

# Mechanics of a Rocket

## Calculating Max Q

James Saslow - 2/1/2023

(\*Everything is in SI units\*)

(\*Falcon 9 Specs/Estimates\*)

$$A = \pi (3.7 / 2)^2;$$

$$C_d = 0.25;$$

$$m_0 = 25600;$$

$$\mu = (395700 / m_0);$$

$$\tau = 205;$$

$$v_e = 5630;$$

(\*Fundamental Constants\*)

$$\alpha = 9.8 / 1000;$$

$$M = 0.028;$$

$$R = 8.314;$$

$$\rho_0 = 1.225;$$

$$T_0 = 293;$$

$$g = 9.81;$$

```
In[ ]:= (*s=NDSolve[{x'[t]+x[t]==0,x[0]==0,x'[0]==1},x[t],{t,0,10}]
Plot[x[t]/.s,{t,0,10}]*)
```


```
tmax = 90;
```

```
s =
```

```
NDSolve[{ $\frac{m_0 v_e}{\tau} \text{Exp}[-t/\tau] - m_0 g (1 + \text{Exp}[-t/\tau]) - \frac{1}{2} \rho_0 (1 - \alpha x[t]/T_0)^{\frac{m_B}{\alpha R}-1} (x'[t])^2 C_d A =$ 
 $m_0 (1 + \text{Exp}[-t/\tau]) x''[t]$ , x[0] == 0, x'[0] == 0}, x, {t, 0, tmax}]
```

```
Plot[Evaluate[x[t] /. s], {t, 0, tmax}, PlotRange -> All]
```

```
Plot[Evaluate[ $\frac{1}{2} C_d A \rho_0 (1 - \alpha x[t]/T_0)^{\frac{m_B}{\alpha R}-1} (x'[t])^2 / 101325 /. s$ ],
{t, 0, tmax}, PlotRange -> All]
```

```
Out[ ]:= {{x -> InterpolatingFunction[ Domain: {{0., 90.}}
Output: scalar ]}}
```

