

## EE5114 Continuous Assessment 1

Main steps	Constraints and additional requirements	Questions to be answered
<ul style="list-style-type: none"> <li>Install <b>MATLAB</b> on your own computer and self-study MATLAB coding basics.</li> <li>Place the MATLAB script file '<b>EE5114_CA1.m</b>' and data file '<b>EE5114_CA1.mat</b>' under the same folder and run the default script.</li> <li>By reading and analyzing the original script and default plots, understand the physical meanings of the data.</li> <li>Add your own MATLAB codes to <b>convert GPS Longitude, Latitude and Altitude to NED x, y, z positions</b> and plot them against time.</li> <li>Write your own MATLAB scripts to <b>implement an EKF</b> to estimate (better) NED positions (in m) and NED velocities (in m/s). Plot them against time.</li> <li><b>Write a report</b> to answer all questions in the 3<sup>rd</sup> column of this table. <b>Provide clear explanations and relevant data plots to support your answers.</b></li> <li>Submit both the report and MATLAB codes to Canvas, zip them to a single file named as '<b>EE5114_CA1_yourmatriculationnumber.zip</b>'</li> </ul>	<ul style="list-style-type: none"> <li>Implement the EKF for data between <math>t_{min}</math> and <math>t_{max}</math> only, where <math>t_{min}</math> is set to be 10 minutes before the 2<sup>nd</sup> time the UAV took off and <math>t_{max}</math> is set to be 5 min after the 2<sup>nd</sup> time the UAV landed.</li> <li>(Non-zero) bias should be considered for accelerometer reading and to be estimated by EKF over time.</li> <li>Codes <ul style="list-style-type: none"> <li>Fully functional codes without run-time errors.</li> <li>Codes should be as clean and modular as possible, and with meaningful comments.</li> <li>Codes should produce all data plots that have been used in your report.</li> </ul> </li> <li>Report <ul style="list-style-type: none"> <li><b>Cover page</b> with report title, your name and matriculation number</li> <li>Main content fonts: <b>Calibri, 11</b></li> <li><b>Not more than 20 pages</b></li> </ul> </li> <li>Deadline for submission: <b>End of mid-term reading week</b>. Check Canvas student submission folder's setting.</li> </ul>	<ol style="list-style-type: none"> <li>By understanding the original codes and observing the default plots of this set of data, can you deduce how many times the UAV had taken off and landed?</li> <li>When did the UAV take off and land for the 2<sup>nd</sup> time? Please provide relevant figures to support your explanations.</li> <li>List the formulas you have used to convert GPS coordinates to NED positions and attach your corresponding codes for this step.</li> <li>List the formulas you have used for the implementation of Kalman filter and attach your corresponding codes for this step.</li> <li>Please explain the physical meanings of your state variables used in your EKF and which coordinate system they are defined in.</li> <li>Note that GPS update rate is slower than IMU update rate for this set of data. Please explain how you have implemented your code to address this practical issue.</li> <li>Please explain what values you have chosen to initialize the state variables and the state covariance matrix P. Why these values?</li> <li>Please explain your choice of Q and R matrices for your EKF implementation.</li> <li>Have the accelerometer bias values converged in the end? What values did they converge on?</li> <li>If considering accelerometer bias is not a constant, but gradually drifts over time, how will you modify the implementation of EKF to address this issue?</li> </ol>