

In [2]:

```
from math import *
import numpy as np
import matplotlib.pyplot as plt

from emukit.core import ContinuousParameter, ParameterSpace
from emukit.core.initial_designs import RandomDesign
from emukit.multi_fidelity.convert_lists_to_array import convert_x_list_to_array, convert_xy_lists_to_arrays
from emukit.model_wrappers.gpy_model_wrappers import GPyMultiOutputWrapper
from emukit.multi_fidelity.kernels import LinearMultiFidelityKernel
from emukit.multi_fidelity.models import GPyLinearMultiFidelityModel

from emukit.multi_fidelity.models.non_linear_multi_fidelity_model import make_non_linear_kernels, NonLinearMultiFidelityModel

import GPy
from GPy.models import GPRegression
from emukit.model_wrappers import GPyModelWrapper

import matplotlib.pyplot as plt
from matplotlib import colors
import mlai.plot as plot
```

In [3]:

```
# Simulation presettings
write_images = False
write_output_txt = False
%run Missile_utils.ipynb
simulation_output = 'range'
```

In [4]:

```
custom_param_names = ['fuelmass', 'drymass']
```



In [5]:

```

# Low-fidelity simulator class
class simple_simulator_mod:
    def __init__(self, fuelmass, drymass, payload, missilediam, params=True):
        self.dMdt = 500
        self.v_ex = 5000

        if params is None:
            self.drymass = 0.8e6
            self.fuelmass = 2e6
            self.diam = 1.65

        else:
            self.drymass = drymass + payload
            self.fuelmass = fuelmass
            self.diam = missilediam

    def get_range(self):
        h=1e-5
        v=1e-5
        gamma = self.to_radians(90)
        thrust = self.v_ex * self.dMdt
        mass = self.drymass + self.fuelmass
        area = pi*(self.diam**2)/4
        G = 6.6743e-11
        M = 5.972e24
        R = 6.371e6
        g = G*M/(R**2)
        dm = self.dMdt
        t=0
        psi=0
        dt=1e-1
        tEND=1e8
        heights = []
        velocity = []
        gammas = []
        cnt=0
        while(h>0):
            if t>tEND:
                break
            try:
                dv = thrust/mass - g*sin(gamma)
            except:
                dv = 0
            dpsi = v*cos(gamma)/(R+h)
            dh = v*sin(gamma)
            if t<5:
                dgamma = -0.1
            else:
                dgamma = dpsi - g*cos(gamma)/v
            #
            print(dgamma)

            if mass <= self.drymass:
                thrust=0
                mass = self.drymass
                dm=0
                if cnt==0:
                    print("THRUST HAS BEEN PUT TO ZERO")
                    cnt+=1
            #
            #
            #
            else:
                mass = mass - dm*dt

            v = v + dt*dv
            #
            print(dh)
            h = h + dt*dh
            heights.append(h)

            gamma = gamma + dt*dgamma
            gammas.append(gamma)
            psi = psi + dt*dpsi
            g = G*M/((R+h)**2)
            t= t+dt
            velocity.append(v)

            #
            plt.plot(velocity)
            #
            plt.show()
            #plt.plot(gammas)
            #plt.show()
            if psi<0:
                print("Rocket Failed (PSI)")
                return 0
            if t>tEND:
                print("Rocket Failed (tend)")
                return 0
            return psi*R/1000

    def eta(self, h, t):
        eta = 0.0
        return eta

```

```

def density(self,h):
    "Calculates air density at altitude"
    rho0 = 1.225 #[kg/m^3] air density at sea level
    if h < 19200:
        rho = rho0 * exp(-h/8420)
    elif h > 19200 and h < 47000:
        rho = rho0 * (.857003 + h/57947)**-13.201
    else:
        #vacuum
        rho = 0.0
    return rho

def temperature(self,h):
    "Calculates air temperature [Celsius] at altitude [m]"
    if h <= 11000:
        #troposphere
        t = 15.04 - .00649*h
    elif h <= 25000:
        #Lower stratosphere
        t = -56.46
    elif h > 25000:
        t = -131.21 + .00299*h
    return t

def pressure(self,h):
    "Calculates air pressure [Pa] at altitude [m]"
    t = self.temperature(h)
    if h <= 11000:
        #troposphere
        p = 101.29 * ((t+273.1)/288.08)**5.256
    elif h <= 25000:
        #Lower stratosphere
        p = 22.65*exp(1.73-.000157*h)
    elif h > 25000:
        p = 2.488 * ((t+273.1)/288.08)**-11.388
    return p

def Cdrag (self,v,h):
    t = self.temperature(h) + 273.15 #convert to kelvin
    a = sqrt(1.4*287*t)
    mach = v/a

    if mach > 5:
        cd = 0.15
    elif mach > 1.8 and mach <= 5:
        cd = -0.03125*mach + 0.30625
    elif mach > 1.2 and mach <= 1.8:
        cd = -0.25*mach + 0.7
    elif mach > 0.8 and mach <= 1.2:
        cd = 0.625*mach - 0.35
    elif mach <= 0.8:
        cd = 0.15
    return cd

def to_radians(self,degree):
    return degree * pi/180

```

In [6]:

```

# Low-fidelity simulation helper
def run_missile_low_sim(custom_params):
    """
    Recives in input an array of custom parameters.
    Each row represents a set of different parameters
    Each column is a different parameter (#cols = len(custom_param_names))
    """
    default_params_IRAQ = {
        'payload':500,
        'missilediam':0.88,
        'fuelmass':5600,
        'drymass':1200,
        'Isp0':226,
        'thrust0':9177.4
    }
    y = np.zeros((custom_params.shape[0], 1))
    for i in range(custom_params.shape[0]):
        # Row: different parameters -> different runs from the simulator
        params_to_use = default_params_IRAQ
        # Overwrite default param variables
        for j in range(custom_params.shape[1]):
            # For each custom variable
            param_name = custom_param_names[j]
            params_to_use[param_name] = custom_params[i,j]

            ## TEMP ## Better customise this
            #if j==0:
            #    print('\nNew simulation \n')
            str_to_print = param_name + ': ' + str(custom_params[i,j])
            #print(str_to_print)
            ##

        # Run simulation
        output_path = 'results/results_' + str(i) + '.txt' # TODO Define better identifier
        sim_obj = simple_simulator_mod(
            fuelmass=params_to_use["fuelmass"],
            drymass=params_to_use["drymass"],
            payload=params_to_use["payload"],
            missilediam=params_to_use["missilediam"],
        )
        Tm = 1.4e4*2000 / (params_to_use["fuelmass"]+params_to_use["drymass"])
        g = 6.6743e-11*5.972e24/(6.371e6**2)
        if Tm < g:
            print("Tm<g -> will fail")
            #print(f'Fuel {params_to_use["fuelmass"]}, dry {params_to_use["drymass"]}')

        y[i, 0] = sim_obj.get_range()
        if y[i, 0] ==0:
            print("FAILED")
    return y

```

In [7]:

```

# High-fidelity simulator helper
def run_missile_high_sim(custom_params):
    """
    Recives in input an array of custom parameters.
    Each row represents a set of different parameters
    Each column is a different parameter (#cols = len(custom_param_names))
    """
    default_params_IRAQ = {
        'payload':500,
        'missilediam':0.88,
        'rvdiam':0,
        'estrange':600,
        'numstages':1,
        'fuelmass':[0,5600],
        'drymass':[0,1200],
        'Isp0':[0,226],
        'thrust0':[0,9177.4]
    }
    y = np.zeros((custom_params.shape[0], 1))
    for i in range(custom_params.shape[0]):
        params_to_use = default_params_IRAQ
        # Overwrite default param variables
        for j in range(custom_params.shape[1]):
            param_name = custom_param_names[j]
            if param_name in ['fuelmass', 'drymass', 'Isp0', 'thrust0']:
                params_to_use[param_name][1] = custom_params[i,j] # OK as long as we are considering missiles with only 1 stage
            else:
                params_to_use[param_name] = custom_params[i, j]

        ## TEMP ## Better customise this
        if j==0:
            print('\nNew simulation \n')
            str_to_print = param_name + ': ' + str(custom_params[i,j])
            print(str_to_print)
        ##

    # Run simulation
    output_path = 'results/results_' + str(i) + '.txt' # TODO Define better identifier
    sim_output = run_one_sim(
        numstages=params_to_use["numstages"],
        fuelmass=params_to_use["fuelmass"],
        drymass=params_to_use["drymass"],
        thrust0=params_to_use["thrust0"],
        Isp0=params_to_use["Isp0"],
        payload=params_to_use["payload"],
        missilediam=params_to_use["missilediam"],
        rvdiam=params_to_use["rvdiam"],
        est_range=params_to_use["estrange"],
        output_path=output_path,
        simulation_output=simulation_output,
    )

    y[i, 0] = sim_output
    return y

```

In [ ]:

## Actual code starting

In [8]:

```

basic_param_spaces = {
    'payload': [10, 2410],
    'missilediam': [0.1, 9.9],
    'rvdiam': [0.1, 9.9],
    'estrange': [100, 4900],
    'fuelmass': [500, 6000], # [500, 7000],
    'drymass': [1000, 3000],
    'Isp0': [100, 800], # [100, 800],
    'thrust0': [10000, 69000],
}

```

In [9]:

```

high_f = run_missile_high_sim
low_f = run_missile_low_sim

num_points = 101

x_dry = np.linspace(1000, 3000, num_points)[: , None]
x_fuel = np.linspace(500, 6000, num_points)[: , None]
x_fmsh, x_dmesh = np.meshgrid(x_fuel, x_dry)
x_plot = np.array([x_fmsh, x_dmesh]).T.reshape(-1,2)

y_plot_low = low_f(x_plot)
print(y_plot_low)

```

```

[[ 212.38444371]
 [ 207.42512151]
 [ 202.57373858]
 ...
 [2412.30249339]
 [2393.15084235]
 [2374.20058179]]

```

In [10]:

```
y_plot_high = high_f(x_plot)
```

```

Angle (deg h): 43.732284489193205
Range (km): 1.0582588623195328
Time (sec): 12.299999999999974
Final results:
Range (km): 26.61990796407633
Apogee (km): 8.112255006544737
Time to target (sec): 86.79999999999934

```

New simulation

```

fuelmass: 500.0
drymass: 1020.0

```

Stage 1 burnout

```

Velocity (km/s): 0.4701545526353664
Angle (deg h): 43.732284489193205
Range (km): 1.0459822915660328

```

In [49]:

```

np.random.seed(1234)
x_train_fuel = np.random.uniform(500, 6000, 12)[: , np.newaxis]
x_train_dry = np.random.uniform(1000, 3000, 12)[: , np.newaxis]
x_train_low = np.column_stack((x_train_fuel, x_train_dry))
x_train_high = np.random.permutation(x_train_low)[:6]
print(x_train_high)

y_train_low = low_f(x_train_low)
y_train_high = high_f(x_train_high)

```

```

[[5769.76644526 1729.7719678 ]
 [1553.35697708 2366.92587034]
 [1999.25932905 1027.53689918]
 [4910.29697644 2765.28238127]
 [2907.50256454 1740.50150958]
 [4789.86694465 2006.16633062]]

```

New simulation

```

fuelmass: 5769.7664452603785
drymass: 1729.7719678027447

```

Stage 1 burnout

```

Velocity (km/s): 1.576964257173826
Angle (deg h): 43.655034996438914
Range (km): 35.66426739809259
Time (sec): 142.099999999999621
Final results:

```

In [50]:

```
y_plot_high.shape
```

Out[50]:

```
(10201, 1)
```

## Plotting simulator-results and test points

In [51]:

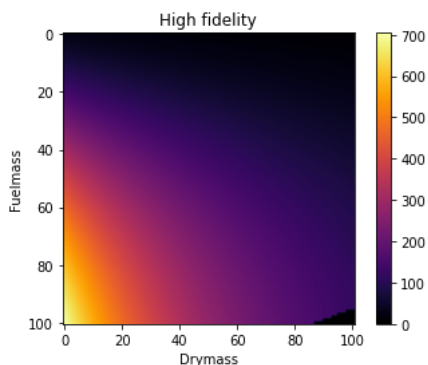
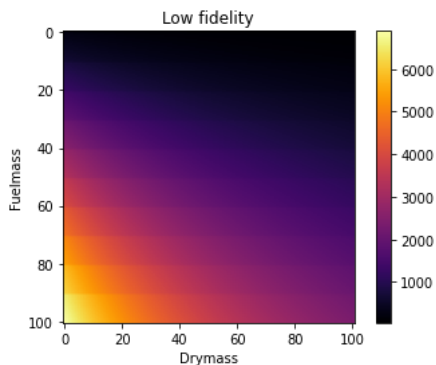
```
y_plot_low = y_plot_low.reshape(num_points**2, 1)
y_plot_high = y_plot_high.reshape(num_points**2, 1)
print(y_plot_low.shape)

# Low-fidelity plot original
y_plot_low_im = y_plot_low.reshape(num_points, num_points)
fig, ax = plt.subplots()
ax.set_xlabel('Drymass')
ax.set_ylabel('Fuelmass')
ax.set_title('Low fidelity')
im_low = ax.imshow(y_plot_low_im, cmap='inferno')
fig.colorbar(im_low, ax=ax)

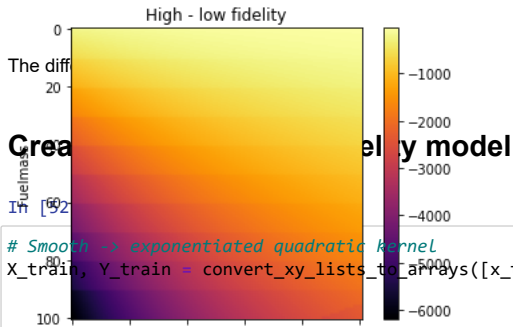
# High-fidelity plot original
y_plot_high_im = y_plot_high.reshape(num_points, num_points)
fig, ax = plt.subplots()
ax.set_xlabel('Drymass')
ax.set_ylabel('Fuelmass')
ax.set_title('High fidelity')
im_high = ax.imshow(y_plot_high_im, cmap='inferno')
x_th_scaled = np.array([(x_train_fuel-500)/(5500)*101, (x_train_dry-1000)/2000*100]).reshape(12,2)
fig.colorbar(im_high, ax=ax)
plt.show()

# Difference
fig, ax = plt.subplots()
ax.set_xlabel('Drymass')
ax.set_ylabel('Fuelmass')
ax.set_title('High - low fidelity')
divnorm=colors.TwoSlopeNorm(vcenter=0.)
im_diff = ax.imshow(y_plot_high_im - y_plot_low_im, cmap='inferno')
fig.colorbar(im_diff, ax=ax)
plt.show()
```

(10201, 1)







```
# Smooth -> exponentiated quadratic kernel
X_train, Y_train = convert_xy_lists_to_arrays([x_train_low, x_train_high],
                                              [y_train_low, y_train_high])

m2_kern = (GPy.kern.RBF(2, lengthscale=500) * \
           GPy.kern.RBF(2, lengthscale=100)) + \
           GPy.kern.Linear(2)
#m2_kern = GPy.kern.Linear(2)

kernels = [m2_kern.copy(), GPy.kern.Linear(2)]
lin_mf_kernel = LinearMultiFidelityKernel(kernels)
gpy_lin_mf_model = GPyLinearMultiFidelityModel(X_train, Y_train, lin_mf_kernel, n_fidelities=2)
gpy_lin_mf_model.mixed_noise.Gaussian_noise.fix(0)
gpy_lin_mf_model.mixed_noise.Gaussian_noise_1.fix(0)
```

Out[52]:

array([0])

In [53]:

```
# Wrapping the model to emukit
lin_mf_model = GPyMultiOutputWrapper(gpy_lin_mf_model, 2, n_optimization_restarts=5)
# Fit the model
lin_mf_model.optimize()
```

Optimization restart 1/5, f = 80244241977.47997  
 Optimization restart 2/5, f = 79622859108.46213  
 Optimization restart 3/5, f = 2994.077140889924  
 Optimization restart 4/5, f = 622.0681757134267  
 Optimization restart 5/5, f = 77689884862.63615

In [54]:

```
# Convert to ndarray representation
X_plot = convert_x_list_to_array([x_plot, x_plot])
X_plot_l = X_plot[:len(x_plot)]
X_plot_h = X_plot[len(x_plot):]

# Find mean and variances
lf_mean_lin_mf_model, lf_var_lin_mf_model = lin_mf_model.predict(X_plot_l)
#lf_std_lin_mf_model = np.sqrt(lf_var_lin_mf_model)
hf_mean_lin_mf_model, hf_var_lin_mf_model = lin_mf_model.predict(X_plot_h)
#hf_std_lin_mf_model = np.sqrt(hf_var_lin_mf_model)
```

Plot of posteriors

In [55]:

```

lf_mean_lin_plot = lf_mean_lin_mf_model.reshape(num_points, num_points)
hf_mean_lin_plot = hf_mean_lin_mf_model.reshape(num_points, num_points)

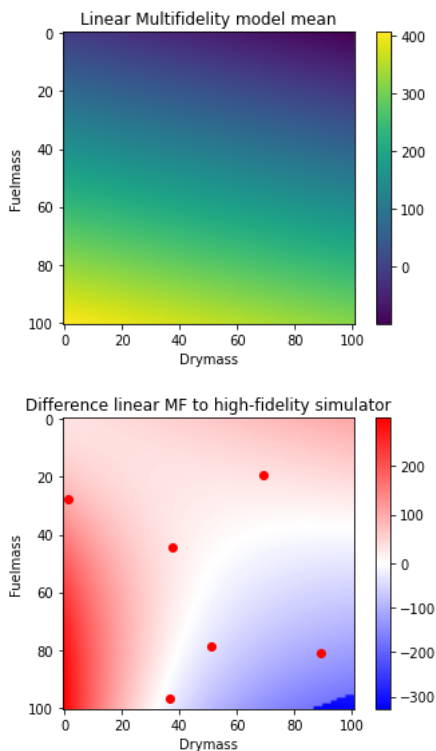
x_th_scaled = x_train_high.copy()
x_th_scaled[:,0] = (x_th_scaled[:,0]-500)/(5500)*101
x_th_scaled[:,1] = (x_th_scaled[:,1]-1000)/(2000)*101

# Mean
fig, ax = plt.subplots()
ax.set_xlabel('Drymass')
ax.set_ylabel('Fuelmass')
ax.set_title('Linear Multifidelity model mean')
im_h_mean = ax.imshow(hf_mean_lin_plot)
fig.colorbar(im_h_mean, ax=ax)
plt.show()

# Difference
fig, ax = plt.subplots()
ax.set_xlabel('Drymass')
ax.set_ylabel('Fuelmass')
ax.set_title('Difference linear MF to high-fidelity simulator')
divnorm=colors.TwoSlopeNorm(vcenter=0.)
im_diff = ax.imshow(y_plot_high_im - hf_mean_lin_plot, cmap='bwr', norm=divnorm)

x_t_scaled = np.array([(x_train_fuel-500)/(5500)*101, (x_train_dry-1000)/2000*100]).reshape(12,2)
ax.plot(x_th_scaled[:,1], x_th_scaled[:,0], 'ro') #Drymass, fuelmass
fig.colorbar(im_diff, ax=ax)
plt.show()

```



In [56]:

```

rmse_mf_lin = np.mean(np.square(y_plot_high_im - hf_mean_lin_plot))
print(rmse_mf_lin)
print(7904)

```

7262.411163153672  
7904

## A standard GP w only high-fidelity data

In [57]:

```
kernel = m2_kern.copy()
high_gp_model = GPy.models.GPRegression(x_train_high, y_train_high, kernel)
high_gp_model.Gaussian_noise.fix(0)
high_gp_model.optimize_restarts(5)
hf_mean_high_gp_model, hf_var_high_gp_model = high_gp_model.predict(x_plot)
hf_std_hf_gp_model = np.sqrt(hf_var_high_gp_model)
```

```
Optimization restart 1/5, f = 36.5172086392213
Optimization restart 2/5, f = 36.517208638843485
Optimization restart 3/5, f = 36.51720863944883
Optimization restart 4/5, f = 36.517208638924004
Optimization restart 5/5, f = 36.517208688110834
```

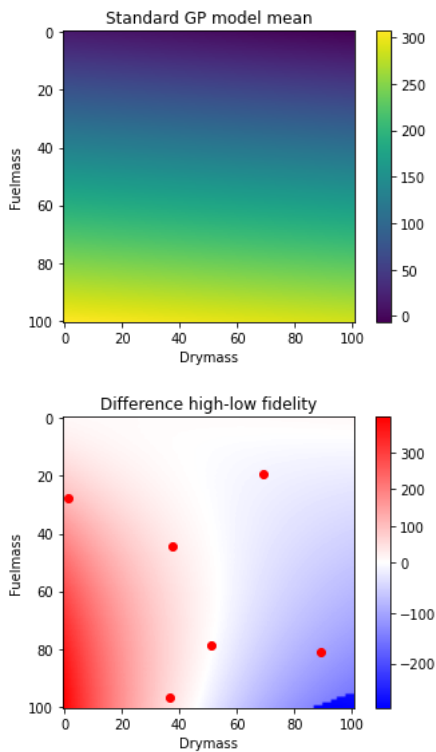
In [58]:

```
# plot the posterior mean for the high-fidelity GP
hf_mean_gp = hf_mean_high_gp_model.reshape(num_points, num_points)

# Mean
fig, ax = plt.subplots()
ax.set_xlabel('Drymass')
ax.set_ylabel('Fuelmass')
ax.set_title('Standard GP model mean')
im_high_mean = ax.imshow(hf_mean_gp)
fig.colorbar(im_high_mean, ax=ax)
plt.show()

# Difference
fig, ax = plt.subplots()
ax.set_xlabel('Drymass')
ax.set_ylabel('Fuelmass')
ax.set_title('Difference high-low fidelity')
divnorm=colors.TwoSlopeNorm(vcenter=0.)
im_diff = ax.imshow(y_plot_high_im - hf_mean_gp, cmap='bwr', norm=divnorm)

x_t_scaled = np.array([(x_train_fuel-500)/(5500)*101, (x_train_dry-1000)/2000*100]).reshape(12,2)
ax.plot(x_th_scaled[:,1], x_th_scaled[:,0], 'ro')
fig.colorbar(im_diff, ax=ax)
plt.show()
```



In [59]:

```
rmse_gp = np.mean(np.square(y_plot_high_im - hf_mean_gp))
print(rmse_mf_lin, rmse_gp)
```

```
7262.411163153672 9630.061344702606
```

## Non-linear multifidelity model

In [60]:

```
from emukit.multi_fidelity.models.non_linear_multi_fidelity_model import make_non_linear_kernels, NonLinearMultiFidelityModel

base_kernel = GPy.kern.Linear
#base_kernel = GPy.kern.LinearSlopeBasisFuncKernel(2, start=0, stop=1e10)
kernels = make_non_linear_kernels(base_kernel, 2, X_train.shape[1] - 1)
```

In [61]:

```
nonlin_mf_model = NonLinearMultiFidelityModel(X_train, Y_train, n_fidelities=2, kernels=kernels,
                                              verbose=True, optimization_restarts=5)

for m in nonlin_mf_model.models:
    m.Gaussian_noise.variance.fix(0)
nonlin_mf_model.optimize()
```

```
# Mean and variance predictions
hf_mean_nonlin_mf_model, hf_var_nonlin_mf_model = nonlin_mf_model.predict(X_plot_h)
#hf_std_nonlin_mf_model = np.sqrt(hf_var_nonlin_mf_model)
lf_mean_nonlin_mf_model, lf_var_nonlin_mf_model = nonlin_mf_model.predict(X_plot_l)
#lf_std_nonlin_mf_model = np.sqrt(lf_var_nonlin_mf_model)
```

```
Optimization restart 1/5, f = 25591.766012624346
Optimization restart 2/5, f = 83843211668545.31
Optimization restart 3/5, f = 27094.421406286256
Optimization restart 4/5, f = 119.71260913416377
Optimization restart 5/5, f = 83170191284272.78
Optimization restart 1/5, f = 48.041084826363644
Optimization restart 2/5, f = 43.98375955090663
Optimization restart 3/5, f = 43.479973084019996
Optimization restart 4/5, f = 43.88444375558964
Optimization restart 5/5, f = 44.69301111845855
```

In [62]:

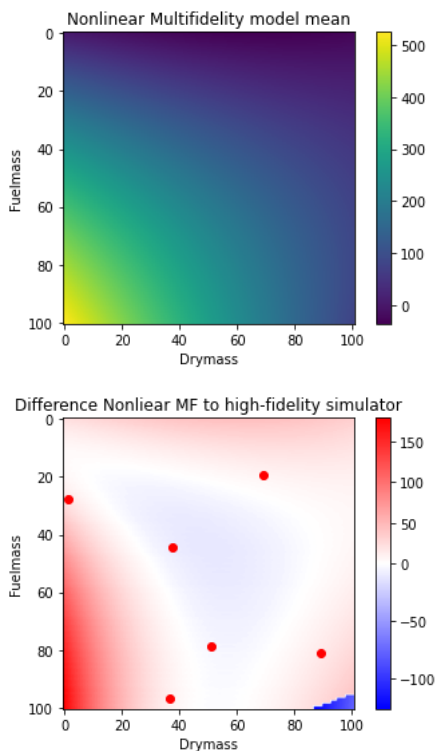
```
# Plot non-linear multifidelity model
hf_mean_hplot = hf_mean_nonlin_mf_model.reshape(num_points, num_points)

x_th_scaled = x_train_high.copy()
x_th_scaled[:,0] = (x_th_scaled[:,0]-500)/(5500)*101
x_th_scaled[:,1] = (x_th_scaled[:,1]-1000)/(2000)*101

# Mean
fig, ax = plt.subplots()
ax.set_xlabel('Drymass')
ax.set_ylabel('Fuelmass')
ax.set_title('Nonlinear Multifidelity model mean')
im_high_mean = ax.imshow(hf_mean_hplot)
fig.colorbar(im_high_mean, ax=ax)
plt.show()

# Difference
fig, ax = plt.subplots()
ax.set_xlabel('Drymass')
ax.set_ylabel('Fuelmass')
ax.set_title('Difference Nonlinear MF to high-fidelity simulator')
divnorm=colors.TwoSlopeNorm(vcenter=0.)
im_diff = ax.imshow(y_plot_high_im - hf_mean_hplot, cmap='bwr', norm=divnorm)

x_t_scaled = np.array([(x_train_fuel-500)/(5500)*101, (x_train_dry-1000)/2000*100]).reshape(12,2)
ax.plot(x_th_scaled[:,1], x_th_scaled[:,0], 'ro')
fig.colorbar(im_diff, ax=ax)
plt.show()
```



In [63]:

```
rmse_mf_nonlin = np.mean(np.square(y_plot_high_im - hf_mean_hplot))
print(rmse_mf_lin)
print(rmse_gp)
print(rmse_mf_nonlin)
```

```
7262.411163153672
9630.061344702606
1103.0607135255602
```

In [ ]: