

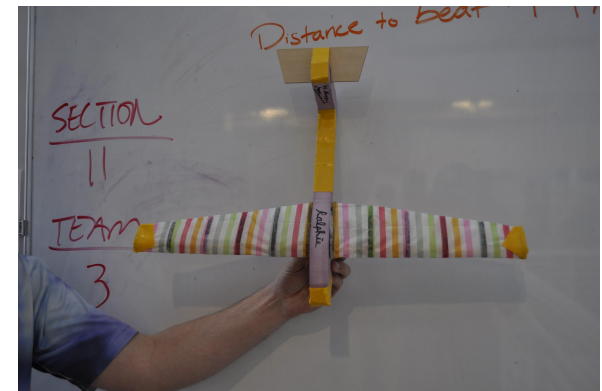
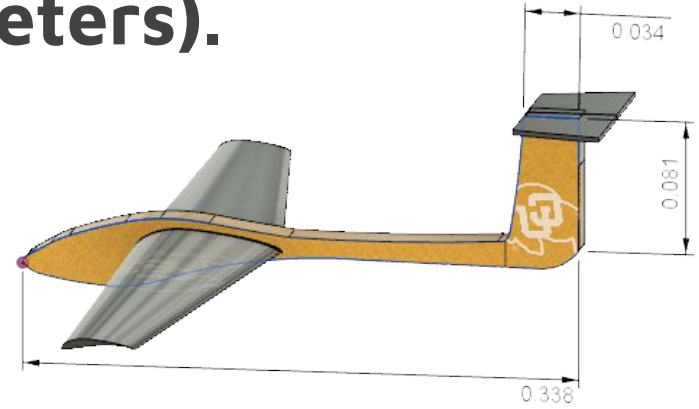
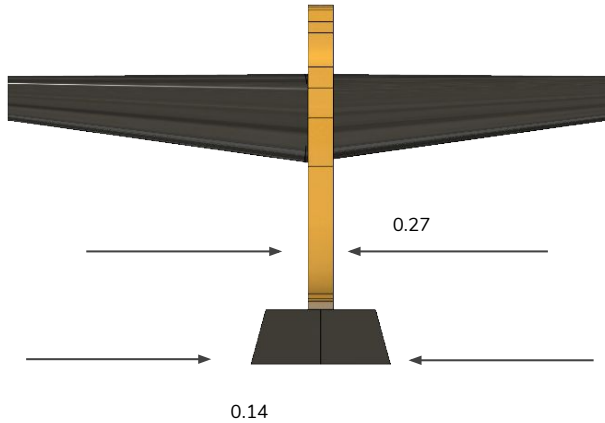
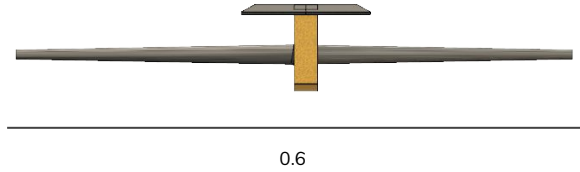
Glider Final Design Package

Design and Presentation by:

Madison Dube, Alexander Larson, James
Guthrie, Samuel D'Souza, Abdullah
Almugairin, Abdulla Al Ameri

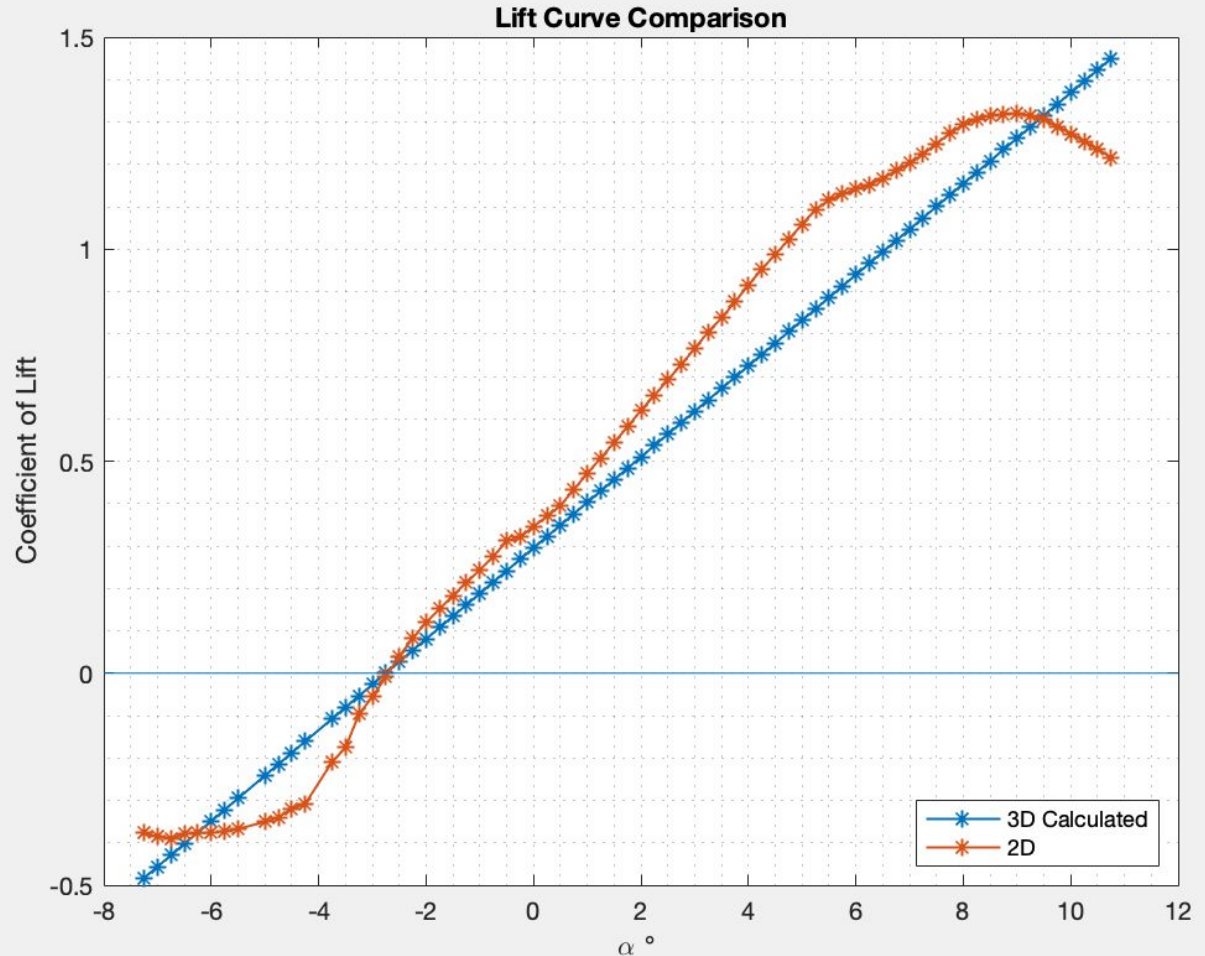


Finalized Design: (in meters).

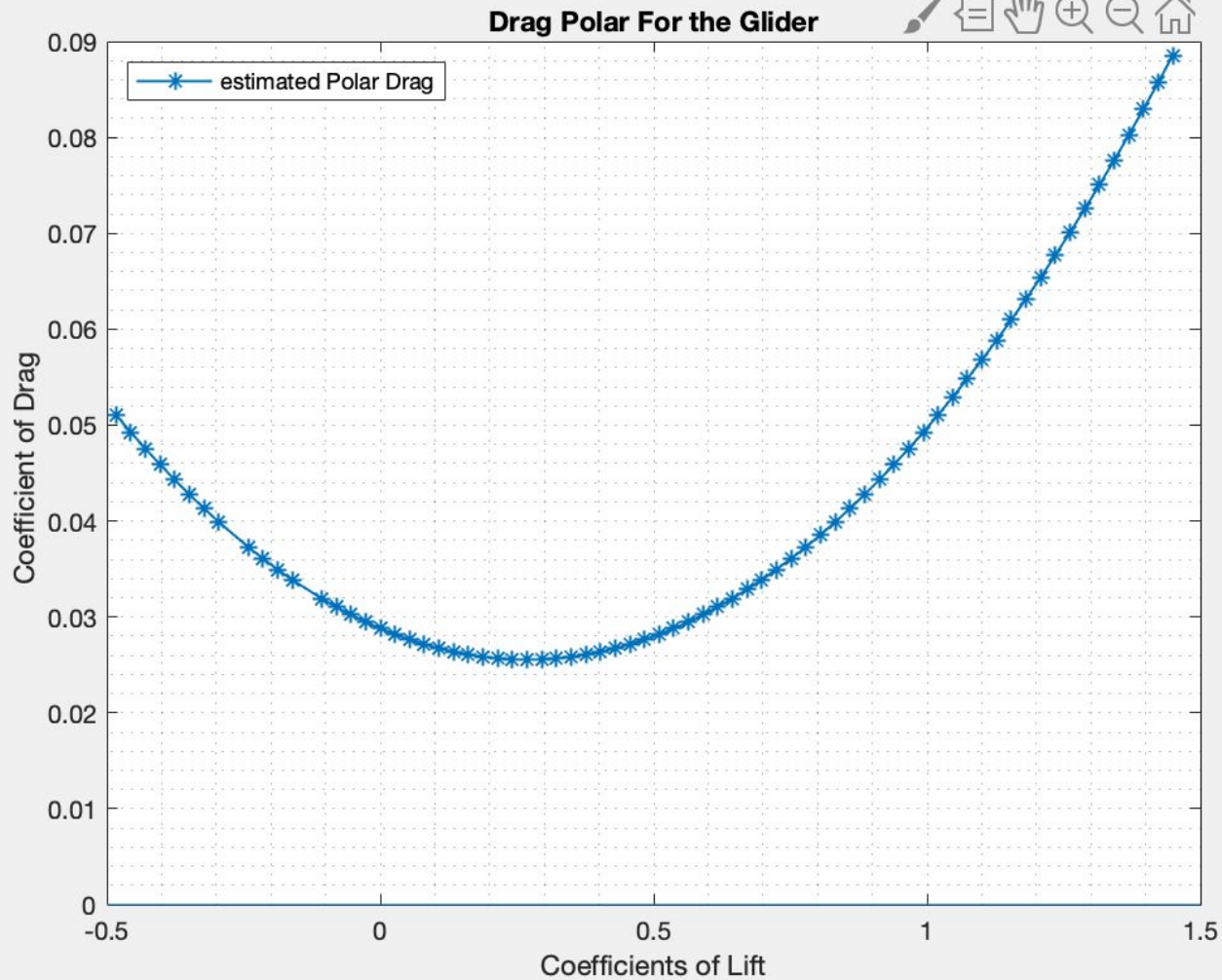


Lift curve

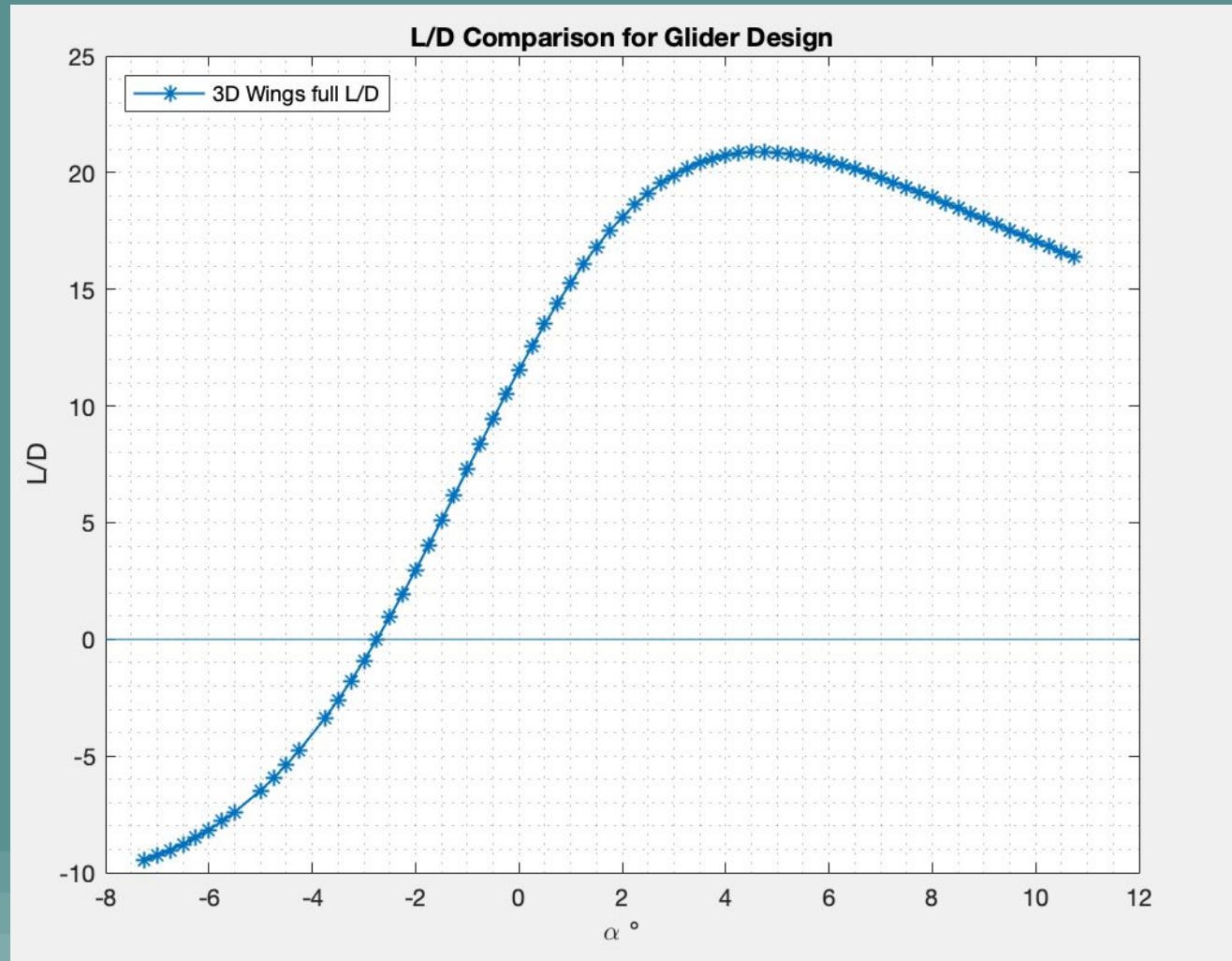
- Airfoil: S1223 (s1223-il)
- $Re = 50,000$



Drag Polar



L/D Curve



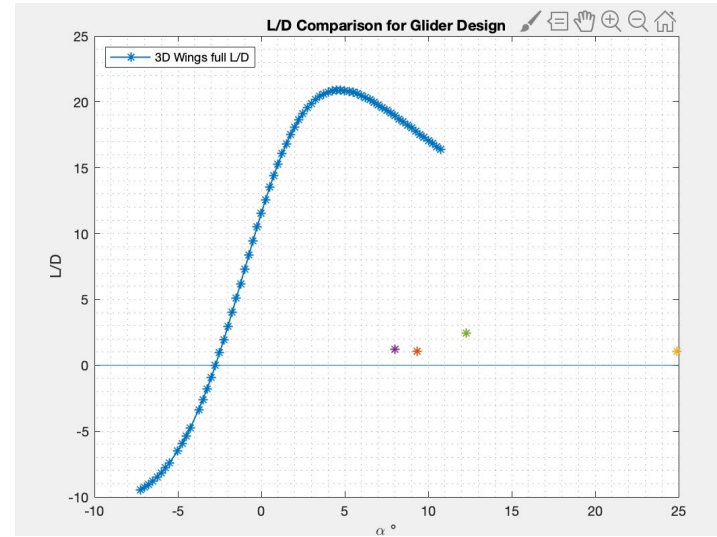
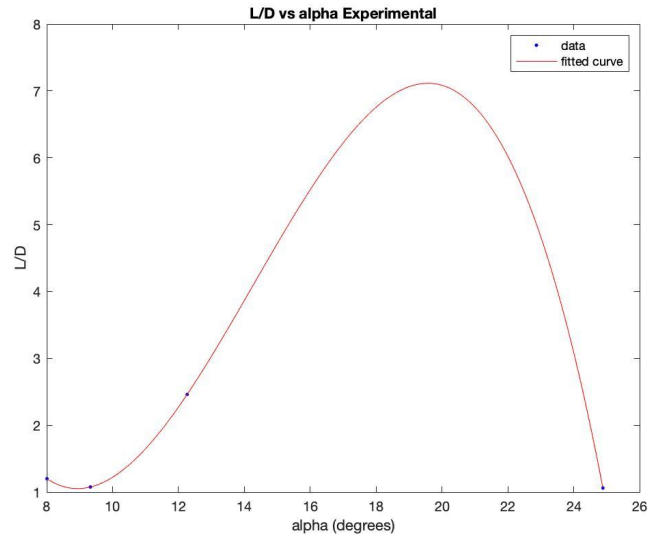
$$\left(\frac{L}{D}\right)_{max} = \left(\frac{C_L}{C_D}\right)_{max} = \frac{R_{max}}{h}$$

$$V_{trim} = \frac{S_{flight}}{t_{flight}}$$

$$C_L = \frac{W_{measured}}{\frac{1}{2} \cdot \rho \cdot V_{trim}^2 \cdot S_{ref}}$$

$$V_{trim} = \frac{\sqrt{h^2 + R^2}}{t_{flight}}$$

$$C_L = a \cdot (\alpha - \alpha_{L=0})$$



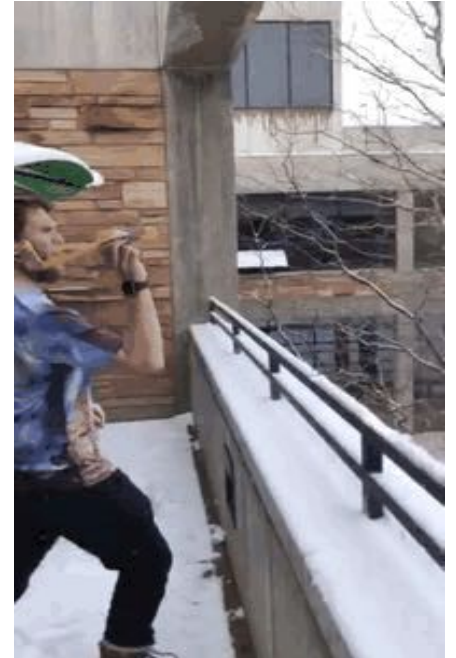


System Requirements 5	Threshold	Design	Flight
Max Glide Range (m)	40 - 80	133	16.1
Max Glide Range Velocity (m/s) (L/D max)	3 - 7	4.68	2.12
Stall Velocity (m/s)		4.5 (based on theoretical CL)	
Elevator Pitch Control (degrees)	+/- 8	+/- 15	+/- 10
Longitudinal Stability (xcg location)	0.0c - 0.3c	0.25c	0.2c
Longitudinal Stability (horizontal tail volume)	0.3 - 0.6	0.553	.585
Lateral Stability (Vert tail volume/spiral)	0.02 - 0.05	.038	0.363
Maximum Wingspan (m)	0.9	0.6	0.555
Total Mass (kg)	0.040 - 0.080	0.053	0.063
Unit Cost (dollars)	20 - 30	15	28



Glider Performance

Investigating test flights of our plane we can analyze what went wrong with our calculations and furthermore the build





Valuable Lessons

- How not to compromise total design to fit one design parameter
- More iterations of manufacturing
- Rethink fuselage design
- Wrong calculated trim velocity
- Adding more tape to the tail, have a checklist for each flight
- Theoretical Calculations go out the door once you start manufacturing
- Plan for manufacturing to some degree initially, and work on refining the design through theoretical calculations

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