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太阳能光热梯级发电系统设计 及其特性研究

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Cascade solar thermal power system design and research
of the key features

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摘 要

随着化石能源的消耗和环境问题的凸显,太阳能作为一种新能源,具有分布广泛、总量巨大、取之不竭、无污染的特点,越来越受到世界各国的重视,被广泛认为是未来最有潜力替代传统化石能源的清洁能源。在发电领域,太阳能光热发电是除了太阳能光伏发电之外的另一种发电形式。与光伏发电相比,光热发电因具有发电平稳,电网兼容性友好,易于与现有化石燃料电厂组合等优点而受到越来越多的关注。已经商业应用的太阳能光热发电技术分为槽式集热发电、碟式集热发电和塔式集热发电三种。三种发电技术各有优缺点:槽式集热发电应用最广,成本较低,但效率也较低;碟式集热发电规模较小,多用于分布式发电;塔式集热发电规模较大,成本较高,目前处于快速发展阶段。综合利用现有发电技术的优缺点,在能量梯级收集和能量梯级利用的思想下,提出采用多种集热发电方式和多种热功循环的梯级系统,是实现大规模太阳能光热发电的一种新颖的可行的技术方案。

本课题以国家国际合作项目专项“太阳能梯级集热发电系统关键技术合作研究”为背景,目标是研究太阳能光热发电装置,利用各种传统型式的太阳能光热发电系统的优缺点以及热力特性,提出并组建、优化太阳能梯级集热发电系统,为探索出可大规模高效率利用太阳能的光热发电技术提供新的方案。主要研究内容和结论包括:

首先,提出太阳能光热梯级集热发电系统的拓扑结构。通过热力特性分析,结合系统中各部件的工作特点,合理布局太阳能光热梯级集热发电系统,利用不同热功循环实现不同品位的能量的梯级利用。合理的梯级发电系统方案才能充分利用发电系统中各部件的性能特点,为创建高效率的太阳能光热梯级发电系统提供基础。本文针对系统中的各组件,组建了多种可行的梯级集热系统拓扑结构。经过系统评估、参数选取、初步计算、方案比较,确定了两种具有代表性的太阳能光热梯级发电系统方案。一种方案同时选用水工质朗肯循环和斯特林循环,利用给水来冷却斯特林机冷腔,回收利用斯特林机放出的热量;另一种方案选用多级有机工质朗肯循环,利用上一级的凝集热来加热下一级的循环工质,实现能量的梯级利用。

其次,针对太阳能光热梯级集热发电系统的各部件建立机理模型。依据目标对象的运行机理,根据物理平衡方程,对系统中的各部件,尤其是系统中的关键部件,如集热器、蒸汽产生系统、汽轮机、斯特林机等,建立起数学模型。各部件的数学模型是由经典理论或是大量实验数据验证的模型,是组建光热梯级集热发电系统模型的基础。对于槽式集热器的集热管和碟式集热器的集热器,建立了热损失模型;对于斯特林机,基于合理的简化和假设,推导出了考虑了各种热损失和不可逆因素的斯特林机

模型。各部件模型使用 **MATLAB** 语言编写,采用面向对象的方法,充分利用了继承、多态等特性,保证了各部件之间既具有独立性又具有关联性。

再次,组建太阳能光热梯级集热发电系统模型。根据所选择的太阳能光热梯级发电系统方案,基于建立好的系统中各部件的模型,利用面向对象语言的继承、组合、多态等特点,组建起梯级集热发电系统模型。研究系统在外及内部因素的耦合作用下主要参数及性能指标的变化规律,掌握其变化机理,建立其性能特性的计算方法。经过组建部件,设置参数,编译环境,完成了各系统方案的系统组建工作,最终完成了拥有自主计算机软件著作权的基于 **MATLAB** 的太阳能光热梯级发电的模拟系统。系统中各部件相对独立,便于更换或改进部件模型;各系统模型的计算结果可以以单个对象的方式方便地查看系统中各个部件的关键参数。

然后,模拟并优化太阳能光热梯级集热发电系统模型。在太阳能光热梯级发电系统性能特性研究的基础上,对系统进行流程优化、结构重构。具体地,通过对系统的蒸汽发生系统进行分析,提出了分阶段加热方法,通过改变导热油的质量流量降低蒸汽发生系统中的传热温差,有效降低了蒸汽发生系统中换热过程中产生的烟损,进而可以提高整个系统的效率。针对梯级系统中的斯特林机组,总结了斯特林机组所具有的五种基本排列形式,并分析了各种排列形式下机组的效率和输出功率的差异,得到了给定冷热源流体条件下斯特林机组最佳的排列方式。

最后,优化太阳能光热梯级集热发电系统的运行参数。针对特定结构方案和运行模式,以梯级发电系统的性能参数和经济指标为目标函数,选择合理的可调节参数,确立各种约束条件,利用现代优化方法,如基因算法、蚁群算法,完成系统的参数优化分析,以及对于独立系统的对比分析。分析结果表明,太阳能光热梯级集热发电系统在一定的参数条件下,相比其对应的独立系统,具有更高的总体光电转换效率。在太阳直射强度为 700 W/m^2 ,碟式集热器出口空气温度为 800°C 的条件下,方案 1 所选用的太阳能光热梯级集热发电系统比对应的独立系统效率提升 5.2%,方案 2 所选用的太阳能光热梯级集热发电系统比对应的独立系统效率提升 15.3%。

关键词： 槽式集热器,碟式集热器,朗肯循环,斯特林循环,斯特林机组,梯级发电

Abstract

With the increasing awareness of problem of fossil energy consumption and environmental pollution, solar energy as a renewable energy, which has the advantages of widely distribution, huge amount, inexhaustible and no pollution, has received much attention by many countries and been regarded as the best potential candidate of the fossil energy. Concentrating solar thermal power generation is another form of power generation technology except solar photovoltaic power generation. Compared to solar photovoltaic, solar thermal power is gaining more attention for its advantages as smooth power generation, good grid compatibility, easy to integrate with existing fossil power plant.

Commercial solar thermal power generation technology is divided into trough collector power generation, dish collector power generation and solar tower power generation. These three types of power generation technologies have their own advantages and disadvantages: trough collector power generation is the most widely used one, its cost is low, however its efficiency is also low; dish collector power generation has high efficiency and smaller capacity, it is used for distributed generation widely; solar tower generation, which has large scales, high efficiency and high cost, is currently in rapid development stage. Based on the idea of energy cascade collection and energy cascade utilization, this paper proposed a cascade system that uses different collector power generation methods and different thermodynamic cycles, which may be a new and feasible technology to realize large-scale solar thermal power generation.

The research is based on the national cooperation project "Collaborative research on key technologies to produce electricity by cascade utilization solar thermal energy" as the background. The objective of this project is to research the equipment of solar thermal power generation system, to propose, develop and optimize a solar thermal cascade system depending on the advantages and disadvantages of the solar thermal power generation system, and to explore a new feasible technology for large-scale solar thermal power generation. The main contents and conclusions of this paper are as follows:

Firstly, the topological structure of solar thermal cascade power generation system was proposed. According to the analysis of thermal characteristics and the working characteristics of each component in the system, rationally arranged topological structures of cascade system were proposed. These systems use different thermodynamic cycles to utilize energy

in different temperature zones. A reasonable cascade generation system can make full use the performance characteristics of the components in the power generation system and provide the foundation for higher efficiency solar thermal cascade generation systems. In this paper, several schemes of feasible topological structures of solar thermal cascade system were set up according to the components in the system. After system evaluation, parameter selection, preliminary calculation and scheme comparison, two representative typical schemes were determined. In one scheme, both Rankine cycle (water as the working fluid) and Stirling cycle are used for power generation. Cooling water of the Rankine cycle is used to cool the hot end of the Stirling engines to recover the released heat. In the other scheme, multiple organic Rankine cycles are used for power generation. Condensation heat of upper cycle is absorbed by lower cycle for energy cascade utilization.

Secondly, mechanism models were established for the components of solar thermal power generation system. The mechanism mathematical models were developed according to the operation mechanism of the target object and physical equations. The key components in the system, such as collectors, steam generating system, steam turbine and Stirling engine, were modeled with details. The mathematical model of each component is a model verified by the classical theory or a large number of experimental data, which is the basic of the model of the cascade solar thermal power generation system. Heat loss models were established for the receivers of trough collector and dish collector. For the Stirling engine, based on the reasonable simplification and hypothesis, the model of the Stirling machine considered various losses and irreversibilities was developed. The component models were developed in MATLAB by using object-oriented method. It makes full use of inheritance and polymorphism to ensure both the independence and the relevance of the components.

Thirdly, the solar thermal cascade generation system models were developed. Based on the selected solar thermal cascade generation systems, solar thermal cascade generation system models were established based on the model of each component in the systems. The object-oriented features of inheritance, combination and polymorphism were used for the model development. The change rules of the main parameters and the performance indexes under the coupling of external and internal factors were studied. The change mechanism was studied and the calculation method of its performance characteristics was established. After setting up the components, setting the parameters and compiling the environment, the paper completes the system construction of each system scheme, and finally completes the simulation system of solar thermal cascade generation based on MATLAB with the copyright of independent computer software. The system components are relatively independent, easy

to replace or improve the parts model; the results of the calculation of the system model can be a single object to easily view the various components of the system key parameters.

Then, simulation and optimization of cascade solar thermal power generation system model. Based on the study of the performance characteristics of solar thermal cascade generation system, the system is optimized and the structure is reconstructed. In particular, by analyzing the steam generation system of the system, a method of staged heating is proposed to reduce the heat transfer temperature difference in the steam generating system by changing the mass flow rate of the heat conduction oil, effectively reducing the heat generated during the heat exchange process in the steam generating system. Which can improve the efficiency of the whole system. Based on Stirling unit in cascade system, five kinds of basic arrangement forms of Stirling unit are summarized, and the difference of unit efficiency and output power under various arrangement forms is analyzed, and a given cold and heat source fluid Stirling unit under the conditions of the best arrangement.

Finally, the operating parameters of solar thermal cascade power generation system are optimized. According to the specific structural scheme and operation mode, the performance parameters and economic indexes of the cascade generation system are taken as the objective function, reasonable adjustable parameters are selected, various constraints are established, and modern optimization methods such as genetic algorithm and ant colony algorithm are used to complete the system Parameter optimization analysis, as well as the independent system for comparative analysis. The results show that solar thermal cascade power generation system has higher overall photoelectric conversion efficiency under certain parameter conditions than its corresponding independent system. Under the condition of direct solar radiation intensity of 700 W/m^2 and dish type collector outlet air temperature of 800°C , the solar thermal cascade power generation system of Scheme 1 is better than the corresponding The efficiency of stand-alone system is increased by 5.2%. The solar thermal cascade power generation system selected in Scheme 2 is 15.3% more efficient than the corresponding independent system.

Key words: parabolic trough collector, parabolic dish collector, Rankine cycle, Stirling cycle, Stirling engine array, cascade powering

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Chapter 1 System modeling

To investigate the performance of the proposed cascade systems, mechanism models of the systems were developed with EES (Engineering Equation Solver) and MATLAB. Bottom-up design method was used for the system modeling. Firstly, the mechanism models of the developed in EES to validate the theoretical relationships of the models. Secondly, the component models were developed in MATLAB by using object-oriented method. It makes full use of inheritance and polymorphism to ensure both the independence and the relevance of the components. Three circuits, air circuit, water circuit and oil circuit, were developed with some specific state parameters in some key components. Energy-based models of these key components were created on the basis of their thermodynamic behavior, heat transfer and the second law.

The following parts introduce models of some key components.

1.1 Component modeling

1.1.1 Parabolic trough collector

Parabolic trough collector consists of a reflector and a receiver. The reflector (mirror) reflects direct solar radiation and concentrates it onto a receiver tube located in the focal line of the parabola. The receiver is typically a metal absorber tube with high absorption rate coating. An outer glass tube is used outside the absorber tube to reduce thermal losses and the space between the absorber tube and the glass tube is usually drawn into a vacuum to further reduce the thermal losses.

Optical loss exists in the reflection process due to optical efficiency terms. The reflection terms can be listed as bellow^[3]:

- Shadowing factor
- Tracking error
- Geometry error
- Clean mirror reflectance
- Dirt on mirrors

- Unaccounted errors

Another term, incident angle modifier $K(\theta)$, should be concerned when the solar irradiation is not normal to the collector aperture. It is a function of the solar incidence angle to the normal of the collector aperture (θ).

The equation was determined from trough collector testing conducted at SNL^[65].

$$K(\theta) = \cos \theta + 0.000884\theta - 0.00005369\theta^2 \quad (1.1)$$

The optical losses are associated with five parameters (see fig. 1-1):

- Reflectivity, ρ : only a fraction of the incident radiation is reflected towards the receiver. The fraction is determined by the reflector type and dirt condition. Reflectivity of commercial parabolic trough mirrors can be assumed to be 0.9 for washed mirrors.
- Intercept factor, γ : a fraction of the direct solar radiation reflected by the mirrors does not reach the glass cover of the absorber tube due to either microscopic imperfections of the reflectors or macroscopic shape errors in the parabolic trough concentrators (e.g., imprecision during assembly). These errors cause reflection of some rays at the wrong angle, and therefore they do not intercept the absorber tube. These losses are quantified by an optical parameter called the intercept factor, γ , that is typically 0.95 for a collector properly assembled.
- Transmissivity of the glass tube, τ : only a fraction of the direct solar radiation reaching the glass cover of the absorber pipe is able to pass through it. The ratio between the radiation passing through the glass tube and the total incident radiation on it, gives transmissivity τ , which is typically 0.93.
- Absorptivity of the absorber selective coating, α_{abs} : this parameter quantifies the amount of energy absorbed by the steel absorber pipe, compared with the total radiation reaching the outer wall of the steel pipe. This parameter is typically 0.95 for receiver pipes with a cermet coating, whereas it is slightly lower for pipes coated with black nickel or chrome.
- Soiling factor, F_e : because of the dirt on reflectors will reduce the reflectivity, it needs to concern the soiling factor. The soiling factor F_e takes into account the progressive soiling of mirrors and glass tubes after washing.

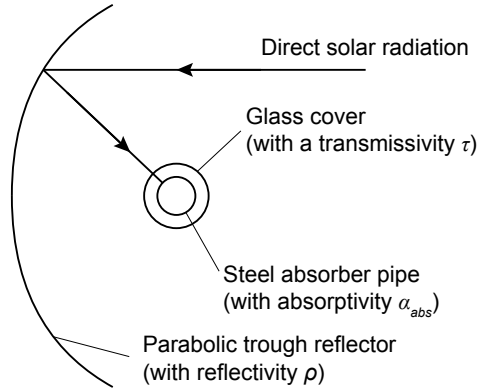


Figure 1-1 Some of the optical parameters of a parabolic trough

The energy pass through the glass tube to the receiver can be obtained by

$$P = I_r w_{tc} L_{tc} \rho \gamma \tau F_e K(\theta) \quad (1.2)$$

The solar energy absorbed by the absorber occurs very close to the outer surface, to simplify the absorption process, it is treated as a uniform heat flux q'' .

$$q'' = \frac{P}{\pi d_o L_{tc}} = \frac{I_r w_{tc} \rho \gamma \tau F_e K(\theta)}{\pi d_o} \quad (1.3)$$

Assume overall heat transfer coefficient $U(T_{abs})$ is uniform for whole length of the collector, and the heat transfer correlation in Appendix B can be used. Figure 1-2 shows the schematic diagram of the thermal analysis of the absorber pipe.

$$\frac{T_o - T_{amb} - \frac{q''}{U(T_{abs})}}{T_i - T_{amb} - \frac{q''}{U(T_{abs})}} = \exp\left(-\frac{U(T_{abs})PL}{\dot{m}c_p}\right) \quad (1.4)$$

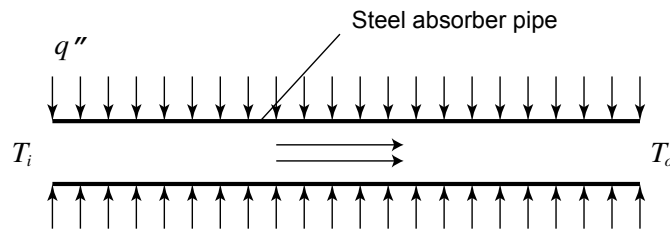


Figure 1-2 Schematic diagram of the absorber pipe

Since the Nusselt number Nu in the pipe is very large (about 1×10^4), small temperature difference exists between the absorber and oil. So the average fluid temperature $(T_i + T_o)/2$ can be used as the average value of T_{abs} , and $U(T_{abs})$ can be obtained by the a second-order polynomial function given by Romero^[66]. The length L required to get the required number of trough collectors in a row can be obtained from Equation (1.4).

1.1.2 Parabolic dish collector

Parabolic dish collector consists of a reflector and a receiver. The reflector (mirror) tracks the sun to reflect direct solar radiation and concentrates it onto a receiver located at the focal point of the reflector. Two axes tracking system needs to be applied for the reflector to continuously follow the daily path of the sun.

In a traditional dish-Stirling system, a Stirling engine is located at the focal point. The Stirling engine has a receiver to absorb the thermal energy from the concentrated sunlights. The receiver consists of an aperture and an absorber. The aperture in a Stirling receiver is located at the focal point of the reflector to reduce the radiation and convection losses. The absorber absorbs the solar radiation and transfers the thermal energy to the working gas of the Stirling engine. An electrical generator, directly connected to the crankshaft of the engine, converts the mechanical energy into electricity.

In the proposed cascade system, a volumetric receiver is located on the focal point. Spiral tube is located in the receiver to absorb the concentrated solar energy. Air (or nitrogen, is used as the heat transfer fluid) flows through the tube to transfer the absorbed energy as the heat source of Stirling engine(s).

The reflector is a key element of the systems. The curved reflective surface can be manufactured by attached segments, by individual facets or by a stretched membranes shaped by a continuous plenum. In all cases, the curved surface should be coated or covered by aluminum or silver reflectors.

Two different methods^[67] are applied for the sun tracking systems:

- Azimuth elevation tracking by an orientation sensor or by calculated coordinates of the sun performed by the local control.
- Polar tracking, where the concentrator rotates about an axis parallel to the earth's axis rotation.

A dish reflector product of SES (Stirling Energy System) is used in this cascade system,

and its key parameters can be found in Table 1.1. The structure of the receiver is shown in fig. 1-3.

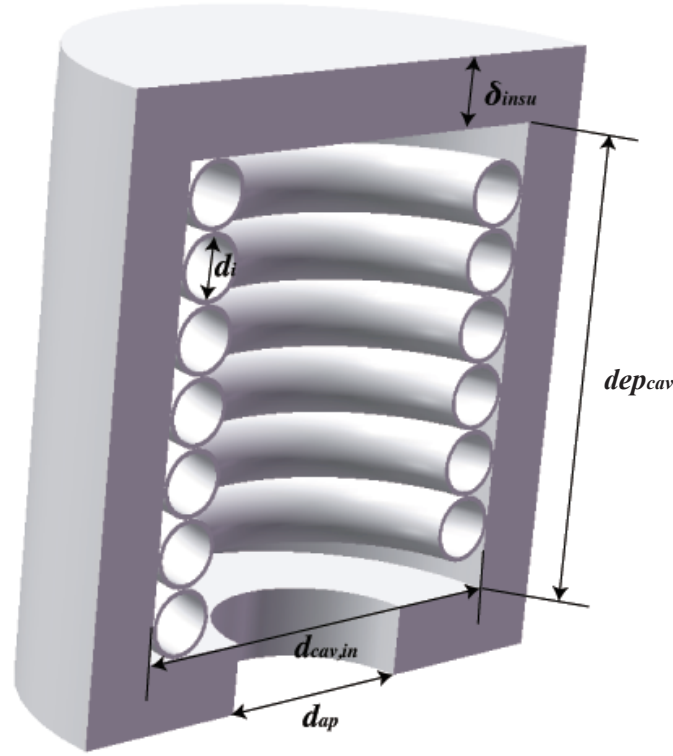


Figure 1-3 The structure of the dish receiver

The dish receiver model concerns the losses include: collector losses due to mirror reflectivity, receiver intercept losses, losses due to shading, and thermal losses. Thermal losses take the largest portion of all those losses, which are due to conduction, convection and radiation. Figure 1-4 shows the thermal network of dish receiver, which concerns the losses:

- Radiation losses reflected off of the receiver cavity surfaces and out of the receiver through the aperture. ($q_{rad,ref}$)
- Conductive losses through the receiver insulating layer. ($q_{cond,tot}$)
- Free convection from the cavity in the absence of wind. ($q_{conv,free}$)
- Forced convection in the presence of wind. ($q_{conv,forc}$)
- Emission losses due to thermal radiation emitted from the receiver aperture. ($q_{rad,emit}$)

Table 1.1 Key parameters of the dish collector

Parameter	Value	Parameter	Value	Parameter	Value
d_{cav}	0.46 m	ϵ_{insu}	0.6	θ_{dc}	45°
δ_{insu}	0.075 m	α_{cav}	0.87	γ	0.97
dep_{cav}	0.23 m	δ_a	0.005 m	$\eta_{shading}$	0.95
d_{ap}	0.184 m	$d_{i,1}$	0.07 m	ρ	0.91
λ_{insu}	0.06 W/(m · K)	A_{dc}	87.7 m ²		

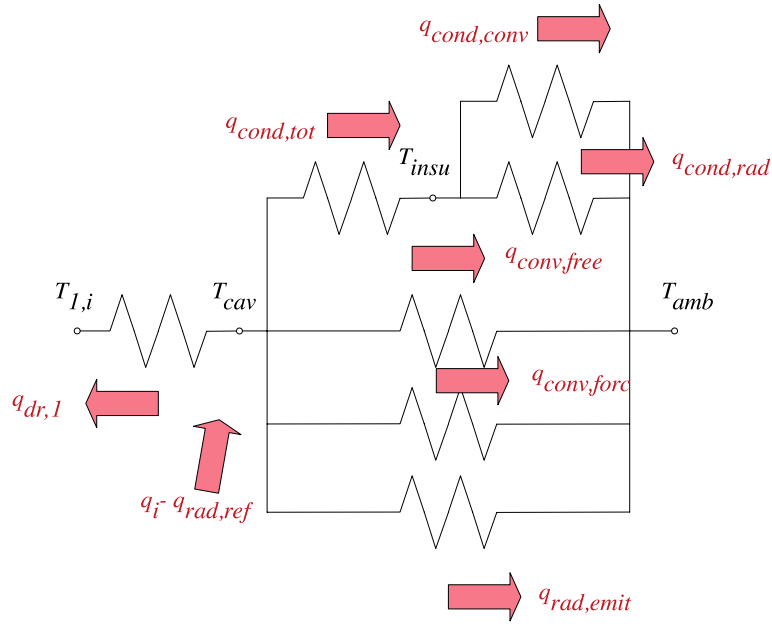


Figure 1-4 Thermal network of dish receiver

To solve the thermal network in fig. 1-4, correlations and relationships of the heat fluxes should be clear.

1. *Inlet energy from the reflector, q_i*

To simplify the model, influences made by receiver blocking and imperfection track

are ignored.

$$q_i = I_r A_{dc} \gamma \eta_{shading} \rho \quad (1.5)$$

In equation 1.5, γ is the intercept factor, $\eta_{shading}$ is the shading factor between different collectors, ρ is the reflectivity of the reflector.

2. *Heat exchange between the HTF and the dish absorber, $q_{dr,1}$*

The heat transfer process between the HTF and the dish absorber is simplified to a heat exchange process of a flow in a uniform temperature heat pipe. So $q_{dr,1}$ can be written as

$$q_{dr,1} = h_{dr,1} A_{dr,1} \Delta T_{ln,dr,1} \quad (1.6)$$

where

$$h_{dr,1} = Nu_{tube} \lambda_{dr,1} / d_{i,1} \quad (1.7)$$

$$Nu_{tube} = c_r Nu'_{tube} \quad (1.8)$$

For helical spiral pipe, multiplier c_r based on curvature ratio can be written as^[68]

$$c_r = 1 + 3.5 \frac{d_{i,1}}{d_{cav} - d_{i,1} - 2\delta_a} \quad (1.9)$$

Nu'_{tube} is the Nusselt number of straight circular tube, which can be obtained by^[69]

$$Nu'_{tube} = 0.027 Re_{tube}^{0.8} Pr_{tube}^{1/3} (\mu_{tube} / \mu_{tube,w})^{0.14} \quad (1.10)$$

and the logarithmic mean temperature difference $\Delta T_{ln,dr,1}$ can be written as

$$\Delta T_{ln,dr,1} = \frac{(T_{cav} - T_{dc,i}) - (T_{cav} - T_{dc,o})}{\ln \frac{T_{cav} - T_{dc,i}}{T_{cav} - T_{dc,o}}} \quad (1.11)$$

3. *Radiation losses reflected off the receiver, $q_{rad,ref}$*

$$q_{rad,ref} = (1 - \alpha_{eff})q_i \quad (1.12)$$

where α_{eff} is the effective absorptivity of the receiver.

$$\alpha_{eff} = \frac{\alpha_{cav}}{\alpha_{cav} + (1 - \alpha_{cav}) \frac{A_{ap}}{A_{cav}}} \quad (1.13)$$

α_{cav} is the absorptivity of the cavity, A_{cav} is the cavity area, A_{ap} is the aperture area.

4. *Conductive losses through the receiver insulating layer, $q_{cond,tot}$*

$$q_{cond,tot} = 2\pi\lambda_{insu}dep_{cav} \frac{T_{cav} - T_{insu}}{\ln((d_{cav} + 2\delta_{insu})/d_{cav})} \quad (1.14)$$

where T_{cav} is the temperature of the cavity wall, T_{insu} is outside temperature of the insulation wall.

5. *Convection losses from the receiver insulating layer, $q_{cond,conv}$*

$$q_{cond,conv} = h_{insu}A_{insu}(T_{insu} - T_{amb}) = \frac{k_{insu}Nu_{insu}A_{insu}(T_{insu} - T_{amb})}{d_{cav} + 2\delta_{insu}} \quad (1.15)$$

where Nu_{insu} can be obtained by the correlation for flow over a circular cylinder.^[70]

6. *Radiation losses from the receiver insulating layer, $q_{cond,rad}$*

$$q_{cond,rad} = \epsilon_{insu}A_{insu}\sigma(T_{insu}^4 - T_{amb}^4) \quad (1.16)$$

7. *Free convection from the cavity in the absence of wind, $q_{conv,free}$*

Ma^[71] conducted tests to determine the free convection losses from the receiver for alternative setups, and the data were consistent with Stine and McDonald's free convection correlation. It is assumed that forced convection is independent of free convection in the receiver, so the total convection losses can be represented as the total of the free and forced convection losses as shown in Figure 1-4.

$$q_{conv,free} = h_{free}A_{cav}(T_{cav} - T_{amb}) \quad (1.17)$$

where $h_{free} = k_{film}Nu_{free}/\overline{d_{cav}}$, $\overline{d_{cav}}$ is the effective diameter of the cavity, $\overline{d_{cav}} = d_{cav} - 2d_i - 4\delta_a$. $d_i = 0.066$ m

8. *Force convection from the cavity in the presence of wind, $q_{conv,forc}$*

$$q_{conv,forc} = h_{forc} A_{cav} (T_{cav} - T_{amb}) \quad (1.18)$$

Wu^[72] present a comprehensive review and systematic summarization of convection heat loss from cavity receiver in parabolic dish solar thermal power system. And we choose the correlation presented by Leibfried^[73]. This correlation gives an extended model of Koenig^[74] and Stine^[75] with better results.

For forced convection loss, side-on wind convection loss model given by Ma^[71], which is independent of the aperture orientation, is used

$$h_{forc} = 0.1967 v_{wind}^{1.849} \quad (1.19)$$

9. *Emission losses due to thermal radiation emitted from the receiver aperture, $q_{rad,emit}$*

The emissivity is set equal to the effective absorptivity of the cavity (gray body),

$$\epsilon_{cav} = \alpha_{eff} \quad (1.20)$$

$$q_{rad,emit} = \epsilon_{cav} A_{ap} \sigma (T_{cav}^4 - T_{amb}^4) \quad (1.21)$$

From fig 1-4, it can be found that

$$q_{dr,1} = q_i - q_{rad,ref} \quad (1.22)$$

$$q_{dr,1} = q_{cond,tot} + q_{conv,free} + q_{conv,forc} + q_{rad,emit} \quad (1.23)$$

$$q_{cond,tot} = q_{cond,conv} + q_{cond,rad} \quad (1.24)$$

So the temperature nodes in the thermal network can be solved by these equations.

$q_{dr,1}$ can be obtained by Equation 1.6.

1.1.3 Stirling engine

1.1.3.1 Theoretical Stirling cycle

In a Stirling cycle, there are two isothermal processes that exchange heat with heating and cooling fluids, two isochoric processes that exchange heat with regenerator. In Figure 1-5, the heat absorbed by regenerator in process 4-1 is reused in process 2-3, but only able to heat the working gas from 2 to 3' due to the imperfect regeneration. e is defined as the regenerator effectiveness^[98,122], $e = \frac{T_R - T_L}{T_H - T_L}$, where T_H is the temperature in the hot space, T_L is the temperature in the cold space, T_R is the effective working fluid temperature in the regenerator.

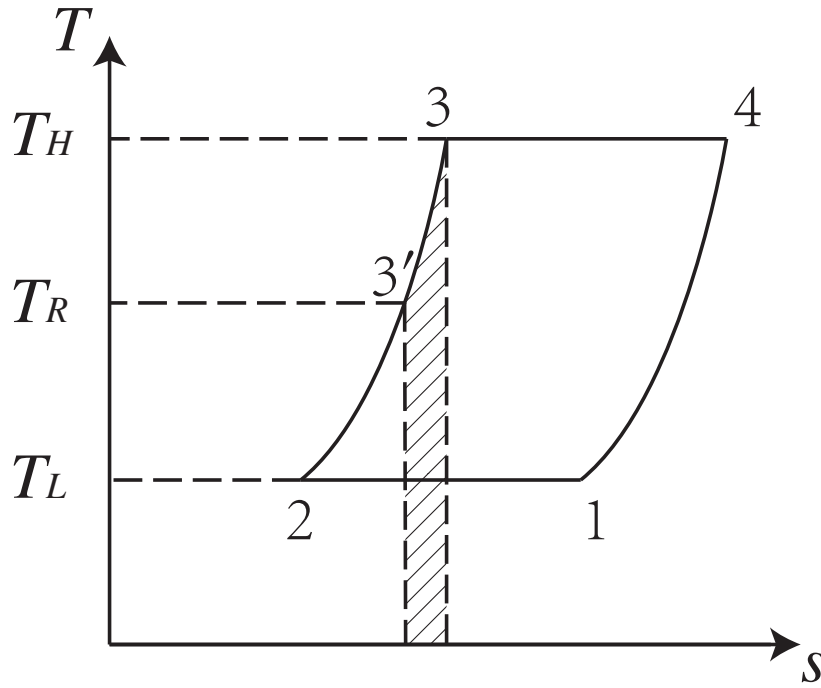


Figure 1-5 T - s diagram of a Stirling cycle

In order to obtain a simplified analytical model, several simplifications were made:

- The working gas in Stirling engines obeys the idea gas law.
- No heat loss to the environment for Stirling engines.

- Overall heat transfer coefficients of the fluids are constant.
- A symmetrical regenerator behavior is assumed^[98,122] so that a simple effectiveness can be obtain by $T_R = \frac{T_H - T_L}{\ln(T_H/T_L)}$.

To consider internal irreversibilities in Stirling cycle made by dead volumes, as described in^[123], total dead volume V_D is divided into heater dead volume V_{DH} , regenerator dead volume V_{DR} and cooler dead volume V_{DC} . There exists a factor K to describe the dead volumes under different temperatures. K is relevant with temperatures in the process and regenerator effectiveness.

$$K = \frac{V_{DH}}{T_H} + \frac{V_{DR}}{T_R} + \frac{V_{DC}}{T_L} \quad (1.25)$$

For the isothermal compression process 1-2, the output work

$$W_{12} = \int_{V_E+V_C}^{V_E} p_{12} dV = -mRT_L \ln \frac{V_E + V_C + KT_L}{V_E + KT_L} \quad (1.26)$$

For the isothermal expansion process 3-4, the output work

$$W_{34} = \int_{V_E}^{V_E+V_C} p_{34} dV = mRT_H \ln \frac{V_E + V_C + KT_H}{V_E + KT_H} \quad (1.27)$$

Define $\gamma_H = \frac{V_E+V_C+KT_H}{V_E+KT_H}$, and $\gamma_L = \frac{V_E+V_C+KT_L}{V_E+KT_L}$, so in a cycle, the theoretical output work

$$W_{th} = W_{12} + W_{34} = mR(T_H \ln \gamma_H - T_L \ln \gamma_L) \quad (1.28)$$

For the isochoric heating process 3'-3, the absorbed heat

$$Q_{3'3} = nc_v(T_H - T_R) = \frac{1-e}{k-1} mR(T_H - T_L) \quad (1.30)$$

For the the isothermal expansion process 3-4, the absorbed heat

$$Q_{34} = W_{34} = mRT_H \ln \gamma_H \quad (1.31)$$

In a cycle, the theoretical absorbed heat

$$Q_{th} = Q_{3'3} + Q_{34} = \frac{1-e}{k-1}mR(T_H - T_L) + mRT_H \ln \gamma_H \quad (1.32)$$

1.1.3.2 Irrevisibilities and losses

1. Non-ideal heat transfer effect

Because of non-ideal heater and cooler, the working fluid temperature (T_H/T_L) in these two heat exchangers is less/higher than the wall temperature (T_{hw}/T_{cw}), respectively. And T_H and T_L can be corrected by the wall temperatures as follows:

$$T_H = T_{hw} - \frac{Q_{se}}{h_h A_{hw}} \quad (1.33)$$

$$T_L = T_{cw} + \frac{(Q - W)s_{se}}{h_c A_{cw}} \quad (1.34)$$

The heat transfer coefficient can be obtained using the following correlation^[110]:

$$h_{h,c} = \frac{\mu c_p f_{Re}}{2D_{h,c} Pr_{h,c}} \quad (1.35)$$

where f_{Re} is a Reynolds friction factor defined as:

$$f_{Re} = 0.0791 Re_{h,c}^{0.75} \quad (1.36)$$

$Re_{h,c}$, $Pr_{h,c}$ and $D_{h,c}$ are Reynolds number, Prandtl number and hydraulic diameter of the heater/cooler exchanger.

2. Effect of pressure drop

Pressure drops in the heat exchangers cause power losses of the Stirling engine. The pressure drops can be obtained by^[104]:

$$\Delta p = -\frac{2f_{Re}\mu uV}{d^2 A} \quad (1.37)$$

where u is the working gas speed, V is volume, A is flow cross-section area.

The net power loss of the Stirling engine due to pressure drop of the heat exchangers can be evaluated by:

$$W_{pd} = \oint \sum_{i=E,C} (\Delta p_i \frac{dV_i}{d\theta}) d\theta \quad (1.38)$$

3. Effect of finite speed of piston and mechanical friction

Due to the finite speed of piston, the pressure on the piston surface is different from the pressure of expansion and compression spaces. It has been demonstrated that the pressure on the piston surface in the expansion process is less than the mean pressure in the expansion space. Similarly, the pressure on the piston surface in the compression process is greater than the mean pressure in the compression space. This means the output work is less than the theoretical value. Besides, The output work also reduces due to mechanical friction. The output work loss due to finite speed of piston and mechanical friction can be obtained as follows^[110]:

$$W_{fs} = \oint p(\pm \frac{au_p}{c} \pm \frac{\Delta p_f}{p}) dV \quad (1.39)$$

where the sign (+) is used in the compression space, and the sign (-) is used in the expansion space. p is the mean pressure in the compression/expansion space, u_p is velocity of the piston, c is the average speed of molecules and Δp_f is the pressure loss due to mechanical friction. Δp_f , a and c can be obtained by^[124]:

$$\Delta p_f = 0.97 + 0.009s_{se} \quad (1.40)$$

$$a = \sqrt{3k} \quad (1.41)$$

$$c = \sqrt{3RT} \quad (1.42)$$

4. Energy losses due to internal conduction

The temperature differs from the heater and cooler, heat losses from heater to cooler exists due to internal conduction through the walls of regenerator.^[107] The internal

conduction loss in a cycle can be obtained by follows:

$$Q_{id} = \frac{k_r A_r}{L_r s_{se}} (T_{hw} - T_{cw}) \quad (1.43)$$

where, k_r , A_r and L_r denote the regenerator matrix conductivity, regenerator length, and regenerator conductive area respectively.

5. Energy losses due to shuttle conduction

The displacer shuttles between the expansion and compression space. It absorbs heat during the hot end of its stroke and releases it during the cold end of its stroke. This heat loss can be estimated as^[125]:

$$Q_{sc} = 0.4 \frac{Z^2 k_p D_p}{J L_d s_{se}} (T_H - T_L) \quad (1.44)$$

where, Z , k_p , D_p , J and L_d denote the displacer stroke, piston thermal conductivity, displacer diameter, gap between the displacer and the cylinder, and length of the displacer respectively.

So, in a Stirling engine, the total absorbed heat in a cycle

$$Q = Q_{th} + Q_{id} + Q_{sc} \quad (1.45)$$

the output work

$$W = W_{th} - W_{pd} - W_{fs} \quad (1.46)$$

Power of the Stirling engine

$$P = W s_{se} \quad (1.47)$$

Efficiency of the Stirling engine

$$\eta = W/Q \quad (1.48)$$

1.1.3.3 Model validation

Evaluation of the developed thermal model was performed by considering the GPU-3 Stirling engine as a case study. Design specifications of the GPU-3 Stirling engine are indicated in Table 1.2. The thermal efficiency and power of the proposed Stirling engine model

was compared with previous thermal models and experimental data as shown in Table 1.3 and Table 1.4.

It can be found that the proposed model has much better agreement with the experimental results than previous thermal models at various rotation speeds and mean effective pressures. It is required to mention that in all thermal models both power W and input heat Q were determined by the thermal process of heat transfer between the wall and working gas. In the proposed model, W and Q are obtained by Equation 1.33 and 1.34. Therefore all the three parameters W , Q and η are determined by the thermal model and input parameters to the model. These input parameters includes heater, cooler, mean effective pressure, type of working gas and geometrical specification of the engine.

Table 1.3 and 1.4 indicate that when mean effective pressure of the engine increases from 2.76 MPa to 6.90 MPa, best performance (efficiency and power) prediction of the proposed model exists. When rotation speed increases from 16.67 Hz to 58.33 Hz, error in prediction of performance of the proposed model increases. The proposed model may have the best performance prediction at a low rotation speed, with mean effective pressure between 4.14 MPa and 5.52 MPa.

However, there is still some discrepancy between the the simulation results of proposed model and the experimental data. In the future researches, more accurate models of Stirling engine may be developed by considering other irreversibilities such as heat loss to the environment, gas spring hysteresis, and etc. It is worth pointing that there are more accurate Stirling engine models. For example, polytropic simulation models of Stirling engine show higher accuracy than our proposed model^[110,114]. However, the model needs more costly calculations and the polytropic indexes are engine-specific.

1.1.3.4 Heat transfer between the engine and the fluids

For a Stirling engine thermal process, the wall temperatures of the heater and cooler are considered to be uniform and constant. The heat transferred between the wall and the fluids is

$$(T_w - T)UdA = q_m c_p dT \quad (1.49)$$

Table 1.2 Design specifications of the GPU-3 Stirling engine^[110,126]

Parameter	Value
Engine type	β
Working gas	Helium
Mass of the working gas	1.136 g
<i>Heater</i>	
Number of tubes	40
Tube external diameter	4.83×10^{-3} m
Tube internal diameter	3.02×10^{-3} m
Tube length (cylinder side)	0.1164 m
Tube length (regenerator side)	0.1289 m
<i>Cooler</i>	
Number of tubes	312
Tube external diameter	1.59×10^{-3} m
Tube internal diameter	1.09×10^{-3} m
Average tube length	4.61×10^{-2} m
<i>Regenerator</i>	
Number of regenerator	8
Regenerator internal diameter	2.26×10^{-2} m
Regenerator length	2.26×10^{-2} m
Diameter of regenerator tube	4×10^{-5} m
Material	Stainless steel
<i>Volume</i>	
Swept Vol. (expansion/compression)	120.82/114.13 cm ³
Clearance Vol. (expansion/compression)	30.52/28.68 cm ³
Dead Vol. (heater/cooler/regenerator)	70.28/13.18/50.55 cm ³

Table 1.3 Thermal efficiency of the models and experimental data (at $T_{hw} = 922\text{ K}$ and $T_{cw} = 288\text{ K}$)

Rotation speed (Hz)	Mean effective pressure (MPa)	The simple analysis (variable $Pr^{[104]}$)			The adiabatic analysis (simple $\Pi^{[107]}$)			The proposed Stirling engine model			Experimental efficiency ^[110]
		Value (%)	Error (%)	Average error (%)	Value (%)	Error (%)	Average error (%)	Value (%)	Error (%)	Average error (%)	
16.67		38.72	18.22		32.48	11.98		28.16	7.66		20.50
25.00		36.16	15.46		31.21	10.51		27.75	7.05		20.70
33.33		33.79	15.79		29.45	11.45		27.43	9.43		18.00
41.67	2.76	31.48	16.28	17.90	27.45	12.25	12.85	27.17	11.97	12.10	15.20
50.00		29.12	17.32		25.21	13.41		26.94	15.14		11.80
58.33		29.74	24.34		22.89	17.49		26.74	21.34		5.40
25.00		35.65	10.85		32.29	7.49		27.29	2.49		24.80
33.33		33.52	9.62		30.40	6.50		26.94	3.04		23.90
41.67	4.14	31.48	10.18	11.46	28.39	7.09	8.28	26.65	5.35	6.65	21.30
50.00		29.45	11.25		26.33	8.13		26.39	8.19		18.20
58.33		27.40	15.40		24.21	12.21		26.17	14.17		12.00
41.67		31.20	8.70		28.59	6.09		26.24	3.74		22.50
50.00	5.52	29.33	10.53	10.82	26.62	7.82	8.11	25.97	7.17	7.48	18.80
58.33		27.44	13.24		24.62	10.42		25.73	11.53		14.20
50.00		29.07	10.37		26.61	7.91		25.62	6.92		18.70
58.33	6.90	27.29	13.09	11.73	24.67	10.47	9.19	25.37	11.17	9.05	14.20

Table 1.4 Output power of the models and experimental data (at $T_{hw} = 922$ K and $T_{cw} = 288$ K)

Rotation speed (Hz)	Mean pressure (MPa)	The simple analysis (variable P_r [10 ⁴])			The adiabatic analysis (simple Π [10 ⁷])			The proposed Stirling engine model			Experiment (kW) [110]
		Value (kW)	Error (%)	Average error (%)	Value (kW)	Error (%)	Average error (%)	Value (kW)	Error (%)	Average error (%)	
16.67		1.796	119.02		1.772	116.10		0.861	4.98		0.82
25.00		2.555	128.13		2.500	123.21		1.253	11.88		1.12
33.33		3.215	165.70		3.117	157.60		1.632	34.88		1.21
41.67	2.76	3.769	211.49	272.03	3.615	198.76	254.71	2.001	65.37	104.84	1.21
50.00		4.195	303.37		3.973	282.08		2.362	127.12		1.04
58.33		4.505	704.46		4.203	650.54		2.715	384.82		0.56
25.00		3.844	114.75		3.761	110.11		1.818	1.56		1.79
33.33		4.856	120.73		4.708	114.00		2.362	7.36		2.20
41.67	4.14	5.734	136.94	259.70	5.501	127.31	158.41	2.890	19.42	39.83	2.42
50.00		6.462	174.98		6.126	160.68		3.405	44.89		2.35
58.33		7.030	306.36		6.573	279.94		3.908	125.90		1.73
41.67		7.645	133.08		7.334	123.60		3.742	14.09		3.28
50.00	5.52	8.655	163.87	180.02	8.206	150.18	164.91	4.401	34.18	43.68	3.28
58.33		9.470	243.12		8.858	220.94		5.045	82.79		2.76
50.00		10.788	174.50		10.223	160.13		5.362	36.44		3.93
58.33	6.90	11.840	399.58	287.04	11.071	367.13	263.63	6.140	159.07	97.75	2.37

with $T(0) = T_i$, $T(A) = T_o$,

$$\frac{T_o - T_w}{T_i - T_w} = \exp\left(-\frac{UA}{q_m c_p}\right) \quad (1.50)$$

For a Stirling engine, T_{hw} or T_{cw} can be used to substitute T_w to get the relationships between $T_{i,h}$, $T_{o,h}$ and T_{hw} , or $T_{i,c}$, $T_{o,c}$ and T_{cw} respectively.

$$\frac{T_{o,h} - T_{hw}}{T_{i,h} - T_{hw}} = \exp\left(-\frac{U_h A_h}{q_{m,h} c_{p,h}}\right) \quad (1.51)$$

$$\frac{T_{o,c} - T_{cw}}{T_{i,c} - T_{cw}} = \exp\left(-\frac{U_c A_c}{q_{m,c} c_{p,c}}\right) \quad (1.52)$$

Heat transferred from heating fluid to Stirling engine in a cycle

$$q_{m,h} c_{p,h} (T_{i,h} - T_{o,h}) / s_{se} = Q \quad (1.53)$$

Heat transferred from Stirling engine to cooling fluid in a cycle

$$q_{m,c} c_{p,c} (T_{o,c} - T_{i,c}) / s_{se} = Q - W \quad (1.54)$$

1.1.4 Other parts

1.1.4.1 Steam generating system

The steam generating system can be divided into preheater, generator, superheater. They are all heat exchangers.

1.1.4.2 Power generating system

1.1.4.3 Condenser

1.1.4.4 Deaerator

1.2 Component connection

1.3 Determination of state parameters

1.4 Steam generating system

1.5 Steam extraction and regeneration system

1.6 Stirling engine array

Stirling engine array is used in the cascade system, Figure 1-6 shows the layout of the Stirling engine array. Each Stirling engine in the Stirling engine array has the identical parameters: $U_{se,1} = 30 \text{ W}/(\text{m}^2 \cdot \text{K})$, $U_{se,2} = 150 \text{ W}/(\text{m}^2 \cdot \text{K})$, $A_{se,1} = 6 \text{ m}^2$, $A_{se,2} = 6 \text{ m}^2$, $k_{se} = 1.4$, $\gamma_{se} = 3.375$, $n_g = 7.84 \times 10^{-2} \text{ mol}$, $s_{se} = 10 \text{ s}^{-1}$.

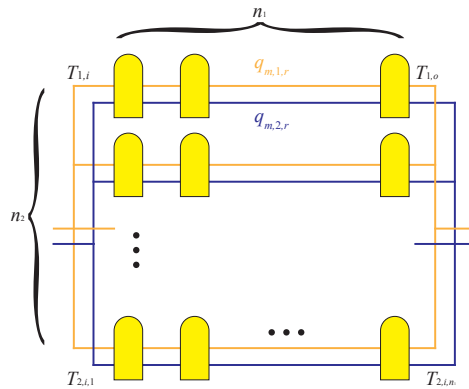


Figure 1-6 Layout of Stirling engines

Depending on the direction of heating and cooling flows, there are two possible flow types: parallel flow and counterflow. Figure 1-7b show the heat transfer diagrams of the two flow types.

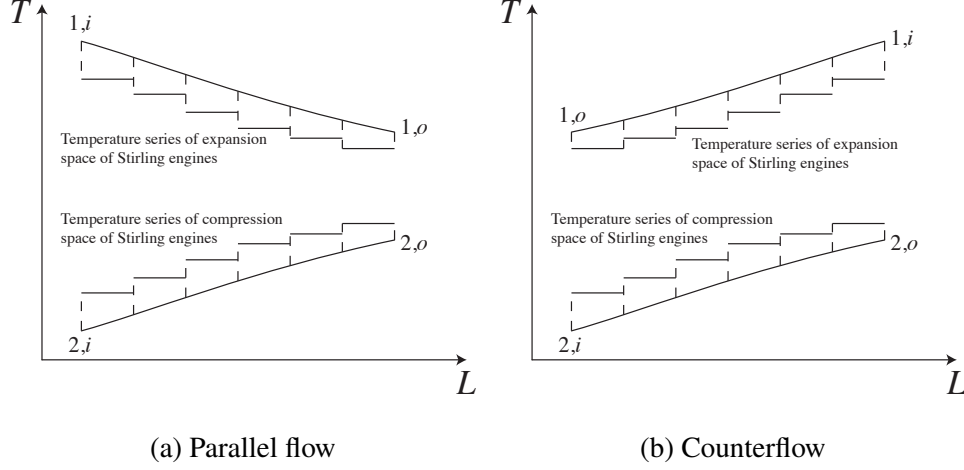


Figure 1-7 Heat transfer diagram of parallel flow and counterflow

By solving above equations, efficiency of the Stirling engine array η_{sea} can be obtained by $\eta_{sea} = 1 - \frac{\dot{m}_2(h_{2,o,n_1} - h_{2,i,1})}{\dot{m}_1(h_{1,i,1} - h_{1,o,n_1})}$, power of each Stirling engine in column x , $P_{se,x} = \dot{m}_{1,r}(h_{1,i,x} - h_{1,o,x})\eta_{se,x}$, and power of the Stirling engine array $P_{sea} = \eta_{sea}\dot{m}_1(h_{1,i,1} - h_{1,o,n_1})$.

1.7 Introduction

With the emphasis on energy and environmental impact, recently gaining attention is focused on Stirling engine for its high efficiency, low maintenance requirement, flexibility on energy sources, no pollution, low explosion risk. Due to its closed cycle, it can use almost any heat source, which makes it compatible with alternative and renewable energy sources.

The Stirling engine is widely used on solar dish system, known as dish-Stirling system. In a traditional dish-Stirling system, each Stirling engines is put on the focus of a parabolic dish to use the heat collected by the dish receiver for power generation. However, solar dish systems are not widely applied yet for its small capacity. Its capacity is mainly constrained by two factors: dish collector size and Stirling engine size. The dish size is limited for the cost and difficulty of production of large mirrors. The Stirling engine size is limited for its low power-to-weight ratio. The tracking feature of dish collector limits the weight of engine on the focal point. Besides, the size of engine is limited for it overlaps part of the collector.

This paper presents new arrangements for dish-Stirling system demonstrated in Fig-

ure 1-8 to solve the limitations. Stirling engines are put on the ground as a Stirling engine array (SEA). Heat collected by multiple dish collectors, or other types of collectors, or combination of different types of collectors, even heat from different heat sources is supplied to the SEA. Since the engines are static installations on the ground where space and weight are not at a premium any more, they can reach higher capacities. They can be connected in a different connection type and their performance may be improved compared to the traditional arrangement in which they are put on the focal points of each dish collector separately.

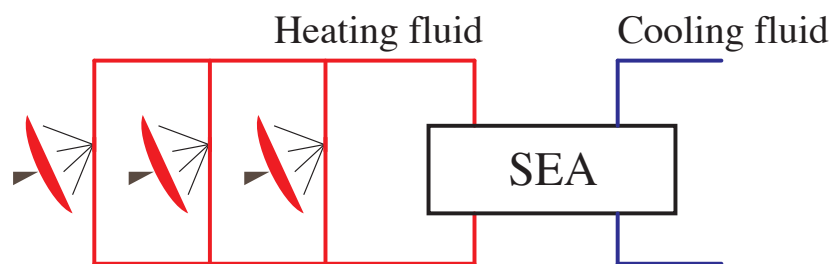


Figure 1-8 New arrangement of dish-Stirling system

Great attention has focused on the application using parabolic dish to collect heat. Some researchers investigated the impact of various parameters on the optical and thermal performance of the solar dish receivers using Monte Carlo Ray Tracing Method (MCRTM) and/or numerical modeling methods^[76–82]. Some researchers considered the applications with different ways to use the collected heat. Loni et al.^[83] considered a system using parabolic dish for an organic Rankine cycle. In the proposed system, thermal oil is used as the working fluid to transport the collected heat for the organic Rankine cycle. Wang et al.^[84] proposed an inverse design method for a cavity receiver used in solar dish Brayton system. Craig et al.^[85] evaluated a parabolic dish tubular receiver used in a dish Brayton cycle. An approach for incorporating a complex geometry like a tubular receiver generated using CFD software into SolTrace was developed. Aichmayer et al.^[86] presented a hybrid solar micro gas-turbine system integrating volumetric solar dish receiver. Concerning the solar dish receiver integration, both pressurized and atmospheric configurations have been considered. Lovegrove et al.^[87] presented an idea of using parabolic dish array to provide heat for ammonia based thermochemical energy storage. In this regard, using parabolic dishes to provide heat for SEA is applicable.

A large number of studies have been done on Stirling engine analysis. To describe a Stirling engine's behavior precisely is a difficult task due to the various losses and irreversibilities in the engine. Researchers have done a lot of work to build a precise Stirling engine model. Different models of Stirling engines were developed using empirical^[88–93], analytical^[94–103] and numerical methods^[104–114].

Among these methods, numerical methods obtain the most accurate models. Urieli and Berchowitz^[104] proposed an adiabatic model of Stirling engine considering some irreversible effects such as non-ideal heat transfer processes and pressure drop effect using numerical methods. The model is known as Simple model. Many researchers developed more accurate models based on the Simple model by using alternative methods or including more loss mechanisms. Ni et al.^[105] proposed an improved Simple analytical model which considers the influence mechanism of rotary speed, pressure and working gas in the view of heat/power losses for Stirling engine performance. Jia et al.^[106] developed a numerical model of free-piston engine generator. The dynamic equations have been linearized to simplify the model to a one-degree forced vibration system with various damping. The solving time of the proposed fast response model can be significantly reduced comparing to previous numerical models. Strauss and Dobson^[107] proposed an alternative method to calculate the regeneration heat loss and pumping losses, which is more suitable for preliminary engine design and optimization, known as Simple II model. Abbas^[108] considered the effects of non-ideal regeneration, shuttle loss and heat conduction losses based on Simple model. Araoz et al.^[109] developed a rigorous Stirling engine model with adiabatic working spaces, isothermal heat exchangers. It considers dead volumes, and imperfect regeneration, mechanical pumping losses due to friction, limited heat transfer and thermal losses on the heat exchangers. The model is suitable for different engine configurations (α , β and γ engines). Babaelahi and Syyaadi^[110] proposed a new numerical thermal model based on polytropic expansion/compression processes. Differential equations in the expansion/compression processes were modified to polytropic processes in the new model. New model shows a better performance prediction compared with previous models.

With the development of finite-time thermodynamics, many researchers studied the the finite-time thermodynamic performance of the Stirling engine. This analysis can also be used in the case of irreversible machines further considering the internal irreversibilities of a Stirling engine such as friction, pressure drop and entropy generation^[111]. Wu et al.^[112] de-

veloped a numerical model considering finite-time effect to find out the relationship between the net power output and thermal efficiency of the engine. Li et al.^[113] developed a mathematical model of a high temperature differential dish-Stirling system with finite-time thermodynamics. Finite-rate heat transfer, regenerative heat losses, conductive thermal bridging losses and finite regeneration processes of the Stirling engine were considered in the model. Hosseinzade^[114] presented a new closed-form thermal model for the thermal simulation of Stirling engines based on the combination of polytropic analysis of expansion/compression processes and the concept of finite speed thermodynamics. Instead of finite-time method, Ahmadi et al.^[115] proposed a finite-speed thermodynamic analysis based on the first law of thermodynamics for a closed system with finite speed and the direct method. The effects of heat source temperature, regenerating effectiveness, volumetric ratio, piston stroke as well as rotational speed are included in the analysis. Chen et al.^[116] developed a heat-engine cycle model using finite-time thermodynamics. The model, considered the losses due to heat-resistance, heat leaks and internal irreversibility, is applicable for generalized irreversible universal steady-flow heat-engine cycles.

On the other side of the researches, multi-objective optimization algorithms were used considering multi-variables to obtain a better performance was paid for attention by numbers of researchers recently^[117–120]. Ahmadi et al.^[117] performed the thermodynamic analysis of solar dish Stirling engine based on the finite-time thermodynamics. Then the NSGA-II algorithm was applied. Three objectives, thermal efficiency, entransy loss rate and power output, were set as the objectives and three well known decision making methods have been employed in the algorithm. Li et al.^[118] developed a multi-objective optimization model of a solar energy powered gamma type Stirling engine using Finite Physical Dimensions Thermodynamics (FPDT) method by multi-objective criteria. Genetic algorithm was used to get the Pareto frontier, and optimum points were obtained using the decision different making methods. Results show that total thermal conductance, hot temperature, stroke and diameter ratios can be improved. Patel and Savsani^[119] developed a strategy for multi-objective optimization for Stirling engines using TS-TLBO (tutorial training and self learning inspired teaching-learning-based optimization) algorithm. An application example with eleven decision variables and three objectives are considered. Luo et al.^[120] proposed a multiple optimization method that combines multiple optimization algorithms including differential evolution, genetic algorithm and adaptive simulated annealing. The optimization considers

five decision variables, including engine frequency, mean effective pressure, temperature of heating source, number of wires in regenerator matrix, and the wire diameter of regenerator for maximum efficiency and output power.

However, the literature review indicates that the analysis of arrangements of Stirling engines, classification and performance of the SEA, has not been reported till now. In this regard, this paper investigated the connection types of SEA and its influence on SEA performance. Connection types of SEA were classified and 5 basic connection types were put forward. According to Organ's theory^[121], one equivalent analytical Stirling engine model always exists for different types (α type, β type and γ type) of engines. To find out the influence of connection type of SEA and to avoid falling into the problem of developing specific Stirling engine model, a Stirling engine model based on some assumptions and simplifications was developed. This model was evaluated using experimental data of the General Motor GPU-3 Stirling engine prototype. Imperfect regeneration and some irreversibilities were considered. Heat transfer analysis of Stirling engine with heating and cooling fluids was also included. SEA models of different connection types were built depending on the engine model. Impacts of different parameters on the performance of these models were analyzed.

1.8 Connection types of SEA

For a single Stirling engine, the heat transfer processes between fluids and engine are independent and irrelevant with the direction of the flows, which means the efficiency and power are not affected by the direction of fluids. However, for an SEA, the connection type will affect the temperature profiles through the array and the specific work production, both of which will determine the efficiency and power of the SEA. It is practically significant to investigate the influence of connection type of an SEA on its performance. Then it is essential to classify the SEAs based on the connection type.

5 basic connection types of SEA were summarized according to the direction-irrelevant feature of Stirling engine, as shown in Figure 1-9. Type 1 is parallel flow, Type 2 is serial flows in the same order, Type 3 is serial flows in the reverse order, Type 4 is heating fluid in serial flow and cooling fluid in parallel flow and Type 5 is heating fluid in parallel flow and cooling fluid in serial flow.

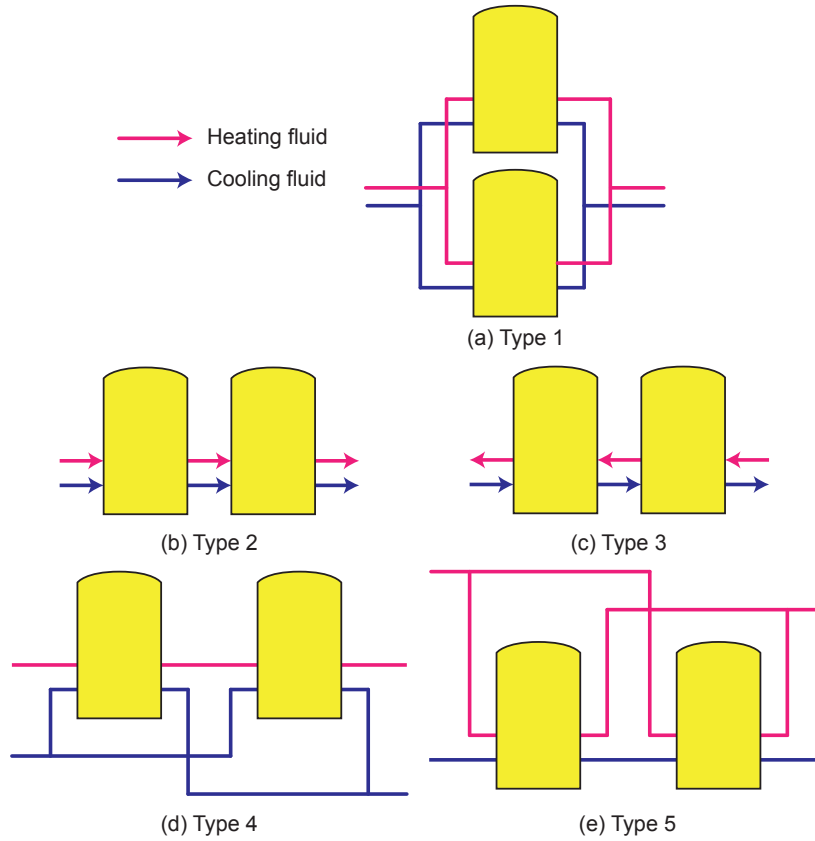


Figure 1-9 5 basic connection types of SEA

All other connection types are the combination of these 5 basic connection types. For instance, an SEA in Figure 1-10 is the combination of Type 2 and Type 4.

1.9 Thermodynamic analysis of Stirling engine

1.10 Modeling of the SEAs

To determine the performance of an SEA, models of all the Stirling engines need to be built depending on their thermodynamic characteristic. Stirling engines are chosen to have the same parameters including the same speed s_{se} . This is especially practical when using SEA for power generation, where the output power frequency should be constant. The speed of Stirling engine can be calibrated by speed controller system^[127]. To eliminate interference

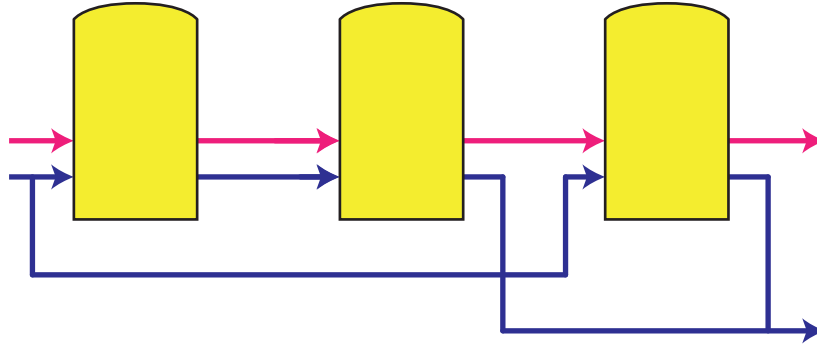


Figure 1-10 An instance of connection type of an SEA

of other factors, heating and cooling fluids are chosen to have same parameters for different connection types of SEAs. To clearly find out the performance differences of different SEAs, large temperature differences of the heating/cooling fluids after heat exchange with the engines are preferred. Air is chosen as the cooling fluid instead of commonly used water to avoid small temperature rise and evaporation after the cooling process. Chosen parameters of Stirling engines and heating/cooling fluids in SEAs are shown in Table 1.5.

Table 1.5 Parameters of SEA models

Parameter	Value	Parameter	Value
Heating fluid	Air	$q_{m,h}$	0.4 kg/s
Cooling fluid	Air	$T_{i,h}$	1000 K
n_{se}	6	$p_{i,h}$	5×10^5 Pa
s_{se}	25 Hz	$q_{m,c}$	0.4 kg/s
p_{se}	5 MPa	$T_{i,c}$	300 K
$U_h A_h$	180 W/K	$p_{i,c}$	5×10^5 Pa
$U_c A_c$	180 W/K		

In an SEA, there are 2 flows as shown in Figure 1-9. In a serial flow, each engine's mass flow rate is q_m , and from the flow's direction, for $2 \leq x \leq n_{se}$, $T_{i,x} = T_{o,x-1}$. In a parallel flow, each engine's mass flow is q_m/n_{se} , for $2 \leq x \leq n_{se}$, $T_{i,x} = T_{i,h}$.

MATLAB was used as the programming tool to build the model of SEAs, and CoolProp was used to provide fluid properties for MATLAB program. 5 basic SEA models composed of the aforementioned Stirling engines and fluids were built. To compare SEA connection types under various conditions, several parameters are investigated to find out their effects on SEA performance.

Figure 1-11 shows the solution algorithm of the SEA model. As it is shown, known inlet parameters of the fluids, the performance of a Stirling engine can be obtained from flowchart (a) according to the equations derived; similarly, known inlet parameters of heating fluid and outlet parameters of cooling fluid, the performance of a Stirling engine can be obtained from flowchart (b). Flowchart (c) shows the algorithm to solve the SEA model iteratively depending on different connection types. The levenberg-marquardt algorithm is applied to numerically solve the non-linear equations in the flowcharts.

1.11 Result Analysis

SEA models with specified parameters in Table 1.5 were built and calculated. Results are shown in Table 1.6, it can be found that under specified parameters Type 3 has the highest efficiency and output power, while Type 1 has the lowest efficiency and output power.

Table 1.6 Results of SEA models under specified parameters

Parameter	Value	Parameter	Value
η_1	0.2215	P_1	8022 W
η_2	0.2273	P_2	8483 W
η_3	0.2277	P_3	8512 W
η_4	0.2227	P_4	8116 W
η_5	0.2263	P_5	8399 W

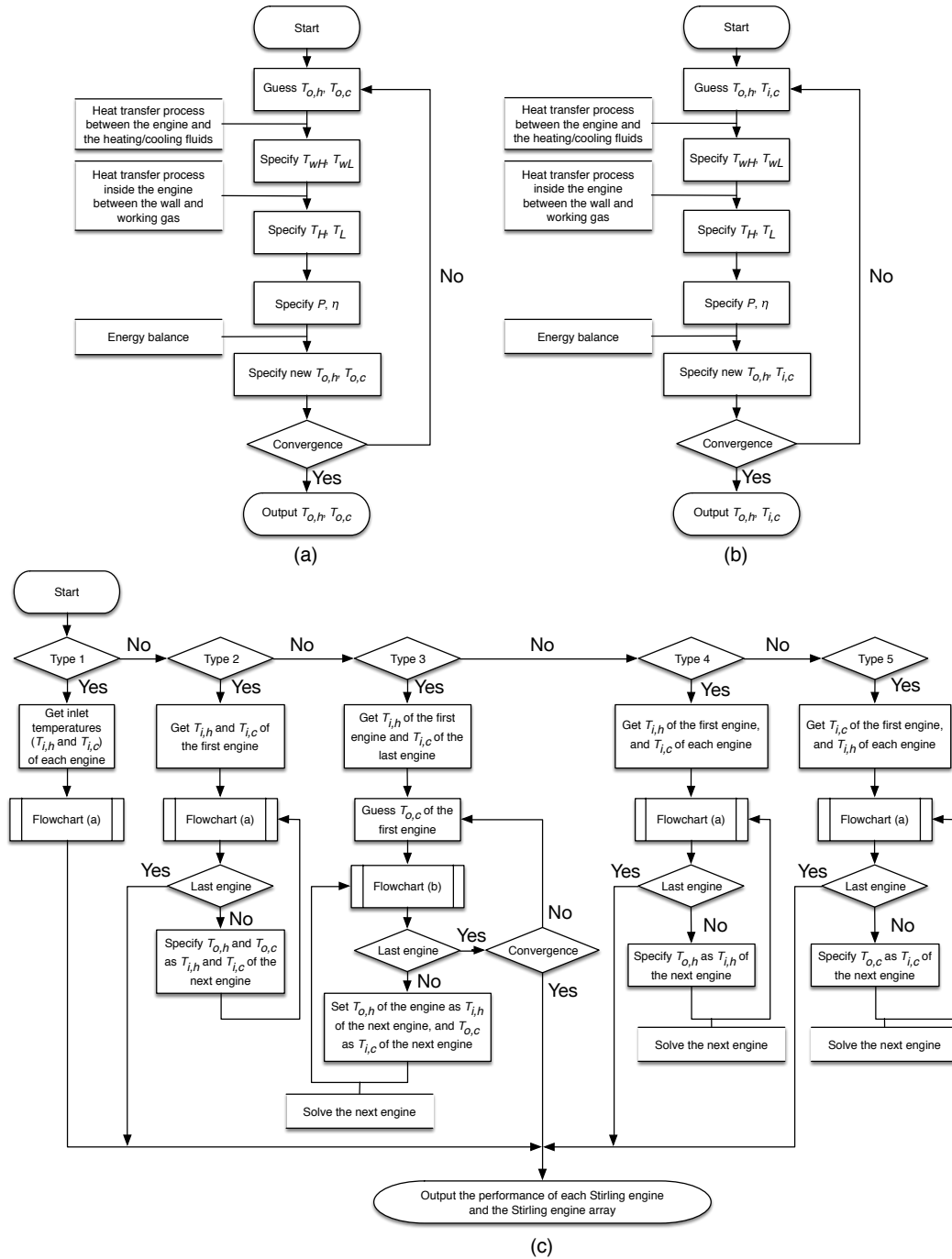


Figure 1-11 Flowcharts of the SEA model for performance analysis of the SEAs

1.11.1 Effectes of $T_{i,h}$

According to Carnot cycle efficiency formula, the temperature of heating fluid determines the efficiency of Stirling engine array. For a Stirling engine, lower temperature heating fluid leads to a lower efficiency. The efficiency and output power may drop to 0 due to its insufficient heating fluid temperature to drive the engine.

Curves of performance of SEAs and $T_{i,h}$ are shown in Figure 1-12. As it is shown, with the increase of $T_{i,h}$, both η and P increase for all SEAs. For some types of SEA, when $T_{i,h}$ is lower than a critical temperature, some of the engines in the SEA will not work and there will be turning points on the $\eta - T_{i,h}$, $P - T_{i,h}$ curves. E.g. for SEA of Type 1, when $T_{i,h}$ is lower 820 K, all the engines stop working, turning points at 820 K can be found on the $\eta - T_{i,h}$, $P - T_{i,h}$ curves in Figure 1-12.

From curves in Figure 1-12, it can be concluded that Type 2 and Type 3 have the best performance, and Type 2 has the best adaptability for lower $T_{i,h}$. All engines in Type 2 begin to work at 730 K.

1.11.2 Effects of $q_m c_p$

According to Equation 1.53, 1.54, $q_m c_p$ (both $q_{m,h} c_{p,h}$ and $q_{m,c} c_{p,c}$) will affect the heat transfer process, which is one of the vital factor for the performance of SEA.

Curves of performance of SEAs and $q_{m,h} c_{p,h}$ are shown in Figure 1-13. For a large $q_{m,h} c_{p,h}$ (> 800 W/K), Type 2, Type 3 and Type 5 have similar performance, which can be interpreted as the cooling fluid has the same properties for the two types of SEAs, and for a large $q_{m,h} c_{p,h}$, the heating fluid has similar effect after diverged. Similar performance of Type 1 and Type 4 can be also interpreted for the same reason.

Curves of performance of SEAs and $q_{m,c} c_{p,c}$ are shown in Figure 1-14. For a connection type of SEA, the performance improves with the increase of $q_{m,c} c_{p,c}$. For a large $q_{m,c} c_{p,c}$ (> 800 W/K), Type 2 and Type 3 have similar performance, which means the flow order doesn't affect the performance of SEA with a large $q_{m,c} c_{p,c}$. There exists an intersection point (at 830 W/K) of curves of Type 4 and Type 5. For a larger $q_{m,c} c_{p,c}$, Type 4 has a better performance, and vice versa. This can be interpreted that larger $q_{m,c} c_{p,c}$ weaken the drawback of larger

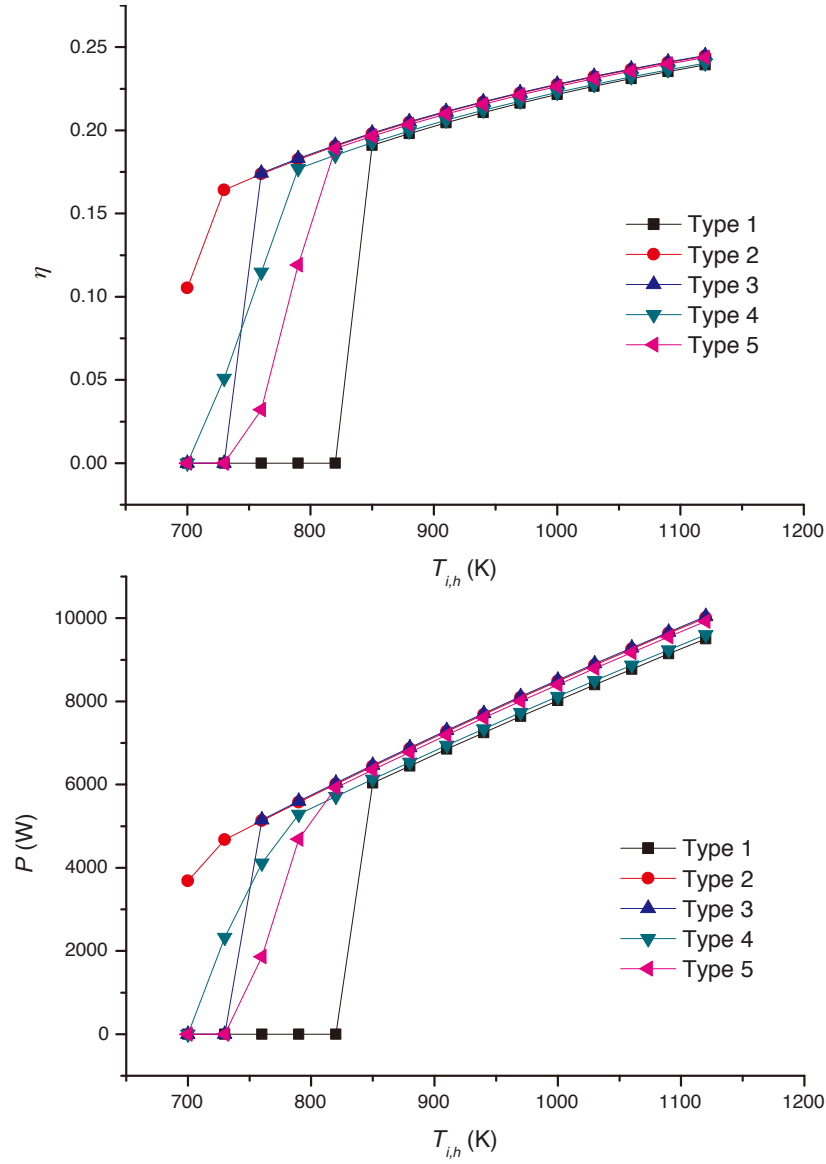


Figure 1-12 Influence of $T_{i,h}$ on efficiency and power of SEA

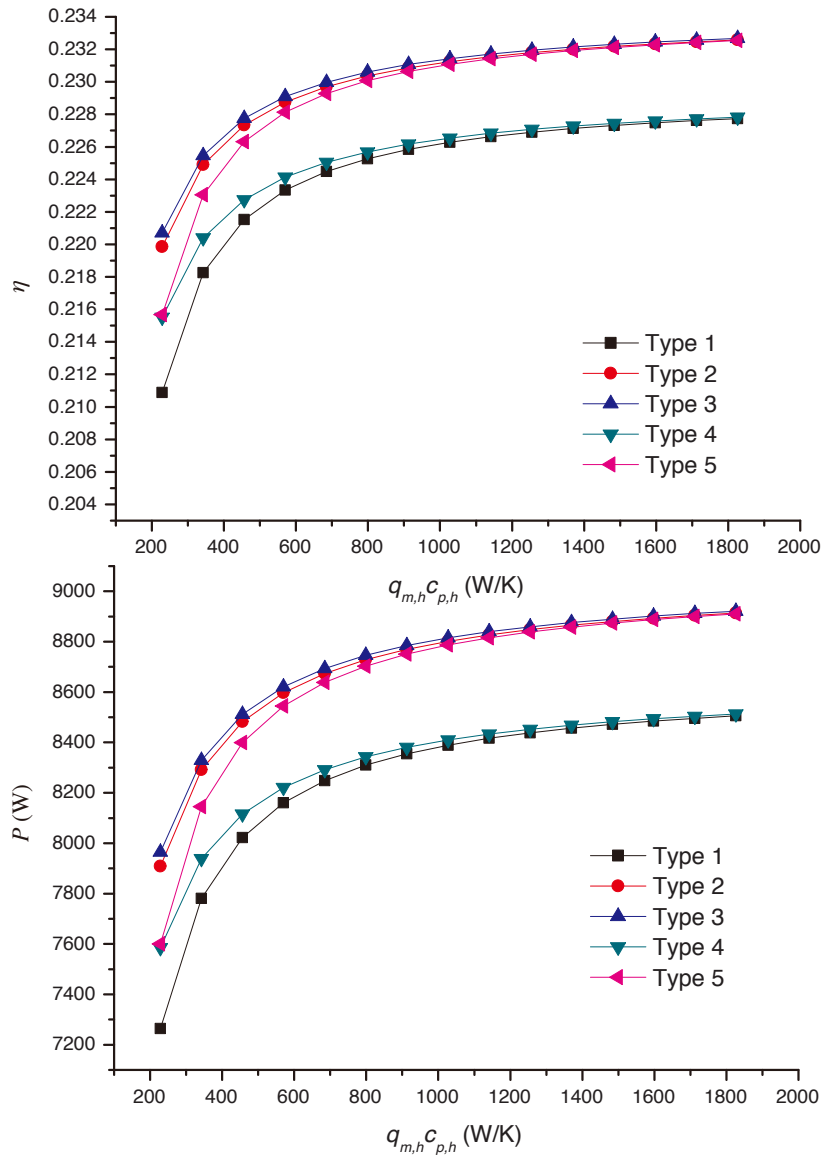


Figure 1-13 Influence of $q_{m,h}c_{p,h}$ on efficiency and power of SEA

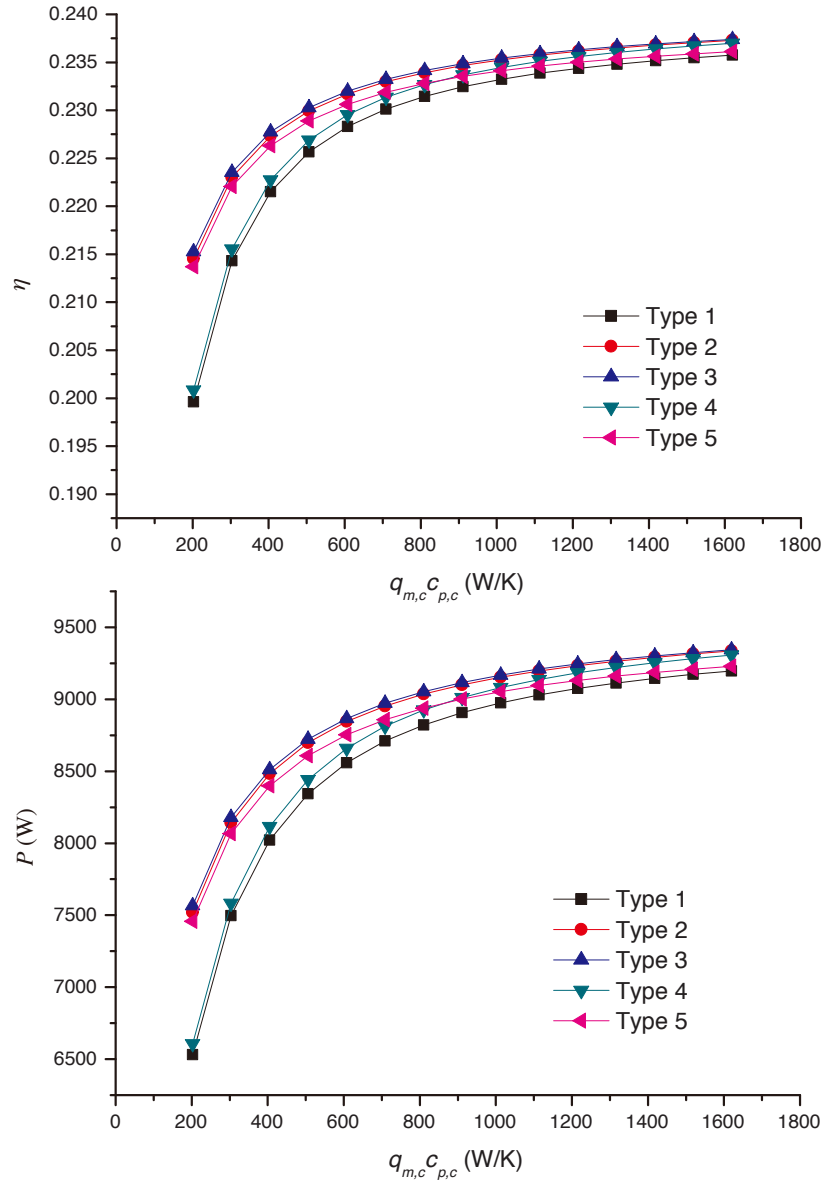


Figure 1-14 Influence of $q_{m,c}c_{p,c}$ on efficiency and power of SEA

temperature rise of parallel flow, while for the heating fluid, temperature drop of serial flow is smaller than parallel flow.

1.11.3 Effects of n_{se}

By varying the number of engines in SEA, the performance levels changed accordingly. n_{se} may affect both the flow rates and temperatures of fluids of each engine. Figure 1-15 shows curves of performance of SEAs with different n_{se} . As it is shown, with an increase of n_{se} leads to a reduction of η for all SEAs due to smaller heating and cooling average temperature difference for more engines. For some types of SEA, when n_{se} is larger than a critical value, some of the engines in the SEA will not work and the curves will dive. E.g. for SEA of Type 1, when n_{se} is larger than 9, all the engines stop working, turning points at 9 can be found on the $\eta - n_{se}$, $P - n_{se}$ curves in Figure 1-15.

For Type 1, when $n_{se} \geq 10$, all engines stop working for given heating and cooling fluids due to small $q_m c_p$. For Type 2 and Type 3, every engine in the SEAs works, by increasing n_{se} , η reduces due to smaller temperature difference of the fluids, and P increases due to more operating engines. For Type 4, by checking results, it can be found that when $n_{se} = 13$, the last engine doesn't work; when $n_{se} = 14$, only the first 10 engines will work; when $n_{se} = 15$, the working engine number drops to 9. For Type 5, by checking results, it can be found that when $n_{se} = 12$, the last 2 engines stop working; when $n_{se} = 13$, only the first 8 engines will work; when $n_{se} = 14$, the working engine number drops to 6; when $n_{se} = 15$, the working engine number drops to 4.

1.12 Conclusion

Connection type may change the flow rates and temperatures of the fluids, as a result the performance of the SEA will be different depending on the connection schemes. In order to compare performance of SEAs with different connections, five basic connection types of SEA were summed up according to flow type and flow order.

Analytical Stirling engine model was created to develop the SEA models for the investigation of influence of connection types. Imperfect regeneration and cycle irreversibility of Stirling engine cycle and heat exchange process between fluids and engine were considered in the model. The model was evaluated by considering the prototype GPU-3 Stirling engine

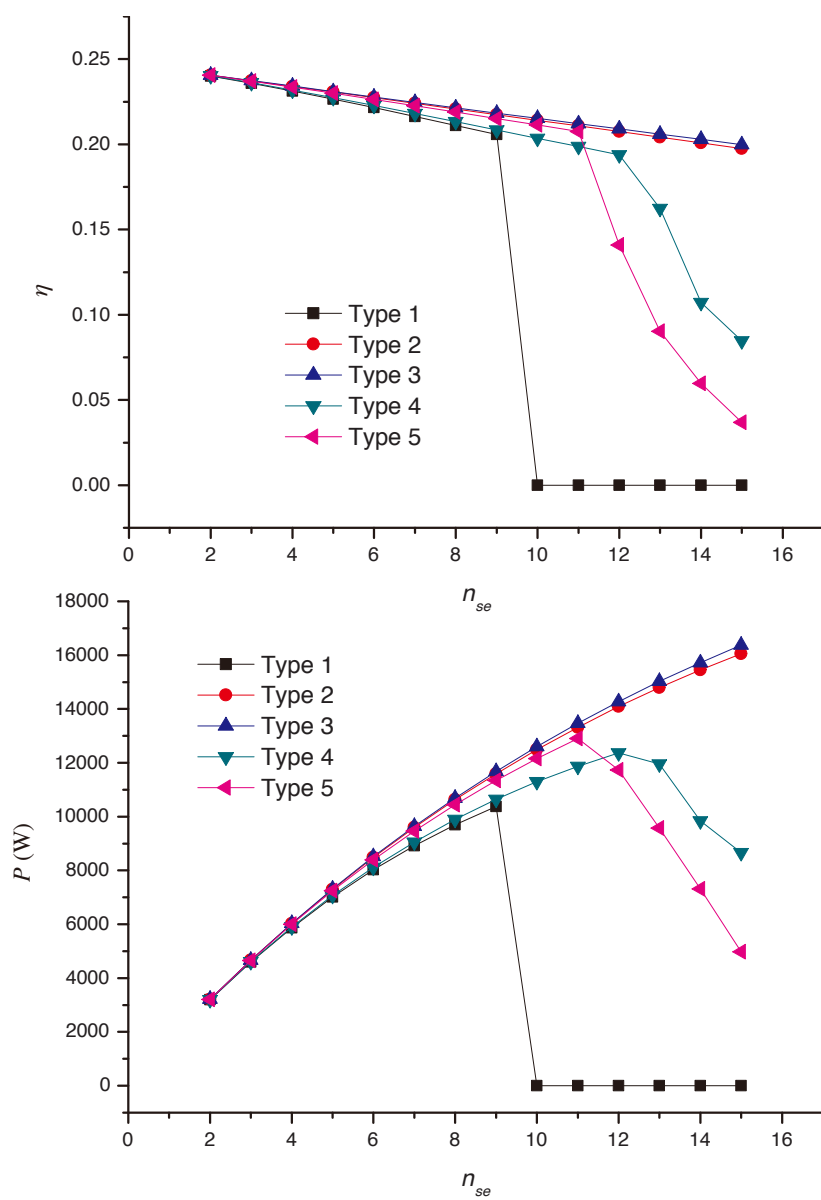


Figure 1-15 Influence of n_{se} on efficiency and power of SEA

as a case study. Result shows that the proposed model predicted the performance with higher accuracy than the previous models. Models of SEAs were developed to calculate the performance under different parameters to find out impacts of $T_{i,h}$, $q_{m,h}c_{p,h}$, $q_{m,c}c_{p,c}$ and n_{se} on different connection types.

It was found that, as expected, decrease $T_{i,h}$ and $q_{m}c_p$ will weaken the performance of SEA of all connection types. However, for some connection types, there exists a critical temperature below which some engines stop working. This needs to be considered for SEA connection type selection, especially when $T_{i,h}$ is low. For given $T_{i,h}$, $T_{i,c}$ and $q_{m}c_p$, Type 2 has the best performance and adaptability.

From Figure 1-12-1-15, it can be found that Type 2 and Type 3 have the best performance, while Type 1 has the worst performance. Type 2 and Type 3 have similar performance under different parameters ($T_{i,h}$, $T_{i,c}$ and $q_{m}c_p$), which means the flow order has little influence on the performance of an SEA.

For a certain connection type, increase n_{se} will reduce the efficiency of SEA. For some connection types, increase n_{se} will reduce the output power P due to inoperative engines and smaller output power engines. Thus it is important to choose the number of engines for some connection types of SEA.

As a conclusion, SEA of serial flows has the best performance and adaptability under different parameters. Given heating and cooling fluids, using serial flow is the best choice for the connection type of an SEA.

It is important to note that, in the future researches, the experiments of influence of connection type on SEA's performance can be carried out to verify the conclusions in this paper.

1.13 Steam Generators

Acknowledge

This is the acknowledgement part.

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Appendix A Heat Transfer Under Constant Temperature

Assuming U , T_c , q_m, c_p to be constant, for given T_i ,

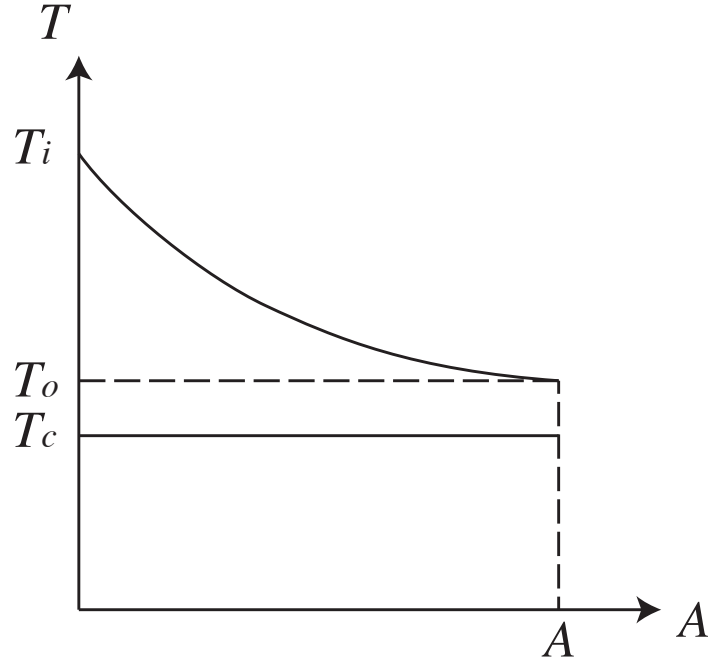


Figure 1-1 Diagram of heat transfer under constant temperature

For $A(x) = Px$, x from 0 to L , while $T(x)$ from T_i to T_o ,

$$q_m c_p dT(x) = (T_c - T(x))UPdx \quad (\text{A.1})$$

so,

$$\frac{dT(x)}{dx} = -\frac{UP}{q_m c_p} (T(x) - T_c) \quad (\text{A.2})$$

$$T_g(x) = T_p(x) + T_h(x) \quad (\text{A.3})$$

where $T_g(x)$ is the general solution, $T_p(x)$ is the particular solution, $T_h(x)$ is the homogeneous solution.

$$-\frac{UP}{q_m c_p}(T_p(x) - T_c) = 0 \quad (\text{A.4})$$

$$T_p(x) = T_c \quad (\text{A.5})$$

$$\frac{dT_h(x)}{dx} = -\frac{UP}{q_m c_p}T_h(x) \quad (\text{A.6})$$

$$\int_{T_h(x)=T_h(0)}^{T_h(x)=T_h(L)} \frac{dT_h(x)}{T_h(x)} = -\int_{x=0}^{x=L} \frac{UP}{q_m c_p} dx \quad (\text{A.7})$$

$$\frac{T_h(L)}{T_h(0)} = \exp\left(-\frac{UPL}{q_m c_p}\right) \quad (\text{A.8})$$

that is

$$\frac{T_g(L) - T_p(L)}{T_g(0) - T_p(0)} = \exp\left(-\frac{UA}{q_m c_p}\right) \quad (\text{A.9})$$

$$\frac{T_o - T_c}{T_i - T_c} = \exp\left(-\frac{UA}{q_m c_p}\right) \quad (\text{A.10})$$

Appendix B Thermal gradient under constant heat flux

Assuming U , T_c , \dot{m} , c_p , q'' to be constant, for given T_i ,

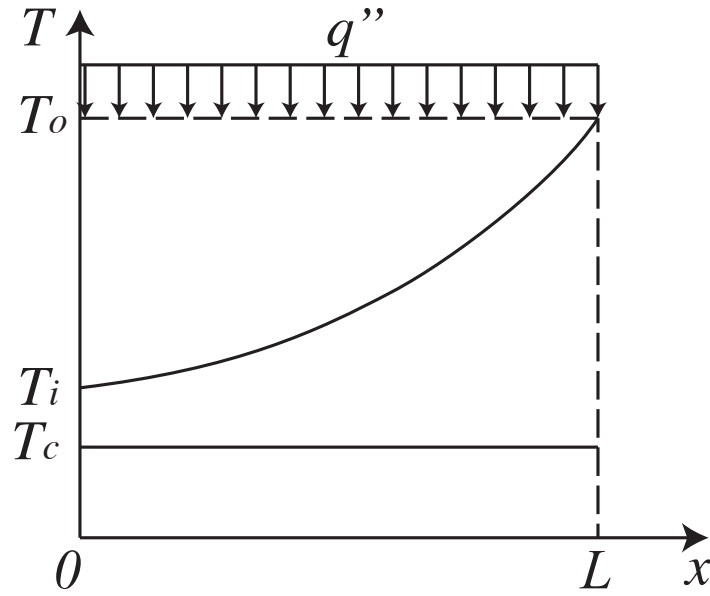


Figure 2-1 Diagram of heat transfer with one constant temperature heat source and constant heat flux

For $A(x) = Px$, x from 0 to L , while $T(x)$ from T_i to T_o ,

$$q_m c_p dT(x) = (T_c - T(x))UPdx + q''Px \quad (\text{B.1})$$

so

$$\frac{dT(x)}{dx} = -\frac{UP}{q_m c_p}T(x) + \frac{q''P + UPT_c}{q_m c_p} \quad (\text{B.2})$$

$$T_g(x) = T_p(x) + T_h(x) \quad (\text{B.3})$$

where $T_g(x)$ is the general solution, $T_p(x)$ is the particular solution, $T_h(x)$ is the homogeneous solution.

$$-\frac{UP}{q_m c_p} T_p(x) + \frac{q''P + UPT_c}{q_m c_p} = 0 \quad (\text{B.4})$$

$$T_p(x) = T_c + \frac{q''}{U} \quad (\text{B.5})$$

$$\frac{dT_h(x)}{dx} = -\frac{UP}{q_m c_p} T_h(x) \quad (\text{B.6})$$

the same as Equation (A.6), so we have

$$\frac{T_g(L) - T_p(L)}{T_g(0) - T_p(0)} = \exp\left(-\frac{UA}{q_m c_p}\right) \quad (\text{B.7})$$

$$\frac{T_o - T_c - \frac{q''}{U}}{T_i - T_c - \frac{q''}{U}} = \exp\left(-\frac{UA}{q_m c_p}\right) \quad (\text{B.8})$$

Appendix C Publication

- [1] Zhang Cheng, Kun Wang. International Conference on Power Engineering: ICOPE 2013: FEA simulation on the alignment of the shafts of three-fulcrum turbine.
- [2] Performance comparison of new and traditional arrangements of a dish-Stirling system
- [3] A multi-stage exergy-loss reduction system for solar parabolic trough power plants
- [4] Zhang Cheng, Zhang Yanping, Arauzo Inmaculada, Gao Wei, Zou Chongzhe. Cascade system using both trough system and dish system for power generation. *Energy Convers Manag* 2017;142:494–503. doi:10.1016/j.enconman.2017.03.073.
- [5] Thermal Modeling of a Pressurized Air Cavity Receiver for a Solar Dish Stirling System
- [6] A solar thermal cascade system, No. 201610806296.5
- [7] A flow control method used in a multi-stage heating system, No. 201610805604.2

Appendix D Formulae

The is the content of the Appendix B