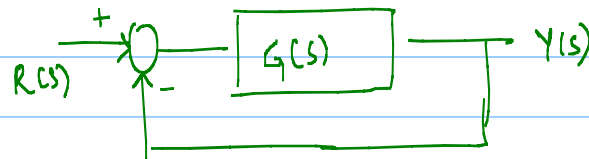


Steady state error:  $e_{ss}$



$$\frac{Y(s)}{R(s)} = \frac{G(s)}{1 + G(s)}$$

$$\begin{aligned} E(s) &= R(s) - Y(s) \\ &= R(s) \left( 1 - \frac{Y(s)}{R(s)} \right) \end{aligned}$$

Let  $\frac{Y(s)}{R(s)} = T(s)$  — Closed loop transfer function

the steady state error,

$$\begin{aligned} e_{ss} &= \lim_{t \rightarrow \infty} e(t) = \lim_{s \rightarrow 0} s E(s) \\ e_{ss} &= \lim_{s \rightarrow 0} s R(s) (1 - T(s)) \end{aligned}$$

$$e_{ss} = \lim_{s \rightarrow 0} s R(s) \left( 1 - \frac{G(s)}{1 + G(s)} \right)$$

$$e_{ss} = \lim_{s \rightarrow 0} \frac{s R(s)}{1 + G(s)}$$

Step input  $R(s) = 1/s$

$$e_{ss} = \lim_{s \rightarrow 0} \frac{s \times 1/s}{1 + G(s)}$$

$$e_{ss} = \lim_{s \rightarrow 0} \frac{1}{1 + G(s)}$$

$$e_{ss} = \frac{1}{1 + \lim_{s \rightarrow 0} G(s)}$$

$$e_{ss} = \frac{1}{1 + \lim_{s \rightarrow 0} G(s)} = \frac{1}{1 + K_p}$$

where,

$$K_p = \lim_{s \rightarrow 0} G(s)$$

— Static position error constant

Ramp input:  $R(s) = 1/s^2$

$$e_{ss} = \lim_{s \rightarrow 0} \frac{s \times 1/s^2}{1 + G(s)}$$

$$= \lim_{s \rightarrow 0} \frac{1}{s + sG(s)}$$

$$e_{ss} = \frac{1}{\lim_{s \rightarrow 0} sG(s)}$$

$$e_{ss} = \frac{1}{\lim_{s \rightarrow 0} s G(s)} = \frac{1}{K_V}$$

$$K_V = \lim_{s \rightarrow 0} s G(s)$$

Static Velocity error Constant.

Parabolic input:  $R(s) = 1/s^3$

$$e_{ss} = \lim_{s \rightarrow 0} \frac{s \times 1/s^3}{1 + G(s)}$$

G

$$e_{ss} = \frac{1}{\lim_{s \rightarrow 0} s^2 G(s)}$$

$$e_{ss} = \frac{1}{\lim_{s \rightarrow 0} s^2 G(s)} = \frac{1}{K_a}$$

$$K_a = \lim_{s \rightarrow 0} s^2 G(s)$$

Static acceleration error constant

General Structure of  $G(s)$

$$G(s) = \frac{k (s+z_1) \dots (s+z_m)}{s^N (s+p_1) \dots (s+p_n)}$$

→ The term  $s^N$  in the denominator representing a pole of multiplicity  $N$  at origin (no of integrators) denotes the 'TYPE' of the system.

type 0 - No poles at origin

type 1 - one pole at origin.

type 0:

$$\lim_{s \rightarrow 0} G(s) = \frac{k z_1 z_2 \dots z_m}{p_1 p_2 \dots p_n} = K'$$

$$\Rightarrow K_p = \lim_{s \rightarrow 0} G(s) = K'$$

Step input  $e_{ss} = \frac{1}{1+K_p} = K''$

$\Rightarrow$  type 0 systems will track step input with  $e_{ss} = K''$

Ramp input:  $K_v = \lim_{s \rightarrow 0} s G(s) = 0$

$$\Rightarrow e_{ss} = \frac{1}{K_v} = \frac{1}{0} = \infty$$

$\Rightarrow$  type 0 system can't track ramp input

parabolic input:  $K_a = \lim_{s \rightarrow 0} s^2 G(s) = 0$

$$e_{ss} = \frac{1}{K_a} = \infty$$

$\Rightarrow$  type 0 system can't track parabolic input.

type 1:  
(1 pole at origin)

$$K_p = \infty \Rightarrow e_{ss} = \frac{1}{1+K_p} = 0$$

$\Rightarrow$  type 1 systems can track step input

$$K_v = \lim_{s \rightarrow 0} s G(s) = K'$$

$$e_{ss} = \frac{1}{K'}$$

$\Rightarrow$  type 1 system can track ramp input but with constant error  $1/K'$

$$K_a = \lim_{s \rightarrow 0} s^2 G(s) = 0 \Rightarrow e_{ss} = \frac{1}{K_a} = \infty$$

$\Rightarrow$  type 1 system can't track parabolic input

type 2:  
(2 poles at origin)

$$K_p = \infty \quad e_{ss} = 0$$

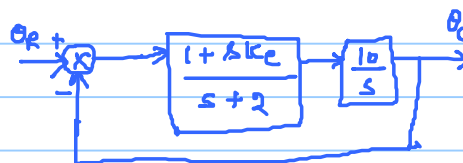
$$K_v = \infty \quad e_{ss} = 0$$

$$K_a = K' \quad e_{ss} = \frac{1}{K'}$$

$\Rightarrow$  type 2 system will follow step & ramp but not parabolic input.

type	Step input		Ramp		Parabolic	
	$K_p$	$e_{ss}$	$K_v$	$e_{ss}$	$K_a$	$e_{ss}$
0	$k'$	$1/(1+k')$	0	$\infty$	0	$\infty$
1	$\infty$	0	$k'$	$1/k'$	0	$\infty$
2	$\infty$	0	$\infty$	0	$k'$	$1/k'$

Determine the values of  $e_{ss}$  (unit ramp input) with and without error rate control  $k_e$ . Given  $\zeta = 0.6$  (with  $k_e$ )



Without  $k_e$

$$G(s) = \frac{10}{s(s+2)}$$

With  $k_e$

$$G(s) = \frac{10(1 + s k_e)}{s(s+2)}$$

$$G(s) = \frac{(1 + s k_e)10}{s(s+2)}$$

Without  $k_e$ :  $k_e = 0$

$$G(s) = \frac{10}{s(s+2)}$$

$$K_p = \lim_{s \rightarrow 0} G(s) = \infty$$

Unit Step input

$$e_{ss} = \frac{1}{\infty} = 0$$

$$K_v = \lim_{s \rightarrow 0} s G(s) = \lim_{s \rightarrow 0} s \times \frac{10}{s(s+2)} = 5$$

Unit Ramp input

$$e_{ss} = \frac{1}{K_v} = 1/5$$

With  $k_e$ :  $k_e = 0.18$

$$G(s) = \frac{10(1 + 0.18s)}{s(s+2)}$$

$$K_p = \lim_{s \rightarrow 0} G(s) = \infty$$

$$e_{ss} = 0$$

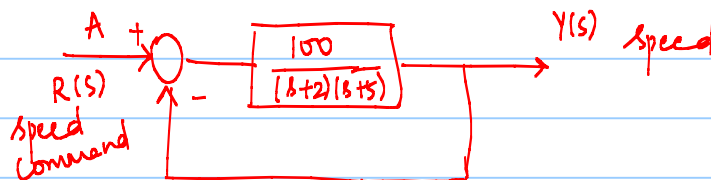
$$k_v = \lim_{s \rightarrow 0} s G(s) = \lim_{s \rightarrow 0} s \frac{10(1+0.1s)}{s(s+2)}$$

$$k_v = 5$$

$$e_{ss} = 1/5$$

⇒ PD Controller can't reduce steady state error but reduces peak overshoot & also settling time.

The following is the model of a racing vehicle that affect the acceleration & speed attainable. The speed of the car is represented by the model shown



- Calculate the steady state error of the car to step command in speed
- Calculate overshoot of the speed to a step command

a)  $G(s) = \frac{100}{(s+2)(s+5)}$  type 0

$$k_p = \lim_{s \rightarrow 0} G(s) = \frac{100}{2 \times 5} = 10$$

$$e_{ss} = \frac{1}{1+k_p} = \frac{1}{11}$$

b)

$$\text{CLTF } T(s) = \frac{G(s)}{1+G(s)} = \frac{100}{s^2 + 7s + 110}$$

$$\omega_n^2 = 110$$

$$\omega_n = \sqrt{110}$$

$$2\zeta\omega_n = 7$$

$$\Rightarrow \zeta = \frac{7}{2\sqrt{110}} = 0.334$$

$$M_p = \frac{e^{-\zeta\pi/\sqrt{1-\zeta^2}}}{\sqrt{1-\zeta^2}}$$

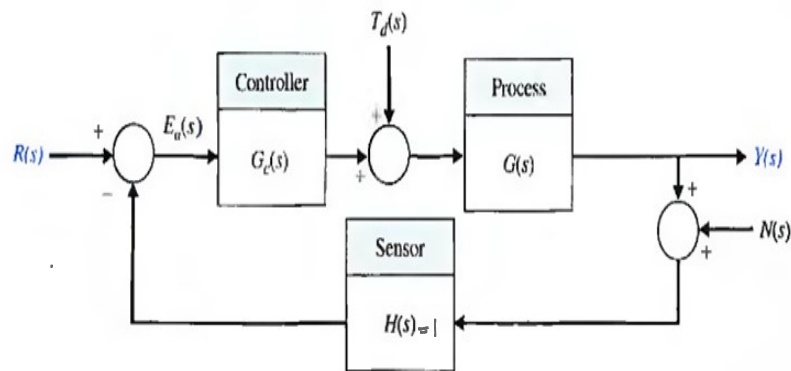
$$M_p = \left( e^{-\pi \zeta / \sqrt{1-\zeta^2}} \right)$$

$$= e^{-\pi \times 0.334 / \sqrt{1-0.334^2}}$$

$$M_p = 0.324$$

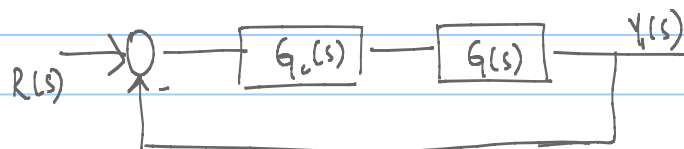
$$\% M_p = 32.4 \%$$

Error Signal Analysis.



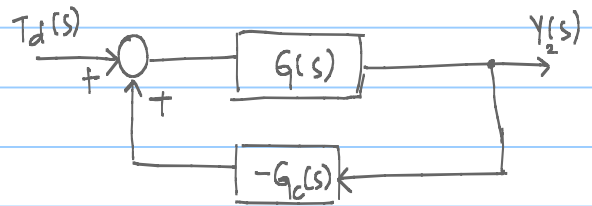
$$E(s) = Y(s) - R(s) = R(s) - Y(s)$$

case (i) :  $T_d(s) = 0 = N(s)$



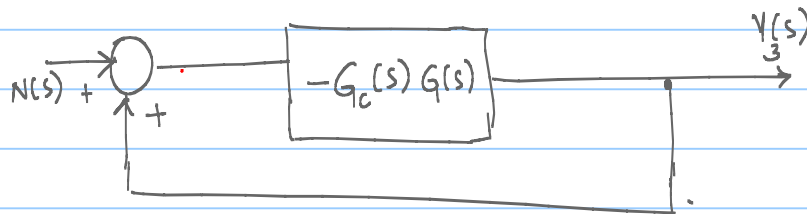
$$Y_1(s) = \frac{G_c(s) G(s)}{1 + G_c(s) G(s)} R(s)$$

Case (ii):  $R(s) = 0 = N(s)$



$$Y_2(s) = \frac{G(s)}{1 + G_c(s) G(s)} T_d(s)$$

Case (iii):  $R(s) = 0 = T_d(s)$



$$Y_3(s) = \frac{-G_c(s) G(s)}{1 - (-G_c(s) G(s))} N(s)$$

$$Y(s) = Y_1(s) + Y_2(s) + Y_3(s)$$

$$Y(s) = \frac{G_c G}{1 + G_c G} R(s) + \frac{G}{1 + G_c G} T_d(s) - \frac{G_c G}{1 + G_c G} N(s)$$

$$\begin{aligned} E(s) &= R(s) - Y(s) \\ &= \left(1 - \frac{G_c G}{1 + G_c G}\right) R(s) - \frac{G}{1 + G_c G} T_d(s) + \frac{G_c G}{1 + G_c G} N(s) \end{aligned}$$

$$E(s) = \frac{1}{1 + G_c G} R(s) - \frac{G}{1 + G_c G} T_d(s) + \frac{G_c G}{1 + G_c G} N(s)$$

Let  $L(s) = G_c(s) G(s)$  — Loop gain

$$E(s) = \frac{1}{1 + L(s)} R(s) - \frac{G}{1 + L(s)} T_d(s) + \frac{L(s)}{1 + L(s)} N(s)$$



Let sensitivity function

$$\underline{S(s)} \triangleq \frac{1}{1+L(s)}$$

Complementary function

$$C(s) = \frac{L(s)}{1+L(s)}$$

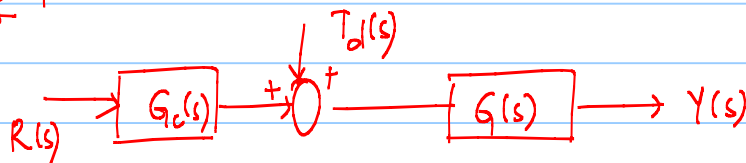
$$\begin{aligned} S(s) + C(s) &= \frac{1}{1+L(s)} + \frac{L(s)}{1+L(s)} \\ &= 1 \end{aligned}$$

$$E(s) = S(s) R(s) - G(s) S(s) T_d(s) + C(s) N(s)$$

→  $\uparrow L(s) \Rightarrow$  effect of  $T_d \downarrow$  in low freq

$\downarrow L(s) \Rightarrow$  effect of  $N \downarrow$  in high frequency range

Open loop :



$$E(s) = R(s) - Y(s)$$

$$Y(s) = G_c(s) G(s) R(s) + G(s) T_d(s)$$

$$E(s) = R(s) - G_c(s) G(s) R(s) - G(s) T_d(s)$$

$$E(s) = (1 - G_c(s) G(s)) R(s) - G(s) T_d(s)$$

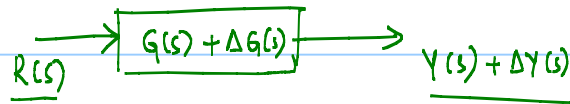
⇒ No way we can reduce the effect of disturbance on the open loop error signal

## Sensitivity to parameter variations:

$$G(s) \longrightarrow G(s) + \Delta G(s)$$

Small perturbation

Open loop system: Let  $T_d(s) = 0$



$$Y(s) + \Delta Y(s) = \underbrace{(G(s) + \Delta G(s))}_{\text{change in the parameter}} R(s)$$

$$E(s) + \Delta E(s) = R(s) - (Y(s) + \Delta Y(s))$$

$$\Delta E(s) = R(s) - (G(s) + \Delta G(s)) R(s) - E(s)$$

$$= R(s) - (G(s) + \Delta G(s)) R(s) - (R(s) - \underbrace{G(s) R(s)}_{\text{change in the parameter}})$$

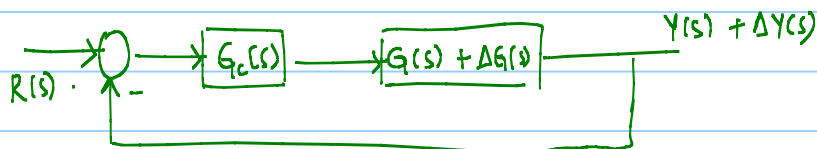
$$\Delta E(s) = \cancel{R(s)} - (\cancel{G(s)} + \Delta G(s)) R(s) - \cancel{R(s)} + \cancel{G(s) R(s)}$$

$$\Delta E(s) = -\Delta G(s) R(s)$$

$\Rightarrow \Delta E(s)$  is directly proportional to *change in the parameter*

$\Rightarrow$  Open loop systems are sensitive to parameter variations

Close loop system:  $T_d(s) = 0$ ,  $N(s) = 0$



$$Y(s) + \Delta Y(s) = \frac{G_c(s) (G(s) + \Delta G(s))}{1 + G_c(s) (G(s) + \Delta G(s))} R(s)$$

$$E(s) + \Delta E(s) = R(s) - (Y(s) + \Delta Y(s))$$

$$E(s) + \Delta E(s) = \left( 1 - \frac{G_c (G + \Delta G)}{1 + G_c (G + \Delta G)} \right) R(s)$$

$$E(s) + \Delta E(s) = \frac{1}{1 + G_c (G + \Delta G)} R(s)$$

$$\Delta E(s) = \frac{1}{1 + G_c (G + \Delta G)} R(s) - \underbrace{E(s)}_{R(s) - Y(s)}$$

$$\Delta E(s) = \frac{1}{1 + G_c(s)G(s)} R(s) - R(s) + \frac{G_c(s)G(s)}{1 + G_c(s)G(s)} R(s)$$

$$\Delta E(s) = \frac{-\Delta G(s) G_c(s)}{(1 + G_c(s)G(s))(1 + G_c(s)G(s))} R(s)$$

$$\text{Let } G_c(s) \Delta G(s) \ll G_c(s)G(s)$$

$$\Delta E(s) \approx \frac{-\Delta G(s) G_c(s)}{(1 + G_c(s)G(s))^2} R(s)$$

→ The change in the error signal is reduced by a factor  $1 + \underbrace{G_c(s)G(s)}_{L(s)}$  over the range of frequencies

→ For large  $L(s)$ ,  $(1 + L(s)) \approx L(s)$

$$\Delta E(s) \approx \frac{-\Delta G(s) G_c(s)}{L(s) (G_c(s)G(s))} R(s)$$

$$\Delta E(s) \approx \frac{-\Delta G(s)}{L(s) G(s)} R(s)$$

⇒ The closed loop system has reduced sensitivity to parameter variations

⇒ CLOSED LOOP SYSTEMS ARE BETTER

Sensitivity of  $T(s)$ :  $T(s)$  - closed loop transfer function

It is the ratio of percentage change in system transfer function ( $T(s)$ ) to the percentage change in the process transfer function ( $G(s)$ )

$$S_G^T = \frac{\Delta T(s)/T(s)}{\Delta G(s)/G(s)}$$

for small incremental changes  $\Delta \rightarrow 0$

$$T(s) = \frac{G}{1 + GH}$$

$$S_G^T = \frac{\partial T(s)}{\partial G(s)} \times \frac{G(s)}{T(s)}$$

## Closed loop transfer

$$T(s) = \frac{G(s) G_c(s)}{1 + G(s) H(s) G_c(s)}$$

$$S_G^T = \frac{\frac{\partial T(s)}{\partial G(s)}}{T(s)} \times \frac{\cancel{G(s)}}{\cancel{G(s)} G_c(s)} \times \frac{1}{1 + G(s) H(s) G_c(s)}$$

$$\frac{\partial T(s)}{\partial G(s)} = \frac{(1 + G(s) H(s) G_c(s)) G_c(s) - G(s) G_c(s) G_c(s) H(s)}{(1 + G(s) H(s) G_c(s))^2}$$

$$= \frac{G_c(s)}{(1 + G(s) H(s) G_c(s))^2}$$

$$S_G^T = \frac{\cancel{G(s)}}{(1 + G(s) H(s) G_c(s))^2} \times \frac{1 + \cancel{G(s)} H(s) G_c(s)}{\cancel{G(s)}}$$

$$S_G^T = \frac{1}{1 + G(s) H(s) G_c(s)}$$

To find the sensitivity w.r.t 'α' parameter

$$S_\alpha^T = \frac{\partial T}{\partial \alpha} \times \frac{\alpha}{T}$$

$$\dot{T} \rightarrow G \rightarrow \underline{\alpha}$$

$$= \frac{\partial T}{\partial G} \frac{\partial G}{\partial \alpha} \times \frac{\alpha}{T} \times \frac{G}{G}$$

$$= \frac{\partial T}{\partial G} \times \frac{G}{T} \times \frac{\partial G}{\partial \alpha} \times \frac{\alpha}{G}$$

$$\boxed{S_\alpha^T = S_G^T S_\alpha^G}$$

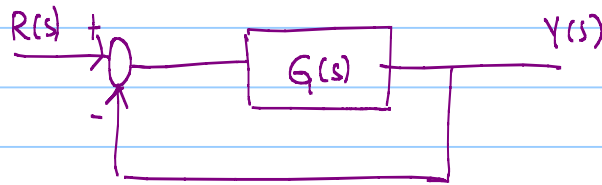
Open loop

$$S_G^G = \frac{\partial G}{\partial G} \frac{G}{G} = 1$$

- A closed loop system is used to track the sun to obtain maximum power from a photovoltaic array. The tracking system may be represented by

$$G(s) = \frac{100}{\tau s + 1} \text{ where } \tau = 3 \text{ seconds nominally}$$

- a) Calculate the sensitivity of this system for a small change in  $\tau \rightarrow S_\tau^T$
- b) Calculate the time constant of the closed loop system response



$$T(s) = \frac{G}{1 + G}$$

$$S_\tau^T = S_G^T S_\tau^G$$

$$S_G^T = \frac{1}{1 + G(s)}$$

$$S_\tau^G = \frac{\partial}{\partial \tau} \left( \frac{100}{1 + s\tau} \right) \times \frac{\tau}{\frac{100}{1 + s\tau}}$$

$$S_\tau^G = \frac{-\tau s}{1 + s\tau}$$

$$S_\tau^T = \frac{1}{1 + \frac{100}{1 + s\tau}} \times \frac{-\tau s}{1 + s\tau}$$

$$S_\tau^T = \frac{-\tau s}{1 + s\tau + 100}$$

$$\tau = 3 \text{ sec}$$

$$S_\tau^T = \frac{-3s}{3s + 101}$$

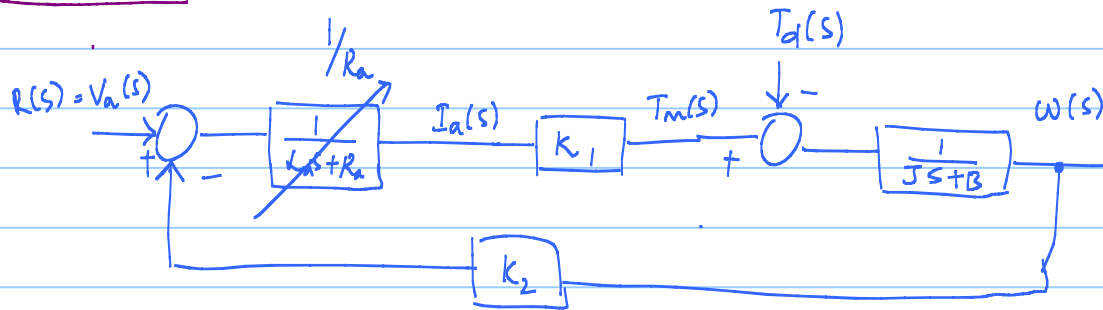
b)  $T(s) = \frac{G}{1 + G} = \frac{100}{3s + 101}$

$$T(s) = \frac{100/101}{3/101 s + 1}$$

$$\text{Time constant} = 3/101$$

## Disturbance signals in feedback control system

DC Motor :

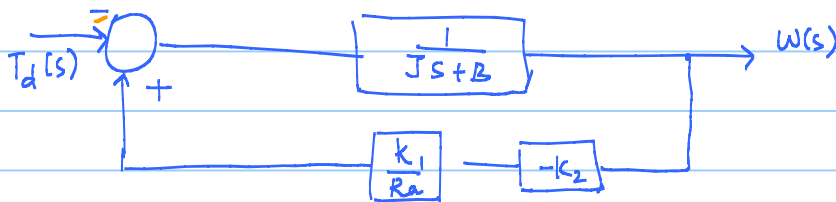


$$L_a \text{ is negligible compared to } R_a \Rightarrow \frac{1}{L_a s + R_a} \approx \frac{1}{R_a}$$

$$\text{Error: } E(s) = R(s) - W(s)$$

Let  $R(s) = 0$ , we will analyse the effect of disturbance

$$E(s) = -W(s)$$



$$\frac{W(s)}{T_d(s)} = - \frac{R_a}{(Js + B)R_a + k_1 k_2}$$

$$= \frac{1/Js + B}{1 - \left(\frac{1}{Js + B}\right) \times \left(-\frac{k_1 k_2}{R_a}\right)}$$

$$E(s) = -W(s)$$

$$= \frac{R_a}{(Js + B)R_a + k_1 k_2} T_d(s)$$

$$\text{Let } T_d(s) = \frac{D}{s}$$

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$T_d(t) = D \quad t \geq 0$  — disturbance signal

$$E(s) = \frac{DRa}{(Js+B)Ra + k_1 k_2} \times \frac{1}{s}$$

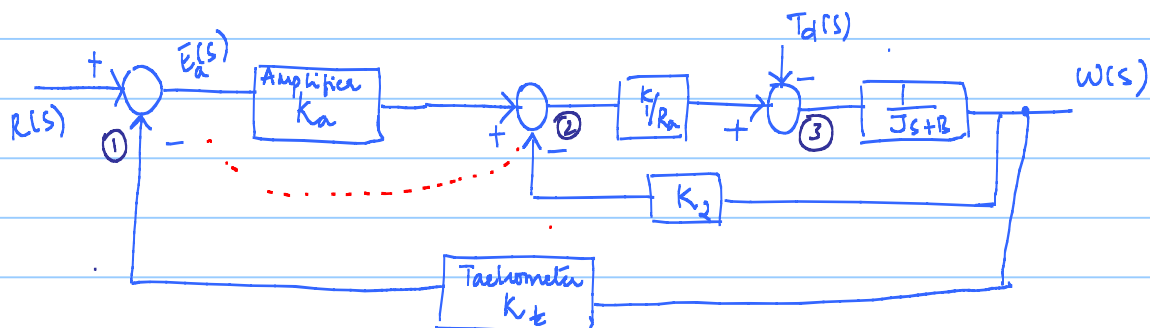
Steady state error

$$e_{ss} = \lim_{t \rightarrow \infty} e(t) = \lim_{s \rightarrow 0} sE(s)$$

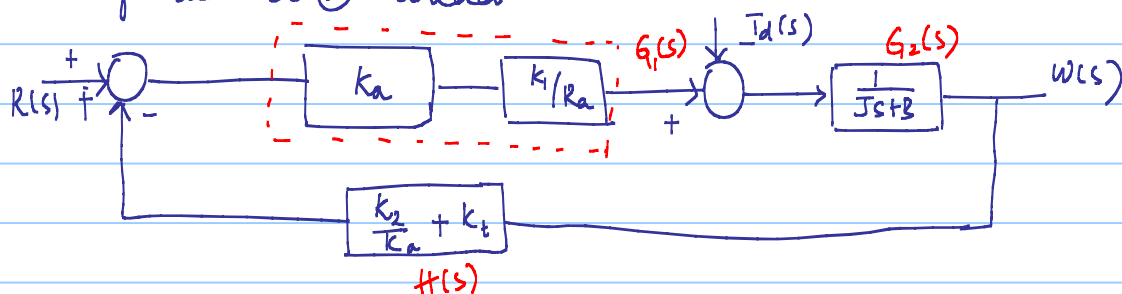
$$e_{ss} = \frac{DRa}{k_1 k_2 + BRa} = -w(\infty)$$

The steady state value of the error cannot be reduced as the values of  $Ra$ ,  $B$ ,  $k_1$ ,  $k_2$  are fixed for the DC motor

$\Rightarrow$  Disturbance upon the open loop system cannot be reduced.



Moving summer (2) ahead the block  $k_1$



$$G_1(s) = \frac{k_a k_1}{Ra}$$

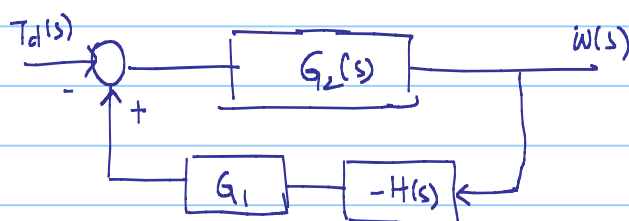
$$G_2(s) = \frac{1}{Js+B}$$

$$H(s) = \frac{k_2 + k_t k_a}{k_a}$$

$$E(s) = R(s) - W(s)$$

$$R(s) = 0$$

$$= -W(s)$$



$$W(s) = \frac{G_2(s)}{1 - G_1 G_2 (-H)} \times -T_d(s)$$

$$W(s) = - \frac{G_2(s)}{1 + G_1 G_2 H} T_d(s)$$

$\underbrace{1 + G_1 G_2 H}_{\approx G_1 G_2 H} \quad G_1 G_2 H \gg 1$

$$E(s) = -W(s) = \frac{G_2}{G_1 G_2 H} T_d(s)$$

$$E(s) = \frac{1}{G_1 H} T_d(s)$$

Let  $T_d(s) = \frac{D}{s}$

$$E(s) = \frac{T_d(s)}{\frac{k_a k_1}{R_a} \times \left( \frac{k_2 + k_t k_a}{k_a} \right)} = \frac{T_d(s)}{\frac{k_1 k_2 + k_1 k_t k_a}{k_a}}$$

$$e_{ss} = \lim_{s \rightarrow 0} s E(s) = \lim_{s \rightarrow 0} s \times \frac{D/s}{\frac{k_1 k_2 + k_1 k_t k_a}{k_a}}$$

$$e_{ss} = \frac{D}{\frac{k_1 k_2}{R_a} + \frac{k_1 k_t k_a}{R_a}}$$

Choose  $k_a$  &  $k_t$  such that the steady state error reduces.  
∴ Closed loop systems have better disturbance rejection.

Control of Transient response:  $T_d(s) = N(s) = 0$

Let us consider the DC motor.

$$\frac{\theta(s)}{V_a(s)} = \frac{k_1}{s(Js+B)(L_a s + R_a) + k_1 k_2 s}$$

$L_a$  is negligible

$$\frac{\theta(s)}{V_a(s)} = \frac{k_1}{s((Js+B)R_a + k_1 k_2)} \Rightarrow \frac{s \theta(s)}{V_a(s)} = \frac{k_1}{(Js+B)R_a + k_1 k_2}$$



step response  $\therefore V_a(s) = \frac{V_a}{s}$

$V_a$  - amplitude of the voltage

$$W(s) = \frac{k_1 V_a}{s(JR_a + BR_a + k_1 k_2)}$$

$$W(s) = \frac{A}{s} + \frac{C}{JR_a s + \underbrace{(BR_a + k_1 k_2)}_{k'}}$$

$$W(s) = \frac{k_1 V_a}{k'} \times \frac{1}{s} - \frac{k_1 V_a \times JR_a / k'}{JR_a s + k'}$$

$$W(s) = \frac{k_1 V_a}{k'} \frac{1}{s} - \frac{k_1 V_a / k'}{s + k' / JR_a}$$

ILT  $\rightarrow$

$$w(t) = \frac{k_1 V_a}{k'} \left( 1 - e^{-k' / JR_a t} \right), t \geq 0$$

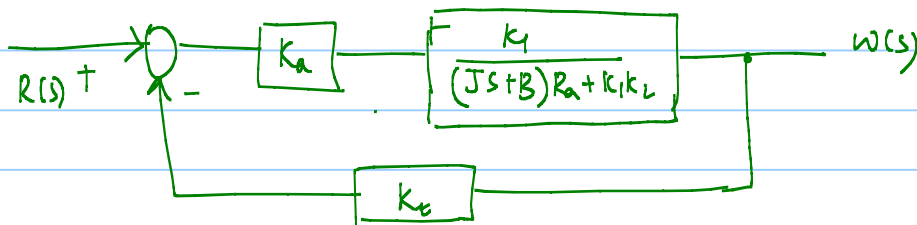
$$w(t) = \frac{k_1 V_a}{BR_a + k_1 k_2} \left( 1 - e^{-\frac{BR_a + k_1 k_2}{JR_a} t} \right), t \geq 0$$

Contributes to transient response

$\Rightarrow$  Transient response ( $t_n, t_p, M_p$ ) cannot be changed as

$B, R_a, k_1, k_2$  &  $J$  are fixed for DC motor.

closed  
loop  
system



$$\frac{W(s)}{R(s)} = \frac{k_a k_1}{((Js+B)R_a + k_1 k_2 + k k_a k_t)}$$

$$R(s) = 1/s$$

$$W(s) = \frac{k_a k_1}{J R_a s + \underbrace{B R_a + k_1 k_2 + k k_a k_t}_{k'}} \times \frac{1}{s}$$

$$W(s) = \frac{A}{s} + \frac{C}{J R_a s + k'}$$

$$A = \left. s W(s) \right|_{s=0} = \left. \frac{k_a k_1}{J R_a s + k'} \right|_{s=0}$$

$$A = \frac{k_a k_1}{k'}$$

$$C = \left. (J R_a s + k') W(s) \right|_{s = -k'/J R_a} = \frac{k_a k_1}{-k'/J R_a}$$

$$W(s) = \frac{k_a k_1}{k'} \frac{1}{s} - \frac{J R_a k_a k_1}{k'} \frac{1}{J R_a (s + k'/J R_a)}$$

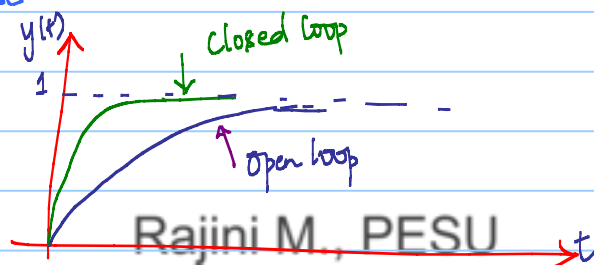
ILT  $\rightarrow$

$$W(t) = \frac{k_a k_1}{k'} \left( 1 - e^{-t k'/J R_a} \right), \quad t \geq 0$$

$$= \frac{k_a k_1}{k'} \left( 1 - e^{-t \left( \frac{B R_a + k_1 k_2}{J R_a} \right)} e^{-t \frac{k k_a k_t}{J R_a}} \right), \quad t \geq 0$$

$k_a$  &  $k_t$  is designed in such a way that the rise time,  $t_p$  &  $M_p$  can be controlled.

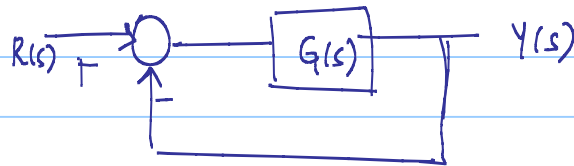
$\therefore$  Closed loop systems have better control over transient response.



The transfer function of the process is given by

$$G(s) = \frac{k}{(s+4)^2} \quad \text{OLTF}$$

Calculate the steady state error for step command 'A'



Method 1:

$$\begin{aligned} E(s) &= R(s) - Y(s) & Y(s) &= \frac{G}{1+G} R(s) \\ &= R(s) - \frac{G}{1+G} R(s) \\ &= \left(1 - \frac{G}{1+G}\right) R(s) \\ &= \frac{R(s)}{1+G(s)} \end{aligned}$$

$$E(s) = \frac{R(s)}{1 + k/(s+4)^2} = \frac{A}{s(1 + k/(s+4)^2)}$$

$$e_{ss} = \lim_{s \rightarrow 0} s E(s) \quad \text{--- Final value Theorem}$$

$$e_{ss} = \lim_{s \rightarrow 0} \cancel{s} \frac{A}{\cancel{s}(1 + k/(s+4)^2)}$$

$$e_{ss} = \frac{A}{1 + k/16}$$

Method 2:  $R(s) = A/s$

$$e_{ss} = \frac{A}{1 + \lim_{s \rightarrow 0} G(s)} = \frac{A}{1 + k_p}$$

$$K_p = \lim_{s \rightarrow 0} G(s) = \frac{K}{16}$$

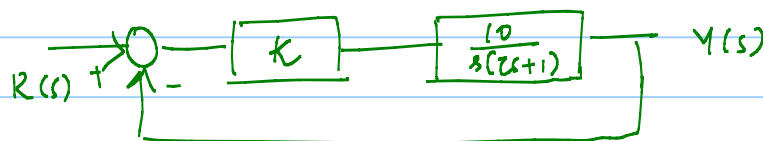
$$e_{ss} = \frac{A}{1 + K/16}$$

b) What should be the value of  $K$  such that error  $e_{ss} = 0.01$ . Let  $A = 1$

$$0.01 = \frac{1}{1 + K/16}$$

$$K = 1584$$

2. Let  $G(s) = \frac{10}{s(\tau s + 1)}$  where  $\tau = 0.001 \text{ sec}$ ,



a) Calculate  $e_{ss}$  (Step input is  $A$ )

b) Calculate the required  $K$  in order to maintain  $e_{ss} = 0.1 \text{ mm}$  for a ramp input  $\frac{10 \text{ cm/s}}{0.1 \text{ m/s}}$   $R(s) = \frac{0.1}{s^2}$

$$a) \quad e_{ss} = \lim_{s \rightarrow 0} (R(s) - Y(s))$$

$$e_{ss} = \lim_{s \rightarrow 0} \frac{s \cdot A/s}{1 + K G(s)} = \lim_{s \rightarrow 0} \frac{A}{1 + K \frac{10}{s(\tau s + 1)}} = 0$$

$$b) \quad e_{ss} = \lim_{s \rightarrow 0} s \left[ \frac{0.1/s^2}{1 + \frac{10K}{s(\tau s + 1)}} \right]$$

$$E(s) = R(s) - Y(s) = \left( 1 - \frac{KG(s)}{1 + KG(s)} \right) R(s)$$

$$E(s) = \frac{R(s)}{1 + KG(s)}$$

$$K_v = \lim_{s \rightarrow 0} s G(s) = \lim_{s \rightarrow 0} \frac{s \times 10K}{s(\tau s + 1)} = 10K$$

$$e_{ss} = \frac{0.1}{K_v} = \frac{0.1}{10K}$$

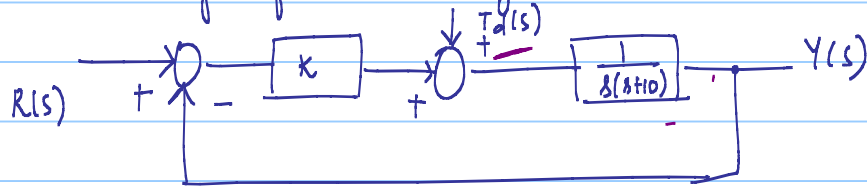
$$K_v = 10K \Rightarrow e_{ss} = \frac{0.1}{K_v} = \frac{0.1}{10K}$$

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$$0.1 \times 10^{-3} = \frac{0.1}{10^4}$$

$$K = 100$$

3. Consider a unity feedback system.



Let  $K = 120$

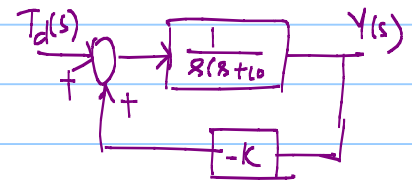
a) Calculate  $e_{ss}$  of the closed loop due to unit step input  $T_d(s) = 0$

b) Calculate steady state response  $Y_{ss} = \lim_{t \rightarrow \infty} y(t)$  when  $T_d(s) = 1/s$

\*  $R(s) = 0$

$$T(s) = \left. \frac{Y(s)}{R(s)} \right|_{T_d(s)=0} = \frac{K}{s(s+10)+K}$$

$$\left. \frac{Y(s)}{T_d(s)} \right|_{R(s)=0} = \frac{1}{s^2+10s+K}$$



$$a) T_d(s) = 0 \quad e_{ss} = \lim_{s \rightarrow 0} s E(s) = \lim_{s \rightarrow 0} s \left( 1 - T(s) \right) R(s) \rightarrow 1/s$$

$\downarrow$   
 $\frac{R(s) - Y(s)}{R(s) - T(s)R(s)}$

$$e_{ss} = (1 - T(0)) = 1 - K/K = 0$$

$$b) R(s) = 0 \quad Y_{ss} = \lim_{s \rightarrow 0} s Y(s) = \lim_{s \rightarrow 0} s \times \frac{T_d(s)}{s^2+10s+K} = \lim_{s \rightarrow 0} s \times \frac{1/s}{s^2+10s+K}$$

$$Y_{ss} = 1/K$$

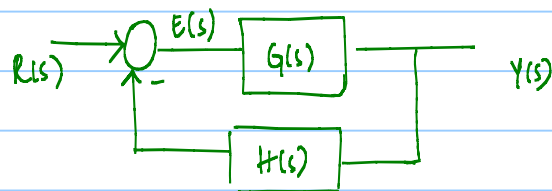
Error with respect to disturbance

$$E(s) = Y(s) \Big|_{R(s)=0}$$

$$e_{ss} = \lim_{s \rightarrow 0} s E(s) = \lim_{s \rightarrow 0} s Y(s) = Y_{ss} = 1/K$$

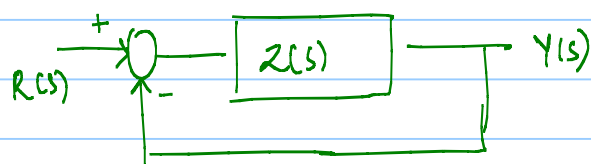
Increase 'K' sufficiently large such that error with respect to disturbance  $T_d$  is small.

Steady state error:  $e_{ss}$



Non unity feedback system.

$$\frac{Y(s)}{R(s)} = \frac{G(s)}{1 + G(s)H(s)}$$



$$\frac{Y(s)}{R(s)} = \frac{Z(s)}{1 + Z(s)}$$

$$\frac{Z(s)}{1 + Z(s)} = \frac{G(s)}{1 + G(s)H(s)}$$

$$\frac{G(s)}{1 + G(s)H(s)} = Z(s) \left( 1 - \frac{G(s)}{1 + G(s)H(s)} \right)$$

$$\Rightarrow Z(s) = \frac{G(s)}{1 + G(s)H(s) - G(s)}$$

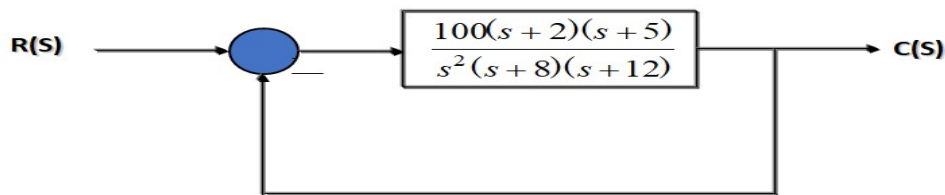
$$\begin{aligned} E(s) &= R(s) - Y(s) \\ &= R(s) \cdot \left( 1 - \frac{Y(s)}{R(s)} \right) \end{aligned}$$

Let  $\frac{Y(s)}{R(s)} = T(s)$  — Closed loop transfer function

the steady state error,

$$e_{ss} = \lim_{t \rightarrow \infty} e(t) = \lim_{s \rightarrow 0} s E(s)$$

- For the system shown in figure below evaluate the static error constants and find the expected steady state errors for the standard step, ramp and parabolic inputs.



$$K_p = \lim_{s \rightarrow 0} G(s)$$

$$K_p = \lim_{s \rightarrow 0} \left( \frac{100(s+2)(s+5)}{s^2(s+8)(s+12)} \right)$$

$$K_p = \infty$$

$$K_v = \lim_{s \rightarrow 0} sG(s)$$

$$K_v = \infty$$

$$K_a = \lim_{s \rightarrow 0} s^2 G(s)$$

$$K_a = \lim_{s \rightarrow 0} \left( \frac{100s^2(s+2)(s+5)}{s^2(s+8)(s+12)} \right)$$

$$K_a = \left( \frac{100(0+2)(0+5)}{(0+8)(0+12)} \right) = 10.4$$

$$e_{ss} = \frac{1}{1 + K_p} = 0$$

$$e_{ss} = \frac{1}{K_v} = 0$$

$$e_{ss} = \frac{1}{K_a} = 0.09$$

Determine the static error constants of the system represented by the OLTF with unity feedback

$$G(s) = \frac{k(s+2)}{s(s^2+7s^2+12s)}$$

Also determine the type & order of the system. Find the  $e_{ss}$  for a unit parabolic input.

Sol: type = 2, order = 4

$$K_p = \lim_{s \rightarrow 0} G(s) = \infty$$

$$K_v = \lim_{s \rightarrow 0} sG(s) = \infty$$

$$K_a = \lim_{s \rightarrow 0} s^2 G(s) = \frac{2k}{12} = k/6$$

$$e_{ss} = \frac{1}{K_a} = \frac{1}{k/6} = \frac{6}{k}$$

Find the static error constants for a system

$$G(s) = \frac{s+10}{s(s^3+5s^2+15s)}$$

assuming UFB, also calculate  $e_{ss}$  when  $r(t) = 2t u(t)$  &  $r(t) = 4t^3 u(t)$

Sol: type = 2, order = 4

$$L[t^n] = \frac{n!}{s^{n+1}}$$

$$K_p = \lim_{s \rightarrow 0} G(s) = \infty$$

$$K_v = \lim_{s \rightarrow 0} sG(s) = \infty$$

$$K_a = \lim_{s \rightarrow 0} s^2 G(s) = \frac{10}{15} = \frac{2}{3}$$

$$R(s) = \frac{2}{s^2}$$

$$e_{ss} = \frac{2}{K_v} = \frac{2}{\infty} = 0$$

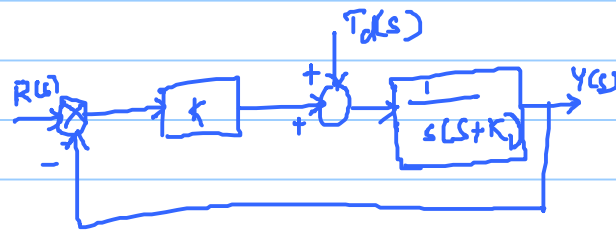
$$R(s) = \frac{8}{s^3}$$

$$e_{ss} = \frac{8}{K_a} = \frac{8}{\frac{2}{3}} = \underline{\underline{12}}$$

$$e_{ss} = \lim_{s \rightarrow 0} s \cdot E(s) = \lim_{s \rightarrow 0} s \cdot \frac{R(s)}{1+G(s)}$$



- Consider the unity feedback system shown. The system has two parameters, the controller gain  $K$  and the constant  $K_1$  in the process.
- A) calculate the sensitivity of CLTF to changes in  $K_1$



sol: To find CLTF,  $T_d(s) = 0$

$$T(s) = \frac{Y(s)}{R(s)} = \frac{\frac{K}{s(s+K_1)}}{1 + \frac{K}{s(s+K_1)}}$$

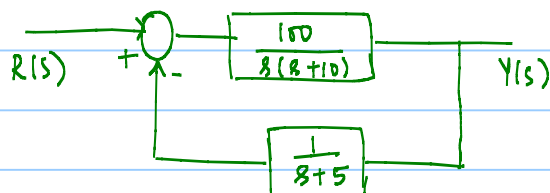
$$= \frac{K}{s^2 + K_1 s + K}$$

$$S_{K_1}^T = \frac{\partial T}{\partial K_1} \cdot \frac{K_1}{T}$$

$$= \frac{-K \cdot s}{(s^2 + K_1 s + K)^2} \cdot \frac{K_1}{\frac{K}{s^2 + K_1 s + K}}$$

$$= \frac{-s K_1}{s^2 + K_1 s + K}$$

make  $K$  as large as possible, To use  $S_{K_1}^T$



Determine the type, order, error constants &  $e_{ss}$  for unit step input

$$G(s) = \frac{100}{s(s+10)} \quad H(s) = \frac{1}{s+5}$$

$$Z(s) = \frac{G}{1 + GH - G}$$

$$Z(s) = \frac{100/s(s+10)}{1 + \frac{100}{s(s+10)} \times \frac{1}{s+5}} = \frac{100}{s(s+10)}$$

① LTF  $Z(s) = \frac{100(s+5)}{s^3 + 15s^2 - 50s - 400}$

type - 0

CLTF  $T(s) = \frac{Z}{1+Z}$

$$= \frac{100(s+5)}{(s^3 + 15s^2 - 50s - 400) + (100s + 500)}$$

Order - 3

$$K_p = \lim_{s \rightarrow 0} Z(s)$$

$$= -5/4$$

$$K_v = \lim_{s \rightarrow 0} s Z(s)$$

$$= 0$$

$$K_a = \lim_{s \rightarrow 0} s^2 Z(s)$$

$$= 0$$

for step  $e_{ss} = \frac{1}{1+K_p}$

$$e_{ss} = \frac{4}{4-5} = -4$$

