

## Midterm Review 8.09

### Hamilton's Principle

$$\delta \int_1^2 dt L(q, \dot{q}, t) = 0$$

$$L = T - V$$

↑ kinetic energy    ↑ ptal energy (forces from V)

$$\frac{d}{dt} \frac{\partial L}{\partial \dot{q}_j} - \frac{\partial L}{\partial q_j} = 0$$

(no constraints yet  $q_j$  indep.)

$$p_i = \frac{\partial L}{\partial \dot{q}_i}$$

eg. friction force  $\vec{F}_i = -h_i(v_i) \frac{\vec{v}_i}{v_i}$ ,  $\vec{v}_i = \dot{\vec{r}}_i$

$$R_j = -\frac{\partial F}{\partial q_j}, \quad F = \sum_i \int_0^{v_i} dv_i' h_i(v_i')$$

### D'Alembert's Principle

constraint forces do

no virtual work

$$\sum_i (\vec{P}_i - \vec{F}_i) \cdot \delta \vec{r}_i = 0$$

↑ virtual displacements  
[inf, satisfy constraints, fixed t]

$$\frac{d}{dt} \frac{\partial L}{\partial \dot{q}_j} - \frac{\partial L}{\partial q_j} = R_j$$

↑ generalized forces not from V

### L with holonomic Constraints

$$f_\alpha(q_1, \dots, q_n, t) = 0$$

- If we don't care about constraint forces use  $K$  eqns  $f_\alpha = 0$  to find  $n-K$  indep coords  $q_j$

$$\frac{d}{dt} \frac{\partial L}{\partial \dot{q}_j} - \frac{\partial L}{\partial q_j} = 0 \quad n-K \text{ eqns}$$

- If we do care, then use Lagrange Mult.  $\lambda_\alpha$  for those constraints

$$\frac{d}{dt} \frac{\partial L}{\partial \dot{q}_j} - \frac{\partial L}{\partial q_j} = \sum_\alpha \lambda_\alpha \frac{\partial f_\alpha}{\partial q_j}$$

$$f_\alpha(q, t) = 0$$

↑ Generalized Forces of Constraint

$n+K$  eqns for  $n+K$  vars

### L with Semi-holonomic Constraints

$$g_\beta = \sum_j a_{\beta j}(q, t) \dot{q}_j + a_{\beta t}(q, t) = 0$$

$$\frac{d}{dt} \frac{\partial L}{\partial \dot{q}_j} - \frac{\partial L}{\partial q_j} = \sum_\beta \lambda_\beta \frac{\partial g_\beta}{\partial q_j}$$



eg. constraints to surfaces

eg. rolling constraints

### Hamilton Eqns

$$H(q, p, t) = \dot{q}_i p_i - L(q, \dot{q}, t)$$

$$\dot{q}_i = \frac{\partial H}{\partial p_i}$$

$$, \quad \dot{p}_i = -\frac{\partial H}{\partial q_i}$$

[2n 1st order eqns]

[q, p equal footing]

### Transformations

• Any Point Trnsfm  $Q_i = Q_i(q, t)$  gives  $L'(Q, \dot{Q}, t) = L(q, \dot{q}, t)$

with  $\frac{d}{dt} \frac{\partial L'}{\partial \dot{Q}_i} - \frac{\partial L'}{\partial Q_i} = 0$  (same form)

• Transformation is canonical  $Q_i = Q_i(q, p, t)$ ,  $P_i = P_i(q, p, t)$   
if there is a  $K(Q, P, t)$  so that

$$\dot{Q}_i = \frac{\partial K}{\partial P_i}, \quad \dot{P}_i = -\frac{\partial K}{\partial Q_i}$$

call  $(q_i, p_i)$ ,  $(Q_i, P_i)$  canonical coordinates

### Cyclic Coords

no  $q_i$  in  $L$  or  $H \Rightarrow \dot{p}_i = 0$   
 $p_i = \text{constant}$

eg. no force on CM coord  $\vec{R} = \frac{1}{M} \sum_i m_i \vec{r}_i = \frac{1}{M} \int dV \rho \vec{r}$   
conj. mom.  $\vec{P} = \text{const.}$

eg. no  $\phi$  dep.,  $p_\phi = \text{ang. mom} = \text{const.}$

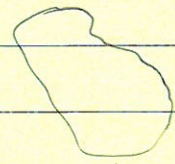
eg. no  $t$  dep.  $\frac{dH}{dt} = \frac{\partial H}{\partial t} = 0$ ,  $H = \text{const}$

may or may not have  $H = E = T + V$

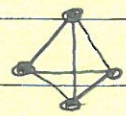


Rigid Bodies

continuous



discrete



6 coords:

3 translations

3 rotations  $(\theta, \phi, \psi)$

Euler Angles

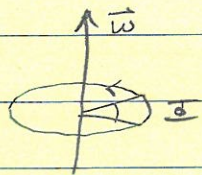
↑ about fixed pt

$$\vec{r}' = U_\psi U_\phi U_\theta \vec{r} = U \vec{r}$$

$$\begin{pmatrix} \cos\theta & \sin\theta & 0 \\ -\sin\theta & \cos\theta & 0 \\ 0 & 0 & 1 \end{pmatrix}$$

$$U^T U = \mathbb{I} \quad \text{orthogonal}$$

General Displacement with one point fixed is instantaneous rotation about some axis



defines angular velocity  $\vec{\omega}(t)$

$$|\vec{\omega}| = \frac{d\theta}{dt}$$

Inertial & Body Axes

$$\left(\frac{d}{dt}\right)_{\text{inertial}} = \left(\frac{d}{dt}\right)_{\text{body}} + \vec{\omega} \times$$

$$\left(\frac{d\vec{r}}{dt}\right)_{\text{inertial}} = \left(\frac{d\vec{r}}{dt}\right)_{\text{body}} + \vec{\omega} \times \vec{r}, \quad \text{2nd } \frac{d}{dt} \text{ derivative gave}$$

Coriolis, Centrifugal Fictitious Forces

$$T = \frac{1}{2} M \dot{\vec{R}}^2 + \frac{1}{2} \vec{\omega}^T \cdot \hat{I}^{(cm)} \cdot \vec{\omega} \quad \text{for rotation about cm point}$$

$$\vec{L} = \hat{I} \cdot \vec{\omega}, \quad T = \frac{1}{2} \vec{\omega} \cdot \vec{L} \quad \text{for rotation about any fixed point}$$

$$\hat{I}_{ab} = \sum_i m_i (\delta_{ab} \vec{r}_i^2 - r_i^a r_i^b)$$

$$\hat{I}_{ab} = \int_{\text{Vol}} dV \rho(\vec{r}) [\delta_{ab} \vec{r}^2 - r_a r_b]$$

} depend on fixed "origin" & axes choice



Translate Origin :  $\hat{I}_{ab}^{(Q)} = \hat{I}_{ab}^{(cm)} + M(\delta^{ab} \vec{R}^2 - R^a R^b)$  (4)

Rotate Axes :  $\hat{I}' = U \hat{I} U^T$

Principal Axes [given  $\hat{I}$  for some origin & axes]

$$\hat{I} \vec{e} = \lambda \vec{e}$$

$$\det(\hat{I} - \lambda \mathbb{1}) = 0$$

3  $\lambda$ 's  
3  $\vec{e}$ 's

3  $\vec{e}$ 's give principal axes,  $\lambda$ 's are principal moments of inertia  
 $I_1, I_2, I_3$

For these axes

$$\hat{I} = \begin{pmatrix} I_1 & 0 & 0 \\ 0 & I_2 & 0 \\ 0 & 0 & I_3 \end{pmatrix}$$

[same origin]

Euler Equations use principal body axes to describe rotation of rigid body about fixed pt.

$$I_1 \dot{\omega}_1 - (I_2 - I_3) \omega_2 \omega_3 = \tau_1$$

etc

$\uparrow \quad \uparrow$

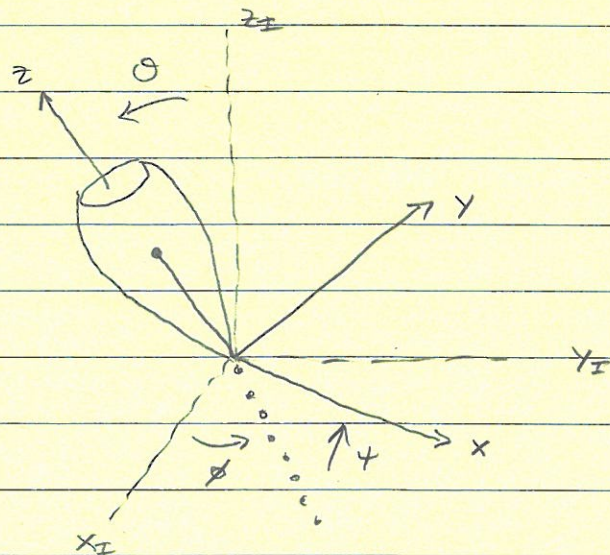
Component in body frame

eg. Symmetric Top

$\dot{\theta}$  nutation

$\dot{\phi}$  precession

$\dot{\psi}$  rotation about z





Vibrations expand about min. of ptal.

$$q_i = q_{0i} + \eta_i$$

$$V = V(q_0) + \frac{1}{2} V_{ij}(q_0) \eta_i \eta_j + \dots$$

$$T = \frac{1}{2} T_{ij}(q_0) \dot{\eta}_i \dot{\eta}_j + \dots$$

$$L = \frac{1}{2} \dot{\eta}^T \hat{T} \dot{\eta} - \frac{1}{2} \eta^T \hat{V} \eta$$

Normal Modes:  $\vec{\eta}^{(k)} = \vec{a}^{(k)} e^{-i\omega^{(k)}t}$

$$\hat{V} \vec{a} = \lambda \hat{T} \vec{a}$$

$$\det(\hat{V} - \lambda \hat{T}) = 0 \quad \text{gives} \quad \lambda^{(k)} = [\omega^{(k)}]^2 \quad 's$$

$$(\hat{V} - (\omega^{(k)})^2 \hat{T}) \cdot \vec{a}^{(k)} = 0 \quad \text{gives} \quad \vec{a}^{(k)} \quad 's$$

norm choice  $\vec{a}^{(l)T} \hat{T} \vec{a}^{(k)} = \delta^{lk}$

General Solution is Superposition  $\vec{\eta} = \text{Re} \sum_k \underbrace{C_k}_{\text{complex}} \vec{\eta}^{(k)}$

Normal coords  $\vec{\eta} = A \vec{\xi}$ ,  $A = \begin{pmatrix} \uparrow \vec{a}^{(1)} & \dots & \vec{a}^{(n)} \\ \downarrow & & \downarrow \end{pmatrix}$

$$L = \frac{1}{2} \sum_i \dot{\xi}_i^2 - \frac{1}{2} \sum_i (\omega^{(i)})^2 \xi_i^2$$

$$\ddot{\xi}_i + (\omega^{(i)})^2 \xi_i = 0$$

### Generating Functions for Canonical Transform

$$Q_i = Q_i(q, p, t) \quad \text{canonical,}$$

$$P_i = P_i(q, p, t)$$

consider inversions

$$\begin{cases} q_i = q_i(Q, P, t) \\ p_i = p_i(Q, P, t) \end{cases}$$

or  $\begin{cases} p_i = p_i(q, Q, t) \\ Q_i = Q_i(q, Q, t) \end{cases} \quad \text{etc}$

$$p_i \dot{q}_i - H = P_i \dot{Q}_i - K + \frac{dF}{dt}$$



(6)

$F = F_1(q, Q, t)$  generates canonical transform through

$$p_i = \frac{\partial F_1}{\partial q_i} = p_i(q, Q, t)$$

$$P_i = -\frac{\partial F_1}{\partial Q_i} = P_i(q, Q, t)$$

$$K = H + \frac{\partial F_1}{\partial t}$$

$F = F_2(q, P, t) - Q_i P_i$  gives  $p_i = \frac{\partial F_2}{\partial q_i}$ ,  $Q_i = \frac{\partial F_2}{\partial P_i}$

$$K = H + \frac{\partial F_2}{\partial t}$$

### Poisson Brackets

$$[u, v]_{q,p} = \sum_i \left( \frac{\partial u}{\partial q_i} \frac{\partial v}{\partial p_i} - \frac{\partial u}{\partial p_i} \frac{\partial v}{\partial q_i} \right)$$

• Fundamental Brackets  $[q_j, q_k]_{q,p} = 0 = [p_j, p_k]_{q,p}$   
 $[q_j, p_k]_{q,p} = \delta_{jk}$

• Insntr  $Q = Q(q, p, t)$  canonical iff  $[Q_j, Q_k]_{q,p} = 0$   
 $P = P(q, p, t)$   $[P_j, P_k]_{q,p} = 0$   
 $[Q_j, P_k]_{q,p} = \delta_{jk}$

• for canonical vars poisson brackets are same  
 $[u, v]_{q,p} = [u, v]_{Q,P}$

• Eqns of Motion  $\frac{du}{dt} = [u, H]_{q,p} + \frac{\partial u}{\partial t}$  any  $u(q, p, t)$   
 $\uparrow$   
 $H$  for  $q, p$

• Conserved  $\frac{du}{dt} = 0 \Leftrightarrow [u, H]_{q,p} + \frac{\partial u}{\partial t} = 0$

eg. Infinitesimal Canonical Transform



# Hamilton - Jacobi

Hamilton's Principal function  $S(q_1, \dots, q_n, \alpha_1, \dots, \alpha_n, t)$

gen. function which transforms to

$$K=0 = H\left(q_1, \dots, q_n, \frac{\partial S}{\partial q_1}, \dots, \frac{\partial S}{\partial q_n}, t\right) + \frac{\partial S}{\partial t} = 0$$

$$\dot{p}_i = 0, \quad \dot{Q}_i = 0$$

time-dependent H-J eqn

$$p_i = \alpha_i = \text{constant}$$

$$p_i = \frac{\partial S}{\partial q_i}$$

$$Q_i = p_i = \frac{\partial S}{\partial \alpha_i} = \text{constant}$$

} solve to get  $q_j = q_j(\alpha, p, t)$   
 $p_j = p_j(\alpha, p, t)$

Hamilton's Characteristic function  $W(q_1, \dots, q_n, \alpha_1, \dots, \alpha_n)$

$$H\left(q_1, \dots, q_n, \frac{\partial W}{\partial q_1}, \dots, \frac{\partial W}{\partial q_n}\right) = \alpha_1$$

H-J eqn  
time indep.

$$p_i = \alpha_i$$

$$p_i = \frac{\partial W}{\partial q_i}$$

W is generating function

$$K = \alpha_1$$

$$p_1 = \frac{\partial W}{\partial q_1} - t$$

$$\dot{Q}_1 = \frac{\partial K}{\partial \alpha_1} = 1, \quad Q_1 = p_1 + t$$

$$p_{i>1} = \frac{\partial W}{\partial q_i}$$

$$\dot{Q}_{i>1} = 0$$

$$Q_{i>1} = p_i$$

Solve by Separating Variables

$$W = W_1(q_1, \alpha) + W_2(q_2, \alpha) + \dots + W_n(q_n, \alpha)$$

Cyclic Coords:  $q_1$  cyclic  $\Rightarrow W_1(q_1, \alpha) = \gamma q_1$

$p_1 = \gamma = \text{constant}$   
(=  $\alpha_2$  say)



eg. Harmonic Osc.

eg. Kepler Problem

$$\alpha_1 = E, \quad \alpha_0 = l = |\vec{L}|$$

$$\alpha_\phi = (\vec{L} \text{ about } \hat{z})$$

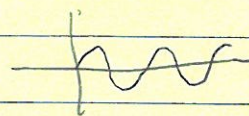
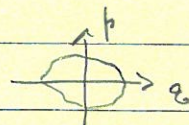
## Action-Angle Variables

1-dim  $H(q, p) = \alpha_1 \Rightarrow p = p(q, \alpha_1)$

Oscillation

 $p, q$  periodic

Rotation

 $p$  periodic $q$  unbounded

Action Variable

$$J = \oint p dq = J(\alpha) = \text{const.}$$

↑  
over  
one  
cycle

consider  $W = W(q, J)$ 

Angle Variable

$$W = \frac{\partial W}{\partial J}, \quad \dot{W} = \frac{\partial H(J)}{\partial J} = \Omega = \text{constant} = \text{frequency}$$

$$\Omega = \frac{1}{\tau}, \quad \tau \text{ period}$$

Many Vars $(q_i, p_i)$  oscillate or rotate, completely separable

$$J_i \equiv \oint p_i dq_i = J_i(\alpha)$$

$$W_i = \frac{\partial W}{\partial J_i} = W_i(q_1, \dots, q_n, J_1, \dots, J_n)$$

$$\dot{W}_i = \frac{\partial H}{\partial J_i} = \Omega_i(J_1, \dots, J_n) \quad \text{frequencies}$$

eg. Kepler

$$E = \frac{-2\pi^2 k^2 m}{(J_r + J_\theta + J_\phi)^2}$$

$$\text{so } \Omega_\theta = \Omega_\phi = \Omega_r$$