

(01- \ddot{x})

Time Response Analysis & Steady State Error Analysis

Def: The time response of a system is the output of the system as a function of time, when subjected to a given input.

* Time response of a system consists of two parts
 Total response ($y(t)$) = Transient response ($y_t(t)$) + Steady state response ($y_{ss}(t)$)

(i) Transient response: Transient response of the system is the portion of total time response during which the output changes from one value to another value. Or y_t is the response before the output reaches the steady state value.

(ii) Steady State response of the system is the response of the system for a given input after a very long time.

In steady state, the output response settles to its final steady state value or steady state value (y_{ss})

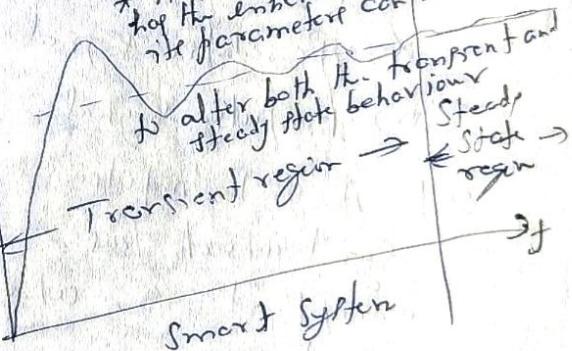
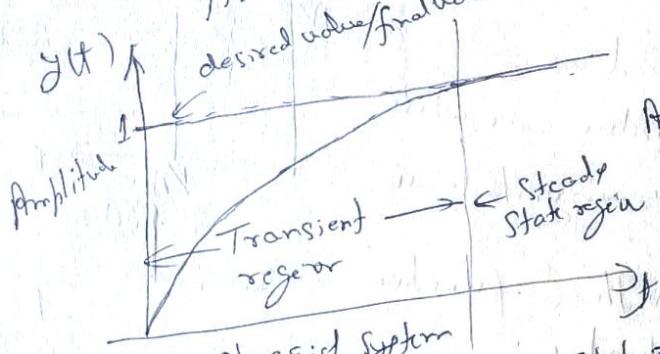


Fig. Step response

Q. What are standard test signals? Discuss in detail.

Ans: Standard test signal or typical test signals: - The signal whose mathematical model is known

(i) Step Signal: $u(t)$ & $u_{st}(t)$

Def: The step signal is a signal

whose value changes from one

level (normally zero) to another level A in zero time. It is denoted by $u(t)$; for $t \geq 0$

$$f(t) = A u(t) \quad \text{when } u(t) = 1; \text{ for } t \geq 0 \\ \& \mathcal{L}[f(t)] = F(s) = \frac{A}{s}$$

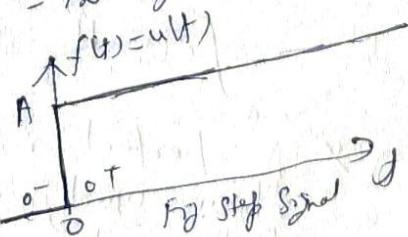


Fig. Step Signal

(ii) Ramp Signal: $\alpha(t)$

Def: The ramp is a signal which starts at a value of zero and increases linearly with time

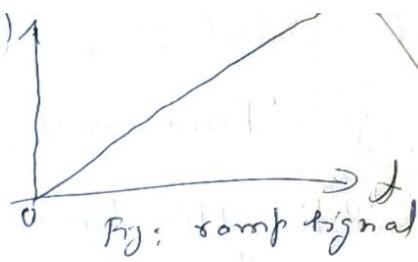


Fig: ramp signal

mathematically; $f(t) = A\alpha(t)$ for $t \geq 0$
 $= 0$ for $t < 0$

$$\mathcal{L}[f(t)] = F(s) = \frac{A}{s^2}$$

If $A=1$, then it is known as unit ramp

* Ramp signal as integral of a step function.

(iii) Parabolic Signal

$$f(t) = \frac{At^2}{2}, \text{ for } t \geq 0$$

$$= 0 \text{ for } t < 0$$

$$\& \mathcal{L}[f(t)] = F(s) = \frac{A}{s^3}$$

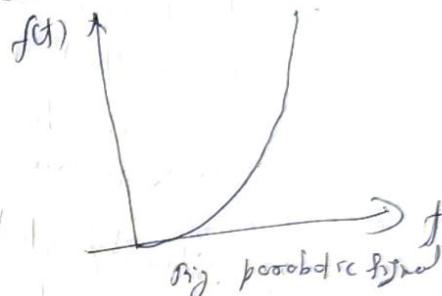


Fig: parabolic signal

* Parabolic signal as integral of a ramp signal

(iv) Impulse Signal

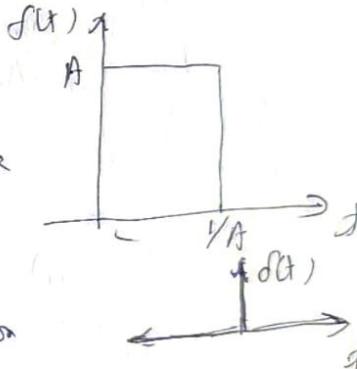
Def: A unit impulse is defined as a signal which has zero value everywhere except $t=0$; where its magnitude is infinity.

It is generally called the δ -function

and has the following property:

$$\epsilon \delta(t) = 0, \quad t \neq 0$$

$$\int \delta(t) dt = 1, \text{ where } \epsilon \text{ tends to zero}$$



* A perfect impulse cannot be achieved in practice, it is usually approximated by a pulse of small width but unit area.

* Mathematically: an impulse function is the derivative of a step function i.e.

$$f(t) = \frac{dy}{dt} = \delta$$

$$\& \mathcal{L}[\delta(t)] = 1$$

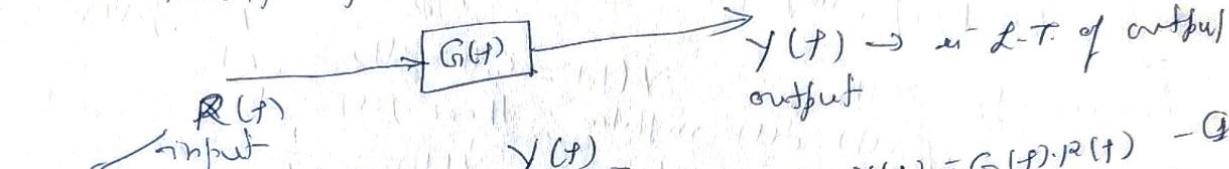
(03-3)

Q. What do you mean by Impulse Response Function?
Discuss in details.

Aq. Impulse Response Function: In a linear, time-invariant system,

For a linear, time-invariant system,

The transfer function $G(s)$ is



L.T. of input $G(s) = \frac{Y(s)}{R(s)}$ output $= Y(s) = G(s) \cdot R(s)$ — ①
convolution × multiplication in the complex domain (for
domain) is equivalent to convolution in the time domain,
so inverse L.T. of eq. ① is given by the following
convolution integral:

$$y(t) = \int_0^t r(\tau) g(t-\tau) d\tau = \int_0^t g(\tau) r(t-\tau) d\tau$$

where $r(t) = \delta(t) = 0$ for $t < 0$ unit impulse input

now consider the output due to unit impulse input
when all the initial conditions are zero,

Since $\mathcal{L}[\delta(t)] = 1 = R(s)$ (say) or $\Delta(s)$

$$\text{i.e. } R(s) = 1 \quad \text{or } \mathcal{L}[r(t)] = \mathcal{L}[G(s)] \quad \text{— ②}$$

from eq ② output $= y(t) = G(s)$

$$\text{or } \mathcal{L}^{-1}[Y(s)] = \mathcal{L}^{-1}[G(s)]$$

or $y(t) = g(t) = \text{impulse-response function}$
of a system, if the inverse

Def: The impulse response of the transfer function.

* This function is also known as weighting function of the system.

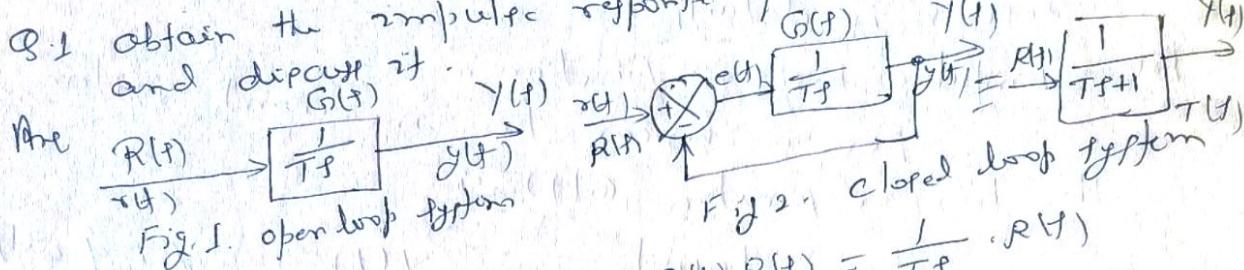
* The function is also known as Green's function and can be used to find the

Weighting function of a system by means of the
system's response to any input $f(t)$, by means of the
Convolution integral

Def: The impulse response function $g(t)$ is the response of a linear
system to a unit impulse input where the initial conditions are zero.
The Laplace transform of this function gives the transfer function
Therefore the transfer function and impulse response function of
a linear time-invariant system contain the same information
about the system dynamics!

(Q4-t)

Time Response of First order System



$$\text{closed) open loop: } Y(s) = G(s) R(s) = \frac{1}{Ts} R(s)$$

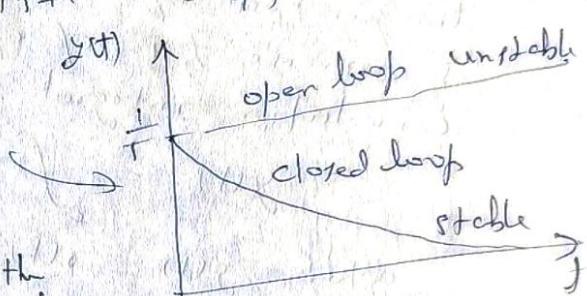
if $\epsilon(s) = \delta(s) = \text{impulse}$, then $R(s) = 1$

$$\therefore Y(s) = \frac{1}{Ts} \Rightarrow Y(s) = \frac{1}{T}$$

$$\text{closed) closed loop: } Y(s) = T(s) \cdot R(s) = \frac{1}{Ts+1} \cdot R(s)$$

$$\text{for impulse } R(s) = 1 \quad \therefore Y(s) = \frac{1}{Ts+1} = \frac{1}{T(s+\frac{1}{T})}$$

$$\therefore Y(s) = \frac{1}{T} e^{-\frac{s}{T}} \quad Y(s)$$



Q2 Derive the expression for the step-response of 1st order system. Also or obtain the step-response of 1st order system. Also define and discuss the concept of time constant.

Ans. case(i) open loop. from fig. 1 (closed) closed loop system

$$Y(s) = \frac{1}{Ts} \cdot R(s) \quad Y(s) = \frac{1}{Ts+1}$$

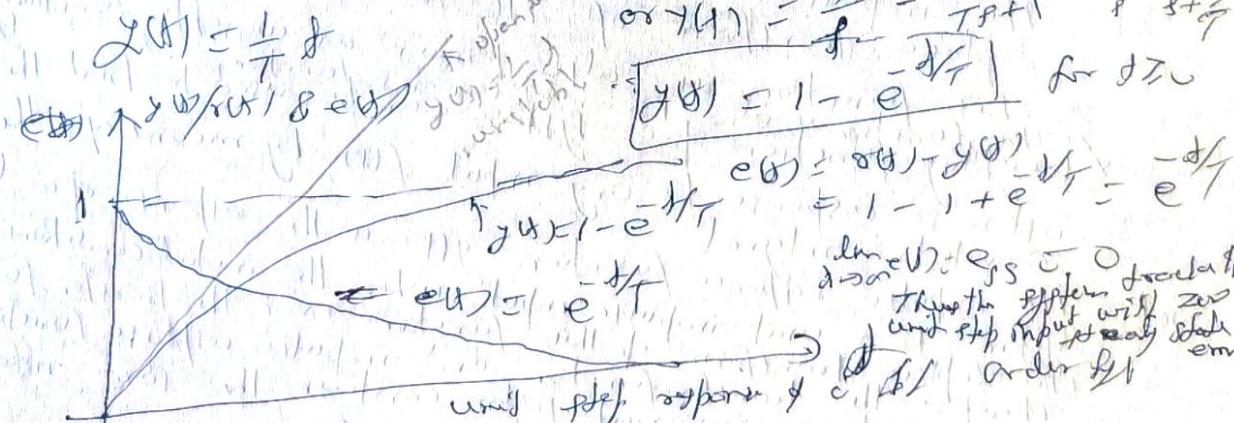
for unit step - $R(s) = \frac{1}{s}$

$$\therefore Y(s) = \frac{1}{Ts} \cdot \frac{1}{s} = \frac{1}{T} \cdot \frac{1}{s}$$

$$Y(s) = \frac{1}{Ts+1} \cdot R(s) = \frac{1}{Ts+1} \cdot \frac{1}{s}$$

$$Y(s) = \frac{1}{s} - \frac{1}{Ts+1} = \frac{1}{s} - \frac{1}{s+\frac{1}{T}}$$

$$Y(s) = 1 - e^{-\frac{s}{T}} \quad \text{for } s > 0$$



(0.5T)

thus output rises exponentially from zero value
to the final value of unity

The initial slope of curve at $t = 0$ is given by

$$\frac{dy}{dt}|_{t=0} = \frac{1}{T} e^{-\frac{t}{T}}|_{t=0} = \frac{1}{T}$$

where T is known as the 'Time Constant' of the system.
Time Constant (T)

$$d\ln y/dt = T$$

$$2B2 = 1 - e^{-\frac{t}{T}} = 1 - e^1 = 1 - 0.368 = 0.632$$

Ques: (1) If the initial step is measured constant,
then after how much time output reaches the
final value, is the time constant of the system

(2) Time constant is the time at which
response reaches 63.2% of its final value.

* Time constant is the measure of how fast
the system tends to reach the final value.

* Smaller the time constant, faster will be the response.

* A larger time constant corresponds to a
sluggish system and a small time constant
corresponds to a ~~fast system~~ system.

* The speed of response can be
quantitatively defined as the time for the output
to become as particular percentage of its final value

$$t=2T, y(2T) = 0.865$$

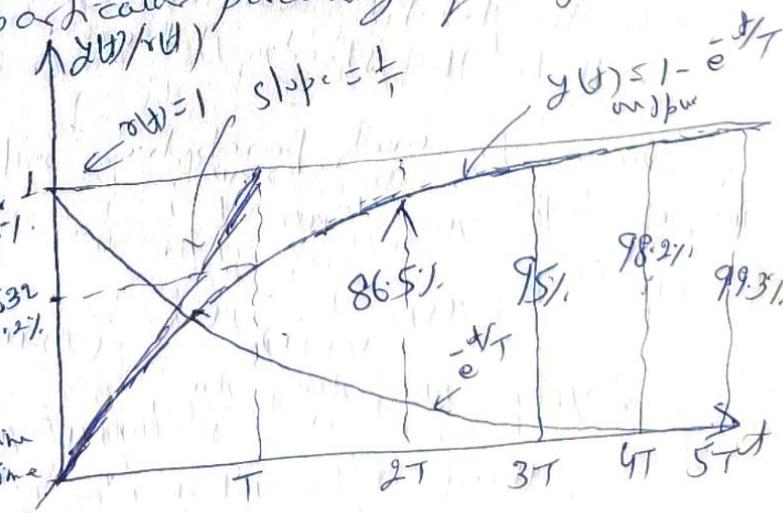
$$t=3T, y(3T) = 0.95$$

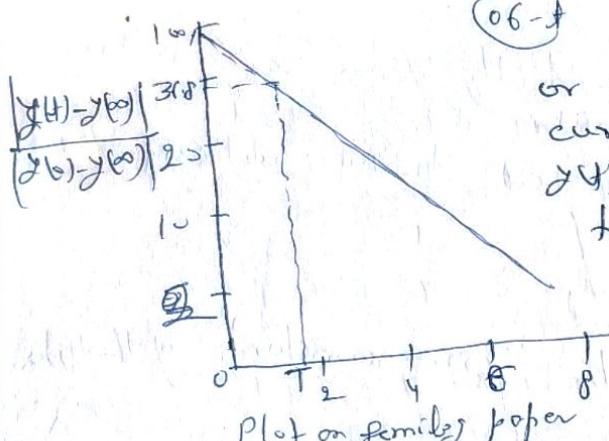
$$t=4T, y(4T) = 0.982$$

$$t=5T, y(5T) = 0.993$$

Thus for $t \geq 4T$ the response $\frac{y(t)}{y(\infty)}$
remains within 2% of the final value
or 63.2%.

* In practice a reasonable estimate of the response time
is the length of the time the response
curve needs to reach the 2% line
of the final value or $4T$ (4 times the time
constant).





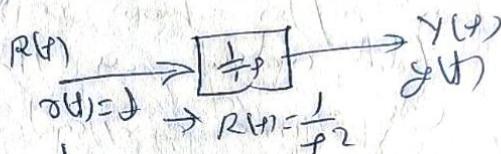
06-t
To determine experimentally whether or not the system is of first order, plot curve $\log|y(t) - y(\infty)|$ as a function of t where $y(t)$ is the output. If the curve turns out to be a straight line, the system is first order. The time constant T can be read from the graph at the time that satisfies the following eq

$$y(T) - y(\infty) = 0.368[y(0) - y(\infty)]$$

Note that instead of plotting $\log|y(t) - y(\infty)|$ vs t , it is convenient to plot $|y(t) - y(\infty)| / |y(0) - y(\infty)|$ vs t on semi-log paper.

Q. Derive the unit-ramp response of 1st-order system and discuss it.

Ans Case (i) open loop



$$Y(s) = \frac{1}{T+s} R(s) = \frac{1}{T+s} \frac{1}{s} = \frac{1}{T+s^2}$$

$$y(t) = \frac{1}{T} t^2$$

Case (ii) closed loop

$$Y(s) = \frac{1}{(Ts+1)} \quad R(s) = \frac{1}{(Ts+1)} \cdot \frac{1}{s^2}$$

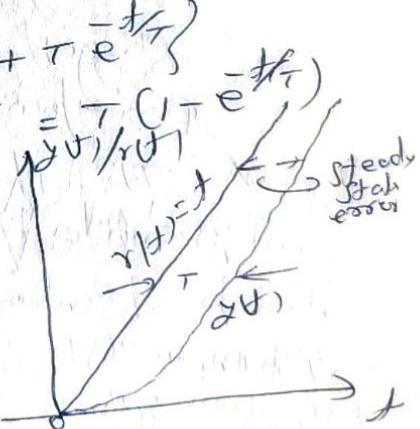
$$\text{or } Y(s) = \frac{A_{11}}{s^2} + \frac{A_{12}}{s} + \frac{B}{Ts+1} = \frac{1}{s^2} - \frac{T}{s} + \frac{T^2}{Ts+1}$$

$$y(t) = t - T + T \cdot e^{-t/T} \quad \text{for } t \geq 0$$

$$\text{error} = e(t) = r(t) - y(t) = t - \{t - T + T e^{-t/T}\} = T - T e^{-t/T} = T(1 - e^{-t/T})$$

$e_{ss} = \lim_{t \rightarrow \infty} e(t) = \lim_{t \rightarrow \infty} T(1 - e^{-t/T}) = T$

* smaller the time constant, the smaller will be the steady state error



Q. Derive the ~~unit~~ parabolic response and discuss it

Ans open loop $\rightarrow R(s) = \frac{1}{T^2} \Rightarrow R(s) = \frac{1}{T^2}$

$$Y(s) = \frac{1}{T^2} R(s) = \frac{1}{T^2} \cdot \frac{1}{s^2} = \frac{1}{T^2 s^2} \Rightarrow y(t) = \frac{1}{T} t^3$$

closed loop

$$Y(s) = \frac{1}{(Ts+1)} R(s) = \frac{1}{(Ts+1)} \cdot \frac{1}{s^2} = \frac{1}{T^2 s^2 (Ts+1)}$$

$$\text{or } Y(s) = \frac{A_{13}}{s^3} + \frac{A_{12}}{s^2} + \frac{A_{11}}{s} + \frac{B}{Ts+1} = \frac{1}{s^3} - \frac{T}{s^2} + \frac{T^2}{s} - \frac{T^3}{Ts+1}$$

$$y(t) = \frac{t^2}{2} - Tt + T^2 - T^2 e^{-t/T}$$

$$e(t) = r(t) - y(t) = \frac{t^2}{2} - \left\{ \frac{t^2}{2} - Tt + T^2 - T^2 e^{-t/T} \right\} = T(t - T + T e^{-t/T})$$

$$e_{ss} = e(\infty) = \infty$$

07- Important property of linear time-invariant systems

For parabolic input, the output $y(t)_{(a)}$ $y(t) = \frac{t^2}{2} - Tt + T^2 - T^2 e^{-t/T}$ (Ans)

For unit ramp input, " " " $y(t) = t - T + T e^{-t/T}$ (Ans),

\downarrow
which is the derivative of parabolic input which is the derivative of parabolic ~~resp~~ response

For unit-step input, which is the derivative of unit-ramp input,

The output $y(t) = 1 - e^{-t/T} \rightarrow$ which is also the derivative of unit step resp

For It unit impulse-input - which is the derivative of unit-step input, the output $y(t)$ is

Conclusion $y(t) = \frac{1}{T} e^{-t/T}$ for $t \geq 0 \rightarrow$ which is also the derivative of step response

- * Thus the response to the derivative of an input signal can be obtained by differentiating the response of the system to the original signal
- * Response to the integral of the original signal can be obtained by integrating the response of the system to the original signal and by determining the integration constant from the zero output initial condition
 \rightarrow These are the property of linear-time invariant system. Linear time-varying systems and non-linear system do not possess this property

Conclusion of 1st order system :-

- (1) ~~we~~ we have seen that if $t \rightarrow \infty$, the output always tends to a finite value, so, first order system can not be unstable.
- (2) First order system is not quick acting as response is exponentially rising.

(8-7)

Step-Response of Second-Order System:

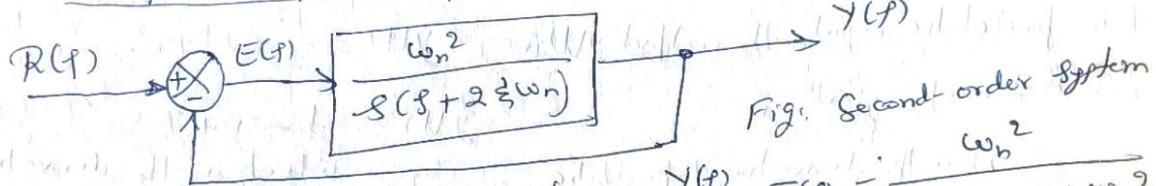


Fig. Second-order system

$$\text{Closed loop transfer function} = \frac{Y(s)}{R(s)} = T(s) = \frac{w_n^2}{s^2 + 2\xi s + w_n^2}$$

where ξ is the damping ratio and w_n is the undamped natural frequency.

The dynamic behaviour of the 2nd-order system can be described in terms of two parameters ξ and w_n .

Case(i) If $\xi = 0$, the system is called undamped.

System and the transient response does not die out. Response is purely oscillatory and lie on the $j\omega$ -axis.

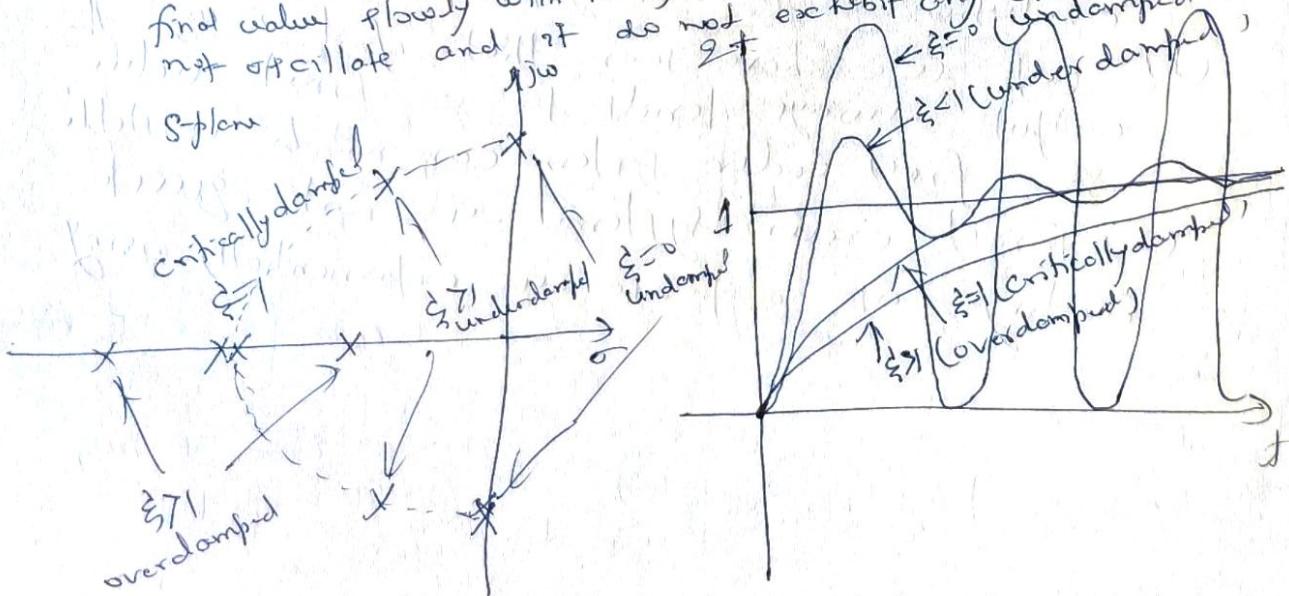
The poles are pure imaginary and lie on the $j\omega$ -axis.

Case(ii) If $0 < \xi < 1$, the closed-loop poles are complex conjugate and lie in the left half of the S-plane.

The system is called underdamped, and the transient response is oscillatory with damping.

Case(iii) If $\xi = 1$, the system is called critically damped. The poles are real, negative and equal. The response rises slowly and reaches the final value. The system do not exhibit any overshoot and transient response do not oscillate.

Case(iv) If $\xi > 1$, the system is called overdamped system, the poles are real, negative and unequal. The output rises to its final value slowly with steady state error. Transient response do not overshoot and it do not exhibit any overshoot.



Underdamped case ($0 < \xi < 1$)

09-f

$$Y(s) = \frac{\omega_n^2}{s^2 + 2\xi\omega_n s + \omega_n^2}$$

$$\xi_1, \xi_2 = \frac{-2\xi\omega_n \pm \sqrt{4\xi^2\omega_n^2 - 4\omega_n^2}}{2}$$

$$= -\xi\omega_n \pm j\omega_n\sqrt{1-\xi^2}$$

$$= -\xi\omega_n + j\omega_d$$

where $\omega_d = \omega_n\sqrt{1-\xi^2} \rightarrow$ damped natural freq

For step input $R(s) = \frac{1}{s}$

$$M(s) = \frac{\omega_n^2}{(s^2 + 2\xi\omega_n s + \omega_n^2)} \cdot \frac{1}{s}$$

$$= \frac{1}{s} - \frac{s + 2\xi\omega_n}{s^2 + 2\xi\omega_n s + \omega_n^2}$$

$$= \frac{1}{s} - \frac{(s + 2\xi\omega_n)}{(s + \xi\omega_n + j\omega_d)(s + \xi\omega_n - j\omega_d)}$$

$$= \frac{1}{s} - \frac{(s + 2\xi\omega_n)}{(s + \xi\omega_n)^2 + \omega_d^2}$$

$$= \frac{1}{s} - \frac{(s + \xi\omega_n)}{(s + \xi\omega_n)^2 + \omega_d^2} - \frac{\xi\omega_n}{(s + \xi\omega_n)^2 + \omega_d^2}$$

using inverse L.T.

$$y(t) = 1 - \frac{1}{2} \left[\frac{s + \xi\omega_n}{(s + \xi\omega_n)^2 + \omega_d^2} \right] - \frac{1}{2} \left[\frac{\xi\omega_n}{(s + \xi\omega_n)^2 + \omega_d^2} \right]$$

$$= 1 - \frac{-\xi\omega_n t}{e^{\xi\omega_n t} \cos \omega_d t} - \frac{1}{2} \left[\frac{\xi\omega_n}{\omega_d} \frac{\omega_d}{(s + \xi\omega_n)^2 + \omega_d^2} \right]$$

$$= 1 - \frac{-\xi\omega_n t}{e^{\xi\omega_n t} \cos \omega_d t} - \frac{\xi\omega_n}{\omega_d} \frac{1}{2} \left[\frac{\omega_d}{(s + \xi\omega_n)^2 + \omega_d^2} \right]$$

$$= 1 - e^{-\xi\omega_n t} \cos \omega_d t - \frac{\xi\omega_n}{\omega_d} e^{-\xi\omega_n t} \sin \omega_d t$$

$$= 1 - e^{-\xi\omega_n t} \cos \omega_d t - \frac{\xi\omega_n}{\omega_n \sqrt{1-\xi^2}} e^{-\xi\omega_n t} \sin \omega_d t$$

$$Z(s) = 1 - e^{-\xi\omega_n t} \left[\cos \omega_d t + \frac{\xi}{\sqrt{1-\xi^2}} \sin \omega_d t \right]$$

$$\text{or } y(t) = 1 - \frac{-\xi\omega_n t}{\sqrt{1-\xi^2}} \left[\sqrt{1-\xi^2} \cos \omega_d t + \frac{\xi}{\sqrt{1-\xi^2}} \sin \omega_d t \right]$$

$$= 1 - \frac{e^{\xi\omega_n t}}{\sqrt{1-\xi^2}} \left[\sin \theta \cos \omega_d t + \cos \theta \sin \omega_d t \right]$$

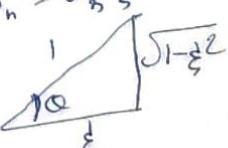
$$= 1 - \frac{e^{\xi\omega_n t}}{\sqrt{1-\xi^2}} \sin [\omega_d t + \theta] = 1 - \frac{e^{\xi\omega_n t}}{\sqrt{1-\xi^2}} \sin \left[\omega_d t + \tan^{-1} \frac{\sqrt{1-\xi^2}}{\xi} \right]$$

$$\begin{aligned} F(s) &\rightarrow e^{-at} \sin \omega_d t \\ \frac{F(s)}{s} &\rightarrow e^{-at} \cos \omega_d t \\ \frac{s+a}{(s+a)^2 + \omega^2} &\rightarrow e^{-at} \frac{-\frac{1}{2}\omega_n t}{\sqrt{1-\xi^2}} \sin \omega_d t \end{aligned}$$

$$\begin{aligned} \frac{s}{s^2 + 2\xi\omega_n s + \omega_n^2} &\rightarrow e^{-at} \frac{\omega_n}{\sqrt{1-\xi^2}} \sin \omega_d t \\ \frac{1}{2} \left[\frac{s + \xi\omega_n}{(s + \xi\omega_n)^2 + \omega_d^2} \right] &\rightarrow e^{-\xi\omega_n t} \cos \omega_d t \end{aligned}$$

$$\begin{aligned} \frac{1}{2} \left[\frac{\omega_d}{(s + \xi\omega_n)^2 + \omega_d^2} \right] &\rightarrow e^{-\xi\omega_n t} \sin \omega_d t \\ \theta &= \tan^{-1} \frac{\sqrt{1-\xi^2}}{\xi} \end{aligned}$$

$$\begin{aligned} \omega_d &= \omega_n \sqrt{1-\xi^2} \\ \omega_d^2 &= \omega_n^2 (1-\xi^2) \\ \omega_d^2 &= \omega_n^2 - \omega_n^2 \xi^2 \end{aligned}$$



$$\cos \theta = \frac{\xi}{\sqrt{1-\xi^2}}$$

$$\sin \theta = \frac{1}{\sqrt{1-\xi^2}}$$

$$\tan \theta = \frac{\sqrt{1-\xi^2}}{\xi}$$

$$\sin(A+B) = \sin A \cdot \cos B + \cos A \cdot \sin B$$

10-8

$$\text{error signal } e(t) = r(t) - y(t) = 1 - y(t)$$

$$= e^{-\xi \omega_n t} \left[\cos \omega_d t + \frac{\xi}{\sqrt{1-\xi^2}} \sin \omega_d t \right] \quad \text{for } t \geq 0$$

- * The error signal exhibits a damped sinusoidal oscillation.
- * At steady state or $t \rightarrow \infty$, no error exists between the input & output i.e. $\lim_{t \rightarrow \infty} e(t) = e_{ss} = 0$
- * If $\xi = 0$, $\omega_d = \omega_n \sqrt{1-\xi^2} = \omega_n \rightarrow$ the response becomes undamped $(\omega_d > \omega_n)$ for $\xi < 0$ and oscillation will continue indefinitely
- * For $\xi = 0$, response $y(t) = 1 - \cos \omega_n t$
- * ω_d is always lower than the ~~ω~~ ω_n
- * An increase in ξ would reduce the damped natural freq ω_d
- * If ξ is increased beyond unity, the response becomes overdamped and will not oscillate.

$$\omega_d = \omega_n \quad \text{for } \xi \leq 0$$

$$\omega_d < \omega_n \quad \text{for } \xi > 0$$

Impulse Response of 2nd order system for $0 < \xi < 1$

$$Y(s) = \frac{\omega_n^2}{(s + \xi \omega_n - j\omega_d)(s + \xi \omega_n + j\omega_d)} \cdot R(s)$$

for impulse $R(s) = 1$

$$\omega_d = \omega_n \sqrt{1 - \xi^2}$$

$$= \frac{A}{s + \xi \omega_n - j\omega_d} \Big|_{s = -\xi \omega_n + j\omega_d} + \frac{B}{s + \xi \omega_n + j\omega_d} \Big|_{s = -\xi \omega_n - j\omega_d}$$

$$A = \frac{\omega_n^2}{s + \xi \omega_n + j\omega_d} \Big|_{s = -\xi \omega_n + j\omega_d} = \frac{\omega_n^2}{-\xi \omega_n + j\omega_d + j\omega_d} = \frac{\omega_n^2}{2j\omega_d}$$

$$= \frac{\omega_n^2}{2j\omega_n \sqrt{1 - \xi^2}} = \frac{\omega_n^2}{2j\sqrt{1 - \xi^2}}$$

$$B = \frac{\omega_n^2}{s + \xi \omega_n - j\omega_d} \Big|_{s = -\xi \omega_n - j\omega_d} = \frac{\omega_n^2}{-\xi \omega_n - j\omega_d + \xi \omega_n - j\omega_d} = \frac{\omega_n^2}{2j\omega_d}$$

$$= -\frac{\omega_n^2}{2j\omega_n \sqrt{1 - \xi^2}} = -\frac{\omega_n^2}{2j\sqrt{1 - \xi^2}}$$

$$Y(s) = \frac{\omega_n}{2j\sqrt{1 - \xi^2}} \left\{ \frac{1}{s + \xi \omega_n - j\omega_d} - \frac{1}{s + \xi \omega_n + j\omega_d} \right\}$$

$$= \frac{\omega_n}{2j\sqrt{1 - \xi^2}} \cdot \frac{1}{\omega_d} \left\{ \frac{\omega_d}{s + \xi \omega_n - j\omega_d} - \frac{\omega_d}{s + \xi \omega_n + j\omega_d} \right\}$$

$$= \frac{\omega_n}{2j\sqrt{1 - \xi^2}} \cdot \frac{1}{\omega_d} \left\{ \frac{\omega_d}{s + (\xi \omega_n - j\omega_d)} - \frac{\omega_d}{s + (\xi \omega_n + j\omega_d)} \right\}$$

$$\begin{aligned}
 y(t) &= \frac{\omega_n}{2j\sqrt{1-\xi^2}} \int e^{-(\xi\omega_n + j\omega_d)t} - e^{-(-\xi\omega_n + j\omega_d)t} \\
 y(t) &= \frac{\omega_n}{2j\sqrt{1-\xi^2}} \left\{ e^{-(\xi\omega_n + j\omega_d)t} - e^{-(-\xi\omega_n + j\omega_d)t} \right\} \\
 &= \frac{\omega_n}{2j\sqrt{1-\xi^2}} \left\{ e^{-\xi\omega_n t} \cdot e^{j\omega_d t} - e^{-\xi\omega_n t} \cdot e^{-j\omega_d t} \right\} \\
 &= \frac{\omega_n}{2j\sqrt{1-\xi^2}} e^{-\xi\omega_n t} \left\{ e^{j\omega_d t} - e^{-j\omega_d t} \right\} \\
 &= \frac{\omega_n}{\sqrt{1-\xi^2}} e^{-\xi\omega_n t} \left(e^{\frac{j\omega_d t}{2j}} - e^{-\frac{j\omega_d t}{2j}} \right) \\
 &= \frac{\omega_n}{\sqrt{1-\xi^2}} e^{-\xi\omega_n t} \cdot \sin \omega_d t
 \end{aligned}$$

or

$$\boxed{y(t) = \frac{\omega_n}{\sqrt{1-\xi^2}} e^{-\xi\omega_n t} \cdot \sin(\omega_n \sqrt{1-\xi^2} t) \xrightarrow{f \neq 0}}$$

Q1 Find the impulse response of the second order system whose transfer function $G(s) = \frac{9}{s^2 + 4s + 9}$

Ans: method 1. comparing $G(s) = \frac{9}{s^2 + 4s + 9} = \frac{\omega_n^2}{s^2 + 2\xi\omega_n s + \omega_n^2}$
 $\Rightarrow \omega_n = 3, 2\xi = 4 \Rightarrow \xi = \frac{4}{2\omega_n} = \frac{4}{2 \cdot 3} = \frac{2}{3} = 0.67$

Impulse response $y(t) = \frac{\omega_n}{\sqrt{1-\xi^2}} e^{-\xi\omega_n t} \sin(\omega_n \sqrt{1-\xi^2} t)$

or $y(t) = \frac{3}{\sqrt{1-(0.67)^2}} \cdot e^{-\frac{2}{3} \cdot 3t} \cdot \sin(3\sqrt{1-(0.67)^2} t)$
 $= 4.025 e^{-2t} \cdot \sin 2.235 t$

method 2 Roots of $s^2 + 4s + 9$ are $s_1, s_2 = \frac{-4 \pm \sqrt{4^2 - 4 \cdot 9}}{2} = -2 \pm j2.236$

$$G(s) = \frac{9}{\{s - (-2 + j2.236)\} \{s - (-2 - j2.236)\}}$$

$$Y(s) = G(s) \text{ since } R(s) = 1$$

$$\begin{aligned} Y(s) &= \frac{A}{s - (-2 + j2.236)} + \frac{A^*}{s - (-2 - j2.236)} \\ &= \frac{-j2.0125}{s - (-2 - j2.236)} + \frac{j2.0125}{s - (-2 + j2.236)} \\ \therefore y(t) &= -j2.0125 e^{(-2+j2.236)t} + j2.0125 e^{(-2-j2.236)t} \\ &= 4.025 e^{-2t} \sin 2.236 t \end{aligned}$$

H.W Q2. Find the impulse response of a second order system whose transfer function is

$$(i) G(s) = \frac{4}{s^2 + 4s + 4}$$

$$\text{Ans: } 4t e^{-2t}$$

$$(ii) G(s) = \frac{4}{s^2 + 2s + 4}$$

$$\text{Ans: } 2.3 e^{-t} \sin(1.732t)$$

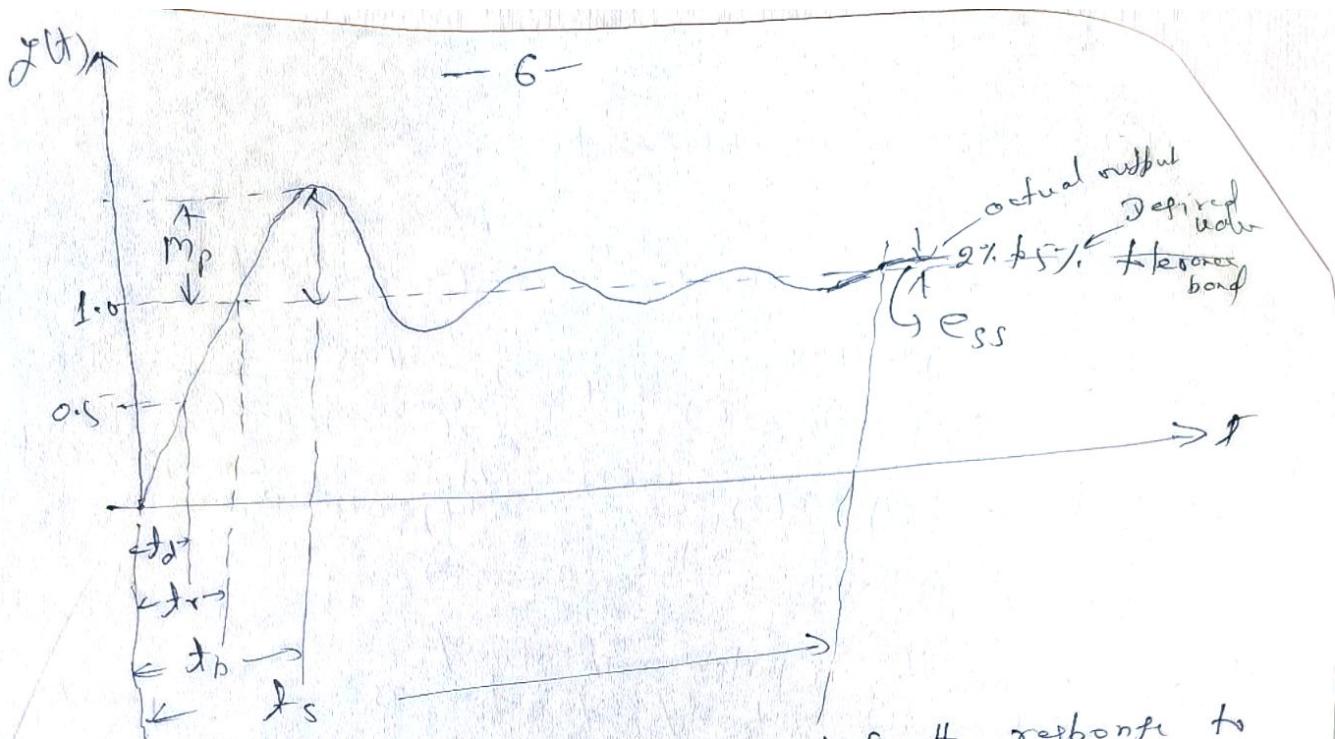
$$(iii) G(s) = \frac{4}{s^2 + 8s + 4}$$

$$\text{Ans: } 1.15 e^{-4t} \sin t (2\sqrt{3}t)$$

Q. Define and discuss transient response specifications of 2nd order system or time domain specification.

Ans. The performance of a control system are express in terms of the transient response to a unit step input because it is easy to generate. The transient response of a control system to a unit step input depends upon the initial conditions. Hence consider a 2nd order system with unit-step input and the system is initially at rest i.e. all initial conditions are zero. Common transient response specifications are

- | | |
|--------------------------|-------------------------------------|
| (1) Delay time (t_d) | (4) Maximum overshoot (M_p) |
| (2) Rise time (t_r) | (5) Settling time (t_s) |
| (3) Peak time (t_p) | (6) Steady state error (e_{ss}) |



- (1) Delay time, t_d → Time required for the response to reach 50% of the final value in first time.
- (2) Rise time, t_r → Time required for the response to rise from 0 to 100% of its final value in the first time for underdamped system. and 10% to 90% for overdamped system.
- (3) Peak time, t_p → Time required for the response to reach the first peak of the time response or first peak overshoot.
- (4) Maximum Overshoot, M_p : — If $y(t)$ is the normalized difference between the peak of the time response and steady state output. Maximum percentage overshoot is defined by $\frac{y(t_p) - y(\infty)}{y(\infty)} \times 100$
- (5) Settling time (t_s) : — Time required for the response to reach and stay within the specified range (2% to 5%) of its final value.
- (6) Steady state error (e_{ss}) → If ϵ is the difference between actual output and desired output of time 't' tends to infinity
- $$e_{ss} = \lim_{t \rightarrow \infty} [x(t) - y(t)] = \lim_{t \rightarrow \infty} \epsilon(t) = E(s)$$

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Expression for time domain specification

(i) Rise time, t_r ,

$$y(t) = 1 - \frac{e^{-\xi \omega_n t}}{\sqrt{1-\xi^2}} \sin \left[\omega_n t + \tan^{-1} \frac{\sqrt{1-\xi^2}}{\xi} \right]$$

$$\text{Put } y(t) = 1 \\ \Rightarrow 1 = 1 - \frac{e^{-\xi \omega_n t}}{\sqrt{1-\xi^2}} \sin \left[\omega_n t + \tan^{-1} \frac{\sqrt{1-\xi^2}}{\xi} \right]$$

$$\text{or } \frac{e^{-\xi \omega_n t}}{\sqrt{1-\xi^2}} \sin \left[\omega_n t + \tan^{-1} \frac{\sqrt{1-\xi^2}}{\xi} \right] = 0$$

$$\text{Since } e^{-\xi \omega_n t} \neq 0$$

$$\therefore \sin \left[\omega_n t + \tan^{-1} \frac{\sqrt{1-\xi^2}}{\xi} \right] = 0 = \sin n\pi$$

$$\text{Put } n=1$$

$$\therefore \omega_n t_r + \tan^{-1} \frac{\sqrt{1-\xi^2}}{\xi} = \pi$$

$$\therefore t_r = \frac{\pi - \tan^{-1} \frac{\sqrt{1-\xi^2}}{\xi}}{\omega_n (\text{=} \omega_n \sqrt{1-\xi^2})}$$

(ii) Peak time t_p

$$y(t) = 1 - \frac{e^{-\xi \omega_n t}}{\sqrt{1-\xi^2}} \sin \left[\omega_n t + \phi \right]$$

$$\text{where } \phi = \tan^{-1} \frac{\sqrt{1-\xi^2}}{\xi}$$

for maximum, $\frac{dy(t)}{dt} = 0$

$$\frac{dy}{dt} = \frac{-\xi \omega_n}{\sqrt{1-\xi^2}} \cos \left[(\omega_n \sqrt{1-\xi^2})t + \phi \right] \omega_n \sqrt{1-\xi^2} + \sin \left[(\omega_n \sqrt{1-\xi^2})t + \phi \right] \frac{\xi \omega_n}{\sqrt{1-\xi^2}} e^{-\xi \omega_n t} = 0$$

$$\text{Since } e^{-\xi \omega_n t} \neq 0$$

on solving

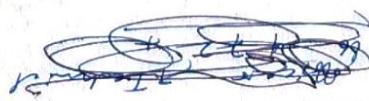
$$t_p = \frac{\pi}{\omega_n \sqrt{1-\xi^2}}$$

First minimum (under transient) occurs at $t = \frac{\pi}{\omega_n \sqrt{1-\xi^2}}$

(iii) Expression for Maximum Overshoot: M_p

Maximum overshoot occurs at peak time $t = t_p$

$$\text{Put } t = t_p = \frac{\pi}{\omega_n \sqrt{1-\xi^2}}$$



$$y(t) = 1 - \frac{-\xi \omega_n \frac{\pi}{\omega_n \sqrt{1-\xi^2}}}{\sqrt{1-\xi^2}} \sin \left[\omega_n \sqrt{1-\xi^2} t + \frac{\pi}{\omega_n \sqrt{1-\xi^2}} + \phi \right]$$

where $\phi = \tan^{-1} \frac{\sqrt{1-\xi^2}}{\xi}$ Then $\sin \phi = \sqrt{1-\xi^2} \phi \sin(\pi + \phi)$

$$= 1 - \frac{-\xi \omega_n}{\sqrt{1-\xi^2}} \sin(\pi + \phi)$$

$$\therefore y(t) = 1 + \frac{\xi \omega_n}{\sqrt{1-\xi^2}} \sin \phi$$

$$y(t)_{\text{max}} = 1 + \frac{\xi \omega_n}{\sqrt{1-\xi^2}}$$

$$\% M_p = y(t)_{\text{max}} - 1 = \frac{\xi \omega_n}{\sqrt{1-\xi^2}}$$

$$\% M_p = e^{-\frac{\pi \xi}{\sqrt{1-\xi^2}}} \times 100$$

(V) Steady state error e_{ss}

$$e_{ss} = \lim_{t \rightarrow \infty} e(t) = 1 - y(t)$$

$$= \lim_{t \rightarrow \infty} \frac{e^{-\xi \omega_n t}}{\sqrt{1-\xi^2}} \sin(\omega_n t + \phi)$$

$$\text{for a ramp input } r(t) = \frac{t}{K_2}$$

$$e_{ss} = \frac{2\xi}{\omega_n} = \frac{1}{K_2}$$

$$e_{ss} = \lim_{t \rightarrow \infty} [r(t) - y(t)]^{50}$$

$$\text{for step input}$$

$$\text{for } 2\% \text{ criterion (of steady state error)} \\ \text{for } \xi = 0.76$$

$$t_s = \frac{4}{\xi \omega_n} = \frac{4}{\sigma} \text{ for } 2\% \text{ criterion (of steady state error)}$$

$$t_s = \frac{3}{\xi \omega_n} = \frac{3}{\sigma} \text{ for } 5\% \text{ criterion (of steady state error)}$$

(iv) Settling time:
inversely proportional
 $t_s \propto \xi \text{ & } \omega_n$

Q1 When a 2nd order control system is subjected to a unit step input, the value of $\xi = 0.5$ and $\omega_n = 6 \text{ rad/sec}$. Determine the rise time, peak time, settling time and a peak overshoot.

Soh. $\xi = 0.5$; $\omega_n = 6 \text{ rad/sec}$

$$(1) \text{ Rise time: } t_r = \frac{\pi}{\omega_n \sqrt{1-\xi^2}} = \frac{\pi}{6 \sqrt{1-(0.5)^2}} = \frac{\pi}{6 \sqrt{0.75}} = \frac{3.14 - 1.047}{5.19} = 0.403 \text{ sec}$$

$$(2) \text{ Peak time, } t_p = \frac{\pi}{\omega_n \sqrt{1-\xi^2}} = \frac{\pi}{6 \sqrt{1-(0.5)^2}} = 0.605 \text{ sec}$$

$$(3) \text{ Settling time } t_s = \frac{4}{\xi \omega_n} = \frac{4}{0.5 \times 6} = 1.33 \text{ sec}$$

$$(4) \text{ Maximum overshoot: } M_p = e^{\frac{-\pi \xi}{\sqrt{1-\xi^2}}} \times 100 = e^{\frac{-\pi \times 0.5}{\sqrt{1-(0.5)^2}}} \times 100$$

$$M_p = 0.163 \times 100 = 16.3\%$$

~~16.3%~~
16.3%

Q. Obtain the unit-impulse response and the unit-step response of a unity feedback system whose open-loop transfer function is

$$G(s) = \frac{2s+1}{s^2}$$

Soln. closed loop tf for $\frac{Y(s)}{R(s)} = \frac{(2s+1)/s^2}{1+(2s+1)/s^2} = \frac{2s+1}{s^2+2s+1} = \frac{2s+1}{(s+1)^2}$

For unit impulse response $R(s) = 1$,

$$\therefore Y(s) = \frac{(2s+1)}{(s+1)^2}$$

$$Y(t) = L^{-1}\left[\frac{(2s+1)}{(s+1)^2}\right] = L^{-1}\left[\frac{A}{s+1} + \frac{B}{(s+1)^2}\right] = L^{-1}\left[\frac{2}{s+1} - \frac{1}{(s+1)^2}\right]$$

$$\text{or } y(t) = 2L^{-1}\left[\frac{1}{s+1}\right] - L^{-1}\left[\frac{1}{(s+1)^2}\right] = 2e^{-t} - t e^{-t} \text{ Ans}$$

For unit step response; $R(s) = \frac{1}{s}$

$$Y(s) = \frac{(2s+1)}{s(s+1)^2} = \frac{A}{s} + \frac{B}{s+1} + \frac{C}{(s+1)^2}$$

$$= \frac{1}{s} - \frac{1}{s+1} + \frac{1}{(s+1)^2}$$

$$y(t) = 1 - e^{-t} + t e^{-t}$$

X Q. The open-loop transfer function of a system with unity feedback gain is given as $G(s) = \frac{20}{s^2+5s+6}$. Determine the damping ratio, maximum overshoot, rise time and break frequency.

$$\text{Ans } \frac{Y(s)}{R(s)} = \frac{G(s)}{1+G(s)} = \frac{\frac{20}{s^2+5s+6}}{1+\frac{20}{s^2+5s+6}} = \frac{20}{s^2+5s+26}$$

$$\text{Comparing with } \frac{\omega_n^2}{s^2+2\zeta\omega_n s+\omega_n^2} \Rightarrow \omega_n^2 = 26 \text{ ; } 2\zeta\omega_n = 5$$

$$\omega_n = \sqrt{26} \text{ ; } \zeta = \frac{5}{2\sqrt{26}} = 0.49$$

$$\zeta = 0.49 \quad \text{rise time } \frac{\pi}{\zeta\omega_n} = \frac{\pi}{0.49\sqrt{26}} \times 100 = 17.1 \text{ sec}$$

$$M_p = \frac{-\pi\sqrt{1-\zeta^2}}{\omega_n} \times 100 = 2 \times \frac{\pi}{0.49\sqrt{26}} \times 100 = 1.059 \text{ rad}$$

$$\text{Tr} = \frac{\pi - \phi}{\omega_n} \text{ where } \phi = \tan^{-1} \frac{\sqrt{1-\zeta^2}}{\zeta} \text{ sec}$$

$$\text{Tr} = \frac{\pi - 1.059}{\omega_n} = \frac{\pi - 1.059}{\sqrt{26} \sqrt{1-(0.49)^2}} = 468.53 \times 10^{-3} \text{ sec}$$

$$t_p = \frac{\pi}{\omega_n \sqrt{1-\zeta^2}} = \frac{\pi}{\sqrt{26} \sqrt{1-(0.49)^2}} = 706.78 \times 10^{-3} \text{ sec}$$

$$= 706.78 \text{ msec}$$

(OJ - Error) Error Analysis

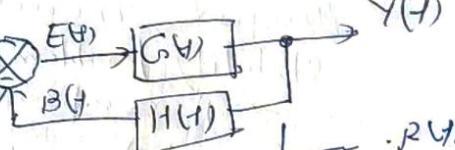
Steady State Error Analysis

Dynamic error Analysis
Generalized error Analysis

Steady State Error Analysis:

Steady State error: g_t is defined as the difference between output and input when $t \rightarrow \infty$.
 $\therefore g_t$ is denoted by e_{ss} or $\lim_{t \rightarrow \infty} e(t)$ i.e. $e_{ss} = \lim_{t \rightarrow \infty} e(t)$

For a ~~unity~~ feedback system:



1. Error transfer function = $\frac{E(s)}{R(s)} = \frac{1}{1 + G(s)H(s)} \Rightarrow e_{ss} = \frac{1}{1 + G(s)H(s)} \cdot R(s)$

$\therefore e_{ss} = \lim_{s \rightarrow 0} e(s) = \lim_{s \rightarrow 0} sE(s) = \lim_{s \rightarrow 0} \frac{sR(s)}{1 + G(s)H(s)} \rightarrow$ using final value theorem

Static Error Constants: These are figures of merit of control systems. The higher the constants, the smaller is the steady-state error.

Order & type of the system:
order → The highest power of the complex variable s in the denominator of the transfer function is called the order of the system → order is 2

$$G(s) = \frac{K}{s^2 + 4s + 9}$$

Type: control system may be classified according to their ability to follow step inputs, ramp inputs, parabolic inputs and so on. The magnitude of the steady-state error due to these individual inputs are indicative of the goodness of the system.

The open loop transfer of a unity feedback system can be written in two standard forms

(a) ~~Constant form~~

$$\text{Pole-Zero form} \quad G(s)H(s) = \frac{K(s+z_1)(s+z_2)\dots(s+z_m)}{(s+p_1)(s+p_2)\dots(s+p_p)}$$

(b) Time constant form

$$G(s)H(s) = \frac{K(T_0s+1)(T_1s+1)\dots(T_ms+1)}{s^n(T_1s+1)(T_2s+1)\dots(T_ps+1)}$$

The gains of the two forms are related by

$$K = K \prod_{j=1}^m z_j$$

(Q2-Err)

In the above two representation ' s^n ' in the denominator represents a pole of multiplicity 'n' at the origin.

Def'n Type - No. of integrations indicated by the open loop tr. for

If $n=0 \rightarrow$ type 0, $n=1 \rightarrow$ type -1, $n=2 \rightarrow$ type 2 etc. and so on

* As the type is increased, accuracy is improved.

However, increasing the type number aggravates the stability problem.

→ A compromise between steady-state accuracy and selective stability is always necessary.

→ In practical system, it is more to have

type-3 or higher system, because it is generally difficult

to design stable systems having more than two

integrations in the forward path.

* Open-loop gain K is directly related to steady-state error.

-: Static Error Constants: -

1) Static Position Error Constant (K_p)

or Position error coefficient.

It is defined using step-input.

$$e_{ss} = \lim_{t \rightarrow \infty} e(t) = \lim_{s \rightarrow 0} sE(s) = \lim_{s \rightarrow 0} s \cdot \frac{1}{1+GH(s)} RH$$

$$\text{for step input } RH = \frac{1}{s}$$

$$e_{ss} = \lim_{s \rightarrow 0} s \cdot \frac{1}{1+GH(s)} \cdot \frac{1}{s} = \lim_{s \rightarrow 0} \frac{1}{1+GH(s)} = \frac{1}{1+G(0)H(0)}$$

$$K_p = \frac{1}{e_{ss}} - 1 = \frac{1 - e_{ss}}{e_{ss}}$$

$$\boxed{\text{or } e_{ss} = \frac{1}{1 + K_p}}$$

$$\text{where } \boxed{K_p = \lim_{s \rightarrow 0} G(s)H(s)}$$

For a type-1 system:

For a type-0 system:

$$K_p = \lim_{s \rightarrow 0} \frac{K(T_0s+1)(T_1s+1)}{(T_1s+1)(T_2s+1)} = K \quad (K_p = \lim_{s \rightarrow 0} \frac{K(T_0s+1)(T_1s+1)}{s(T_1s+1)(T_2s+1)}) = \infty$$

* For a type-0 system, the static position error constant K_p is finite, while for a type-1 or higher system K_p is infinite.

(03-Err)

For a unit step input, error may be of:

$$\text{for type-0 system, } e_{ss} = \frac{1}{1+K_p} \neq \frac{1}{1+K}$$

$$\text{in type-1 or higher system, } e_{ss} = \frac{1}{1+K_p} = \frac{1}{1+\infty} = 0$$

* Thus, the response of a feedback control system to a step input involves a steady-state error if there is no integration in the forward path.

→ If a small error for step inputs can be tolerated, then a type-0 system may be permissible, provided that the gain K is sufficiently large.

→ If the gain K is too large, it is difficult to obtain reasonable relative stability.

→ If zero steady-state error for a step input is desired, the type of the system may be one or higher.

(ii) Static-Velocity Error Constant Or Coefficient (K_v)

K_v is defined using ramp input. For ramp input $R(s) = \frac{1}{s^2}$

$$e_{ss} = \lim_{s \rightarrow 0} s E(s) = \lim_{s \rightarrow 0} s \frac{1}{s^2 + G(s)H(s)} = \lim_{s \rightarrow 0} \frac{s}{s^2 + G(s)H(s)} = \frac{1}{1 + G(0)H(0)}$$

$$e_{ss} = \lim_{s \rightarrow 0} \frac{1}{s + G(s)H(s)} = \frac{1}{\lim_{s \rightarrow 0} s G(s)H(s)} = \frac{1}{K_v}$$

$$\therefore e_{ss} = \frac{1}{K_v} \quad \text{where } K_v = \lim_{s \rightarrow 0} s G(s)H(s)$$

* The term velocity error is used to express the steady-state error for a ramp input. The dimension of the velocity error is the same as the system error. That is, velocity error is not an error in velocity, but it is an error in position due to a ramp input.

For a type-0 system

$$K_v = \lim_{s \rightarrow 0} \frac{s K(T_b s + 1)(T_a s + 1)}{(T_b s + 1)(T_a s + 1)} = 0 \implies e_{ss} = \frac{1}{K_v} = \frac{1}{0} = \infty$$

For a type-1 system

$$K_v = \lim_{s \rightarrow 0} \frac{s K(T_b s + 1)(T_a s + 1)}{s(T_b s + 1)(T_a s + 1)} = K \implies e_{ss} = \frac{1}{K_v} = \frac{1}{K} \rightarrow \text{finite value}$$

For a type-2 system

$$K_v = \lim_{s \rightarrow 0} \frac{s K(T_b s + 1)(T_a s + 1)}{s^2(T_b s + 1)(T_a s + 1)} = \infty \implies e_{ss} = \frac{1}{K_v} = \frac{1}{\infty} = 0$$

- (04-Error) (Ch 9.2)
- * They, a type-0 system is incapable of following a ramp in the steady-state.
 - * Type-1 system with unity feedback can follow the ramp input with a finite error.
 - * Type-2 or higher system can follow a ramp input with zero actuating error at steady state.

(iii) Static acceleration error constant or coefficient (K_a)

Acceleration error is defined using parabolic input.

$$r(t) = \frac{t^2}{2} \text{ for } t > 0 \Rightarrow R(s) = \frac{1}{s^3}$$

$$\text{ess} = \lim_{s \rightarrow 0} s^2 E(s) = \lim_{s \rightarrow 0} s^2 \frac{1}{1 + G(s)H(s)} = \lim_{s \rightarrow 0} \frac{s^2}{1 + G(s)H(s)} s^3 = \frac{1}{1 + G(0)H(0)} s^3$$

$$\text{or } e_{ss} = \lim_{s \rightarrow 0} \frac{1}{s^2 + s^2 G(s)H(s)} = \frac{\lim_{s \rightarrow 0} s^2 G(s)H(s)}{\lim_{s \rightarrow 0} s^2 + s^2 G(s)H(s)} = \frac{1}{K_a}$$

$$\boxed{e_{ss} = \frac{1}{K_a}}$$

$$\boxed{K_a = \lim_{s \rightarrow 0} s^2 G(s)H(s)}$$

Thus, the acceleration error, in the steady state is the difference between the position error due to a parabolic input in or error in position

For a type-0 system:

$$K_a = \lim_{s \rightarrow 0} \frac{s^2 K(T_0 s + 1)(T_1 s + 1)}{(T_1 s + 1)(T_2 s + 1)} = 0 \Rightarrow e_{ss} = \frac{1}{K_a} = \frac{1}{0} = \infty$$

$$\text{For a type-1 system: } \lim_{s \rightarrow 0} \frac{s^2 K(T_0 s + 1)(T_1 s + 1)}{s(T_1 s + 1)(T_2 s + 1)} = 0 \Rightarrow e_{ss} = \frac{1}{K_a} = \frac{1}{0} = \infty$$

$$\text{For a type-2 System: } K_a = \lim_{s \rightarrow 0} \frac{s^2 K(T_0 s + 1)(T_1 s + 1)}{s^2 (T_1 s + 1)(T_2 s + 1)} \leq K \Rightarrow e_{ss} = \frac{1}{K_a} = \frac{1}{K} = \text{finite}$$

For a type-3 system:

$$K_a = \lim_{s \rightarrow 0} \frac{s^2 K(T_0 s + 1)(T_1 s + 1)}{s^3 (T_1 s + 1)(T_2 s + 1)} = \infty \Rightarrow e_{ss} = \frac{1}{K_a} = \frac{1}{\infty} = 0$$

* They, type-0 and type-1 systems are incapable of following a parabolic input in the steady state.

* The type-2 system with unity feedback can follow a parabolic input with a finite actuating error signal.

| Type | Step input $\delta(t) = 1$ | Ramp input $\delta(t) = t$ | Parabolic input $\delta(t) = \frac{t^2}{2}$ |
|---------------|-------------------------------|-------------------------------|--|
| Type-0 System | $\frac{1}{1+K}$ | ∞ | ∞ |
| Type-1 System | 0 | $\frac{1}{K}$ | ∞ |
| Type-2 System | 0 | 0 | $\frac{1}{K}$ |

- * In table, the finite values for steady-state error appear on the diagonal line. Above the diagonal the steady-state errors are infinite, below the diagonal, they are zero.
- * The error constants K_p , K_v & K_a describe the ability of a system to reduce or eliminate steady-state error. Therefore, they are indicative of the steady-state performance.
- * It is generally desirable to increase the error constant while maintaining the transient response within an acceptable range.
- * If there is any conflict between the static velocity error constant and the static ~~acceleration~~ acceleration error constant, then the latter may be considered less important than the former. Then the latter may be considered less important than the former. Then the latter may be considered less important than the former.
- * To improve the steady-state performance we can increase the type of the system by adding an integrator or integrator to the forward path. This introduces additional stability problem.
- * The design of a satisfactory system with more than two integrators in series in the forward path is generally difficult.

Disadvantages of Error Constant method

- (i) The error constant method does not apply to the system with inputs that are sinusoidal, since the final value theorem cannot apply to these cases.
- (ii) Static error constants do not give any information on the steady-state error when the inputs are other than three basic types i.e. step, ramp & parabolic.
- (iii) Static error constants fail to indicate the exact manner in which the error function changes with time.

(13)

overshoot when the system is subjected to a step-input.

Error Series Or Dynamic Error Co-efficient

or Generalized Error co-efficient

In this, the error constant concept is generalized to include inputs of almost any arbitrary function of time. Error-Series is a more versatile representation of the steady-state-error.

Disadvantages of error constant method:

- (1) When the steady-state error is infinite, which actually, is due to the error increases with time;
- the error-constant method does not indicate how the error varies with time.

(2) The error constant method does not apply to systems with inputs that are sinusoidal, since the final value theorem cannot be applied to this case.

Error transfer function: Since $E(s) = \frac{R(s)}{1 + G(s)H(s)}$ — (1)
 Here the transform error function $E(s)$ which could be of the form (1) or any function that is defined by the system error.

Let us express $E(s)$ as:

$$E(s) = W_e(s) R(s) \quad — (2)$$

where $R(s)$ is the Laplace transform of the input $r(t)$, and $W_e(s)$ is the error transfer function.

By using real convolution integral, the error signal $e(t)$ is

$$e(t) = \int_{-\infty}^t W_e(\tau) r(t-\tau) d\tau \quad — (3)$$

(14)

where $w_e(t)$ is the inverse Laplace transform of $W_e(s)$.
 If the first n derivatives of $\tau(t)$ exist
 for all values of t , the function $\tau(t-\tau)$ can
 be expanded into a Taylor series; that is,

$$\tau(t-\tau) = \tau(t) - \tau^{(1)}(t) + \frac{\tau^2}{2!} \tau^{(2)}(t) - \frac{\tau^3}{3!} \tau^{(3)}(t) + \dots \quad (4)$$

where $\tau^{(i)}(t)$ denotes the i^{th} -order derivative of $\tau(t)$ with respect to time.

Since $\tau(t)$ is considered to be zero
 for negative time, the limits of convolution
 integral in eq (3) may be taken from 0 to t .
 Substituting in eq (3), we have

$$\begin{aligned} e(t) &= \int_0^t w_e(\tau) \left[\tau(t) - \tau^{(1)}(t) + \frac{\tau^2}{2!} \tau^{(2)}(t) - \frac{\tau^3}{3!} \tau^{(3)}(t) + \dots \right] d\tau \\ &= \tau(t) \int_0^t w_e(\tau) d\tau - \tau^{(1)}(t) \int_0^t \tau w_e(\tau) d\tau + \tau^{(2)}(t) \int_0^t \frac{\tau^2}{2!} w_e(\tau) d\tau - \dots \end{aligned} \quad (5)$$

Let $\tau_s(t)$ represent the steady-state part of
 the input $\tau(t)$, and $e_s(t)$ be the steady-state
 part of the error, that is due to $\tau_s(t)$. Then
 the steady-state error function, $e_s(t)$,
 may be written as

$$e_s(t) = \tau_s(t) \int_0^\infty w_e(\tau) d\tau - \tau_s^{(1)}(t) \int_0^\infty \tau w_e(\tau) d\tau + \tau_s^{(2)}(t) \int_0^\infty \frac{\tau^2}{2!} w_e(\tau) d\tau - \dots \quad (6)$$

Let us define: $C_0 = \int_0^\infty w_e(\tau) d\tau$

$$C_1 = - \int_0^\infty \tau w_e(\tau) d\tau$$

$$C_2 = \int_0^\infty \tau^2 w_e(\tau) d\tau$$

$$C_n = (-1)^n \int_0^\infty \tau^n w_e(\tau) d\tau$$

{ } (7)

(15)
eq. (6) can be written as.

$$e_s(t) = C_0 \tau_s(1) + C_1 \tau_s^{(1)}(t) + \frac{C_2 \tau_s^{(2)}(t)}{2!} + \dots + \frac{C_n \tau_s^{(n)}(t)}{n!} \quad (8)$$

which is called the error term, and the coefficients $C_0, C_1, C_2, \dots, C_n$ are called the generalized error coefficients or simply the error coefficients.

The error coefficients may be readily evaluated from the error transfer function $W_e(s)$. Since $W_e(s)$ and $w_e(t)$ are related through the Laplace transform, we have

$$W_e(s) = \int_{-\infty}^{\infty} w_e(t) e^{-ts} dt \quad (9)$$

Taking limit on both sides of eq. (9) as t approaches zero, we have

$$\lim_{s \rightarrow \infty} W_e(s) = \lim_{s \rightarrow \infty} \int_{-\infty}^{\infty} w_e(t) e^{-ts} dt \quad (10)$$

Taking the derivative of $W_e(s)$ of eq. (9) with respect to s , we get

$$\frac{dW_e(s)}{ds} = - \int_{-\infty}^{\infty} t w_e(t) e^{-ts} dt \quad (11)$$

from which the error coefficient C_1 is determined as

$$C_1 = \lim_{s \rightarrow \infty} \frac{dW_e(s)}{ds} \quad (12)$$

and others -

$$C_2 = \lim_{s \rightarrow \infty} \frac{d^2 W_e(s)}{ds^2} \quad (13)$$

$$C_3 = \lim_{s \rightarrow \infty} \frac{d^3 W_e(s)}{ds^3} \quad (14)$$

$$C_n = \lim_{s \rightarrow \infty} \frac{d^n W_e(s)}{ds^n} \quad (15)$$

(16)

Alternatively:

$$\text{Error-transfer function } W_e(s) = \frac{E(s)}{R(s)}$$

$$= \frac{1}{K_1} + \frac{1}{K_2} s + \frac{1}{K_3} s^2 + \frac{1}{K_4} s^3 + \dots$$

Arrange E & R in the ascending power of s
and divide them, the coefficient which are
left are the dynamic error ~~function coefficients~~.

Here $K_1, K_2, K_3, K_4, \dots$ are dynamic error
functions.

Here $K_1, K_2, K_3, K_4, \dots$ are dynamic error
coefficients

K_1 — dynamic position error coefficient

K_2 — " velocity "

K_3 — " acceleration "

K_4 — dynamic error coefficient

Q-1 Find the static and dynamic error-coefficients
with unity feedback control system having
the forward path tr. for of

$$\textcircled{i} \quad G_1(s) = \frac{K}{Ts+1} \quad \textcircled{ii} \quad G_2(s) = \frac{\omega_n^2}{s^2 + 2\zeta\omega_n s}$$

Soln:-

a) static error coefficients

$$(i) K_p = \lim_{s \rightarrow 0} G_1(s) = \lim_{s \rightarrow 0} \frac{K}{Ts+1} = K$$

$$K_v = \lim_{s \rightarrow 0} sG_1(s) = \lim_{s \rightarrow 0} s \cdot \frac{K}{Ts+1} = 0$$

$$K_a = \lim_{s \rightarrow 0} s^2 G_1(s) = \lim_{s \rightarrow 0} s^2 \frac{K}{Ts+1} = 0$$

$$(ii) K_p = \lim_{s \rightarrow 0} G_2(s) = \lim_{s \rightarrow 0} \frac{\omega_n^2}{s(s+2\zeta\omega_n)} = \infty$$

$$K_v = \lim_{s \rightarrow 0} sG_2(s) = \lim_{s \rightarrow 0} \frac{s\omega_n^2}{s(s+2\zeta\omega_n)} = \frac{\omega_n}{2\zeta}$$

$$K_a = \lim_{s \rightarrow 0} s^2 G_2(s) = \lim_{s \rightarrow 0} \frac{s^2 \omega_n^2}{s(s+2\zeta\omega_n)} = 0$$

(17)

(b) Dynamic error coefficient:

$$(i) G_1(s) = \frac{K}{Ts + 1}$$

$$\text{Since } W_e(s) = \frac{E(s)}{R(s)} = \frac{1}{1 + G_1(s)H(s)}$$

$$\text{then } H(s) = 1$$

$$\therefore W_e(s) = \frac{E(s)}{R(s)} = \frac{1}{1 + G_1(s)} = \frac{1}{1 + \frac{K}{Ts + 1}}$$

$$= \frac{1 + Ts}{1 + K + Ts}$$

$$= \frac{1}{1+K} + \frac{TK}{(1+K)^2}s + \dots$$

$$\therefore \frac{1}{K_1} = \frac{1}{1+K} \Rightarrow K_1 = 1 + K_p$$

$$\frac{1}{K_2} = \frac{TK}{(1+K)^2}$$

$$\frac{(1+K) + Ts}{1 + \frac{Ts}{1+K}} / \frac{1 + Ts(\frac{1}{1+K} + \frac{TK}{(1+K)^2}s + \dots)}{1 + \frac{Ts}{1+K}}$$

$$\hookrightarrow \frac{Ts(1 - \frac{1}{1+K})}{\frac{Ts}{1+K}}$$

$$\frac{\frac{Ts}{1+K} + \frac{TKs^2}{(1+K)^2}}{-\frac{1}{1+K} - \frac{TKs^2}{(1+K)^2}}$$

$$-\frac{TKs^2}{1+K^2}$$

$$(ii) W_e(s) = \frac{E(s)}{R(s)} = \frac{1}{1 + G_2(s)} = \frac{1}{1 + \frac{\omega_n^2}{s^2 + 2\xi\omega_n s}}$$

$$= \frac{2\xi\omega_n s + s^2}{\omega_n^2 + 2\xi\omega_n s + s^2}$$

$$= \frac{2\xi}{\omega} s + \frac{1 - 4\xi^2}{\omega_n^2} s^2 + \dots$$

$$\therefore \frac{1}{K_1} = 0$$

$$\Rightarrow K_1 = \infty$$

$$\frac{1}{K_2} = \frac{2\xi}{\omega_n}$$

$$\Rightarrow K_2 = \frac{\omega_n}{2\xi}$$

$$\frac{1}{K_3} = \frac{1 - 4\xi^2}{\omega_n^2}$$

$$\Rightarrow K_3 = \frac{\omega_n^2}{1 - 4\xi^2}$$

(18)

Static error coefficient

Q2 Find the dynamic error - coefficient
of a unity feedback control system
with forward path transfer function

$$G(s) = \frac{10}{s(s+1)} \quad \text{due to an input}$$

Find also the steady state error if
described by $r(t) = a_0 + a_1 t + a_2 t^2$.

Soln Steady state errors

$$e_{ss} = e(t) \underset{t \rightarrow \infty}{=} \frac{a_0}{s} E(s)$$

Step I Evaluate $\frac{E(s)}{R(s)}$ for the given problem.

$$\frac{E(s)}{R(s)} = \frac{1}{1+G(s)} \quad \text{for unity feedback system}$$

$$\text{put } s = \frac{1}{1 + \frac{10}{s(s+1)}} = \frac{s(s+1)}{s(s+1) + 10}$$

Step II Arrange in ascending order of s.

$$\frac{E(s)}{R(s)} = \frac{s+s^2}{10+s+s^2}$$

$$= 0.1s + 0.09s^2 + 0.01s^3 + \dots$$

Compare the coefficient of s in this problem to
the standard form which is given below:

$$\frac{E(s)}{R(s)} = \frac{1}{K_1} + \frac{1}{K_2}s + \frac{1}{K_3}s^2 + \frac{1}{K_4}s^3 + \dots$$

$$\therefore \frac{1}{K_1} = 0 \rightarrow K_1 = \infty$$

$$\frac{1}{K_2} = 0.1 \rightarrow K_2 = 10$$

$$\frac{1}{K_3} = 0.09 \rightarrow K_3 = \cancel{\frac{100}{11.1}} \cancel{11.1}$$

$$\frac{1}{K_4} = 0.01 \rightarrow K_4 = \cancel{\cancel{100}}$$

$$\frac{E(s)}{R(s)} = \frac{1}{K_1} + \frac{1}{K_2}s + \frac{1}{K_3}s^2 + \frac{1}{K_4}s^3 + \dots$$

$$\text{Suppose } E(s) = \frac{1}{K_1} R(s) + \frac{1}{K_2} s R(s) + \frac{1}{K_3} s^2 R(s) + \dots$$

Taking inverse Laplace transform of both side.

$$e(t) = \frac{1}{K_1} r(t) + \frac{1}{K_2} \frac{dr(t)}{dt} + \frac{1}{K_3} \frac{d^2r(t)}{dt^2} + \dots$$

Steady state error in terms of dynamic error
coefficient:

(19)

$$e_{ss} = \lim_{t \rightarrow \infty} e(t) = \lim_{t \rightarrow \infty} \left\{ \frac{1}{K_1} r(t) + \frac{1}{K_2} \frac{d r(t)}{dt} + \frac{1}{K_3} \frac{d^2 r(t)}{dt^2} \right\}$$

$$r(t) = a_0 + a_1 t + a_2 t^2$$

$$\therefore \frac{d r(t)}{dt} = a_1 + 2 a_2 t$$

$$\frac{d^2 r(t)}{dt^2} = 2 a_2$$

$$\text{given } K_1 = 0, \quad K_2 = 0.1, \quad \frac{1}{K_3} = 0.09$$

$$e_{ss} = \lim_{t \rightarrow \infty} e(t) = \lim_{t \rightarrow \infty} \left\{ \frac{1}{K_1} (a_0 + a_1 t + a_2 t^2) + \frac{1}{K_2} (a_1 + 2 a_2 t) \right. \\ \left. + \frac{1}{K_3} 2 a_2 + \dots \right\}$$

$$= \lim_{t \rightarrow \infty} \left\{ 0 + 0.1 (a_1 + 2 a_2 t) + 0.09 (2 a_2) \right\}$$

$$= \infty$$

$$e_{ss} = \lim_{t \rightarrow \infty} e(t) = \infty$$

To error is infinity.
we remove $a_2 t^2$ from $\frac{d}{dt}$

for $e_{ss} \neq \infty$,
general input.

\Rightarrow input can only be of 1L type

$a_0 + a_1 t$ & $a_2 t^2$

— — —

dynamic error constants.

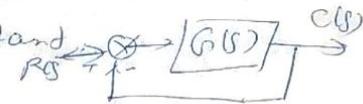
Q. Find the static & dynamic error constants
or Evaluate the steady-state error of a feedback control system by the use of its error-constant
and its error-eliminating method.

$$G(s) = \frac{K}{s+1}$$

Soln:

(20)

Q. For the stable system shown in fig 1, find the position, velocity and error constant when $G(s) = \frac{4(s+2)}{s(s+1)(s+4)}$



Soln: It is a unity feedback system for which,

$$G(s) = \frac{4(s+2)}{s(s+1)(s+4)}$$

$$K_p = \lim_{s \rightarrow \infty} G(s) = \infty$$

$$K_v = \lim_{s \rightarrow 0} sG(s) = 2$$

$$K_a = \lim_{s \rightarrow 0} s^2 G(s) = 0$$

Q. For a unity feedback system, the open loop transfer function is $G(s) = \frac{K(s+s_1)}{s^2(s+s_2)(s+s_3)}$

(a) State the type of the system.

(b) Find K_p , K_v and K_a .

(c) Find the steady state error due to an input which is described by

$$r(t) = (R_1 + R_2 t + R_3 t^2) u_{-1}(t)$$

Soln (a) It is a type 2 system.

$$(b) K_p = \lim_{s \rightarrow \infty} G(s) = \infty; K_v = \lim_{s \rightarrow 0} sG(s) = \infty;$$

$$K_a = \lim_{s \rightarrow 0} s^2 G(s) = \frac{K s_1}{s_2 s_3} = K' \text{ (constant)}$$

(c) The input can be considered as a summation of step, ramp and parabolic inputs. Using superposition theorem,

$$\text{For step input } R_1 u_{-1}(t) \text{, } e_{ss1} = \frac{R_1}{1+K_p} = 0$$

$$\text{For ramp input } R_2 t u_{-1}(t) \text{, } e_{ss2} = \frac{R_2}{K_v} = 0$$

$$\text{For parabolic input } R_3 t^2 u_{-1}(t) \text{, } e_{ss3} = \frac{2 R_3}{K_a} \quad (\text{In } t^2 \Rightarrow \frac{2}{s^3})$$

∴ for the gen. input,

$$e_{ss} = \frac{2 R_3}{K_a} = \frac{2 R_3 s_2 s_3}{s_1 K}$$

Performance Index (P.I.)

Def. A Performance index is a number which indicates the "goodness" of system performance. A control system is considered optimal if the values of the parameters are ~~selected~~ chosen ^{selected} so that the selected performance index is minimum or maximum. The optimal values of the parameters depend directly upon the performance index selected.

Or The performance index is a function whose value indicates how well the actual performance of the system matches with the desired performance.

Parameters : Peak overshoot, settling time,

gain margin, phase margin, steady state error.

Requirements of Performance Index:

(1) The performance specification consists of a single performance index i.e. it must yield a single positive number or zero.

(2) A P.I. must offer selectivity i.e. its power to clearly distinguish between an optimum and non-optimum system.

(3) A P.I. must be a function of the parameters of the system and it must exhibit a minimum or maximum.

Finally, to be practical
(4) A P.I. must be easily computed,
analytically or experimentally.

Various error performance indices are:

(1) Integral Square Error (I.S.E.)

$$I.S.E. = \int_0^{\infty} e^2(t) dt \quad \text{--- (1)}$$

This is the most commonly

let $x(t) \rightarrow$ desired output & $z(t) \rightarrow$ actual output

$$\therefore \text{error } e(t) = x(t) - z(t)$$

unless, $\lim_{t \rightarrow \infty} e(t) = 0$, the performance index approaches infinity.

If $\lim_{t \rightarrow \infty} e(t)$ does not approach zero, we may

$$\text{define } [e(t) = z(\infty) - z(t)] \rightarrow \text{with this}$$

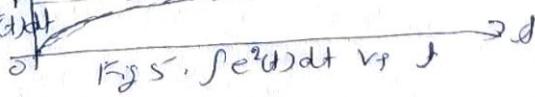
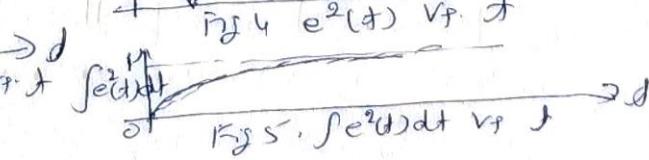
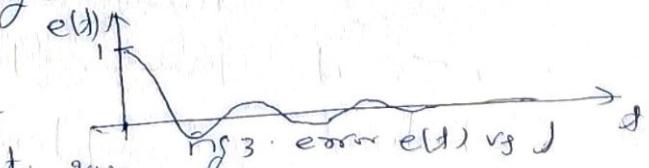
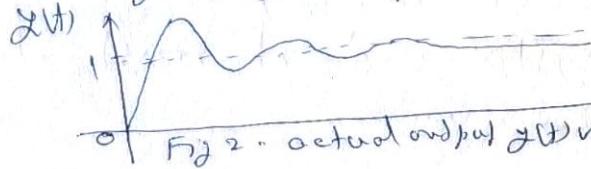
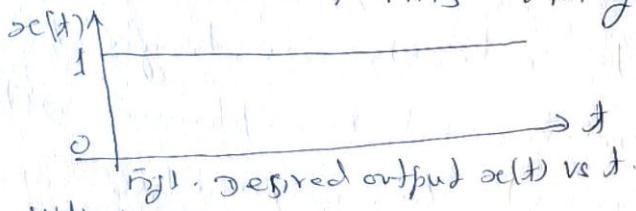
definition of the error, the performance index will yield finite numbers

According to the integral square error (ISE) criterion, the quality of system performance

is evaluated by the following integral $\int_0^{\infty} e^2(t) dt$
where the upper limit ∞ may be replaced

by T which is chosen sufficiently large so that $e(t)$ for $T < t$ is negligible.

The optimal system is the one which minimizes this integral.



- * The characteristics of the P.I. is that it weights large errors heavily and small errors lightly.
- * This criterion is not very selective.
- * A system designed by this criterion tends to show a rapid decrease in a large initial error. Hence the response is fast and oscillatory. Thus the system has poor relative stability.
- * It has practical significance between the minimization of P.I. results in the minimization of power consumption for some systems, such as space craft system.

(2) Integral of time-multipled square error

criterion (ITSE) :-

$$ITSE = \int_0^\infty t e^2(t) dt \quad - (2)$$

The optimal system is one which minimizes the above integral. This criterion has a better selectivity than the integral-square-error criterion.

Initial error is weighted lightly while errors occurring late in the transient response penalized heavily.

(3) Integral-Absolute error criterion (IAE)

The PI defined by IAE is

$$IAE = \int_0^\infty |e(t)| dt \quad - (3)$$

An optimum system based on this criterion is a system which has reasonable damping and a

-4-

satisfactory transient response characteristics.

They are one of most easily applied P.I.

* Minimization of the integral absolute error
is directly related to the minimization
of fuel consumption of Spacecraft system.

* Selectivity is not too good. This P.I. cannot

(*) be easily evaluated by analytical means.
It is practically convenient for analog computer.

(4) Integral - of - time - multiplied absolute error
criterion. (ITAE):

The optimum system is one which
minimizes the following performance index by

$$\int t |e(t)| dt \quad \text{ie. ITAE} = \int_0^\infty t |e(t)| dt \quad (4)$$

A system designed by use of this criterion

has a characteristic that the overshoot in the
transient response is small and oscillations are
well damped. This criterion possesses good
selectivity and is ~~an~~ important over the
integral - absolute - error criterion.

However it is very difficult to evaluate
analytically, although it can be easily
measured experimentally.

(5) integral - of - time - squared absolute error

(6) integral - of - time - cubed absolute error

(7) integral - of - time - multiplied squared absolute error

(8) integral - of - time - multiplied cubed absolute error

-5-

Q1. Consider the system whose closed-loop transfer function is $\frac{C(s)}{R(s)} = \frac{1}{s^2 + 2\xi s + 1}$ ($0 < \xi < 1$)

compute $\int_0^\infty c^2(t) dt$ for a unit-impulse input

soln. For a unit impulse input, $R(s) = 1$, hence

$$C(s) = \frac{1}{s^2 + 2\xi s + 1} = \frac{1}{\sqrt{1-\xi^2}} \cdot \frac{s\sqrt{1-\xi^2}}{(s+\xi)^2 + 1 - \xi^2}$$

Taking inverse Laplace transform.

$$c(t) = \frac{1}{\sqrt{1-\xi^2}} e^{-\xi t} \sin \sqrt{1-\xi^2} t \quad (t \geq 0)$$

$$\begin{aligned} c^2(t) &= \frac{1}{1-\xi^2} e^{-2\xi t} \sin^2 \sqrt{1-\xi^2} t \\ &= \frac{1}{1-\xi^2} e^{-2\xi t} \cdot \frac{1}{2} (1 - \cos 2\sqrt{1-\xi^2} t) \end{aligned}$$

$$\mathcal{L}[c^2(t)] = \frac{1}{2(1-\xi^2)} \left[\frac{1}{s+2\xi} - \frac{s+2\xi}{(s+2\xi)^2 + 4(1-\xi^2)} \right]$$

$$\begin{aligned} \text{Then } \int_0^\infty c^2(t) dt &= \lim_{s \rightarrow 0} \mathcal{L}[c^2(t)] \\ &= \lim_{s \rightarrow 0} \frac{1}{2(1-\xi^2)} \left[\frac{1}{(s+2\xi)} - \frac{(s+2\xi)}{(s+2\xi)^2 + 4(1-\xi^2)} \right] \\ &= \frac{1}{4\xi} \end{aligned}$$

Q2. Compute the following performance index

$$\int_0^\infty |e(t)| dt, \int_0^\infty t |e(t)| dt, \int_0^\infty t [|e(t)|^2 + |e'(t)|^2] dt$$

$$\text{for the system } \frac{C(s)}{R(s)} = \frac{1}{s^2 + 2\xi s + 1} \quad (\xi \geq 1)$$

$$E(s) = \mathcal{L}[e(t)] = R(s) - C(s)$$

Assume that the system is initially at rest and is subjected to a unit step input

Soln. For a unit-step input; $R(s) = \frac{1}{s}$ and

$$\begin{aligned} E(s) &= R(s) - C(s) = \frac{1}{s} - \frac{1}{s^2 + 2\zeta s + 1} \cdot \frac{1}{s} \\ &= \frac{1}{s} \cdot \frac{s^2 + 2\zeta s}{s^2 + 2\zeta s + 1} \\ &= \frac{-\zeta + \sqrt{\zeta^2 - 1}}{2\sqrt{\zeta^2 - 1}} \cdot \frac{1}{s + \zeta + \sqrt{\zeta^2 - 1}} + \frac{\zeta + \sqrt{\zeta^2 - 1}}{2\sqrt{\zeta^2 - 1}} \cdot \frac{1}{s + \zeta - \sqrt{\zeta^2 - 1}} \end{aligned}$$

$$\begin{aligned} (1) \int_0^\infty |e(t)| dt &= \int_0^\infty e(t) dt = \lim_{s \rightarrow 0} E(s) \\ &= \frac{-\zeta + \sqrt{\zeta^2 - 1}}{2\sqrt{\zeta^2 - 1}} \cdot \frac{1}{\zeta + \sqrt{\zeta^2 - 1}} + \frac{\zeta + \sqrt{\zeta^2 - 1}}{2\sqrt{\zeta^2 - 1}} \cdot \frac{1}{\zeta - \sqrt{\zeta^2 - 1}} \\ &= 2\zeta (\zeta \geq 1) \end{aligned}$$

For $\zeta \geq 1$, the system does not overshoot
Hence $|x(t)| = x(t)$ for all $t \geq 0$

$$\begin{aligned} (2) \int_0^\infty t |e(t)| dt &= \int_0^\infty t e(t) dt = \lim_{s \rightarrow 0} \left[-\frac{d}{ds} E(s) \right] \\ &= \lim_{s \rightarrow 0} \left[\frac{-\zeta + \sqrt{\zeta^2 - 1}}{2\sqrt{\zeta^2 - 1}} \cdot \frac{1}{(s + \zeta + \sqrt{\zeta^2 - 1})^2} + \frac{\zeta + \sqrt{\zeta^2 - 1}}{2\sqrt{\zeta^2 - 1}} \cdot \frac{1}{(s + \zeta - \sqrt{\zeta^2 - 1})^2} \right] \\ &= 4\zeta^2 - 1 \quad \text{for } \zeta \geq 1 \end{aligned}$$

Time-Domain Analysis

(1)

Varioues forms of first order transfer functions:

$$(i) G(s) = \frac{K}{Ts+1} \rightarrow \text{Gain-time constant form} - (1)$$

where K is the gain and T is the time constant of the system.

$$(ii) G(s) = \frac{K'}{s+a} \rightarrow \text{Pole-zero form}$$

where K' is the gain of the system.

Eq. (2) can be obtained from eq. (1) of

$$G(s) = \frac{K}{Ts+1} = \frac{\frac{K}{T}}{s + \frac{1}{T}} = \frac{K'}{s+a} \quad - (2)$$

where $K' = \frac{K}{T}$ & $a = \frac{1}{T}$

Q1. Find the open loop gain and the time constant for the following first order system.

$$(i) G(s) = \frac{5}{s+2} \quad (ii) G(s) = \frac{2}{3s+5}$$

↓
Convert it Time Constant
into ~~transfer fn.~~ for $G(s) = \frac{s/4}{\frac{3}{4}s+1}$

$$(i) G(s) = \frac{5/2}{\frac{1}{2}s+1} \quad K = \frac{5}{4} = 0.5$$

so $\text{gain} = 5/2 = 2.5$ & $T = \frac{3}{4} = 0.75$
& $\frac{1}{2} = 0.5$

Q2. Find the impulse response of the 2nd order system

Q2. Find the impulse response of the 2nd order system whose transfer function $G(s) = \frac{9}{s^2+4s+9}$

Soln. Method-1. Comparing $G(s) = \frac{9}{s^2+4s+9}$ with $G(s) = \frac{\omega_n^2}{s^2 + 2\xi\omega_n s + \omega_n^2}$

$$\therefore \omega_n^2 = 9 \Rightarrow \omega_n = 3 \quad 2\xi\omega_n = 4 \Rightarrow \xi = \frac{4}{2\omega_n} = \frac{4}{2 \times 3} = \frac{2}{3} = 0.667$$

Impulse response for $\xi < 1$ is given by

For step input

$$y(t) = 1 - e^{-\xi\omega_n t} \cdot \frac{\sin(\omega_n\sqrt{1-\xi^2}t)}{\sin(\omega_n t + \alpha)}$$

$$C(t) = \frac{\omega_n}{\sqrt{1-\xi^2}} e^{-\xi\omega_n t} \sin(\omega_n\sqrt{1-\xi^2}t)$$

$$= \frac{3}{\sqrt{1-(0.667)^2}} e^{-\frac{2}{3} \cdot 3t} \cdot \sin(3\sqrt{1-(0.667)^2}t)$$

$$= 4.025 e^{-2t} \sin 2.235t$$

Method 2

for $s^2 + 4s + 9$ roots are $s_1, s_2 = \frac{-4 \pm \sqrt{4^2 - 4 \times 9}}{2}$

$$= \frac{-4 \pm \sqrt{16 - 36}}{2} = \frac{-4 \pm \sqrt{-20}}{2} = \frac{-4 \pm j\sqrt{20}}{2} = \frac{-4 \pm j2\sqrt{5}}{2}$$

$$= -2 \pm j2.236$$

$$G(s) = \frac{9}{[s - (-2 + j2.236)][s - (-2 - j2.236)]}$$

Since $R(s) = 1$ for impulse fn

$$\therefore C(s) = G(s)$$

$$\therefore C(t) = \mathcal{L}[G(s)]$$

$$C(s) = \frac{A}{s - (-2 + j2.236)} + \frac{A^*}{s - (-2 - j2.236)}$$

$$= \frac{-j2.0125^-}{s - (-2 + j2.236)} + \frac{j2.0125^+}{s - (-2 - j2.236)}$$

$$C(t) = -j2.0125^- e^{(-2+j2.236)t} + j2.0125^+ e^{(-2-j2.236)t}$$

$$= 4.025 e^{-2t} \cdot \sin 2.236 t$$

Time domain specifications

$$\textcircled{1} \quad T_r = \frac{\pi - \tan^{-1} \frac{\sqrt{1-\xi^2}}{\xi}}{\omega_n \sqrt{1-\xi^2}} = \frac{\pi - \phi}{\omega_d} \quad \text{for } 0 < \xi < 1$$

$$\textcircled{2} \quad T_p = \frac{\pi}{\omega_d} = \frac{\pi}{\omega_n \sqrt{1-\xi^2}}$$

$$(3) \quad M_p = e^{-\frac{\xi \pi}{\sqrt{1-\xi^2}}} \quad \% M_p = e^{\frac{-\xi \pi}{\sqrt{1-\xi^2}}} \times 100$$

$$\textcircled{4} \quad \text{Settling time} = T_s = -\ln \frac{[0.02] \sqrt{1-\xi^2}}{\xi \omega_n}$$

$$\boxed{T_s = \frac{4}{\xi \omega_n}} \rightarrow \text{optimal for } 2\% \text{ criterion}$$

$$\boxed{T_s = \frac{3}{\xi \omega_n}} \rightarrow \text{optimal for } 5\% \text{ criterion}$$

$$\textcircled{5} \quad T_s = \frac{4}{\xi \omega_n} = 4T \quad \& \quad \text{time constant} = \frac{1}{\xi \omega_n}$$

$\frac{4}{\xi \omega_n} = \frac{1}{\omega_n}$ for ramp input

$$T_s = \frac{3}{\xi \omega_n} = 3T \quad \& \quad \text{tolerance band}$$

$(5) e_{ss} = \lim_{t \rightarrow \infty} [1 - C(t)] = 0$ for unit step input

Q2 When a second order control system is subjected to a unit step input, the values of $\xi = 0.5$ & $\omega_n = 6 \text{ rad/sec}$
 Determine the rise time, peak time, settling time and peak overshoot

Ans. given $\xi = 0.5$ & $\omega_n = 6 \text{ rad/sec}$

$$(i) t_r = \frac{\pi - \tan^{-1} \frac{\sqrt{1-\xi^2}}{\xi}}{\omega_n \sqrt{1-\xi^2}} = \frac{\pi - \tan^{-1} \frac{\sqrt{1-(0.5)^2}}{0.5}}{6 \sqrt{1-(0.5)^2}} = \frac{3.14 - 1.047}{5.19} = 0.403 \text{ sec}$$

$$(ii) t_p = \frac{\pi}{\omega_n \sqrt{1-\xi^2}} = \frac{\pi}{6 \sqrt{1-(0.5)^2}} = 0.605 \text{ sec}$$

$$(iii) t_s = \frac{4}{\xi \omega_n} = \frac{4}{0.5 \times 6} = 1.33 \text{ sec}$$

$$(iv) M_p = e^{\frac{-\pi \xi}{\sqrt{1-\xi^2}} \times 100} = e^{-\frac{\pi \times 0.5}{\sqrt{1-(0.5)^2}} \times 100} = 0.163 \times 100 = 16.3\%$$

Q3. Obtain the unit step and unit impulse response of the following.

$$\frac{C(s)}{R(s)} = \frac{10}{s^2 + 2s + 10}$$

$$\begin{aligned} \text{Ans. } \frac{C(s)}{R(s)} &= \frac{10}{s^2 + 2s + 10}; \text{ for unit step } R(s) = \frac{1}{s} \\ \therefore C(s) &= \frac{10}{s(s^2 + 2s + 10)} = \frac{10}{s(s+1+j3)(s+1-j3)} \\ \therefore C(t) &= L \left[\frac{1}{s} + \frac{-0.5 - j0.166}{s+1+j3} + \frac{-0.5 + j0.166}{s+1-j3} \right] \\ &= 1 + (-0.5 - j0.166) e^{(-1+j3)t} + (-0.5 + j0.166) e^{(-1-j3)t} \\ &= 1 - \bar{e}^t [0.5(e^{j3t} + e^{-j3t}) + j0.166(e^{-j3t} - e^{j3t})] \\ &= 1 - \bar{e}^t [0.5 \cos 3t + j0.33 \sin 3t] \\ &= 1 - 1.053 \bar{e}^t \sin(3t + 71.736^\circ) \end{aligned}$$

Q4. The closed loop tr. fr. is given by $G(s) = \frac{s(s^2 + 9s + 19)}{s^3 + 7s^2 + 14s + 8}$
 Determine the response of the system when a unit step is applied as input

$$\text{Ans. } C(s) = G(s) \cdot R(s) = \frac{s(s^2 + 9s + 19)}{(s^3 + 7s^2 + 14s + 8)} \cdot \frac{1}{s} = \frac{s^2 + 9s + 19}{s^3 + 7s^2 + 14s + 8}$$

$$C(s) = \frac{11}{3(s+1)} - \frac{5}{2(s+2)} - \frac{1}{6}(s+4)$$

$$\therefore C(t) = \frac{11}{3} \bar{e}^t - \frac{5}{2} \bar{e}^{-2t} - \frac{1}{6} \bar{e}^{-4t}$$

-4-

Q.1. A control system having $\omega_n = 5 \text{ rad/sec}$ and $\xi = 0.6$. Determine rise time, peak time, maximum overshoot and settling time when the system is subjected to unit step input and settling time when the system is subjected to unit step input.

Given undamped natural angular freq. $= \omega_n = 5$, damping ratio $\xi = 0.6$

$$(i) \text{ Damped natural angular velocity} = \omega_d = \omega_n \sqrt{1-\xi^2} = 5 \sqrt{1-0.36} = 4$$

$$(ii) \text{ Real part of the roots of the char. eq.} = \sigma = -\xi \omega_n = -5 \times 0.6 = -3$$

$$(iii) \text{ Rise time} = t_r = \frac{\pi - \phi}{\omega_d} = \frac{\pi - \tan^{-1} \frac{\sqrt{1-\xi^2}}{\xi}}{\omega_n \sqrt{1-\xi^2}} = \frac{3.14 - 0.93}{4} = 0.55 \text{ sec}$$

$$(iv) \phi = \tan^{-1} \frac{\sqrt{1-\xi^2}}{\xi} = \tan^{-1} \frac{\sqrt{1-(0.6)^2}}{0.6} = 0.93$$

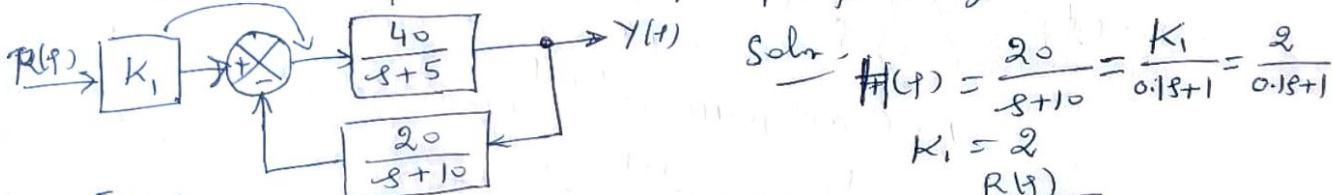
$$(v) \text{ Peak time} = t_p = \frac{\pi}{\omega_d} = \tan^{-1} \frac{3.14}{4} = 0.785$$

$$(vi) \text{ Maximum overshoot, } M_p = e^{\frac{-\xi \pi}{\sqrt{1-\xi^2}}} = e^{\frac{-0.6 \pi}{\sqrt{1-(0.6)^2}}} = 0.095 \\ = 9.5\%$$

$$(vii) \text{ Settling time for } 2\% \text{ tolerance band} = 4T = \frac{4}{\sigma} = \frac{4}{-3} = 1.33 \text{ sec}$$

$$\text{ " " " } + 5T = 5 \times 1.33 = 6.65 \text{ sec}$$

Q.2. Determine the approximate value of K_1 and calculate the steady state error for a unit step input for the system shown in Fig. 1.



$$\text{Schr. } H(s) = \frac{20}{s+10} = \frac{K_1}{0.1s+1} = \frac{2}{0.1s+1}$$

$$K_1 = 2$$

$$E(s) = \frac{R(s)}{1+K_1 G(s) H(s)}$$

Fig. 1

$$\frac{Y(s)}{R(s)} = \frac{K_1 G(s)}{1+K_1 G(s) H(s)} = 8$$

$$\text{Steady state error} = e_{ss} = \lim_{s \rightarrow 0} s E(s) = \lim_{s \rightarrow 0} s \cdot \frac{K_1 G(s)}{1+K_1 G(s) H(s)} = \frac{K_1 G(0)}{1+K_1 G(0) H(0)} = \frac{K_1}{1+K_1 H(0)}$$

$$= \lim_{s \rightarrow 0} \frac{1}{1+K_1 \frac{40}{s+5} \cdot \frac{20}{s+10}} = \frac{1}{1+2 \cdot \frac{40}{5} \cdot \frac{20}{10}} = \frac{1}{1+2 \cdot 8} = \frac{1}{17} = 0.03$$

Time-Domain Analysis -5-

$$\textcircled{1} \quad t_r = \frac{\pi - \tan^{-1} \frac{\sqrt{1-\xi^2}}{\xi}}{\omega_n \sqrt{1-\xi^2}}$$

$$\textcircled{2} \quad t_p = \frac{\pi}{\omega_n \sqrt{1-\xi^2}} - \frac{\pi \xi}{\sqrt{1-\xi^2}}$$

$$\textcircled{3} \quad \% M_p = e^{-\frac{\pi \xi}{\sqrt{1-\xi^2}}} \times 100$$

$$\textcircled{4} \quad t_s = \frac{\pi}{\xi \omega_n}$$

$$\textcircled{5} \quad e_{ss} = \frac{2\xi}{\omega_n} \quad \begin{matrix} \text{for ramp input} \\ \leq 0 \quad \text{for step input} \end{matrix} \quad = \frac{1}{K_{ou}}$$

Q4 The open loop transfer function of a servo system with unity feedback is given by

$$G(s) = \frac{10}{(s+2)(s+5)}$$

Determine the damping ratio, undamped natural frequency of oscillation. What is the percentage overshoot & a unit step input.

Sol: $G(s) = \frac{10}{(s+2)(s+5)}, H(s)=1$

$$\text{Ch. eq. } 1 + G(s)H(s) = 1 + \frac{10}{(s+2)(s+5)} = 0$$

$$\text{or } s^2 + 7s + 20 = 0$$

$$\text{Comparing with } s^2 + 2\xi \omega_n s + \omega_n^2 = 0$$

$$\omega_n^2 = 20 \rightarrow \omega_n = 4.472 \text{ rad/sec}$$

$$2\xi \omega_n = 7 \Rightarrow \xi = 0.7826$$

$$\text{or } 2 \times \xi \times 4.472 = 7$$

$$M_p = e^{-\frac{\pi \xi}{\sqrt{1-\xi^2}}} \times 100 \Rightarrow M_p = 1.92 \%$$

—6—

Q. A unity feedback system is characterised by an open loop tr. fn. $G(s) = \frac{K}{s(s+10)}$. Determine the gain K so that the system will have a damping ratio of 0.5. For this value of K, determine settling time, peak overshoot and peak time for a unit step input. Soln. Given $G(s) = \frac{K}{s(s+10)}$, $H(s) = 1$ & $\frac{C(s)}{R(s)} = \frac{\frac{K}{s(s+10)}}{1 + \frac{K}{s(s+10)}} = \frac{K}{s^2 + 10s + K}$

(i) Comparing with the std. eqn $\frac{\omega_n^2}{s^2 + 2\zeta\omega_n s + \omega_n^2} = \frac{K}{s^2 + 10s + K}$

$$\omega_n^2 = K \quad \& \quad 2\zeta\omega_n = 10$$

$$\text{Given } \zeta = 0.5, \quad \text{or } 2 \times 0.5 \omega_n = 10 \Rightarrow \omega_n = \frac{10}{1.5} = 6.67$$

$$\therefore K = (\omega_n)^2 = (10)^2 = 100$$

$$(ii) t_s = \frac{\pi}{\zeta\omega_n} = \frac{\pi}{(0.5)(10)} = \frac{\pi}{5} = 0.8 \text{ sec}$$

$$(iii) M_p = 100 - \frac{\zeta\pi}{\sqrt{1-\zeta^2}} = 100 e^{\frac{-0.5\pi}{\sqrt{1-(0.5)^2}}} = 16.3\%$$

$$(iv) t_p = \frac{\pi}{\omega_n} = \frac{\pi}{\omega_n \sqrt{1-\zeta^2}} = \frac{\pi}{10 \sqrt{1-(0.5)^2}} = \frac{\pi}{10(0.866)} \\ = \frac{3.14}{8.66} = 0.362 \text{ sec}$$

Q. For a closed loop tr. fn given by $\frac{C(s)}{R(s)} = \frac{\omega_n^2}{s^2 + 2\zeta\omega_n s + \omega_n^2}$ determine the value of ζ and ω_n so that the system responds to a step input with approximately 5% overshoot and with a settling time of 2 sec.

Ans t_s (for 2% crit. damped) = $\frac{\pi}{\zeta\omega_n} = 2 \text{ sec.} \Rightarrow \frac{\zeta}{\omega_n} = 2$.

~~for 5% crit. damped~~ Peak overshoot = $M_p = 5\%$.
 $\Rightarrow e^{-\frac{\zeta\pi}{\sqrt{1-\zeta^2}}} = 5\%$. $\left\{ \begin{array}{l} \text{using natural log on both sides} \\ \frac{\zeta\pi}{\sqrt{1-\zeta^2}} = 3 \end{array} \right.$
 or $e^{-\frac{\zeta\pi}{\sqrt{1-\zeta^2}}} = 0.05$ $\left\{ \begin{array}{l} \text{using natural log on both sides} \\ \frac{\zeta\pi}{\sqrt{1-\zeta^2}} = 3 \end{array} \right. \Rightarrow \zeta = 0.69$

$$\zeta\omega_n = 2 \Rightarrow \omega_n = \frac{2}{\zeta} = \frac{2}{0.69} = 2.896 \text{ rad/sec}$$

$$\frac{C(s)}{R(s)} = \frac{\omega_n^2}{s^2 + 2\zeta\omega_n s + \omega_n^2} = \frac{(2.896)^2}{s^2 + 2(2)s + (2.896)^2} = \frac{8.396}{s^2 + 4s + 8.396}$$

-7-

- Q. For the ~~given~~ system described by a differential equation $\frac{d^2y}{dt^2} + 4 \frac{dy}{dt} + 8y = 8x$ where x is the input & y is the output

Determine the undamped natural freq., damping ratio, damped natural freq., peak time and settling time.

Ans Overall tr-fn. = $\frac{Y(s)}{X(s)} = \frac{8}{s^2 + 4s + 8}$

Comparing $\frac{8}{s^2 + 4s + 8}$ with $\frac{\omega_n^2}{s^2 + 2\zeta\omega_n s + \omega_n^2}$

$\omega_n^2 = 8$ & $2\zeta\omega_n = 4$

undamped natural freq. = $\omega_n = \sqrt{8}$ rad/sec

Damping ratio = $\zeta = \frac{4}{2\omega_n} = \frac{4}{2\sqrt{8}} = \frac{1}{\sqrt{2}} = 0.707$

Damped natural freq. = $\omega_d = \omega_n \sqrt{1 - \zeta^2}$

$$= \sqrt{8} \sqrt{1 - (\frac{1}{\sqrt{2}})^2} = \sqrt{8} \sqrt{0.5} = 2 \text{ rad/sec}$$

Peak overshoot = $M_p = e^{-\frac{\pi\zeta}{\sqrt{1-\zeta^2}}} \times 100$
 $= e^{-\pi \times (0.707) / \sqrt{1 - (0.707)^2}} \times 100 = 4.33\%$

Settling time = $t_s = \frac{4}{\zeta\omega_n} = \frac{4}{(0.707)\sqrt{8}} = 2.2 \text{ sec}$

- Q. A unity feedback system is characterized by the closed loop transfer function given by

$$G(s) = \frac{s(s^2 + 9s + 19)}{s^3 + 7s^2 + 14s + 8}$$

Determine the response of the system when a unit step is applied as input

Ans: $Y(s) = G(s)R(s) = \frac{1}{s} \cdot \frac{s(s^2 + 9s + 19)}{s^3 + 7s^2 + 14s + 8} = \frac{1}{s} \cdot \frac{11}{3(s+1)} + \frac{5}{2} \frac{1}{s+2} - \frac{1}{6(s+4)}$

$$\therefore y(t) = \frac{11}{3} e^{-t} - \frac{5}{2} e^{-2t} - \frac{1}{6} e^{-4t} \quad \text{Ans}$$

-8-

Q Obtain the unit step response and unit impulse response of the following system.

$$\frac{Y(s)}{R(s)} = \frac{10}{s^2 + 2s + 10}$$

Soln: $\frac{Y(s)}{R(s)} = \frac{10}{s^2 + 2s + 10}$; For unit step input $R(s) = \frac{1}{s}$

$$\Rightarrow Y(s) = \frac{10}{s(s^2 + 2s + 10)} = \frac{A}{s} + \frac{B}{s+0.5+j0.166} + \frac{B^*}{s+0.5-j0.166}$$

$$Y(t) = \mathcal{L}^{-1} \left[\frac{1}{s} + \frac{-0.5-j0.166}{s+1+j3} + \frac{-0.5+j0.166}{s+1-3j} \right] \quad \begin{array}{l} A = -0.5-j0.166 \\ B = -0.5-j0.166 \\ B^* = -0.5+j0.166 \end{array}$$

$$= 1 + (-0.5-j0.166)e^{(-1-3j)t} + (-0.5-j0.166)e^{(1+j3)t}$$

$$= 1 - e^{-t} \left[(0.5+j0.166)e^{-j3t} + (0.5-j0.166)e^{j3t} \right]$$

$$= 1 - e^{-t} [0.5(e^{j3t} + e^{-j3t}) + j0.166(e^{-j3t} - e^{j3t})]$$

$$= 1 - e^{-t} [\cos 3t + j0.33 \sin 3t]$$

$$= 1 - e^{-t} \left(\sqrt{1^2 + 0.33^2} \right) \left(\frac{1}{\sqrt{1^2 + 0.33^2}} \cos 3t + j \frac{0.33}{\sqrt{1^2 + 0.33^2}} \sin 3t \right)$$

$$= 1 - 1.053 e^{-t} [0.9496 \cos 3t + 0.3134 \sin 3t]$$

$$= 1 - 1.053 e^{-t} [\sin 71.74^\circ \cos 3t + \cos 71.74^\circ \sin 3t]$$

$$y(t) = 1 - 1.053 e^{-t} \sin (3t + 71.736^\circ)$$

Alternative method. on comparing with standard 2nd order system

$$\frac{10}{s^2 + 2s + 10} = \frac{\omega_n^2}{s^2 + 2\xi\omega_n s + \omega_n^2} \Rightarrow \omega_n = \sqrt{10} \text{ and } \xi = \frac{1}{\sqrt{10}}$$

$$y(t) = 1 - \frac{e^{-\xi\omega_n t}}{\sqrt{1-\xi^2}} \sin (\omega_n t + \phi) = 1 - \frac{e^{-t}}{\sqrt{1-\frac{1}{10}}} \sin \left(\sqrt{10} \sqrt{1-\frac{1}{10}} t + \phi \right)$$

$$\phi = \tan^{-1} \frac{\frac{1}{\sqrt{10}}}{\frac{1}{\sqrt{10}}} = 71.736^\circ$$

$$= 1 - 1.053 e^{-t} \sin (3t + 71.736^\circ)$$

For unit impulse:

$$Y(s) = \frac{10}{s^2 + 2s + 10} = \frac{10}{(s+1)^2 + 3^2}$$

$$\text{For } \mathcal{L}^{-1} \left[\frac{b}{(s+a)^2 + b^2} \right] = e^{-at} \sin bt$$

$$y(t) = 3.33 e^{-t} \sin 3t$$

