

ASEN 5044, Fall 2024

Statistical Estimation for Dynamical Systems

Lecture 05: LTI IVPs Wrap Up; Nonlinear State Space Systems and Linearization

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Tuesday 09/10/2024



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Announcements

- **Prof. Ahmed out of country next week (SPIE Defense & Security Conference in UK)**
 - **No live classes next Tues 09/17 or Thurs 09/19, BUT pre-recorded Lecture Videos to be posted – WATCH THEM!!** (will need them for HW 2 and Quiz)
 - **Also: TF Aidan will cover Prof. Ahmed's hours next Wed 09/18 4:30-6 pm, AERO N353**
- **HW 1 Due Fri 09/13 at 11:59 pm**
- **Submit to Gradescope (linked via Canvas) –**
 - **All submissions must be legible!!! – zero credit otherwise**
- **Advanced Questions: these are optional/extra credit (follow instructions)**
- **HW 2 will be posted Thurs 09/12, due Fri 09/16.**
- **Quiz 2 solutions to be posted later today**
- **Quiz 3: this Friday-Tuesday via Canvas**
- **Office hours this week (in person + Zoom):**
 - **Prof. Ahmed: Wed, 4:30-6 pm, AERO N353**
 - **TF Aidan Bagley: Wed 12-1:30 pm, AERO N353**
 - **TF Jiho Lee: Tues 2:30-3:30 pm, AERO N253**
 - **TF Collin Hudson: Mon 1-2 pm, AERO 303**

Overview

Last time: Unforced LTI State Space IVP Solutions and the Matrix Exponential

$$\dot{x}(t) = Ax(t), x(t_0) = x_0 \longrightarrow x(t) = \underline{e^{A(t-t_0)}} \cdot x(t_0)$$
$$e^{A(t-t_0)} \triangleq \sum_{i=0}^{\infty} \frac{A^i (t-t_0)^i}{i!} = I + A(t-t_0) + \frac{A^2 (t-t_0)^2}{2!} + \dots$$

Today:

- Wrap up general LTI SS IVP solutions with forcing inputs
- Choice and Transformation of State Representations
- Nonlinear systems and “standard form” nonlinear state space models
- Linearization and transformation to linear SS models

READ: Chapter 1.6-1.8 in Simon book

General Solution to Forced LTI Matrix-Vector IVPs

- Recall: General LTI state space model with inputs given by

$$\dot{x} = Ax(t) + Bu(t), \quad x(t_0) = x_0, \quad u(t) \neq 0 \text{ for } t \geq 0$$

where $x(t) \in \mathbb{R}^n$, $u(t) \in \mathbb{R}^m$, $A \in \mathbb{R}^{n \times n}$, $B \in \mathbb{R}^{n \times m}$

- If non-zero $u(t)$ and initial condition $x(t_0)$, then general LTI solution is:

⊛
$$x(t) = \underbrace{e^{A(t-t_0)} x(t_0)}_{\substack{\text{unforced response} \\ \text{("free response") due to IC's}}} + \underbrace{\int_{t_0}^t \underline{e^{A(t-\tau)}} \underline{B} \underline{u}(\tau) d\tau}_{\substack{\text{forced response due to input (o IC's)} \\ \text{[convolution integral]}}}$$

"l.s.m" in matlab ←

(note: for LTV systems, general solution is: $x(t) = \underbrace{\Phi(t, t_0)}_{STM} x(t_0) + \int_{t_0}^t \underbrace{\Phi(t, \tau)}_{STM} B(\tau) u(\tau) d\tau$)

Choice and Transformation of State Representations

- The $[A,B,C,D]$ matrices are not unique for given set of linear ODEs
- Infinitely many possible $[A,B,C,D]$ -- governed by choice of state x
- These choices are all related by invertible similarity transformations
- Example for 1D mass system again: $x = \begin{bmatrix} v \\ d \end{bmatrix}$, $A = \begin{bmatrix} 0 & 0 \\ 1 & 0 \end{bmatrix}$, $\dot{x} = Ax$ (unforced)

→ Suppose we want to change basis to get new state vector $\tilde{x} = \begin{bmatrix} d \\ v \end{bmatrix}$

→ clearly, $\tilde{x} = Tx$ where $T = \begin{bmatrix} 0 & 1 \\ 1 & 0 \end{bmatrix}$ [invertible similarity transform]

→ so $\dot{\tilde{x}} = \frac{d}{dt}(Tx) \stackrel{\text{(product rule)}}{=} T\dot{x} + \frac{d}{dt}(T)x = T(Ax) \rightarrow \dot{\tilde{x}} = TAX$

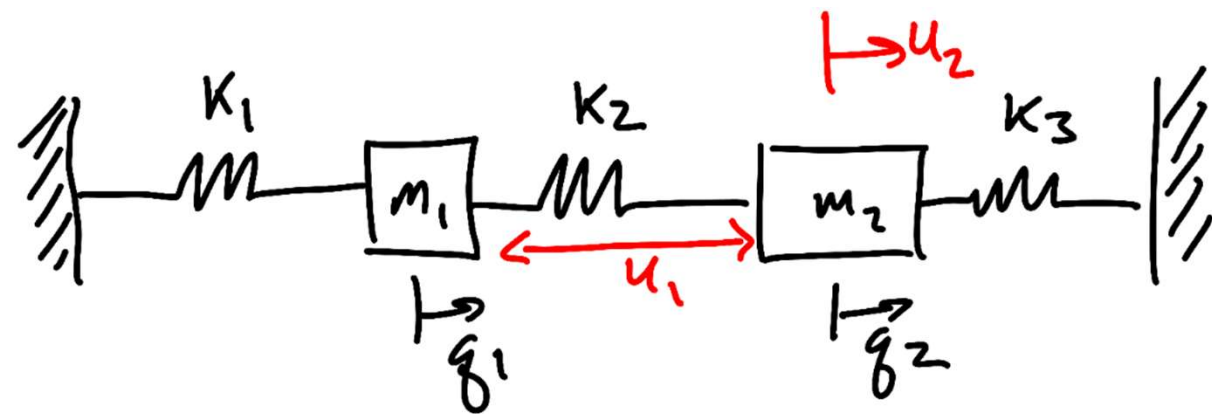
But since $\tilde{x} = Tx \rightarrow x = T^{-1}\tilde{x} \rightarrow \dot{\tilde{x}} = \underbrace{TAT^{-1}}_{\substack{2 \times 2 \text{ Sq. matrix} \\ \rightarrow \text{call it } \tilde{A}}} \tilde{x} = \tilde{A}\tilde{x} \Rightarrow \boxed{\dot{\tilde{x}} = \tilde{A}\tilde{x}}$
 where $\tilde{A} = TAT^{-1}$
 (holds for any similarity transform)

* Note: by similar reasoning: can be generalized to
 $B \rightarrow \tilde{B}$, $C \rightarrow \tilde{C}$, $D \rightarrow \tilde{D}$.

Also: note that y & u Do NOT Transform under T - only the internal ^{state} variables in x do!

Linear vs. Nonlinear System Models

- Linear dynamics/ODEs = good approx. for many physical laws, but not all!
- Example: 2 mass / 3 spring system
- Physical springs and actuators always have nonlinear behavior – but sometimes we can ignore these for *a priori*/first principles models in control/estimation



$$x = [q_1(t), \dot{q}_1(t), q_2(t), \dot{q}_2(t)]^T$$

$$u = [u_1(t), u_2(t)]^T$$

$$y = [q_1(t), q_2(t)]^T$$

u_1 = relative actuator;
 u_2 = absolute actuator

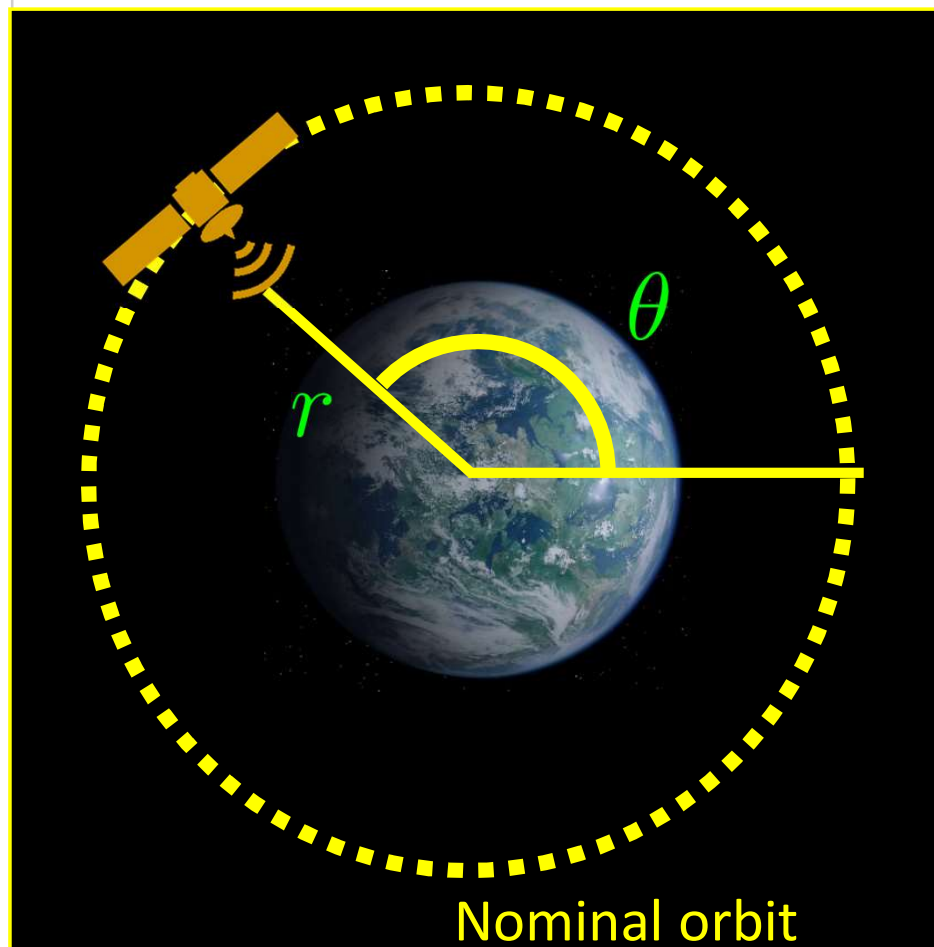
- For $k_1 = k_2 = k_3 = 1$ N/m and $m_1 = m_2 = 1$ kg, can use basic physics to get eqs. of motion and express as LTI SS model:

$$\dot{x} = Ax(t) + Bu(t) \quad A = \begin{bmatrix} 0 & 1.0 & 0 & 0 \\ -2.0 & 0 & 1.0 & 0 \\ 0 & 0 & 0 & 1.0 \\ 1.0 & 0 & -2.0 & 0 \end{bmatrix} \quad B = \begin{bmatrix} 0 & 0 \\ -1 & 0 \\ 0 & 0 \\ 1 & 1 \end{bmatrix}$$

$$y(t) = Cx(t) + Du(t) \quad C = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 \end{bmatrix} \quad D = \begin{bmatrix} 0 & 0 \\ 0 & 0 \end{bmatrix}$$

What to do about Nonlinear ODEs?

- But most systems have intrinsically nonlinear effects that are not obviously/easily modeled by linear physical relationships
- **What if a priori/first principles give nonlinear (NL) dynamics?**
- Example: equation for orbit plane motion of satellite



$$\ddot{r} - \dot{\theta}^2 r = -\frac{\mu}{r^2} + a_r$$
$$r\ddot{\theta} + 2\dot{r}\dot{\theta} = a_i$$

a_i, a_r : in-track and radial accelerations
[due to thrust, drag, SRP, grav anomalies, etc.]
 r, θ : states for spacecraft

NL systems can get very nasty and weird...

- For now, only focus on NL sys with “sufficiently smooth” nonlinearities

- i.e. derivatives exist for state vars and are bounded

Lipschitz Continuous: for some function $f(x)$, $\|f(x_1) - f(x_2)\| \leq c \cdot \|x_1 - x_2\|$, $\forall x_1, x_2$ “sufficiently close”
for some constant c (c : Lipschitz constant)

- Most cases: want to keep NL sys near operating point/condition

- Equilibrium: set of x and u such that $\dot{x} = 0$

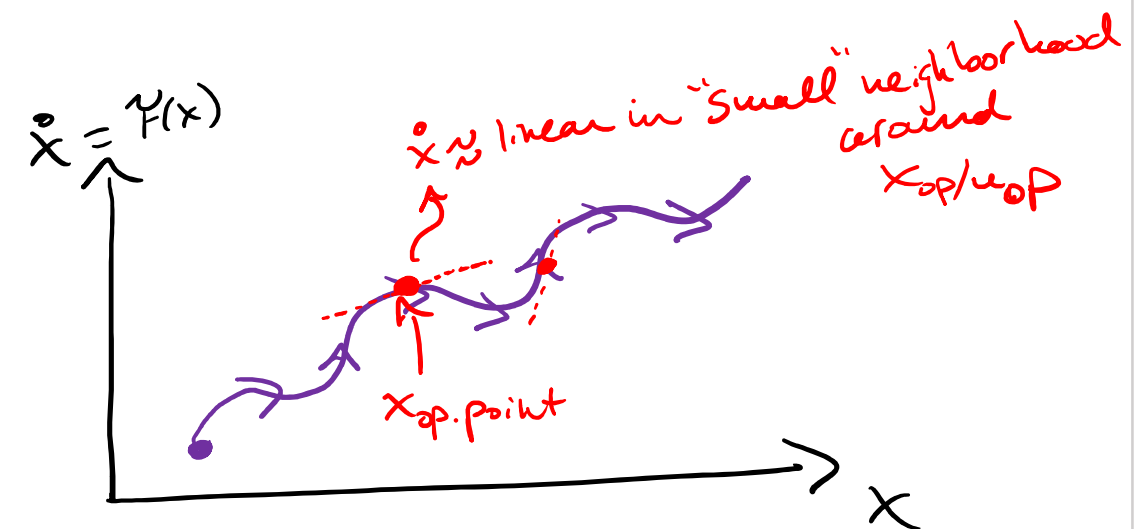
(i.e. some $\{x_{eq}, u_{eq}\}$ exists s.t. $\dot{x} = \mathcal{F}(x, u)|_{x_{eq}, u_{eq}} = 0$)

- Nominal trajectory: known valid solutions $x_{nom}(t)$ and $u_{nom}(t)$ to the nonlinear vector ODE $\dot{x} = \mathcal{F}(x, u)$,

i.e. such that $\dot{x} = \mathcal{F}(x, u)|_{x_{nom}(t), u_{nom}(t)}$ ~~linear~~

- Can look at dynamics of “small” perturbations near op pt

- If perturbations small enough, system behaves (almost) linearly!



Linearization of NL ODEs via Multivariable Taylor Series

- Idea: express NL ODEs in (non-linear) standard state vector form \rightarrow do Taylor expansion near operating point \rightarrow drop higher order terms (HOTs) \rightarrow re-arrange to linear SS model

Given set of nonlinear ODEs, express in standard nonlinear state space form by picking state vars & stacking into a vector expressed w/ time derivatives as follows:

$$\textcircled{*} \dot{x} = \begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \vdots \\ \dot{x}_n \end{bmatrix} = \begin{bmatrix} f_1(x,u,t) \\ f_2(x,u,t) \\ \vdots \\ f_n(x,u,t) \end{bmatrix} = f(x,u,t) \rightarrow \text{Stack (vector) of } n \text{ NL ODEs}$$

$$\textcircled{*} y = \begin{bmatrix} y_1 \\ \vdots \\ y_p \end{bmatrix} = \begin{bmatrix} h_1(x,u,t) \\ \vdots \\ h_p(x,u,t) \end{bmatrix} = h(x,u,t) \rightarrow \text{Stack (vector) of } p \text{ NL algebraic equations}$$

$$\text{Also, let } u = \begin{bmatrix} u_1(t) \\ \vdots \\ u_m(t) \end{bmatrix} \text{ (input vector).}$$

Linearization (cont'd)

- Suppose a nominal solution or operating point is known/given, i.e. $x_{nom}(t)$ & $u_{nom}(t)$ given

Define slight perturbations from nom. op. point:

$$\text{perturbation vectors } \begin{cases} \delta x(t) \triangleq x(t) - x_{nom}(t) \longrightarrow x(t) = x_{nom}(t) + \underline{\delta x}(t) \\ \delta u(t) \triangleq u(t) - u_{nom}(t) \longrightarrow u(t) = u_{nom}(t) + \underline{\delta u}(t) \end{cases}$$

→ Plug in expressions for $x(t)$ & $u(t)$ into NL ODES & do vector Taylor Series expansion:

$$\dot{x}(t) = \tilde{F}(x, u, t) \iff \dot{x}(t) = \dot{x}_{nom}(t) + \delta \dot{x}(t) \iff \tilde{F}(x, u, t) = \tilde{F}(x_{nom} + \delta x, u_{nom} + \delta u, t)$$

→ ^(vector) Taylor Series Exp.: $\dot{x}(t) = \tilde{F}(x_{nom} + \delta x, u_{nom} + \delta u, t)$

$$= \tilde{F}(x_{nom}, u_{nom}, t) + \left[\frac{\partial \tilde{F}}{\partial x} \right]_{x_{nom}, u_{nom}} \delta x + \left[\frac{\partial \tilde{F}}{\partial u} \right]_{x_{nom}, u_{nom}} \delta u + \text{HOTS (higher order terms)}$$

→ Likewise: $y(t) = y_{nom} + \delta y = h(x_{nom} + \delta x, u_{nom} + \delta u, t)$

$$= h(x_{nom}, u_{nom}, t) + \left[\frac{\partial h}{\partial x} \right]_{x_{nom}, u_{nom}} \delta x + \left[\frac{\partial h}{\partial u} \right]_{x_{nom}, u_{nom}} \delta u + \text{HOTS}$$

Linearization (cont'd)

- Partial derivative matrices = **Jacobians** w.r.t. x and u

$$\left[\frac{\partial \mathcal{F}}{\partial x} \right] \Big|_{x_{nom}, u_{nom}} = \begin{bmatrix} \frac{\partial \mathcal{F}_1}{\partial x_1} & \frac{\partial \mathcal{F}_1}{\partial x_2} & \cdots & \frac{\partial \mathcal{F}_1}{\partial x_n} \\ \vdots & \vdots & \cdots & \vdots \\ \frac{\partial \mathcal{F}_n}{\partial x_1} & \frac{\partial \mathcal{F}_n}{\partial x_2} & \cdots & \frac{\partial \mathcal{F}_n}{\partial x_n} \end{bmatrix} \Big|_{x_{nom}, u_{nom}} \quad \rightarrow \boxed{n \times n \text{ matrix}}$$

$$\left[\frac{\partial \mathcal{F}}{\partial u} \right] \Big|_{x_{nom}, u_{nom}} = \begin{bmatrix} \frac{\partial \mathcal{F}_1}{\partial u_1} & \frac{\partial \mathcal{F}_1}{\partial u_2} & \cdots & \frac{\partial \mathcal{F}_1}{\partial u_m} \\ \vdots & \vdots & \cdots & \vdots \\ \frac{\partial \mathcal{F}_n}{\partial u_1} & \frac{\partial \mathcal{F}_n}{\partial u_2} & \cdots & \frac{\partial \mathcal{F}_n}{\partial u_m} \end{bmatrix} \Big|_{x_{nom}, u_{nom}} \quad \rightarrow \boxed{n \times m}$$

(similarly for $\frac{\partial h}{\partial x} \Big|_{nom}$ and $\frac{\partial h}{\partial u} \Big|_{nom}$)
 $[p \times n]$
 $[p \times m]$

For small enough δx , δu , can neglect HOTS

- Get linearized eqs for dynamics of perturbations δx , δy w.r.t. δu "near" nominal op point:

$$(i) \quad \dot{x}(t) = \dot{x}_{nom}(t) + \delta \dot{x} \approx \dot{f}(x_{nom}, u_{nom}, t) + \left. \frac{\partial f}{\partial x} \right|_{nom} \delta x + \left. \frac{\partial f}{\partial u} \right|_{nom} \delta u + \text{HOTS}$$

$$(ii) \quad y(t) = y_{nom}(t) + \delta y \approx h(x_{nom}, u_{nom}, t) + \left. \frac{\partial h}{\partial x} \right|_{nom} \delta x + \left. \frac{\partial h}{\partial u} \right|_{nom} \delta u + \text{HOTS}$$

→ left with

$$\delta \dot{x} = \left[\left. \frac{\partial f}{\partial x} \right|_{nom} \right] \delta x + \left[\left. \frac{\partial f}{\partial u} \right|_{nom} \right] \delta u$$

$$\delta y = \left[\left. \frac{\partial h}{\partial x} \right|_{nom} \right] \delta x + \left[\left. \frac{\partial h}{\partial u} \right|_{nom} \right] \delta u$$

Let

$$\bar{x} = \delta x$$

$$\bar{y} = \delta y$$

$$\bar{u} = \delta u$$

$$\bar{A}|_{nom} = \left. \frac{\partial f}{\partial x} \right|_{nom}$$

$$\bar{B}|_{nom} = \left. \frac{\partial f}{\partial u} \right|_{nom}$$

$$\bar{C}|_{nom} = \left. \frac{\partial h}{\partial x} \right|_{nom}$$

$$\bar{D}|_{nom} = \left. \frac{\partial h}{\partial u} \right|_{nom}$$

Perturbation Dynamics

valid only if δx & δu stay small!

Looks like SS $[A, B, C, D]$ matrices for linear system BUT Now depends on x_{nom} & u_{nom} !

$$\dot{\bar{x}} = \bar{A}|_{nom} \bar{x} + \bar{B}|_{nom} \bar{u}$$

$$\bar{y} = \bar{C}|_{nom} \bar{x} + \bar{D}|_{nom} \bar{u}$$

Example (please read on your own!):

2nd order NL ODE (no input/unforced)

$$\ddot{z} + (1 + z)\dot{z} - 2z + 0.5z^3 = 0 \quad (\Leftrightarrow \ddot{z} = 2z - 0.5z^3 - (1 + z)\dot{z})$$

Step 1: define state variables and put into standard NL SS form:

$$\begin{array}{l} x_1 = z \\ x_2 = \dot{z} \end{array} \rightarrow x = \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} \rightarrow \dot{x} = \begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} x_2 \\ 2x_1 - 0.5x_1^3 - (1 + x_1)x_2 \end{bmatrix} = \begin{bmatrix} \dot{z} \\ \ddot{z} \end{bmatrix}$$

$$\rightarrow \mathcal{F}(x) = \begin{bmatrix} \mathcal{F}_1 \\ \mathcal{F}_2 \end{bmatrix} = \begin{bmatrix} x_2 \\ 2x_1 - 0.5x_1^3 - (1 + x_1)x_2 \end{bmatrix} = \dot{x} \quad (\text{now in standard NL SS form})$$

$$\text{Define output } y = \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} = \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} \quad (\text{already linear – no linearization needed!})$$

→ Suppose we now linearize dynamics around *equilibrium points*...

Example (please read on your own!):

2nd order NL ODE (cont'd)

Step 2: look for eq. points to use as x_{nom} op. point:

Equilibrium points: solutions of $\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} |_{x_1, x_2} = \begin{bmatrix} 0 \\ 0 \end{bmatrix} \rightarrow \begin{bmatrix} \mathcal{F}_1(x) \\ \mathcal{F}_2(x) \end{bmatrix} |_{x_1, x_2} = \begin{bmatrix} x_2 \\ 2x_1 - 0.5x_1^3 - (1 + x_1)x_2 \end{bmatrix}$
(i.e. $\dot{x} = \mathcal{F}[x] = 0$)

→ So, we must have:

$$x_2 = 0$$

$$\text{and } 2x_1 - 0.5x_1^3 - (1 + x_1)x_2 = 0$$

→ Solve for the roots of the 2nd equation (since $x_2 = 0$ is known via first eq.)

→ Get 3 equilibrium points: $x_{eq,1} = \begin{bmatrix} 0 \\ 0 \end{bmatrix} \left(= \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} \right)$
 $x_{eq,2} = \begin{bmatrix} 2 \\ 0 \end{bmatrix}$
 $x_{eq,3} = \begin{bmatrix} -2 \\ 0 \end{bmatrix}$

Now we take Jacobians of NL ODE with respect to x and evaluate at these different x_{nom} operating points

Example (please read on your own!):

2nd order NL ODE (cont'd)

Step 3: Find Jacobians at x_{nom} points

$$\left[\frac{\partial \mathcal{F}}{\partial x} \right] \bigg|_{x_{nom}} = \begin{bmatrix} \frac{\partial \mathcal{F}_1}{\partial x_1} & \frac{\partial \mathcal{F}_1}{\partial x_2} \\ \frac{\partial \mathcal{F}_2}{\partial x_1} & \frac{\partial \mathcal{F}_2}{\partial x_2} \end{bmatrix} \bigg|_{x_{nom}} = \begin{bmatrix} 0 & 1 \\ (2 - \frac{3}{2}x_1^2 - x_2) & -(1 + x_1) \end{bmatrix} \bigg|_{x_{nom}}$$

$$\begin{aligned} \rightarrow \text{for } x_{eq,1} : \quad & \bar{A}_{1,nom} = \begin{bmatrix} 0 & 1 \\ 2 & -1 \end{bmatrix} \\ x_{eq,2} : \quad & \bar{A}_{2,nom} = \begin{bmatrix} 0 & 1 \\ -4 & -3 \end{bmatrix} \\ x_{eq,3} : \quad & \bar{A}_{3,nom} = \begin{bmatrix} 0 & 1 \\ -4 & 1 \end{bmatrix} \end{aligned}$$

3 different linearized ODEs
 $\dot{\bar{x}} = \bar{A}_i \bar{x}$ for $\bar{x} = \delta x$
for different eq. points $i = 1, 2, 3$

Example (please read on your own!):

2nd order NL ODE (cont'd)

Step 4: Put into LTI SS form: what is the state vector?

$$\bar{x} = \delta x = \begin{bmatrix} \delta x_1 \\ \delta x_2 \end{bmatrix} = \text{perturbation state vector } (\neq \text{total state vector } x(t)!!)$$

→ What is actual total state of NL system at any time (w.r.t. op. pt.)?

$$\begin{aligned} x(t) &= x_{nom}(t) + \delta x(t) \\ &= \begin{bmatrix} x_{nom,1}(t) \\ x_{nom,2}(t) \end{bmatrix} + \begin{bmatrix} \delta x_1 \\ \delta x_2 \end{bmatrix} \\ &= \begin{bmatrix} z_{nom,i}(t) \\ \dot{z}_{nom,i}(t) \end{bmatrix} + \begin{bmatrix} \delta z \\ \delta \dot{z} \end{bmatrix} \\ &\quad (\text{for eq. pt. } i \text{ in example}) \end{aligned}$$

