

Design and Fabrication of RC Delivery Aircraft: Col. Pollo J. Rosso



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Main Aircraft Specifications

| | | | | | |
|-----------------------------|-------|-----------------------|----------|----------------|------|
| Empty Weight (kg) | 1.4 | Chord (m) | Span (m) | Area (m^2) | AR |
| Gross Weight (kg) | 2.125 | Wing | 0.2 | 1.35 | 0.27 |
| Payload 1 Weight (kg) | 0.68 | Horizontal Stabilizer | 0.12 | 0.58 | 0.07 |
| Payload 2 Volume (cm^3) | 700 | Vertical Stabilizer | 0.15 | 0.27 | 0.04 |
| Thrust-to-weight | 0.5 | | | | 1.8 |
| Cruise Speed (m/s) | 20 | | | | |



Mission

Location: Mission Bay Field (SEFSD)

- Mission 1:** Perform successful empty weight flight
- Mission 2:** Take off with both payload 1 and 2, successfully eject payload 2 midair at a designated point, land safely with payload 1

$$J(x) = \frac{r_{wp} W_p(x) + r_{vp} V_p(x) - c_e E_f(x) - c_c}{T_f(x)} - \frac{c_{wg} W_g(x)}{t_l} - c_f$$

Flight Score

Key Variables: W_p = payload 1 weight, V_p = payload 2 volume, T_f = flight time, W_g = gross weight, E_f = energy consumption

Design Decision: We focused on maximizing the weights carried, the key factor influencing our flight score

Aerodynamics

Wing Airfoil Selection: NACA 4412

Key Aerodynamic Parameters:

- $C_{Lmax} = 1.3622$, $C_{Dmin} = 0.0363$, $C_{D0} = 0.035$
- Maximum L/D of 35.838 @ 13.25° AOA
 - Taildragger design achieves desired angle

Empennage Airfoil Selection: NACA 0012

Planform Area (in % of wing):

- Horizontal Stabilizer: 19%, Vertical Stabilizer: 11%

Propeller Choice: At 9000 RPM, the 9x6E easily meets the 3N minimum thrust for nominal cruise speed

Stability & Performance

Stability Parameters (% from MAC):

- Neutral Point: 58
- CG (Empty Flight): 24.73
- CG (Payloads 1 & 2): 24.04
- CG (Payload 1): 23.97

Takeoff Distance: 14.7 m

Cruise Speed: 20 m/s

Stall Speed: 10 m/s

Climb Angle: 15°

