

State Feedback \mathcal{H}_∞ Optimal Controller

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\mathcal{H}_∞ Optimal Controller

Motivation

- \mathcal{H}_2 Optimal Control
 - ▶ *disturbance error reduction*
 - ▶ *sensor noise error reduction*
- \mathcal{H}_∞ Optimal Control
 - ▶ *disturbance error reduction*
 - ▶ *sensor noise error reduction*
 - ▶ ***tolerant to uncertainties*** – *easier to formulate in \mathcal{RH}_∞ than \mathcal{RH}_2*

	$\ u\ _2$	$\ u\ _\infty$	pow (u)
$\ y\ _2$	$\ \hat{G}(j\omega)\ _\infty$	∞	∞
$\ y\ _\infty$	$\ \hat{G}(j\omega)\ _2$	$\ G(t)\ _1$	∞
pow (y)	0	$\leq \ \hat{G}(j\omega)\ _\infty$	$\ \hat{G}(j\omega)\ _\infty$

∞ -norm of system is pretty useful

Kalman-Yakubovich-Popov (KYP) Lemma

Lemma: Suppose $\hat{G}(s) = \begin{bmatrix} A & B \\ C & D \end{bmatrix}$. Then the following are equivalent conditions.

1. The matrix A is Hurwitz and

$$\|\hat{G}\|_\infty < 1.$$

2. There exists a matrix $X > 0$ such that

$$\begin{bmatrix} C^* \\ D^* \end{bmatrix} \begin{bmatrix} C & D \end{bmatrix} + \begin{bmatrix} A^*X + XA & XB \\ B^*X & -I \end{bmatrix} < 0.$$

- Very useful – relates transfer matrix (frequency domain) inequality to state space conditions
- Convenient way to evaluate \mathcal{H}_∞ norm of transfer matrix

Full State-Feedback \mathcal{H}_∞ Control

One of three formulations

Given system

$$\begin{aligned}\dot{x} &= Ax + B_u u + B_w w, \\ z &= Cx + D_u u + D_w w.\end{aligned}$$

Theorem Controller $u = Kx$ internally stabilizes and minimizes $\|G_{w \rightarrow z}\|_\infty$ iff there exists W , and $X > 0$ such that following optimization problem has solution (A, B_u) stabilizable

$$\min_{X, W} \gamma$$

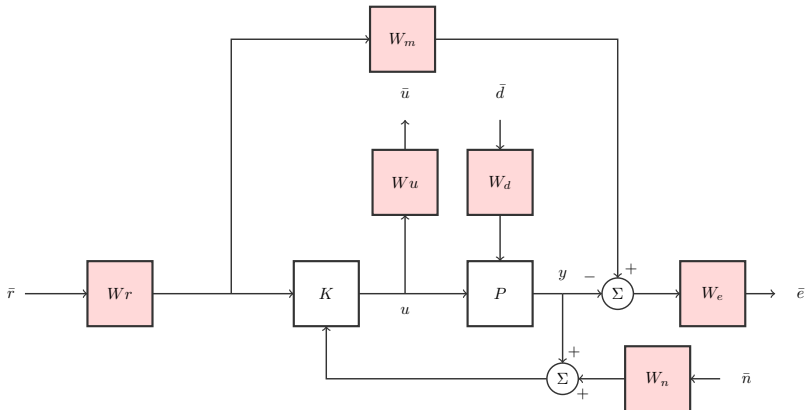
subject to

$$X > 0, \begin{bmatrix} (AX + B_u W) + (*)^T & B_w & (CX + D_u W)^T \\ B_w^T & -\gamma I & D_w^T \\ (CX + D_u W) & D_w & -\gamma I \end{bmatrix} < 0,$$

with $K = WX^{-1}$.

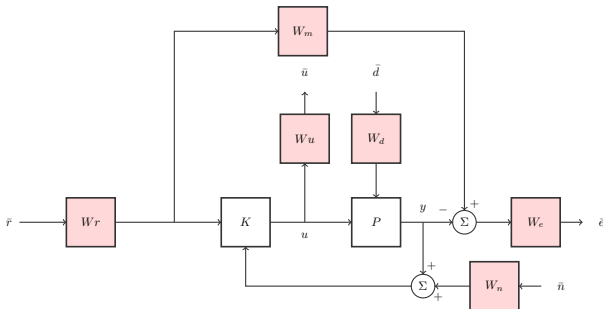
Weighted Performance

For both \mathcal{H}_∞ and \mathcal{H}_2 control



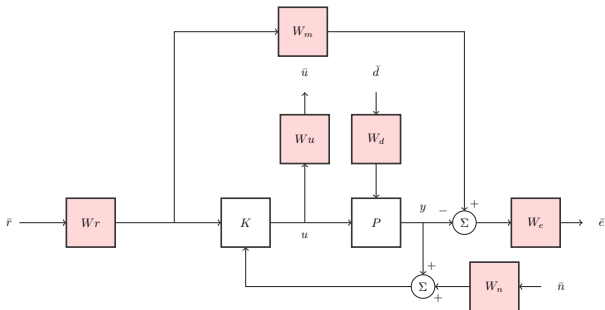
Standard interconnection

Frequency Dependent Weights



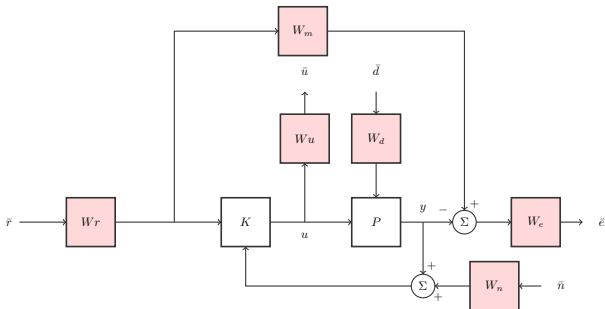
- Some signals may be more important than others
- Signals may not be measured in the same metric
- May be interested in keeping signals small in certain frequency range

Frequency Dependent Weights

 W_r, W_d, W_n 

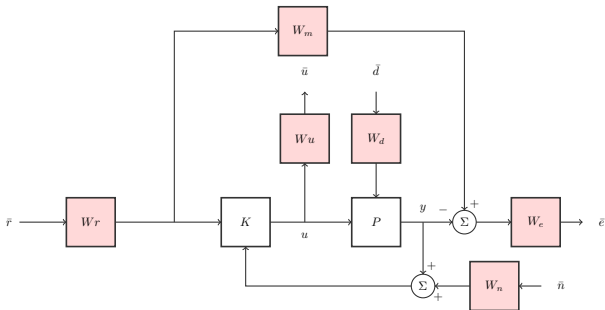
- W_r : specifies frequency content of $r(t)$ – Pilot models, etc.
- W_d : specifies frequency content of $d(t)$ – gust models, road vibration, etc.
- W_n : specifies frequency content of sensor noise – comes from manufacturer.

Frequency Dependent Weights

 W_u 

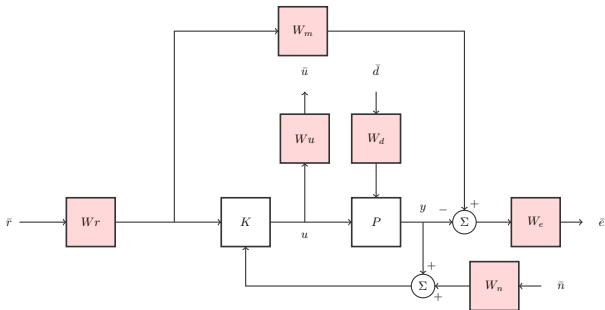
- W_u : defines the **reciprocal of** desired frequency content of $u(t)$
- Can be used to
 - ▶ *include control magnitude, rate constraints*
 - ▶ *specify desired controller roll off – not excite high-frequency uncertain modes*

Frequency Dependent Weights

 W_e 

- W_e : defines the reciprocal of desired error at each frequency

Frequency Dependent Weights

 W_m 

- W_m : Defines the model for model-matching formulation
- Desired response to $r(t)$ is given by response of model W_m
- E.g. second order response – can relate to rise time, overshoot, settling time

\mathcal{H}_∞ Loopshaping – $P(j\omega)C(j\omega)$

Define desired loop shape using weights

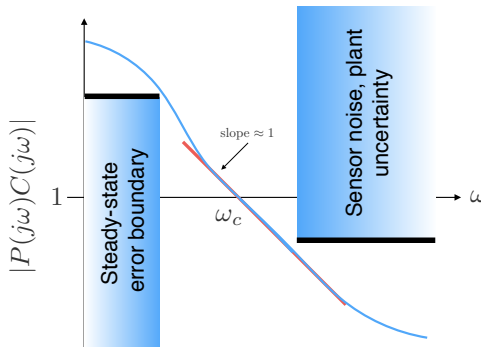
Develop conditions on the Bode plot of the open loop transfer function

- Sensitivity $\frac{1}{1+PC}$
- Steady-state errors: slope and magnitude at $\lim_{\omega} \rightarrow 0$
- Robust to sensor noise
- Disturbance rejection
- Controller roll off \implies not excite high-frequency modes of plant
- Robust to plant uncertainty

Look at Bode plot of $L(j\omega) := P(j\omega)C(j\omega)$

Frequency Domain Specifications

Constraints on the shape of $L(j\omega)$

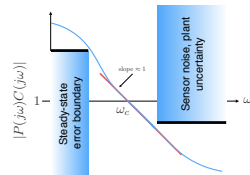
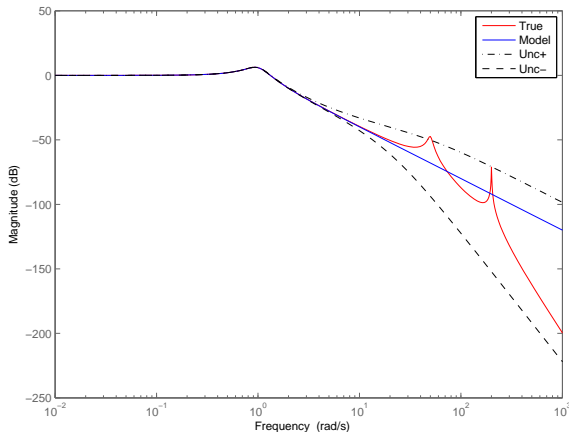


- Choose $C(j\omega)$ to ensure $|L(j\omega)|$ does not violate the constraints
- Slope ≈ -1 at ω_c ensures $PM \approx 90^\circ$
stable if $PM > 0 \implies \angle PC > -180^\circ$

Plant Uncertainty

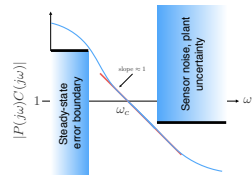
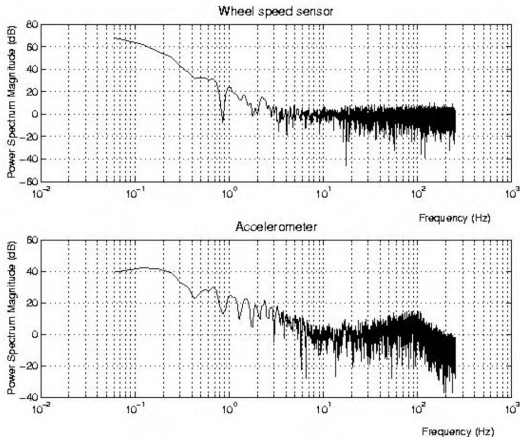
$$P(j\omega) = P_0(j\omega)(1 + \Delta P(j\omega))$$

Bode Diagram



Sensor Characteristics

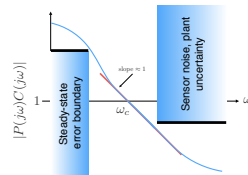
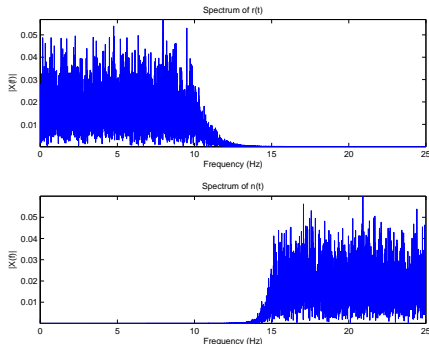
Noise spectrum



$$G_{yn} = -\frac{PC}{1 + PC}$$

Reference Tracking

Bandlimited else conflicts with noise rejection

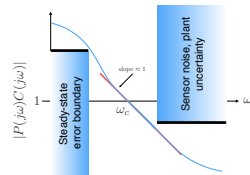
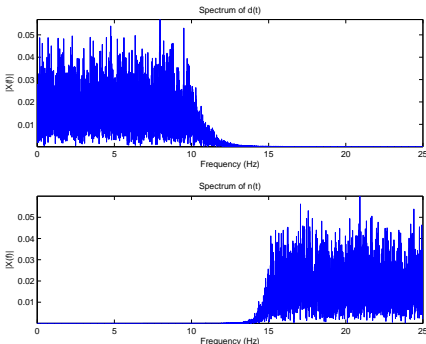


$$G_{yr} = \frac{PC}{1 + PC}$$

$$G_{yn} = -\frac{PC}{1 + PC}$$

Disturbance Rejection

Bandlimited else conflicts with noise rejection



$$G_{yd} = \frac{P}{1 + PC}$$

$$G_{yn} = -\frac{PC}{1 + PC}$$