

ASEN 5044 - HW 7, Fall 2024, Jash Bhalavat

ASEN 5044
Fall 2024
Jash Bhalavat

①

HW 7

Problem 1 →

$$\ddot{\xi} = -\Omega \eta, \quad \ddot{\eta} = \Omega \xi$$

$$x = \begin{bmatrix} \xi \\ \dot{\xi} \\ \eta \\ \dot{\eta} \end{bmatrix} \rightarrow \ddot{x} = \begin{bmatrix} \ddot{\xi} \\ \ddot{\dot{\xi}} \\ \ddot{\eta} \\ \ddot{\dot{\eta}} \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & -\Omega \\ 0 & 0 & 0 & 1 \\ 0 & \Omega & 0 & 0 \end{bmatrix} \begin{bmatrix} \xi \\ \dot{\xi} \\ \eta \\ \dot{\eta} \end{bmatrix} = A x$$

confirming → $e^{A\Delta t} = I + A\Delta t + \frac{A^2}{2} \Delta t^2 + \dots$

$$A^2 = \begin{bmatrix} 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & -\Omega \\ 0 & 0 & 0 & 1 \\ 0 & \Omega & 0 & 0 \end{bmatrix} \begin{bmatrix} 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & -\Omega \\ 0 & 0 & 0 & 1 \\ 0 & \Omega & 0 & 0 \end{bmatrix} = \begin{bmatrix} 0 & 0 & 0 & -\Omega \\ 0 & -\Omega^2 & 0 & 0 \\ 0 & \Omega & 0 & 0 \\ 0 & 0 & 0 & -\Omega^2 \end{bmatrix} \quad A^4 = \begin{bmatrix} 0 & 0 & 0 & \Omega^3 \\ 0 & \Omega^4 & 0 & 0 \\ 0 & -\Omega^3 & 0 & 0 \\ 0 & 0 & 0 & \Omega^4 \end{bmatrix}$$

$$A^3 = \begin{bmatrix} 0 & 0 & 0 & -\Omega \\ 0 & -\Omega^2 & 0 & 0 \\ 0 & \Omega & 0 & 0 \\ 0 & 0 & 0 & -\Omega^2 \end{bmatrix} \begin{bmatrix} 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & -\Omega \\ 0 & 0 & 0 & 1 \\ 0 & \Omega & 0 & 0 \end{bmatrix} = \begin{bmatrix} 0 & -\Omega^2 & 0 & 0 \\ 0 & 0 & 0 & \Omega^3 \\ 0 & 0 & 0 & -\Omega^2 \\ 0 & -\Omega^3 & 0 & 0 \end{bmatrix}$$

$$e^{A\Delta t} = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix} + \Delta t \begin{bmatrix} 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & -\Omega \\ 0 & 0 & 0 & 1 \\ 0 & \Omega & 0 & 0 \end{bmatrix} + \frac{\Delta t^2}{2} \begin{bmatrix} 0 & 0 & 0 & -\Omega \\ 0 & -\Omega^2 & 0 & 0 \\ 0 & \Omega & 0 & 0 \\ 0 & 0 & 0 & -\Omega^2 \end{bmatrix} + \frac{\Delta t^3}{3!} \begin{bmatrix} 0 & -\Omega^2 & 0 & 0 \\ 0 & 0 & 0 & \Omega^3 \\ 0 & 0 & 0 & -\Omega^2 \\ 0 & -\Omega^3 & 0 & 0 \end{bmatrix} + \dots$$

$$e^{A\Delta t} = \begin{bmatrix} 1 & \Delta t - \frac{\Omega^2}{3!} \Delta t^3 + \dots & 0 & -\Omega \frac{\Delta t^2}{2} + \Omega^3 \frac{\Delta t^4}{4!} + \dots \\ 0 & 1 - \frac{\Omega^2}{2} \Delta t^2 + \dots & 0 & -\Omega \Delta t + \frac{\Omega^3}{3!} \Delta t^3 + \dots \\ 0 & \frac{\Omega}{2} \Delta t^2 - \frac{\Omega^3}{4!} \Delta t^4 + \dots & 1 & \Delta t - \frac{\Omega^2}{3!} \Delta t^3 + \dots \\ 0 & \Omega \Delta t - \frac{\Omega^3}{3!} \Delta t^3 + \dots & 0 & 1 - \frac{\Omega^2}{2} \Delta t^2 + \frac{\Omega^4}{4!} \Delta t^4 + \dots \end{bmatrix}$$

$$\sin(\Omega \Delta t) = \Omega \Delta t - \frac{\Omega^3 \Delta t^3}{3!} + \dots$$

$$\cos(\Omega \Delta t) = 1 - \frac{\Omega^2 \Delta t^2}{2!} + \frac{\Omega^4 \Delta t^4}{4!} - \dots$$

$$e^{A\Delta t} = \begin{bmatrix} 1 & \frac{\sin(\Omega \Delta t)}{\Omega} & 0 & -\frac{1 - \cos(\Omega \Delta t)}{\Omega} \\ 0 & \cos(\Omega \Delta t) & 0 & -\sin(\Omega \Delta t) \\ 0 & \frac{1 - \cos(\Omega \Delta t)}{\Omega} & 1 & \frac{\sin(\Omega \Delta t)}{\Omega} \\ 0 & \sin(\Omega \Delta t) & 0 & \cos(\Omega \Delta t) \end{bmatrix} \quad \therefore A = \begin{bmatrix} 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & -\Omega \\ 0 & 0 & 0 & 1 \\ 0 & \Omega & 0 & 0 \end{bmatrix}$$

$$x_{k+1} = f x_k \quad \text{where } f = e^{A\Delta t}$$

$$\begin{bmatrix} \xi_{k+1} \\ \dot{\xi}_{k+1} \\ \eta_{k+1} \\ \dot{\eta}_{k+1} \end{bmatrix} = \begin{bmatrix} 1 & \sin(\Omega \Delta t)/\Omega & 0 & -(1-\cos(\Omega \Delta t))/\Omega \\ 0 & \cos(\Omega \Delta t) & 0 & -\sin(\Omega \Delta t) \\ 0 & -(1-\cos(\Omega \Delta t))/\Omega & 1 & \sin(\Omega \Delta t)/\Omega \\ 0 & \sin(\Omega \Delta t) & 0 & \cos(\Omega \Delta t) \end{bmatrix} \begin{bmatrix} \xi_k \\ \dot{\xi}_k \\ \eta_k \\ \dot{\eta}_k \end{bmatrix}$$

Problem 2 $\rightarrow \Delta t = 0.5 \text{ s}, \Omega = 0.045 \frac{\text{rad}}{\text{s}}, x(0) \sim \mathcal{N}(\mu(0), P(0))$

a) $x_1 = Fx_0 \rightarrow x_2 = Fx_1 = F(Fx_0) = F^2x_0 \rightarrow x_k = F^kx_0$

For $k=0, 2, 3, \dots, N$, the recursive formula to predict the mean of linear Gauss

lecture 24 dynamical system into the future is:

slides 9-11

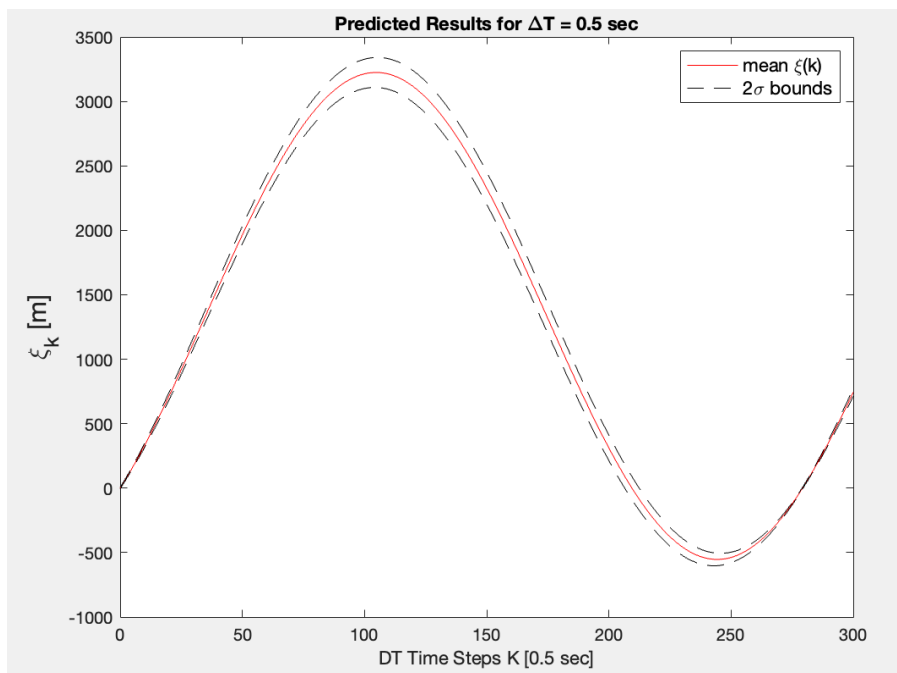
$$\mu(k+1) = F\mu(k) + G\mu(k) \stackrel{0}{=} F\mu_k = F^k\mu(0) \quad | \quad P_{k+1} = F P_k F^T + Q \stackrel{0}{\rightarrow} P_1 = F P_0 F^T$$

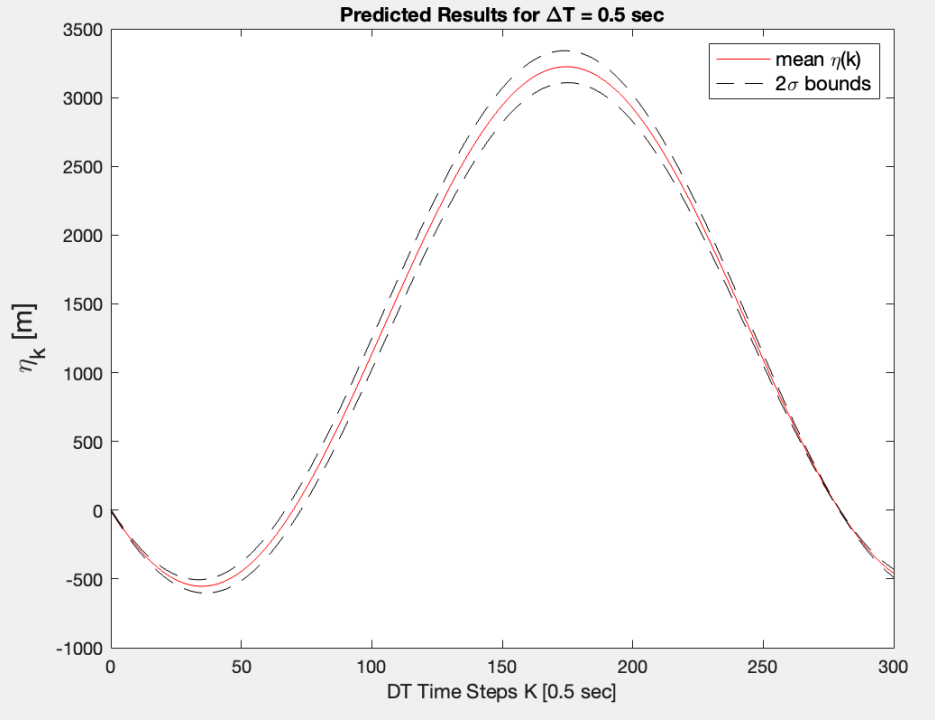
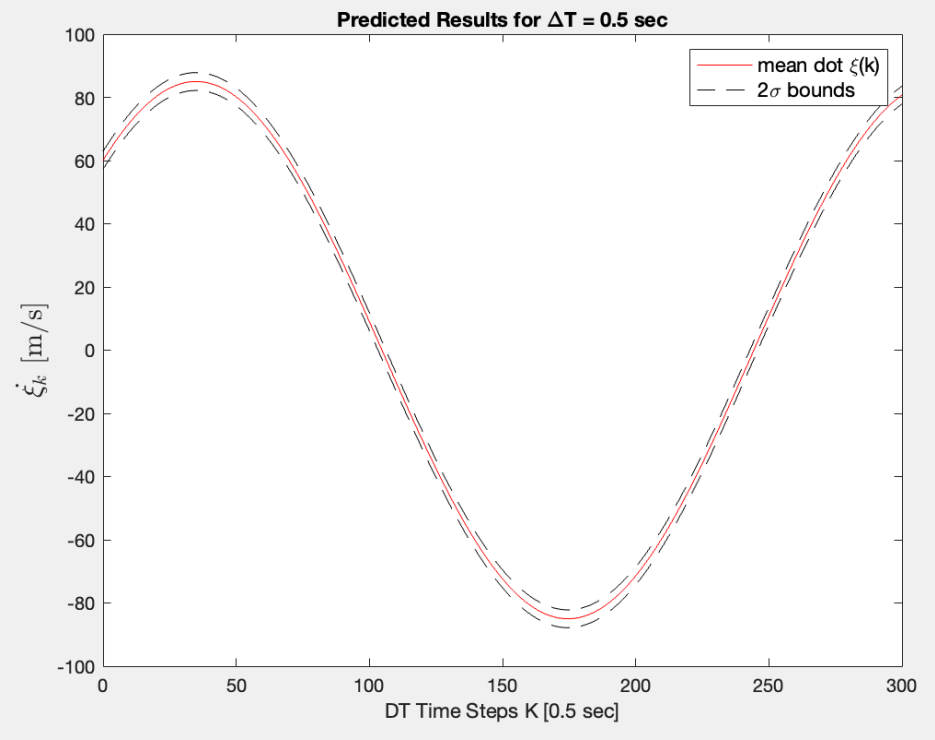
$$\mu(k+1) = F^{k+1}\mu(0) \rightarrow \mu(N) = F^N\mu(0) \quad | \quad P_2 = F_1 F_0 F_1^T F^T = F^2 P_0 F^{T^2}$$

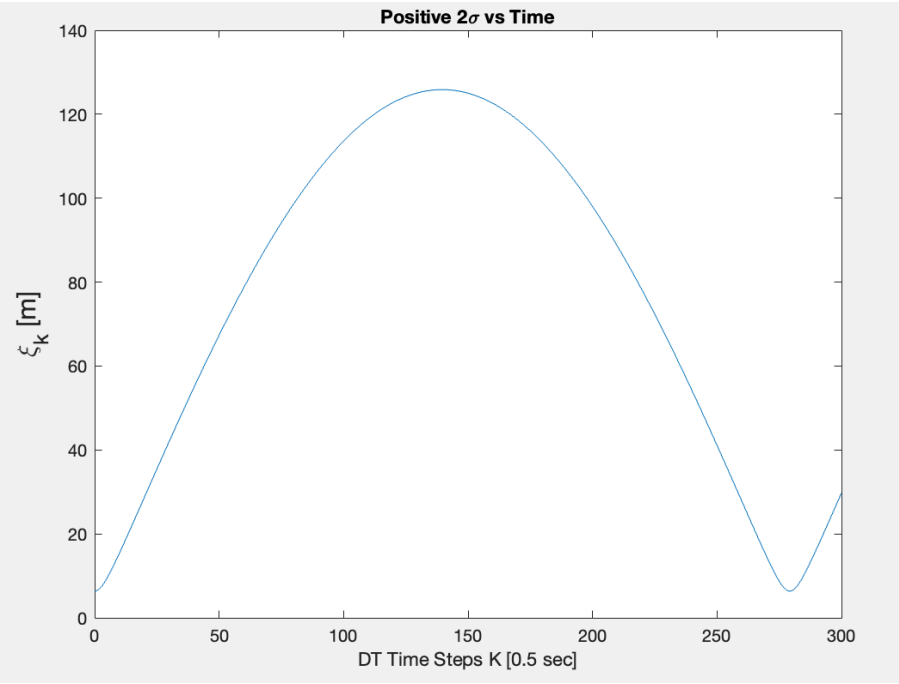
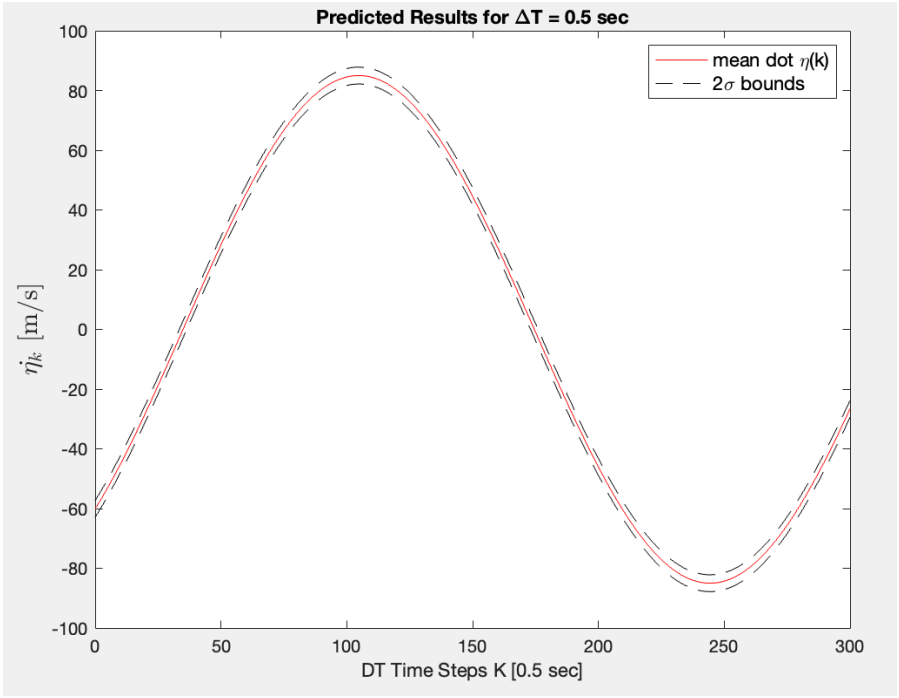
$$P_{k+1} = F^{k+1} P_0 F^{T^{k+1}} \rightarrow P_N = F^N P_0 (F^T)^N$$

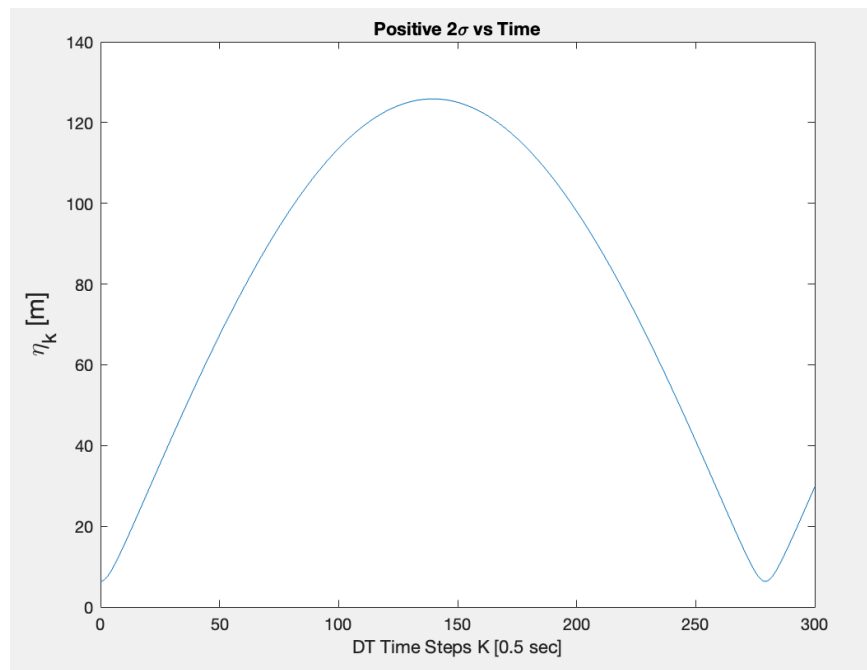
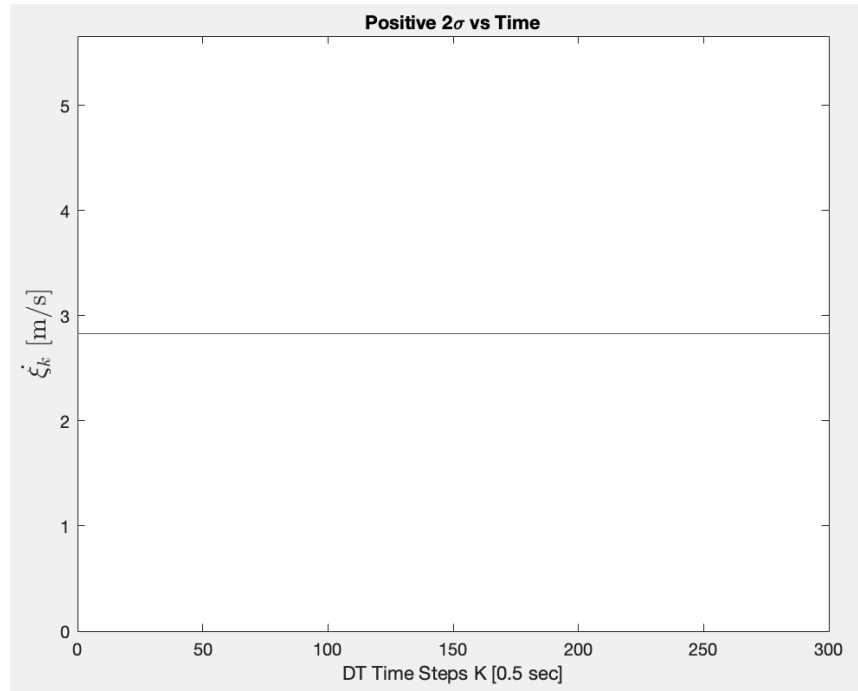
b) $\mu(0) = [0 \text{ m}, 85 \cos(\frac{\pi}{4}) \frac{\text{m}}{\text{s}}, 0 \text{ m}, -85 \sin(\frac{\pi}{4}) \frac{\text{m}}{\text{s}}]^T$

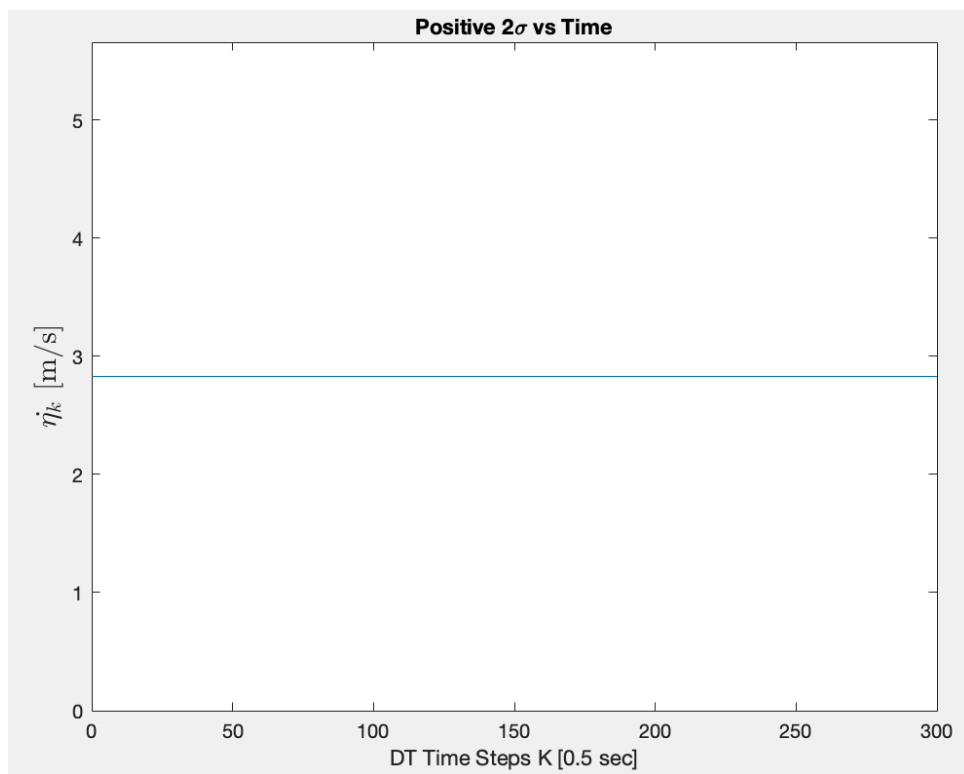
$P(0) = \text{diag}([10 \text{ m}^2, 2(\frac{\text{m}}{\text{s}})^2, 10 \text{ m}^2, 2(\frac{\text{m}}{\text{s}})^2]) \quad N=300$











Problem 3 → a) $r_c(k) = r_a(k) - r_b(k) \rightarrow r_a(k) = [\xi_a(k), \eta_a(k)]^T$, $r_b(k) = [\xi_b(k), \eta_b(k)]^T$
 $E[r_c(k)] = E[r_a(k) - r_b(k)] = E[r_a(k)] - E[r_b(k)]$
 $x_a(0) \sim \mathcal{N}(u_a(0), p_a(0)) \rightarrow \text{From Prob 2} \rightarrow x_a(k) \sim \mathcal{N}(F_a^k u_a(0), F_a^k p_a(0) F_a^{T^k})$
 similarly $\rightarrow x_b(k) \sim \mathcal{N}(F_b^k u_b(0), F_b^k p_b(0) F_b^{T^k})$
 $\mu_{ra}(k) = [F_a^k u_a(0)]_{1,3} \leftarrow 1^{st}, 3^{rd} \text{ components of vector}$
 $P_{ra}(k) = [F_a^k p_a(0) F_a^{T^k}]_{1,1,3,3} \leftarrow 1^{st}, 3^{rd} \text{ diagonal components of matrix}$
 $\mu_{rb}(k) = [F_b^k u_b(0)]_{1,3}$
 $P_{rb}(k) = [F_b^k p_b(0) F_b^{T^k}]_{1,1,3,3}$
 $\therefore \mu_{rc}(k) = \mu_{ra}(k) - \mu_{rb}(k) \rightarrow r_c(k) \sim \mathcal{N}(\mu_{rc}(k), P_{rc}(k))$
 $\sigma_{rc}^2(k) = P_{ra}(k) + P_{rb}(k) = P_{rc}(k)$
 b) $P(-\infty < r_c(k) \leq X) = \int_{-\infty}^X p(r_c(k)) dr_c(k) = C(X)$
 Aircrafts will collide when $\Delta \xi_c \in [-\xi_R, \xi_R]$ and $\Delta \eta_c \in [-\eta_R, \eta_R]$

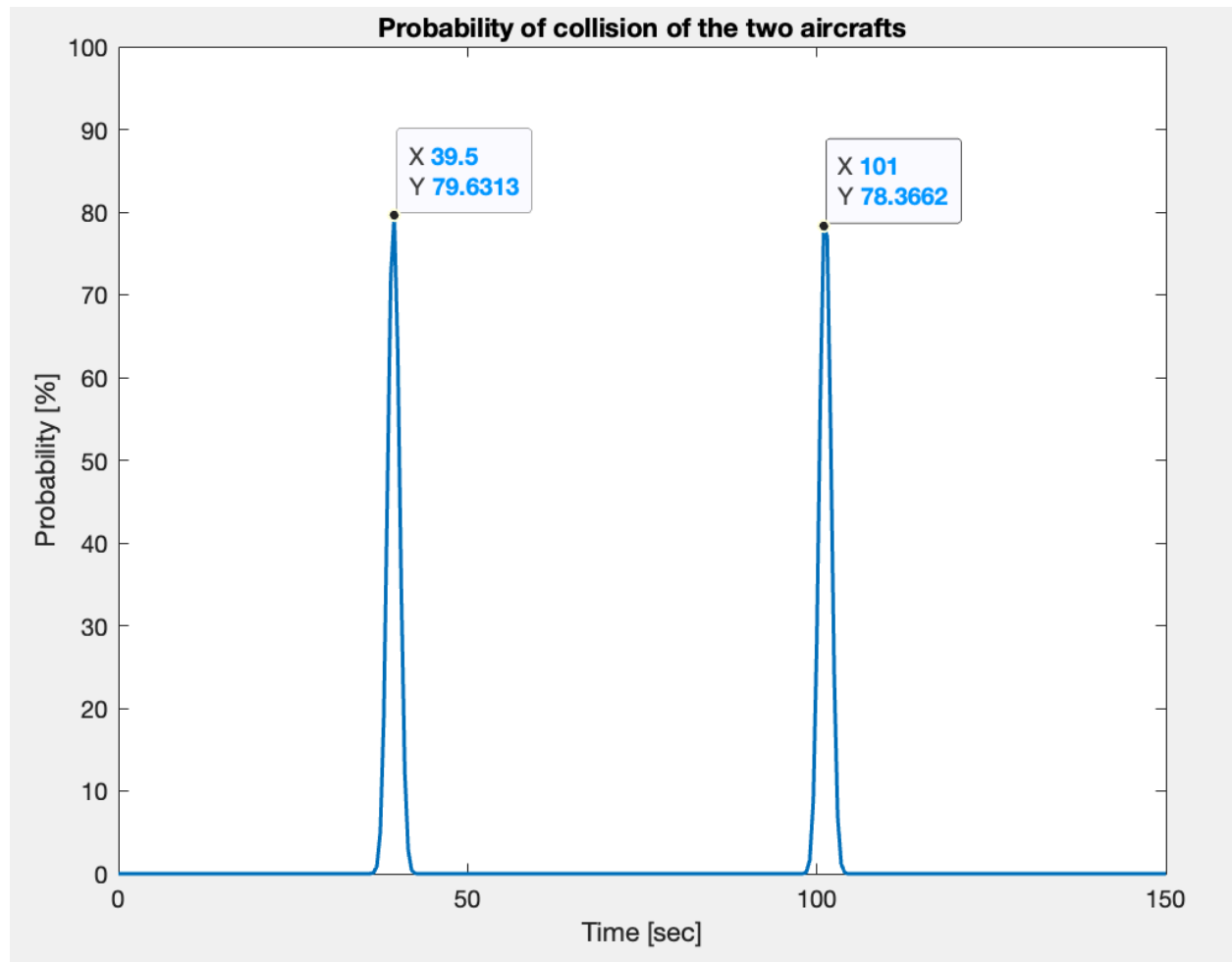
$$r_c(k) = [\Delta \xi(k), \Delta \eta(k)]^T \text{ let } X_R = [\xi_R, \eta_R]^T$$

so, aircrafts will collide if $-X_R \leq r_c(k) \leq X_R$

$$P(-X_R \leq r_c(k) \leq X_R) = \int_{-X_R}^{X_R} p(r_c(k)) dr_c(k) \text{ where } r_c(k) \sim \mathcal{N}(\mu_{rc}(k), P_{rc}(k))$$

$$P(-X_R \leq r_c(k) \leq X_R) = C(X_R) - C(-X_R)$$

Part c



- The aircrafts are in the greatest danger of colliding at time 39.5 seconds and 101 seconds.
- Outside of these two times (and some margin around them), the probability of collision is almost always zero. That makes sense because these two aircrafts are moving almost circularly in opposite directions which causes certain times to have a very high probability of collision and other times to have zero probability of collision.

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```
clear; clc; close all;

% ASEN 5044, HW 7
% Fall 2024
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syms delta_t
syms omega

A = [0 1 0 0; 0 0 0 -omega; 0 0 0 1; 0 omega 0 0];

e_at = expm(A * delta_t);

delta_t = 0.5;
omega = 0.045;

% Omega times delta_t
odt = omega * delta_t;

e_at_given = [1 sin(odt)/omega 0 -(1 - cos(odt))/omega; 0 cos(odt) 0
-sin(odt); 0 (1-cos(odt))/omega 1 sin(odt)/omega; 0 sin(odt) 0 cos(odt)];

u_0 = [0; 85*cos(pi/4); 0; -85*sin(pi/4)];
P_a_0 = diag([10, 2, 10, 2]);

T = 300;

F = e_at_given;

mu = [u_0];
P = [P_a_0];
P_1_2sig = 2 * sqrt(P(1, 1));
P_2_2sig = 2 * sqrt(P(2, 2));
P_3_2sig = 2 * sqrt(P(3, 3));
P_4_2sig = 2 * sqrt(P(4, 4));

for i = 1:T
    mu(:, i+1) = F^i * u_0;
    F_t = F';
    P_temp = F^i * P_a_0 * F_t^i;
    P(:, i+1) = get_diag(P_temp);
    P_1_2sig(i+1) = 2 * sqrt(P(1, i+1));
    P_2_2sig(i+1) = 2 * sqrt(P(2, i+1));
    P_3_2sig(i+1) = 2 * sqrt(P(3, i+1));
    P_4_2sig(i+1) = 2 * sqrt(P(4, i+1));
end
```

```

function out = get_diag(mat)
    len = length(mat);
    for i = 1:len
        out(i, 1) = mat(i, i);
    end
end

k = 0:T;

figure()
plot(k, mu(1, :), 'r')
hold on
plot(k, mu(1, :) + P_1_2sig, '--k')
plot(k, mu(1, :) - P_1_2sig, '--k')
hold off
title("Predicted Results for \DeltaT = 0.5 sec")
legend("mean \xi(k)", "2\sigma bounds", 'FontSize', 11)
xlabel("DT Time Steps K [0.5 sec]")
ylabel("\xi_k [m]", 'FontSize', 15)

figure()
plot(k, mu(2, :), 'r')
hold on
plot(k, mu(2, :) + P_2_2sig, '--k')
plot(k, mu(2, :) - P_2_2sig, '--k')
hold off
title("Predicted Results for \DeltaT = 0.5 sec")
legend("mean dot \xi(k)", "2\sigma bounds", 'FontSize', 11)
xlabel("DT Time Steps K [0.5 sec]")
ylabel('$\dot{\xi}_k$ [m/s]', 'Interpreter','latex', 'FontSize', 15)

figure()
plot(k, mu(3, :), 'r')
hold on
plot(k, mu(3, :) + P_3_2sig, '--k')
plot(k, mu(3, :) - P_3_2sig, '--k')
hold off
title("Predicted Results for \DeltaT = 0.5 sec")
legend("mean \eta(k)", "2\sigma bounds", 'FontSize', 11)
xlabel("DT Time Steps K [0.5 sec]")
ylabel("\eta_k [m]", 'FontSize', 15)

figure()
plot(k, mu(4, :), 'r')
hold on
plot(k, mu(4, :) + P_4_2sig, '--k')
plot(k, mu(4, :) - P_4_2sig, '--k')
hold off
title("Predicted Results for \DeltaT = 0.5 sec")

```

```

legend("mean dot \eta(k)", "2\sigma bounds", 'FontSize', 11)
xlabel("DT Time Steps K [0.5 sec]")
ylabel('$\dot{\eta}_k$ [m/s]', 'Interpreter','latex', 'FontSize', 15)

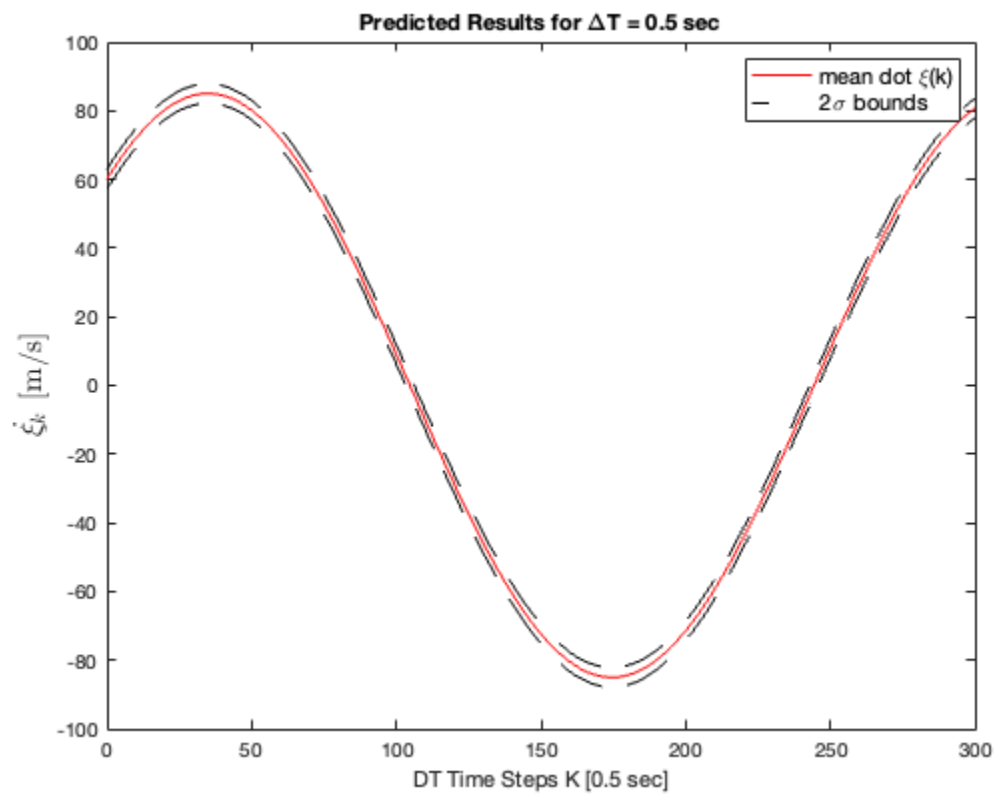
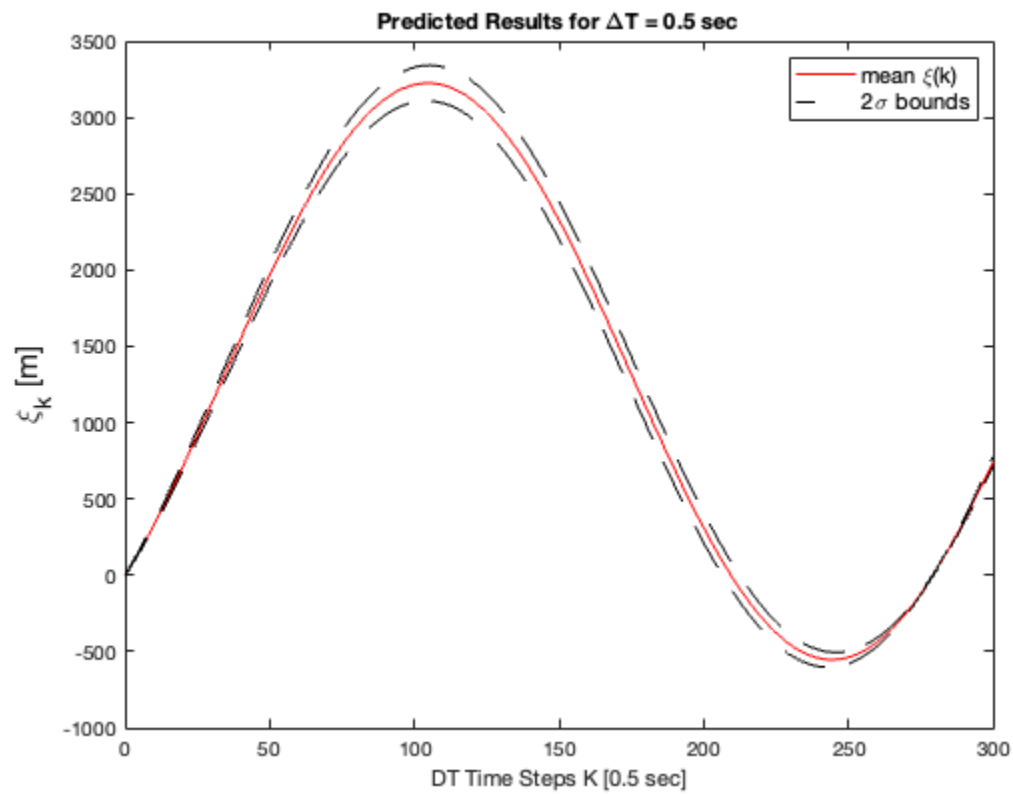
figure()
plot(k, P_1_2sig)
title("Positive 2\sigma vs Time")
xlabel("DT Time Steps K [0.5 sec]")
ylabel("\xi_k [m]", 'FontSize', 15)

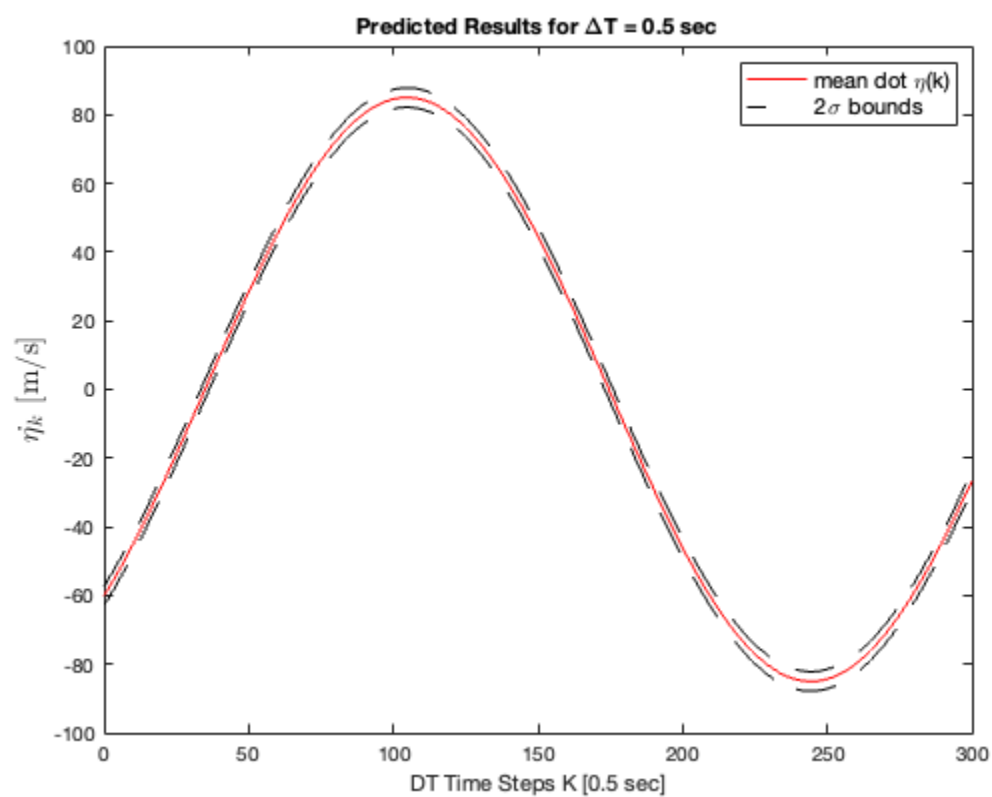
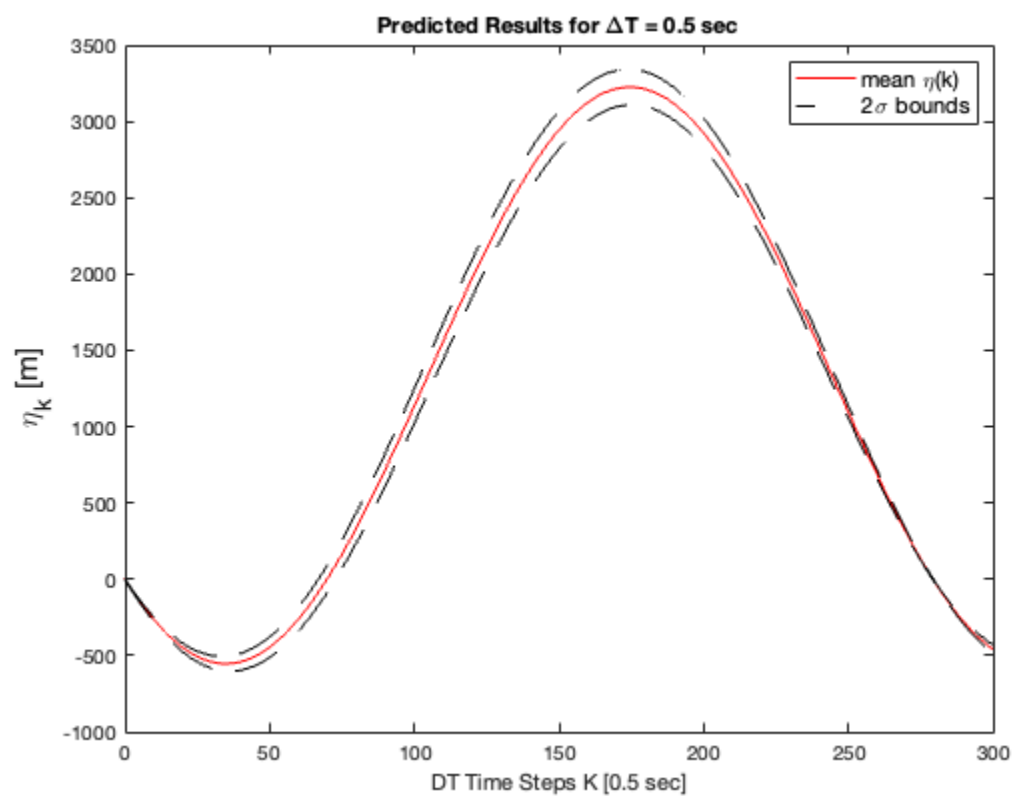
figure()
plot(k, P_2_2sig)
title("Positive 2\sigma vs Time")
xlabel("DT Time Steps K [0.5 sec]")
ylabel('$\dot{\xi}_k$ [m/s]', 'Interpreter','latex', 'FontSize', 15)
ylim([0 2*P_2_2sig(1)])

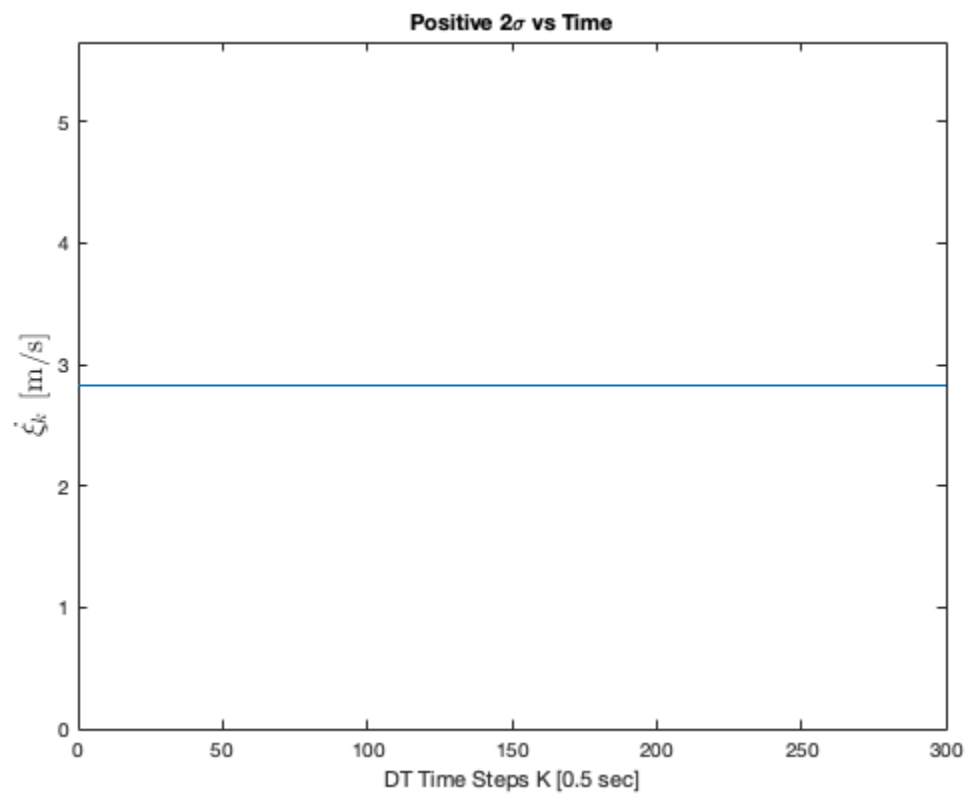
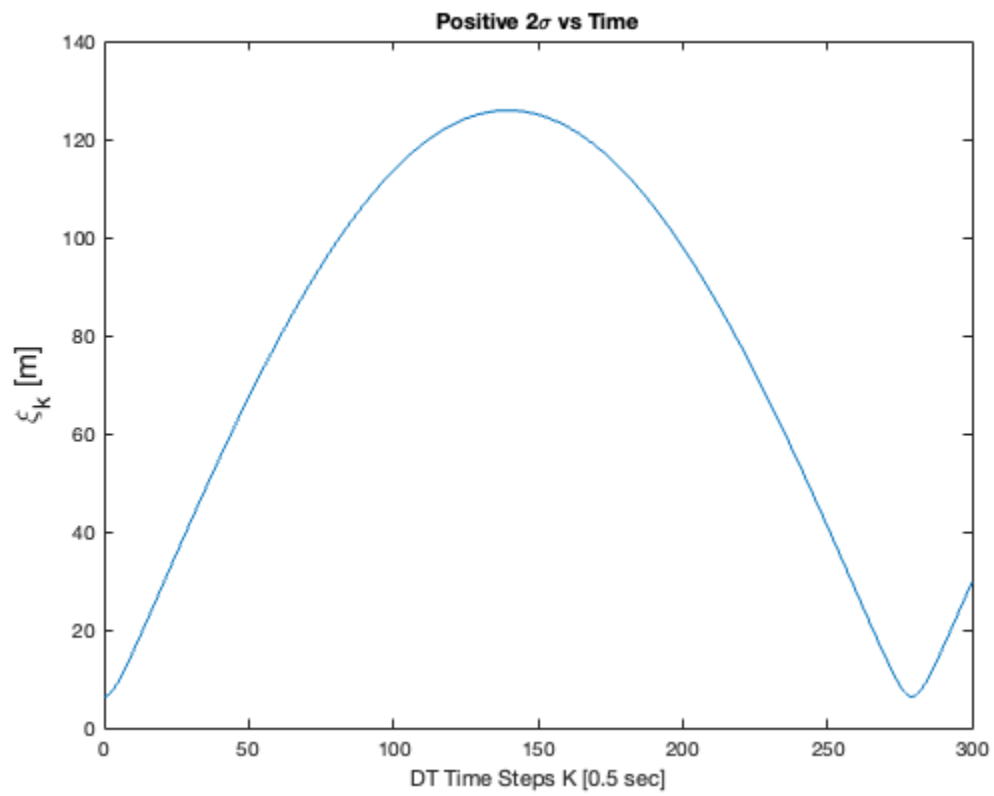
figure()
plot(k, P_3_2sig)
title("Positive 2\sigma vs Time")
xlabel("DT Time Steps K [0.5 sec]")
ylabel("\eta_k [m]", 'FontSize', 15)

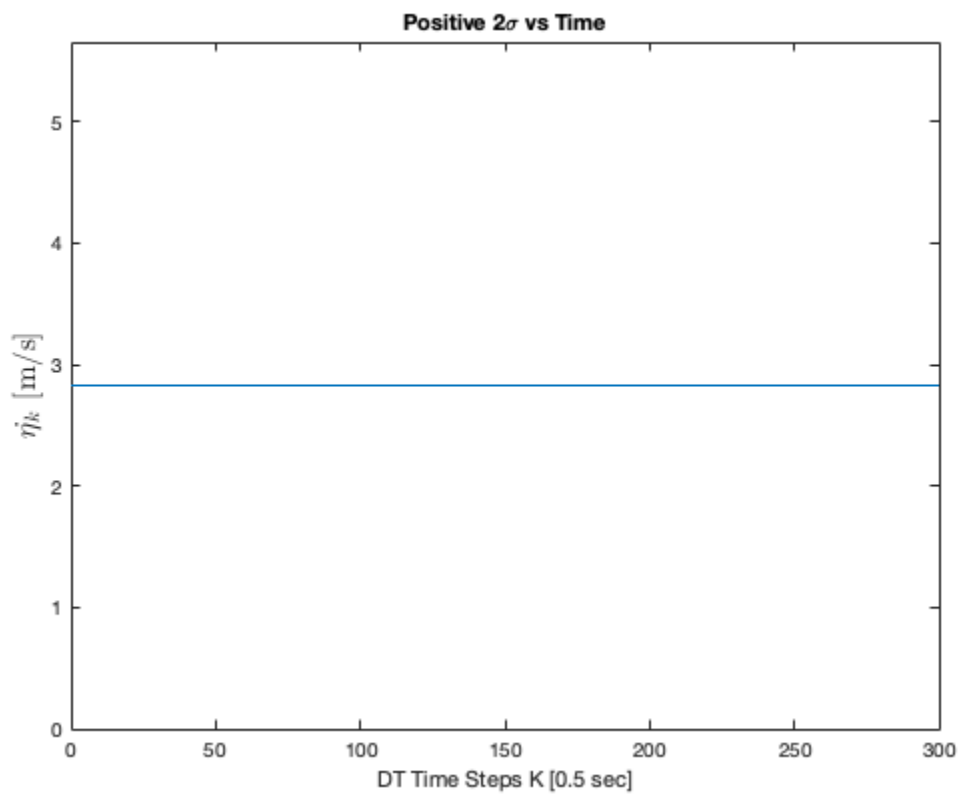
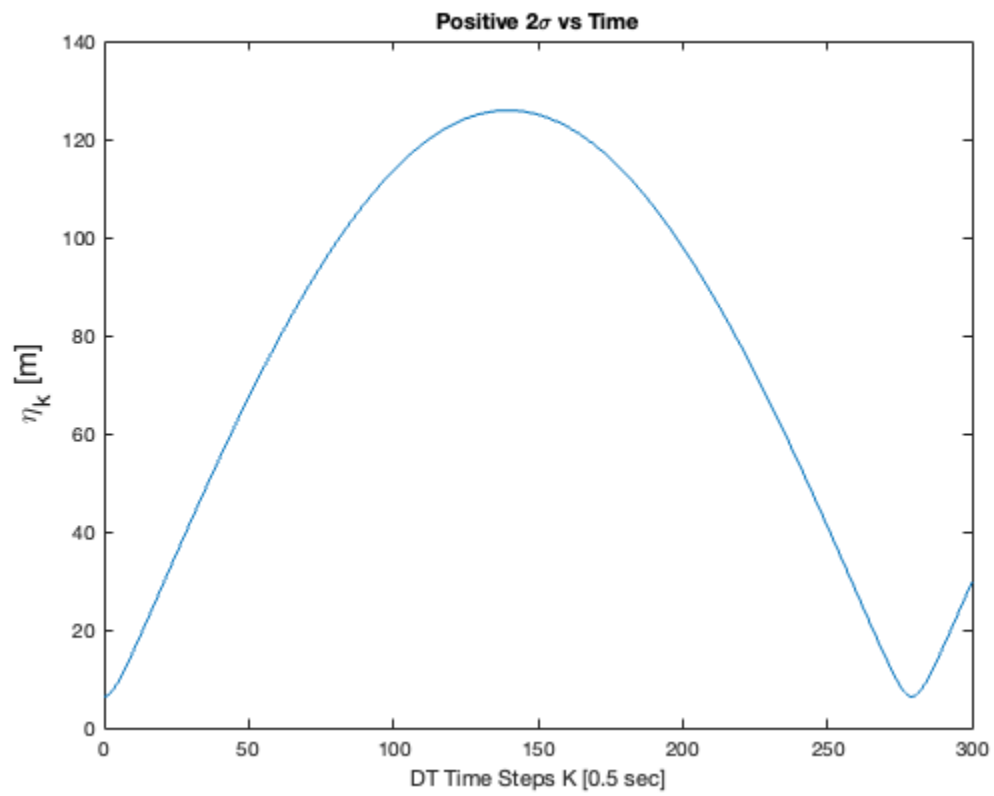
figure()
plot(k, P_4_2sig)
title("Positive 2\sigma vs Time")
xlabel("DT Time Steps K [0.5 sec]")
ylabel('$\dot{\eta}_k$ [m/s]', 'Interpreter','latex', 'FontSize', 15)
ylim([0 2*P_4_2sig(1)])

```









Problem 3

```
delta_t = 0.5;

omega_a = 0.045;
omega_b = -0.045;

odt_a = omega_a * delta_t;

F_a = [1 sin(odt_a)/omega_a 0 -(1-cos(odt_a))/omega_a;
        0 cos(odt_a) 0 -sin(odt_a);
        0 (1-cos(odt_a))/omega_a 1 sin(odt_a)/omega_a;
        0 sin(odt_a) 0 cos(odt_a)];

odt_b = omega_b * delta_t;

F_b = [1 sin(odt_b)/omega_b 0 -(1-cos(odt_b))/omega_b;
        0 cos(odt_b) 0 -sin(odt_b);
        0 (1-cos(odt_b))/omega_b 1 sin(odt_b)/omega_b;
        0 sin(odt_b) 0 cos(odt_b)];

u_a_0 = [0; 85*cos(pi/4); 0; -85*sin(pi/4)];
u_b_0 = [3200; 85*cos(pi/4); 3200; -85*sin(pi/4)];

u_a = [u_a_0];
u_b = [u_b_0];

P_a_0 = diag([10, 4, 10, 4]);
P_b_0 = diag([11, 3.5, 11, 3.5]);

P_a = [P_a_0];
P_b = [P_b_0];

T = 150;
t = 0:delta_t:T;

for i = 1:T/delta_t
    u_a(:, i+1) = F_a^i * u_a_0;
    P_a((4*i + 1):(4*(i+1)), :) = F_a^i * P_a_0 * (F_a')^i;

    u_b(:, i+1) = F_b^i * u_b_0;
    P_b((4*i + 1):(4*(i+1)), :) = F_b^i * P_b_0 * (F_b')^i;
end

u_c = u_a - u_b;
P_c = P_a + P_b;

u_r_c = [u_c(1, :); u_c(3, :)];
P_r_c = [P_c(1, 1), P_c(1, 3); P_c(3, 1), P_c(3, 3)];

xi_R = 100;
eta_R = 100;
x_R = [xi_R; eta_R];
```

```

cdf_x_R = mvncdf(x_R, u_r_c(:,1), P_r_c);
cdf_neg_x_R = mvncdf(-x_R, u_r_c(:,1), P_r_c);

probability_of_collision = cdf_x_R - cdf_neg_x_R;

for i = 1:T/delta_t
    P_c_temp = [P_c((4*i + 1):(4*(i+1))),:];
    P_r_c_current = [P_c_temp(1,1), P_c_temp(1,3); P_c_temp(3,1),
P_c_temp(3,3)];
    P_r_c((2*i + 1):(2*(i+1)), :) = P_r_c_current;

    cdf_x_R = mvncdf(x_R, u_r_c(:,i), P_r_c_current);
    cdf_neg_x_R = mvncdf(-x_R, u_r_c(:,i), P_r_c_current);

    probability_of_collision(i+1) = cdf_x_R - cdf_neg_x_R;
end

figure()
plot(t, probability_of_collision*100, 'LineWidth',1.5)
xlabel("Time [sec]")
ylabel("Probability [%]")
ylim([0, 100])
title("Probability of collision of the two aircrafts")

```

