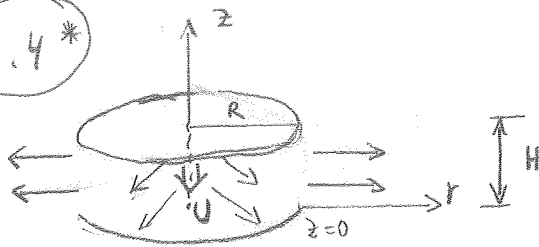


1.4 *

A plate moves with U $\bar{u} = ?$ (assume that u_r independent of z)flow incompressible $\hat{=}$ $\rho = \text{const.}$

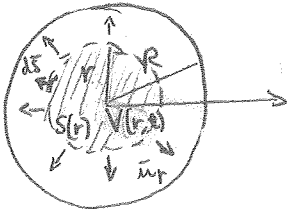
Solution:

$$\frac{dV}{dt} = - (V - \text{flux out})$$

Looking from above:

$$V(r, t) = \pi r^2 (H_0 - Ut)$$

$$\Rightarrow \frac{dV}{dt} = - \pi r^2 U \quad (1)$$

Total V -flux through $S(r)$

$$Q = \iint_S \bar{u} \cdot d\bar{s} = u_r \cdot 2\pi r (H_0 - Ut) \quad (2)$$

Now (1) & (2) \Rightarrow

$$- \pi r^2 U = - u_r \cdot 2\pi r (H_0 - Ut)$$

$$\Leftrightarrow u_r = \frac{Ur}{2(H_0 - Ut)}$$

 u_z is obtained from $\nabla \cdot \bar{u} = 0$

$$\frac{1}{r} \frac{\partial}{\partial r} (r u_r) + \frac{\partial u_z}{\partial z} = 0$$

$$\Rightarrow \frac{\partial u_z}{\partial z} = - \frac{1}{r} \frac{\partial}{\partial r} \left[\frac{U r^2}{2(H_0 - Ut)} \right] = - \frac{U}{H_0 - Ut}$$

$$\Rightarrow u_z = - \frac{Uz}{H_0 - Ut} + \text{const.}$$

$$\text{Boundary conditions: } \begin{cases} u_z(0, t) = 0 \\ u_z(H_0 - Ut, t) = -U \end{cases}$$

$$\Rightarrow \text{const.} = 0$$

$$\Rightarrow u_z = - \frac{Uz}{H_0 - Ut}$$

1.5

A model of 2-D "stationary" turbulent flow, $\bar{u} = (u_x, u_y)$

$$u_x = \sum_i A_i \cos(k_i x) \sin(k_i y)$$

a) $u_y = ?$

b) P (for one mode)

Solution:

a) If only one mode \Rightarrow

$$u_x = A \cos(kx) \sin(ky)$$

$$\text{incomp. } (s = \text{const.}) \Rightarrow \nabla \cdot \bar{u} = 0 \Leftrightarrow \frac{\partial u_x}{\partial x} + \frac{\partial u_y}{\partial y} = 0$$

$$\Rightarrow \frac{\partial u_y}{\partial y} = + k A \sin(kx) \sin(ky)$$

$$\Rightarrow u_y = -A \sin(kx) \cos(ky) \left(+ C(x) \right)$$

= 0, elementary mode we sum up

All modes

$$u_y = - \sum_i A_i \sin(k_i x) \cos(k_i y) \leftarrow \text{all elementary modes}$$

b) Pressure for 1 mode

$$\text{Euler equation } \Rightarrow (\bar{u} \cdot \nabla) \bar{u} = - \frac{1}{\rho} \nabla P$$

$$\Rightarrow \begin{cases} u_x \frac{\partial u_x}{\partial x} + u_y \frac{\partial u_x}{\partial y} = - \frac{1}{\rho} \frac{\partial P}{\partial x} \end{cases} \quad (1)$$

$$\begin{cases} u_x \frac{\partial u_y}{\partial x} + u_y \frac{\partial u_y}{\partial y} = - \frac{1}{\rho} \frac{\partial P}{\partial y} \end{cases} \quad (2)$$

$$\Rightarrow \begin{cases} -kA^2 \cos(kx) \sin(kx) \sin^2(ky) - kA^2 \cos(kx) \sin(kx) \cos^2(ky) = - \frac{1}{\rho} \frac{\partial P}{\partial x} \\ -kA^2 \cos^2(kx) \cos(ky) \sin(ky) - kA^2 \sin^2(kx) \cos(ky) \sin(ky) = - \frac{1}{\rho} \frac{\partial P}{\partial y} \end{cases}$$

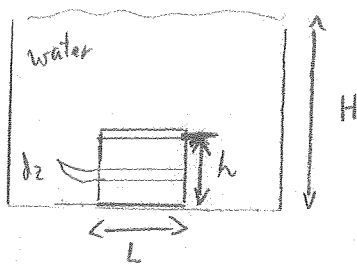
$$\text{Since } \sin(kx) \cos(kx) = \frac{1}{2} \sin(2kx)$$

$$\Rightarrow \begin{cases} \frac{\partial P}{\partial x} = \frac{1}{2} \rho A^2 k \sin(2kx) \\ \frac{\partial P}{\partial y} = \frac{1}{2} \rho A^2 k \sin(2ky) \end{cases} \Rightarrow \begin{cases} P = -\frac{1}{4} \rho A^2 \cos(2kx) + f_1(y) \\ P = -\frac{1}{4} \rho A^2 \cos(2ky) + f_2(x) \end{cases}$$

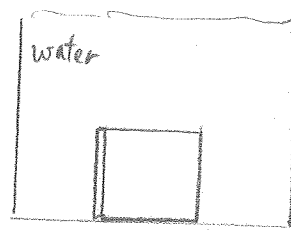
$$\Rightarrow P = -\frac{1}{4} \rho A^2 [\cos(2kx) + \cos(2ky)]$$

1.8

a)



b)



Find F and T acting on the gate. P_{atm} is neglected.

Solution:

Force on the gate caused by water pressure in both of the cases a and b.

$$d\vec{F} = -P d\vec{S} \Rightarrow \vec{F} = - \int_{S_0} P d\vec{S} \quad , \quad \frac{dP}{dz} = -\rho g \Rightarrow P = -\rho g z (+P_0)$$

$$d\vec{S} = +L dz \hat{e}_y \quad \leftarrow P_{atm}$$

$$\Rightarrow d\vec{F} = (-1) \cdot (-\rho g z) \cdot (+L dz) \hat{e}_y \quad z < 0 \Rightarrow d\vec{F} \uparrow \uparrow -\hat{e}_y$$

surface of the container \rightarrow gate

$$\Rightarrow F = - \int_{-H}^{-H+h} \rho g L z dz = -\rho g L \left[\frac{z^2}{2} \right]_{-H}^{-H+h}$$

$$= \frac{1}{2} \rho g L \left[H^2 - (H-h)^2 \right]$$

$$= \cancel{H^2} - \cancel{H^2} + 2Hh - h^2$$

$$= \frac{1}{2} \rho g L h (2H-h) \quad , \quad \vec{F} \uparrow \uparrow -\hat{e}_y$$

Torque

$$a) \quad \vec{T} = \vec{r} \times \vec{F}$$

$$\vec{r} \perp d\vec{F} \Rightarrow dT = r dF \quad , \quad d\vec{T} \uparrow \uparrow -\hat{e}_x$$

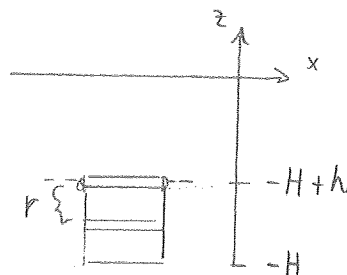
$$\begin{cases} r = -H + h - z & (z < 0) \\ dF = P ds = -\rho g L z dz \end{cases}$$

$$\Rightarrow T = -\rho g L \int_{-H}^{-H+h} z(-H+h-z) dz$$

$$= -\rho g L \left[-(H-h) \frac{z^2}{2} - \frac{z^3}{3} \right]_{-H}^{-(H-h)}$$

$$= \rho g L \left[\frac{H^3}{3} - \frac{(H-h)^3}{3} + \frac{(H-h)^3}{2} - (H-h) \frac{H^2}{2} \right] = \frac{1}{6} \rho g L h^2 (3H-h)$$

continues...

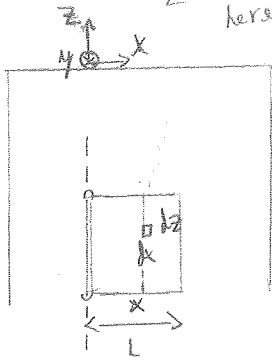


1.8

... continues

Let's fix coord. axes
here

b)



$$dT = x dF, \quad d\vec{T} \parallel -\hat{e}_z$$

$$\Rightarrow dT = x P dx dz$$

$$\Rightarrow T = \int_0^L \int_{-H}^{-H+h} x (-\rho g z) dx dz$$

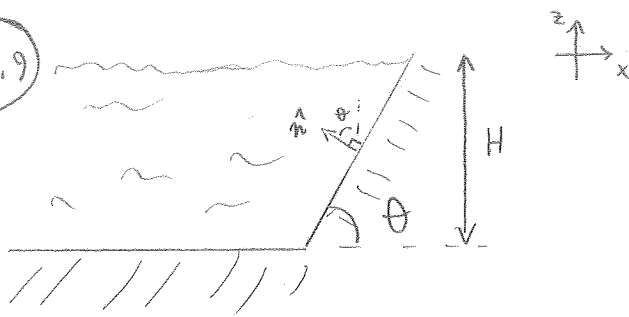
$$= \underbrace{\int_0^L x dx}_{= \frac{L^2}{2}} \int_{-H}^{-H+h} (-\rho g z) dz$$

$$= \frac{L}{2} \underbrace{\int_{-H}^{-H+h} (-\rho g L z) dz}_{= F}$$

$$= \frac{L}{2} \cdot F$$

$$= \frac{1}{4} \rho g L^2 h (2H - h)$$

P 1.9



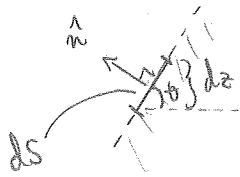
$$p = p_0 - \alpha z$$

Force on the wall $F_{\text{total}} = ?$

Solution: $d\vec{F} = -p ds \hat{n}$, $\hat{n} = -\hat{e}_x \sin\theta + \hat{e}_z \cos\theta$

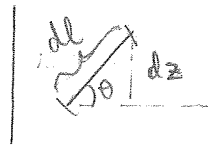
$$\frac{dp}{dz} = -\rho g = -\rho_0 g + g \alpha z$$

$$\Rightarrow p = -\rho_0 g z + g \alpha \frac{z^2}{2} \quad (\text{neglecting } p_{\text{atm}})$$



$$ds = L_y \cdot dl$$

$$= L_y \cdot \frac{dz}{\sin\theta}$$



$$\frac{dz}{dl} = \sin\theta$$

$$\Rightarrow dl = \dots$$

$$\Rightarrow \vec{F} = -\hat{n} F$$

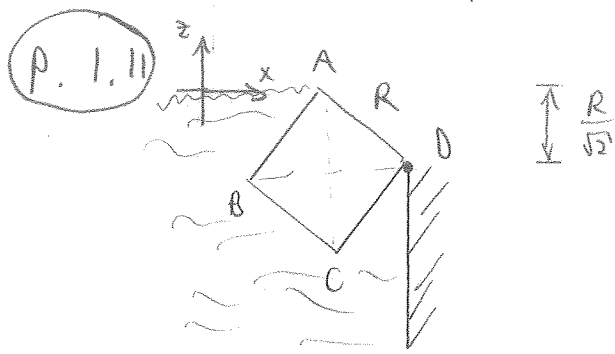
$$\Rightarrow F = \int_{-H}^0 \left(-\rho_0 g z + g \alpha \frac{z^2}{2} \right) L_y \frac{dz}{\sin\theta}$$

$$= \left[-\rho_0 g \frac{z^2}{2} + g \alpha \frac{z^3}{6} \right] \bigg|_{-H}^0 \frac{L_y}{\sin\theta}$$

$$= \frac{L_y}{\sin\theta} \left(\rho_0 g \frac{H^2}{2} + g \alpha \frac{H^3}{6} \right)$$

$$\Rightarrow \vec{F} = -\hat{n} \frac{g L_y}{2 \sin\theta} H^2 \left(\rho_0 + \frac{\alpha H}{3} \right)$$

$$\Rightarrow \underline{\underline{\vec{F}/\text{unit length} = \frac{\vec{F}}{L_y} = -\hat{n} \frac{g}{2 \sin\theta} H^2 \left(\rho_0 + \frac{\alpha H}{3} \right)}}$$



Horizontal force

$$F_x = F_{AB} + \cancel{F_{BC}} - \cancel{F_{CD}} = F_{AB}$$

$= -F_{BC}$

$$p = -\rho g z$$

$$dF_x = \int_{-\frac{R}{\sqrt{2}}}^0 (-\rho g z) \cdot \widetilde{L_y} dz = -\rho g L_y \left[\frac{z^2}{2} \right]_{-\frac{R}{\sqrt{2}}}^0 = \rho g L_y \frac{R^2}{4}$$

$$\Rightarrow \frac{dF_x}{\text{unit length}} = \frac{dF_x}{L_y} = \rho g \frac{R^2}{4}$$

Vertical force

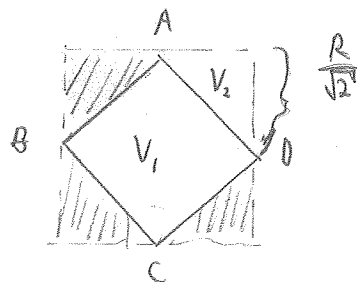
$$F_z = F_{BCD} - F_{AB}$$

Archimedes

$$= \rho_{\text{fluid}} g V_{\text{body}}$$

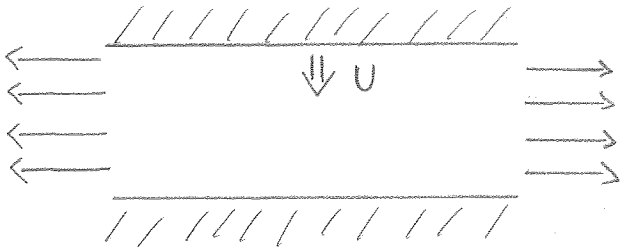
$$= \rho g (V_1 + V_2)$$

$$= \rho g L_y \left(R^2 + \frac{1}{2} \frac{R^2}{2} \right) = \frac{5}{4} \rho g L_y R^2$$



$$\Rightarrow \frac{dF_z}{\text{unit length}} = \frac{dF_z}{L_y} = \frac{5 \rho g R^2}{4}$$

Example:



$$u_x = \frac{U}{H_0 - y} x$$

$$u_y = -\frac{U}{H_0 - y} y$$

Streamlines ?

Solution:

$$\frac{dx}{u_x} = \frac{dy}{u_y}$$

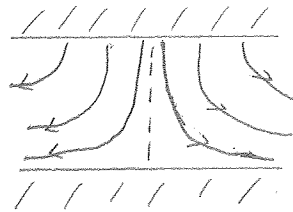
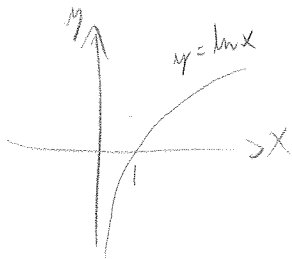
$$\Rightarrow (H_0 - y) \frac{dx}{x} = - \frac{dy}{y} (H_0 - y)$$

$$\Rightarrow \frac{dx}{x} + \frac{dy}{y} = 0$$

$$\Rightarrow \ln x + \ln y = C'$$

$$\Rightarrow xy = C, \quad C = \text{const.}$$

$$\Rightarrow y = \frac{C}{x}$$



$$\frac{dy}{dx} = \frac{u_y}{u_x}$$

$$d\vec{r} = dx \hat{e}_x + dy \hat{e}_y$$

$$\ln(xy) = C'$$

$$xy = e^{C'} = C$$

(2.2)

The Rankine vortex

$$\begin{cases} u_\theta = \Omega r & , r < R \\ u_\theta = \Omega R^2/r & , r > R \end{cases}$$

Shape of the free surface $z_f = f(r) = ?$ Euler \Rightarrow $\Delta h = ?$

Solution: r : $-\frac{u_\theta^2}{r} = -\frac{1}{\rho} \frac{\partial p}{\partial r}$ (1)

z : $0 = -\frac{1}{\rho} \frac{\partial p}{\partial z} - g$ (2)

a) $r < R$ $\int (1) \& \int (2)$

$$\frac{\partial p}{\partial r} = \rho \Omega^2 r$$

$$\Rightarrow p = \frac{1}{2} \rho \Omega^2 r^2 - \rho g z + p_0$$

$$p_0 = p_{atm} \Rightarrow \text{at } z_f: p_{z=z_f} = p_{atm}$$

$$\Rightarrow \underline{z_f = \frac{\Omega^2 r^2}{2g}}$$

 $z=0 = z_f \text{ at } r=0 \leftarrow \text{our choice}$ b) $r > R$ $\int (1) \& \int (2)$

$$\frac{\partial p}{\partial r} = \frac{\rho \Omega^2 R^4}{r^3}$$

$$(3) \Rightarrow$$

$$p = -\frac{1}{2} \frac{\rho \Omega^2 R^4}{r^2} - \rho g z' + \underbrace{p_0}_{= p_{atm}}$$

$$\text{Now } z' = z - \Delta h$$

$$\Rightarrow p = -\frac{1}{2} \frac{\rho \Omega^2 R^4}{r^2} - \rho g z + \rho g \Delta h + p_{atm}$$

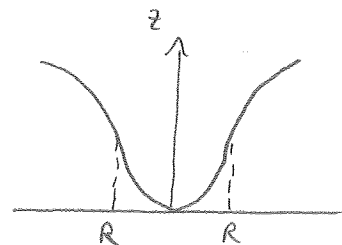
$$p_{z=z_f} = p_{atm}$$

$$\Rightarrow \cancel{p_{atm}} = -\frac{1}{2} \frac{\rho \Omega^2 R^4}{r^2} - \rho g z_f + \rho g \Delta h + \cancel{p_{atm}}$$

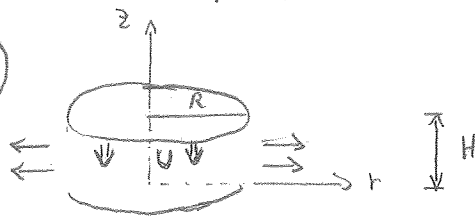
$$\Leftrightarrow z_f = -\frac{1}{2} \frac{\Omega^2 R^4}{g r^2} + \Delta h$$

$$\text{Now } z_f|_{R_-} = z_f|_{R_+} \Leftrightarrow \frac{\Omega^2 R^2}{2g} = -\frac{\Omega^2 R^4}{2g R^2} + \Delta h \Rightarrow \underline{\Delta h = \frac{\Omega^2 R^2}{g}}$$

$$\Rightarrow z_f = \begin{cases} \frac{\Omega^2 r^2}{2g} & , r < R \\ -\frac{\Omega^2 R^4}{2g r^2} + \frac{\Omega^2 R^2}{g} & , r > R \end{cases}$$



1.4 **



$P(r) = ?$, $P(r=0) = ?$, $F_{p, flow} = ?$ at the bottom plate $z=0$

Solution: From 1.4* \Rightarrow

$$\begin{cases} u_r = \frac{Ur}{2(H_0 - Ut)} \\ u_z = -\frac{Uz}{H_0 - Ut} \end{cases}$$

At the bottom plate $u_z = 0$ ($z=0$)

A quasi-stationary flow $\Rightarrow H_0 - Ut \approx H_0$

Bernoulli's eq. $P + \frac{1}{2}\rho u^2 = \text{const.}$

($u_z=0$) \Rightarrow

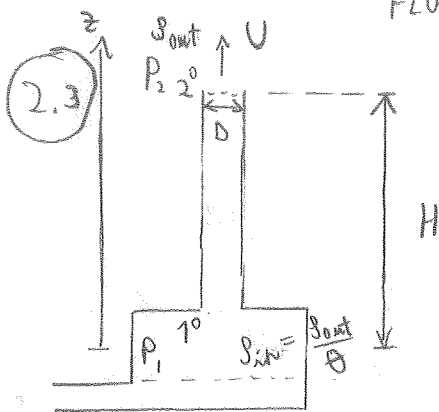
$$P + \frac{1}{2}\rho u_r^2 = P_{atm} + \frac{1}{2}\rho u_r^2 \Big|_{r=R}$$

$$\Rightarrow P = P_{atm} + \frac{1}{2}\rho U^2 \left(\frac{R^2}{4H_0^2} - \frac{r^2}{4H_0^2} \right) = \underbrace{P_{atm} + \frac{3U^2}{8H_0^2} (R^2 - r^2)}$$

$$\Rightarrow \underbrace{P(r=0) = P_{atm} + \frac{3U^2 R^2}{8H_0^2}}$$

Total force due to the flow:

$$\begin{aligned} F_{p, flow} &= \iint_S (P - P_{atm}) ds = \int_0^R \frac{3U^2}{8H_0^2} (R^2 - r^2) \underbrace{2\pi r dr}_{= ds} \\ &= \frac{3U^2}{8H_0^2} \cdot 2\pi \left[R^2 \frac{r^2}{2} - \frac{r^4}{4} \right] \Big|_0^R \\ &= \underbrace{\frac{3U^2 \pi R^4}{16H_0^2}} \end{aligned}$$



At the exit of the stack $P = P_2$

$$Q_{out} = ? = US$$

Solution: (slow inside)

Bernoulli's integral $\frac{P}{S_{in}} + \frac{u^2}{2} + gz = \text{const.}$

$$(P + S_{in} \frac{u^2}{2} + S_{in} gz = \text{const.})$$

Now at 1° (inside the stove burning chamber):

$$\begin{cases} P_1 = P_2 + S_{out} g H \\ U_1 \approx 0 \\ z = 0 \end{cases}$$

and at 2° (exit of the stack):

$$\begin{cases} P_2 = P_2 \\ U_2 = U \\ z = H \\ S_{in} = S_{out} / \theta \end{cases}$$

Bernoulli $1^\circ = 2^\circ \Rightarrow$

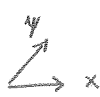
$$\underbrace{\frac{P_2 + S_{out} g H}{S_{in}} + 0 + 0}_{1^\circ, \text{ stove}} = \underbrace{\frac{P_2}{S_{in}} + \frac{U^2}{2} + g H}_{2^\circ, \text{ exit of the stack}}$$

$$\Rightarrow \theta g H = \frac{U^2}{2} + g H$$

$$\Leftrightarrow U^2 = 2(\theta - 1) g H$$

$$\Rightarrow Q = US = \frac{\pi}{4} D^2 \sqrt{2(\theta - 1) g H}$$

2.4



Horizontal tube, $\vec{F}_{\text{on the wall}} = ?$
 $= \vec{F}$

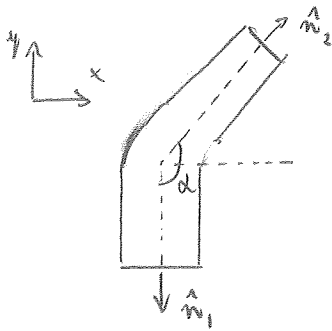
Solution:

$$S_1 U_1 = S_2 U_2 \Rightarrow U_2 = \frac{S_1}{S_2} U_1 \quad (1)$$

Bernoulli's integral

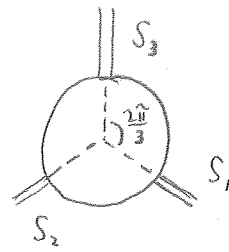
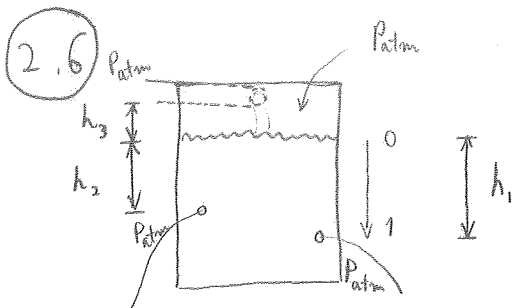
$$\frac{p_1}{\rho} + \frac{U_1^2}{2} = \frac{p_2}{\rho} + \frac{U_2^2}{2} \quad (2)$$

$$p_2 = p_1 + \frac{\rho U_1^2}{2} \left(1 - \frac{S_1^2}{S_2^2} \right)$$

(1) & (2) \Rightarrow 

$$\begin{cases} \vec{F} \stackrel{(1,2)}{=} - \sum_i (\rho U_i^2 + p_i) S_i \hat{n}_i & (= - \int_{S_1, S_2} [(\rho \vec{U}) \vec{U} \cdot d\vec{S} + p d\vec{S}]) \\ \hat{n}_1 = -\hat{e}_y \\ \hat{n}_2 = -\hat{e}_y \cos \alpha + \hat{e}_x \sin \alpha \end{cases}$$

$$\begin{aligned} \vec{F} &= -(\rho U_1^2 + p_1) S_1 (-\hat{e}_y) \\ &\quad - \left[-\rho \frac{S_1^2}{S_2^2} U_1^2 + - \left(p_1 + \frac{\rho U_1^2}{2} \left(1 - \frac{S_1^2}{S_2^2} \right) \right) \right] S_2 \underbrace{(-\hat{e}_y \cos \alpha + \hat{e}_x \sin \alpha)}_{\hat{n}_2} \\ &= \underbrace{(p_1 + \rho U_1^2) S_1 \hat{e}_y + \left[p_1 + \frac{1}{2} \rho U_1^2 \left(1 + \frac{S_1^2}{S_2^2} \right) \right] S_2 (\hat{e}_y \cos \alpha - \hat{e}_x \sin \alpha)} \end{aligned}$$



$$\bar{F}_{\text{on the tank}} = ? = \bar{F}$$

Formal equation $\bar{F} = - \sum_i (p U_i^2 + p_i) S_i \hat{n}_i$ (1.22)

$p_i = p_{\text{atm}}$ for all jets and p_{atm} generally the tank, $\Rightarrow p_i$ may be neglected

$$\Rightarrow \text{more convenient eq. } \bar{F} = - \sum_i Q_i S \bar{U}_i, \quad Q_i = \bar{U}_i \cdot S_i \hat{n}_i$$

Let's find first U_1 and U_2 :

Bernoulli's integral \Rightarrow velocity at the surface $\ll U_1$

$$0 \rightarrow 1: \quad \frac{p_{\text{atm}}}{\rho} + 0 + 0 = \frac{p_{\text{atm}}}{\rho} + \frac{U_1^2}{2} - gh_1 \Rightarrow U_1 \approx \sqrt{2gh_1}$$

Similarly $U_2 \approx \sqrt{2gh_2}$

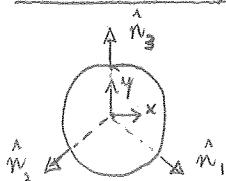
Total discharge: $Q_{\text{out}} = S_1 \sqrt{2gh_1} + S_2 \sqrt{2gh_2}$

Discharge of the coming water: $Q_{\text{in}} = Q_{\text{out}}$

$$\Rightarrow U_3 = \frac{S_1}{S_3} \sqrt{2gh_1} + \frac{S_2}{S_3} \sqrt{2gh_2}$$

Vertical velocity component of the coming fluid $U_{\text{vert}} = \sqrt{2gh_3}$

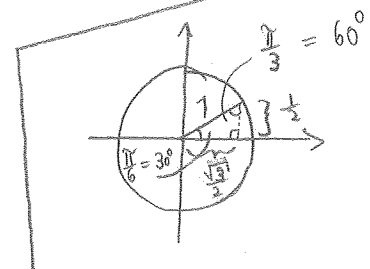
Horizontal forces:



$$\hat{n}_3 = \hat{e}_y$$

$$\hat{n}_1 = \frac{\sqrt{3}}{2} \hat{e}_x - \frac{1}{2} \hat{e}_y$$

$$\hat{n}_2 = -\frac{\sqrt{3}}{2} \hat{e}_x - \frac{1}{2} \hat{e}_y$$



$$\bar{F}_{\text{horiz}} = -\hat{e}_y S U_3^2 S_3 - \left(\frac{\sqrt{3}}{2} \hat{e}_x - \frac{1}{2} \hat{e}_y \right) S_1 S \cdot 2gh_1 - \left(-\frac{\sqrt{3}}{2} \hat{e}_x - \frac{1}{2} \hat{e}_y \right) S_2 S \cdot 2gh_2$$

$$= -\hat{e}_y \frac{S}{S_3} \left(S_1 \sqrt{2gh_1} + S_2 \sqrt{2gh_2} \right)^2 + \frac{1}{2} \hat{e}_y S 2g (h_1 S_1 + h_2 S_2)$$

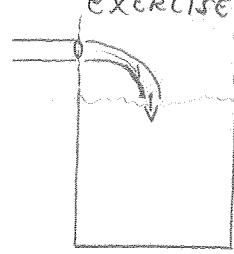
$$- \frac{\sqrt{3}}{2} \hat{e}_x S 2g (S_1 h_1 - S_2 h_2)$$

$$= \hat{e}_x \sqrt{3} S g (S_2 h_2 - S_1 h_1) + \hat{e}_y S g \left[h_1 S_1 + h_2 S_2 - \frac{2}{S_3} \left(S_1 \sqrt{h_1} + S_2 \sqrt{h_2} \right)^2 \right]$$

continues...

2.6 ... continues

$$\begin{aligned}
 \bar{F}_{\text{vert}} &= -\hat{e}_z \rho Q_{\text{in}} U_{\text{vert}} \\
 &= -\hat{e}_z \rho \sqrt{2gh_3} (S_1 \sqrt{2gh_1} + S_2 \sqrt{2gh_2}) \\
 &= -\hat{e}_z 2\rho g \sqrt{h_3} (S_1 \sqrt{h_1} + S_2 \sqrt{h_2})
 \end{aligned}$$



b) $\lambda(t) = ?$

Solution:

Solution:

$$\begin{cases} P_{atm} + \rho \frac{U_2^2}{2} = P_3 + \rho \frac{U_3^2}{2} \\ U_2 S_2 = U_3 S_3 \end{cases} \Rightarrow P_3 = P_{atm} - \rho \frac{U_2^2}{2} \left(\frac{S_2^2}{S_3^2} - 1 \right)$$

Bernoulli's integral in the reservoir (assuming $U_0 \ll U_1$)

$$\begin{aligned} \cancel{P_{atm}} + \rho g h &= \overbrace{P_3 + \rho g L}^{P_2} + \rho \frac{V_2^2}{2} \\ &= \cancel{P_{atm}} + \rho g L + \rho \frac{V_1^2}{2} - \frac{V_2^2}{2} \left(\frac{S_2^2}{S_3^2} - 1 \right) \end{aligned}$$

$$\Rightarrow U_1^2 = U_2^2 \left(\frac{S_2^2}{S_3^2} - 1 \right) - 2q(L-h)$$

In addition

In addition

$$U_1 = \frac{S_0}{S_1} U_0 = -\frac{S_0}{S_1} \frac{dh}{dt}$$

$$\Rightarrow \left(\frac{dh}{dt}\right)^2 = \frac{S_1^2}{S_0^2} \left(2g(h-L) + U_2^2 \left(\frac{S_2^2}{S_3^2} - 1 \right) \right)$$

The flow in pipe stops $\Leftrightarrow U_1 = 0$ at

$$h_{do} = L - \frac{V_2^2}{2g} \left(\frac{S_2^2}{S_3^2} - 1 \right)$$



\Rightarrow reservoir empty $\Leftrightarrow h_{\infty} = 0$

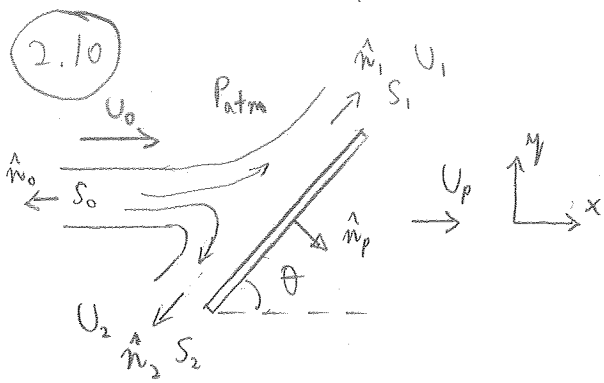
$$\Rightarrow 0 = L - \frac{V_{2, \text{empty}}^2}{2g} \left(\frac{S_2^2}{S_3^2} - 1 \right)$$

$$\Rightarrow U_{2, \text{empty}} = \sqrt{\frac{2q/L}{S_2 - S_3}} S_3 \quad a)$$

$$\Rightarrow - \int_{h_0}^h \frac{dh}{\sqrt{h-h_\infty}} = \frac{S_1}{S_2} \sqrt{2g} \sqrt{h-h_\infty}$$

$$\Rightarrow -2\sqrt{h-h_{\infty}} + 2\sqrt{h_0-h_{\infty}} = \frac{S_1}{S_2} \sqrt{2gt^2}$$

$$\Rightarrow h = h_{\infty} + (h_0 - h_{\infty}) \left(1 - \frac{S_1}{S_2} \sqrt{\frac{g t^2}{2(h_0 - h_{\infty})}} \right)^2 \quad b)$$



a) $S_1 = ?$, $S_2 = ?$

b) $\bar{F}_p = ?$

c) maximal work $W_{\max} = ?$
 assuming no losses, no gravity

Solution: In the rest frame of the plate: $U_0' = U_0 - U_p$

Bernoulli's integral $\Rightarrow \frac{(U_0')^2}{2} + \frac{P_{atm}}{\rho} = \frac{U_1^2}{2} + \frac{P_{atm}}{\rho} = \frac{U_2^2}{2} + \frac{P_{atm}}{\rho}$

$\Leftrightarrow U_0' = U_1 = U_2$

Mass conservation $\Rightarrow S_0 U_0' = S_1 U_1 + S_2 U_2 \Rightarrow S_0 = S_1 + S_2$

The net force on the fluid

$0 = - \oint_S \bar{u} \cdot d\bar{s} + \bar{F}_{ext}$

$\Rightarrow \bar{F}_{ext} = \rho S_0 (U_0')^2 \hat{n}_0 + \rho S_1 (U_0')^2 \hat{n}_1 + \rho S_2 (U_0')^2 \hat{n}_2$

$\left. \begin{aligned} \frac{dp}{dt} &= -(\bar{p} - \text{flux out}) \\ &+ \bar{F}_{\text{on the fluid}} \\ &= \bar{F}_{ext} \end{aligned} \right|$

$\hat{n}_2 = -\hat{n}_1$

$\bar{F}_{ext} \uparrow \uparrow -\hat{n}_p$, $\hat{n}_p = \hat{e}_x \sin \theta - \hat{e}_y \cos \theta$

\hat{n}_p the external force along the plate is zero (the fluid is ideal)

$\Rightarrow \hat{n}_1 \cdot \bar{F}_{ext} = 0 = S_0 (\hat{n}_0 \cdot \hat{n}_1) + S_1 - S_2$
 $\hat{n}_0 \cdot \hat{n}_1 = -\cos \theta$

Now $\begin{cases} S_1 - S_2 = S_0 \cos \theta \\ S_1 + S_2 = S_0 \end{cases} \Rightarrow \begin{cases} S_1 = \frac{S_0}{2} (1 + \cos \theta) \\ S_2 = \frac{S_0}{2} (1 - \cos \theta) \end{cases}$ a)

Checking: $\cos \theta = 0$, $\theta = \frac{\pi}{2} \Rightarrow S_1 = S_2$ ok

$\theta = 0$, $\cos \theta = 1 \Rightarrow S_1 = S_0$, $S_2 = 0$ ok

continues...

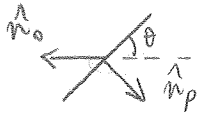
2.10 ... continues

b) Looking for the force \Rightarrow

$$\hat{n}_p \cdot \vec{F}_{\text{ext}} = \oint S_0 (U_0')^2 \underbrace{\hat{n}_0 \cdot \hat{n}_p}_{=0} + \oint S_1 (U_0')^2 \underbrace{\hat{n}_1 \cdot \hat{n}_p}_{=0} + \oint S_2 (U_0')^2 \underbrace{\hat{n}_2 \cdot \hat{n}_p}_{=0}$$

$$= -\cos\left(\frac{\pi}{2} - \theta\right)$$

$$= -\sin\theta$$



$$\Rightarrow F_p = -F_{\text{ext}} = -\oint S_0 (U_0')^2 (\hat{n}_0 \cdot \hat{n}_p) = \oint S_0 (U_0')^2 \sin\theta$$

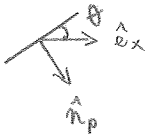
Going to the laboratory frame

$$U_0' = U_0 - U_p$$

$$\Rightarrow \vec{F}_p = \hat{n}_p \oint S_0 (U_0 - U_p)^2 \sin\theta$$

c) $\frac{\text{Work}}{\text{unit time}} = \text{power} = W = \vec{F}_p \cdot \vec{U}_p$

$$= \oint S_0 (U_0 - U_p)^2 U_p \sin\theta \underbrace{\hat{e}_x \cdot \hat{n}_p}_{= \cos\left(\frac{\pi}{2} - \theta\right)} = \sin\theta$$



$$W = \oint S_0 (U_0 - U_p)^2 U_p \sin^2\theta$$

$$W_{\text{max}} = ?$$

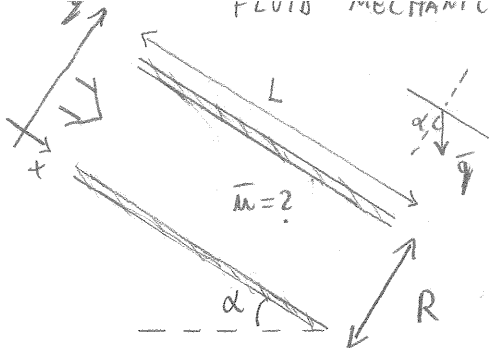
$$\frac{dW}{dU_p} = 0 \Rightarrow \oint S_0 \sin^2\theta [(U_0 - U_p)^2 - 2U_p(U_0 - U_p)] = 0$$

Maximum corresponds to

$$U_p = \frac{U_0}{3}$$

$$\Rightarrow W_{\text{max}} = \frac{4}{27} \oint S_0 U_0^3 \sin^2\theta$$

(5.10)

velocity $\bar{u} = ?$ discharge $Q = ?$

$$\rho \left(\frac{\partial \bar{u}}{\partial t} + (\bar{u} \cdot \nabla) \bar{u} \right) = -\nabla p + \rho \bar{g} + \mu \nabla^2 \bar{u}$$

x-direction: the channel open to atmosphere $\Rightarrow \frac{\partial p}{\partial x} = 0$

$$u_x = u_x(y) \Rightarrow (\bar{u} \cdot \nabla) \bar{u} = 0$$

$$\Rightarrow \mu \frac{\partial^2 u_x}{\partial y^2} + \rho g \sin \alpha = 0$$

$$\Rightarrow \frac{\partial^2 u_x}{\partial y^2} = -\frac{\rho g}{\mu} \sin \alpha$$

$$\Rightarrow u_x = -\frac{\rho g}{\mu} \sin \alpha \frac{y^2}{2} + By + C$$

Boundary conditions: $u_x = 0$ at $y = 0 \Rightarrow C = 0$

$$u_x = 0 \text{ at } y = R \Rightarrow -\frac{\rho g}{\mu} \sin \alpha \frac{R^2}{2} + BR = 0$$

$$\Rightarrow B = \frac{\rho g}{\mu} \sin \alpha \frac{R}{2}$$

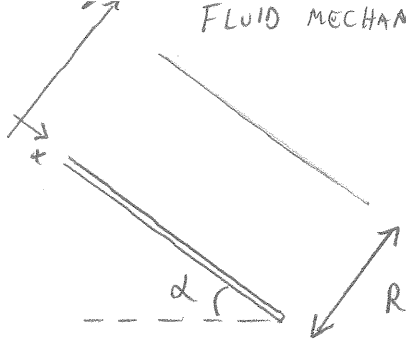
$$\Rightarrow u_x = \frac{\rho g}{2\mu} \sin \alpha (R - y)y$$

The discharge per L_z

$$\frac{Q}{L_z} = \int_0^R u_x dy = \frac{\rho g}{2\mu} \sin \alpha \int_0^R (Ry - y^2) dy = \frac{\rho g}{2\mu} \sin \alpha \left(\frac{R^3}{2} - \frac{R^3}{3} \right)$$

$$= \frac{\rho g}{12\mu} \sin \alpha R^3$$

(5.11)

 $\downarrow \bar{g}$

$u_x = ?$

$u_{x, \max} = ?$

$$\mathcal{S} = \mathcal{S}_{x,y} = \frac{F}{S} \Big|_{y=0} = ? \quad \text{Force on the bottom plate}$$

Navier - Stokes $\mu \frac{\partial^2 u_x}{\partial y^2} + \rho g \sin \alpha = 0$

$$\Rightarrow u_x = - \frac{\rho g \sin \alpha}{\mu} \frac{y^2}{2} + B y + C$$

Boundary conditions:

$$u_x = 0 \quad \text{at} \quad y = 0 \quad \Rightarrow C = 0$$

$$\frac{\partial u_x}{\partial y} = 0 \quad \text{at} \quad y = R \quad \triangleq \text{zero stress} \quad \left(\mathcal{S} = \mu \frac{\partial u_x}{\partial y} \right)$$

$$\Leftrightarrow \frac{\partial u_x}{\partial y} \Big|_{y=R} = \left(- \frac{\rho g \sin \alpha}{\mu} y + B \right) \Big|_{y=R} = 0 \quad \Rightarrow B = \frac{\rho g \sin \alpha}{\mu} R$$

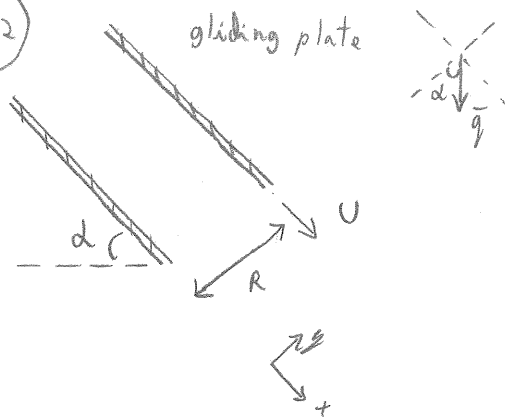
$$\Rightarrow u_x = \frac{\rho g \sin \alpha}{2\mu} (2R - y) y$$

$$u_{x, \max} = u_x \Big|_{y=R} = \frac{\rho g \sin \alpha}{2\mu} R^2$$

Viscous stress (force) acting on the lower plate

$$\frac{F}{S} = \mu \frac{\partial u_x}{\partial y} \Big|_{y=0} = \mu \frac{\rho g \sin \alpha}{\mu} R = \underline{\underline{\rho g R \sin \alpha}}$$

5.12



$$U = ?$$

$$\frac{\text{plate mass}}{\text{unit surface area}} = \frac{M}{S}$$

Solution: Balance of forces on the (upper) plate $\Leftrightarrow Mg \sin \alpha = \int S$ (*)

Navier-Stokes along x

$$\mu \frac{\partial^2 u_x}{\partial y^2} = - \rho g \sin \alpha \quad \Rightarrow \quad u_x = - \frac{\rho g \sin \alpha}{2\mu} y^2 + A y + B$$

Boundary conditions: $u_x = 0$ at $y = 0 \Rightarrow B = 0$

$$u_x = U \text{ at } y = R \Rightarrow A = \frac{U}{R} + \rho g \frac{\sin \alpha}{2\mu} R$$

Then
$$u_x = \frac{\rho g \sin \alpha}{2\mu} y(R - y) + \frac{U y}{R}$$

\Rightarrow Viscous stress \int at $y = R$

$$\begin{aligned} \int &= \mu \left. \frac{\partial u_x}{\partial y} \right|_{y=R} = \mu \left[\frac{\rho g \sin \alpha}{2\mu} R - \frac{\rho g \sin \alpha}{2\mu} \cdot 2R + \frac{U}{R} \right] \\ &= \mu \frac{U}{R} - \frac{\rho g \sin \alpha}{2} R \end{aligned}$$

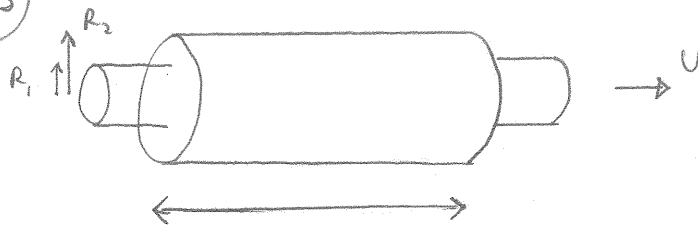
Now (*) $\Rightarrow \int = \mu \frac{U}{R} - \frac{\rho g \sin \alpha}{2} R = \frac{M g \sin \alpha}{S}$

$$\Rightarrow (M + \frac{1}{2} \rho R S) g \sin \alpha = \mu \frac{U}{R} S$$

$$\Rightarrow U = \frac{R g \sin \alpha}{\mu S} (M + \frac{1}{2} \rho R S)$$

or
$$U = \frac{M g R \sin \alpha}{\mu S} + \frac{\rho R^2 g \sin \alpha}{2\mu}$$

5.13



$$u(r) = ?$$

$$\Delta P > 0$$

Solution:

$$-\frac{\partial P}{\partial z} + \mu \frac{1}{r} \frac{\partial}{\partial r} \left(r \frac{\partial u_z}{\partial r} \right) = 0$$

$$\frac{\partial P}{\partial r} = 0 \Rightarrow \frac{\partial P}{\partial z} = \frac{\Delta P}{L} = \text{const.}$$

$$\Rightarrow \frac{1}{r} \frac{\partial}{\partial r} \left(r \frac{\partial u_z}{\partial r} \right) = \frac{\Delta P}{\mu L} \quad | \int$$

$$\Rightarrow u_z = \frac{\Delta P}{\mu L} \frac{r^2}{4} + A \ln r + B$$

Boundary conditions:

$$u_z = U \quad \text{at } r = R_1$$

$$u_z = 0 \quad \text{at } r = R_2$$

Then

$$\frac{\Delta P}{4\mu L} R_1^2 + A \ln R_1 + B = U$$

$$\frac{\Delta P}{4\mu L} R_2^2 + A \ln R_2 + B = 0$$

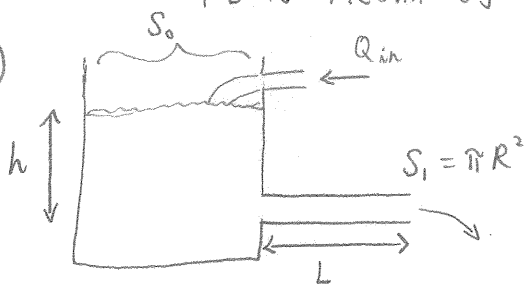
$$\frac{\Delta P}{4\mu L} (R_1^2 - R_2^2) + A \ln \frac{R_1}{R_2} = U \Rightarrow A = \ln \frac{R_2}{R_1} \left[U - \frac{\Delta P}{4\mu L} (R_1^2 - R_2^2) \right]$$

$$\Rightarrow B = -\ln \frac{R_2}{R_1} \left[\right] \ln R_2 - \frac{\Delta P}{4\mu L} R_2^2$$

$$\Rightarrow u_z = \frac{\Delta P}{4\mu L} r^2 + \left[U + \frac{\Delta P}{4\mu L} (R_2^2 - R_1^2) \right] \ln \frac{R_2}{R_1} \cdot \ln r - \left[U + \frac{\Delta P}{4\mu L} (R_2^2 - R_1^2) \right] \ln \frac{R_2}{R_1} \ln R_2 - \frac{\Delta P}{4\mu L} R_2^2$$

$$= \frac{\Delta P}{4\mu L} (r^2 - R_2^2) + \left[U + \frac{\Delta P}{4\mu L} (R_2^2 - R_1^2) \right] \frac{\ln(r/R_2)}{\ln(R_1/R_2)}$$

5.14



$$h(t) = ?$$

Pressure variations for the tube $\Delta P = \rho g h$

Navier-Stokes in the tube

$$-\frac{\partial P}{\partial x} + \mu \frac{1}{r} \frac{\partial}{\partial r} \left(r \frac{\partial u_x}{\partial r} \right) = 0$$

$$\frac{\partial P}{\partial x} = -\frac{\Delta P}{L} = -\frac{\rho g h}{L}$$

$$\frac{1}{r} \frac{\partial}{\partial r} \left(r \frac{\partial u_x}{\partial r} \right) = -\frac{\rho g h}{\mu L}$$

$$\Rightarrow u_x = \frac{\rho g h}{4\mu L} (R^2 - r^2)$$

$$Q_{out} = \int_0^R u_x 2\pi r dr = \frac{\pi}{8} \frac{\rho g h}{\mu L} R^4$$

$$Q_{out} - Q_{in} = -S_0 \frac{dh}{dt}$$

$$\Rightarrow \frac{dh}{dt} = \frac{Q_{in}}{S_0} - \frac{\pi \rho g R^4}{8\mu L S_0} h$$

Water level $h = ?$

$$\text{When } \frac{dh}{dt} = 0 \Leftrightarrow h_{\infty} = \frac{8\mu L Q_{in}}{\pi \rho g R^4}$$

$$\text{Typical time scale } \tau \left(= \frac{h_{\infty}}{U_0} \right) = \frac{8\mu L S_0}{\pi \rho g R^4}$$

$$\text{Then } \frac{dh}{dt} = -\frac{h - h_{\infty}}{\tau}$$

$$\Rightarrow \ln(h - h_{\infty}) - \ln(h_0 - h_{\infty}) = -\frac{t}{\tau}$$

$$\Rightarrow h = h_{\infty} + (h_0 - h_{\infty}) \exp(-t/\tau)$$

$$\frac{\partial}{\partial r} \left(r \frac{\partial u_x}{\partial r} \right) = -\frac{\rho g h}{\mu L} r$$

$$\Rightarrow r \frac{\partial u_x}{\partial r} = -\frac{\rho g h}{\mu L} \frac{r^2}{2} + A$$

$$\Rightarrow \frac{\partial u_x}{\partial r} = -\frac{\rho g h}{\mu L} \frac{r}{2} + \frac{A}{r}$$

$$\Rightarrow u_x = -\frac{\rho g h}{\mu L} \frac{r^2}{4} + A \ln r + B$$

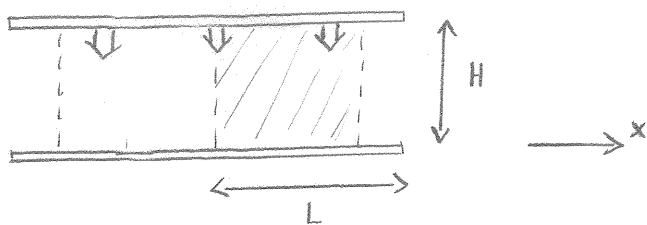
Boundary cond.

$$u_x > 0 \text{ at } r=0 \Rightarrow A=0$$

$$u_x = 0 \text{ at } r=R$$

$$\Rightarrow B = \frac{\rho g h}{4\mu L} R^2$$

5.15



Viscous walls,
far from centre $u_x(y) \sim$ Poiseuille

$$U_{\max, x} = U_{\max} = ?$$

Solution:

$$V = x (H_0 - Ut) L_z$$

$$\frac{dV}{dt} = -x U L_z = -Q_{\text{out}}$$

Quasi-Poiseuille flow:

$$u_x \approx A y (H - y)$$

$$U_{\max} \Rightarrow \frac{du_x}{dy} = 0 \Leftrightarrow A(H - y) - Ay = 0$$

$$\Rightarrow y = \frac{H}{2}$$

$$\Rightarrow U_{\max} = A \frac{H}{2} H - A \frac{H^2}{4} \Rightarrow U_{\max} = A \frac{H^2}{4}$$

$$\Leftrightarrow A = \frac{4 U_{\max}}{H^2}$$

$$\Rightarrow u_x = 4 U_{\max} \frac{y}{H} \left(1 - \frac{y}{H}\right)$$

$$Q_{\text{out}} = \int_0^H u_x L_z dy$$

$$= L_z 4 U_{\max} \int_0^H \frac{y}{H} \left(1 - \frac{y}{H}\right) dy$$

$$= \int_0^H \left(\frac{4y}{H} - \frac{4y^2}{H^2}\right) dy = \left[\frac{4y^2}{2H} - \frac{4y^3}{3H^2} \right]_0^H = \frac{H}{6}$$

$$= L_z 4 U_{\max} \frac{H}{6}$$

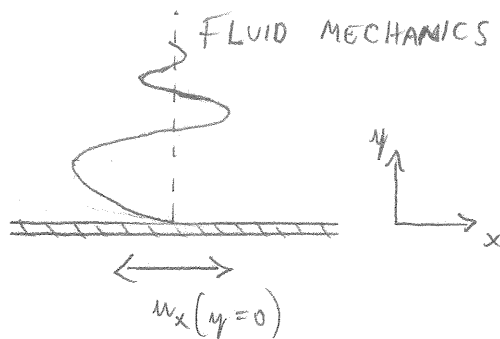
$$\text{Now } \cancel{L_z} \frac{2 U_{\max} H}{3} = x U L_z$$

 \Rightarrow

$$U_{\max} = \frac{3}{2} U \frac{x}{H}$$

$$\begin{cases} x = L \\ U_{\max} = \frac{3}{2} U \frac{L}{H} \end{cases}$$

5.24



$$u_x(y, t) = ?$$

$$u_x(y=0) = U \sin(\omega t) = \text{Im}[U \exp(i\omega t)]$$

Navier-Stokes $\frac{\partial u_x}{\partial t} = \nu \frac{\partial^2 u_x}{\partial y^2}$

$$u_x = W(y) \exp(i\omega t)$$

k can be complex
↓

$$\Rightarrow i\omega W \exp(i\omega t) = \nu \frac{d^2 W}{dy^2} \exp(i\omega t) \Rightarrow W \propto \exp(ky)$$

$$\Rightarrow k^2 = i\frac{\omega}{\nu} \Rightarrow k = (\pm) \frac{1+i}{\sqrt{2}} \sqrt{\frac{\omega}{2\nu}}$$

$$\text{or } k = -(1+i) \sqrt{\frac{\omega}{2\nu}}$$

"minus" $\hat{=}$ damping solution

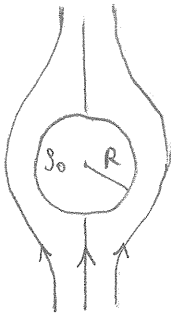
$$\begin{aligned} (1+i)^2 &= 1 + 2i - 1 \\ &= 2i \\ \Rightarrow i &= \frac{(1+i)^2}{2} \end{aligned}$$

Then $u_x = A \exp(i\omega t - i\sqrt{\frac{\omega}{2\nu}} y - \sqrt{\frac{\omega}{2\nu}} y)$

Boundary conditions: $u_x = \text{Im}[U \exp(i\omega t)]$ at $y=0 \Rightarrow A=U$

$$\Rightarrow u_x = U \exp(-\sqrt{\frac{\omega}{2\nu}} y) \sin(\omega t - \sqrt{\frac{\omega}{2\nu}} y)$$

(E 1)

 $\bar{g} \downarrow$ 

$$\rho_f = \alpha \rho_0, \quad \alpha < 1, \quad Re \ll 1 \quad (\text{slow flow})$$

$$U(t) = ?$$

Solution:

$$M \frac{dU}{dt} = Mg - F_{\text{Archim.}} - F_{\text{drag}}$$

$$M = \frac{4\pi}{3} R^3 \rho_0, \quad F_{\text{Arch.}} = \alpha \rho_0 \frac{4\pi}{3} R^3 g = \alpha Mg$$

$$F_{\text{drag}} = 6\pi\mu UR \quad (\text{Stokes formula})$$

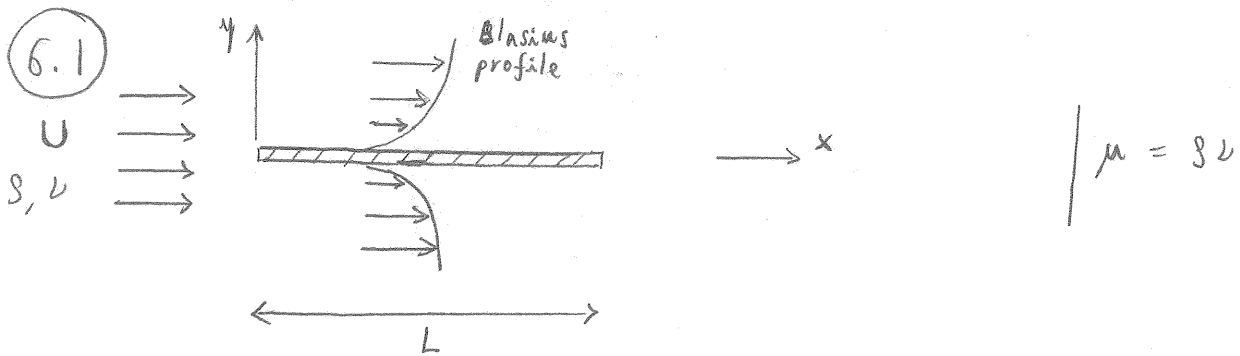
$$\Rightarrow \frac{dU}{dt} = (1-\alpha)g - \frac{6\pi\mu R}{M} U$$

$$\text{Introducing } \tau = \frac{M}{6\pi\mu R} = \frac{\frac{4\pi}{3} \rho_0 R^3}{6\pi\mu R} = \frac{2\rho_0 R^2}{9\mu}$$

$$\Rightarrow \frac{dU}{dt} = (1-\alpha)g - \frac{U}{\tau} \quad | \int$$

$$\Rightarrow \ln \left[(1-\alpha)g\tau - U \right]_{U=0}^U = -\frac{t}{\tau}$$

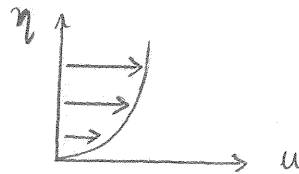
$$\Rightarrow \frac{(1-\alpha)g\tau - U}{(1-\alpha)g\tau} = \exp\left(-\frac{t}{\tau}\right) \quad \Rightarrow \underline{U = (1-\alpha)g\tau \left[1 - \exp\left(-\frac{t}{\tau}\right)\right]}$$



Blasius profile $u = U f(\eta)$

$$\eta = y/\delta(x)$$

$$\frac{df}{d\eta} \approx 0.33 \text{ at } \eta = 0$$



$$F_{\text{drag}}/L_z = ?$$

Solution:

$$F_{\text{drag}}/L_z = \overbrace{2 \int}^{\text{top + bottom}} \mu \frac{\partial u}{\partial y} \bigg|_{\eta=0} dx$$

$$= 2\mu U \underbrace{\left(\frac{df}{d\eta}\right) \bigg|_{\eta=0}}_{= 0.33} \underbrace{\frac{\partial \eta}{\partial y}}_{= \frac{1}{\delta(x)}} dx$$

$$\Rightarrow F_{\text{drag}}/L_z = 0.66 \frac{\mu U}{\delta(x)} dx$$

$$\delta(x) = \sqrt{\frac{Lx}{U}}$$

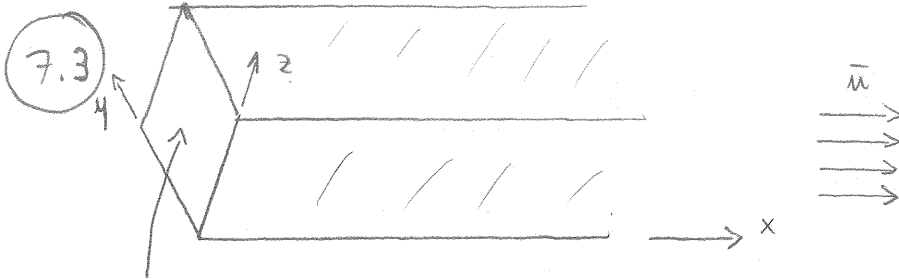
$$\mu = 8L$$

$$= 0.66 \cdot 8L U \sqrt{\frac{U}{Lx}} dx$$

Now

$$F_{\text{drag}}/L_z = 0.66 \cdot 8L U \sqrt{\frac{U}{L}} \int_0^L \frac{dx}{\sqrt{x}} = 2 \cdot 0.66 \cdot 8L U \sqrt{\frac{U}{L}} \sqrt{L}$$

$$= 1.32 \cdot 8L U \sqrt{\frac{UL}{L}}$$



$$T_0 + \theta \sin\left(\frac{\pi y}{R}\right) \sin\left(\frac{\pi z}{R}\right)$$

$$\bar{u} = U \hat{e}_x$$

$$T(x, y, z) = ?$$

Solution:

$$\frac{\partial T}{\partial x} + (\bar{u} \cdot \nabla) T = \chi \nabla^2 T$$

temperature drift

$$\Rightarrow \frac{U}{\chi} \frac{\partial T}{\partial x} = \frac{\partial^2 T}{\partial x^2} + \frac{\partial^2 T}{\partial y^2} + \frac{\partial^2 T}{\partial z^2}$$

$$T = T_0 + \vartheta(x) \sin\left(\frac{\pi y}{R}\right) \sin\left(\frac{\pi z}{R}\right)$$

$$\cancel{\sin} \cancel{\sin} \frac{U}{\chi} \frac{d\vartheta}{dx} = \frac{d^2 \vartheta}{dx^2} \cancel{\sin} \cancel{\sin} - 2 \frac{\pi^2}{R^2} \vartheta \cancel{\sin} \cancel{\sin}$$

$$\vartheta = A \exp(-\alpha x)$$

$$\Rightarrow \alpha^2 + \frac{U}{\chi} \alpha - 2 \frac{\pi^2}{R^2} = 0$$

$$\Rightarrow \alpha = -\frac{U}{2\chi} \pm \sqrt{\frac{U^2}{4\chi^2} + 2 \frac{\pi^2}{R^2}}$$

$$\Rightarrow T = T_0 + \theta \exp(-\alpha x) \sin\left(\frac{\pi y}{R}\right) \sin\left(\frac{\pi z}{R}\right)$$

(7.4)

Boussinesq convection $\rho \approx \text{const}$ in all terms except for $\rho \bar{g}$, where $\rho = \rho_0 + \Delta \rho$

$$\Delta \rho \approx \left(\frac{\partial \rho}{\partial T} \right)_p \Delta T = -\alpha \rho_0 \Delta T, \quad \text{designating } \Delta T = \vartheta$$

\uparrow
constant

Equations:

$$\nabla \cdot \bar{u} = 0$$

$$\frac{\partial \bar{u}}{\partial t} + (\bar{u} \cdot \nabla) \bar{u} = -\frac{1}{\rho_0} \nabla p + \frac{1}{\rho_0} (\rho_0 + \Delta \rho) \bar{g} + \nu \nabla^2 \bar{u}$$

$$\frac{\partial \vartheta}{\partial t} + (\bar{u} \cdot \nabla) \vartheta = \chi \nabla^2 \vartheta$$

stationary

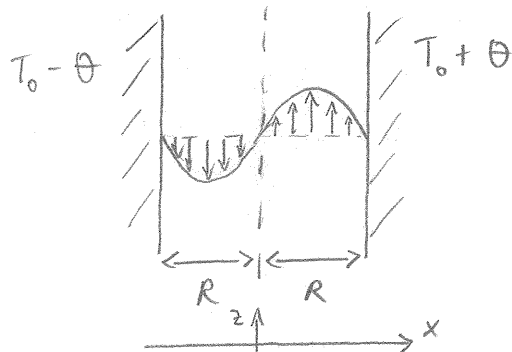
Navier - Stokes:

$$\frac{\partial \bar{u}}{\partial t} + (\bar{u} \cdot \nabla) \bar{u} = -\frac{1}{\rho_0} \nabla p + \bar{g} - \alpha \vartheta \bar{g} + \nu \nabla^2 \bar{u}$$

stationary

$\bar{u} = u_z(x) \hat{e}_z$
shear

pressure coincides with the hydrostatic one



$$\bar{u} = u_z(x) \hat{e}_z$$

$$T(x) = ?$$

$$u_z(x) = ?$$

$$\bar{g} = -g \hat{e}_z$$

$$T = T_0 + \vartheta, \quad \vartheta = \vartheta(x)$$

$$u_z \frac{\partial}{\partial z} u_z = g \alpha \vartheta + \nu \frac{\partial^2 u_z}{\partial x^2}$$

$$u_z \frac{\partial}{\partial z} \vartheta = \chi \nabla^2 \vartheta$$

Then $\frac{\partial^2 \vartheta}{\partial x^2} = 0 \Rightarrow \vartheta = Ax + B$, Boundary conditions

$\vartheta(x=0) = 0 \Rightarrow B = 0$

$\vartheta(x=R) = \theta \Rightarrow A = \frac{\theta}{R}$

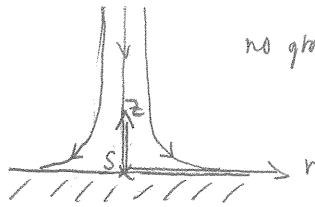
$$\Rightarrow T = T_0 + \theta \frac{x}{R}$$

And $g \alpha \vartheta + \nu \frac{\partial^2 u_z}{\partial x^2} = 0 \Leftrightarrow \frac{\partial^2 u_z}{\partial x^2} = -\frac{g \alpha \theta}{\nu R} x$

$$\Rightarrow u_z = -\frac{g \alpha \theta}{6 \nu R} x^3 + Ax + B, \quad \text{Boundary conditions}$$

$u_z = 0 \text{ at } x = \pm R \Rightarrow u_z = \frac{g \alpha \theta}{6 \nu R} (R^2 - x^2) x$

2.13



no gravity

$$\phi(r, z) = ? \quad \begin{cases} u_r = ? \\ u_z = ? \end{cases}$$

Streamlines = ?

 $p(r, z) = ?$

$$\text{Let's look for } \phi(r, z) = \phi_1(r) + \phi_2(z)$$

$$\nabla^2 \phi = 0 \Leftrightarrow \frac{1}{r} \frac{\partial}{\partial r} \left(r \frac{\partial \phi}{\partial r} \right) + \frac{\partial^2 \phi}{\partial z^2} = 0$$

$$\text{Then } \frac{1}{r} \frac{\partial}{\partial r} \left(r \frac{\partial \phi_1}{\partial r} \right) + \frac{\partial^2 \phi_2}{\partial z^2} = 0 \Rightarrow \begin{cases} \frac{1}{r} \frac{\partial}{\partial r} \left(r \frac{\partial \phi_1}{\partial r} \right) = A & (1) \\ \frac{\partial^2 \phi_2}{\partial z^2} = -A & (2) \end{cases}$$

$$\text{Now } \phi_2 = -\frac{A z^2}{2} + B_2 z + \underbrace{C_2}_{=0}, \quad \text{Choosing } \phi_2(z=0) = 0 \Rightarrow C_2 = 0$$

$$\Rightarrow u_z = \frac{\partial \phi}{\partial z} = \frac{\partial \phi_2}{\partial z} = -A z + B_2$$

$$\text{Bound. cond. } u_z = 0 \text{ at } z=0 \xRightarrow{B_2=0} \underline{u_z = -A z} \quad \text{and} \quad \underline{\phi_2 = -\frac{A z^2}{2}}$$

$$(1) \Rightarrow r \frac{\partial \phi_1}{\partial r} = \frac{A r^2}{2} + B_1$$

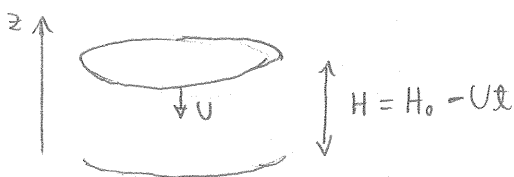
$$\Rightarrow \underline{\phi_1 = \frac{A r^2}{4} + \underbrace{B_1}_{=0} \ln r + \underbrace{C_1}_{=0}}, \quad \text{Choosing } \phi_1(r=0) = 0 \Rightarrow B_1 = 0, C_1 = 0$$

$$\Rightarrow \underline{\phi = \phi_1 + \phi_2 = \frac{A r^2}{4} - \frac{A z^2}{2}} \quad \text{and} \quad \underline{u_r = \frac{\partial \phi}{\partial r} = \frac{\partial \phi_1}{\partial r} = \frac{A r}{2}}$$

$$\text{Streamlines: } \frac{dz}{u_z} = \frac{dr}{u_r} \Leftrightarrow -\frac{dz}{z} = 2 \frac{dr}{r} \Rightarrow \int -\frac{dz}{z} + \text{const}' = 2 \ln r \Rightarrow \ln r^2 + \ln z = \text{const}' \Rightarrow \underline{z r^2 = \text{const.}}$$

$$p(r, z) = ? \quad \text{Bernoulli} \Rightarrow p + \frac{1}{2} \rho u^2 = p_s + 0$$

$$\Rightarrow \underline{p = p_s - \frac{1}{2} \rho A^2 \left(z^2 + \frac{r^2}{4} \right)}$$

Example

$$u_z = -U \Big|_{z=H_0-Ux}$$

On the other hand

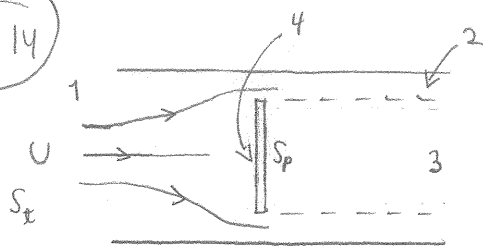
$$u_z = -A z \Big|_{z=H_0-Ux}$$

$$\Rightarrow U = A \cdot (H_0 - Ux)$$

$$\Leftrightarrow A = \frac{U}{H_0 - Ux}$$

$$\text{Then } \phi = \phi(x) \text{ and Bernoulli: } \rho \frac{\partial \phi}{\partial x} + \frac{1}{2} \rho u^2 + p = f(x)$$

2.14



$$a) \Delta P = P_4 - P_3 = ?$$

$$P_4 = P_1 + \frac{1}{2} \rho U^2$$

$$P_2 + \frac{1}{2} \rho U_2^2 = P_1 + \frac{1}{2} \rho U^2$$

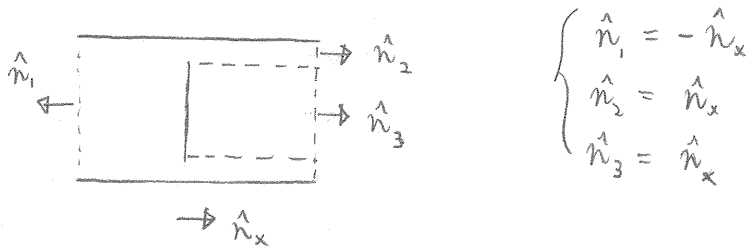
$$\Rightarrow P_3 = P_2 = \underbrace{P_1 + \frac{1}{2} \rho U^2}_{P_4} - \frac{1}{2} \rho U_2^2 \Rightarrow \Delta P = \frac{1}{2} \rho U_2^2$$

$$\text{From continuity eq. } U S_t = U_2 (S_t - s_p)$$

$$\Rightarrow \Delta P = \frac{1}{2} \rho U^2 \frac{s_p^2}{(S_t - s_p)^2}$$

b) $F_{\text{drag}} = ?$, by using momentum conservation

$$\vec{F}_{\text{drag}} = \vec{F} = - \sum (P_i + \rho U_i^2) S_i \hat{n}_i$$



$$\begin{cases} \hat{n}_1 = -\hat{n}_x \\ \hat{n}_2 = \hat{n}_x \\ \hat{n}_3 = \hat{n}_x \end{cases}$$

$$P_2 = P_3$$

$$\begin{cases} F = P_1 S_t + \rho U^2 S_t - P_2 (S_t - s_p) - \rho U_2^2 (S_t - s_p) - P_3 s_p \\ P_2 = P_1 + \frac{1}{2} \rho U^2 - \frac{1}{2} \rho U_2^2 \end{cases}$$

$$\Rightarrow F = \cancel{P_1 S_t} + \rho U^2 S_t - \cancel{P_1 S_t} - \frac{1}{2} \rho U^2 S_t + \frac{1}{2} \rho U_2^2 S_t - \rho U_2^2 (S_t - s_p)$$

$$U_2 = \frac{S_t U}{S_t - s_p} = \frac{1}{2} \rho U^2 \left[S_t + \frac{S_t^3}{(S_t - s_p)^2} - 2 \frac{S_t^2 (S_t - s_p)}{(S_t - s_p)^2} \right]$$

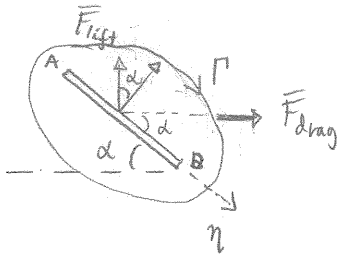
$$= \frac{1}{2} \rho U^2 \left[\frac{S_t^3 - 2 S_t^2 s_p + S_t s_p^2 + S_t^3 - 2 S_t^2 s_p + 2 S_t^2 s_p}{(S_t - s_p)^2} \right]$$

$$= \frac{1}{2} \rho U^2 \frac{S_t s_p^2}{(S_t - s_p)^2}$$

$$F = \frac{C_{\text{drag}}}{2} \rho U^2 s_p$$

$$\Rightarrow C_{\text{drag}} = \frac{S_t s_p}{(S_t - s_p)^2}$$

2.15



Jukowsky - Kutta $\Rightarrow \frac{F_{lift}}{L_z} = -\rho U \Gamma$

On the other hand elementary pressure force

$$\frac{F_{lift}}{L_z} = \int_A^B (P_{bot} - P_{top}) d\eta \cos \alpha$$

$$\frac{F_{drag}}{L_z} = \int_A^B (P_{bot} - P_{top}) d\eta \sin \alpha$$

$$\Rightarrow \frac{F_{drag}}{F_{lift}} = \tan \alpha$$

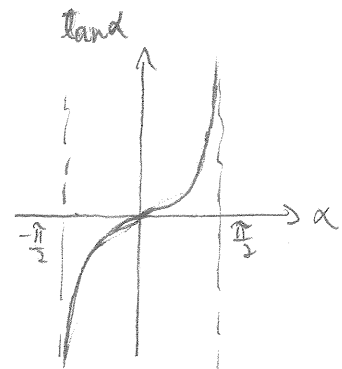
Then

$$\frac{F_{drag}}{L_z} = -\rho U \Gamma \tan \alpha$$

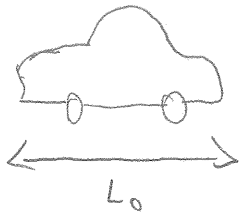
Checking : $\alpha = 0 \Rightarrow F_{drag} = 0$

$$\alpha \rightarrow \frac{\pi}{2} \Rightarrow F_{drag} \rightarrow \infty$$

$\hat{=}$ the model
does not work



5.1



$$\frac{L_0}{L_m} = 10$$

$$\nu_0 = \text{air} = 1.5 \cdot 10^{-5} \frac{\text{m}^2}{\text{s}}$$

$$\nu_m = \text{water} = 10^{-6} \frac{\text{m}^2}{\text{s}}$$

a) $U_m = ?$

$$Re_0 = Re_m$$

$$\Leftrightarrow \frac{U_0 L_0}{\nu_0} = \frac{U_m L_m}{\nu_m}$$

$$\left| \begin{array}{l} Re = \frac{\rho U L}{\mu} \\ \nu = \frac{\mu}{\rho} \end{array} \right.$$

$$\Rightarrow U_m = \frac{L_0 \nu_m}{L_m \nu_0} U_0 = 10 \cdot \frac{10^{-6} \frac{\text{m}^2}{\text{s}}}{1.5 \cdot 10^{-5} \frac{\text{m}^2}{\text{s}}} U_0$$

$$= \underline{\underline{\frac{2}{3} U_0}}$$

b) $F_0 = ?$

$$F' = \frac{F_0}{\rho_0 U_0^2 L_0^2} = \frac{F_m}{\rho_m U_m^2 L_m^2}$$

$$\left| [F] = [\rho U^2 L^3] \right.$$

$$\Rightarrow F_0 = \frac{\rho_0 U_0^2 L_0^2}{\rho_m U_m^2 L_m^2} F_m$$

$$\rho_0 = 1.2 \text{ kg/m}^3$$

$$\rho_m = 10^3 \text{ kg/m}^3$$

$$a) \Rightarrow \left(\frac{U_0 L_0}{U_m L_m} \right)^2 = \left(\frac{L_0}{L_m} \right)^2$$

$$\Rightarrow F_0 = \frac{\rho_0 L_0^2}{\rho_m L_m^2} F_m = \underline{\underline{0.27 F_m}}$$

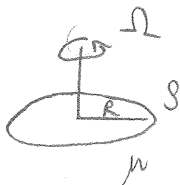
Dimensional analysis:

Buckingham Pi Theorem <sup>units are in balance
(the same in the both sides)</sup>

A dimensionally homogeneous equation involving n variables can be reduced to a relationship among $n-m$ dimensionless products, where m is the minimum number of reference dimensions (= number of basic dimensions)

- Eight steps:
1. List all the variables that are involved in the problem
 2. Express each of the variables in terms of basic dimensions
 3. Determine the required number of pi terms
 4. Select a number of repeating variables, ^(no two repeating variables can have the same dimension) where the number required is equal to the number of reference (=basic) dimensions.
 5. Form a pi term by multiplying one of the nonrepeating variables by the product of the repeating variables, each raised to an exponent that will make the combination dimensionless.
 6. Repeat Step 5 for each of the remaining nonrepeating variables.
 7. Check all the resulting pi terms to make sure they are dimensionless.
 8. Express the final form as a relationship among the pi terms, and think about what it means!

5.3



$$T \propto \Omega$$

$$T \text{ on } R?$$

$$[T] = [FR] = [\rho \Omega^2 R^2 R]$$

1. R, Ω, ρ, μ, T
2. $[R] = [L], [\Omega] = [t^{-1}], [\rho] = \left[\frac{M}{L^3}\right], [\mu] = \left[\frac{M}{L t}\right], [T] = \left[\frac{M}{L^2} \frac{L^2}{t^2} L^3\right]$
3. $\begin{cases} n=5 \\ m=3 \end{cases} ([L], [t], [M]) \Rightarrow n-m=2 \text{ pi terms}$
4. 3 repeating variables (R, Ω, ρ)
5. $[\Pi_1] = \left[\frac{\rho R^2 \Omega}{\mu} \right] = \left[\frac{M}{L^3} \cdot \frac{L^2}{t^2} \cdot \frac{1}{t} \cdot \frac{L^3}{M} \right]$ dimensionless $\rightarrow Re$ continues...

$$Re = \frac{\rho(\Omega R) R}{\mu} = \frac{\rho \Omega R^2}{\mu}$$

5.3 ... continues

6. $[\Pi_2] = \left[\frac{T}{\rho \Omega^2 R^2 R^3} \right] = \left[\frac{\frac{N \cdot s^2}{L}}{\frac{kg}{L^3} \cdot \frac{1}{s^2} \cdot \frac{1}{L^5}} \right]$ dimensionless

8. $\Rightarrow \Pi_2 = f(\Pi_1)$

$\Leftrightarrow \frac{T}{\rho \Omega^2 R^5} = f(Re) \Rightarrow T = \rho \Omega^2 R^5 \cdot C \frac{\mu}{\rho R R^2} = \underline{C \mu \Omega R^3}$

$T \propto \Omega$
 $= \frac{C}{Re}$

Vitaly's way: R, Ω, ρ, μ , $T \propto \Omega$
 $Re = \frac{\rho(\Omega R) \cdot R}{\mu} = \frac{\rho \Omega R^2}{\mu}$
 T on R ?

Dimension of torque

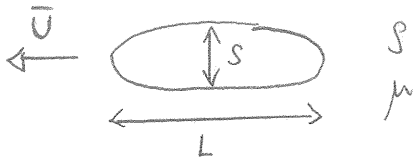
$$[T] = [FR] = [\rho v^2 R^2 R] = [\rho \Omega^2 R^2 R^3] = [\rho \Omega^2 R^5]$$

$$T' = \frac{T}{T_{scaling}} = \frac{T}{\rho \Omega^2 R^5} = f(Re)$$

$$\Rightarrow T = \rho \Omega^2 R^5 f\left(\frac{\rho \Omega R^2}{\mu}\right)$$

$$T \propto \Omega = C \rho \Omega^2 R^5 \cdot \frac{\mu}{\rho R R^2} = \underline{C \mu \Omega R^3}$$

(5.4)



$$F \propto \frac{1}{\sqrt{\rho}}$$

$$F(L) = ?$$

$$F \propto \sqrt{S}$$

Solution: Dimensionless parameters

$$Re = \frac{\rho U L}{\mu} \quad , \quad \frac{S}{L^2}$$

$$[F] = [\rho U^2 S]$$

$$F' = \frac{F}{\rho U^2 S} = f\left(\frac{\rho U L}{\mu}, \frac{S}{L^2}\right)$$

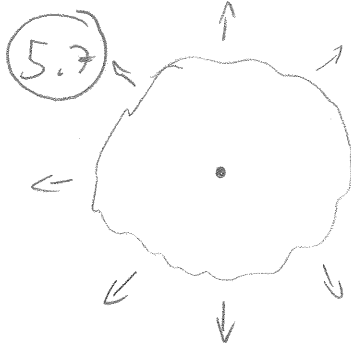
Assuming power laws \Rightarrow

$$F \propto \rho U^2 S \left(\frac{\rho U L}{\mu}\right)^n \left(\frac{S}{L^2}\right)^m = \rho^{n+1} U^{n+2} S^{m+1} L^{n-2m} \mu^{-n}$$

Now $n+1 = -\frac{1}{2} \Rightarrow n = -\frac{3}{2}$

$m+1 = \frac{1}{2} \Rightarrow m = -\frac{1}{2}$

$$\Rightarrow F \propto \sqrt{\frac{\rho U S^3}{L}}$$



$$R = R(t) = ?$$

Sedov - Taylor strong shock

$$E_{kin} \sim \frac{1}{2} M v^2$$

$$\rightarrow E, \rho, R, t$$

\rightarrow only 1 dimensionless parameter

$$[E] = [M v^2] = \left[M \frac{L^2}{t^2} \right] = \left[\rho \frac{L^5}{t^2} \right]$$

$$[\rho] = \left[\frac{M}{L^3} \right]$$

$$\left[\frac{E}{\rho} \right] = \left[\frac{L^5}{t^2} \right]$$

Only one dimensionless parameter $\frac{E t^2}{\rho R^5} = \text{const}$

$$\Rightarrow R = \text{const} \left(\frac{E t^2}{\rho} \right)^{1/5}$$

$$\Rightarrow R \propto t^{2/5}$$

~~~~~