

# Chap 11 review

1. Knowing that  $g = 9.80 \frac{\text{m}}{\text{s}^2}$  at sea level and that  $R_E = 6.37 \times 10^6 \text{ m}$ , we find that the value of  $g$  in  $\frac{\text{m}}{\text{s}^2}$  at a distance  $R_E$  from the surface of the Earth is

- a. 1.23.      b. 2.45.      c. 4.90.      d. 7.35      e. 9.80      ANS: B

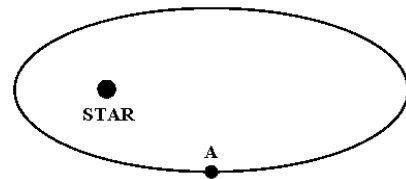
1. B.  $\because g = \frac{GM_E}{R_E^2}$      $g' = \frac{GM_E}{(2R_E)^2} = \frac{g}{4} = \frac{9.8}{4} = 2.45 \text{ m/s}^2$

2. The period of a satellite circling planet Nutron is observed to be 84 s when it is in a circular orbit with a radius of  $8.0 \times 10^6 \text{ m}$ . What is the mass of planet Nutron?

- a.  $6.2 \times 10^{28} \text{ kg}$       b.  $5.0 \times 10^{28} \text{ kg}$       c.  $5.5 \times 10^{28} \text{ kg}$   
d.  $4.3 \times 10^{28} \text{ kg}$       e.  $3.7 \times 10^{28} \text{ kg}$       ANS: D

2. D. circular orbit  $\frac{mv^2}{r} = \frac{m4\pi^2 r}{T^2} = \frac{GMm}{r^2}$   
 $\Rightarrow M_N = \frac{4\pi^2 r^3}{GT^2} = \frac{4\pi^2 (8 \times 10^6)^3}{6.67 \times 10^{-11} \times 84^2} = 4.29 \times 10^{28}$

3. The figure below shows a planet traveling in a counterclockwise direction on an elliptical path around a star located at one focus of the ellipse. When the planet is at point A,



- a. its speed is constant.      b. its speed is increasing      c. its speed is decreasing  
d. its speed is a maximum.      e. its speed is a minimum      ANS: C




4. An asteroid revolves around the Sun with a perihelion 0.5 AU and an aphelion of 7.5 AU. What is its period of revolution?

- a. 4 years      b. 8 years      c. 16 years      d. 32 years      e. 64 years      ANS: B

4. B. 1 AU ~ ave distance from earth to sun  
Perihelion = 近日点, aphelion = 远日点  
for earth  $\frac{a^3}{T^2} = \frac{(1 \text{ AU})^3}{1 \text{ yr}^2} = \frac{(0.5 + 7.5 \text{ AU})^3}{T^2} = \frac{64}{T^2}$   
 $\Rightarrow T = 8 \text{ yrs.}$


5. A spacecraft (mass =  $m$ ) orbits a planet (mass =  $M$ ) in a circular orbit (radius =  $R$ ). What is the minimum energy required to send this spacecraft to a distant point in space where the gravitational force on the spacecraft by the planet is negligible?

- a.  $GmM/(4R)$  b.  $GmM/R$  c.  $GmM/(2R)$  d.  $GmM/(3R)$  e.  $2GmM/(5R)$  ANS: C

5. C.   $E_R = -\frac{GmM}{2R}$  . go to  $\infty$  . total  $E = 0$   
 $\Rightarrow \Delta E = 0 - E_R = GmM/2R$


6. An object is released from rest at a distance  $h$  above the surface of a planet (mass =  $M$ , radius =  $R < h$ ). With what speed will the object strike the surface of the planet? Disregard any dissipative effects of the atmosphere of the planet.

- a.  $\left[ \frac{2GMh}{R(R+h)} \right]^{1/2}$  b.  $\left[ \frac{2GM}{R} \right]^{1/2}$  c.  $\left[ \frac{2GM(h-R)}{Rh} \right]^{1/2}$  d.  $\left[ \frac{2GM}{R+h} \right]^{1/2}$   
 e.  $\left[ \frac{2GM}{R-h} \right]^{1/2}$  ANS: A

6. A.   $E_{\text{mech}} = K + U_g = 0 - \frac{GmM}{(R+h)} = \frac{1}{2}mv^2 - \frac{GmM}{R}$   
 $\Rightarrow v^2 = 2 \left( \frac{GM}{R} - \frac{GM}{R+h} \right) = 2GM \frac{h}{R(R+h)}$   
 $\therefore v = \sqrt{2GMh/R(R+h)}$

7. Planet Zero has a mass of  $5.0 \times 10^{23}$  kg and a radius of  $2.0 \times 10^6$  m. A space probe is launched vertically from the surface of Zero with an initial speed of 4.0 km/s. What is the speed of the probe when it is  $3.0 \times 10^6$  m from Zero's center?

- a. 3.0 km/s b. 2.2 km/s c. 1.6 km/s d. 3.7 km/s e. 5.9 km/s ANS: B

7. B.   $E_{\text{mech}} = \frac{1}{2}m(4 \text{ km/s})^2 - \frac{GmM}{R}$   
 $M = 5 \times 10^{23} \text{ kg}, R = 2 \times 10^6 \text{ m}$   $= \frac{1}{2}mv^2 - \frac{GmM}{3 \times 10^6}$   
 $\Rightarrow v^2 = 16 \times 10^6 + 2GM \left( \frac{1}{3 \times 10^6} - \frac{1}{2 \times 10^6} \right) = 4.9 \times 10^6$   
 $\Rightarrow v = \sqrt{4.9 \times 10^6} = 2.2 \text{ km/s}$


8. The escape velocity at the surface of Earth is approximately 11 km/s. What is the mass, in units of  $M_E$  (the mass of the Earth), of a planet with twice the radius of Earth for which the escape speed is twice that for Earth?

- A)  $2 M_E$  B)  $4 M_E$  C)  $8 M_E$  D)  $1/2 M_E$  E)  $1/4 M_E$  Ans: C

8. C.  $V_{esc} = \sqrt{\frac{2GM_E}{R_E}}$ ,  $V'_{esc} = \sqrt{\frac{2GM}{2R_E}} = 2 V_{esc} = 2 \sqrt{\frac{2GM_E}{R_E}}$   
 $\Rightarrow M = 8M_E$

9. A satellite of mass  $m$  circles a planet of mass  $M$  and radius  $R$  in an orbit at a height  $2R$  above the surface of the planet. What minimum energy is required to change the orbit to one for which the height of the satellite is  $3R$  above the surface of the planet? ANS: A

- a.  $\frac{GmM}{24R}$       b.  $\frac{GmM}{15R}$       c.  $\frac{GmM}{12R}$       d.  $\frac{2GmM}{21R}$       e.  $\frac{3GmM}{5R}$

9. A.   $E_i = \frac{U_i}{2} = \frac{-GmM}{2 \times 3R}$ ,  $E_f = \frac{U_f}{2} = \frac{-GmM}{2 \times 4R}$   
 $\Rightarrow E_f - E_i = \frac{GmM}{R} \left( \frac{1}{6} - \frac{1}{8} \right) = \frac{GmM}{24R}$

10. An energy of 13.6 eV is needed to ionize an electron from the ground state of a hydrogen atom. Selecting the longest wavelength that will work from the those given below, what wavelength is needed if a photon accomplishes this task?

- a. 60 nm      b. 80 nm      c. 70 nm      d. 90 nm      e. 40 nm      ANS: D

10. D.  $13.6 \text{ eV} = hf = \frac{hc}{\lambda} \Rightarrow \lambda = \frac{hc}{13.6 \text{ eV}} = \frac{4.14 \times 10^{-15} \text{ eV} \cdot \text{s} \cdot 3 \times 10^8 \text{ m/s}}{13.6 \text{ eV}}$   
 $\Rightarrow \lambda = 0.913 \times 10^{-7} \text{ m} = 91.3 \text{ nm}$

11. For the following allowed transitions, which photon would have the largest wavelength when an electron "jumps" from one energy level, characterized by the quantum number  $n$ , to another?

- a.  $n = 2$  to  $n = 1$   
 b.  $n = 3$  to  $n = 2$   
 c.  $n = 3$  to  $n = 1$   
 d.  $n = 1$  to  $n = 3$   
 e.  $n = 4$  to  $n = 1$       ANS: B

11. B. for  $E_n = -\frac{13.6}{n^2}$   $\therefore n=2 \rightarrow 1$ ,  $E_2 - E_1 = -13.6 \left( \frac{1}{2^2} - \frac{1}{1^2} \right) = 13.6 \times \frac{3}{4}$   
 $n=3 \rightarrow 2$ ,  $E_3 - E_2 = -13.6 \left( \frac{1}{3^2} - \frac{1}{2^2} \right) = 13.6 \times \frac{5}{36}$   
 $n=3 \rightarrow 1$ ,  $E_3 - E_1 = -13.6 \left( \frac{1}{3^2} - 1 \right) = 13.6 \times \frac{8}{9}$   
 $n=1 \rightarrow 3 = E_1 - E_3 = -13.6 \times \frac{8}{9}$   
 $n=4 \rightarrow 1 = 13.6 \left( 1 - \frac{1}{4^2} \right) = 13.6 \times \frac{15}{16}$   
 for  $\Delta E = hf = \frac{hc}{\lambda} \therefore \Delta E \uparrow, \lambda \downarrow \Rightarrow \min \lambda: n=3 \rightarrow 2$


12. Suppose a beam of electrons is incident on a collection of hydrogen atoms, all of which are in the lowest energy state ( $n = 1$ ). What is the minimum energy the electrons can have if they are to excite the hydrogen atoms into the  $n = 2$  state?

ANS: 10.2 eV

12.  $\Delta E (n=2 \rightarrow 1) = -13.6 \left( \frac{1}{2^2} - 1 \right) = 13.6 \times \frac{3}{4} = 10.2 \text{ eV}$   
 $e^-$  need 10.2 eV to excite H atom from  $n=1 \rightarrow 2$

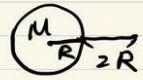
13. An asteroid revolves around the Sun with a period of 8 years. If its perihelion is 0.5 AU find the aphelion of the asteroid. Note, 1 AU is the average distance between the Earth and the Sun.

A. 4.5 AU      B. 6.0 AU      C. 7.5 AU      D. 8.5 AU      E. 9.5 AU

13. C.   $\therefore \frac{a^3}{T^2} = \text{const} = \frac{(1 \text{ AU})^3}{1 \text{ yr}^2} = \frac{(0.5 + x/2)^3}{8}$   
 $\therefore \left( \frac{0.5 + x}{2} \right)^3 = 64 = 4^3 \therefore 0.5 + x = 8 \therefore x = 7.5 \text{ AU}.$


14. An object is released from rest at a distance  $2R$  above the surface of a planet (mass =  $M$ , radius =  $R$ ). With what speed will the object strike the surface of the planet? Disregard any dissipative effects of the atmosphere of the planet.

A.  $\sqrt{\frac{2GM}{3R}}$       B.  $\sqrt{\frac{GM}{R}}$       C.  $\sqrt{\frac{4GM}{3R}}$       D.  $\sqrt{\frac{5GM}{3R}}$       E.  $\sqrt{\frac{2GM}{R}}$

14. C.   $\therefore E = K + U = 0 - \frac{GMm}{3R} = \frac{1}{2}mv^2 - \frac{GMm}{R}$   
 $\Rightarrow \frac{1}{2}v^2 = \frac{GM}{R} \left( 1 - \frac{1}{3} \right) = \frac{GM}{R} \cdot \frac{2}{3} \Rightarrow v = \sqrt{\frac{4GM}{3R}}$

15. A satellite of mass  $m$  circles a planet of mass  $M$  and radius  $R$  in an orbit at a height  $R$  above the surface of the planet. What minimum energy is required to change the orbit to one for which the height of the satellite is  $2R$  above the surface of the planet?

A.  $\frac{GmM}{24R}$       B.  $\frac{GmM}{15R}$       C.  $\frac{GmM}{12R}$       D.  $\frac{GmM}{9R}$       E.  $\frac{GmM}{8R}$

15. C.  circular orbit  $E = U/2$   
 $\therefore E_i = \frac{-GmM}{2 \times 2R}$        $E_f = \frac{-GmM}{2 \times 3R}$        $\Delta E = E_f - E_i = \frac{GmM}{R}$   
 $\therefore \Delta E = E_f - E_i = \frac{GmM}{R} \left( \frac{1}{4} - \frac{1}{6} \right) = \frac{GmM}{12R}$