{\sqrt{z^{\scriptscriptstyle T} z}}, \qquad z = B^{\scriptscriptstyle T} P e \ , \qquad oldsymbol{
ho} \ \ ext{is a function to choose}$$

$$z^{T}\left(-
horac{z}{\sqrt{z^{T}z}}+\eta
ight)\leq -
ho\|z\|+\|z\|\|\eta\|=\|z\|\left(-
ho+\|\eta\|
ight)$$

To sum up, we search for a scalar function $\rho(\cdot)$ such that

$$(-\boldsymbol{\rho} + \|\boldsymbol{\eta}\|) \leq 0 \iff \|\boldsymbol{\eta}\| \leq \boldsymbol{\rho}$$

and

$$\|\eta(e, \mathbf{w})\| \le \alpha \|\mathbf{w}\| + \gamma_1 \|e\| + \gamma_2 \|e\|^2 + \gamma_3, \quad \alpha < 1$$

with

$$oldsymbol{w} = -oldsymbol{
ho}(\cdot) rac{z}{\sqrt{z^{{\scriptscriptstyle T}} z}}$$

To sum up, we search for a scalar function $\rho(\cdot)$ such that

$$(-\boldsymbol{\rho} + \|\boldsymbol{\eta}\|) \leq 0 \iff \|\boldsymbol{\eta}\| \leq \boldsymbol{\rho}$$

and

$$\|\eta(e, \mathbf{w})\| \le \alpha \|\mathbf{w}\| + \gamma_1 \|e\| + \gamma_2 \|e\|^2 + \gamma_3, \quad \alpha < 1$$

with

$$oldsymbol{w} = -oldsymbol{
ho}(\cdot) rac{z}{\sqrt{z^{{\scriptscriptstyle T}} z}}$$

These two inequalities imply the next one

$$\alpha \left\| \rho(\cdot) \frac{z}{\sqrt{z^T z}} \right\| + \gamma_1 \left\| e \right\| + \gamma_2 \left\| e \right\|^2 + \gamma_3 \le \rho(\cdot)$$

To sum up, we search for a scalar function $\rho(\cdot)$ such that

$$(-\boldsymbol{\rho} + \|\boldsymbol{\eta}\|) \leq 0 \iff \|\boldsymbol{\eta}\| \leq \boldsymbol{\rho}$$

and

$$\|\eta(e, \mathbf{w})\| \le \alpha \|\mathbf{w}\| + \gamma_1 \|e\| + \gamma_2 \|e\|^2 + \gamma_3, \quad \alpha < 1$$

with

$$oldsymbol{w} = -oldsymbol{
ho}(\cdot) rac{z}{\sqrt{z^{{\scriptscriptstyle T}} z}}$$

These two inequalities imply the next one

$$\alpha \rho(\cdot) + \gamma_1 \|e\| + \gamma_2 \|e\|^2 + \gamma_3 \le \rho(\cdot)$$

To sum up, we search for a scalar function $\rho(\cdot)$ such that

$$(-\boldsymbol{\rho} + \|\boldsymbol{\eta}\|) \leq 0 \iff \|\boldsymbol{\eta}\| \leq \boldsymbol{\rho}$$

and

$$\|\eta(e, \mathbf{w})\| \le \alpha \|\mathbf{w}\| + \gamma_1 \|e\| + \gamma_2 \|e\|^2 + \gamma_3, \quad \alpha < 1$$

with

$$oldsymbol{w} = -oldsymbol{
ho}(\cdot) rac{z}{\sqrt{z^{{\scriptscriptstyle T}} z}}$$

These two inequalities imply the next one

$$\alpha \rho(\cdot) + \gamma_1 \|e\| + \gamma_2 \|e\|^2 + \gamma_3 \le \rho(\cdot)$$

$$oldsymbol{
ho(e)} \geq rac{1}{1-lpha} \left[\gamma_1 \left\| e
ight\| + \gamma_2 \left\| e
ight\|^2 + \gamma_3
ight]$$

Final Form of the Controller

Given a trajectory $q = q_d(t)$, consider the closed loop system

$$egin{array}{ll} M(q)\ddot{q}+C(q,\dot{q})\dot{q}+g(q)=oldsymbol{u} \ oldsymbol{u} &=& \left[M(q)+igtriangle M
ight]oldsymbol{v}+\left[C(q,\dot{q})+igtriangle C
ight]\dot{q}+\left[g(q)+igtriangle g
ight] \ oldsymbol{v} &=& \ddot{q}_d(t)-K_p(q-q_d(t))-K_d(\dot{q}-\dot{q}_d(t))+oldsymbol{w} \ oldsymbol{v} &=& egin{cases} -oldsymbol{
ho}(e)rac{z}{\sqrt{z^Tz}}, & ext{if } z=B^TPe \neq 0 \ 0, & ext{if } z=B^TPe=0 \end{cases} \end{array}$$

where $\rho(\cdot)$ is any function that satisfies to inequality

$$ho(e) \geq rac{1}{1-lpha} \left[\gamma_1 \left\| e
ight\| + \gamma_2 \left\| e
ight\|^2 + \gamma_3
ight]$$

where constants α , γ_1 - γ_2 are from the inequality

$$\|\eta(\cdot)\| \le \alpha \|\mathbf{w}\| + \gamma_1 \|e\| + \gamma_2 \|e\|^2 + \gamma_3, \quad \alpha < 1$$

and

$$\eta(q,\dot{q},v) = M^{-1} \left[riangle M oldsymbol{v} + riangle C \dot{q} + riangle g
ight], \quad e = \left[egin{array}{c} (q-q_d) \ (\dot{q}-\dot{q}_d) \end{array}
ight]$$