

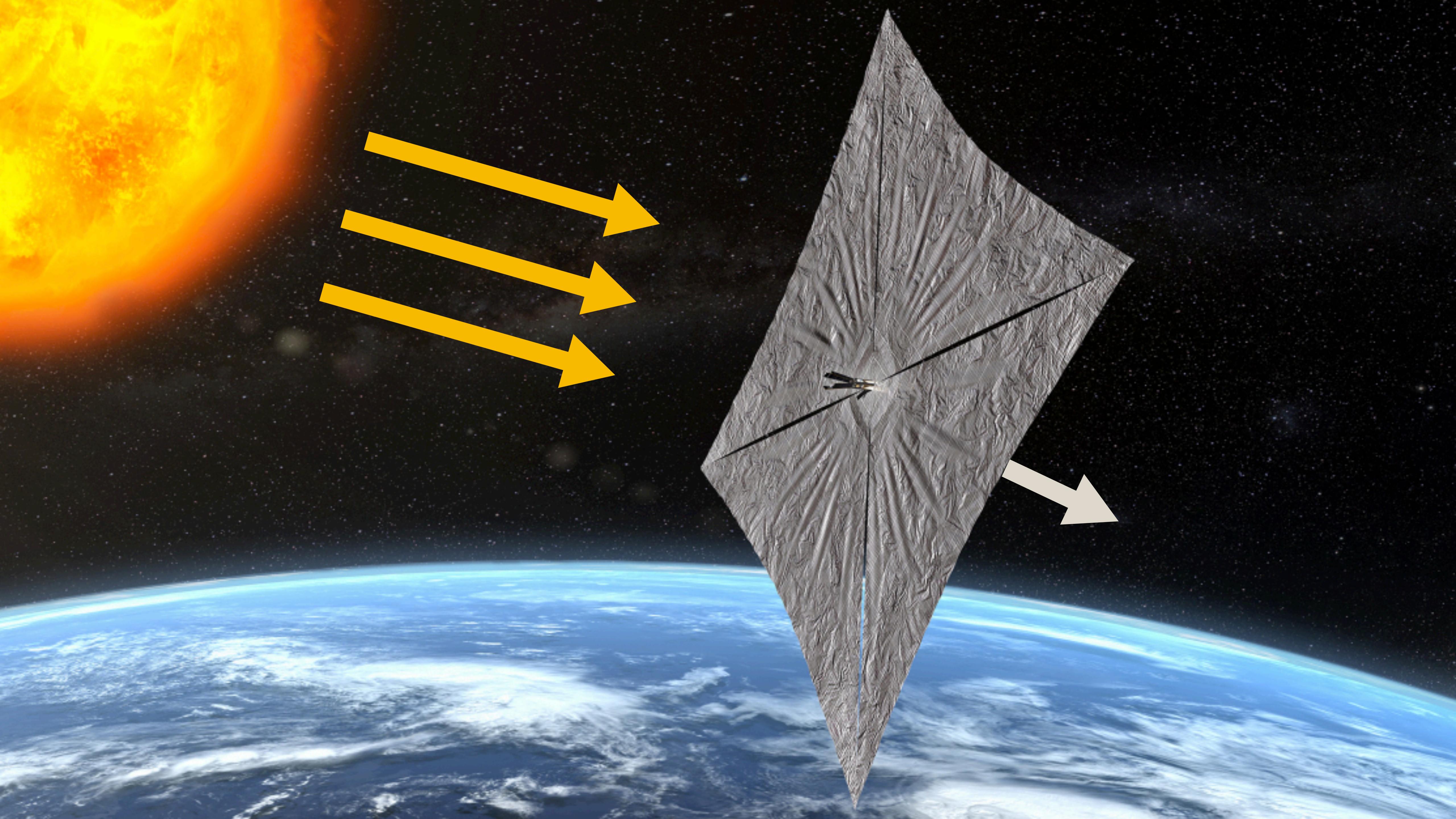
CONTROL OF SOLAR SAILS

J.-B. Caillau, L. Dell'Elce, A. Herasimenka, J.-B. Pomet

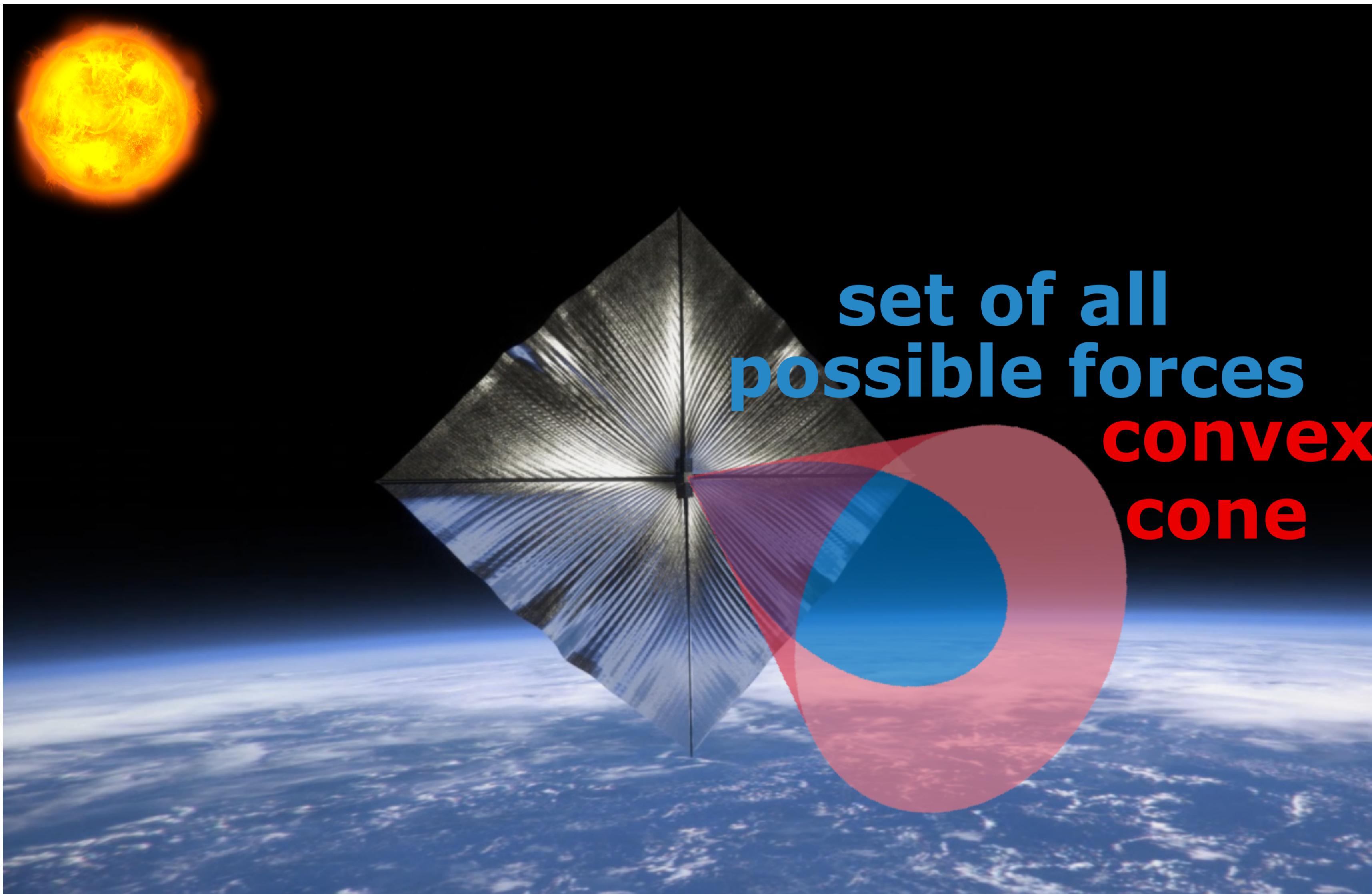
Université Côte d'Azur, CNRS, Inria, LJAD, France

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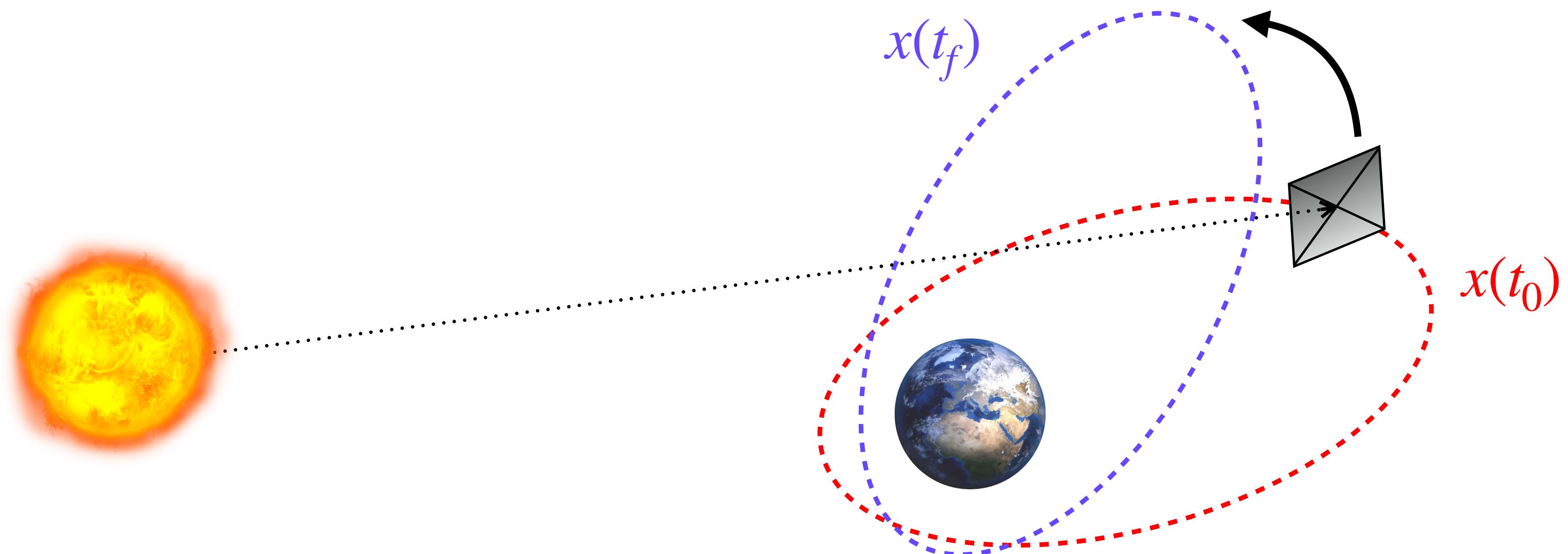
Zhejiang U., March 2025



Non-ideal sail: a cone-constrained control problem

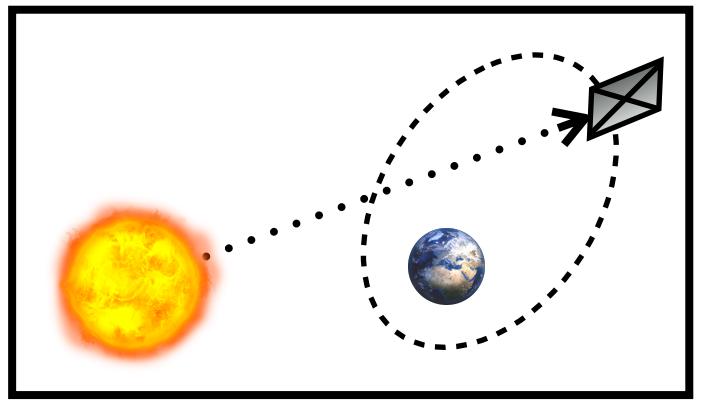


Can non-ideal sails arbitrarily change their orbit?



Is it possible to generate any $x(t_0) \rightarrow x(t_f)$?

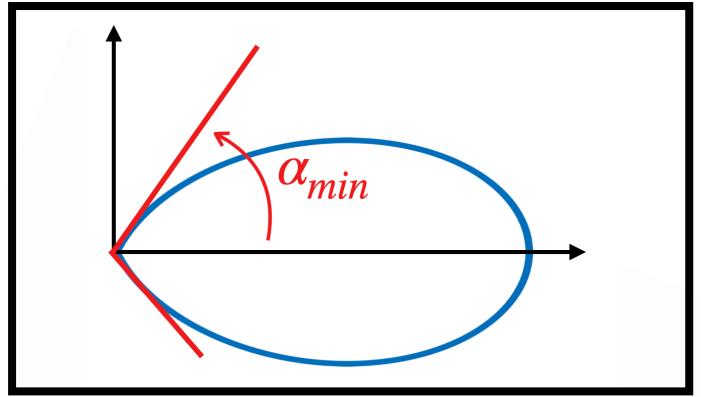
Outline



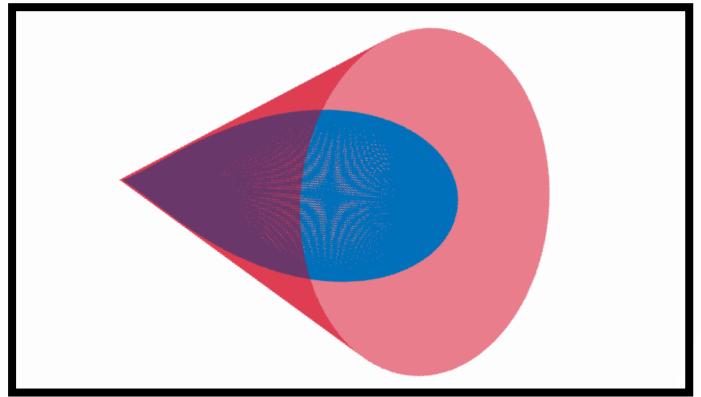
1. Dynamics of the system

$$\left\langle p_I, \frac{dI(f, u)}{dt} \right\rangle \geq 0$$

2. Necessary condition for local controllability

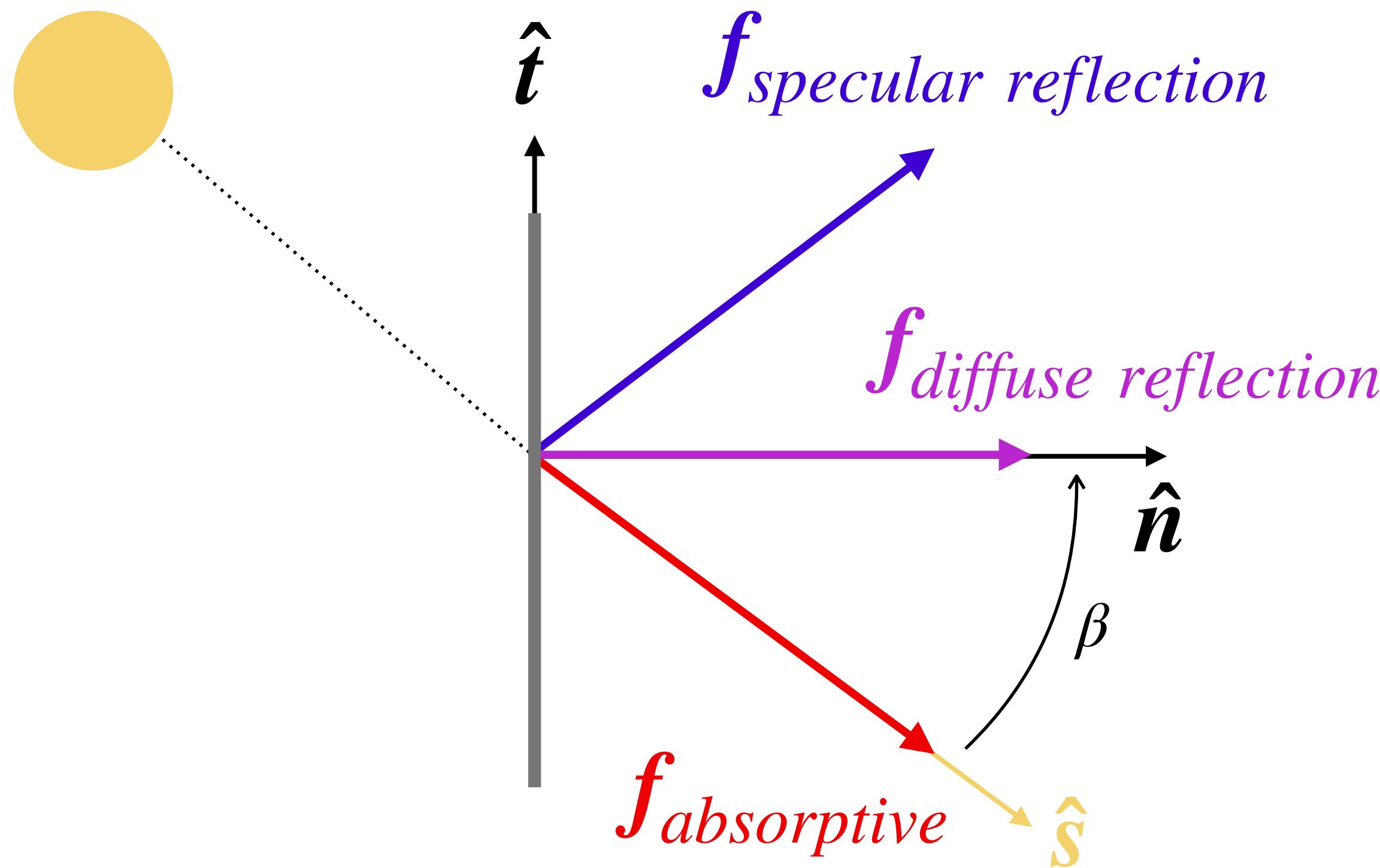


3. Minimum optical requirements



4. Generalization to other problems

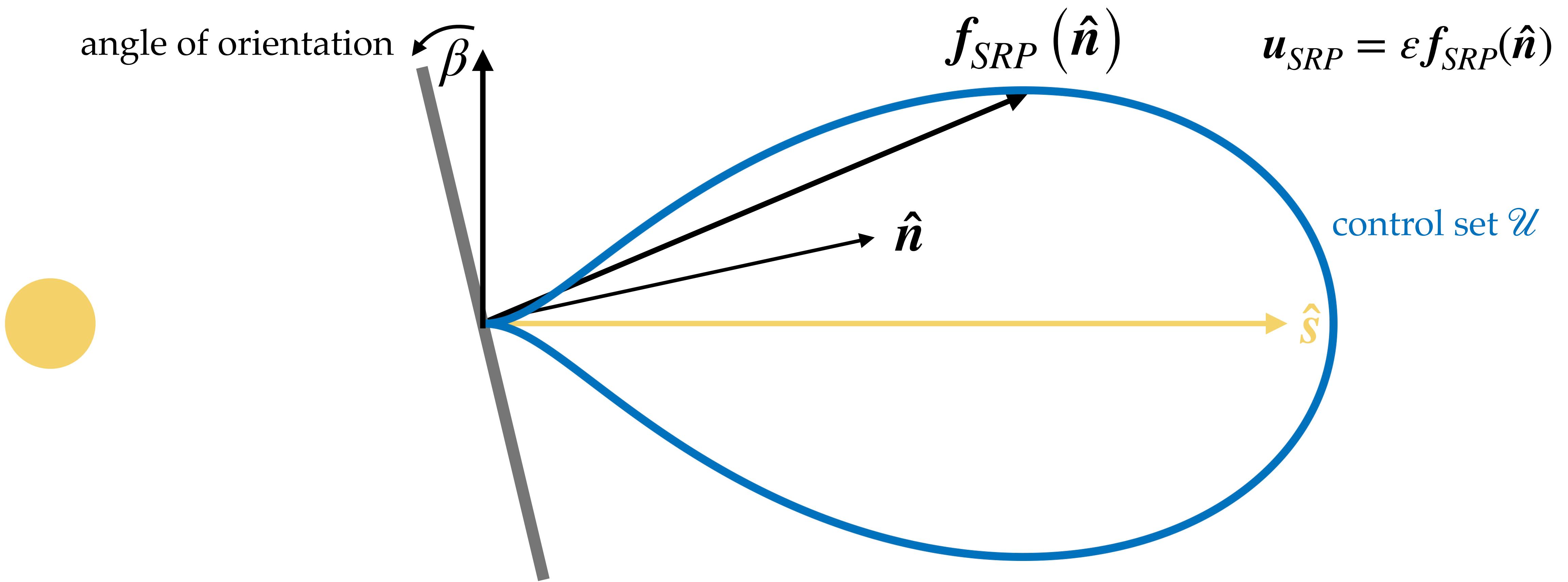
1. Force components of solar sail



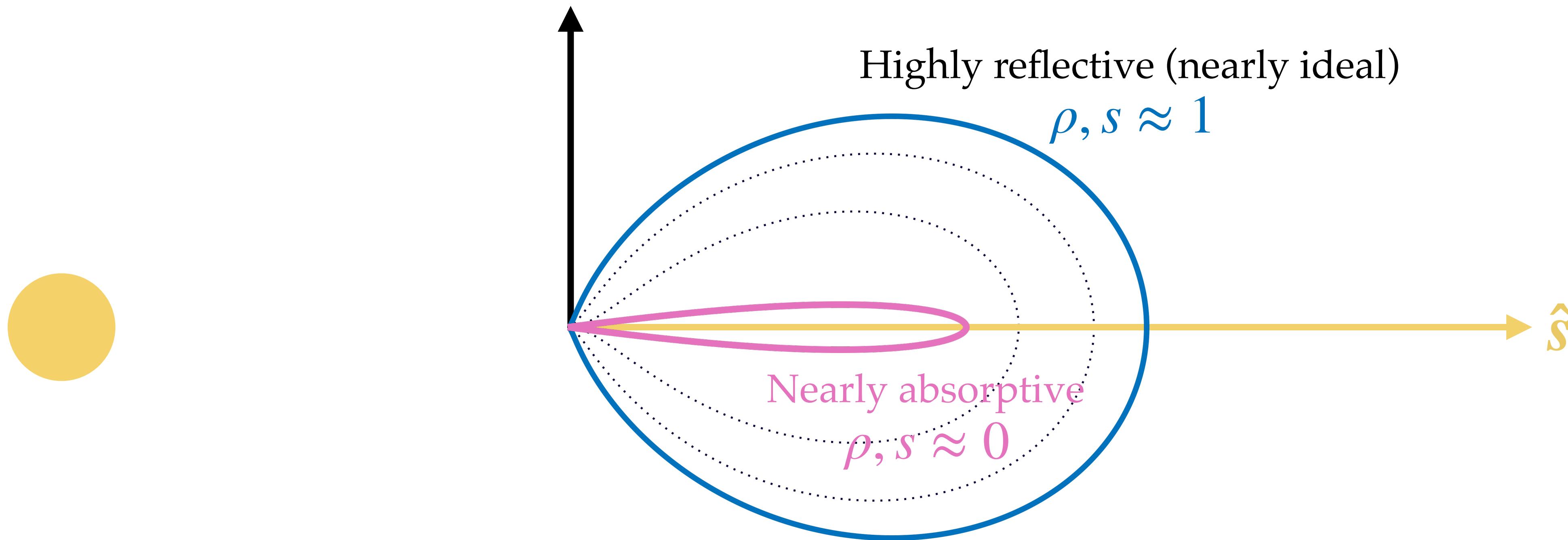
$$f_{SRP} = f_{absorptive} + f_{specular\ reflection} + f_{diffuse\ reflection}$$

1. Control set

$$\dot{x} = F^0(x) + \sum_i u_i F^i(x), \quad u \in \mathcal{U}$$

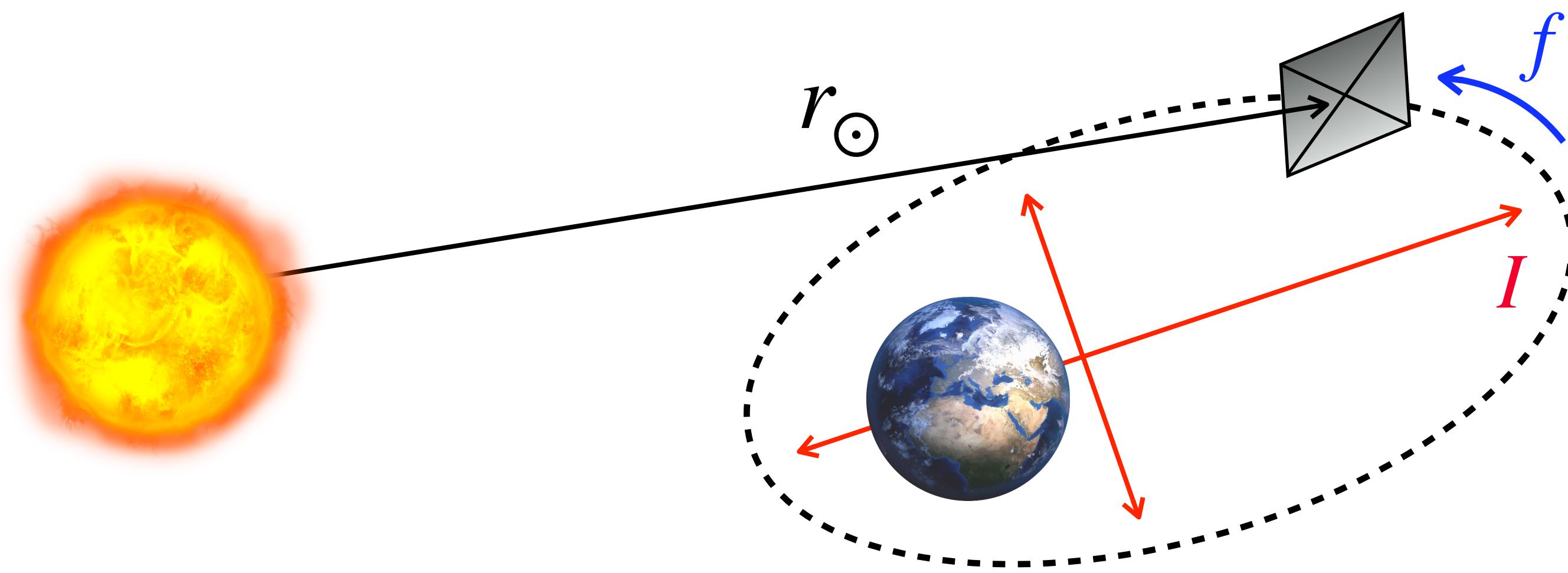


1. Parametrisation of the control set



$\rho, s \in [0,1]$ portion of specular, diffuse reflection

1. Dynamical system



Assumptions:

- No eclipses
- Sun motion neglected over one orbit
- SRP is the only perturbation

$$\dot{x} = F^0(x) + \sum_i u_i F^i(x), \quad u \in \mathcal{U}, \quad i = 1, 2, 3$$

with $x = (\mathbf{I}, \mathbf{f})$, $\mathbf{I} \in M$, $\mathbf{f} \in \mathbb{S}^1$, F^0, F^i given by Gauss variational equations

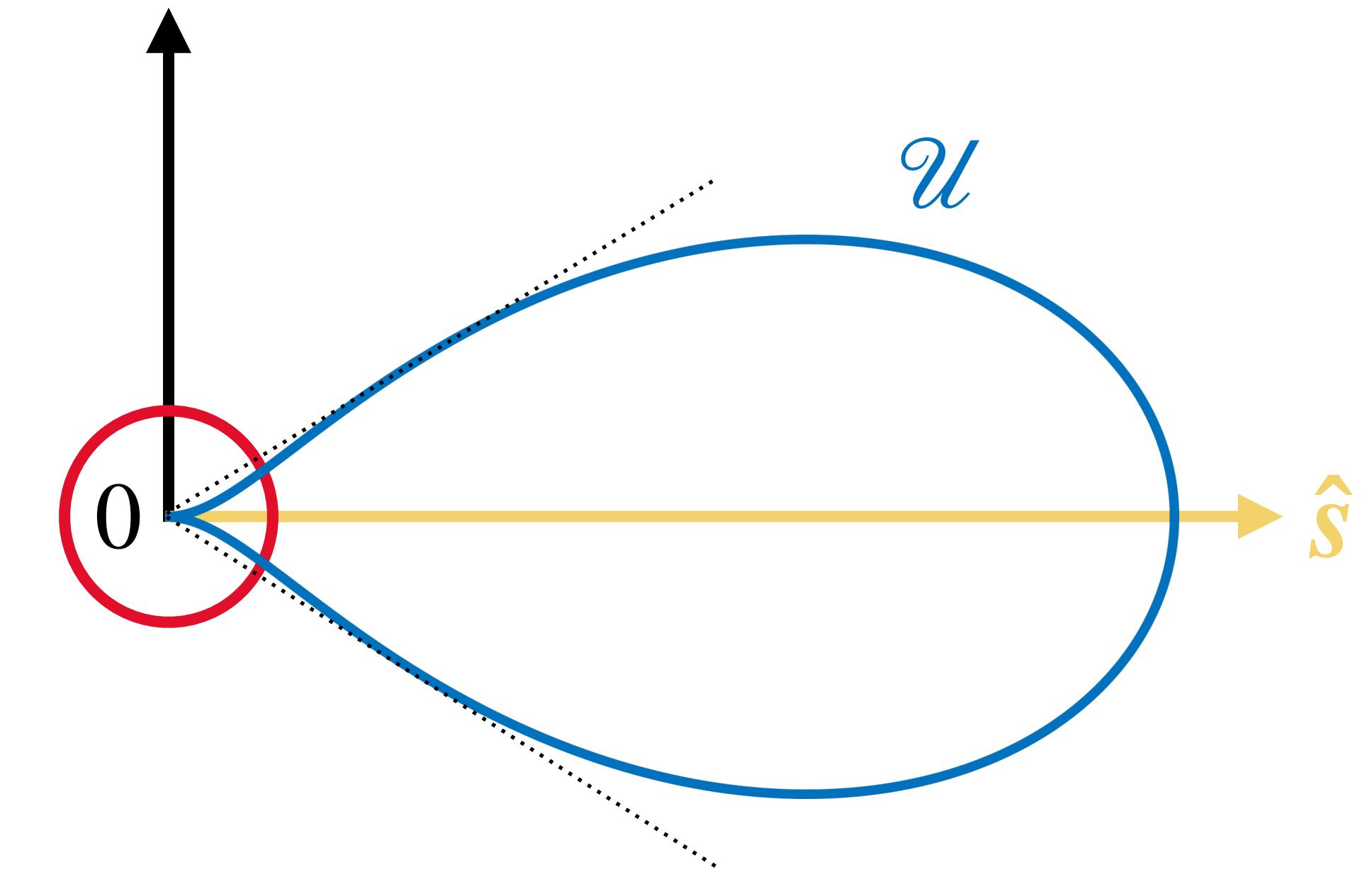
2. Classical approach using Lie brackets

Controllability if [Jurdjevic, 1996]:

i) Periodic drift $\rightarrow \text{OK}$

ii) Bracket generating
 OK if $\rho > 0$
first integral if $\rho = 0$

iii) $\text{Conv}(\mathcal{U})$ is neighbourhood of the origin $\rightarrow \text{FAIL}$



2. Proposition on controllability

Under the conditions:

- (i) system is bracket generating,
- (ii) control set \mathcal{U} contains the origin,
- (iii) $\forall I \in M, \quad \text{cone} \left\{ \sum_i u_i F^i(I, f), u \in \mathcal{U}, f \in \mathbb{S}^1 \right\} = T_I M,$

the system is controllable* in the following sense: for any $(I_0, f_0), (I', f') \in M \times \mathbb{S}^1$, there is a time T and a integrable control $u \in \mathcal{U}$ that drives (I_0, f_0) to (I', f') at time T .

* The proof is available in [H., Caillau, Dell'Elce, Pomet, 2022]

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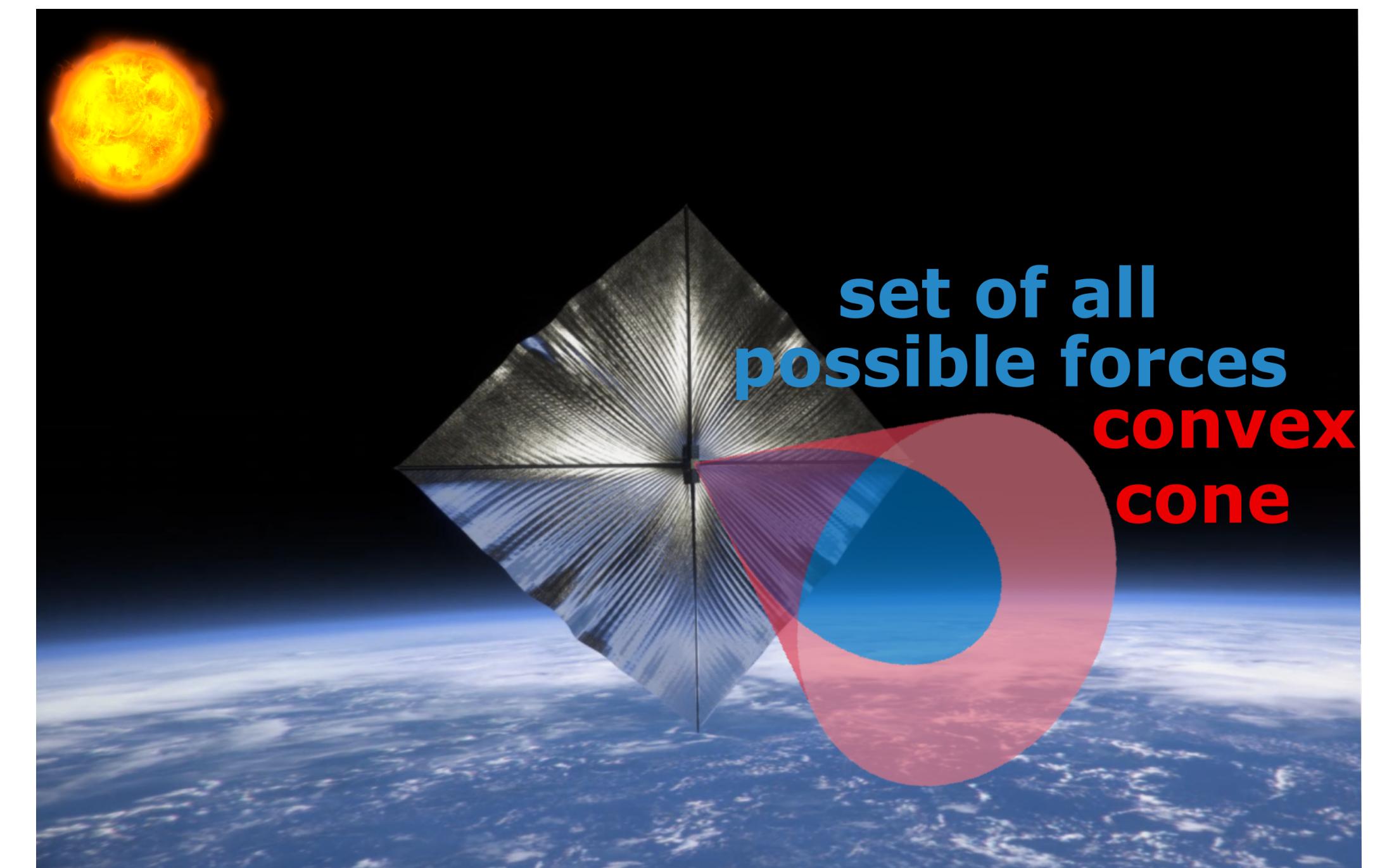
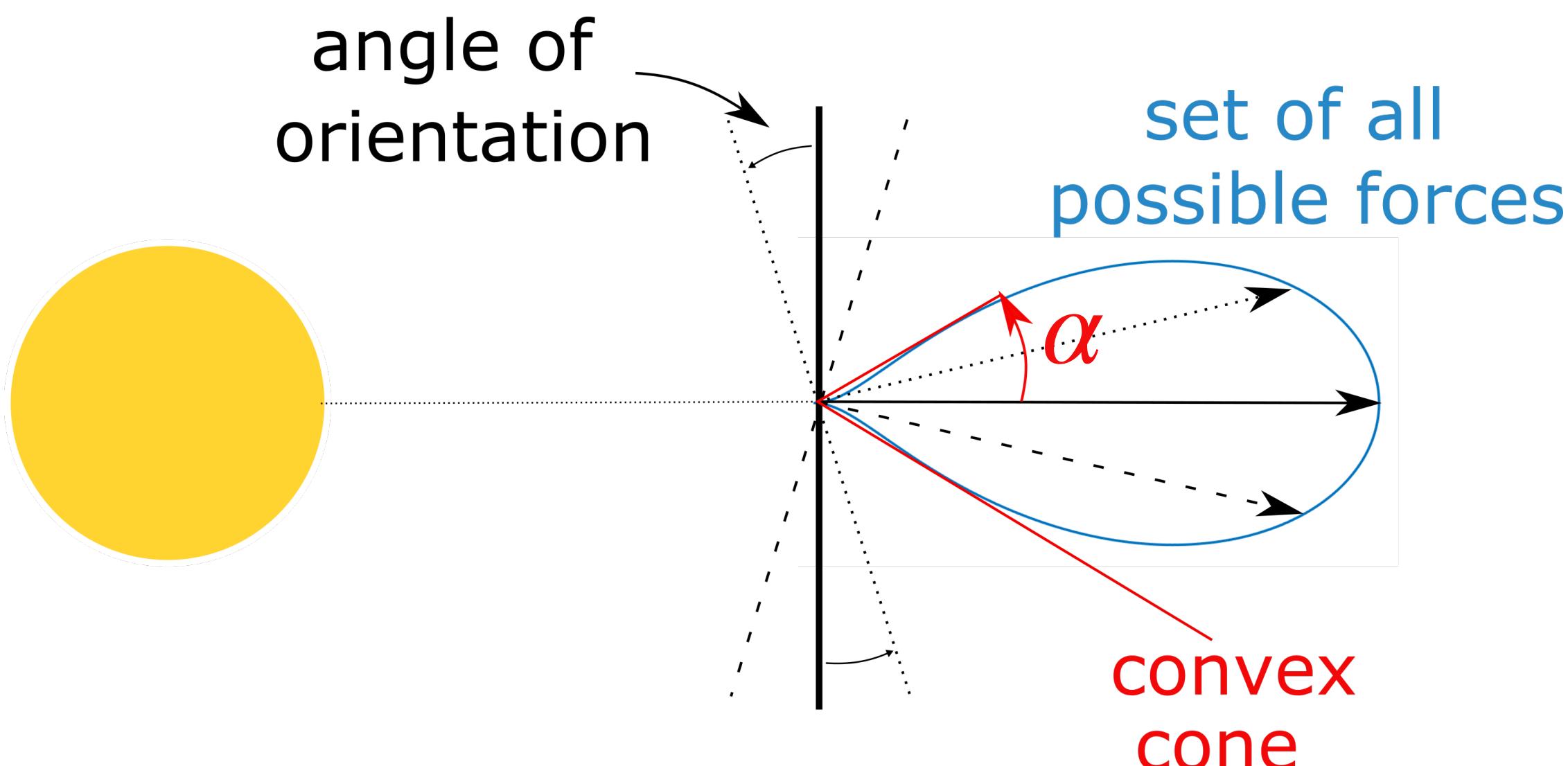
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2. Convexification of the control set

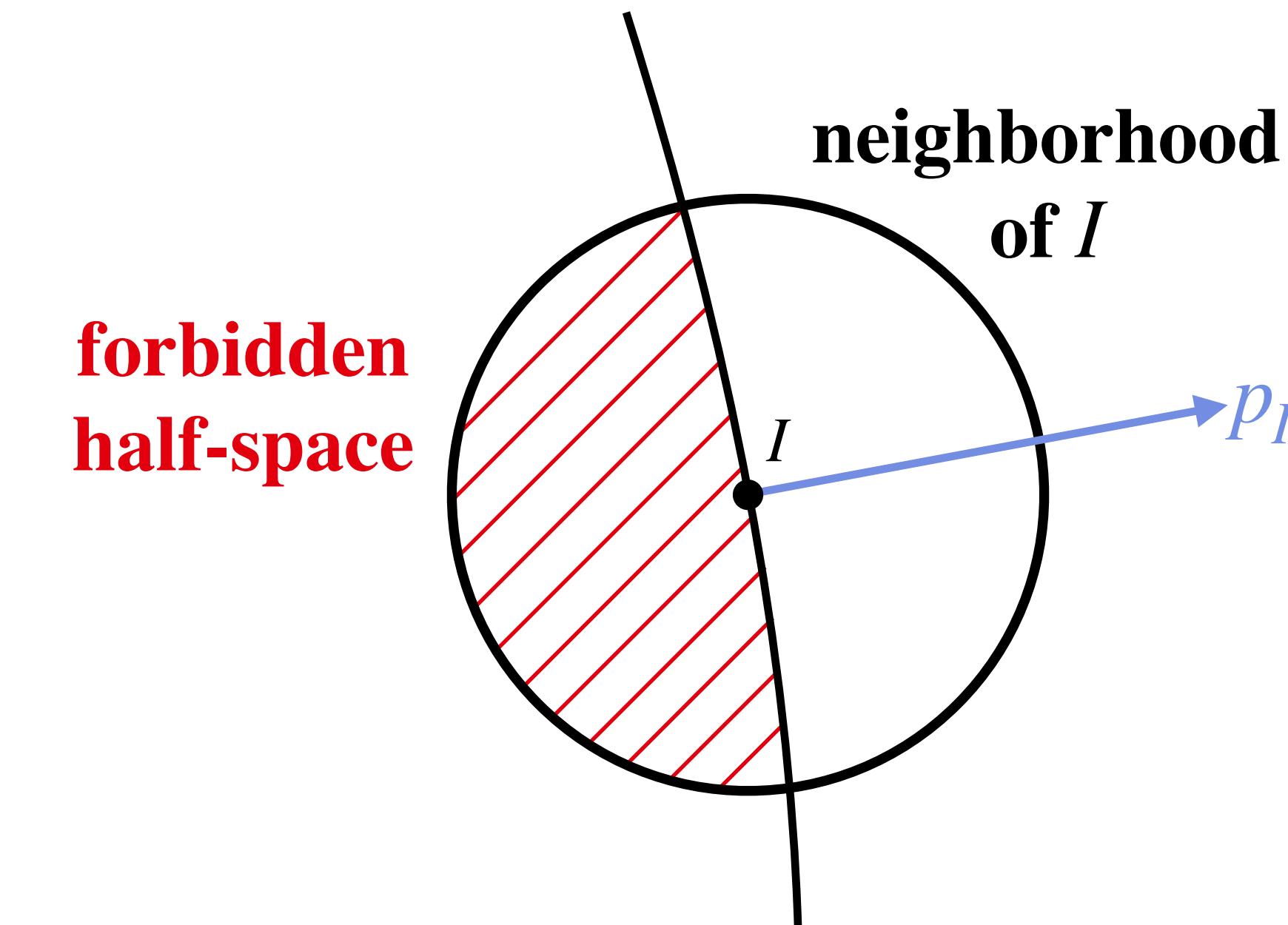
$$u \in \mathcal{U} \subset K_\alpha := \text{cone}(\mathcal{U})$$



2. How to verify the condition?

Negation: If \exists a one-form $p_I \in T^*M$, s.t.

$$\left\langle p_I, \frac{dI(f, u)}{dt} \right\rangle \geq 0, \quad \forall f \in \mathbb{S}^1, u \in \partial K_\alpha \rightarrow \text{not locally controllable}$$



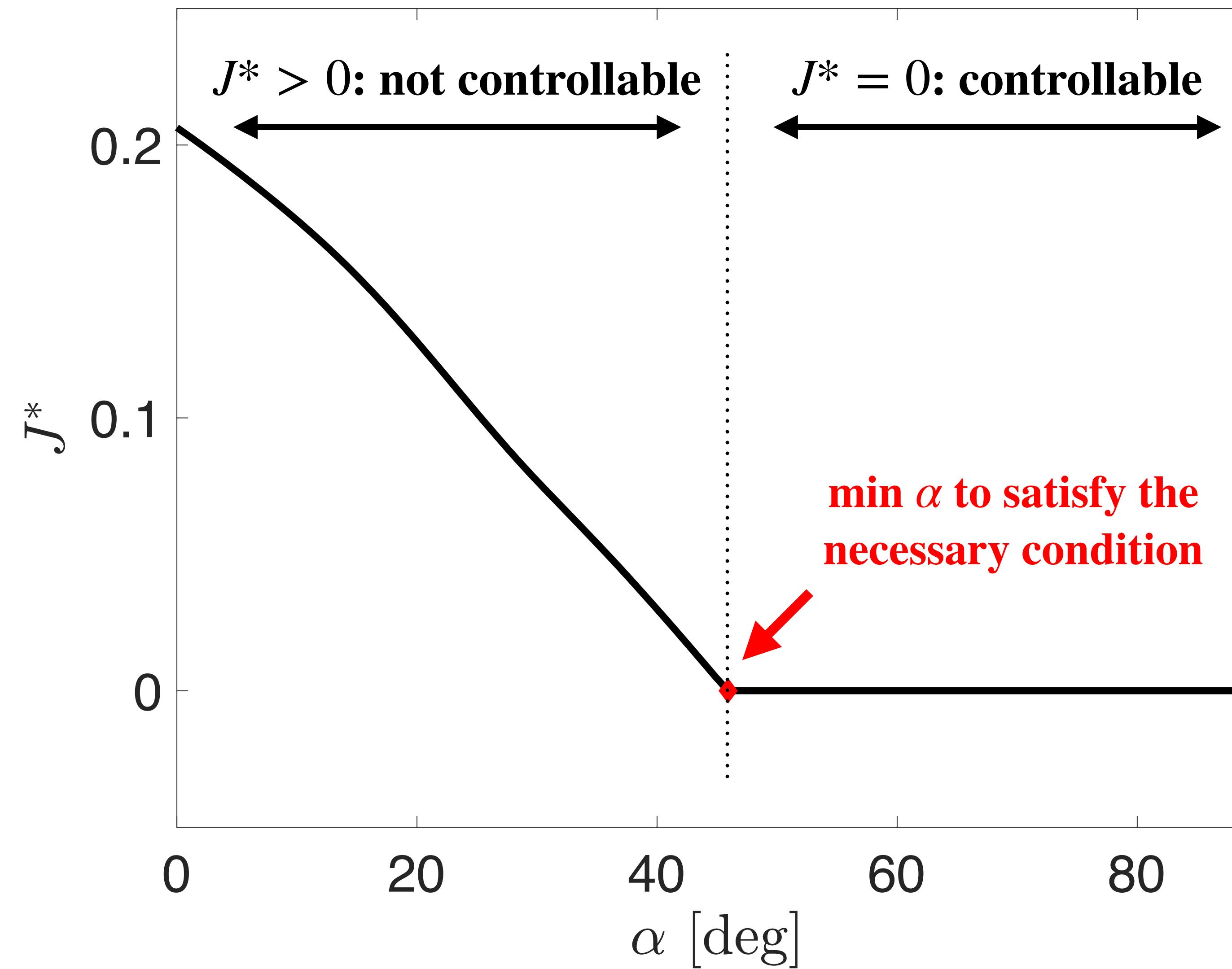
2. Recast into a convex optimisation problem

$$\max_{J, \|p_I\| \leq 1} J \quad \text{s.t.}$$

$$\left\langle p_I, \frac{dI(f, \mathbf{u})}{dt} \right\rangle \geq J, \quad \forall f \in \mathbb{S}^1, \forall \mathbf{u} \in \partial K_\alpha, \|\mathbf{u}\| = 1$$

If $J^* > 0 \rightarrow$ not locally controllable

2. Exploitation: minimum optical requirements



2. Recast into a convex optimisation problem

$$\max_{J, \|p_I\| \leq 1} J \quad \text{s.t.}$$

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Difficulties:

$$u(f) \in \partial K_\alpha, \quad f \in \mathbb{S}^1$$

↓

positivity
constraint

↓

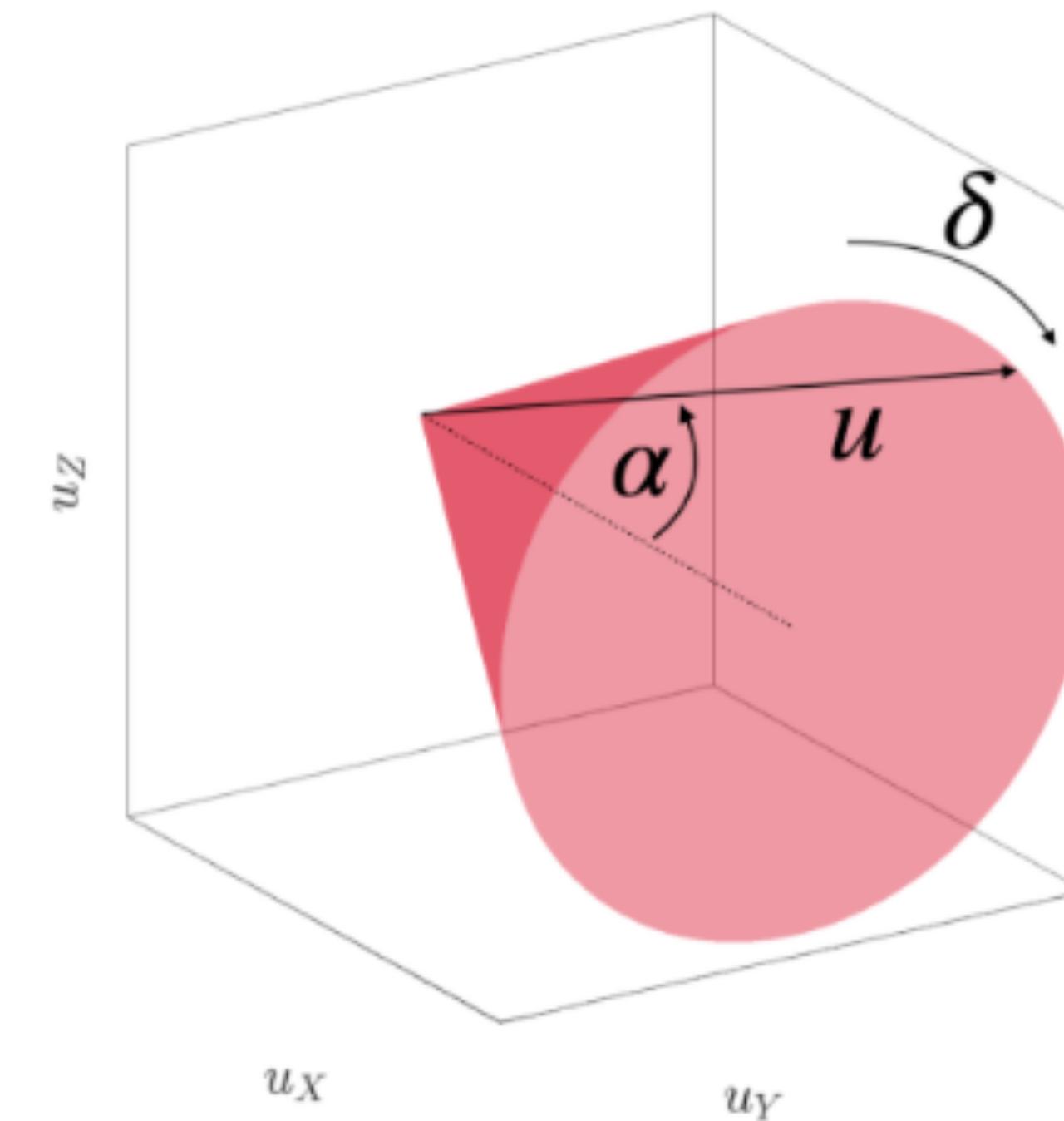
infinite
dimension

2. Numerical solution of the semi-infinite problem

Parametrization of the cone

Controls are combinations of generators εu

$$u = \begin{bmatrix} \cos \alpha \\ \cos \delta \sin \alpha \\ \sin \delta \sin \alpha \end{bmatrix}$$



2. Numerical solution of the semi-infinite problem

Fourier transform (exact) of the dynamics

$$\frac{dI}{dt} = \varepsilon \sqrt{\frac{a(1 - e^2)}{\mu}} G(I, f) R(I, f) \mathbf{u}$$

$$G = \begin{pmatrix} 0 & 0 & \frac{\sin(\gamma_3 + f)}{\sin \gamma_2 (1 + e \cos f)} \\ 0 & 0 & \frac{\cos(\gamma_3 + f)}{1 + e \cos f} \\ -\frac{\cos f}{e} & \frac{2+e \cos f}{1+e \cos f} \frac{\sin f}{e} & \frac{\cos(\gamma_3 + f)}{1 + e \cos f} \\ \frac{2ae}{1-e^2} \sin f & \frac{2ae}{1-e^2} (1 + e \cos f) & 0 \\ \sin f & \frac{e \cos^2 f + 2 \cos f + e}{1 + e \cos f} & 0 \end{pmatrix}$$

$(1 + e \cos f) G(I, f) R(I, f)$ is a trigonometric polynomial of degree 2 in f

2. Numerical solution of the semi-infinite problem

Positive polynomials [Nesterov, 2000; Dumitrescu, 2007]

Leverage on formalism of squared functional systems:

$$\begin{aligned}\Phi(f, \delta) &= \left[1, e^{i\delta} \right]^T \otimes \left[1, e^{if}, e^{2if} \right]^T \\ &= \left[1, e^{if}, e^{2if}, e^{i\delta}, e^{if} e^{i\delta}, e^{2if} e^{i\delta} \right]^T\end{aligned}$$

$\Lambda_H : \mathbb{C}^N \rightarrow \mathbb{C}^{N \times N}$ a linear operator s.t. $\Lambda_H(\Phi(f, \delta)) = \Phi(f, \delta)\Phi^H(f, \delta)$

and $\Lambda_H^* : \mathbb{C}^{N \times N} \rightarrow \mathbb{C}^N$ its adjoint operator.

2. Numerical solution of the semi-infinite problem

Positive polynomials [Nesterov, 2000; Dumitrescu, 2007]

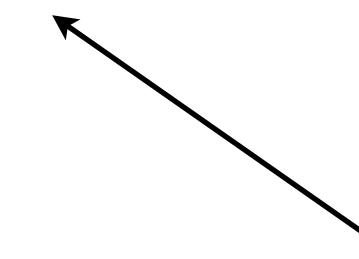
Then for any trigonometric bivariate polynomial

$$\langle p_I, \tilde{G}(I, f)\mathbf{u} \rangle - J \geq 0, f \in \mathbb{S}^1, \mathbf{u} \in \partial K_\alpha \Leftrightarrow \exists Y \succeq 0$$

such that $\tilde{G}\mathbf{u} p_I - e_1 J = \Lambda_H^*(Y)$



Fourier coefficients in the basis Φ



defined with Toeplitz matrices

2. Recast into a convex optimisation problem

$$\max_{J, \|p_I\| \leq 1} J \quad \text{s.t.}$$

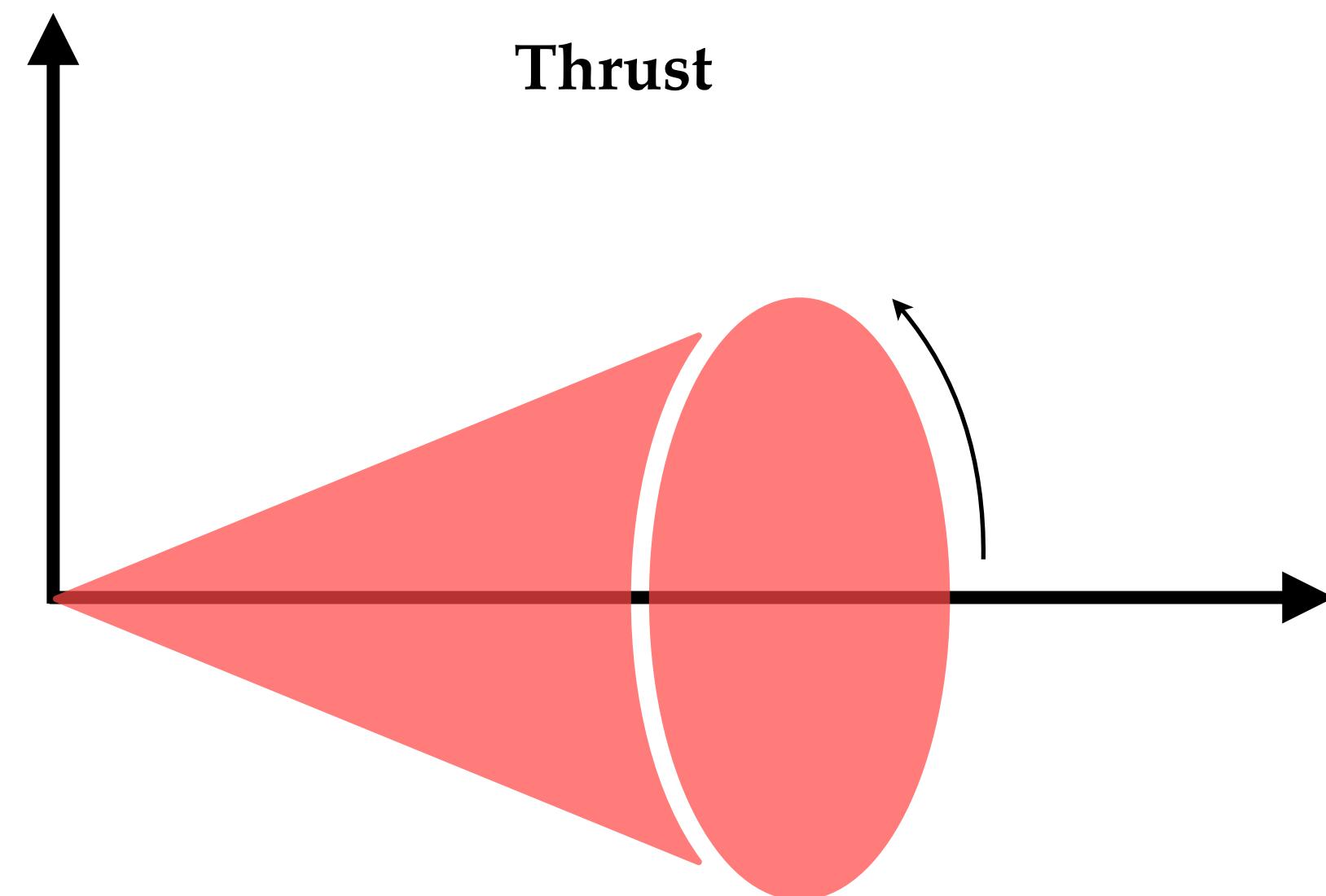
$$\left\langle p_I, \frac{dI(f, u)}{dt} \right\rangle \geq J, \quad \forall f \in \mathbb{S}^1, \forall u \in \partial K_\alpha, \|u\| = 1$$



LMI

2. Effective test of the necessary condition

$\min_{u \in K_\alpha} \alpha$ such that the system is controllable

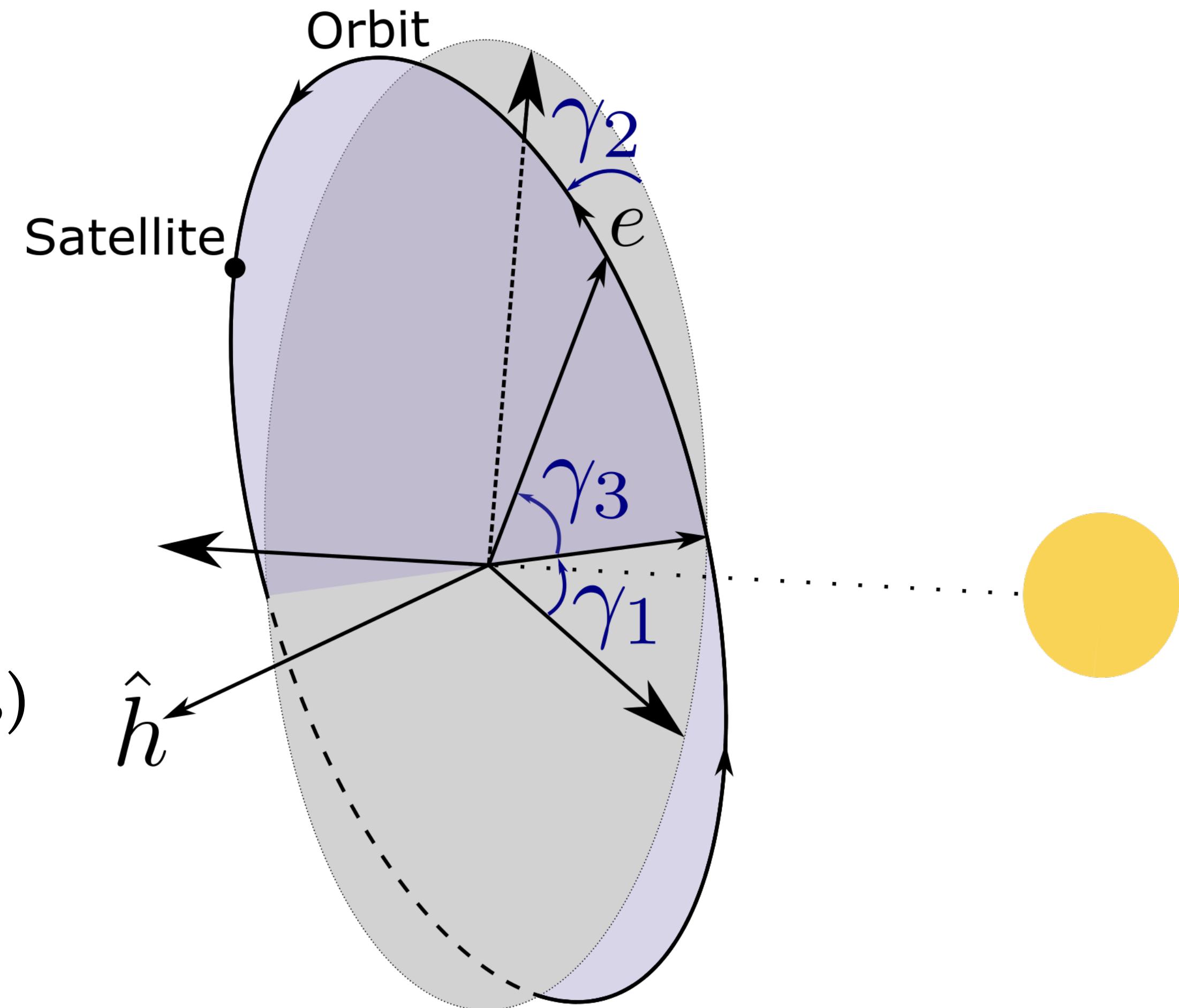


2. Convenient choice of orbital elements

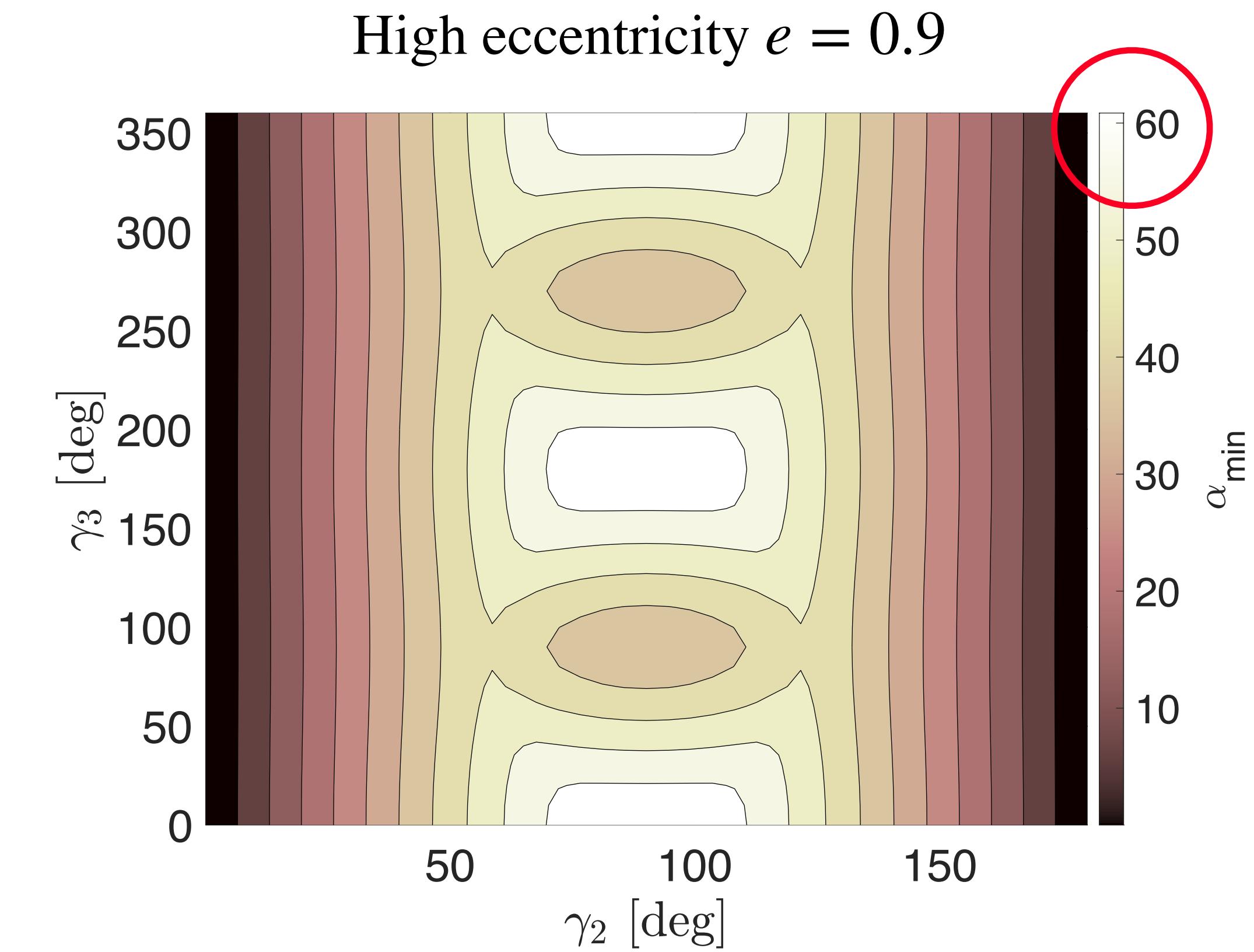
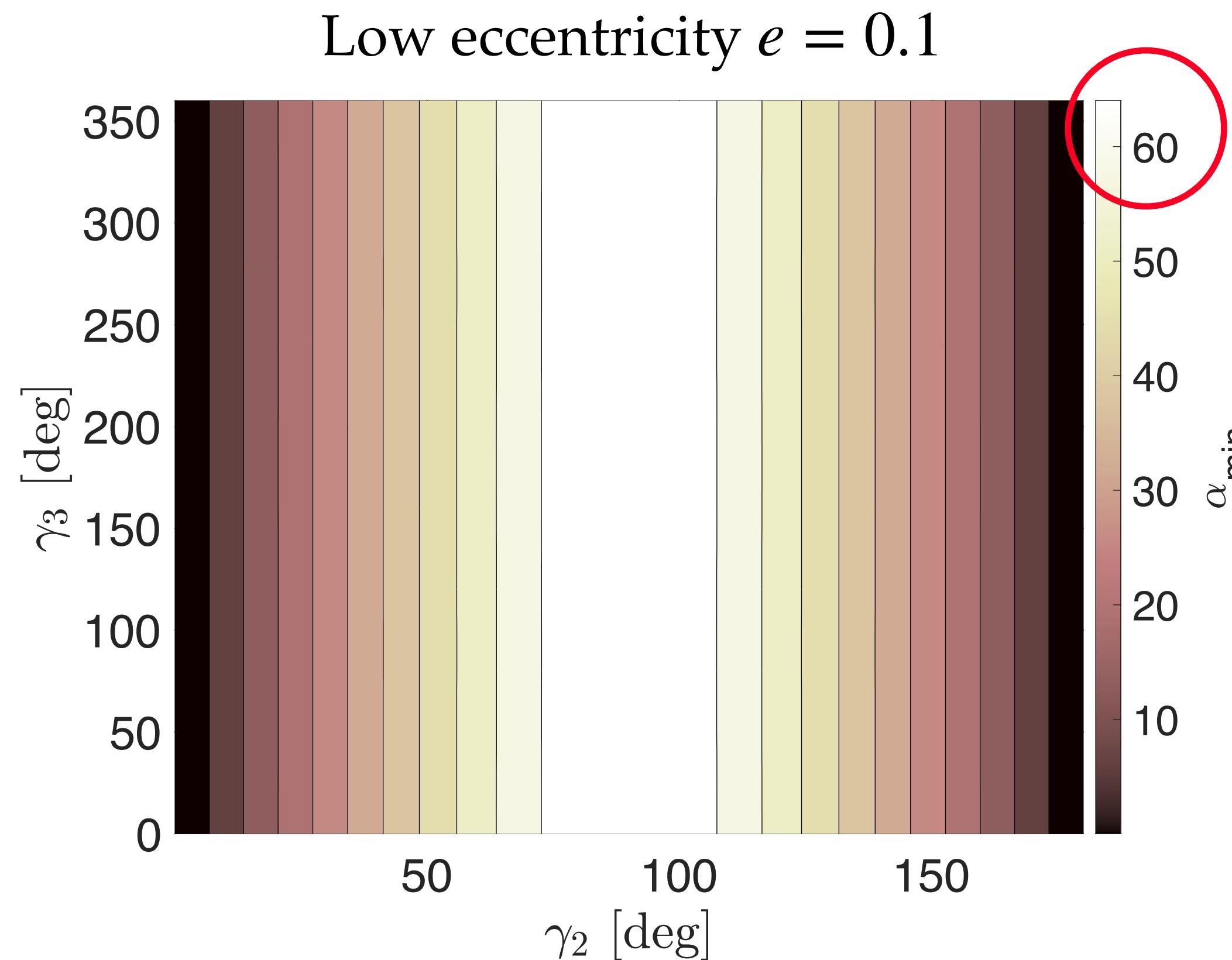
$$I = (\gamma_1, \gamma_2, \gamma_3, a, e)$$

Problem independent of:

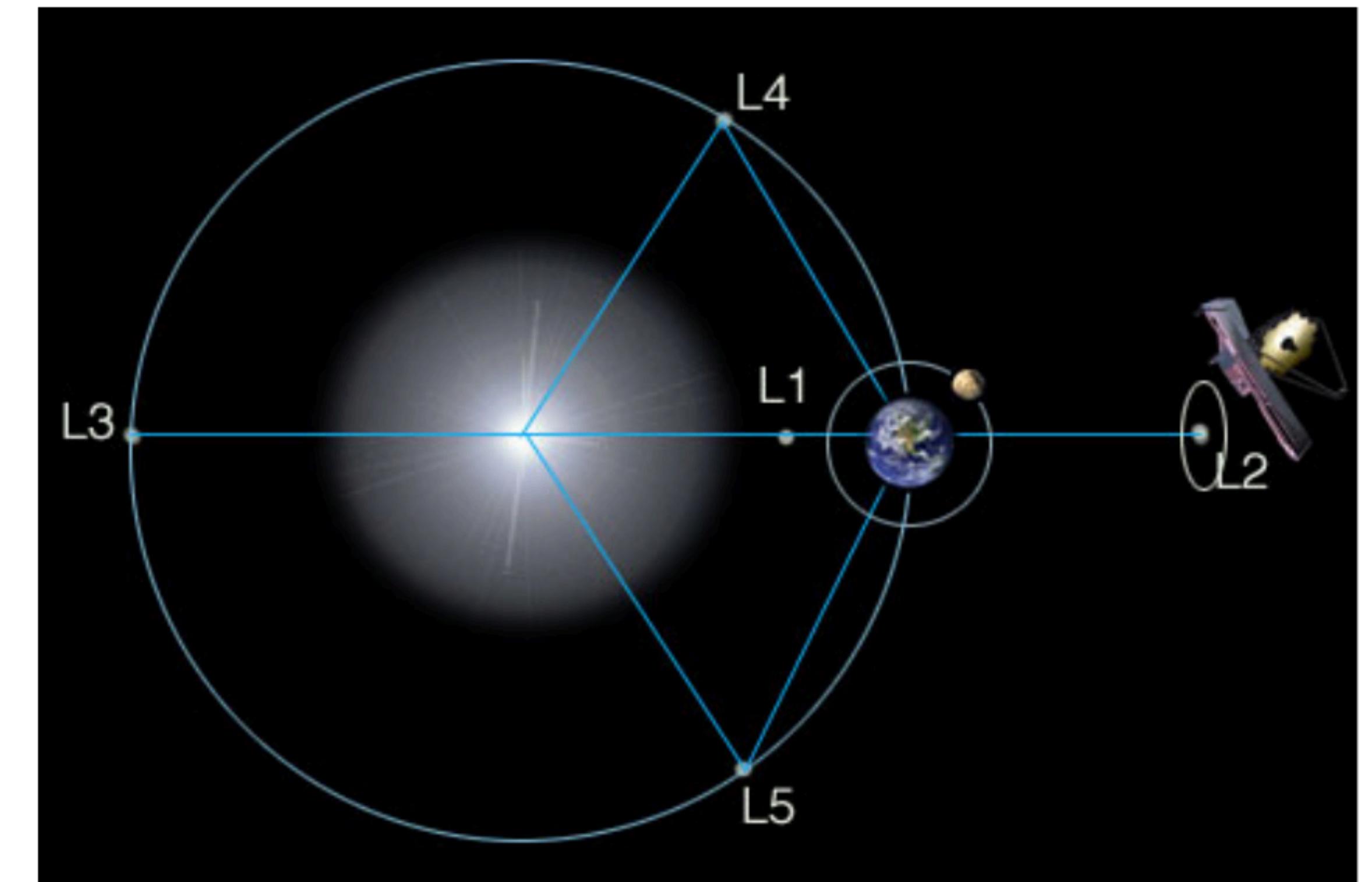
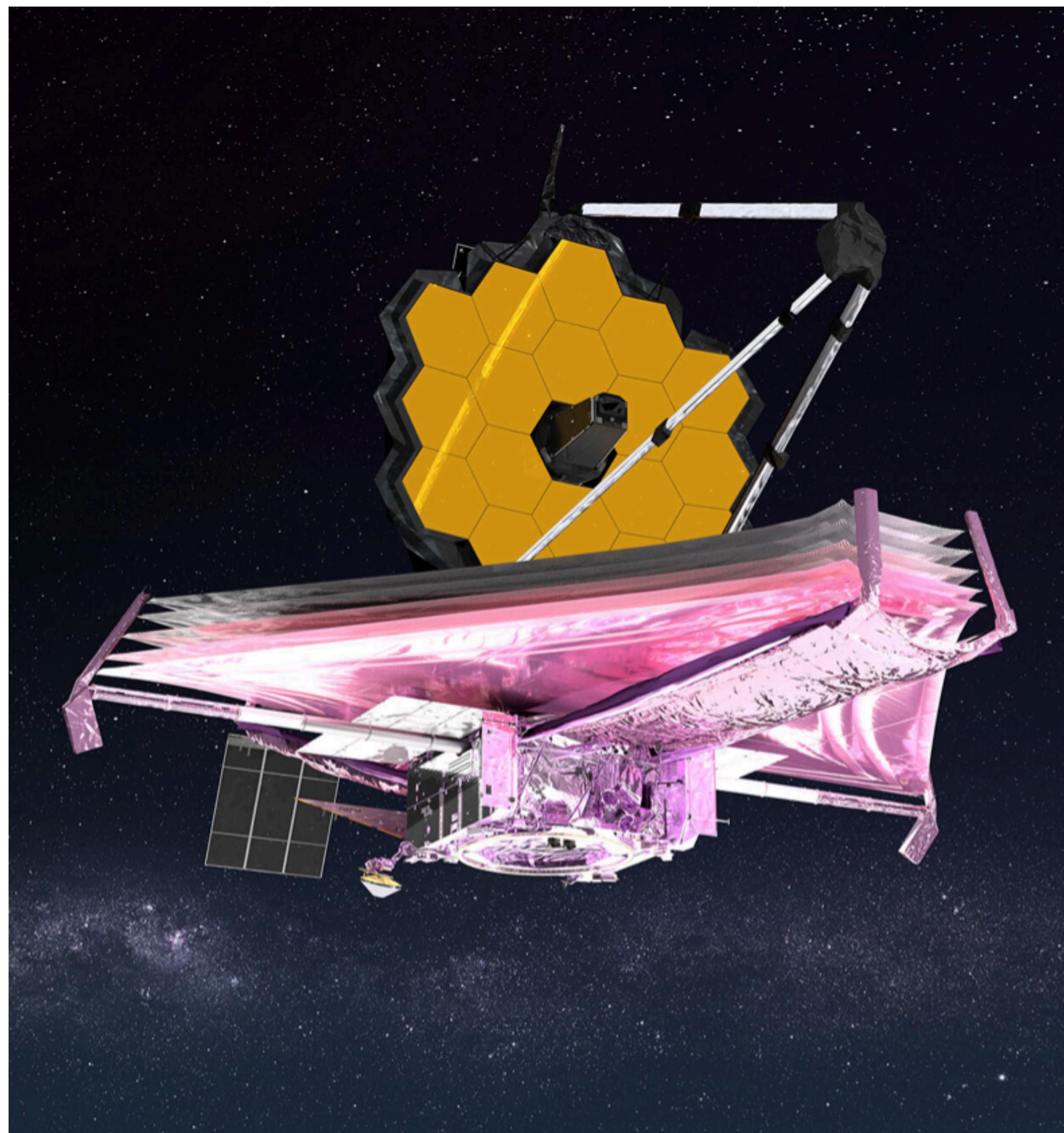
- γ_1 (axial symmetry)
- $a, \mu \rightarrow$ (planet-independent results)



3. Results: minimal requirement

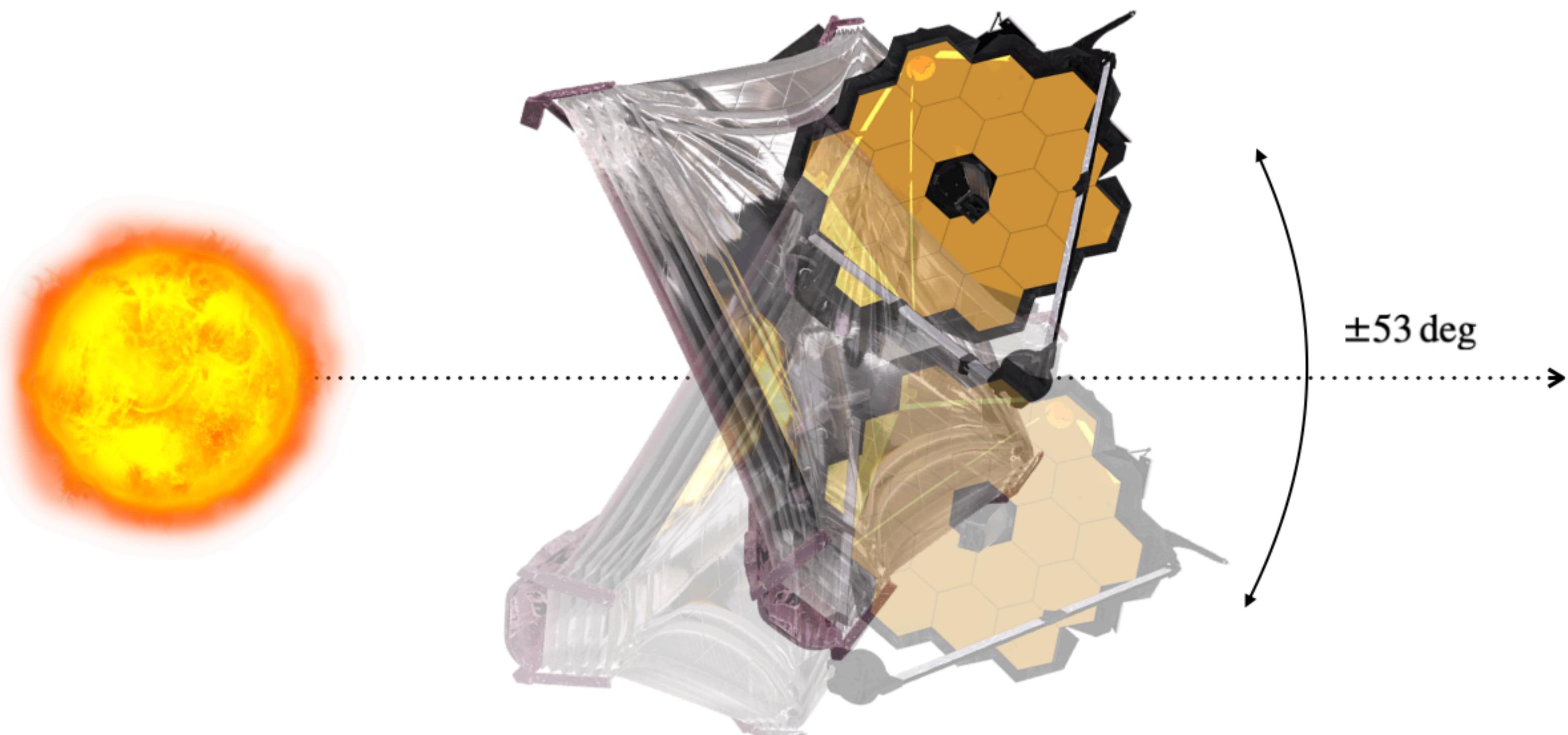


4. Other systems with the same constraint

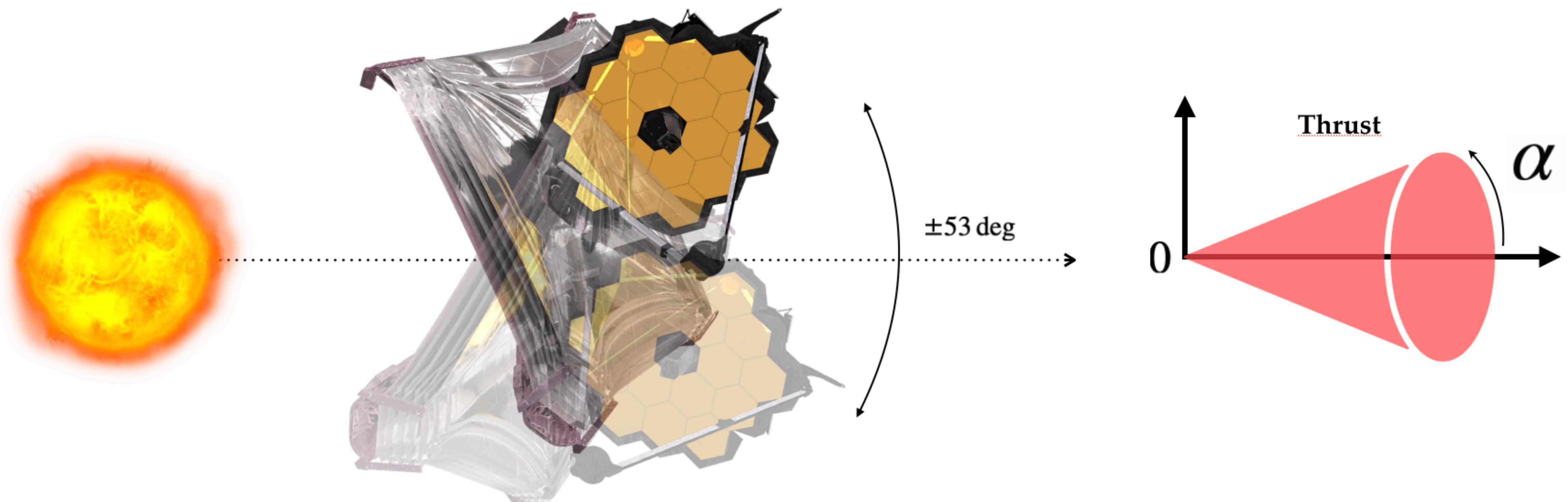


Source: NASA

4. James Webb Space Telescope attitude constraint



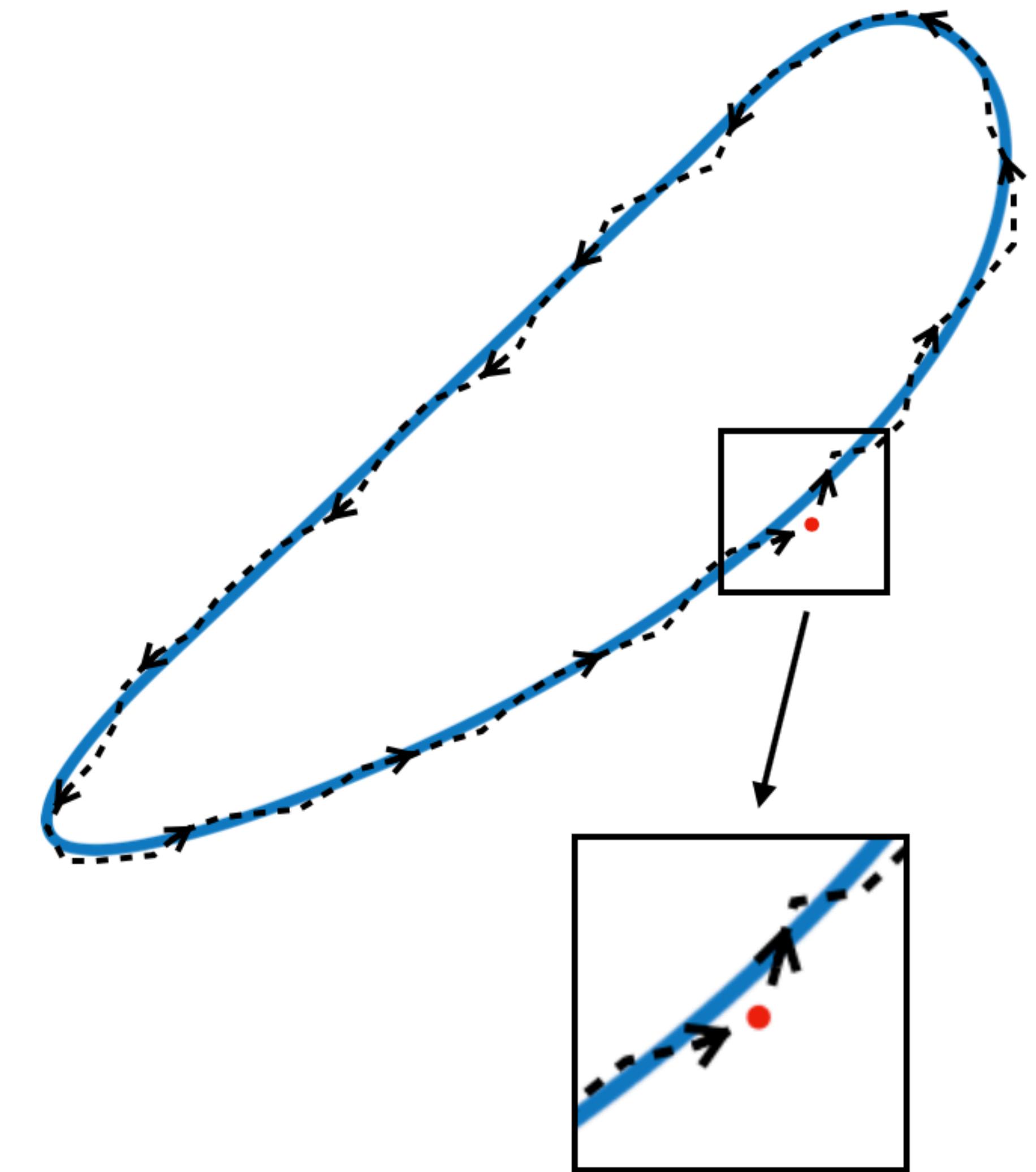
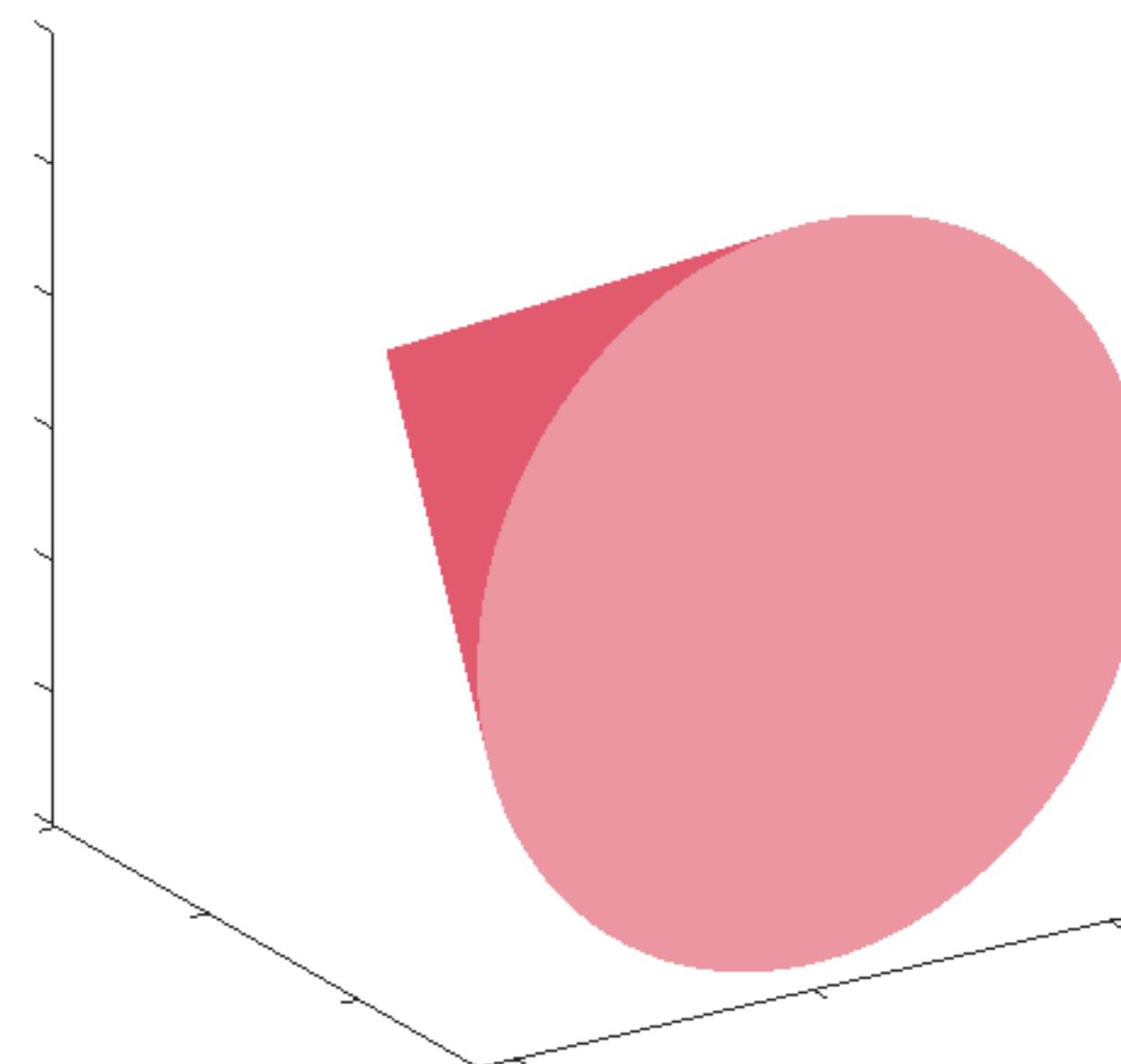
4. James Webb Space Telescope attitude constraint



4. Dynamics of any periodical orbit

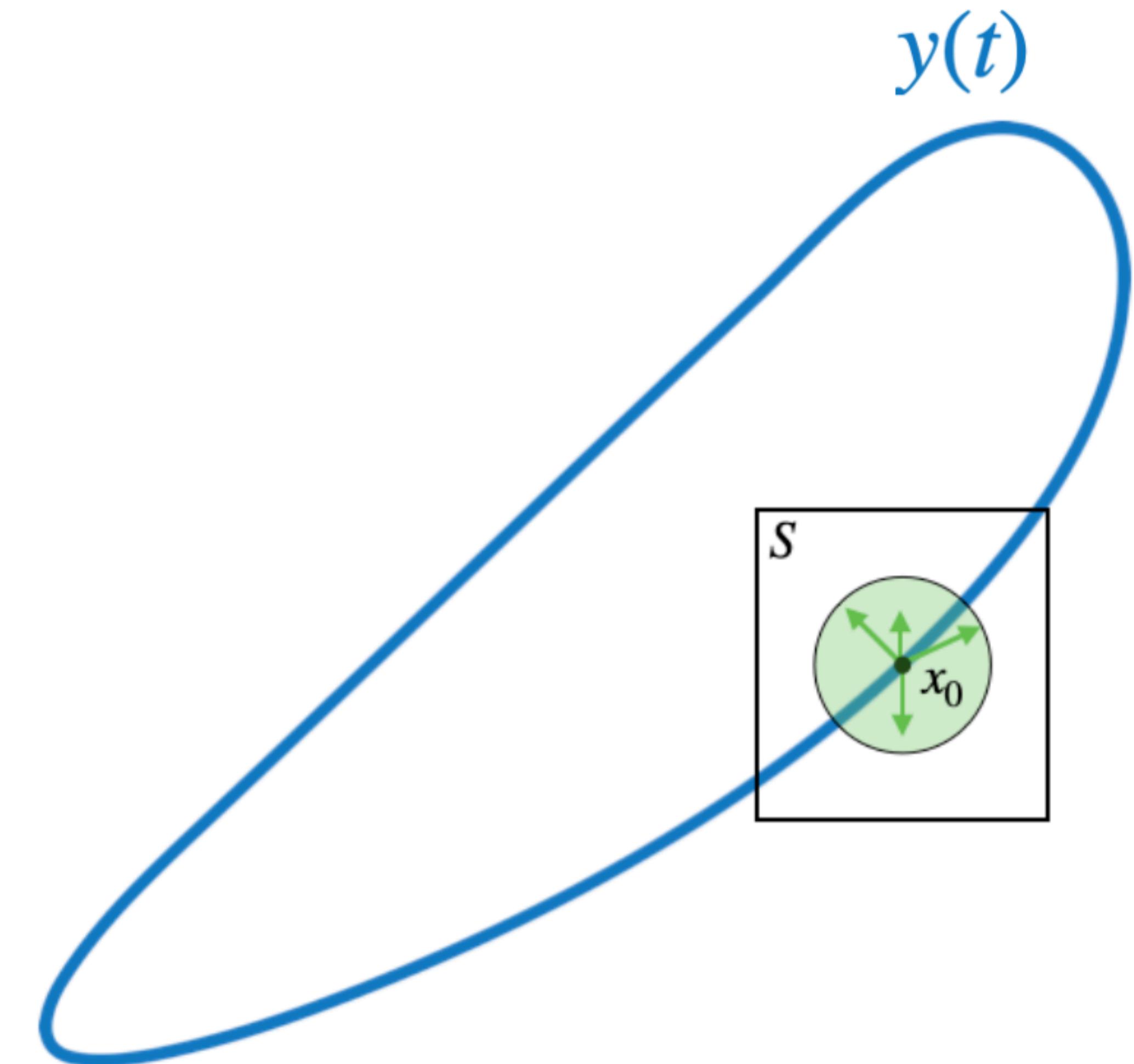
$$\frac{dx}{dt} = f(x) + B(x) u, \quad x \in M \subseteq \mathbb{R}^n$$

$$u \in K_\alpha \subseteq \mathbb{R}^m, \|u\| \leq \varepsilon$$



4. Linearization over the periodical trajectory

$$\begin{cases} \frac{dy}{dt} = f(y) \\ y(0) = y(T) = x_0 \\ S(x_0) = 0 \end{cases}$$

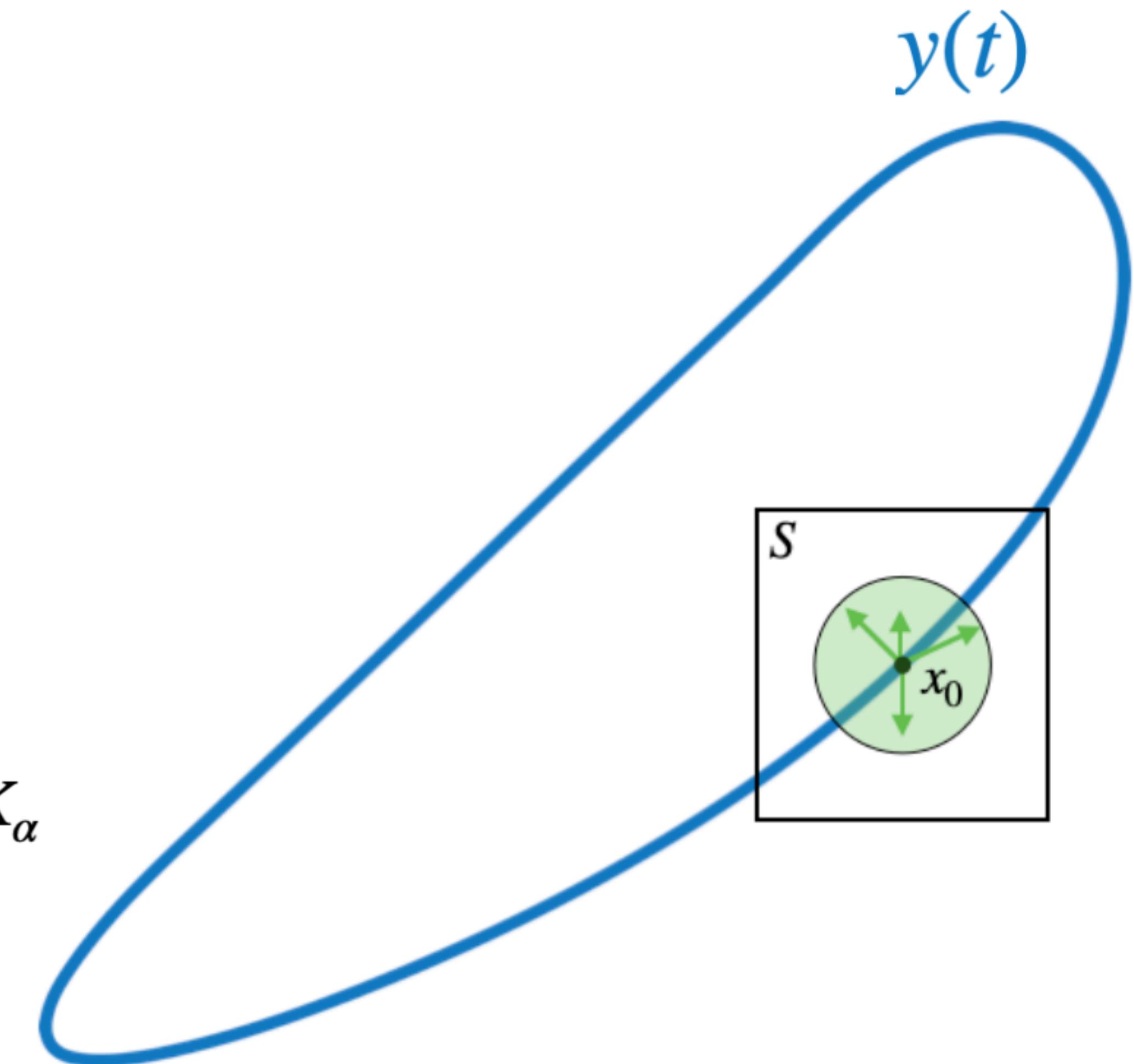


4. Linearization over the periodical trajectory

$$\delta x(t) = \Phi(t, x_0) \delta x_0$$

$$\frac{d\delta x}{dt} = \left. \frac{df}{dx} \right|_y \delta x + B(y) u$$

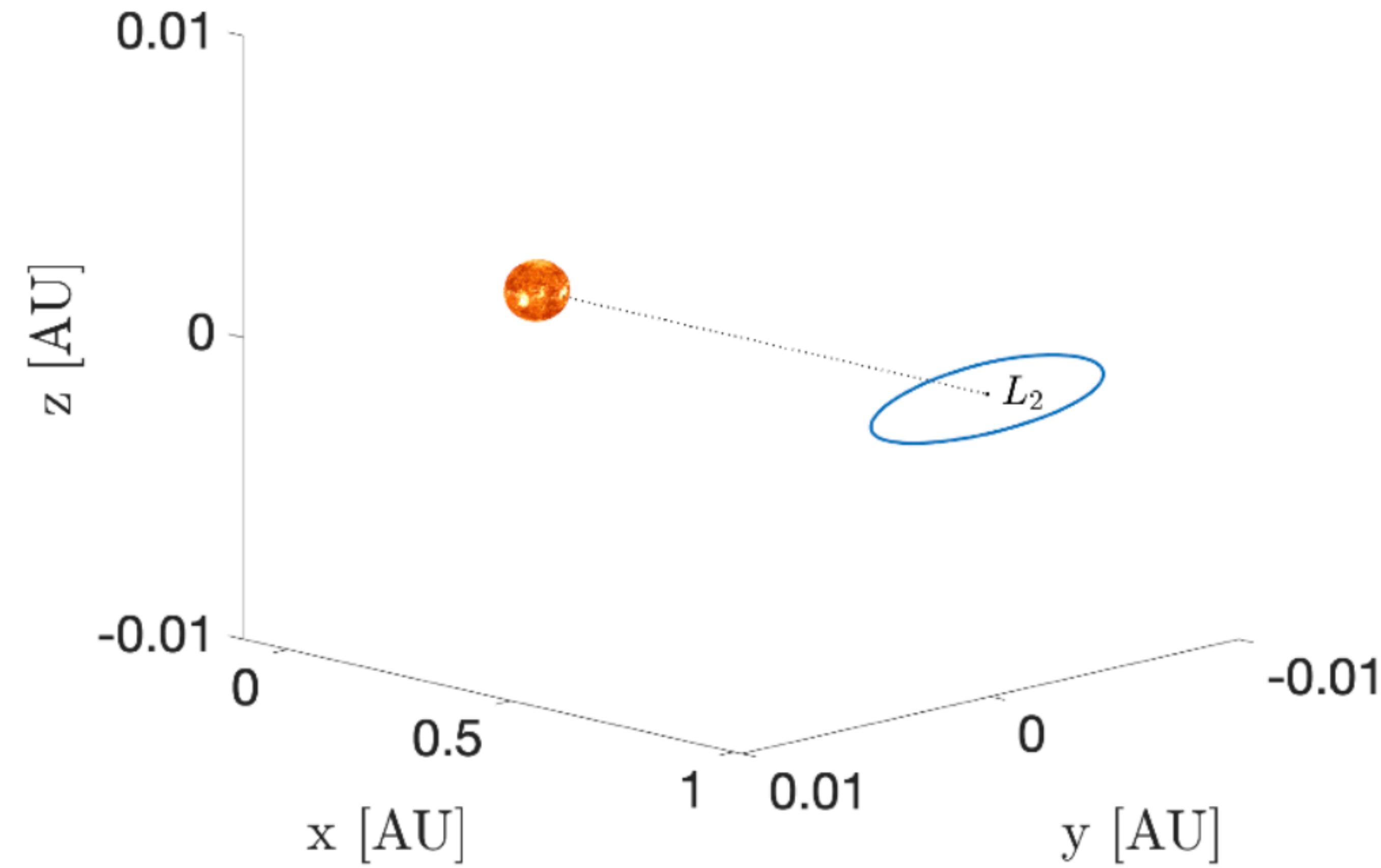
$$\frac{d\delta x_0}{dt} = \Phi^{-1}(t, x_0) B(y(t)) u, \quad \delta x_0 \in T_{x_0} S, \quad u \in K_\alpha$$



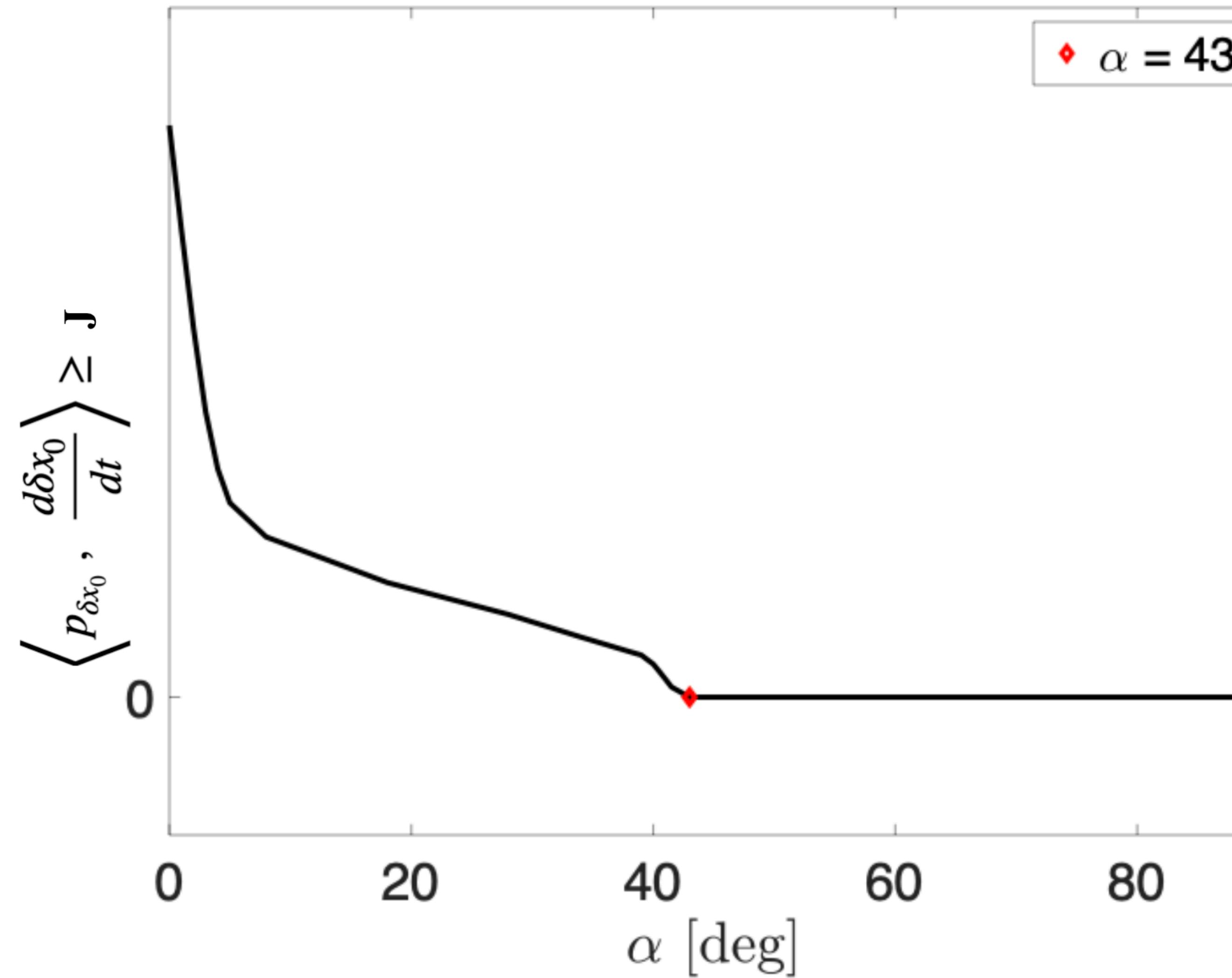
4. Difference with the previous problem

Not exact trigonometrical polynomials → truncation of the Fourier series

4. Halo orbit around L₂



4. Minimum cone angle for thrust directions



References

- Caillau, J.-B.; Dell'Elce, L.; Herasimenka, A.; Pomet, J.-B. On the controllability of nonlinear systems with a periodic drift. *SIAM J. Control Optim.*, to appear.
- Caillau, J.-B.; Dell'Elce, L.; Herasimenka, A.; Pomet, J.-B. Optimal control of a solar sail. *Optimal Control Appl. Methods*, **45** (2024), 2837-2855.
- Herasimenka, A.; Dell'Elce, L.; Caillau, J.-B.; Pomet, J.-B. Controllability Properties of Solar Sails. *J. Guidance Control Dyn.* **46** (2023), no. 5, 900-909.
- Herasimenka, A.; Dell'Elce, L.; Farres, A. Controllability of satellites on periodic orbits with cone-constraints on the thrust direction. *Space Flight Mechanics Meeting*, Austin, 2023.