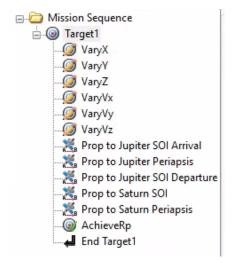
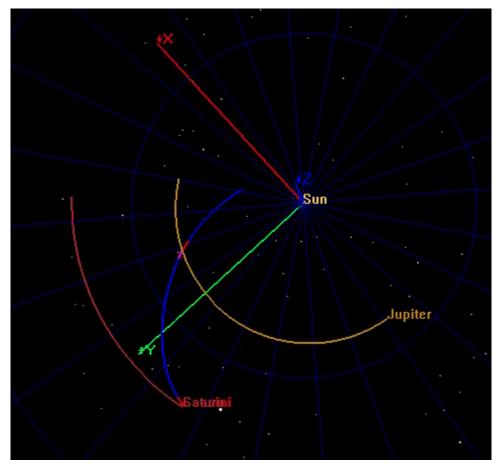
Problem 2

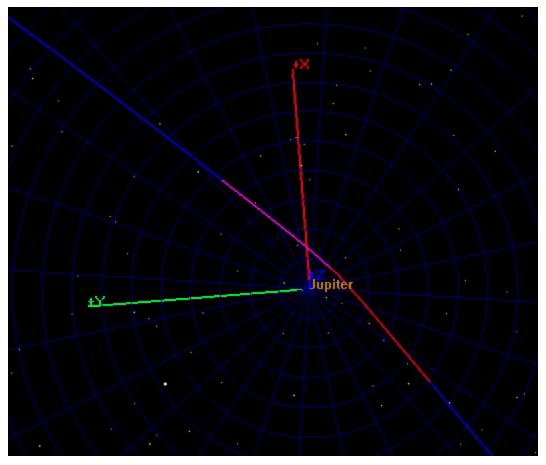
Part a



Mission Sequence Screenshot



Heliocentric View of Trajectory (Blue)



Jupiter-Centered View of Trajectory (Blue)

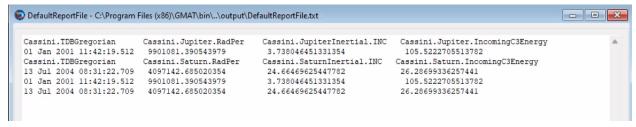
Part b

- The report lists the radius of periapsis as 9,901,081.3905 km. It is given that the radius of Jupiter is 71,492 km. So, the altitude at periapsis is 9,829,589.3905 km
- The epoch in Gregorian TDB date format is 01 Jan 2001 11:42:19.512.
- o The inclination is 3.7380 deg
- The v-infinity along the hyperbola is square root of Incoming C3.
 v-infinity is 10.2724 km/s

Part c

- The report lists the radius of periapsis as 4,097,142.6850 km. It is given that the radius of Jupiter is 60,268 km. So, the altitude at periapsis is 4,036,874.685 km
- The epoch in Gregorian TDB date format is 13 Jul 2004 08:31:22.709.

- The inclination is 24.6647 deg
- The v-infinity along the hyperbola is square root of Incoming C3.
 v-infinity is 5.1271 km/s
- The report file is shown below



Part d

- As seen above the radius of periapsis is not 81,000 km and the date of arrival is not July 1, 2004.
- This may be because the GMAT simulation doesn't model any trajectory correction maneuvers that actual Cassini spacecraft may have conducted before it reached Saturn.

Part e

```
*** Targeting Completed in 8 iterations.

*** The Targeter converged!

Final Variable values:

    Cassini.X = 298694286.772
    Cassini.Y = 201077211.165
    Cassini.Z = -4192849.59156
    Cassini.VX = 5.28832476238
    Cassini.VY = 22.0972626796
    Cassini.VZ = -0.262665585358

Mission run completed.

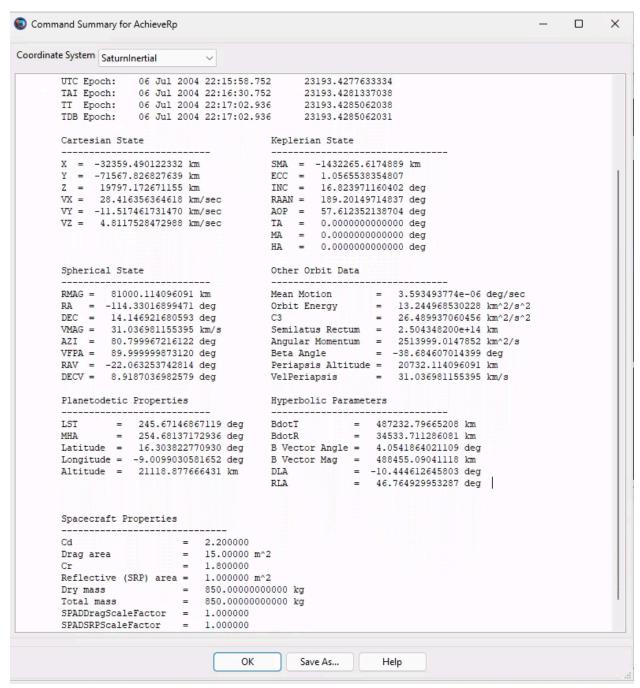
===> Total Run Time: 118.621 seconds
```

New Initial State Vector

- The targeter required 8 iterations to get to these values.
- Difference in state vector
 - X 0.0000575 km
 - Y 0.0001812 km
 - Z 0.000004223 km
 - Vx 0.00156924506263 km/s
 - Vy 0.00436424588517 km/s
 - Vz 0.00010459917061 km/s

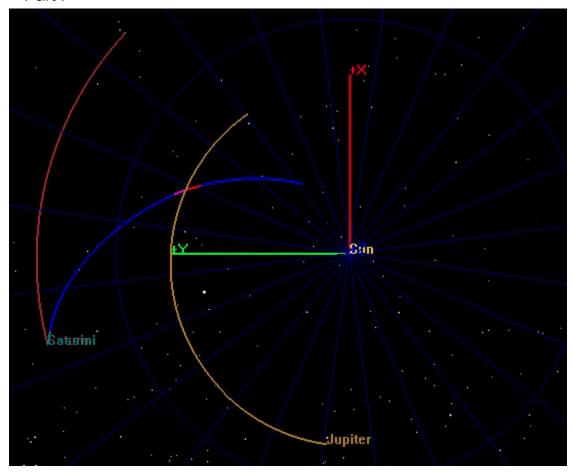
 Plot below shows the achieved radius of periapsis - 81000.114096 km. The plot below that also shows the radius of periapsis (RMAG)

Control Variable	Current Value	Last Value	Difference
Cassini.X	298694286.7720575	298694286.7720575	0
Cassini.Y	201077211.1651812	201077211.1651812	0
Cassini.Z	-4192849.591564223	-4192849.591564223	0
Cassini.VX	5.28832476237644	5.28832476237644	0
Cassini.VY	22.09726267962946	22.09726267962946	0
Cassini.VZ	-0.2626655853580731	-0.2626655853580731	0
Constraints	Desired	Achieved	Difference
(==) Cassini.Saturn.RadPer	81000	81000.11409609072	0.1140960907214321

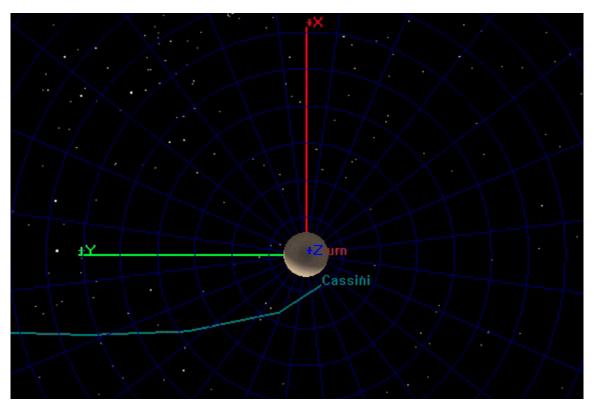


 Such a small change in the initial state vector caused such a big change because the change in initial positions changed the position of Jupiter and Saturn that provided the gravity assists. Their change in position changed the trajectory slightly, but at a solar system scale, even such a small change can drastically change the trajectory of the spacecraft. For perspective, the initial radius of periapsis (4e6 km in part c) is ~0.2% of Saturn's orbital radius. Hence, small initial changes cause big ripples.

• Part f



Heliocentric View of Trajectory (Blue)



Saturn-Centered View of Trajectory (Blue)