```json
 1  {
 2    "flow_conditions" : {          Wind tunnel conditions
 3      "reynolds" : "1.85e6 - 4.0e6",   Reynolds number, provided or calculated
 4      "mach" : "0.3 - 0.83",           Mach number, provided or calculated
 5      "freestream": "",                Freestream velocity [m/s], provided or calculated
 6      "alpha": "0, 2, 3, 4, 6, 8"      Geometric angles of attack [deg]
 7    },
 8    "uncertainty" : {          Experimental uncertainties, 1σ unless stated otherwise
 9      "x" : "",                        Uncertainty in pressure orifice location [x/c]
10      "cp" : "0.005 + 0.01|Cp|",       Uncertainty in pressure coefficient
11      "alpha": "0.04",                 Uncertainty in angle of attack [deg]
12      "mach": "0.003"                  Uncertainty in Mach number
13    },
14    "airfoil": {          Airfoil information
15      "name": "NACA 0012",             Airfoil name
16      "camber": "N",                   Is the airfoil cambered? "Y" or "N"
17      "supercritical": "N",            Is the airfoil supercritical? "Y" or "N"
18      "application": "General"         Airfoil applications: "General", "Rotor", "Wind Tunnel", etc.
19    },
20    "source": {          Source material
21      "name": "AGARD-AR-138",          Source name
22      "year": "1979",                  Year of publication
23      "page": "A1-A13",                Pages of where the data was presented
24      "type": "tabulated",             Original data format: "graph", "tabulated", etc.
25      "url": "https://www.sto.nato.int/publications/AGARD/AGARD-AR-138/AGARD-AR-138.pdf"   URL of source PDF file
26    },
27    "meta" : {          Information related to data digitization
28      "num_cases": "19",               Number of digitized pressure distributions
29      "num_remaining": "",             If not finished, how many remain?
30      "notes": ""                      Any points of note
31    }
32  }
```

Annotations (descriptions shown to the right of each line):

- Line 2 `"flow_conditions"` — Wind tunnel conditions
- Line 3 `"reynolds"` — Reynolds number, provided or calculated
- Line 4 `"mach"` — Mach number, provided or calculated
- Line 5 `"freestream"` — Freestream velocity [m/s], provided or calculated
- Line 6 `"alpha"` — Geometric angles of attack [deg]
- Line 8 `"uncertainty"` — Experimental uncertainties, 1σ unless stated otherwise
- Line 9 `"x"` — Uncertainty in pressure orifice location [x/c]
- Line 10 `"cp"` — Uncertainty in pressure coefficient
- Line 11 `"alpha"` — Uncertainty in angle of attack [deg]
- Line 12 `"mach"` — Uncertainty in Mach number
- Line 14 `"airfoil"` — Airfoil information
- Line 15 `"name"` — Airfoil name
- Line 16 `"camber"` — Is the airfoil cambered? "Y" or "N"
- Line 17 `"supercritical"` — Is the airfoil supercritical? "Y" or "N"
- Line 18 `"application"` — Airfoil applications: "General", "Rotor", "Wind Tunnel", etc.
- Line 20 `"source"` — Source material
- Line 21 `"name"` — Source name
- Line 22 `"year"` — Year of publication
- Line 23 `"page"` — Pages of where the data was presented
- Line 24 `"type"` — Original data format: "graph", "tabulated", etc.
- Line 25 `"url"` — URL of source PDF file
- Line 27 `"meta"` — Information related to data digitization
- Line 28 `"num_cases"` — Number of digitized pressure distributions
- Line 29 `"num_remaining"` — If not finished, how many remain?
- Line 30 `"notes"` — Any points of note