

# Data Analysis Techniques for Continuous Gravitational Wave Searches

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# Abstract

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# Acknowledgements

# Declaration

*DECLARATION*

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# Acronyms

**A | B | C | E | F | G | L | M | N | P | R | S | U**

**A**

**ASD** amplitude spectral density. [vii](#), [103](#)

**B**

**BBH** binary black hole. [1](#), [6](#), [7](#)

**BNS** binary neutron star. [1](#), [6](#), [8](#)

**C**

**CBC** compact binary coalescence. [5–7](#), [62](#), [67](#), [68](#)

**CFS** Chandrasekhar, Friedman and Schutz. [10](#)

**CMB** cosmic microwave background. [7](#)

**CNN** convolutional neural network. [v–vii](#), [25](#), [67–69](#), [74](#), [78](#), [81–84](#), [87–99](#)

**CW** continuous gravitational wave. [v–vii](#), [1](#), [2](#), [5–8](#), [16](#), [22–26](#), [28](#), [54](#), [59](#), [61](#), [62](#), [65–70](#),  
[74](#), [78](#), [81](#), [84](#), [87](#), [90](#), [95–97](#), [99](#), [102](#), [103](#), [105–107](#), [116](#), [122](#), [124](#)

**E**

**EM** electromagnetic. [7](#), [68](#)

**EOS** equation of state. [6](#), [8](#)

**F**

**FFT** fast Fourier transform. [vii](#), [44](#), [45](#), [47](#), [48](#), [56](#), [68](#), [69](#), [106](#), [107](#), [116](#), [118](#), [121](#)

**G**

**GR** general relativity. 1

**GRB** gamma ray burst. 7

**GW** gravitational-wave. vi, 1, 5–12, 16–18, 23, 27–29, 32, 35, 50, 67, 68, 84, 91, 99, 102–107, 112, 117

## L

**LIGO** Laser Interferometer Gravitational-wave Observatory. vi, vii, 1, 11–13, 15, 16, 23, 25, 27, 29, 49, 54, 59, 62, 67, 81, 84, 87, 96, 103, 104, 106–108, 110–112

**LISA** laser interferometer space antenna. 6, 8

## M

**MCMC** Markov-Chain Monte Carlo. 21, 52

**MDC** mock data challenge. vii, 25, 47, 52–55, 57, 59, 62, 64, 65, 85, 89, 90, 95

**MSU** million standard units. 25

## N

**NSBH** neutron star black hole. 6

## P

**PEM** physical environment monitor. 105

**PSD** power spectral density. 32, 46, 47, 50, 67, 84, 91, 116, 117

## R

**RMS** root median square. 51, 53–55, 65

## S

**SFT** short Fourier transform. 23, 24, 28, 32, 36, 37, 39–41, 43, 46, 47, 49, 51, 53, 54, 62, 64–66, 87, 88, 93, 107, 108, 110–113

**SNR** signal-to-noise-ratio. 16, 23, 39–41, 43, 46–48, 50–60, 62, 65, 84, 87, 89–94, 97–99, 116–118, 121, 123

**SSB** solar system barycenter. 17

## U

**UCD** up, centre or down. 31, 34, 35, 38, 39

# Chapter 1

## Introduction

Gravitational-waves (GWs) were first predicted in 1915 as a consequence of Einstein’s general theory of relativity [1]. They are theorised as ripples in the fabric of space-time. The first observational evidence that GWs exist came from observations of the Hulse-Taylor binary [2, 3]. This observation showed a binary pulsar system whose time of periastron was advancing. i.e. the orbit was decaying. This implies that the periastron time separation was decreasing with time. If the separation of two orbiting objects is decreasing then the system is losing energy somewhere. The loss in energy matched the general relativity (GR) prediction which assumed the energy was lost to GWs. This gave hope of GWs existence and helped lead the way to designing instruments which could directly detect them. The first direct detection of gravitational waves was made in 2015 when the two Laser Interferometer Gravitational-wave Observatory (LIGO) detectors in the US [4] identified a signal from a binary black hole (BBH) system. This was not only the first observation of GWs but gave information on a yet unobserved astrophysical system. This has since been followed by many more detections of BBH signals involving LIGO and Virgo including [5, 6]. In 2017 the LIGO detectors observed the first binary neutron star (BNS) system [7] which had a corresponding electromagnetic counterpart. This allowed verification of the source from optical counterparts and started the era of multi-messenger astronomy. These detections opened up the field of gravitational wave astronomy, where many more detections are expected to give more information on the universe and objects within it.

As well as searching for BBH and BNS signals, there are many efforts to detect other types of GW signals. This thesis focuses on efforts to search for a particular type of GW which are thought to originate from rapidly rotating neutron stars. In chapters 1 and 2 I will review introductory material. This includes a general introduction to the generation of GWs in Sec. 1.1 and their sources in Sec. 1.2. I will then introduce instruments used to detect GW in Sec. 1.3. In Chapter 2 I will introduce the general model for continuous gravitational waves (CWs) and current methods used to detect them. Chapters 3, 4 and

[5](#) will go into detail about techniques developed by the author to search for [CW](#) signals. Finally I will summarise this work and discuss future developments in chapter [6](#).

## 1.1 Gravitational waves

In general relativity, gravity is thought of as the curvature of space-time and matter moves according to this curvature. The matter in the universe also has an effect on the curvature of the space-time. The larger the mass of matter the more the space-time is distorted. Space-time can generally be described by Einstein's field equations,

$$G_{\mu\nu} = \frac{8\pi G}{c^4} T_{\mu\nu}. \quad (1.1)$$

where  $G_{\mu\nu}$  is the Einstein tensor and  $T_{\mu\nu}$  is the stress-energy tensor. The stress energy tensor essentially describes the matter and energy in the universe. Its components contain information on the density of energy and momentum. The Einstein tensor contains information on the curvature of the universe. This can be derived directly from the metric tensor  $g_{\mu\nu}$  which describes the geometry of the universe. Einstein's equations then explain how the curvature of space-time changes with the energy and matter within it. In empty space one can assume that the geometry of space-time is flat, i.e. there is no curvature to space-time. The metric tensor for this can then be defined as,

$$g_{\mu\nu} = \eta_{\mu\nu} = \begin{pmatrix} -1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \end{pmatrix}. \quad (1.2)$$

Each index of this matrix refers to a space-time dimension, i.e.  $x^0 = t$ ,  $x^1 = x$ ,  $x^2 = y$  and  $x^3 = z$ . Measuring a distance  $dx$  in space-time can be different for different observers, therefore, one needs a measure which is invariant for every observer. This is the space-time interval  $ds$ , also known as the line element, between two 'events' in space-time. This is defined as,

$$ds^2 = g_{\mu\nu} dx^\mu dx^\nu. \quad (1.3)$$

As in Einstein's notation this is a sum over the indices  $\mu$  and  $\nu$ . Eq. [1.3](#) can be thought to describe the space-time 'distance' between the two events. For flat space-time,  $\eta_{\mu\nu}$ , this can then be written as,

$$ds^2 = -c^2 dt^2 + dx^2 + dy^2 + dz^2. \quad (1.4)$$

The Einstein equations Eq. [1.1](#) then demonstrate how the curvature of space-time  $G_{\mu\nu}$  depends on the matter and energy distribution  $T_{\mu\nu}$  within it.

A gravitational wave can be described as a ripple in this space time. The simplest way to visualise this is just a small time dependent change to the flat space-time metric  $\eta_{\mu\nu}$ . In linearised theory of gravity, the space-time metric  $g_{\mu\nu}$  can be defined as,

$$g_{\mu\nu} = \eta_{\mu\nu} + h_{\mu\nu}, \quad (1.5)$$

where  $\eta_{\mu\nu}$  is the metric for flat space-time and  $h_{\mu\nu}$  is some perturbation, where  $|h_{\mu\nu}| \ll 1$  [8]. In this linearised theory the perturbations to the metric tensor are assumed to be small, therefore, Einstein's field equations can be solved such that the solution is a plane wave. I will skip over lots of the calculation here, however, more information on this derivation can be found in [8] **JOE: more references** By using  $g_{\mu\nu}$  from Eq. 1.5, we can write the linearised Einstein equations as,

$$\square h_{\mu\nu} = -16\pi T_{\mu\nu}, \quad (1.6)$$

where  $\square$  is the d'Alembert operator which in flat space is defined by,

$$\square = -\frac{1}{c^2} \frac{\partial^2}{\partial t^2} + \frac{\partial^2}{\partial x^2} + \frac{\partial^2}{\partial y^2} + \frac{\partial^2}{\partial z^2} \quad (1.7)$$

In empty space there is no matter, therefore, all the components of the stress energy tensor are zero, i.e.  $T_{\mu\nu} = 0$ . This allows Eq. 1.6 to be reduced to,

$$\square h_{\mu\nu} = 0. \quad (1.8)$$

This then takes the form or a common physics problem known as a wave equation. This follows the same form as in electrodynamics, therefore, the general solutions can be written down as,

$$h_{\mu\nu} = A_{\mu\nu} e^{ik_\alpha x^\alpha}, \quad (1.9)$$

where each component of  $h_{\mu\nu}$  is a sinusoid travelling along vector  $k_\alpha$  with amplitude  $A_{\mu\nu}$  [9]. At this point the set of equations are not simple, the symmetric tensor  $A_{\mu\nu}$  has 10 independent components. This can be greatly simplified by choosing a different gauge (coordinate system) where the metric perturbation is both transverse and traceless (TT) [8]. This is just a choice of coordinate system which does not change any current assumptions. A traceless metric is one where the sum of the diagonal elements are 0 and a transverse metric is when the oscillations are perpendicular to the direction of travel. This gauge imposes two conditions: one is that  $h_{\mu\nu}$  is traceless, i.e. that the sum of the diagonal elements are 0 and the other is that  $h_{\mu\nu}$  is transverse. The transverse element means that the oscillations of the wave happen perpendicular to the direction of travel. At this point we can choose that the wave is travelling in the  $z$  direction which means that

$k = (\omega, 0, 0, k)$ . By then adopting the TT gauge there are only two unique components to the metric such that the perturbation is defined as,

$$h_{\mu\nu} = \begin{pmatrix} 0 & 0 & 0 & 0 \\ 0 & h_+ & h_\times & 0 \\ 0 & h_\times & -h_+ & 0 \\ 0 & 0 & 0 & 1 \end{pmatrix} e^{i(kt-wt)}. \quad (1.10)$$

The two unique components are then the two polarisations of gravitational waves,  $h_+$  and  $h_\times$ . The affect of each of the polarisations on a ring of test particles can be seen in Fig. 1.1 where the gravitational wave is travelling out of the page.

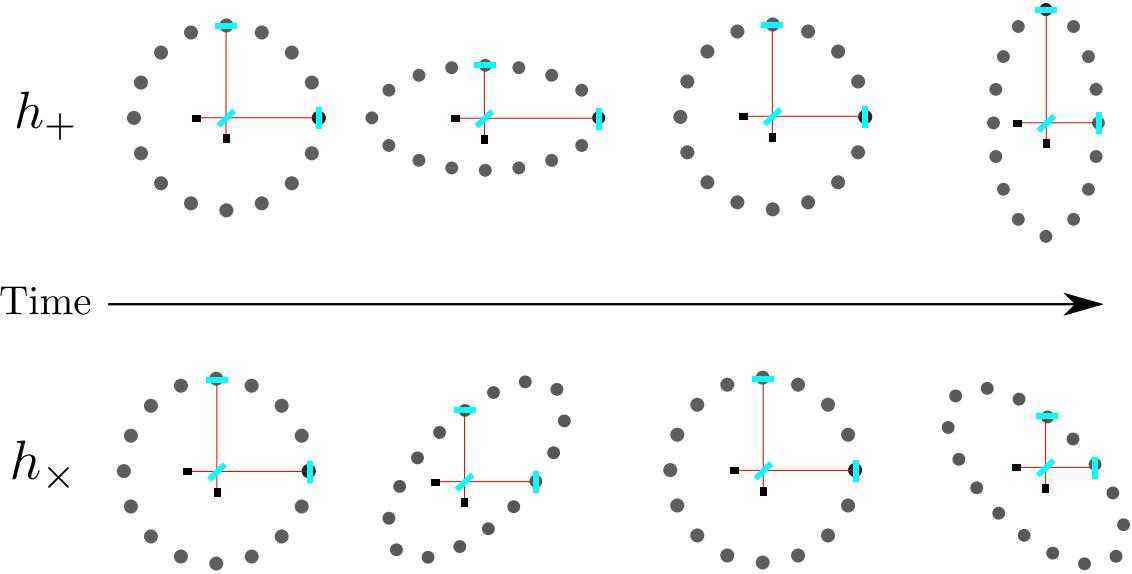


Figure 1.1: Shows how the plus and cross polarisation's affect a ring of test particles. This assumes the wave is travelling out of the page and the effects have been greatly exaggerated. This also shows an example of how this effects the test masses of an interferometer. This will be described in more detail in Sec. 1.3.

### Generating gravitational waves

To generate gravitational waves we go back to Eq. 1.6 where we include the stress-energy term on the right hand side. Following the derivation in [8], one can find that the gravitational wave amplitude is related to the second moment of the mass distribution. The second moment of the mass distribution  $I_{\mu\nu}$  is defined as,

$$I_{\mu\nu}(t) = \int \rho(t, \mathbf{x}) x^\mu x^\nu d^3x, \quad (1.11)$$

where  $\rho$  is the mass density, and  $x_i$  and  $x_j$  are the coordinates [8]. This is the quadrupole moment tensor without the trace subtracted. The gravitational wave amplitude is then defined as,

$$h_{\mu\nu} = \frac{2}{r} \frac{d^2 I_{\mu\nu}(t - r)}{dt^2}. \quad (1.12)$$

This has a slight modification in the TT gauge, see [8], however, has the same relationship between the mass quadrupole and the **GW** amplitude. This shows that for a **GW** to be generated, the second derivative of the mass quadrupole moment is needed. A mass quadrupole moment only exists when the mass distribution is not spherically symmetric. Therefore, a mass which is asymmetric and accelerating will produce a **GW**.

Systems which will produce detectable **GWs** are generally rapidly rotating high mass systems which have some asymmetry around their rotation axis. The sources of these **GW** will be described in the following section.

## 1.2 Sources and signals

There are many potential sources for **GW**. The expected sources can be split into 3 general categories based on their signal type: Transient, Stochastic and **CWs**. These categories are chosen based on the length of the signal and how well modelled the signal is. Fig. 1.2 shows an example of each of the signals and their category. In the sections that follow, I will give an overview of the potential sources of each of these signal categories and their wave-forms.

### 1.2.1 Transient

Transient sources of gravitational waves give short duration signals which are observable from milliseconds to tens of seconds depending on the source. Some of these sources will emit signals for a much longer time, however these are at a lower frequency and lower amplitude and not observable by current ground based detectors. Transient signals can be further split into two categories based on how well they are modelled. **CBCs** have well modelled wave-forms and bursts are generally from un-modelled or unknown sources.

#### Compact Binary Coalescence

**CBCs** originate from the slow in-spiral and merge of two compact objects which are gravitationally bound. The objects in-spiral as they lost energy to the radiation of gravitational waves. Dependent on the masses of the two objects, the gravitational waves generated by the system can be detected by ground based detector such as LIGO [10] and Virgo [11].

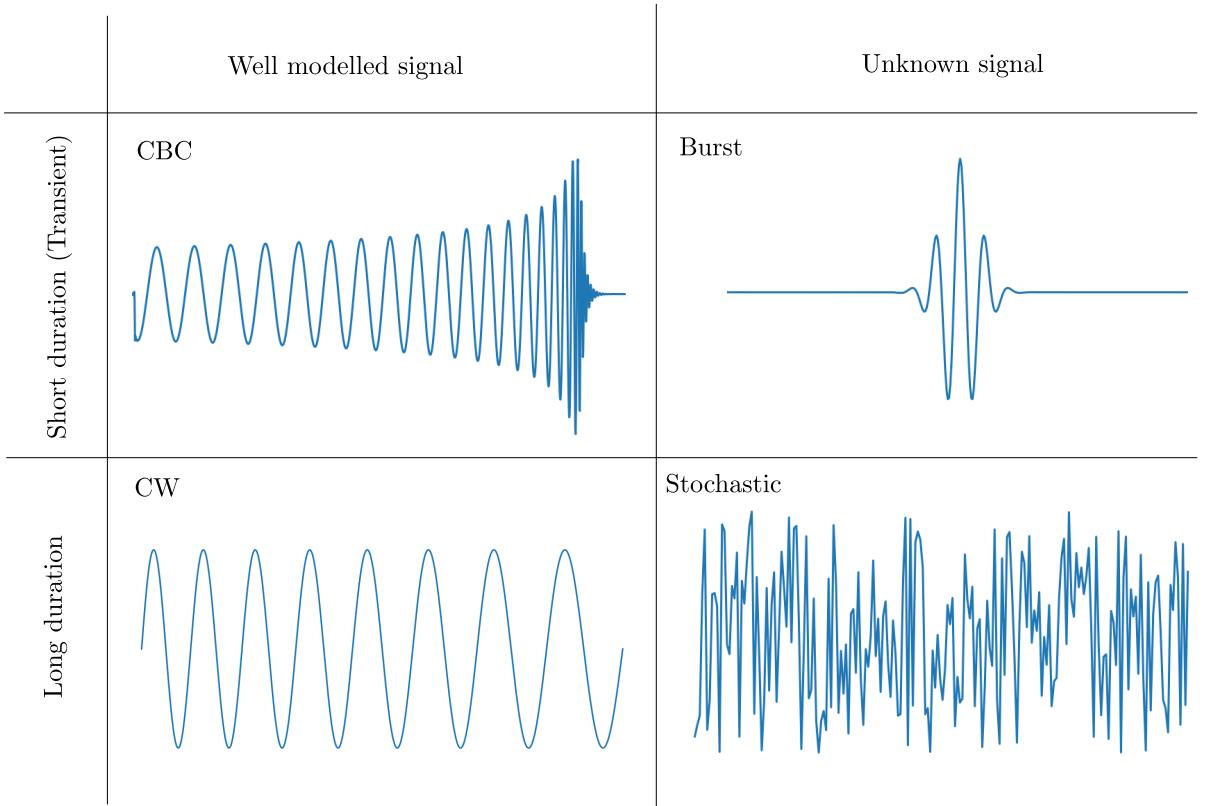


Figure 1.2: Each **GW** signal type can be categorised based on its signal length and how well the signal is modelled. Transient signals which are short duration, include both well modelled **compact binary coalescence (CBC)** signals and unknown **Burst** signals. Long duration signals include well modelled **CW** signals and unknown **Stochastic** signals.

In fact, the only detections to date have been originated from this source which include [4, 5, 7].

The compact objects referred to here are either either black holes or neutron stars. There are generally 3 types of **CBC** source: **BBH**, **BNS** and **neutron star black hole (NSBH)**. The general structure of the wave-form is the same for each of these and follows a ‘chirp’ where the **GW** frequency increases with time until merger. An example of this is shown in Fig. 1.2. For higher mass systems such as **BBH** these signals are detectable by ground based detectors for  $< 1\text{ s}$ . For lower mass systems such as **BNS** they can be detected for longer periods  $\mathcal{O}(10)\text{s}$ . The lower frequency parts of the signal (i.e. earlier times) aim to be detected by future space based detectors such as **laser interferometer space antenna (LISA)** [12].

In systems which have a neutron star, during the in-spiral when the objects are close, the neutron star can deform due tidal interactions between the objects [13, 14]. This becomes useful as it will affect the generated waveform and can help place limits on and determine the **equation of state (EOS)** for the dense matter in a neutron star [13, 14]. **BNS** systems also offer a way to observe objects in multiple different channels, or what is known as multi-messenger astronomy. This is where the object can be viewed in the

electromagnetic (EM) spectrum as well as in gravitational waves. This offers much in the field of astronomy as it can aid in the measurement of the Hubble constant [15]. Observations of BBH systems can also give information on how black holes and BBHs form, more details on this can be found in [16, 17].

### Burst

Burst sources are also short duration however, are un-modelled or difficult to model. This means that the exact wave-form of the signal is unknown. There are a few possible reasons for the lack in knowledge of the waveform: the physics of the system is too complicated to model in a reasonable amount of time or there is no model of the source. As there is no model to generate wave-forms, burst searches cannot use matched filtering as in CBC searches. Rather burst searches look for short bursts in power which is coherent between multiple detectors [18, 19]. There are a number of systems which could potentially emit a short duration burst signals. These include core collapse supernovae [20], gamma ray bursts (GRBs) [21], cosmic strings [22] and other unknown sources. Detecting GW from one of these sources could offer more insight into the processes inside hostile environments.

As burst searches are un-modelled, they are sensitive to almost any signal which is coherent between detectors. This allows them to also search for signals from CBC as well as any GW signal from an unknown source.

#### 1.2.2 Stochastic

The stochastic background has no signal model, however, is expected to be a persistent source of GW in the background of the detector. The stochastic background is the incoherent sum of many unresolved GW signals. The source of these signals can be anything from cosmological sources such as cosmic strings to CBC signals. These signals can be thought of as the GW analogue of the cosmic microwave background (CMB). The signal is assumed to be isotropic such that it can be observed at any point on the sky [23]. As the stochastic background is essentially just noise, it is not possible to detect with a single detector [23]. Rather, searches for the stochastic background correlate signals between multiple detectors [24, 23]. When detected, these signals may be able to offer insights into the early universe and its formation.

#### 1.2.3 Continuous waves

CWs are long duration signals which which can be well modelled. The signals last for times greater than the observation runs of ground based detectors and in general have a fixed or slowly varying frequency. There are a number of potential sources of CWs including CBCs long before their merger. Long before a CBC merger, the two compact objects are

orbiting each other whilst slowly in-spiralling. This will emit a long duration, almost fixed frequency sinusoid as its signal. This signal however, is at lower frequency than ground based detectors can detect, therefore space based detectors such as LISA [12] are expected to observe this type of CW.

The primary source for many CWs searches is rapidly rotating neutron stars. Neutron stars originate when a massive star collapses and are the remnant of this collapse, they are objects with incredibly high density and are highly magnetised. They have masses around  $1.4 - 2 M_{\odot}$  contained in a star with radius of  $\sim 10$  km with magnetic fields ranging from  $10^8 - 10^{15}$  G [25]. Despite many observations in the electromagnetic spectrum and a large amount of research, these objects are not well understood. A key part of neutron stars which is not understood is the EOS. A review of the current understanding can be found in [106]. The EOS relates quantities such as the pressure, temperature and volume of a neutron star and dictates how the neutron star matter behaves. Observations of GWs from neutron stars can place limits on the EOS of this type of matter. These observations have already been made in the form of BNS mergers [7]. However, independent observations of rapidly rotating neutron stars can add to this understanding.

For a neutron star to emit a gravitational wave it needs to have some asymmetry in its mass distribution around its rotation axis, this follows from Eq. 1.12. There are a number of different mechanisms which could cause this and emit GWs, some of these are reviewed in [26, 27, 28, 29]. Here I will summarise two main theories: Neutron star mountains and neutron star oscillations.

## Mountains

One of more likely mechanisms for detectable GW emission from neutron stars is from ‘mountains’ on the surface of the star. These are permanent deformations on the surface which are non axisymmetric, i.e. the deformation is not symmetric around the rotation axis. Fig. 1.3 shows an exaggerated example of a deformation.

This deformation or asymmetry can be quantified by the ellipticity  $\epsilon$  of the neutron star. This is defined using the principal moments of inertial,

$$\epsilon = \frac{I_{xx} - I_{yy}}{I_{zz}}, \quad (1.13)$$

where  $I_{zz}, I_{xx}, I_{yy}$  are the principal moment of inertia. This is when the star is rotating around the  $z$  axis so  $I_{zz}$  is along the rotation axis.

There are a number of theories which describe the origin of this axisymmetry. If the pulsar is in a binary system and accreting material from its companion star, the material can be funnelled towards the magnetic poles by the magnetic field, thereby causing a hot spot [28]. This ‘hot spot’ could cause a deformation on the surface of the star which is not

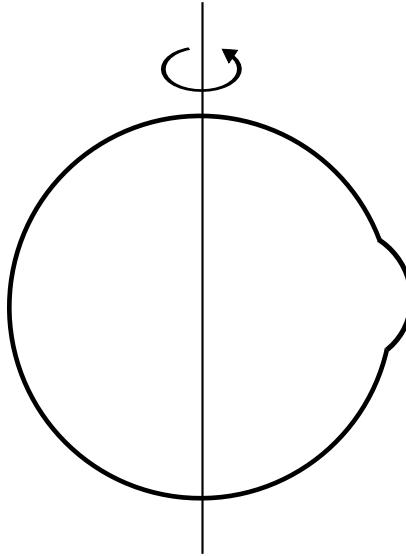


Figure 1.3: The neutron star has some ellipticity which mean that the star deforms such that it is not symmetric around its rotation axis. This is thought of as a ‘mountain’ on the surface of the neutron star. The diagram above shows an extremely exaggerated view of what this could look like.

axi-symmetric. The magnetic stresses from strong magnetic fields within the star, could potentially also cause non axi-symmetric deformations to the star. Finally the spin down of the pulsar itself could cause stresses in the crust of the star until the point of breaking, its then after this break which could leave a distortion in the crust [30]. More details on the signal waveform of this type of **GW** and methods to search for it will be explained in Sec. 2.

### Neutron star oscillations

There are a number of oscillation modes within a star such as f-modes, p-modes and r-modes [30]. Each of these waves are oscillations in the star similar to oscillations in the earth which are used for seismology. The difference between the different modes are the restoring force bringing the perturbed state back to equilibrium. For example, f-modes use gravity as the restoring force where the oscillations happen in the crust of the star. The more promising of these for gravitational wave emission and detection is the r-mode [29]. These are oscillations in the neutron superfluid part of the star, where the restoring force is the Coriolis effect from the rotation of the star. Fig. 1.4 shows an highly exaggerated view of a neutron star with an oscillation mode travelling in each direction. If these modes are excited in a non-rotating star, then they will emit **GW** where the **GW** carries away angular momentum [31]. For the mode travelling in a clockwise direction, this angular momentum is positive and the mode travelling anti-clockwise the angular momentum is

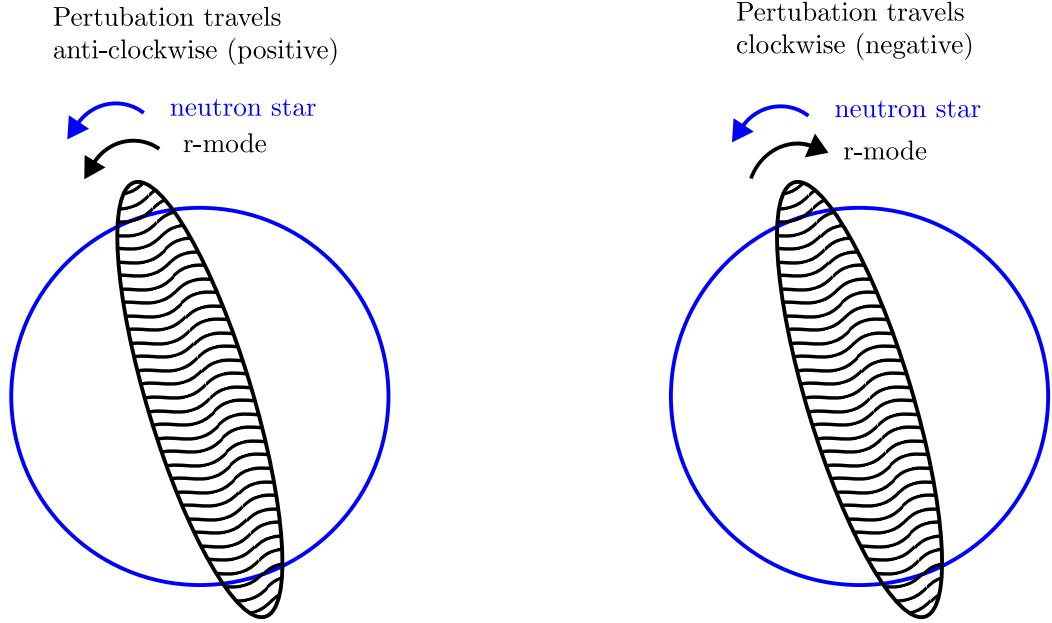


Figure 1.4: The r-modes can travel in either direction in the star. in the case when the r-mode is moving clockwise and the neutron star is moving anticlockwise, if the neutron star is rotating fast enough it can cause unstable emission of **GW**. This image was reconstructed from [31].

negative. Therefore, the **GW** is taking away either positive or negative angular momentum depending on the direction of rotation. The emission of **GW** damps the modes and the magnitude of the perturbation decreases making them extremely difficult to detect. Now if the neutron star is rotating, this can lead to an effect called the **Chandrasekhar, Friedman and Schutz (CFS)** instability [32, 33]. As the rotation speed of the neutron star increases, there are two different effects on the modes travelling in opposite directions. For the mode travelling anti-clockwise with the stars rotation, the mode will appear to be travelling faster, therefore, will emit more **GW** taking away more angular momentum. This means that this mode will be damped more rapidly. The interesting affect is for the mode travelling clockwise, opposite to the neutron stars rotation. At a certain rotation rate, the mode will be ‘frozen’ from the observers perspective and no **GW** will be emitted. As the rotation rate increases further, the mode will appear to travel anti-clockwise to an observer, i.e. the mode is dragged in the opposite direction by the stars rotation. Here it is key to remember that this mode had negative angular momentum as in the neutron stars frame it is still travelling clockwise. As the mode rotates its emits positive angular momentum, which is then subtracted from the modes negative angular momentum. The magnitude of the angular momentum then increases such that more **GW** are released. This effect causes the amplitude of the oscillation to grow and therefore become unstable.

Therefore, a neutron star is unstable to **GW** emission if it is rotating sufficiently fast [29]. For a more detailed view on how r-modes generate **GW** see [34, 31]

## 1.3 Detectors

The theory mentioned above and the indirect detection of gravitational waves from the Hulse-Taylor binary pulsar system left little doubt as to whether **GW** existed. The real challenge was to design an instrument which could directly detect gravitational waves. There were a number of different proposed methods for the design of the instrument which includes: resonant bar detectors, both ground based and space based interferometers, pulsar timing arrays and cosmic microwave background (CMB) detectors. Resonant bar detectors were initially designed and built by Joseph Weber [35]. These are large cylinders of metal which should resonate as a gravitational wave passes by. There are a few different designs of this type of detector, including an omni-directional design [36]. The majority of these detectors are no longer operational. Pulsar timing arrays aim to use the accurate arrival time of pulses from millisecond pulsars to measure **GW** [37]. As a **GW** passes between the pulsar and the observer, the arrival time of the pulses should change. Whilst a detection has not been made using pulsar timing arrays, these methods are still in use. Cosmic microwave background detectors aimed to look for evidence of gravitational waves in the polarisation's of the CMB [38]. These use a range of detectors to look at the CMB however, are yet to confirm a detection of a **GW** signal. The most commonly known design of a **GW** detector is the ground based interferometer, these made the first detection of **GW** in 2015 [4]. These are the focus of this section as the analysis that will follow uses data from the **LIGO** detectors in the USA [39, 10] and Virgo detector in Italy [11, 40].

### 1.3.1 Laser Interferometers

Laser interferometers use the interference of light to measure a length with high precision. The majority of this section will focus on ground based interferometers such as **LIGO** and Virgo [10, 11]. A simple design of an interferometer is shown in Fig. 1.5a. A laser beam is fired at a beam splitter which splits the light equally down two perpendicular arms. Each of these beams is reflected from a mirror at the end of either arm. The light then returns to the beam splitter where the two beams are combined and sent to a photo-detector. At the output, there is an interference pattern between the two beams. If the length of one of the arms is changed then the interference pattern will change as the phase of one beam changes with respect to the other. The phase difference of the light can be related to the wavelength of the light,  $\lambda_l$  and the length of the detectors arms  $L$  by,

$$\Delta\phi \sim \frac{\Delta L}{\lambda_l}, \quad (1.14)$$

where  $\Delta\phi$  is the phase change and  $\Delta L$  is the difference in the arm lengths [41]. An interferometer can then measure small changes in the mirrors position. This can be used in gravitational wave detection as the mirrors at the end of each arm of the interferometer can be treated as ‘free’ test masses. Fig. 1.1 shows the effect of a **GW** in free test masses. In the interferometer, this affect essentially changes the relative lengths of the two arms, i.e.  $\Delta L$  in Eq. 1.14. The change of the interference pattern with time is then related to the **GW**. In practice the interference is held at a single point by adjusting the position of the two mirrors. The **GW** is then related to the readout of the mirrors control system.

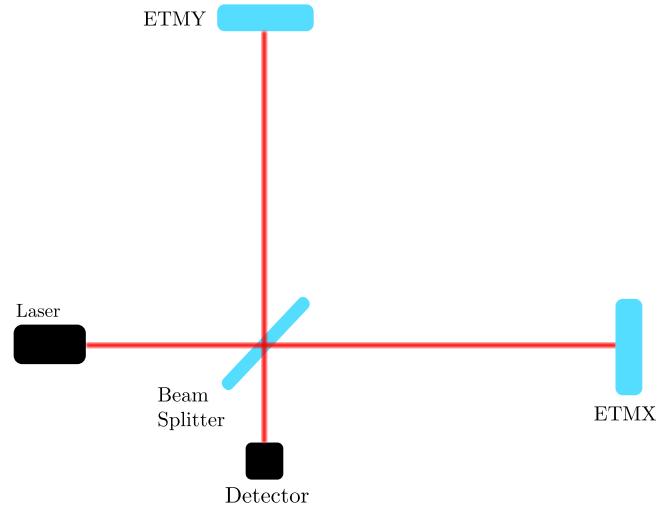
If the mirrors at the end of the arms (ETMX and ETMY) are placed further from the beam splitter, then the phase difference between the arms at the beam splitter for the same length change of one arm would be greater. This means that increasing the length of the detectors arms increases the sensitivity. A method to achieve a similar affect without physically increasing the arm length is to use a Fabry-Perot cavity [10], this is shown in Fig. 1.5b. This is where a semi-transparent mirror is placed between the beam splitter and end mirror in each arm (ITMX and ITMY). Light enters this cavity and reflects back and forth between the two mirrors (ITMX and ETMX) a number of times before returning to the beam splitter. This increases the time the light spends in one arm and is equivalent to increasing the arm length. Actual ground based **GW** detectors such as **LIGO** [39] and Virgo [11] are much more complicated than described above. They use many techniques to increase the sensitivity some of which are outlined in [10, 39]. Many of these techniques are designed to reduce non-astrophysical affects on the detector, some of these effects and solutions are listed in Sec. 1.3.1.

### Detector response

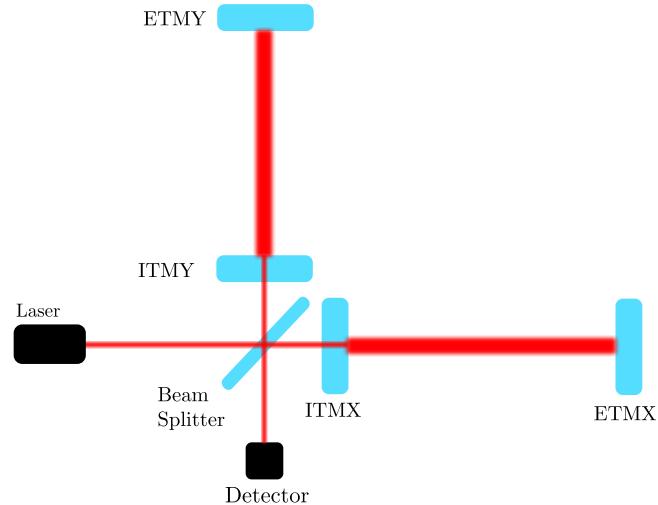
An important factor to know when using detector data to search for astrophysical signals is the detectors response. This measures how sensitive the detector is to different directions. An example of the antenna response for **LIGO** is in Fig. 1.6 where the detectors arms lie on the x and y axis of the image. This is clear when thinking about how a gravitational wave affects the test masses as in Fig. 1.1. As the **GW** is transverse to its propagation, when the detector is face on to the source, there will be a maximum change in the arm lengths and therefore a maximum sensitivity. In the same way the sensitivity will be at a minimum when edge to the source.

### Noise sources

To increase the sensitivity of the **LIGO** detectors, any effect on the output of the interferometer which is not astrophysical ideally needs to be reduced. This involves understanding what causes certain noise features in the detector, and how the affect of these can be limited. Within the detector, there are many sources of noise. Some of these noise sources



(a) Simple interferometer.



(b) Fabry-Perot interferometer.

Figure 1.5: **LIGO** uses interferometry to make a detection. This has many additional features, one of which is shown above. ETMY and ETMX refer to the end test masses, which are just mirrors at the end of the interferometer arms. The beam splitter splits the Laser beam equally to each arm, this them recombines the beams back to the detector. ITMY and ITMX refer to the internal test masses, these create a Fabry-Perot cavity in the interferometers arms which can build up laser power.

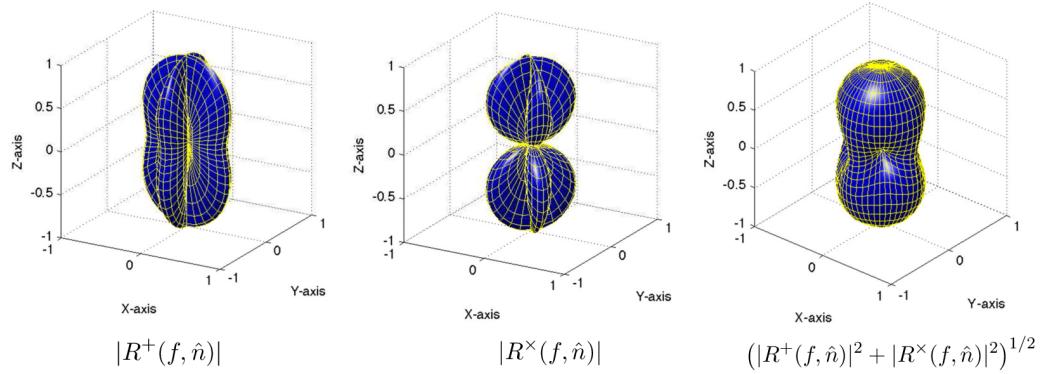


Figure 1.6: The antenna response is shown as in [24] for the plus, cross and average polarisation's. The detectors arms lie on the x and y axis in the above plots.

and how they limit the detectors strain sensitivity are all shown in Fig. 1.7 from [10]. Here I will summarise some of the limiting sources and also sources which become useful for understanding later sections.

**Seismic noise** This originates from the seismic activity of the earth and effects lower frequencies. This can be oscillations such as earthquakes or ocean waves. Seismic waves cause the mirrors to oscillate and induce a change in the length of the arm. This is reduced by having multi stage suspensions in the detectors which essentially absorb the seismic oscillations [42].

**Coating noise** This is in general due to two main factors, the thermal noise of the coating and Brownian noise. The Brownian noise is from the mechanical dissipation in the coating and the thermal noise is due to thermal dissipation. The Brownian noise is the dominant factor as shown in Fig. 1.7. These effects are limited by using different coatings on the mirrors [43].

**Quantum noise** Quantum noise is a fundamental limit due to the statistical uncertainty of counting photons. This limits the sensitivity at many frequencies. There are methods to reduce this include squeezing of the light [44].

**Electronics noise** Whilst this is not shown in Fig. 1.7, this becomes important to searches described later. These have a different effect on the detector which is more narrow-

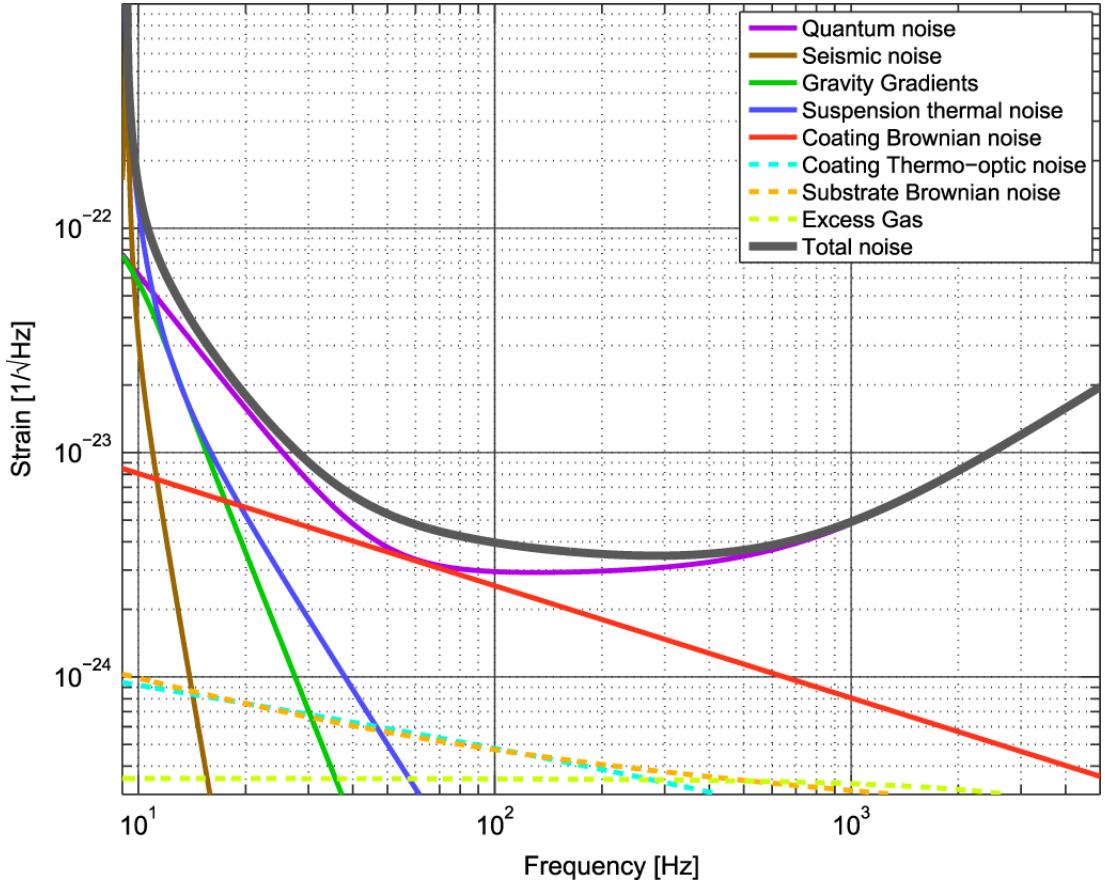


Figure 1.7: The different noise sources affect the sensitivity of the **LIGO** detectors at different frequencies. This shows the various sources how they affect the noise curve [10].

band frequency lines. This is generated by the digital and analogue electronics that are used to measure the signal.

There are also many other sources of noise in the detector which I have not listed. However, these are often not the limiting cases of noise or are not relevant to this thesis. In Sec. 5 I will go into more detail about specific noise sources in the detector known as instrumental lines and how they can be monitored and potentially removed.

# Chapter 2

## Searching for continuous gravitational waves

Continuous gravitational waves have particular challenges when it comes to their detection. The nature of **CW** are that they are long duration, this means that they would be observed for the entirety of a detectors observing runs. The signals also have an intrinsically small amplitude which is below the noise floor of current ground based detectors such as **LIGO**. This means the for a detection, the entire observing runs data will be needed to accumulate enough **signal-to-noise-ratio (SNR)** for a signal to be observed. Given that **LIGO** samples as  $\sim 16$  kHz (generally downsampled to  $\sim 4$  kHz) this leaves a huge amount of data which needs to be searched through. As will be described in Sec. 2.3.1 and 2.3.2, this requires a large amount of computational resources to perform these searches. For some types of search, the parameter space can also be very large, this only adds to the computational time and in some cases makes it infeasible.

Whilst I have described the potential sources of the signal and its approximate signal type in Sec. 1.2.3, to perform a search the wave-form of a signal and how it is observed is needed. In this section I will go into more detail on the ‘mountain’ model in Sec. 1.3 and its wave-form description. This model is then used in various search methods for **CW** signals. In Sec. 2.3 I will overview a subset of current searches for **CW** signals. Sec. 2.4 explains the motivation for the majority of the work in this thesis.

### 2.1 Continuous signal model

The model of a **GW** signal from a pulsar is relatively simple, it is a quasi-sinusoidal signal. This means that the signal is a sinusoid with a slowly varying frequency. One reason for the slow variance in the frequency is due to the energy loss to **GW** as the pulsar spins down. Here the signal is modelled to originate from an isolated triaxial neutron star rotating around a principal axis. The parameters of each pulsar can be split into two sections: the

Doppler components ( $\alpha, \delta, f$ ) and its amplitude components ( $\psi, \phi_0, \iota, h_0, \theta$ ). This ignores any orbital parameters which would be present if the star was in a binary systems and higher order frequency derivatives. They are defined as follows: the sky positions  $\alpha$  and  $\delta$  refer to the right ascension and declination.  $f$  refers to the source frequency and its derivatives.  $\psi$  and  $\phi_0$  and  $h_0$  are the **GW** polarisation, initial phase and amplitude respectively.  $\iota$  is the inclination angle which is how much the source is tilted relative to the observer.  $\theta$  is the ‘wobble angle’ or the angle between the rotation axis and the symmetry axis of the neutron star.

The definition of the **GW** from a neutron star here follows that in [27, 45, 46]. The amplitude of the **GW** can be defined as,

$$h(t) = F_+(t)h_+(t) + F_\times(t)h_\times(t), \quad (2.1)$$

where  $h_+, h_\times$  are the plus and cross polarisations functions as in Eq.1.10 and  $F_+, F_\times$  are the antenna pattern functions to the two polarisations. These are defined by,

$$\begin{aligned} h_+(t) &= h_0 \frac{1 + \cos^2(\iota)}{2} \cos(\Phi(t)) \\ h_\times(t) &= h_0 \cos(\iota) \sin(\Phi(t)) \end{aligned} \quad (2.2)$$

The plus and cross polarised components then depend on the **GW** amplitude  $h_0$ , the inclination angle of the source  $\iota$  and the phase evolution of the **GW**. Here I have chosen to assume a small wobble angle  $\theta$ , however, this is included in [45]. The phase of the wave  $\Phi(t_{\text{SSB}})$  at the **solar system barycenter (SSB)** can be defined as,

$$\Phi(t_{\text{SSB}}) = \phi_0 + 2\pi \left[ f(t_{\text{SSB}} - t_0) + \frac{1}{2} \dot{f}(t_{\text{SSB}} - t_0)^2 + \dots \right]. \quad (2.3)$$

This consists of an initial phase  $\phi_0$  which is the phase at time  $t_0$ , the frequency of the signal  $f$  and its derivative  $\dot{f}$  at time  $t_0$ . Here we show the phase to second order, however, this can be easily extended if necessary. The time at the **SSB**  $t_{\text{SSB}}$  can be transformed to the time  $t$  at the detector by,

$$t_{\text{SSB}} = t - \frac{\mathbf{r}_d \cdot \mathbf{k}}{c} + \delta_t. \quad (2.4)$$

Here  $\mathbf{r}_d$  is the position of the detector with reference to the **SSB**,  $\mathbf{k}$  is a unit vector in the direction of the source. This essentially takes into account the Doppler shift of the signal due to the movement of the detector, i.e. as the earth rotates and orbits the sun.  $c$  is the speed of light and  $\delta t$  is extra corrections from the Einstein, Binary and Shapiro delay [].

The amplitudes  $h_0$  in Eq. 2.2 are defined by,

$$h_0 = \frac{16\pi^2 G}{c^4} \frac{\epsilon I f^2}{r}, \quad (2.5)$$

where  $G$  is the gravitational constant,  $c$  is the speed of light,  $\epsilon$  is the ellipticity of the star,  $f$  is the sum of the frequency of rotation of the star and the frequency of precession,  $r$  is the distance to the star and  $I_{zz}$  is the moment of inertia with respect to the rotation axis  $z$ . The ellipticity of the star  $\epsilon$  is a measure of the distortion of the star around its rotation axis and is defined by,

$$\epsilon = \frac{I_{xx} - I_{yy}}{I_{zz}}, \quad (2.6)$$

where  $I_{xx}$ ,  $I_{yy}$  and  $I_{zz}$  are the moments of inertia for each axis.

In Eq. 2.1,  $F_+(t)$  and  $F_\times(t)$  are the antenna pattern functions of the detector. These describe how sensitive a detector is to a particular location on the sky at any given time. The amplitude of the signal will vary dependent on the orientation and location of the detector relative to the source. This is described in Sec. 1.3 and the response to sky location is shown in Fig. 1.6. These components are defined in [45] as,

$$\begin{aligned} F_+(t) &= \sin \zeta [a(t) \cos(2\psi) + b(t) \sin(2\psi)], \\ F_\times(t) &= \sin \zeta [b(t) \cos(2\psi) - a(t) \sin(2\psi)], \end{aligned} \quad (2.7)$$

where  $\zeta$  is the angle between the arms of the detectors,  $\psi$  is the polarisation angle of the **GW** and  $a(t)$  and  $b(t)$  are defined in [45] and relate the sky location to the orientation of the detector at a given time. A full derivation of this can be found in [45] where each of these terms are expanded.

Eqs. 2.1 - 2.7 then describe the amplitude and phase evolution of a signal at a given detector location and orientation.

## 2.2 Bayes Theorem

A key part in understanding the different methods to search for **GW** or any data analysis, is understanding probability and statistics. This gives understanding of the random processes underlying all measured quantities. Whilst there are generally two approaches to statistics: Frequentist and Bayesian, here I will focus on the Bayesian approach.

### 2.2.1 Basic probability

Initially I will define some basic concepts of probability. We can define the probability of some event  $A$  as  $p(A)$  where probabilities follow  $0 \leq p(A) \leq 1$  and some other event  $B$  which has a probability  $p(B)$  and follows  $0 \leq p(B) \leq 1$ .

**Union** A union is the probability of either event  $A$  happening or event  $B$  happening. This is written as,  $p(A \cup B)$ .

**Intersection** An intersection is then the probability that both event  $A$  and an event  $B$  happens. This is written as  $p(A \cap B)$ .

**Independent and dependent Events** If the events  $A$  and  $B$  are independent, i.e. the event  $A$  does not affect the outcome of event  $B$ , then,

$$p(A \cap B) = p(A)p(B). \quad (2.8)$$

However, if the event  $A$  is dependent on event  $B$ , i.e. the event  $A$  affects event  $B$  or vice versa, then the joint probability of both events is,

$$p(A \cap B) = p(A)p(B | A) = p(B)p(A | B). \quad (2.9)$$

Here  $p(B | A)$  means the probability of event  $B$  happening given that event  $A$  has happened.

**Conditional probability** Conditional probability arises from situations where the outcome of one event will affect the outcome of future events. The definition of this arises from the the dependent events defined above in Eq. 2.9,

$$p(A | B) = \frac{p(A \cap B)}{p(B)}. \quad (2.10)$$

**Bayes Theorem** Bayes theorem can then be defined using conditional probabilities. i.e we can use

$$p(A | B) = \frac{p(A \cap B)}{p(B)} \quad \text{and} \quad p(B | A) = \frac{p(A \cap B)}{p(A)} \quad (2.11)$$

such that then,

$$p(B)p(A | B) = p(A)p(B | A) \quad (2.12)$$

and this is rearranged to Bayes theorem,

$$p(A | B) = \frac{p(A)p(B | A)}{p(B)} \quad (2.13)$$

## 2.2.2 Bayesian Inference

We can take Bayes theorem from Sec. 2.2.1 and apply it to a problem which involves inferring some parameters from some model. Here we can relabel the events  $A$  and  $B$  with

the data  $\mathbf{d}$  and the parameters  $\boldsymbol{\theta}$  of some model  $I$ . Eq. 2.2.2 then becomes,

$$p(\boldsymbol{\theta} \mid \mathbf{d}, I) = \frac{p(\boldsymbol{\theta}, I)p(\mathbf{d} \mid \boldsymbol{\theta}, I)}{p(\mathbf{d} \mid I)} \quad (2.14)$$

where each of the components are assigned names:  $p(\boldsymbol{\theta} \mid \mathbf{d})$  is the posterior distribution,  $p(\boldsymbol{\theta})$  is the prior distribution,  $p(\mathbf{d} \mid \boldsymbol{\theta})$  is the likelihood and  $p(\mathbf{d})$  is the evidence.

**Posterior** The posterior distribution describes the probability of a parameter  $\theta$  in some model  $I$  given some data  $d$ . For many problems this is the distribution which is most useful as it informs you the most likely set of parameters of your model given some observation.

**Prior** The Prior distribution is a key part of Bayesian statistics. This distribution describes any information which you have prior to the observation. This is a distribution defined by the user, where you define a distribution of the parameters based on what you expect to be true.

**Likelihood** The likelihood is where the observation is included in the calculation. This tells you how probable it is to get the observed data  $d$  given the model  $I$  with the set of parameters  $\theta$ .

**Evidence** The evidence is the probability of the data itself given the choice of model. This is found by integrating the likelihood over all possible values of  $\boldsymbol{\theta}$  weighting them by our prior belief of that value of  $\boldsymbol{\theta}$ . This is known as a marginal distribution and is defined by,

$$p(\mathbf{d} \mid I) = \int p(\boldsymbol{\theta}, I)p(\mathbf{d} \mid \boldsymbol{\theta}, I)d\boldsymbol{\theta}. \quad (2.15)$$

Bayes theorem then gives a description of the probability distribution of some parameters in a model given some observation. Often when using Bayesian statistics the aim is to find posterior distribution of parameters. There are very few cases where this can be calculated analytically, therefore, numerical methods are almost always used to find the posterior. This can be difficult to calculate numerically especially in problems where the parameters space has many dimensions. The most difficult part to calculate is the evidence in Eq. 2.15, this involves calculating an integral over all possible parameters. There is however, a way around having to calculate this. For any given mode  $I$ , the evidence  $p(\mathbf{d} \mid I)$  is the same for any set of parameters  $\boldsymbol{\theta}$  in Eq. 2.14. The evidence is then just a normalisation factor for the posterior distribution. When different models are not being compared, and we assume the model  $I$  to be true, we no longer need to calculate the evidence. The posterior distribution can then be found by sampling,

$$p(\boldsymbol{\theta} \mid \mathbf{d}, I) \propto p(\boldsymbol{\theta}, I)p(\mathbf{d} \mid \boldsymbol{\theta}, I). \quad (2.16)$$

To find this posterior you could then calculate (sample) the value for every point in parameter space. This however, is very computationally expensive and often the posterior distribution is located in a small area in parameter space. Therefore, the majority of the time is sampling a area of parameter space where the posterior is close to zero, and this is not particularly useful. A method titled **Markov-Chain Monte Carlo (MCMC)** was proposed [47] to deal with this issue, more information on this can be found in [48, 49]. This builds up the posterior distribution by randomly jumping around in the parameter space. It starts by calculating the posterior value for a particular point in parameter space. Then it will randomly jump to another parameter space point. The posterior can then be calculated again, if the posterior value is higher then the jump is ‘accepted’. This just means that the parameter values of this point are stored. If the posterior value is lower than the previous step then the jump is accepted with some probability. This means that there is a random chance that a value lower than the current is accepted. As the accepted positions aim for areas where the posterior is higher, **MCMC** does not waste time calculating areas in parameter space of low posterior values. The samples which were accepted then build the posterior distribution.

In certain situations it can be useful to calculate the evidence in Eq.2.15. For example, if there are two different models which could represent the data, the evidence can be used to determine which of the two models is more likely. This is known as a Bayes factor where two models  $I_1$  and  $I_2$  are compares and is defined as,

$$B = \frac{p(\mathbf{d} | I_1)}{p(\mathbf{d} | I_2)} \quad (2.17)$$

This then requires the calculation of the evidence. To estimate the evidence efficiently a method known as Nested sampling can be used, this is explained in detail in [50, 51]. By calculating the Bayes factor, which is similar to a likelihood ratio, one can find the posterior odds of a particular model by using,

$$\frac{p(I_1 | \mathbf{d})}{p(I_2 | \mathbf{d})} = \frac{p(I_1)}{p(I_2)} \frac{p(\mathbf{d} | I_1)}{p(\mathbf{d} | I_2)}. \quad (2.18)$$

The can be written as *posterior odds* = *prior odds*  $\times$  *bayes factor*. This is then a comparison of how likely different models are given some observation.

The methods described above then provide a way to estimate parameters of a model given some data. Also this provides a way to compare different models given some observation. In following sections the methods described above are used to estimate various parameters.

## 2.3 Continuous wave searches

There are many different methods to search for continuous gravitational waves. They can be split into three general categories: Targeted searches, Directed searches and All-sky searches. The main difference between these different categories is the amount which is known about the source prior to the search. For targeted searches the sky position ( $\alpha, \delta$ ) and rotation frequency are known from electromagnetic observations, i.e. X-ray, radio or  $\gamma$ -ray. Directed searches have information on the sky position ( $\alpha, \delta$ ) but not the rotation frequency. For all-sky searches, there is no prior knowledge of the pulsar, therefore, is a search for unknown pulsars. In general the searches in each of these categories use two distinct techniques: Fully coherent searches and Semi-coherent searches

### 2.3.1 Fully coherent

A Fully coherent search generally uses a pre generated waveform which follows the model described in Sec. 2.1. This contains all the phase information of the signal. The set of parameters which generated the waveform which ‘matches’ the best can then be considered as the optimum set of parameters given the data. This is known as a matched filter [] and is used in **CW** searches in [46].

The matched filter maximises the signal to noise ratio for a given filter, in this case the filter is our **CW** model. The matched filter used for **CW** models is defined in [52] and it titled the  $\mathcal{F}$ -statistic. This essentially maximises a likelihood with respect to the parameters. If one assumes that the noise  $n$  is Gaussian and zero mean, the data  $x$  can be written as,

$$x(t) = n(t) + h(t). \quad (2.19)$$

The likelihood can then be written as,

$$\log \Lambda = (x | h) - \frac{1}{2} (h | h) \quad (2.20)$$

where the product  $(x | y)$  is defined as,

$$(x | y) = 4\mathcal{R} \int_{-\infty}^{\infty} \frac{\tilde{x}^X(f)\tilde{y}^{X*}(f)}{S^X(f)} df. \quad (2.21)$$

This is fulle expanded into the  $\mathcal{F}$ -statistic in [45], however, it is essentially this likelihood function which is maximised.

Targeted searches look for a specific pulsar which has been observed in the electro-magnetic spectrum. These observations give information such as the sky position and the frequency evolution of the source. Using knowledge of the earths position around the sun, which is well known, one can use the accurate sky position and frequency of a known

source to find its phase evolution in Eq. 2.3. This mean that for this type of search one can maximise the likelihood with respect to the parameters  $h_0$ ,  $\phi_0$ ,  $\iota$  and  $\psi$ . Another method which uses templates is described in [46], this uses a Bayesian approach.

This type of search can take long periods of time. This is due both to the size of the parameter space and the amount of data which needs to be searched. CW searches need long observation times in order to accumulate the required SNR for detection. Therefore, most searches use data from an entire LIGO observing run which can last for  $\mathcal{O}(1)$  years. Given that the sampling rate for the GW channel is 16 kHz usually downsampled to  $\sim 4$  kHz, the quantity of data is large.

Whilst the fully coherent matched filter searches have methods to reduce the computational time for known sources, in all-sky and directed searches, this type of search is no feasible. This is because all-sky and directed searches have a wider parameter space, therefore, enough templates need to be made to sufficiently cover the large parameter space. This task quickly becomes impossible for coherent matched filtering for an entire observing run due to the amount of time needed. This problem led to the development of semi-coherent searches which will be introduced in the next section.

### 2.3.2 Semi coherent

Semi-coherent searches offered a solution to searching over large parameters spaces and large amounts of data. As is directed and all-sky searches the phase evolution of the source is not known, one cannot use a coherent search for the entire observing run. It may however, be possible to approximately describe the phase for a shorter length of time known as the coherence time,  $T_{coh}$ . The general idea of a semi-coherent search is to break the data-set into smaller section which each can be analysed coherently. The coherent analysis can use the matched filter as described in Sec. 2.3.1 or another method such as a Fourier transform. The results from each of these individual sections can be combined incoherently using various methods which will be summarised later. This method can greatly reduce the time taken for the analysis depending on the coherence length, however, will always come with some loss in sensitivity.

There are many different types of semi-coherent search which use various methods to incoherently combine the coherently analysed results. I will summarise some of these searches below, some of these searches were summarised and compared in [53]. Many of these searches use a set of 1800s long Fourier transforms as the input data, known as short Fourier transforms (SFTs). This is a default for many all-sky CW searches, where it assumes that the signal remains within one frequency bin during that 1800s.

**Stack-slide** Stack uses a set of Fourier transforms of the data known as SFTs, specifically it uses the power spectrum of these. Each of the separate Fourier transforms

(segments) is shifted up or down relative to the others to account for the Doppler modulation of the source. The power from each can then be stacked. More explanation of this can be found in [54, 55]

**Hough** The Hough transform is based on the stack-slide algorithm. The main difference is that the detection statistic for each segment is assigned a weight of 0 or 1 depending if it crossed a detection threshold. The Hough transform can the create a ‘Hough map’ which gives a view of the data in parameter space. This approach is explained in greater detail in [56, 57]. This method has been applied in two main ways known as Sky Hough [56] and Frequency Hough [57, 58].

**Einstein@Home** Einstein at home uses the  $\mathcal{F}$ -statistic mentioned above in various stages. It has a hierarchical structure where it starts with a coarse parameter space with shorter coherence times. This search then provides a list of candidates from this run in coarse parameter space. The parameter space is then more finely sampled around the parameters of the candidates and this process is repeated. The search can also increases the coherence length when searching around given candidates to improve the sensitivity of the search. This algorithm has many additions which are explained in more detail in [59, 60, 53]. This provides the most sensitive all-sky CW search, however, uses a large amount of computing power. This is achieved by using a distributed computing projects, more details can be found at [61].

**Time domain  $\mathcal{F}$ -statistic** The time domain  $\mathcal{F}$ -statistic splits the data into narrowband segments of length  $\sim 2$  days [53]. Then a coherent search using the  $\mathcal{F}$ -statistic is applied to each of these segments. Values of this statistic above a threshold are stored. Coincidences are then found in each segment, where candidates are selected best on a given threshold. This is explained in greater detail in [62, 53].

**Powerflux** Powerflux uses a standard set of 1800s SFTs. For each point in parameter space, the power in this set of SFTs along the frequency track is recorded. This power is then weighted depending on the antenna pattern and noise of the detector. In longer stretches of  $\sim 1$  month, the weighted power is summed. Any point in parameter space which produces high power in each of these stretches is identified as a potential signal. This search can then be repeated around each candidate with a finer resolution in parameter space. This is explained in more detail and tested in [63, 53, 64]

**Viterbi** The Viterbi algorithm [65] has been used in [107, 66, 67, 68, 69] to search for a CWs with randomly wandering spin frequency. This algorithm was applied to specific sources, where the  $\mathcal{F}$ -statistic is used on short duration segments which are then incoherently combined using the Viterbi algorithm.

Table 2.1: From [53], shows the computational cost for the first 4 months of advanced LIGO for each search. One million standard units (MSU), where one standard unit is one core-hour on a standard core. ‘Expected computational costs of searches using the first four months of advanced LIGO data with each search pipeline. These estimates are for a different data observing time from that of the MDC, and do not cover the same parameter space as each other or the MDC. The Einstein@Home searches uses the computing resources of the Einstein@Home project and is designed to run for 6 - 10 months in the Einstein@Home grid.’

Pipeline	Expected runtime of O1 search
Powerflux	6.8 MSU
Time domain $\mathcal{F}$ -statistic	1.6 MSU
Frequency Hough	0.9 MSU
Sky Hough	0.9 MSU
Einstein@Home	100-170 MSU

Each of these searches uses a large computational cost. In [53] a mock data challenge (MDC) was conducted to compare the sensitivity of some of the searches, where an expected runtime for an O1 search was presented. Result in O1 for some of these searches can now be found in [70]. The results of this are shown in Tab.2.1. Even the fastest of these searches takes close to 1 million core-hours to search through four months of data. This is one of the larger current issues in CW searches as this is a slow process and running computing clusters can be costly.

## 2.4 Motivation

The searches described in Sec. 2.3 are computationally expensive, where the fastest takes  $\sim$ 1 million core-hours to search through 4 months of O1 data. Many of these searches use well-modelled signals to compare to the data. This leads to the parameter space having to be finely sampled such that it is sufficiently covered. The motivation for much of the work then follows from these points. The aim was to develop searches which used minimal computational resources and could work outside of the CW model in Sec. 2.1. This thesis then outlines algorithms which can reduce this computational time of searches for CW whilst retaining as much sensitivity to CW signals as possible. There are two main sections which follow, Sec. 3 will outline an essentially un-modelled CW search method which uses the Viterbi algorithm. Sec 4 will then outline a method which used machine learning, specifically convolutional neural networks (CNNs) as both its own CW search and an extension to the search described in Sec. 3. Following chapters then explain applications of these searches.

# Chapter 3

## SOAP: A generalised application of the Viterbi algorithm to searches for continuous gravitational-wave signals.

The SOAP search is a semi-coherent [CW](#) search algorithm that aims to reduce the computational time needed to find a potential signal. The algorithm looks through narrow-banded time-frequency spectrograms of data to find the ‘most probable track’ in frequency through it. This ‘most probable track’ is then the most likely track which a pulsars frequency would follow. The motivation of the search is simple, if we looked at a frequency band in a spectrogram as in Fig. 3.1, we could find every possible randomly wandering track from a starting frequency bin to an end frequency bin. For each of these tracks the sum of the spectrogram power along the track can be found such that for each track there is a single value. Fig. 3.1 shows a histogram of a subset of these values. If there is a signal present in this band then it could be assumed that the signal which gives the highest sum of spectrogram power is the track which is most likely to follow the signals frequency track. In Fig. 3.1 the optimum track in red shows gives a statistic value of  $\sim 1700$  which is far outside the main distribution of summed powers. The red track follows that an injected signal. This demonstrates that the sum of the spectrogram power along a track which follows a signal is outside the distribution of tracks randomly walk through noise. Therefore, it can be assumed that if the frequency track with the highest sum of spectrogram power is found, then the corresponding track is most likely for follow a signal. Now as calculating all possible tracks is computationally expensive. Given that in the example in Fig. 3.1, the spectrogram is frequency 180 bins wide  $M$  and 400 time segments long  $N$  and after each segment the track has  $T$  possible options to jump to (in this case it is 180 options), The total number of possible tracks is  $T^{M \times N}$ . For this spectrogram this value is  $180^{72000}$ , this is an unreasonable number of tracks to possibly calculate. This is where the Viterbi algorithm [65] is useful as it can efficiently find the track which gives the maximum sum

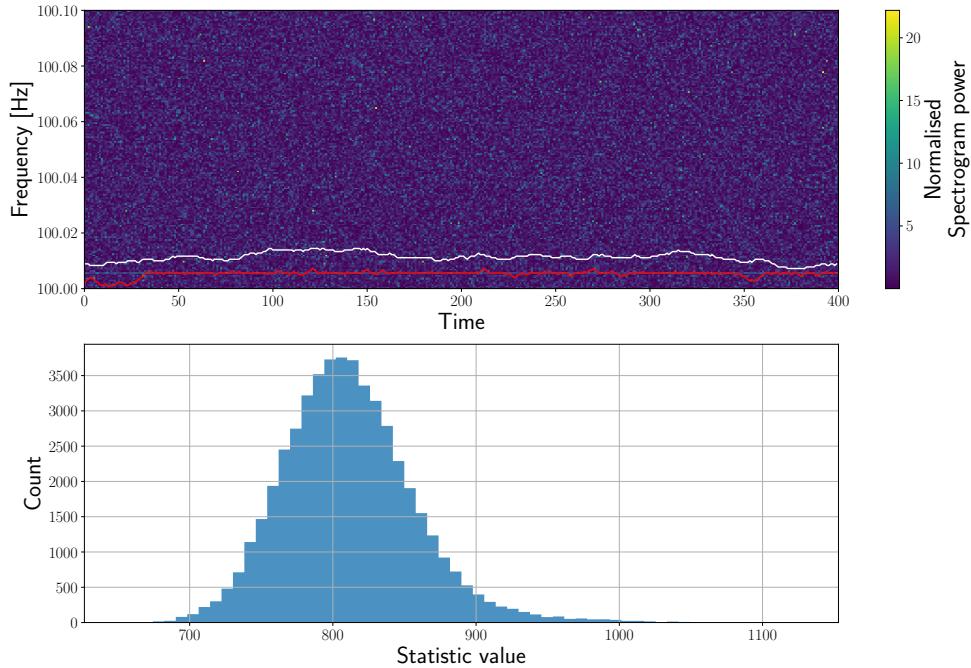


Figure 3.1: This figure shows an example of a time-frequency spectrogram which is typical of LIGO data which is searched through. Here an instrumental line has been injected at 100.006 Hz. The white track shows a random walk track through this spectrogram whereas the red line shows the track which gives the highest sum of detector power. The second panel shows a histogram of a subset of all paths which can be found through the given spectrogram from start to finish. This is a subset as the total number of paths is too large to calculate. The value of the statistic which comes from the optimal path is  $\sim 1700$ . This is much larger than any of the random tracks in our subset and much larger than the mean of all tracks.

of power. For an equivalent search the Viterbi algorithm would have to do  $T * M * N$  calculations for find the optimum track. The method of how this is done is explained in the following sections.

The majority of this chapter that follows has been reviewed and published as in [82]. This was work done by the author under the supervision of Prof. Graham Woan and Dr. Chris Messenger. The exceptions are work in Sec. 3.9, Sec. 3.11, Sec. 3.11.3 and Sec. 3.11.4 which is supplementary material.

### 3.1 Introduction

One of the main targets for current ground based GW detectors, including LIGO [39, 10] and Virgo [40, 11], are sources of continuous gravitational waves. These are long-duration,

quasi-monochromatic sinusoidal signals that are well-modelled by a Taylor series expansion in the signal phase. A likely source of such signals are rapidly spinning non axisymmetric neutron stars. A number of possible emission mechanisms are outlined in [71, 72].

These types of **GWs** are expected to give strain amplitudes that are significantly below the detector’s noise spectral density, and need sensitive search algorithms for detection. The most sensitive method is to use a coherent matched filter which requires knowledge of the waveform beforehand such that it can be coherently correlated with the data. This approach is used in searches for gravitational signals from known pulsars such as [46, 73, 45, 108, 74]. For broad parameter space searches, where the parameters of the signal are unknown, a large number of template waveforms must be used to sufficiently cover the parameter space. This approach rapidly becomes computationally impractical as the search space grows, so semi-coherent search methods have been developed to deliver the maximum overall sensitivity for a given computational cost. Semi-coherent searches break the data up into sections of either time or frequency and perform a coherent analysis on these sections separately. These intermediate results can then be recombined incoherently in a number of different ways to form the final search result outlined in [75, 76] and references therein.

The analysis that we present here is known as SOAP [77] and is based on the Viterbi algorithm [65]. The algorithm models a process that has a discrete number of states at discrete time steps, and computes the set of states which gives the highest probability (suitably defined) given the data. Our implementation of SOAP is intended as a stand-alone search which is naturally non-parametric and has broad applications to both searches for known signal types and signals which have an unknown frequency evolution. The algorithm works in time-frequency plane, where our ‘states’ are represented by the time and frequency coordinates of a potential signal. We can then find the most probable set of frequencies a possible signal could have, i.e., we can find the most probable track in frequency as a function of time. This is not the first application of the Viterbi algorithm to **GW** data. Another variant of the algorithm [109] has recently been used, amongst other applications, as part of a **CW** search to track a pulsar with randomly wandering spin frequency [107, 66, 67, 68, 69]. We develop an alternative version which is aimed to be applied more generally to search for any long duration signals using just **SFTs**.

In the next section we will describe the Viterbi algorithm and the basic SOAP implementation to **GW** time-frequency data. We then describe additional features to the algorithm, including the use of data from multiple detectors. As well as this we describe methods used to ignore instrumental effects in the data, such as incoherently summing data and a ‘line aware’ statistic. In the final section as well as a test of the computational cost of the search, we show results of a search performed on datasets of increasing complexity: Gaussian noise with no gaps (i.e., contiguous in time), Gaussian noise with gaps

simulating real data more accurately, and finally real LIGO data taken during the sixth science run.

## 3.2 Viterbi algorithm

The Viterbi algorithm is an efficient method for determining the most probable set of states (a single ‘track’ of steps on the time-frequency plane) in a Markov model dependent on data, where the model has a discrete number of states at each step. Rather than computing the probability of every possible track and selecting the most probable, the algorithm maximises this probability after every discrete step. As a result, a partial track which cannot ultimately be the most probable is rejected before the next step is calculated, and only a fraction of all possible tracks need to be computed to find the one that is most probable.

In this work we apply the Viterbi algorithm to a GW strain time-series to find the most probable track of a single variable-frequency signal in the noisy data. We divide the time series into  $N$  equal-length and contiguous segments  $\mathbf{x}_j$ , defining the set  $D \equiv \{\mathbf{x}_j\}$ . The ‘states’ in the model correspond to the frequencies a signal could have in each segment. A ‘track’ is a list of such frequencies  $\boldsymbol{\nu} \equiv \{\nu_j\}$ , where  $\nu_j$  is the frequency in the segment  $\mathbf{x}_j$ .

Our objective is to calculate the most probable track given the data, i.e., the track that maximises  $p(\boldsymbol{\nu} | D)$ . Using Bayes theorem, this posterior probability can be written as

$$p(\boldsymbol{\nu} | D) = \frac{p(\boldsymbol{\nu})p(D | \boldsymbol{\nu})}{p(D)}, \quad (3.1)$$

where  $p(\boldsymbol{\nu})$  is the prior probability of the track,  $p(D | \boldsymbol{\nu})$  is the likelihood of the track (i.e., the probability of the data given the track) and  $p(D)$  is the model evidence (or marginalised likelihood).

The Viterbi algorithm treats the track as the result of a Markovian process, such that the current state depends only on the previous state. It is therefore useful to split the track’s prior into a set of transition probabilities such that

$$\begin{aligned} p(\boldsymbol{\nu}) &= p(\nu_{N-1}, \dots, \nu_1, \nu_0) \\ &= p(\nu_{N-1} | \nu_{N-2})p(\nu_{N-2} | \nu_{N-3}) \dots p(\nu_1 | \nu_0)p(\nu_0) \\ &= p(\nu_0) \prod_{j=1}^{N-1} p(\nu_j | \nu_{j-1}), \end{aligned} \quad (3.2)$$

where  $p(\nu_0)$  is the prior probability that the signal in the first time step has a frequency  $\nu_0$  and  $p(\nu_j | \nu_{j-1})$  is the prior ‘transition’ probability for  $\nu_j$  given the frequency at the last step was  $\nu_{j-1}$ .

The noise in each of the segments can be treated as independent, so the likelihood

component in Eq. 3.1 can be factorised as

$$p(D \mid \boldsymbol{\nu}) = \prod_{j=0}^{N-1} p(\mathbf{x}_j \mid \nu_j), \quad (3.3)$$

where  $p(\mathbf{x}_j \mid \nu_j)$  is the likelihood of our signal having a frequency  $\nu_j$  in the  $j$ th segment.

Using Eqs. 3.1, 3.2 and 3.3, the posterior probability is then

$$p(\boldsymbol{\nu} \mid D) = \frac{p(\nu_0)p(\mathbf{x}_0 \mid \nu_0) \prod_{j=1}^{N-1} p(\nu_j \mid \nu_{j-1})p(\mathbf{x}_j \mid \nu_j)}{\sum_S \left\{ p(\nu_0)p(\mathbf{x}_0 \mid \nu_0) \prod_{j=1}^{N-1} p(\nu_j \mid \nu_{j-1})p(\mathbf{x}_j \mid \nu_j) \right\}}, \quad (3.4)$$

where in the denominator we must sum over all possible tracks  $S$ . We require the specific track, or set of frequencies,  $\hat{\boldsymbol{\nu}}$  that maximises the posterior probability. Therefore, as the denominator in Eq. 3.4 is a sum over all possible tracks, the track which maximises the posterior is the same track which maximises the numerator on the right-hand side of Eq. 3.4, i.e.,

$$p(\hat{\boldsymbol{\nu}} \mid D) \propto \max_{\boldsymbol{\nu}} \left[ p(\nu_0)p(\mathbf{x}_0 \mid \nu_0) \prod_{j=1}^{N-1} p(\nu_j \mid \nu_{j-1})p(\mathbf{x}_j \mid \nu_j) \right]. \quad (3.5)$$

This track also maximises the log of the probability and can be written as,

$$\begin{aligned} \log p(\hat{\boldsymbol{\nu}} \mid D) &= \max_{\boldsymbol{\nu}} \left\{ \log p(\nu_0) + \log p(\mathbf{x}_0 \mid \nu_0) \right. \\ &\quad \left. \sum_{j=1}^{N-1} \left[ \log p(\nu_j \mid \nu_{j-1}) + \log p(\mathbf{x}_j \mid \nu_j) \right] \right\} + \text{const.} \end{aligned} \quad (3.6)$$

The Viterbi algorithm finds the most probable track  $\hat{\boldsymbol{\nu}}$  by calculating the quantities in Eq. 3.6 for each frequency at each time step. In the following sections we explain how this is achieved in practice.

### 3.3 The transition matrix

An important concept when using the Viterbi algorithm is the ‘transition matrix’  $T$ , which is defined as the matrix that stores the prior log-probabilities  $\log p(\nu_j \mid \nu_{j-1})$ . These transition probabilities depend only on the size and direction of the transition, and in our case correspond to a jump in frequency when moving from the  $(j - 1)$ th to the  $j$ th state. It is within the transition matrix that we impose some loose model constraints. For example it is usual in the time-frequency plane for frequencies to only have discrete

values (frequency bins) and a track might only be allowed to move by one bin in each time step, restricting it to a **up, centre or down (UCD)** transition or ‘jump’ or equivalently setting the size of the first dimension of the transition matrix  $n_1 = 3$ . We can also impose that the transition probabilities are independent of the current track location in frequency, i.e.  $p(\nu_j \mid \nu_{j-1}) = p(\nu_{j+k} \mid \nu_{j+k-1})$ . This leads to the transition matrix containing only three numbers, corresponding to the three prior log-probabilities that the track was in the corresponding **UCD** frequency bin at the previous time step. These numbers are chosen to reflect the prior probability of a frequency deviation in the track and depend on the class of signals that one wishes to detect. For the majority of examples that follow, a symmetric transition matrix is used, i.e. the probability of a transition up a frequency bin is equal to the probability of a transition down a frequency bin. This allows us to parameterise the one dimensional transition matrix with a single value, this value is the ratio of the probability of a transition to the same frequency bin, to either up or down a frequency bin.

In later sections we will consider more complex situations in which the transition matrix describes the prior probability associated with sequences of even earlier transitions (‘memory’) and the case where there are multiple detectors. In these cases the number of dimensions of the transition matrix can grow substantially to account for the extra complexity of the problem.

### 3.4 Single detector

We will first consider the simple case of a single dataset  $D$ , generated by a single gravitational wave detector, and consider only a one-dimensional transition matrix. We will make use of discrete Fourier transforms so that frequencies, and hence the track frequencies, are also discrete. These frequencies will be indexed by  $k$  and therefore  $\nu_j \rightarrow \nu_{j,k} = k(j)\Delta f$  where  $\Delta f = 1/T$  is the frequency bin width for a segment of duration  $T$ .

The Viterbi algorithm determines the most probable track on the time-frequency plane by calculating the value of Eq. 3.6 for every discrete Fourier frequency, incrementally in time. In other words, at each time segment it finds the most probable earlier track which ends at each particular frequency. On reaching the final segment it can look back to identify the most probable track connecting segment 1 to segment  $N$ .

There are two main components to Eq. 3.6: the transition probabilities  $p(\nu_j \mid \nu_{j-1})$  and the likelihoods  $p(\mathbf{x}_j \mid \nu_j)$ . The transition probabilities are pre-calculated and stored in a transition matrix according to Sec. 3.3 above. To calculate the likelihood we follow the approach of [110] which gives, under the assumption of a single sinusoidal signal in

additive Gaussian noise in data segment  $\mathbf{x}_j$ ,

$$p(\mathbf{x}_j \mid \nu_{j,k}, \sigma_{j,k}, I) \propto \exp [C(\nu_{j,k})]. \quad (3.7)$$

where  $C_{j,k}(\nu_{j,k})$  is the Schuster periodogram normalised to the noise variance at frequency  $\nu_{j,k}$  of segment  $j$ . This is equivalent to the log-likelihood, and is defined as

$$C(\nu_{j,k}) \equiv C_{j,k} = \frac{1}{\sigma_{j,k}^2} \frac{1}{N_s} \left| \sum_{r=0}^{N_s-1} x_{j,r} e^{i\nu_{j,k} t_r} \right|^2, \quad (3.8)$$

where  $N_s$  is the number of data points in each segment and  $t_r$  is the time corresponding to  $x_{j,r}$ , the  $r$ th sample in the  $j$ th data segment.  $\sigma_{j,k}^2$  is the noise variance and is calculated as an estimate of the noise **power spectral density (PSD)** in the  $k$ th sample and the  $j$ th data segment. It is worth noting at this point that it is also possible to write this as a likelihood ratio, and therefore write out detection statistic as a log-odds ratio, however, we will discuss this in more depth in Sec. 3.8. The log-likelihoods of each segment can be calculated at discrete frequencies before running the algorithm by computing the power spectra for each segment from discrete Fourier transforms of the data. In the **GW** field these standard data forms are known as **SFTs**.

The Viterbi algorithm records two quantities for each frequency and time bin: The first,  $V_{j,k}$ , contains the value defined by Eq. 3.6, which is the log-probability of the most probable path ending in position  $j, k$ . The second,  $A_{j,k}$ , is the transition, or ‘jump’, used to achieve the most probable path. The algorithm can be divided into three main sections: initialisation, iteration and identification. These three sections are described in pseudo-code in Alg. 3.1 and a simple demonstration of the algorithm at work is shown in Fig. 3.2.

**Initialisation** The two parts of Eq. 3.6,  $\log p(\nu_0)$  and  $\log p(\mathbf{x}_0 \mid \nu_0)$ , must be computed before the main recursive part of the algorithm can start. Therefore, the initialisation section (lines 5–8) in Alg. 3.1 calculates the first column in the lower panel of Fig. 3.2. A priori, there is no preferred initial frequency, so we take the log-prior  $\log p(\nu_{0,k})$  to be uniform over the complete frequency range. As a result, this does not affect the maximisation for any jump, therefore, can be omitted from the calculation. We then use the pre-calculated log-likelihood values  $C_{0,k}$  to fill the track probabilities  $V_{0,k}$ . There is no previous position to jump from in this case, so the transition probabilities are irrelevant and  $A_{0,k}$  are set to zero.

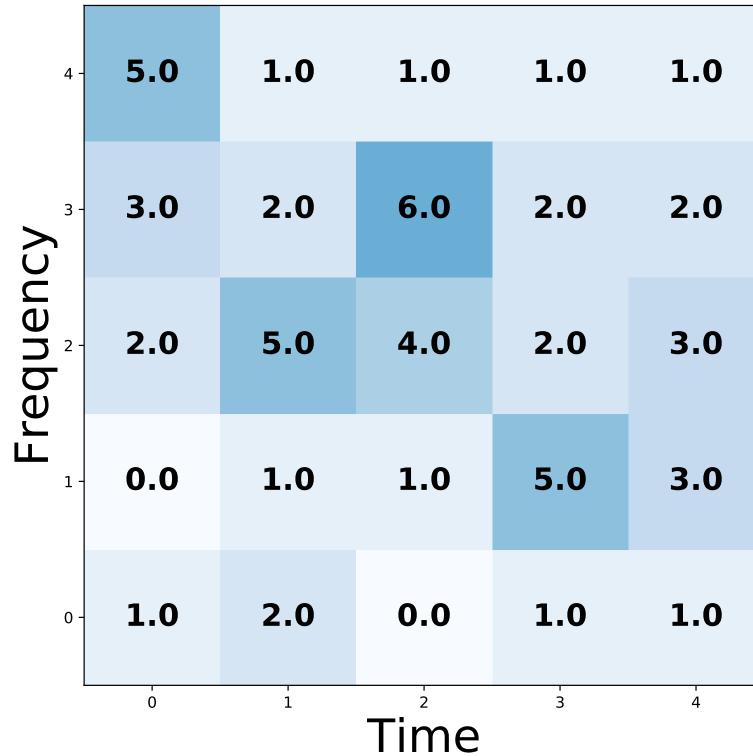
**Iteration** The main part of the calculation is the sum in Eq. 3.6. Lines 11–16 in Alg. 3.1 calculate the most probable tracks that end at each frequency bin for each segment

```

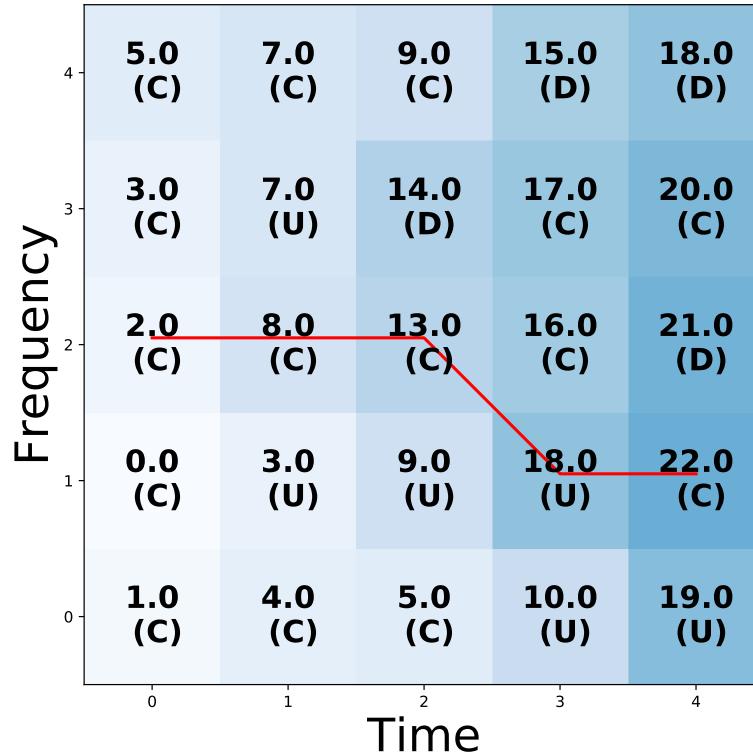
1: Input:  $C, T$  {log-likelihood,transition matrix}
2: Output:  $\hat{\nu}, V, A$  {most probable track, track probabilities, jumps}
3:
4: Initialisation
5: for Frequency ( $\nu_{0,k}$ ),  $k = 0 \rightarrow M - 1$  do
6:    $V_{0,k} = C_{0k}$ 
7:    $A_{0,k} = 0$ 
8: end for
9:
10: Iteration
11: for Segment,  $j = 0 \rightarrow N - 1$  do
12:   for Frequency ( $\nu_{j,k}$ ),  $k = 0 \rightarrow M - 1$  do
13:      $V_{j,k} = \max_i(C_{j,k} + T_i + V_{j-1,j+i})$ 
14:      $A_{j,k} = \operatorname{argmax}_i(C_{j,k} + T_i + V_{j-1,j+i})$ 
15:   end for
16: end for
17:
18: Identification
19:  $\hat{\nu}_{N-1} = \operatorname{argmax}_k(V_{N-1,k})$ 
20: for Segment,  $j = N - 1 \rightarrow 0$  do
21:    $\hat{\nu}_j = \hat{\nu}_{j+1} + A_{j,\nu_{k+1}}$ 
22: end for

```

ALGORITHM 3.1: The Viterbi algorithm in pseudo-code.  $N$  is the number of segments,  $M$  is the number of frequency bins in each segment. Here the maximisations over  $i$  run between  $\pm(n_1 - 1)/2$  where  $n_1$  is the size of the transition matrix. The values from Eq. 3.6 are stored in  $V$ , and the jumps are stored in  $A$ . The most probable track is denoted by  $\hat{\nu}$ .



(a) The input data



(b) The log-probabilities, jumps, and most probable path

Figure 3.2: Fig. 3.2a shows the observed data, i.e the log-likelihood values  $C_{j,k}$ . Fig. 3.2b shows the calculated log-probabilities  $V_{j,k}$ .  $A_{j,k}$  is shown in parentheses, where the UCD components correspond to  $i = [-1, 0, 1]$  respectively. The red line shows the path that gives the maximum probability. The transition matrix for the UCD jumps is  $[0, 1, 0]$  and corresponds to the un-normalised prior log-probabilities of these jumps occurring.

by using

$$V_{j,k} = \max_i(C_{j,k} + T_i + V_{j-1,k+i}), \quad (3.9)$$

where  $i$  is the size and direction of the jump. For example, in Fig. 3.2 columns 1–4 are calculated in order using Eq. 3.9, where it maximises over three possible previous positions in frequency. These positions are the frequency bins **UCD** of the current position. The size and direction of the jump,  $i$ , which gives the maximum probability is then saved to  $A_{j,k}$ . These are shown in parentheses below the log-probabilities in Fig. 3.2 where **UCD** correspond to values of  $i = [-1, 0, 1]$  respectively.

**Identification** The final stage of the algorithm identifies the most probable track. This is done by initially finding the highest log-probability values in the final time segment,  $\max_k(V_{N-1,k})$  (line 19 in Alg. 3.1). In the lower panel of Fig. 3.2 this is located at position  $j, k = 4, 1$  with  $V_{4,1} = 22$ . To find the track which corresponds to this, the values in  $A_{jk}$  are followed backwards from this position (lines 20–21). For example, in Fig. 3.2 the final position is  $j, k = 4, 1$  and  $A_{j,k} = \text{Center} = 0$ , this means that at the previous segment the most probable track was at position  $j, k = 4-1, 1+0 = 3, 1$ . At this time  $A_{3,1} = R = 1$ , therefore, the next track element is at  $j, k = 3-1, 1+1 = 2, 2$ . This then continues until  $j = 0$  whereupon these retraced positions constitute the most probable track, highlighted in red in Fig. 3.2.

The most probable track is the one traced backwards from the highest probability final segment frequency position. However, tracks can also be traced back from any of the end-frequency positions, returning the most probable track conditional on a given final position. Such tracks should not be confused with the being equal to the second, third, fourth, etc. most probable tracks. Information regarding the rankings and properties of all possible tracks (excluding the most probable and conditionally most probable tracks) is lost during the maximisation procedures computed at each stage in the algorithm – a necessary consequence of the algorithm’s speed and efficiency.

### 3.5 Multiple detectors

If there are  $Q$  detectors operating simultaneously we have  $Q$  sets of data which can be combined appropriately to provide input to the Viterbi search described above. We must also modify the allowed transitions encoded within the transition matrix to take account of the extra prior constraints that are now available.

The received instantaneous frequency of a given astrophysical signal will be nearly the same for all ground-based **GW** detectors, and our algorithm should be sensitive to tracks that show this consistency in frequency. However there *will* be small differences between the frequencies measured at detectors that are not co-located, due to differential Doppler

shifts caused by Earth rotation. As a result the signal could fall in different frequency bins at each detector.

To account for these small differences in signal tracks in each detector, we reference the observed tracks to a third (pseudo) detector located at the centre of the Earth which would be insensitive to Earth spin. The signal frequencies in each real detector are then allowed to vary within a certain number of frequency bins from the track in the reference detector. In the examples that follow, we only consider the possibilities that the track in each real detector is no more than one frequency bin away from the reference track. We can tune the length of the [SFTs](#) to ensure this is a valid assumption. As well as differences in signal frequency, due to antenna patterns and other effects, the measured signal amplitude may differ between the detectors. In the following example we assume that the signal has the same amplitude in each detector, however, in Sec. 3.8 we discuss the case where they differ.

We will now show how the algorithm in Sec. 3.4 can be modified to handle a two-detector network (i.e.,  $Q = 2$ ), however any number of detectors can easily be accommodated. In the two detector case the joint probability of two (real) tracks,  $\nu^{(1)}$  and  $\nu^{(2)}$ , and the geocentric track  $\nu$ , given the data, is

$$\begin{aligned} p(\nu, \nu^{(1)}, \nu^{(2)} | D^{(1)}, D^{(2)}) &\propto p(\nu)p(\nu^{(1)}, \nu^{(2)} | \nu) \\ &\quad p(D^{(1)} | \nu^{(1)})p(D^{(2)} | \nu^{(2)}), \end{aligned} \tag{3.10}$$

where  $D^{(1)}$  and  $D^{(2)}$  represent the data from the two detectors. The main difference between this and that described in Sec. 3.4 is that the track probabilities  $V_{j,k}$  are stored for the geocentric pseudo-detector. The main iterative calculation (defined for the single detector case in Eq. 3.9) now becomes

$$V_{j,k} = \max_{i,l,m}(C_{j,k+l}^{(1)} + C_{j,k+m}^{(2)} + T_{i,l,m} + V_{j-1,k+i}), \tag{3.11}$$

where  $C^{(1)}$  and  $C^{(2)}$  refer to the log-likelihoods in detectors 1 and 2 respectively and the transition matrix  $T$  is an  $n_1 \times n_2 \times n_3$  matrix, where  $n_1$  dimension refers to the jump from the previous time step,  $n_2$  and  $n_3$  refer to the relative frequency positions in each real detector. The transition matrix is now three-dimensional and holds the prior log-probabilities of  $p(\nu)$  and  $p(\nu^{(1)}, \nu^{(2)} | \nu)$ . We now need to maximise over three indices:  $i, l$  and  $m$ . The index  $i$  refers to the size and direction of the jump at the geocentre (as before). The indices  $l$  and  $m$  refer to the number of frequency bins by which the two real tracks deviate from the geocentre track. For example, if the most probable track in the geocentred detector is in bin  $j, k = 5, 12$  and the values of  $i, l, m = 0, -1, 1$ , then detector 1 is in position  $j, k = 5, 11$  and detector 2 is in position  $j, k = 5, 13$  and the geocentred track was in the position  $j, k = 4, 12$  at the previous time step. As a result, the track at

the geocentre is only affected by Doppler modulations from the Earth’s orbit whereas the tracks in the real detectors include Doppler modulations from the Earth’s spin.

At every time step the frequency bin position for each real detector is forced to be within  $n_l$  or  $n_m$  bins of the track in the geocentred detector, where  $n_l$  and  $n_m$  depend on how much each detector could possibly be Doppler shifted. As mentioned previously, we only consider the case where  $n_l = 1$  and  $n_m = 1$ , allowing the track from each real detector to be at most one frequency bin away from the geocentred track position. While we tune the **SFT** length to keep this condition for different frequencies, it is also possible to tune the values of  $n_l$  and  $n_m$  to get a similar effect. The implementation of the multi-detector algorithm is similar to the single detector case described in Sec. 3.4. However in the single detector case there is only a single variable to be maximised over for each time-frequency bin. This variable is the frequency jump from the position in the previous segment. For the multi-detector case there are at least three variables to be maximised over: the probability of the jump,  $i$ , at the geo-centre and the probability of the signal being in the surrounding positions in each of  $Q$  real detectors,  $l, m, \dots$ . The values of  $i, l, m, \dots$  are then saved to  $A_{j,k}$  and are ultimately used to reconstruct the most probable consistent tracks in each real detector.

As in Sec. 3.4, there are three main sections: Initialisation, iteration, and the identification. For the multi-detector case each element is modified as follows.

**Initialisation** The first-row calculation (lines 5–8) in Alg. 3.1, are now modified to additionally maximise over the real detector track positions  $l$  and  $m$ . For each time-frequency bin the maximum sum of the log-likelihoods is saved together with the frequency locations of the corresponding tracks in the real detectors. The index  $i = 0$  is kept constant as there is no previous position.

**Iteration** To process the subsequent time segments, lines 13–14 in Alg. 3.1 are modified to account for two (or more) detectors. Line 13 of Alg. 3.1 is changed to calculate Eq. 3.11, the log-probability of a track at the geocentre ending in bin  $j, k$  given that signal is in the real detector positions of  $j, k+l$  and  $j, k+m$ . Line 14 is then modified so that  $A_{j,k}$  stores the jump values,  $i$ , and the real detector positions,  $l$  and  $m$ , which returned the highest probability.

**Identification** The most probable track is identified in the same way as for the single detector case, first by finding the maximum value in the final time step of  $V_{j,k}$  (line 19 in Alg. 3.1). The track at the geocentre can then be found by iteratively following the jump values stored in  $A_{j,k}$  back from this position. The track in each of the real detectors is determined by using the values of  $l$  and  $m$  indices also stored in  $A_{j,k}$  to find the relative position of the track in each real detector compared to the geocentre.

This method can be extended to more than two detectors by including additional datasets and expanding the corresponding number dimensions of the maximisation procedures in the iterative steps.

## 3.6 Memory

In this section we extend the basic Viterbi algorithm to improve its sensitivity to non-stochastic signals where there is some knowledge of its frequency evolution. We do this by including a form of ‘memory’ and this extension applies to both the single and multiple-detector cases. Rather than considering only the previous step in our decision-making process, we now include the previous  $m + 1$  steps and expand the transition matrix to include these values. A memory of  $m = 0$  therefore corresponds to the methods described in previous sections. With a non-zero memory the transition matrix can a-priori make certain sequences of jumps more probable and assign different prior probabilities for these jump sequences e.g., ‘up then centre’ may be less preferable to ‘centre then centre’. As a result we can increase the chance of the most probable track matching an expected astrophysical signal. In a single detector search with a memory of  $m = 1$ , if we only allow **UCD** transitions, then for every frequency bin we save 3 values. These are proportional to the log-probabilities of a track coming from a **UCD** bin in the previous time step, where the maximisation is over the corresponding **UCD** bins two time steps back. Eq. 3.11 then is then modified to,

$$V_{j,k,s} = \max_h (C_{j,k} + T_{s,h} + V_{j-1,k+s,k+s+h}), \quad (3.12)$$

where  $s$  and  $h$  refer to the **UCD** jumps at the time step  $j - 1$  and  $j - 2$  respectively. Similar to the previous two sections, the algorithm is split into three parts: initialisation, iteration, and the track identification:

**Initialisation** The initialisation process needs to populate the first  $m + 1$  steps before the main iteration can start. At the first time step, the elements  $V_{0,k,s}$  are set to the log-likelihoods  $C_{0,k}$  as in Sec. 3.4. There is no previous time step, so the element  $s$  is not relevant. At the second time step,  $V_{1,k,s}$  is calculated using Eq. 3.12, where there is no maximisation over  $h$ , it is assumed to be 0, or a center jump. As there is no data before  $j = 0$ , the maximisation at this point will always return the jump which has the largest prior probability, which in this case is a center jump. Therefore, the maximisation returns the same value for all frequency bins and can be set to a center jump.

**Iteration** For all following time steps the values for each element of  $V_{j,k,s}$  in Eq. 3.12 are calculated. This quantity is proportional to the log-probability of the track ending

in time-frequency bin  $j, k$ , which was in the previous position of  $j - 1, k + s$ . The corresponding value of  $h$  that maximised the log-probability of the track is recorded in  $A_{j,k,s}$ .

**Identification** The most probable track is identified in a similar way to the non-memory cases, by finding the highest-valued last element,  $V_{N-1,k,s}$ . The values of  $s$  and  $h$  are then followed back to find the most probable track. As an example, let us assume the most probable track finishes in bin  $j, k, s = 10, 5, 0$ , where the value of  $m$  is  $A_{10,5,0} = 1 = \text{up}$ . The previous position is then  $j, k, s = 10 - 1, 5 + s, m = 10 - 1, 5 + 0, 1 = 9, 5, 1$  with a value  $A_{9,5,1} = 0 = \text{Center}$ , and the next track position is  $j, k, s = 9 - 1, 5 + 1, 0 = 8, 6, 0$  etc. The values of  $j, k$  along this track describes most probable path.

The number of elements over which one must search increases rapidly with memory length, and has a strong impact on the computational cost of the analysis. For the single detector Viterbi approach the number of calculations made is  $3 \times N \times M$  if we only allow UCD jumps, where  $N$  and  $M$  are the number of time and frequency bins respectively. When memory is included this increases to  $3^{m+1} \times N \times M$ .

## 3.7 Summed input data

In this section a method of incoherently-summing a set of SFTs to increase the SNR of a signal in a segment is outlined. To be more precise, it is actually the log-likelihoods which are summed, i.e. the quantity in Eq. 3.8. We can write the new summed set of data  $F_j$  as,

$$F_j = \sum_i^{N_s} C_{i,k} \quad (3.13)$$

where  $N_s$  is the number of SFTs to sum together and the log-likelihood  $C(\nu_{i,k})$  is defined in Eq. 3.8. We can see this is possible by looking at Eq. 3.7, where we can use the product of likelihoods,

$$\begin{aligned} p(D \mid \nu) &\propto p(x_1, x_2 \dots x_n \mid \nu) \\ &\propto p(x_1 \mid \nu) \dots p(x_n \mid \nu) \\ &\propto \exp \left( \sum_i C_{j,k} \right). \end{aligned} \quad (3.14)$$

If the data contains gaps where the detector was not observing, then we fill the gaps in the power spectrum with a constant value which is the expectation value of the log-likelihood. The procedure of filling in the gaps of the data is completed before any summing. There-

fore, the data should have the same mean regardless of how much real data is in each sum. In the examples that follow, we sum the **SFTs** over the length of one day.

The main motivation for summing the data is to increase the **SNR** of a signal in the segments. The risk is that a signal can move between adjacent frequency bins during a day. To reduce this risk, we choose the frequency bin width such that it is more likely that a signal will be contained within a single frequency bin that cross a bin edge. In practise, to ensure that this is true, the segment or **SFT** length and the number of segments which are summed can be tuned for each search. As well as increasing the **SNR**, summing over one day should average out the antenna pattern. This means that the log-likelihood value in any bin should be more similar between detectors, however, there is still some variation due to the sky localisation and polarisation.

This also has two main effects on the transition matrix, the first is that as each segment of data is now one day long, a jump between frequency bins is far more likely, therefore, the transition matrix elements are modified to account for this. The second is that as the data is averaged over one day, the signal should remain in the same frequency bin between detectors, therefore, there is no longer a need for the multi-dimensional transition matrix described in Sec. 3.5.

The volume of the data is also reduced by a factor of  $1/N_s$ , therefore, the time taken for the algorithm to run is also reduced by the same factor.

## 3.8 Line-aware statistic

The multiple-detector algorithm described in Sec. 3.5 returns the most probable track of a common signal assumed to be in Gaussian noise. As a consequence the algorithm will return large values of the log-likelihood even if there are inconsistent values of **SFT** power between the detectors, either from non-Gaussian noise or because the signal is not equally strong in the two detectors. However a signal with unequal power in the two detectors is more likely to be a non-Gaussian instrumental line than an astrophysical signal. The line-aware statistic described in this section is designed to make the search more robust to such instrumental artefacts within realistic non-Gaussian data whilst maintaining sensitivity to astrophysical signals.

For most of the analysis examples presented here we use data which is the incoherent sum of 30-minute normalised **SFTs** over a day (described in more detail in Sec. 3.7). As a result the effects of the detector antenna patterns and of differential Doppler shifts are significantly reduced, and any signal should have a broadly similar summed log-likelihood in the same frequency bin in each detector. The statistic can then be modified such that we expect a similar log-likelihood in each detector.

We first consider the model of Gaussian noise with no signal present. Within a sin-

gle summed segment, the likelihood of Gaussian noise at frequency  $\nu$  is given by a  $\chi^2$  distribution,

$$p(F_j|\nu_j, M_N, I) = \frac{1}{2^{d/2}\Gamma(d/2)} F_j^{d/2-1} \exp\left\{-\frac{F_j}{2}\right\} \quad (3.15)$$

where  $F_j$  is the frequency domain power summed over sub-segments within a single day, as described in Sec. 3.7 and  $d$  is the number of degrees of freedom, equal to twice the total number of summed SFTs.  $M_N$  represents the model that the data is simply Gaussian noise. In the presence of a signal (model  $M_S$ ), the power should follow a non central  $\chi^2$  distribution in which the non-centrality parameter  $\lambda$  is the square of the **SNR**, ( $\lambda = \rho_{\text{opt}}^2$ ), i.e.

$$p(F_j|\nu_j, \lambda, M_S, I) = \frac{1}{2} \exp\left\{-\frac{F_j + \lambda}{2}\right\} \left(\frac{F_j}{\lambda}\right)^{d/4-1/2} I_{d/2-1}\left(\sqrt{\lambda F_j}\right). \quad (3.16)$$

If a signal is present we therefore expect the **SFT** powers in both detectors to follow Eq. 3.16. Assuming for the moment that the noise variance is the same in both, we can determine the evidence for model  $M_S$  by marginalising over  $\lambda$ ,

$$p(F_j^{(1)}, F_j^{(2)} | \nu_j, M_S, I) = \int_0^\infty p(\lambda, w_S) \\ p(F_j^{(1)} | \nu_j, \lambda, M_S, I) p(F_j^{(2)} | \nu_j, \lambda, M_S, I) d\lambda. \quad (3.17)$$

We set the prior on  $\lambda$  to be an exponential distribution of width  $w$ , this is done somewhat arbitrarily as we expect the majority of signals to have a low **SNR**. This distribution follows,

$$p(\lambda, w) = \exp\left(-\frac{\lambda}{w}\right). \quad (3.18)$$

On the other hand, if an instrumental line is present in one of the detectors we expect to see signal-like power in that detector and noise-like power in the other. The evidence for this ‘line’ model ( $M_L$ ) is therefore

$$p(F_j^{(1)}, F_j^{(2)} | \nu_j, M_L, I) = \int_0^\infty p(\lambda, w_L) \\ \left[ p(F_j^{(1)} | \nu_j, M_N, I) p(F_j^{(2)} | \nu_j, \lambda, M_S, I) \right. \\ \left. + p(F_j^{(1)} | \nu_j, \lambda, M_S, I) p(F_j^{(2)} | \nu_j, M_N, I) \right] d\lambda, \quad (3.19)$$

The third option to consider is the simple case of approximately Gaussian noise in both

of the detectors,

$$\begin{aligned} p(F_j^{(1)}, F_j^{(2)} \mid \nu_j, \lambda, M_G, I) &= p(F_j^{(1)} \mid \nu_j, M_G, I) \\ &\quad p(F_j^{(2)} \mid \nu_j, M_G, I). \end{aligned} \tag{3.20}$$

The posterior probability of model  $M_{\text{GL}}$ , which contains the probability of Gaussian noise or Gaussian noise with a line in one detector, (taken as mutually exclusive) is

$$\begin{aligned} p(M_{\text{GL}} \mid F_j^{(1)}, F_j^{(2)}, \nu_j, I) &= p(M_G \mid F_j^{(1)}, F_j^{(2)}, \nu_j, I) \\ &\quad + p(M_L \mid F_j^{(1)}, F_j^{(2)}, \nu_j, I), \end{aligned} \tag{3.21}$$

where we assume that  $M_G$  and  $M_L$  are mutually exclusive.

We can now find the posterior odds ratio for the presence of a signal over noise or a line,

$$\begin{aligned} O_{\text{SGL}}(F_j^{(1)}, F_j^{(2)} \mid \nu_j) &= \frac{p(M_S \mid F_j^{(1)}, F_j^{(2)}, \nu_j)}{p(M_{\text{GL}} \mid F_j^{(1)}, F_j^{(2)}, \nu_j)} = \frac{p(M_S \mid F_j^{(1)}, F_j^{(2)}, \nu_j)}{p(M_G \mid F_j^{(1)}, F_j^{(2)}, \nu_j) + p(M_L \mid F_j^{(1)}, F_j^{(2)}, \nu_j)} \\ &= \frac{p(M_S)p(F_j^{(1)}, F_j^{(2)} \mid M_S, \nu_j)}{p(M_G)p(F_j^{(1)}, F_j^{(2)} \mid M_G, \nu_j) + p(M_L)p(F_j^{(1)}, F_j^{(2)} \mid M_L, \nu_j)} \\ &= \frac{p(F_j^{(1)}, F_j^{(2)} \mid M_S, \nu_j)p(M_S)/p(M_G)}{p(F_j^{(1)}, F_j^{(2)} \mid M_G, \nu_j) + p(F_j^{(1)}, F_j^{(2)} \mid M_L, \nu_j)p(M_L)/p(M_N)} \end{aligned} \tag{3.22}$$

In practice it is convenient to use the log odds ratio,

$$\begin{aligned} \log \left[ O_{\text{SGL}}(F_j^{(1)}, F_j^{(2)}) \right] &= \log \left[ p(F_j^{(1)}, F_j^{(2)} \mid M_S) \right] \\ &\quad - \left[ \log \left( p(F_j^{(1)}, F_j^{(2)} \mid M_G) \right. \right. \\ &\quad \left. \left. + p(F_j^{(1)}, F_j^{(2)} \mid M_L)p(M_L)/p(M_G) \right) \right] \end{aligned} \tag{3.23}$$

As we are only interested in the maximum of  $\log \left[ O_{\text{SGL}}(F_j^{(1)}, F_j^{(2)}) \right]$ , the factor  $\log [p(M_S)/p(M_G)]$  can be dropped from the expression.

In this version of the Viterbi algorithm, rather than storing a value proportional to the log-probabilities as in Sec. 3.5, here we store a value proportional to the log-odds ratio. Here we take the log-odds ratio defined in Eq. 3.23, which is the log-odds of a signal having a similar power in each detector, and add the log-prior odds  $p(\boldsymbol{\nu} \mid M_S)/(p(\boldsymbol{\nu} \mid M_N) + p(\boldsymbol{\nu} \mid M_L))$  which is the log-prior or any particular track. By assuming that the track transitions for the line and noise model are equally probable for any jump, we set the denominator of

the prior-odds is a constant  $b$ . This then means Eq. 3.11 is modified to,

$$\begin{aligned}\hat{V}_{i,j} = \max_{k,l,m} & (T_{k,l,m} + b + V_{i-1,j+k} \\ & + \log [O_{\text{SGL}}(F_j^{(1)}, F_j^{(2)})]) ,\end{aligned}\quad (3.24)$$

where  $\hat{V}$  refers to a log-odds ratio. The maximised statistic now has three tuneable parameters: the width,  $w_s$  in Eq. 3.18, on the prior for a signal **SNR** squared,  $p_s(\lambda)$ , the width,  $w_L$  of the prior in the case of a line,  $p_L(\lambda)$ , and the ratio of the prior on the line and noise models,  $p(M_L)/p(M_G)$ . These parameters are optimised for each search, where we initially estimate the **SNR** of a signal we hope to be sensitive to in each time slice, then use this as a guide for the width of the signal prior. This is then repeated for an expected line **SNR** and this is used for the width of the line prior. The ratio of line and noise models runs in the range 0 to 1, we set this limit as we do not expect an instrumental line to be as likely as Gaussian noise in any particular frequency bin.

This line-aware statistic can be applied in a more powerful way when we use multiple detectors and is similar to the approach in [102]. The multiple-detector algorithm described in Sec. 3.5 returns the most probable track of a common signal assumed to be in Gaussian noise. As a consequence the algorithm will return large values of the log-likelihood even if there are inconsistent values of **SFT** power between the detectors, either from non-Gaussian noise or because the signal is not equally strong in the two detectors. However a signal with unequal power in the two detectors is more likely to be a non-Gaussian instrumental line than an astrophysical signal. The line-aware statistic described in this section is designed to make the search more robust to such instrumental artefacts within realistic non-Gaussian data whilst maintaining sensitivity to astrophysical signals.

For most of the analysis examples presented here we use data which is the incoherent sum of 30-minute normalised **SFTs** over a day (described in more detail in Sec. 3.7). As a result the effects of the detector antenna patterns and of differential Doppler shifts are significantly reduced, and any signal should have a broadly similar summed log-likelihood in the same frequency bin in each detector. The statistic can then be modified such that we expect a similar log-likelihood in each detector.

In a similar way to the single-detector case, we can write out the evidence for each of the three models as follows. If a signal is present we therefore expect the **SFT** powers in both detectors to follow Eq. 3.16. Assuming for the moment that the noise variance is the same in both, we can determine the evidence for model  $M_S$  by marginalising over  $\lambda$ ,

$$\begin{aligned}p(F_j^{(1)}, F_j^{(2)} | \nu_j, M_S, I) &= \int_0^\infty p(\lambda, w_s) \\ & p(F_j^{(1)} | \nu_j, \lambda, M_S, I) p(F_j^{(2)} | \nu_j, \lambda, M_S, I) d\lambda.\end{aligned}\quad (3.25)$$

We set the prior on  $\lambda$  the same as in the single detector case in Eq. 3.18. In this case, if an instrumental line is present in one of the detectors we expect to see signal-like power in that detector and noise-like power in the other. The evidence for this ‘line’ model ( $M_L$ ) is therefore

$$\begin{aligned} p(F_j^{(1)}, F_j^{(2)} | \nu_j, M_L, I) &= \int_0^\infty p(\lambda, w_L) \\ &\quad \left[ p(F_j^{(1)} | \nu_j, M_N, I) p(F_j^{(2)} | \nu_j, \lambda, M_S, I) \right. \\ &\quad \left. + p(F_j^{(1)} | \nu_j, \lambda, M_S, I) p(F_j^{(2)} | \nu_j, M_N, I) \right] d\lambda, \end{aligned} \quad (3.26)$$

The third option is the simple case of approximately Gaussian noise in both of the detectors,

$$\begin{aligned} p(F_j^{(1)}, F_j^{(2)} | \nu_j, \lambda, M_G, I) &= p(F_j^{(1)} | \nu_j, M_G, I) \\ &\quad p(F_j^{(2)} | \nu_j, M_G, I). \end{aligned} \quad (3.27)$$

We can now find the posterior odds ratio for the presence of a signal over noise or a line by following the same steps as in Eq. 3.22. Once again we write this as a log-odds ratio,

$$\begin{aligned} \log \left[ O_{S/GL}^{(2)}(F_j^{(1)}, F_j^{(2)}) \right] &= \log \left[ p(F_j^{(1)}, F_j^{(2)} | M_S) \right] \\ &\quad - \left[ \log \left( p(F_j^{(1)}, F_j^{(2)} | M_G) \right. \right. \\ &\quad \left. \left. + p(F_j^{(1)}, F_j^{(2)} | M_L) p(M_L) / p(M_G) \right) \right] \end{aligned} \quad (3.28)$$

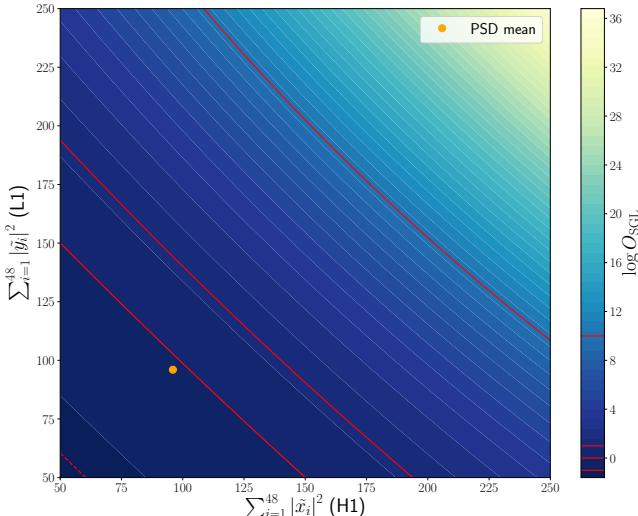
The factor  $\log [p(M_S)/p(M_G)]$  can again be dropped from the expression.

For the multi-detector case we then modify Eq. 3.11 to,

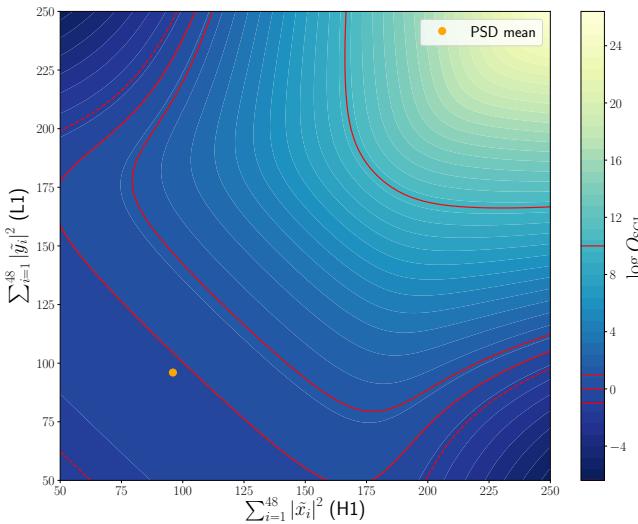
$$\begin{aligned} \hat{V}_{i,j} &= \max_{k,l,m} (T_{k,l,m} + b + V_{i-1,j+k} \\ &\quad + \log \left[ O_{S/GL}^{(2)} \left( F_j^{(1)}, F_j^{(2)} \right) \right]), \end{aligned} \quad (3.29)$$

where  $\hat{V}$  refers to a log-odds ratio. This is then optimised over the same three parameters as the single detector case.

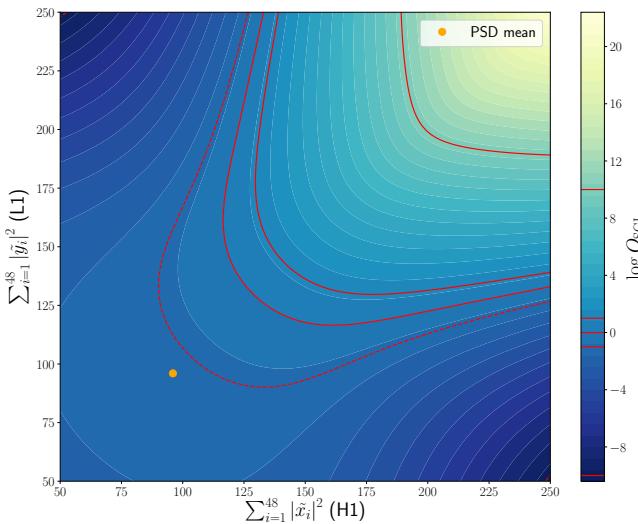
Fig. 3.3 shows an example of the output of the statistic in Eq. 3.28 for different **fast Fourier transform (FFT)** powers  $F$ .



(a) This shows the distribution of the lines aware statistic plotted against the **FFT** power in each detector. This example is for parameters  $p_s(\lambda) = 4, p_l(\lambda) = 0$  and  $p(M_L)/p(M_G) = 0$ . So the line part of the statistic is not operating.



(b) This shows the distribution of the lines aware statistic plotted against the **FFT** power in each detector. This example is for parameters  $p_s(\lambda) = 4, p_l(\lambda) = 5$  and  $p(M_L)/p(M_G) = 0.03$ . Here we include the line part of the statistic.



(c) This shows the distribution of the lines aware statistic plotted against the **FFT** power in each detector. This example is for parameters  $p_s(\lambda) = 4, p_l(\lambda) = 5$  and  $p(M_L)/p(M_G) = 1$ . Here the effect of lines is expected to be larger than the previous panel on the search. Therefore, the statistic forces the two detectors to have more similar power.

Figure 3.3: Lookup tables using the line aware statistic in Eq. 3.29.

### 3.9 Line aware statistic for consistent amplitude

In Sec. 3.8 the ‘line aware’ statistic was designed to penalise high **SFT** powers in a single detector and reward powers which have a similar **SNR**. This is often a useful statistic to use when the detectors have similar sensitivities, however, this is not always the case. During an observing run of a gravitational wave detector, their sensitivity will vary with time due fluctuating or new noise sources or potentially upgrades which can increase the sensitivity. A change in the sensitivity, or noise floor, affects the **SNR** of a possible signal in the data, i.e. a lower noise floor results in a higher **SNR**. In this section the above ‘line aware’ statistic is modified to account for the difference in sensitivities of the detectors, and therefore search for a consistent amplitude between detectors as opposed to **SNR**.

There are two main factors which are taken into account when determining how sensitive a detector is in a particular time interval: the **PSD** of detector and the duty cycle. The **PSD** of the detector is essentially how sensitive the detector is at that time and the duty cycle is the fraction of time in a given interval that the detector was collecting data. A decrease in the duty cycle and an increase in the **PSD** will decrease the **SNR** and vice-versa. To search for consistent amplitude Eq.3.24 is modified by weighting each detector by its **PSD** and duty cycle.

The definition of **SNR** is taken from [] as,

$$\rho_0^2 = \frac{h_0^2 T}{S} (\alpha_1 A + \alpha_2 B + \alpha_3 C), \quad (3.30)$$

where  $\rho_0$  is the optimal **SNR**,  $h_0$  is the signal amplitude,  $T$  is the time of observation,  $S$  is the noise **PSD** and the term in brackets include the antenna pattern of the detector. The signal with amplitude  $h_0$  will be the same amplitude at both detectors (H and L), therefore we can relate the **SNR** in each detector by,

$$\rho_L^2 = \frac{\rho_H^2 S_H T_L}{S_L T_H} \frac{(\alpha_1 A_L + \alpha_2 B_L + \alpha_3 C_L)}{(\alpha_1 A_H + \alpha_2 B_H + \alpha_3 C_H)}. \quad (3.31)$$

For the majority of the analysis that follows the **SFTs** are summed over one day, this will be explained in greater detail in Sec. 3.7. The components in the above equation which have the form  $(\alpha_1 A + \alpha_2 B + \alpha_3 C)$ , account for the antenna pattern of the earth as it rotates. These can be approximated to be the same for the two detectors H1 and L1 as we will essentially be averaging out the daily modulation by summing **SFTs**. Therefore we can simplify the above Eq. 3.31 to,

$$\rho_L^2 = \frac{\rho_H^2 S_H T_L}{S_L T_H} = l \rho_H^2. \quad (3.32)$$

This then gives a factor  $l = S_H T_L / S_L T_H$  which relates the **SNR** of each detector, where  $S$

and  $T$  are values that are known for a given data-set prior to running the search.

This ratio of **SNRs** can be included in the integral over **SNR** for the signal model in Eq. 3.25 as follows,

$$p(F_j^{(1)}, F_j^{(2)} | \nu_j, M_S, I) = \int_0^\infty p(\lambda, w_s) p(F_j^{(1)} | \nu_j, \lambda, M_S, I) p(F_j^{(2)} | \nu_j, l\lambda, M_S, I) d\lambda. \quad (3.33)$$

Similarly, the line model in Eq. 3.26 can be modified as,

$$\begin{aligned} p(F_j^{(1)}, F_j^{(2)} | \nu_j, M_L, I) = \int_0^\infty & p(\lambda, w_L) \left[ p(F_j^{(1)} | \nu_j, M_N, I) p(F_j^{(2)} | \nu_j, l\lambda, M_S, I) \right. \\ & \left. + p(F_j^{(1)} | \nu_j, \lambda, M_S, I) p(F_j^{(2)} | \nu_j, M_N, I) \right] d\lambda. \end{aligned} \quad (3.34)$$

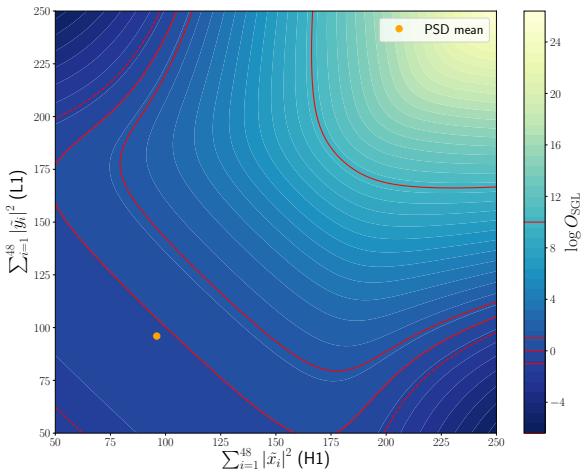
Fig. 3.4 shows an example of the values of the statistic described in Eq. 3.34 plotted against a range of **FFT** powers from each detector. This demonstrated how the statistic accounts for a difference in sensitivity on detectors by allowing the **FFT** power or effectively **SNR** to vary more.

In Fig. 3.4 we show an example of two detectors with large differences in sensitivity, and how the statistic which takes this into account can improve the search sensitivity in this case.

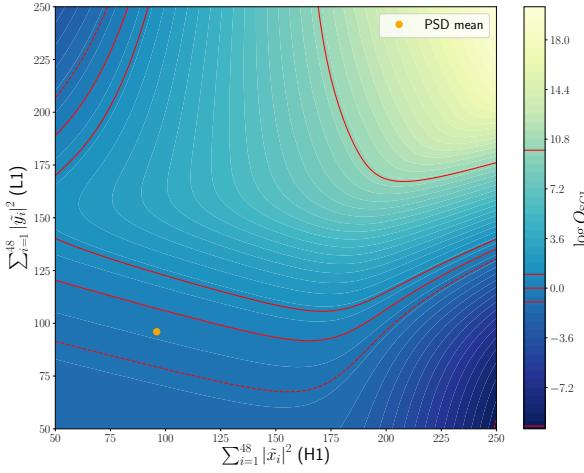
## 3.10 Testing the algorithm

The sensitivity of the algorithm was tested by searching for artificial signals from isolated pulsars added to three types of noise-like data: continuous Gaussian noise, Gaussian noise but with periods of missing data, and real detector data (the S6 **MDC** [53]). The S6 **MDC** refers to a standardised set of simulated signals which are injected into real data, this set is also what is used for the injections into the two Gaussian noise cases. We describe each of the tests in more detail in Sec. 3.10.1, 3.10.2 and 3.10.3, but several common pre-processing steps are performed before running these datasets through the Viterbi algorithm:

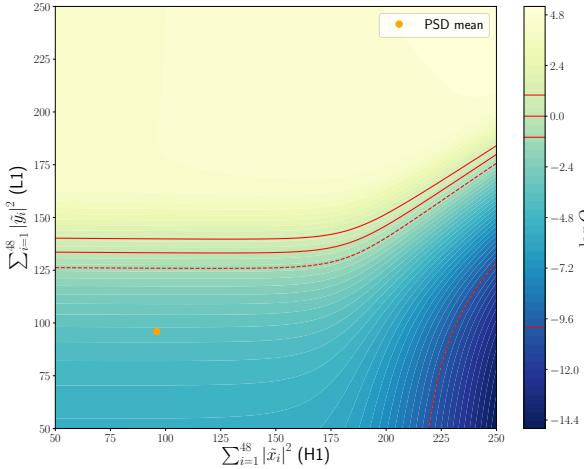
1. We read **SFTs** generated from 1800 s stretches of data in 2 Hz bands between 100 and 200 Hz. The **SFTs** length is chosen to ensure that any signal is likely to be contained within the width of a single frequency bin during the length of one day, rather than being split across the bin edges (see below).
2. We estimate the noise **PSD** for each **SFT** by calculating a running median over frequency using LALSuite code `XLALSFTtoRnmed` [78], this includes a bias factor to convert this to the mean and has a width of 100 bins. We then normalise the **SFT**



(a) This shows the distribution of the lines aware statistic plotted against the **FFT** power in each detector. This example is for parameters  $p_s(\lambda) = 2, p_l(\lambda) = 0$  and  $p(M_L)/p(M_G) = 0$ . So the line part of the statistic is not operating.



(b) This shows the distribution of the lines aware statistic plotted against the **FFT** power in each detector. This example is for parameters  $p_s(\lambda) = 2, p_l(\lambda) = 2$  and  $p(M_L)/p(M_G) = 1$ . Here the line part of the statistic has the same **SNR** and the signal part, i.e. we expect the **SNR** of a signal to be similar to that of a line.



(c) This shows the distribution of the lines aware statistic plotted against the **FFT** power in each detector. This example is for parameters  $p_s(\lambda) = 2, p_l(\lambda) = 10$  and  $p(M_L)/p(M_G) = 1$ . Here we expect the **SNR** of a line to be larger than a signal.

Figure 3.4: Lookup tables using the line aware statistic for consistent amplitude as in Sec. 3.9. Each of these use the parameters  $p_s(\lambda) = 4, p_l(\lambda) = 5$  and  $p(M_L)/p(M_G) = 0.03$ .

by dividing it by its running median, giving the noise-like parts of the spectrum a mean power of approximately one.

3. The **SFTs** are then summed over one day, as described in Sec. 3.7. The signal parameters are chosen so that within the frequencies of the search, the signal will not fall in more than two frequency bins over this period.

The differential Doppler shift of a signal seen at two detector sites due to the Earth's rotation  $\Delta f_{\text{rot}}^{(1,2)}$  is simply

$$\Delta f_{\text{rot}}^{(1,2)} = \frac{(\mathbf{v}^{(1)} - \mathbf{v}^{(2)}) \cdot \hat{\mathbf{s}}}{c} f_0, \quad (3.35)$$

where  $\mathbf{v}^{(1,2)}$  is the velocity of detector 1, 2 in an inertial reference frame,  $f_0$  is the instantaneous signal frequency in the frame,  $\hat{\mathbf{s}}$  is the unit vector in the direction of the source and  $c$  is the speed of light. The maximum difference in frequency seen by the two **LIGO** detectors is

$$\Delta f_{\text{rot}} \approx 6.5 \times 10^{-7} f_0, \quad (3.36)$$

so the frequency measured from a source in the equatorial plane with  $f_0 = 200$  Hz will differ by up to  $1.3 \times 10^{-4}$  Hz in the two detectors. This is  $\sim 4$  times smaller than the frequency bin width of 1800 s **SFTs** ( $5.6 \times 10^{-4}$  Hz), so signals at frequencies lower than this are likely to appear in the same frequency bin in the two detectors. Therefore, whilst at higher frequencies we still allow the signal to be in different frequency bins between the detectors, in the following searches, we do not allow this.

4. The data is then split into 0.1 Hz sub-bands which are overlapping by 0.05 Hz. These were chosen to ensure that signals are contained within a sub-band over the year. On these timescales the important contributions to the frequency evolution are the spin-down rate of the pulsar and the Doppler shift due to the earth orbit. To investigate the doppler shift, we can look at a signal at 200 Hz, using Eq. 3.35 we can calculate the maximum shift in frequency due to the earth's orbit as,

$$\Delta f_{\text{orbit}} = \frac{2\pi R_o}{T_o} \frac{1}{c} f_0 \approx 9.9 \times 10^{-5} f_0, \quad (3.37)$$

where  $T_o$  and  $R_o$  are the earth orbit time and radius. This gives a maximum doppler shift of 0.019 Hz, this is a  $\sim 1/5$  of the width of a sub-band, therefore, is more likely to be totally contained within a sub-band than crossing over the edge. To account for the cases where the signal frequency crosses over the edge of a sub-band, the sub-bands overlap by 0.05 Hz so that the majority of the signals should be completely contained within at least one of the sub-bands. To investigate the spin-down of the

pulsar, we look at the length of data,  $T = 4.05 \times 10^7$  s and we choose a sub-band width of 0.1 Hz. For a signal to drift over the width of a whole sub-band we would need f-dot of,

$$\frac{df}{dt} > \left| \frac{-0.1}{4.05 \times 10^7} \right| = 2.4 \times 10^{-9} \text{ Hz/s}. \quad (3.38)$$

The majority of the injections that follow satisfy this condition, signals which are greater than this, and therefore drift over multiple bands, are vetoed from the search.

5. The two detector Viterbi algorithm is then run using the line aware statistic (see Sec. 3.8). There are 4 parameters which we optimise in this search. The transition probabilities, where we have one parameter  $\tau$  which is the ratio of the probability of going straight to the probability of going either up or down. Due to the averaging procedure, the signals received at each detector are forced to follow a common track which is equal to the ‘imaginary’ detectors track. The other three parameters,  $w_S$ ,  $w_L$  and  $p(M_L)/p(M_N)$ , are described in Sec. 3.8.
6. The algorithm then returns the most probable track though the data, and the value  $\propto$  the log-odds in the final time step, i.e., the maximum final value,  $\max_j(V_{N,j})$ , in Eq. ??, which is then our detection statistic.

As an example of what the algorithm returns, Fig. 3.5 shows the tracks in the two detectors, H1 and L1. This also shows the log-odds ratio of ending in any frequency bin, i.e., all the elements in Eq. ???. In this figure, each time segment of the odds ratios have been normalised such that the sum of the odds ratios is 1.

In the following tests there are two main quantities which we use to determine the sensitivity. These are sensitivity depth  $\mathcal{D}$  and the optimal **SNR**  $\rho$ . The sensitivity depth,  $\mathcal{D}$ , is defined in [79] as,

$$\mathcal{D}(f) = \frac{\sqrt{S_h(f)}}{h_0}, \quad (3.39)$$

where  $S_h(f)$  is the single-sided noise **PSD** and  $h_0$  is the **GW** amplitude. The optimal **SNR** is defined as,

$$\rho^2 = \sum_X 4\Re \int_0^\infty \frac{\tilde{h}^X(f)\tilde{h}^{X*}(f)}{S^X(f)} df, \quad (3.40)$$

where  $X$  indexes the detectors and  $\tilde{h}(f)$  is the Fourier transform of the time series of the signal  $h(t)$ . This expression is defined in [52] for a double-sided **PSD** and we have defined it for the more common single-sided case.

### 3.10.1 S6 injections into gapless Gaussian noise

The first test involves injecting signals into Gaussian noise. The power spectrum of a Gaussian noise time-series follows a  $\chi^2$  distribution with two degrees of freedom, therefore,

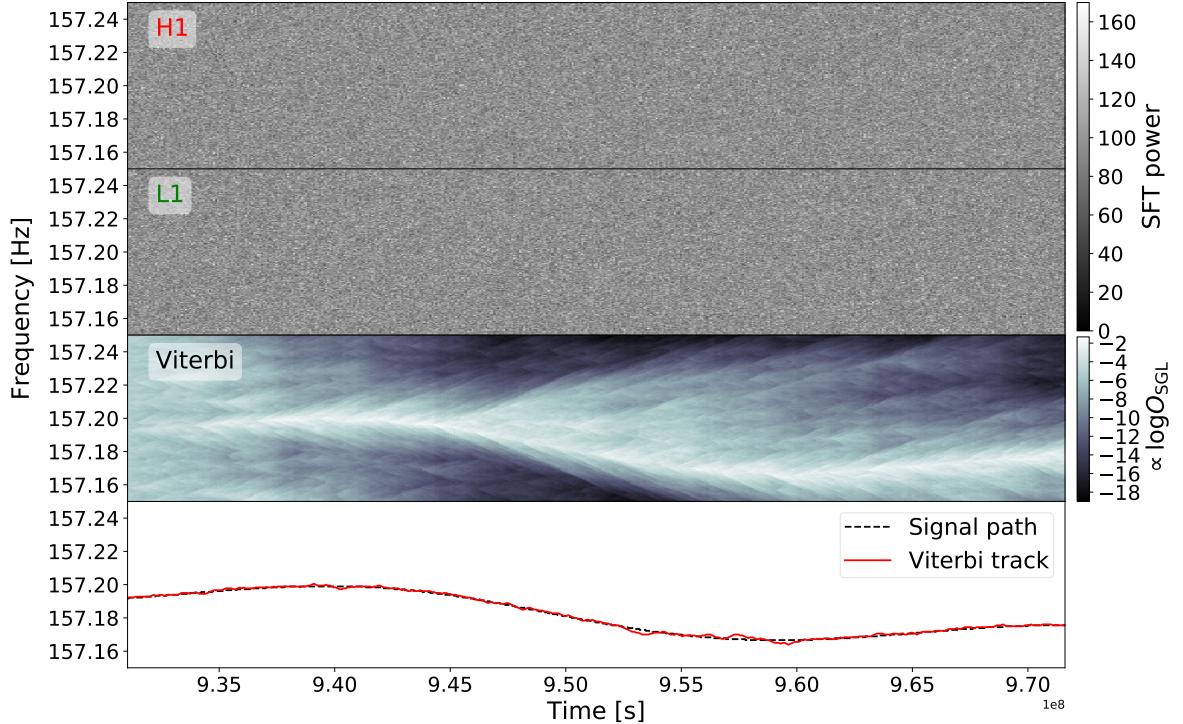


Figure 3.5: The results that the SOAP algorithm returns from an injection with an optimal SNR of 90, i.e., the SNR in H1 is 64 and the SNR in L1 is 62. The signal is injected into Gaussian noise, where the 1800 s SFTs have been summed over 1 day. The top panel shows a simulation of summed SFTs from H1, the second panel shows the same for L1, the third panel shows the values proportional to the log-odds ratios in Eq. ???. The log-odds have been normalised such that the sum of all the odds ratios in every time bin are equal to 1. The bottom panel shows the injected signal track (black dotted) and the track found in the ‘imaginary’ detector by the two-detector SOAP search with the line-aware statistic (red), both of these tracks are at the geo-centre. In this case the root median square (RMS) of the difference between the Viterbi track and injected signal track was  $\sim 1$  bin, where 1 bin is 0.00056 Hz wide.

as we search through the power spectrum, we generate spectrograms which follow a  $\chi^2$  distribution. These spectrograms are 0.1 Hz wide and are set at 0.05 Hz intervals between 100 Hz and 200 Hz. The bins are 1./1800 Hz wide and 1800s long, where the total length of data is the same as S6, i.e.,  $\sim 1.3$  years. We then generate the signals, where the pulsars parameters are fixed to the same values as the injections in the S6 **MDC** in this band, these values are outlined in [53].

The values of  $f_0$  for the injections were not always centred in a sub-band, therefore a number of sub-bands contained only part of the injected signal. These sub-bands were ignored as they contaminated the signal statistics and only the sub-band which contained the whole signal was accepted. This reduced the number of sub-bands from 2000 to 1762 with the removal of 238 sub-bands containing only part of a signal. This set also includes signals that drift across multiple sub-bands due to their high spin-down rate. Only two signals were removed due to their spin-down values, which were  $> 5 \times 10^{-9}$  Hz/s, these were the two hardware injections in the 100-200 Hz band.

For each injection the SOAP algorithm returns the detection statistic described in Sec. 3.8 and 3.10. We calculate a false alarm rate, which is the fraction of bands that have no injection that do exceed a given threshold. This is set to 1% and is used as a detection threshold. We then take all of the bands and if they pass the threshold we set them as detected, i.e., 1, and if they do not they are set as not detected, i.e., 0. This then leaves us with a set of binomial data, where the efficiency curves later in the paper are sigmoids which have been fitted to this. The sigmoid follows,

$$s(x; x_0, k) = \frac{1}{1 - \exp(-k(x - x_0))}. \quad (3.41)$$

The fit is done by sampling the posterior, i.e.,

$$p(x_0, k | b) \propto p(x_0, k)p(x | x_0, k), \quad (3.42)$$

where  $p(x_0, k)$  is the prior and we set to a flat prior and  $p(x | x_0, k)$  is the likelihood function which is defined by,

$$p(\bar{x} | x_0, k) = \prod_{j=0}^n \frac{n!}{k!(n-k)!} s(x_j | x_0, k)^k (1 - s(x_j | x_0, k))^{n-k}. \quad (3.43)$$

To plot the efficiency curves and lower and upper error bounds, we sample Eq. 3.42 using **MCMC** and then take the mean and the 5th and 95th percentiles respectively for each point in SNR or depth and plot these. Fig. 3.6a and 3.6c then show the efficiency curves for the analyses plotted against the signals optimal **SNR** and depth respectively. The

Table 3.1: Table shows the ranges of the search parameters and their optimised values for injections into gapless Gaussian noise, Gaussian noise with gaps and the S6 MDC. For gapless Gaussian noise and Gaussian noise with gaps, there are 10 parameter values spaced linearly between the limits. For the S6 MDC the parameters,  $\tau$ ,  $w_L$  and  $w_S$  were distributed in log space between the limits and  $p(M_L)/p(M_N)$  is distributed uniformly.

	$\tau$	$w_S$	$w_L$	$p(M_L)/p(M_N)$
<b>Gapless Gaussian</b>				
limits	[1.0,1.3]	[0.1,5.0]	None	0.0
optimised	1.1	2.06	None	0.0
<b>Gaussian with gaps</b>				
limits	[1.0,1.3]	[0.1,5.0]	None	0.0
optimised	1.1	2.06	None	0.0
<b>S6 MDC</b>				
limits	[1.0,1.1]	[0.1,5.0]	[0.1,6.0]	[0.0,1.0]
optimised	1.00000001	4.0	5.0	0.0387

parameters of the search and their optimised values are shown in Tab. 3.1. Where we set the prior on the line model to 0 as this part is irrelevant to this search due to the lack of lines in the data.

From this we can determine that in Gaussian noise without gaps, the Viterbi algorithm can detect to an SNR of  $\sim 60$  and a depth of  $\sim 33 \text{ Hz}^{-1/2}$  with 95% efficiency at a 1% false alarm.

Fig. 3.6b and 3.6d, show the RMS of the difference between the injected signal track and the track found by Viterbi for SNR and sensitivity depth respectively. This shows that at SNR of 60, where we are detecting signals with a 95% efficiency, the signals have a mean RMS of  $\sim 2$  frequency bins. Here one bin width is 0.00056 Hz therefore, we have and RMS of  $\sim 0.0012$  Hz.

### 3.10.2 S6 injections into Gaussian noise with gaps

In the second test, we attempt to more closely mirror the S6 MDC [53] in two stages. The first uses the same injection method as Sec. 3.10.1 however, removes the SFTs where there are gaps in S6. The second uses the same injection method again including gaps, however, uses a different value for the noise floor for each SFT, this is calculated for each band and SFT from S6 data.

Both detectors in S6 had a duty cycle of  $\sim 50\%$  [80], which means that there are sections of time where there is no data in either one or both detectors. In the sections

where one detector is observing but the other is not, the multi-detector statistic will not behave correctly as it only has access to data from a single detector. In these sections we switch from using the multi-detector statistic to the single-detector statistic using the same parameters, these are both defined in Sec. 3.8.

The process of removing sub-bands and generating efficiency curves is the same as in Sec. 3.10.1.

We set a 1% false alarm rate and generate an efficiency curve for **SNR** and depth in Fig. 3.6a and Fig. 3.6c respectively. From these efficiency plots we can see to an **SNR** of  $\sim 72$  or a depth of  $\sim 13 \text{ Hz}^{-1/2}$  at a 95% confidence with a false alarm of 1%.

The parameters of the search which were optimised and their optimised values are shown in Tab. 3.1.

In Fig. 3.6b and 3.6d show the **RMS** of the difference between the injected signal track and the track found by Viterbi for **SNR** and sensitivity depth respectively. This shows that at **SNR** of 72, where we are detecting signals with a 95% efficiency, the signals have a mean **RMS** of  $\sim 10$  frequency bins (0.0056 Hz).

### 3.10.3 Tests on the S6 MDC

For a more direct comparison to other **CW** searches and to see how the algorithm performs with real data, we test the two-detector SOAP algorithm using the S6 **MDC**. We focus this search on the 100–200 Hz band, there are two main reasons for this, one being that this is **LIGO**s most sensitive band and the other is that for much higher frequencies the signal will drift over larger frequency ranges, therefore, our **SFT** length will have to be changed. Here the 1800 s **SFTs** are split as in Sec. 3.10, whereafter normalisation, the data is split into 0.1 Hz wide sub-bands overlapping by 0.05 Hz.

The two-detector SOAP algorithm using the line-aware statistic in Sec. 3.8 is then run on each sub-band under the assumption that the detectors have the same sensitivity. For this search we have four parameters which we optimise, the ranges and optimised values are shown in Tab. 3.1.

As in Sec. 3.10.2, only the sub-bands which contained the entire frequency evolution of the signal were selected. Out of the 2000 sub-bands, 238 were removed due the sub-band only containing part of the signals frequency evolution. The main difference between the analysis for Gaussian noise and real data is that the real data is contaminated with instrumental lines. This means that whilst the techniques described in Sec. 3.8 reduce the number of contaminated bands with a high statistic value, there are still instrumental lines which are coincident between the detectors and which could not be removed with these techniques. Within the data there are large number of lines at integer Hertz, which are seen in coincidence between the two detectors, these are thought to originate from digital electronics [81]. Therefore the frequency bins  $\pm 1$  bin of each integer frequency in Hertz

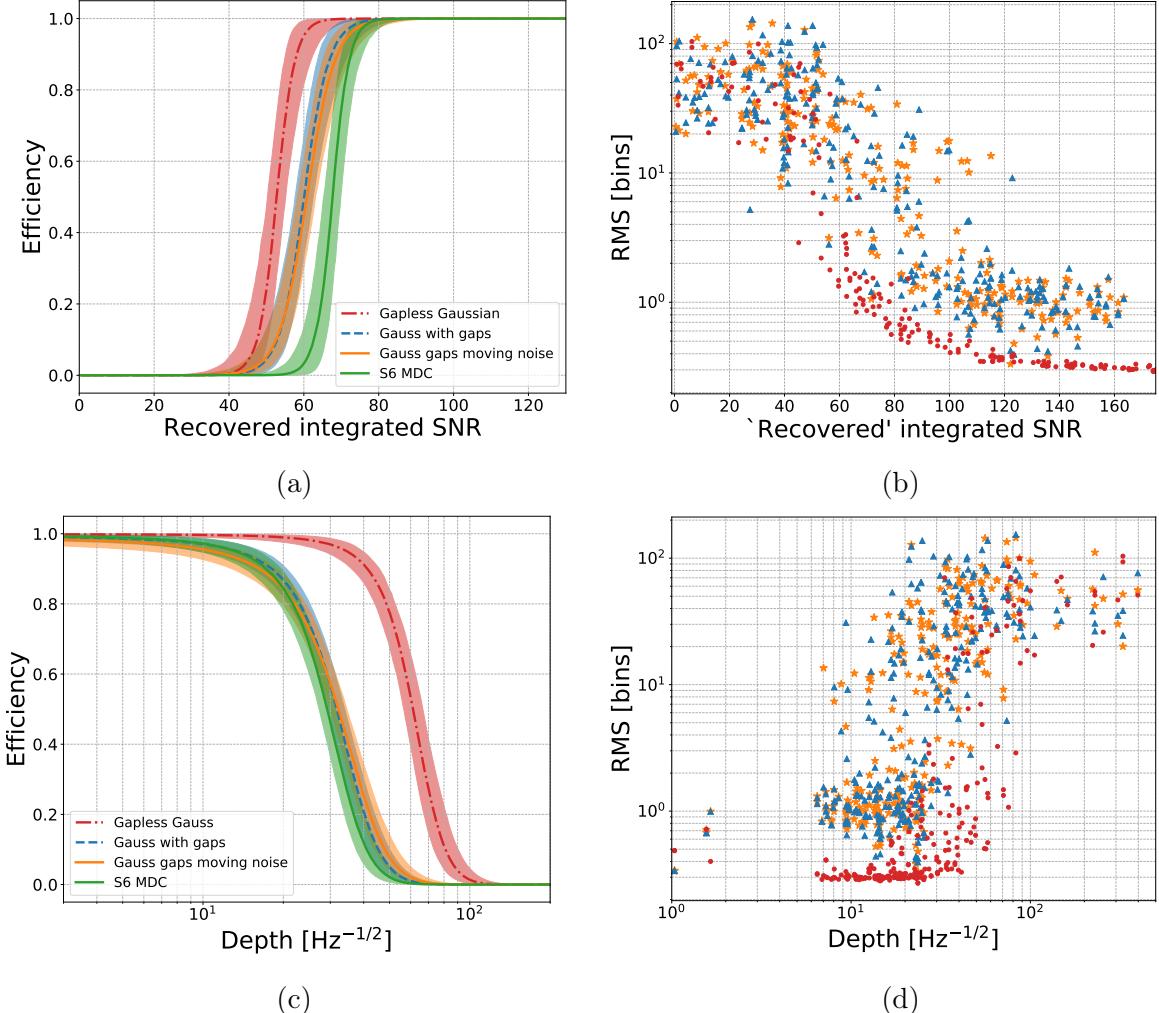


Figure 3.6: Panels 3.6a and 3.6c show the detection efficiency as a function of **SNR** and depth respectively. Here **SNR** is the the integrated **SNR** which we would expect to recover from the available data. The four curves refer to injections into gapless Gaussian noise (red), Gaussian noise with gaps in data, where the noise floor is either fixed (blue-dashed) or it is moving with time (orange) in the same way as the **S6 MDC** and injections into real data i.e., the **S6 MDC**. In the gapless Gaussian noise case, the recovered integrated **SNR** refers to the **SNR** the injection would have if it had the same amount of data as in the cases with gaps. The curves are made by fitting a sigmoid Eq. 3.39 to binomial detection data with a 1% false alarm rate, as explained in Sec. 3.10.1, the error bounds are the 5% and 95% intervals. At 95% efficiency and a 1% false alarm rate, this shows we can detect to an **SNR** of  $\sim 60$  and a sensitivity depth of  $\sim 34 \text{ Hz}^{-1/2}$  for gapless Gaussian noise and an **SNR** of  $\sim 69$  and  $72$  and a sensitivity depth of  $\sim 13 \text{ Hz}^{-1/2}$  and  $\sim 10 \text{ Hz}^{-1/2}$  for the Gaussian with gaps case with fixed noise floor and moving noise floor respectively. For the **S6 MDC** we can detect an **SNR** of  $\sim 74$  and a sensitivity depth of  $\sim 13 \text{ Hz}^{-1/2}$ . Panels 3.6b and 3.6d show the **RMS** of the difference between the injected signal track and the track found by SOAP as a function of **SNR** and sensitivity depth respectively. This is shown in units of bins where each bin is  $0.00056 \text{ Hz}$  wide.

were removed and filled with the expectation value of the noise. To remove instrumental effects at other frequencies, the sub-bands which gave values of our statistic above a chosen threshold were investigated by eye. In this case 344 sub-bands were investigated, and any which were contaminated were vetoed. From these 344 sub-bands, 193 were removed from the analysis. The predominant feature in the bands which were removed were broad spectral features which lasted the whole run. Therefore, out of the 2000 sub-bands which are searched over, a total number of 431 sub-bands were removed.

The process to calculate the efficiency curves is the same as in Sec. 3.10.2 and 3.10.1.

Fig. 3.6c and Fig. 3.6a show the efficiency curves for **SNR** and depth respectively. These show that we can detect and **SNR** of  $\sim 74$  and a sensitivity depth of  $\sim 13 \text{ Hz}^{-1/2}$  with an efficiency of 95% at a false alarm of 1%. These results can then be compared to other searches in the S6 MDC comparison paper [53]. Whilst we only search in the 100 - 200 Hz range, the closest comparison in [53] is the test in the 40 - 500 Hz range, such as in Fig. 4 in [53]. Here our algorithm sits roughly in the middle of all other searches in terms of sensitivity.

## 3.11 Optimisation of Line-aware statistic.

For the above searches we used optimised versions on the line aware statistic, however, we have yet to explain how this was optimised. The aim is to find the best parameters for any given search; the four parameters are  $\tau, p_S(\lambda), p_L(\lambda)$  and  $p(M_L)/p(M_G)$ . We find the optimum values empirically by running the entire search for each parameter value that need to be tested. This is possible as the search is relatively fast, this will be explained in Sec .3.11.5. The line aware statistic is time consuming to calculate, therefore, to reduce the computational time, it is pre-calculated and places into lookup tables such that it is calculated once and called many times. These lookup tables were calculated for values of **FFT** power  $F$  in the range 1 to 400 in each of the detectors as shown in Fig. 3.3. For each parameter the ranges in which they were optimised were chosen based on expected **SNRs** of each of the injections.

We can the use a measure of the sensitivity of that search and pick the lookup table which gives the highest sensitivity. We measure the sensitivity by taking the value of **SNR** which is at 80% efficiency at 1% false alarm.

### 3.11.1 Gaussian noise simulations

For injections into Gaussian noise, we know that there are no instrumental lines, therefore, we do not need to optimise over the ‘lines’ part of the statistic and can set the parameter  $p(M_L)/p(M_G)$  to zero which renders the parameter  $p_L(\lambda)$  redundant. This then reduces the complexity of the problem by leaving us with only two parameters to optimise over,

Table 3.2: Table shows the ranges of the search parameters and their optimised values for injections into Gaussian noise and the S6 MDC. For Gaussian noise there are 30 parameter values spaced linearly between the limits. For the S6 MDC the transition matrix parameters,  $\tau$ , had three values space between the limits. This is because the search is relatively insensitive to this parameter. The parameters  $w_L$ ,  $w_S$  and  $p(M_L)/p(M_N)$  had 10 parameters distributed in linearly between the limits.

	$\tau$	$w_S$	$w_L$	$p(M_L)/p(M_N)$
<b>Gaussian noise</b>				
limits	[1.0,1.1]	[0.1,7.0]	None	0.0
<b>S6 MDC</b>				
limits	[1.0,1.1]	[0.1,10.0]	[0.1,20.0]	[0.0,0.3]

$\tau$  and  $p_S(\lambda)$ . Whilst this optimisation was partially done in Sec. 3.10, with the result in Tab. 3.1, this is repeated more completely here. The parameters were optimised in the range shown in Tab. 3.2.

For each point in Fig. 3.7, the entire SOAP search was run using the corresponding parameters as input. Efficiency curves are then generated for each of these runs and the values for the SNR at 80% efficiency are recorded. Fig. 3.7 then shows the 80% efficiency for each parameter value. There appears not to be any single value which gives an optimum, however, the dark stripe in Fig. 3.7 running from the bottom left to the red cross is the combinations of parameters which give the best result. The point where the red lines cross is the parameters used in previous searched in Sec. 3.10.1 and 3.10.2. This falls on the line where the algorithm performs best. Venturing far from this ‘optimum’ line does not change the results a great deal as the SNR does not change much. The search is then not particularly sensitive to choice of parameters in Gaussian noise.

### 3.11.2 S6 MDC injections

As the S6 MDC data-set is real detector data, there are many examples of instrumental lines. This is where we expect the line-aware statistic to have the greatest effect in rejecting lines. Here the all of the four parameters are optimised over in the ranges described in Tab. 3.1. This greatly increases the number of lookup tables which need to be generated and therefore the number of time the search needs to be run. Fig. 3.8 shows the slices through parameters space for each of the parameters where the values are the SNRs at 80% efficiency. These slices are made by taking the mean across the parameters not shown in the figure. The range chosen for the transition matrix parameter  $\tau$  as small with a small number of points, this is because the impact of this parameter was expected to be small. Fig. 3.8 demonstrates that changing this parameter does not have much affect on the

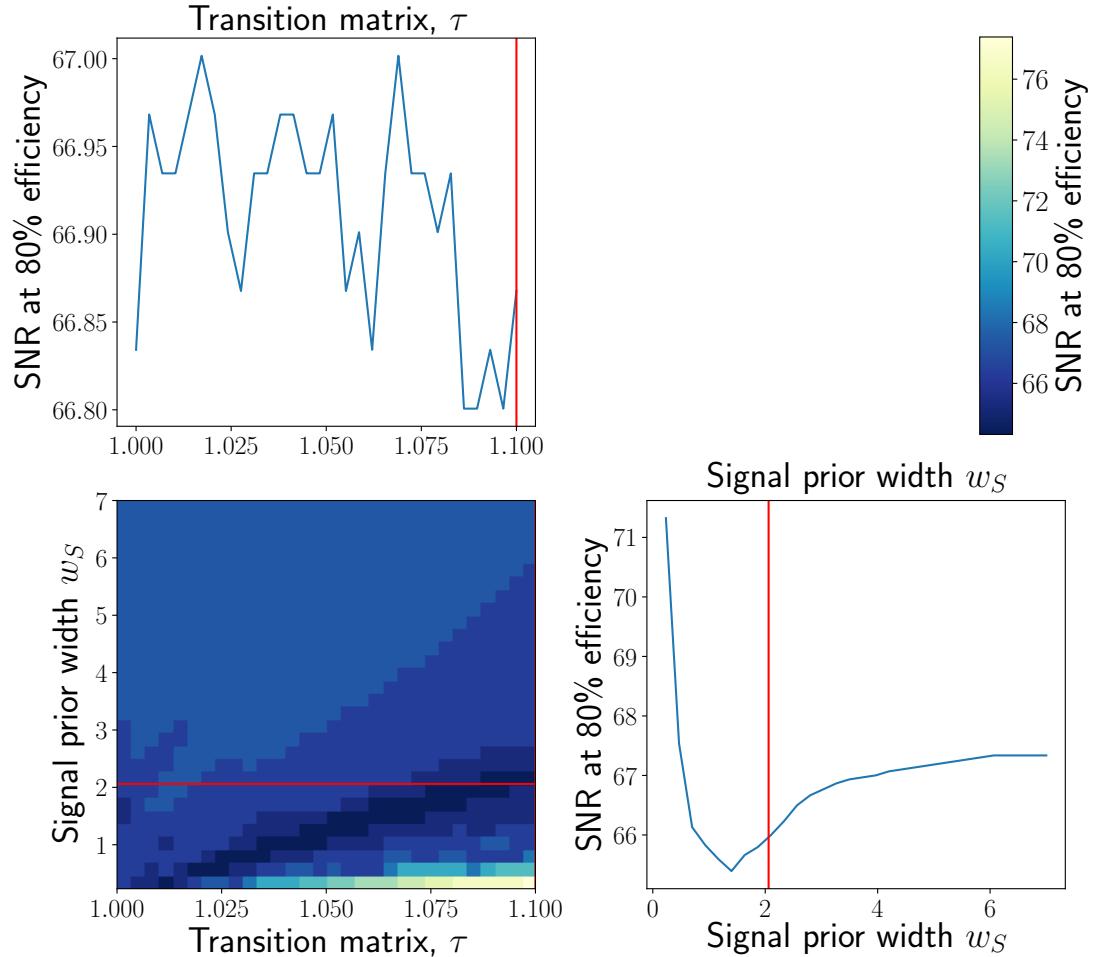


Figure 3.7: In Gaussian noise the transition matrix parameter  $\tau$  and the width of the prior on the signal case  $w_S$  were optimised. The key part to remember when reading this plot is that the lower the value of **SNR** the better the search has performed. Therefore darker blue areas are when the search performed better. This map shows that there is a line in parameter space where the search performed best. Also in Gaussian noise, the search is not that sensitive to the choice of parameter. The red lines on here shows the parameters used in the searches in this section.

final sensitivity. The parameter  $w_S$  appears to have a clear location around a value of 3 where the sensitivity is at a minimum. This is not far from the value of the parameter which is used for all other searches in this thesis (the red line). The parameters  $w_L$  and  $p(M_L)/p(M_G)$  show that the sensitivity drops substantially as the values increase. In these examples the optimum value may be out of the range chosen here. Whilst this could be improved slightly by running with a larger parameter space, this is a time consuming process. The results in Sec. 3.10 show that the current values of the parameters (red lines in Fig. 3.8) are sufficiently sensitive when compared to other searches. To choose optimum parameters for this search, the minimum point for each parameter was chosen. This is the

minimum in each of the diagonal plots in Fig .3.8. These values are,

$$\begin{aligned}\tau &= 1.0 \\ w_S &= 3.0 \\ w_L &= 20.0 \\ p(M_L)/p(M_G) &= 0.2879.\end{aligned}\tag{3.44}$$

To visualise how these better optimised parameters are to the values used in Sec. 3.10 we can compare how they perform on a simulated signal in a time-frequency spectrogram. In one of the detectors a narrow instrumental line can also be simulated. In Fig. 3.9 an example a **CW** signal injected into Gaussian noise in both detectors. In the Gaussian noise optimised case the search is looking for high power in both detectors which the strong instrumental line satisfied when the astrophysical signal is weak. The S6 optimised statistic looks for more consistent **SNR** in each of the detectors. The two different lines show the statistic optimised on the Gaussian noise injections and the S6 **MDC** injections. This demonstrates how the ‘line aware’ statistic improves the robustness of the algorithm against non astrophysical signals.

### 3.11.3 Sensitivity with frequency

All of the above tests, the search was conducted in the range from 100-200 Hz. This was chosen to be within the most sensitive band of **LIGO** as this is where a signal is most likely to be discovered. However, signals can appear at much higher frequencies also. Therefore, it is important to see how the sensitivity of the search varies with the frequency.

For this test we simulated **CW** signals in Gaussian noise with no gaps in data. The injections used the same source parameters as in the a6 **MDC** [53] and the tests above. This has the exception that the integrated recovered **SNR** of the signal is sampled uniformly between 50 and 500. These injections were then made at frequencies of 100,250,500,750,1000,1500 and 2000 Hz, where the band width is 2 Hz. i.e. the simulations were in frequency bands 100-102 Hz, 250-252 Hz etc. The setup of the search was the same as in the above sections. Here each sub-band is 0.1 Hz wide, and the parameters of the SOAP search were as in Tab. 3.1.

Fig.3.10 shows the resulting efficiency curves from each of these tests. This is for a 1% false alarm rate, which means that 1% of sub-bands which contained no injection crossed the detection threshold. This plot shows how the sensitivity of the search drops as the frequency increases. This is perhaps unfair to the algorithm as we used the setup of the search which has been optimised for the range 100-200 Hz. Optimising the search means, choosing the parameters of SOAP, the key parameter which will affect this is the transition matrix. As the simulated signals frequency is increased, the scale of the Doppler

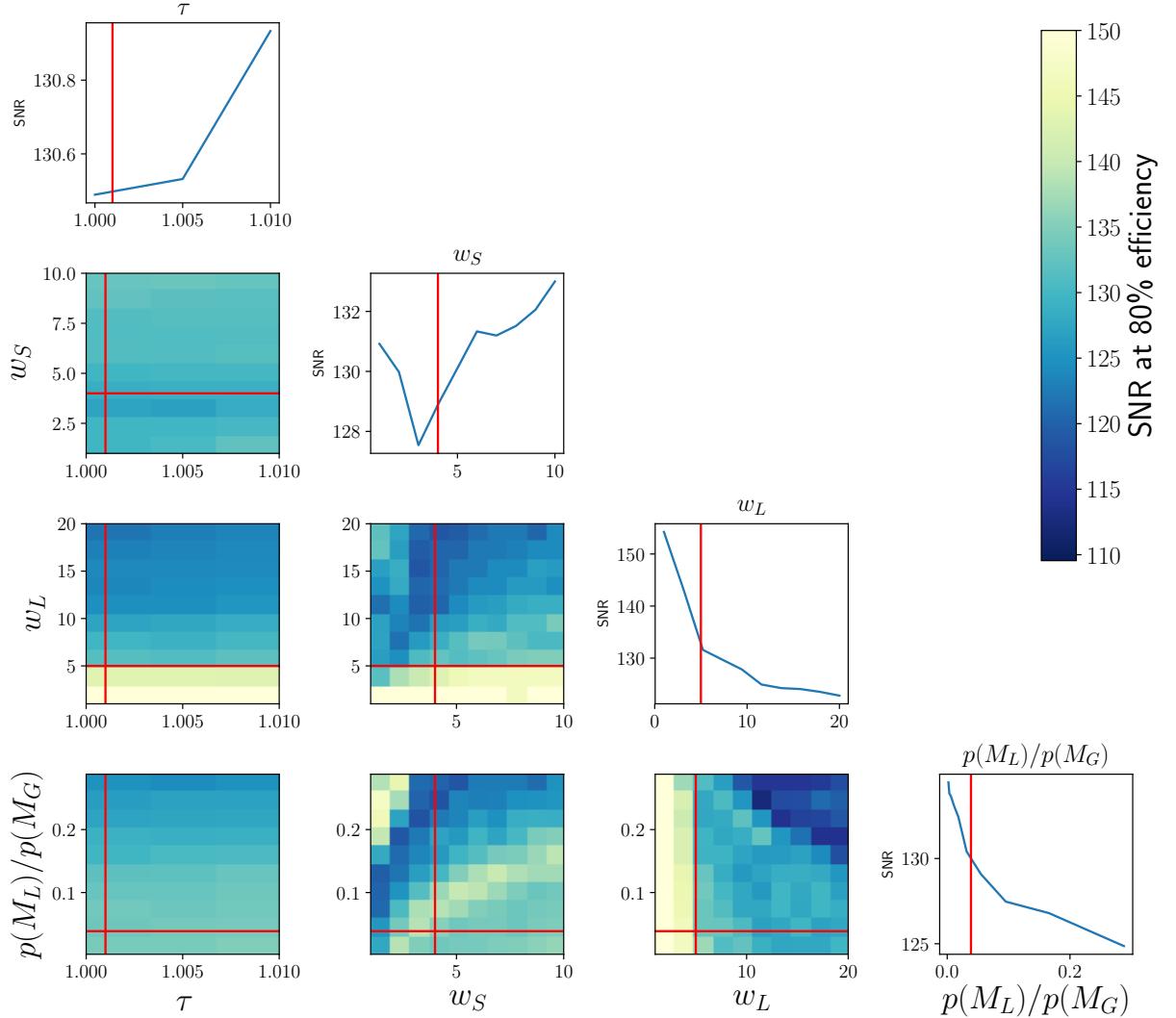


Figure 3.8: When using real S6 data, all four parameters of the search were optimised over on simulations in real data. The plot above shows the **SNR** at 80% efficiency for each of the parameters where the ranges are in Tab. 3.2. Lower values of **SNR** mean the search is performing better. The red lines show the parameters used in the searches in this section and the sections that follow. Whilst this does not seem optimal, the search does not underperform much using the current choice of parameters (red).

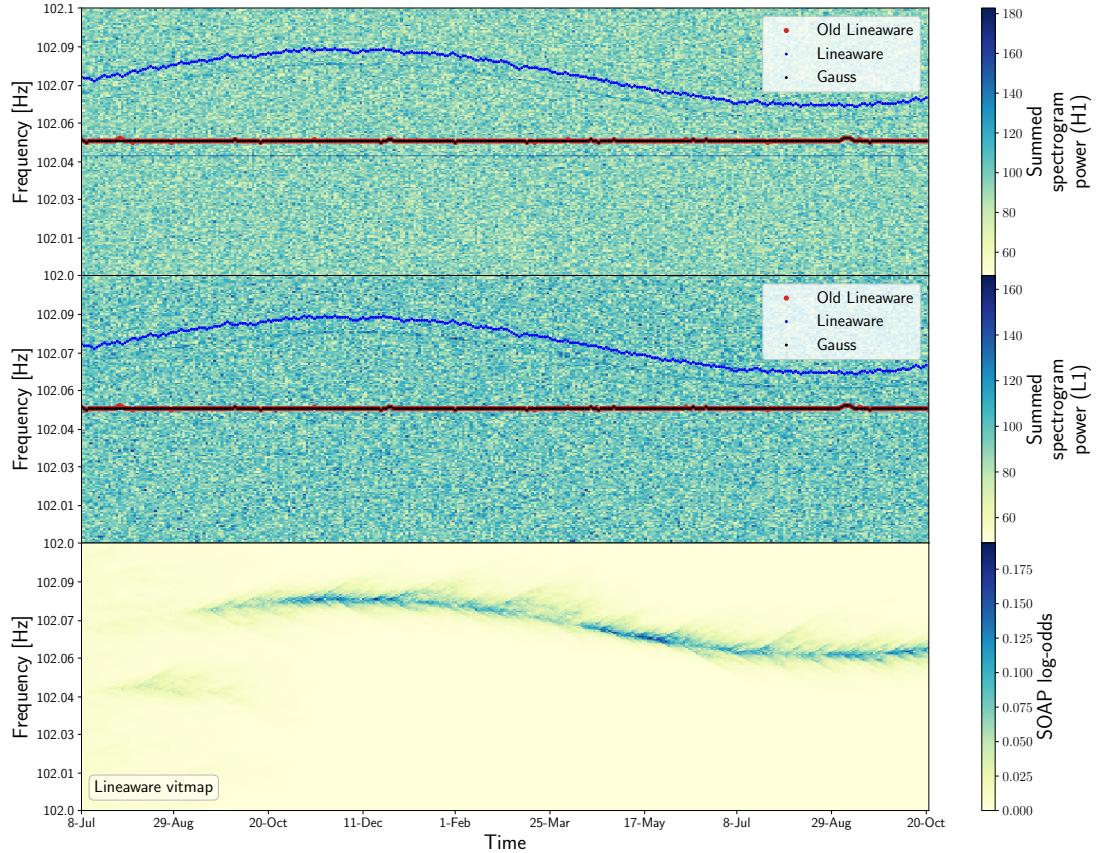


Figure 3.9: As a visualisation of the line aware statistic working, an astrophysical signal was simulated in both the H1 and L1 detectors. In one detector an instrumental line was added. In the top panel above the tracks identified by different values of the line aware statistic are shown shifted up by 10 frequency bins. This allows the underlying feature to be identified. In the top two panels the sinusoidal nature of the **CW** signal can be identified in the data, and was identified by the line aware statistic using the S6 MDC optimal values (blue line). The top panel also contains a narrow spectral line at  $\sim 102.5$  Hz. The values of the statistic optimised for the Gaussian noise case do not identify the signal but identify the spectral line as the most likely signal. The final panel shows the Viterbi map for the statistic values optimised in the S6 MDC.

modulation will also increase. This means that at higher frequencies the signal is more likely to jump more than a single frequency bin. The current setup of the search does not allow this size of jump, therefore, would struggle to identify this type of track. The other main factor which will decrease this sensitivity is the sub-band width of 0.1 Hz. Similarly to before, as the frequency increases the scale of the Doppler modulation will increase as in Eq. 3.37. For example at 1000 Hz, the Doppler shift is  $\sim 0.1$  Hz, the signal is then more likely to not be fully contained within a frequency band. Therefore, the search can not accumulate all of the injected **SNR**

Whilst these parameters of the search can be changed to depend on frequency, this was meant as a small test to see how sensitivity varies with frequency.

### 3.11.4 Searching for non CW sources

Whilst SOAP was designed to search for sources of **CWs**, when set up correctly, it can be applied to searching for many other types. This is because the search is essentially un-modelled, and in its simplest form looks for tracks of high power in a spectrogram.

#### CBC

CBC signals cover a wide frequency range and are very short signals in **LIGO** detectors. In previous SOAP searches the default length of **SFT** has been 1800 s, this is not suitable when searching for **CBC** signals. We can take shorter time base for the **SFTs**, however, to get a required resolution, these can overlap. **JOE: section not complete**

### 3.11.5 Computational cost

One of the main strengths of this search is the drastically reduced computational cost when compared to other current **CW** searches. The scaling of the computing cost can be estimated for a single detector by looking at the number of calculations that need to be made. The number of calculations for a single detector search,  $N_{\text{calcs}}^{(1)}$  is,

$$N_{\text{calcs}}^{(1)} = n_1^m NM, \quad (3.45)$$

where  $n_1$  is the size of the transition matrix,  $N$  is the number of **SFTs**,  $M$  is the number of frequency bins and  $m$  is the amount of memory described in Sec. 3.6. Where the computing cost scales linearly with the number of frequency bins and **SFTs**. In the following test we ignore ‘memory’ and look at the time taken for the single detector search where the time taken to read and save data is ignored. Here the data is the same size as the S6 **MDC** for a single detector search and the search is over a 0.1 Hz band, where we set  $n_1 = 3$ . This test, and the following test, was run locally on a MacBook Air with a 1.3 GHz Intel Core

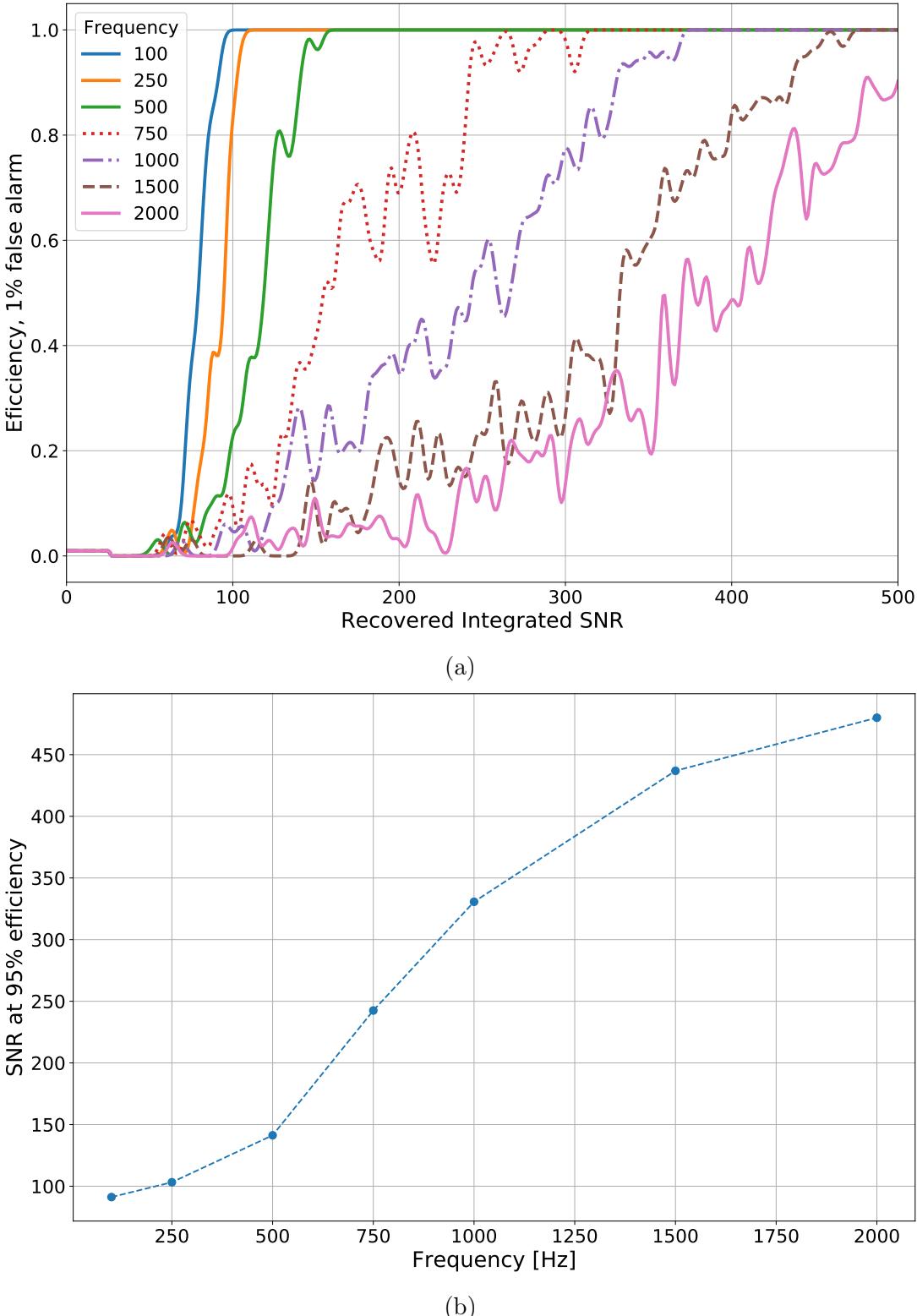


Figure 3.10: The sensitivity of the SOAP search in this configuration decreases as the frequency of the pulsar increases. This setup of data for the search however, was optimised for the 100–200 frequency band and can be changed for different frequencies. 3.10a shows the efficiency curves with 1% false alarm rate for each frequency. 3.10b shows the values from the efficiency curves at 90% efficiency.

i5 processor . We can then write the time taken , $T$ , as,

$$T = 0.56 \text{ sec} \left( \frac{N}{22538} \right) \left( \frac{M}{180} \right) \left( \frac{N_{\text{bands}}}{1} \right), \quad (3.46)$$

where  $N_{\text{bands}}$  is the number of different frequency bands. For the multiple,  $Q$ , detector case, we can then generalise Eq. 3.45 and write the number of calculations  $N_{\text{calcs}}^{(Q)}$  as,

$$N_{\text{calcs}}^{(Q)} = NMn_1^m \prod_{q=1}^Q n_{q+1}, \quad (3.47)$$

where  $n_1$  is the first dimension of the transition matrix,  $Q$  is the number of detectors and  $n_{q+1}$  is the size of the transition matrix element which refers to detector  $q$ . For our tests we set  $n_1 = n_{q+1} = 3$  and use 2 detectors i.e.,  $Q = 2$  which each have the same size data as the previous test. The actual time taken to run however, depends on the version of the algorithm which is run. For example, including the line aware statistic slows the search slightly. For the two detector where two SFT powers are summed,

$$T = 1.35 \text{ s} \left( \frac{N}{22538} \right) \left( \frac{M}{180} \right) \left( \frac{N_{\text{bands}}}{1} \right). \quad (3.48)$$

The same search now including the line aware statistic, which is implemented using a lookup table, changes this to,

$$T = 25.7 \text{ s} \left( \frac{N}{22538} \right) \left( \frac{M}{180} \right) \left( \frac{N_{\text{bands}}}{1} \right). \quad (3.49)$$

Other searches, excluding Einstein@home which takes on the order of months to run ( $> 100$  million core-hours [53]), take  $1 - 10$  million core-hours [53]. This search should take  $\mathcal{O}(10^3)$  core-hours.

## 3.12 Discussion

In this paper we describe an application of the Viterbi algorithm, called SOAP, to search for continuous sources of gravitational waves. This paper outlines the method and derives the statistics behind the method in a consistent Bayesian formalism. It then presents the results from the first set of tests of sensitivity for the SOAP algorithm on three separate datasets.

We tested SOAP on a set of fake isolated pulsar signals in the 100 – 200 Hz range, based on 1800s SFTs summed over one day. The three datasets that included these signals comprised continuous Gaussian noise, Gaussian noise but with temporal gaps corresponding to LIGO dead times in the S6 data run, and real data, i.e., the S6 MDC. Although a major

attraction of SOAP is its sensitivity to a wide range of signal types, in the tests above it was optimised to detect isolated pulsar signals below 100 Hz with low spin-down to offer a comparison with other CW searches. From these tests, by setting a 95% efficiency and a false alarm of 1%, we found that in the case of continuous Gaussian data we could detect a signal with an optimal SNR of  $\sim 60$  and a depth of  $\sim 33 \text{ Hz}^{-1/2}$  with an RMS of the difference between the injected and Viterbi track being  $\sim 2$  frequency bins (0.0012 Hz). When gaps were introduced into the data to simulate S6 we could detect a signal with an SNR  $\sim 72$  and a depth of  $\sim 10 \text{ Hz}^{-1/2}$ , with an RMS of  $\sim 10$  bins (0.0056 Hz). The drop in sensitivity here is simply because there is  $\sim 50\%$  less data compared to the previous case. Finally, in the S6 MDC we could detect a signal with an SNR  $\sim 74$  and a depth of  $\sim 13 \text{ Hz}^{-1/2}$ . These real data contain non-Gaussian artefacts such as instrumental lines and this causes a further drop in sensitivity. Whilst not a full comparison to other searches in the S6 MDC [53], as we only tested on a subset of the bands, this search has a sensitivity which is comparable to some other CW searches, however offers a massive increase in speed.

We chose the specific frequency band to search over as the data which we used, i.e., the summed data, becomes less effective at frequencies much higher than 200 Hz, and using the parameters of our simulations, signals can spread over many frequency bins in a day, reducing sensitivity further, however this can be mitigated by using shorter SFTs or performing their summation over 12 (rather than 24) hours.

The methods described in this paper present a basic approach for gravitational-wave signal searches using SOAP. However there are several further developments that could increase its sensitivity. Some of these are outlined below:

One of the main features which reduces the sensitivity of the search is non-gaussianities within the data, namely instrumental lines. Although we have a statistic which penalises these features, in some cases it will also penalise a strong signal. For example, when the amplitude of the noise floor is high for one detector or the duty factor is lower, the signal will appear more like an instrumental line to this statistic. We hope to improve the search statistic in the future by searching for consistent amplitudes as opposed to consistent SNR, i.e, the statistic will take the amplitude of the noise floor and the duty factor into account.

One variation of this method which has been described in this paper is ‘memory’, which is where the tracks jump in frequency is determined by the previous  $n$  jumps. This has yet to be fully tested, however, we expect that this will increase our sensitivity to signals where have a better idea of their frequency evolution. This however, comes at a cost in computational time which we can estimate given Eq. 3.47 in Sec. 3.11.5.

Further additions to the search include using the Fourier transform of the SFT power along the Viterbi track as a detection statistic. If the Viterbi track follows that from an astrophysical signal, then we should see the effects of the antenna pattern in this Fourier

transform as a peak at half a sidereal day. If the track follows something which is not astrophysical then this should not be seen this peak in this Fourier transform. This only applies to the search directly on the **SFTs** not the summed data, as the antenna pattern variations will have been averaged out in the summing.

As well as searching for astrophysical signals, SOAP can also be used to search for and identify instrumental lines. Here we use single detector data, or multiple channels from a single detector, to identify quasi-monochromatic features on the data for further study.

Whilst this paper presents initial tests on sensitivity, further tests will be needed for a full comparison to other **CW** search methods. This search, however, aims to look for signals which may not follow the standard frequency evolution and is intended to return potentially interesting candidates for a more sensitive followup.

# Chapter 4

## Machine learning for continuous wave searches

Machine learning is a term which was used by Arthur Samuel in 1959. He described it as a "Field of study that gives computers the ability to learn without being explicitly programmed" []. This can be thought of as a subset of artificial intelligence. With the development in computing in recent years along with the easier access to large data-sets, machine learning has become much more accessible. One machine learning technique which is widely used is deep neural networks. They are used extensively in classification problems as well as others. **JOE: want to write a little bit more here**

This chapter aims to give an overview of neural networks, specifically **CNNs** and their application to a **CW** search. The majority of this section is written in a paper which is yet to be published, however, is aimed to be submitted soon. The differences are, Sec. 4.3, 4.4 and 4.5 which describe the operation of neural networks have more detail in this thesis than in the paper draft. Sec. 4.10 is additional material which is currently not included in the paper draft.

### 4.1 Introduction

Gravitational wave detectors such as **LIGO** [39, 10] and **VIRGO** [11, 40] search for a number of different targets. Some targets such as **CBCs** have been observed [7, 5, 4], however, other primary sources such as **CWs** are yet to be observed. **CWs** are well modelled quasi-sinusoidal signals with a duration much longer than observing times of detectors. The source of these signals is thought to be rapidly rotating neutron stars which can emit **GW** if there is some asymmetry around its rotation axis. This can be caused by various mechanisms as described in [71]. These signals have small amplitude, which if detected will be below the noise **PSD** of the detector. Therefore, sensitive search algorithms are needed to find the signals. These algorithms generally fall into three categories: Targeted,

directed, and all-sky searches, listed in order of how much is known a priori about the source from **EM** observations.

In targeted searches the sky position, frequency, and its derivatives are assumed to be well known, in directed searches only the sky position is known and in all-sky searches the sky position and frequency of the source is unknown. The most sensitive of these are targeted searches which use coherent matched filtering [46, 45]. These use template waveforms which are generated using the information already known about the source, then correlated this with the data. Directed and all-sky searches have a much broader parameter space to search, therefore, many templates are needed to sufficiently cover the parameter space. Using the coherent matched filter for broader parameter space searches becomes unfeasible due to the amount of computing time that is needed. This led to the development of semi-coherent searches where the data is divided up into smaller segments which can be analysed separately and then the results can be recombined incoherently using various methods [76, 75]. Semi-coherent searches result in a trade off between sensitivity and computing time.

The analysis here is presented mainly as an addition to an existing semi-coherent search algorithm titled SOAP [82]. This is a fast and largely un-modelled search which finds tracks of high **FFT** power in time-frequency spectrograms. When applied to multiple detectors using a line-aware statistic, SOAP looks for frequency bins which have both a high power and are similar in each detector. This means that at a given frequency at a given time, SOAP will penalise frequencies where the **FFT** power is largely different in each detector. The algorithmic details summarised in Sec. 3.

One effect which limits the sensitivity of SOAP and many other **GW** searches is noise artefacts known as ‘instrumental lines’. These can be anything from long duration fixed frequency or wandering lines to shorter duration fixed frequency transients. There are certain types of instrumental line which the SOAP search can struggle to distinguish from an astrophysical signal even with the development of a ‘line aware’ statistic in [82]. Currently the method used to reduce the effect of these lines is to manually look at the SOAP output and the spectrograms for each sub-band to determine whether the sub-band is contaminated by instrumental effects. This process is slow, requires a lot of human input and is subject to human error. When the search runs over a larger bandwidth, it will no longer be practical to look through all bands.

We aim to automate how the search deals with instrumental lines by using **CNNs**. These have been used extensively in image classification and we explain this in more detail in Sec. ???. **CNNs** have already been shown to detect gravitational wave signals from **CBCs** in [83, 84, 85] and other deep learning techniques have been used in searching for **CW** signals in [86].

In Sec. 4.2 we will summarise the basics of how the SOAP search works. In Sec. 4.3,

[4.4](#) and [4.5](#) we explain how CNNs operate and how they are trained. Sec. [4.6](#) will describe how this is applied to an CW search. This includes the structure of the CNN in Sec. [4.6.1](#) and the entire search from raw data to results in Sec. [4.8](#). Finally in Sec. [4.9](#) we show the results from this search and compare to similar analyses.

## 4.2 SOAP

SOAP [82] is an un-modelled search for long duration signals which is based on the Viterbi algorithm [65]. In its most simple form SOAP analyses a spectrogram to find the time-frequency track which gives the highest sum of FFT power. If a signal is present and sufficiently loud then this is the track which is most likely to come from some signal. In [82] the algorithm has been expanded to search through multiple detectors as well as including a statistic to penalise artefacts in the data from the instrument as opposed to from an astrophysical source.

Fig. [4.1](#) shows an example of the spectrogram data which is searched through and the outputs of SOAP; the three main output components are the frequency track, the Viterbi statistic and the Viterbi map.

**Viterbi track** The Viterbi track is the most probable track through time-frequency data given a choice of statistic.

**Viterbi statistic** The Viterbi statistic is the sum of the individual statistics along the Viterbi track. In the analysis that follows, the ‘line-aware’ Viterbi statistic is used. This is the sum of the log-odds ratios,  $p_{\text{signal}}/(p_{\text{line}} + p_{\text{noise}})$  along the track. This is defined in more detail in [82]

**Viterbi map** The Viterbi map value of the Viterbi statistic for every time-frequency bin in the spectrogram. This represents the most probable track which ends in any time-frequency bin. In the Viterbi maps, each time slice is normalised individually, i.e., each vertical slice has been normalised such that the sum of their exponentiated values is equal to 1. This way each pixel in the image can be interpreted as a value related to the log-probability that there is a signal in the bin at that time.

To determine whether an simulated astrophysical signal has been detected, in [82] we used the Viterbi ‘line aware’ statistic alone described above. The ‘line-aware’ statistic reduced the affect of instrumental lines on the analysis, however the ‘line-aware’ statistic is still contaminated by certain types of line. For example, the statistic is affected by broad wandering lines as they offer high power tracks in both detectors. To reduce the effect of these instrumental lines, we looked through the spectrograms and Viterbi maps

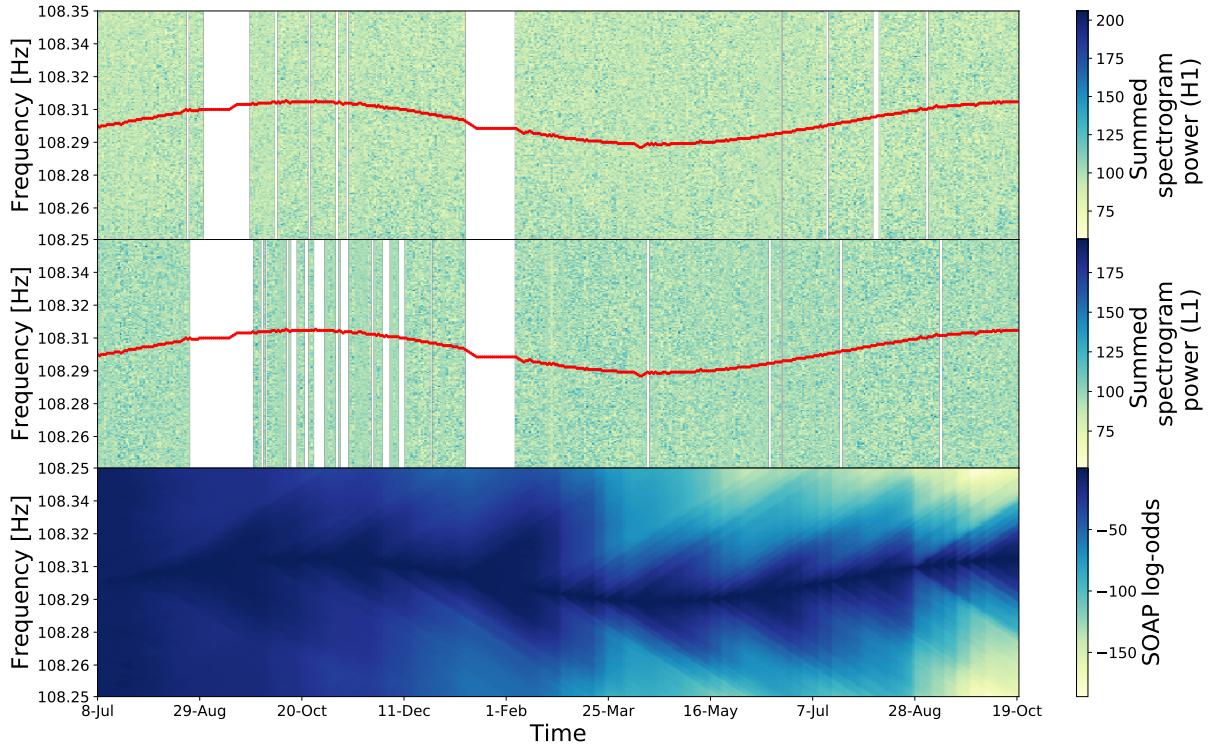


Figure 4.1: This plot shows the inputs and outputs of the SOAP search. The top two panels are time-frequency spectrograms which have been pre-processed as described in Sec. 4.8. This data is a 0.1 Hz wide frequency band from the S6 observing run [ ] **JOE: reference** which includes a string simulated CW signal. The white areas in the spectrograms are gaps in data when the detector was not operating. These both have the optimal track found by SOAP overlaid. The bottom panel shows the normalised Viterbi map, the intensity of a pixel in this image relates to the log-probability that a track ends in a particular frequency bin at a given time.

of individual bands by eye as in Fig. 4.1. Bands which appeared to be contaminated were then removed from the search.

In this search the spectrograms and the Viterbi map contained extra information over the Viterbi statistic. We aim to utilise this in addition to the Viterbi statistic to replace the process of removing contaminated bands ‘by eye’ and therefore automating the search. A useful tool which can be used to classify this extra information is convolutional neural networks.

### 4.3 Neural networks

Throughout this section I will summarise one machine learning technique which are known as Neural networks. Neural networks, as the name may suggest, was developed as a way for a computer to mimic the neurons in the brain. To understand why this would be useful, a common example used is the ability for an algorithm to identify hand written

digits. This seems like a simple task as a brain can complete with ease. However, writing a traditional algorithm to perform this same task is very difficult. The algorithm would have to identify a particular shape which has a huge amount of variation. Neural networks offer a way to deal with this problem as they can be trained on large datasets. This is similar to how a human brain is ‘trained’. In a lifetime of a brain many examples of different symbols are seen and each time a new one is seen the brain ‘updates’ itself based on what is observed. This process is essentially replicated for a neural network, where the algorithm can be updated such that it can correctly identify each digit.

This process can be replicated in a neural network, where it has many parameters which can be modified or ‘trained’. These parameters and their application are grouped into objects called neurons, many of these neurons are used to build a neural network.

### 4.3.1 Neurons

Neurons are the building blocks of any neural network. They perform simple operations on any number of input values and then output a single value. The output  $o$  of a neuron is defined by the equation,

$$o = f \left( b + \sum_{i=1}^N w_i x_i \right), \quad (4.1)$$

where  $b$  is the bias,  $x$  is the input,  $w$  is the weights,  $f$  is the activation function,  $o$  is the output and  $N$  is the number of inputs. Here the input  $x$  represents either the data which is input, i.e the pixels of the image which contains the digit in the example above, or the output of another neuron. The weights  $w$  then represents how important each of those data points are to this problem, or specifically this neuron. The bias  $b$  is then just an extra factor which can shift the data by a fixed value. The activation function  $f$  is then a function which can have many forms, in the simplest case in a neuron known as a ‘perceptron’, it provides a cut where any value above a given threshold is 1 and any below is 0, this will be explained in more detail in Sec. 4.3.3. However, there are many different types of activation functions which can be applied to different situations. This will be explained in more detail in Sec. 4.3.3.

In the example in Fig. 4.2 I have shows a neuron which has 4 input variables, or 4 input data points. When a network is trained, or when it learns, the weights applied to each of the inputs and the bias are updated to better represent the input data. This training procedure is explained in more detail in Sec. ?? Many neurons are then used in combination with each other to develop a neural network which can be applied to more complex problems.

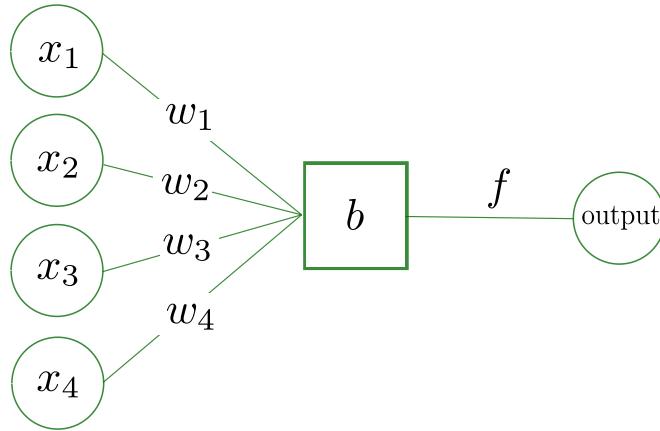


Figure 4.2: Basic neuron showing the four input parameters  $x_n$  and their corresponding weights  $w_n$ . These are multiplied and summed as in Eq. 4.1. A bias  $b$  is added and this value is passed through and activation function  $f$  to the output.

### 4.3.2 Network structure

The structure of a neural network is defined by the user and there is no set way to design a network. However, the general layout of a neural network is defined by structures called layers, sometimes known as fully connected layers. These are rows of  $N$  neurons which all take the same input such that there is  $N$  output values. An example of a simple neural network is shown in Fig. 4.3. The first layer is the input layer, this is just the data points from an input example. In the example of hand drawn digits, this would be the pixels from the image of the digit. The final layer represents the information that you intend the network to extract from the input data. In the hand drawn digit example, this could have 10 output values corresponding to each digit 0-9. Each of these outputs is then a value which is related to the probability of that digit being present in the image.

When designing a network, the user will have a defined input layer size from the data and a set number for the output layer which represents, for a classification example, the number of output classes. The number of hidden layers and the number of neurons in those hidden layers can be arbitrarily changed. In general if the data contains more complex information the size or complexity of the network will need to be increased for it to be able to extract the information. If there is a small number of training examples and a large and complex network, it may be able to learn the input data set as opposed to the general information that they represent.

### 4.3.3 Activation functions

The activation function is how the output of a neuron is transformed. The most simple activation function is a cut as described in Sec. 4.3.1, however, this type of activation

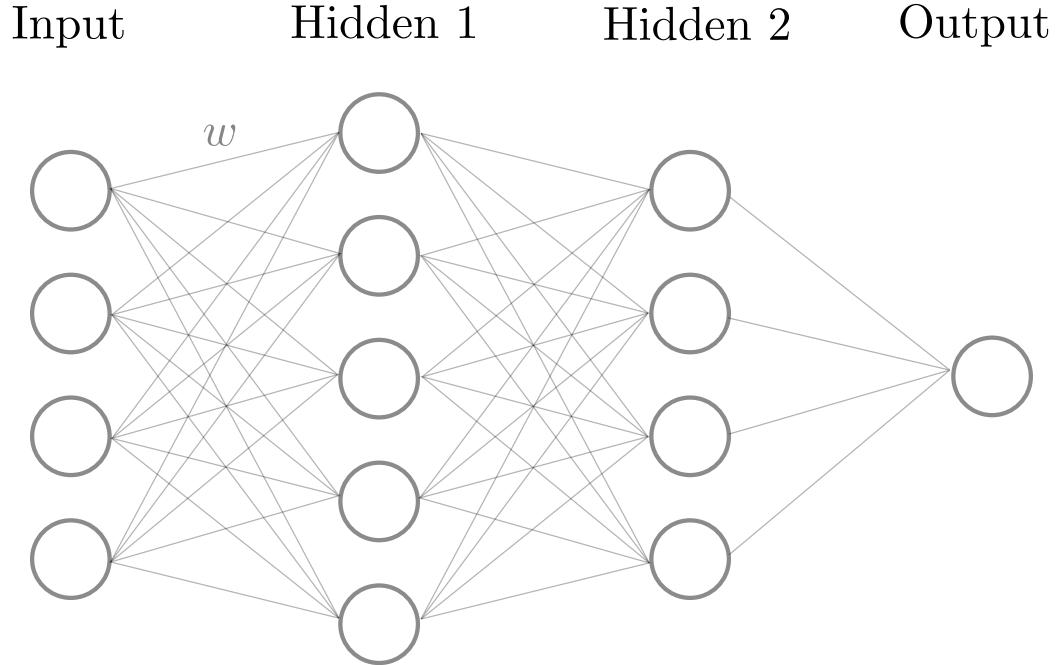


Figure 4.3: A neural network is structured with layers. Each of the circles in these layers are neurons as described in Sec. 4.3.1 and Fig. 4.2. The networks contain an input layer which is usually the data which you would like to analyse. Then this passes to a number of ‘hidden’ layers, in the above diagram there are two. Hidden layers are just layers which exist between the input and the output. The output layer is then the desired output, above I have chosen a single neuron as output. This is such that the network could classify the input to a value between 0 and 1. Every neuron in a layer is connected to the output of all neurons in the previous layer.

does not perform well. Activation functions are generally based on a few properties. The activation function is generally non-linear, this allows networks with multiple layers to be used to approximate a function. A linear activation function means that any number of layers in a network is equivalent to a single layer network. Another property which is desired in activation function is that it is continuously differentiable. This is to allow algorithms such as gradient descent to optimise the network. The functions are found to perform better if they are monotonic and smooth. There are many choices when defining this in the network, some of the available options are shown in Fig. 4.4. One of the more commonly used activation function is the LeakyRELU function, this is explained in more detail in [87]. In the work that follows we use the Leaky RELU function and the sigmoid function.

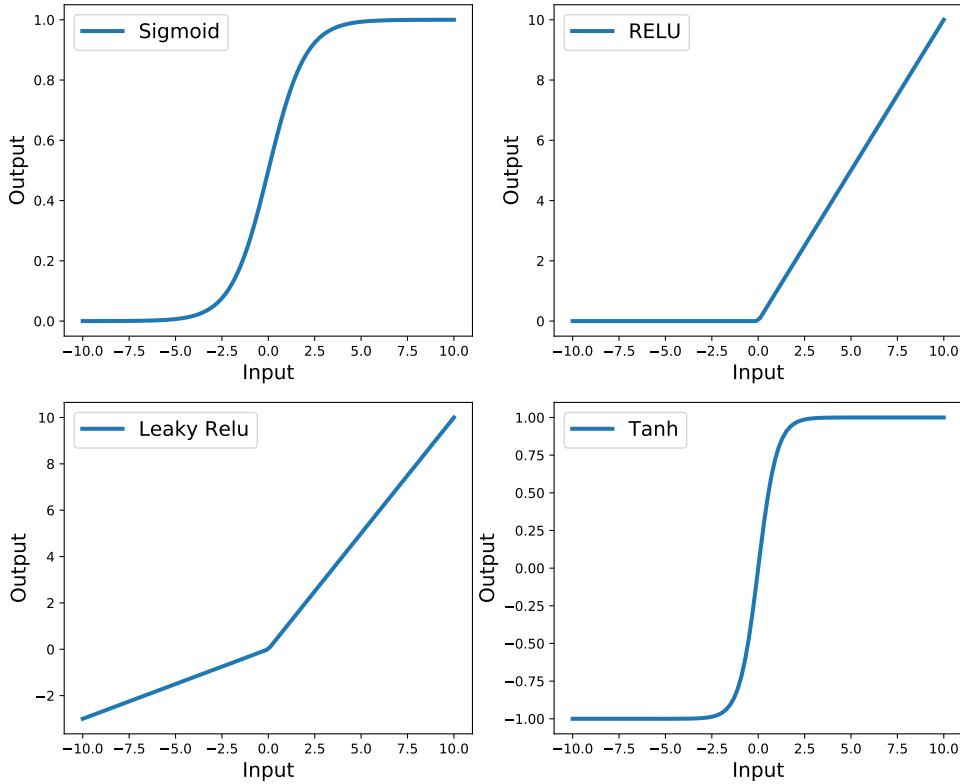


Figure 4.4: There are many different activation functions which are used, and essentially any function can be defined if necessary. Above is shown a subset of the more commonly used functions. The linear function is not used, however, is there to compare to common non linear function.

## 4.4 Convolutional Neural Networks

CNNs are a different type of deep neural network than in Sec. 4.3. They are a types of network which are primarily used in image processing and recognition [88, 89, 90, 91]. Here the general idea is summarised. A CNN is designed to take in data, identify different features within that data and classify what those features or combinations of those features mean. In the context of this work the input data is a time-frequency spectrogram which may contain a simulated CW signal. The output is then a single number which represents if a signal is present. Here a value of 1 represents a signal and 0 is not a signal. A CNN can learn how to identify features by being trained on many examples of the input data where the output is known. For example, an input spectrogram with a simulated CW signal would have an output value of 1. Given the set of training examples, the many parameters of the CNN can be updated such that it gives the best result for any new image. This process is the same as neural networks in Sec. 4.3 and will be described in greater detail in Sec. 4.5

The key features of CNNs which distinguish them from ordinary neural networks is some additional types of layers including: Convolutional layers and max pooling layers.

### 4.4.1 Convolutional layers

Convolutional layers have some similarities to standard fully connected layers as described in Sec. 4.3.2. The main difference being how the weights are applied to the inputs. A fully connected neural network would flatten this image and apply Eq. 4.1 to the inputs. This involves having a separate weight for each of the input pixels in an image. A convolutional layer however, filters the image and outputs a filtered image of the same size (the image can be a different size it depends how the layer was set up). This convolution is defined by,

$$O_{i,j} = f \left( \sum_m \sum_n F_{m,n} x_{i-m, j-n} \right), \quad (4.2)$$

where  $O$  is the output image,  $x$  is the input image,  $F$  is the convolutional filter and  $f$  is the activation function. The weights of the filter  $F_{m,n}$  are what are updated when the network is trained. Fig. 4.5 shows an example of a 6x6 image and the results of filtering the image using Eq. 4.2 with two different filters  $F$ . In this case the network has 4 parameters for each filtered image which can be updated as opposed to the 36 which a full connected network would have for a single neuron.

Fig. 4.5 demonstrates how a filter which matches a feature in an image can highlight that particular feature. i.e. the diagonal line in the bottom left of the input is enhanced by Filter 1, which matches that feature. When this type of layer is trained, the weights of the filter are updated. The filter should then ideally match the feature which is intended to be extracted.

A convolutional layer has a number of different hyper-parameters which can be varied when setting up a model. Below I list each of the adaptable parameters and what they do.

**Filter size** The filter size is the size and shape of the convolutional filter. In Fig. 4.5 we use a filter size of  $2 \times 2$ . The filter does not have to be square, however must be less than the dimensions of the image.

**Number of filters** The number of filters can be any value. If you have  $K$  filter kernels, then the convolutional layer will output  $K$  filtered images. In Fig. 4.5 we use two filters and therefore, the output of the layer is two images.

**Activation function** The activation function is generally kept the same for each of the layers, however this can be set here. The different types have been explained in Sec. 4.3.3 and are applied as in Eq. 4.2.

**Stride** A normal convolutional layer applies a filter by multiplying by a filter, then shifting over by one pixel and repeating. Applying a stride mean rather than shifting by one pixel, one shifts by a number greater than one. This reduces the size of the output

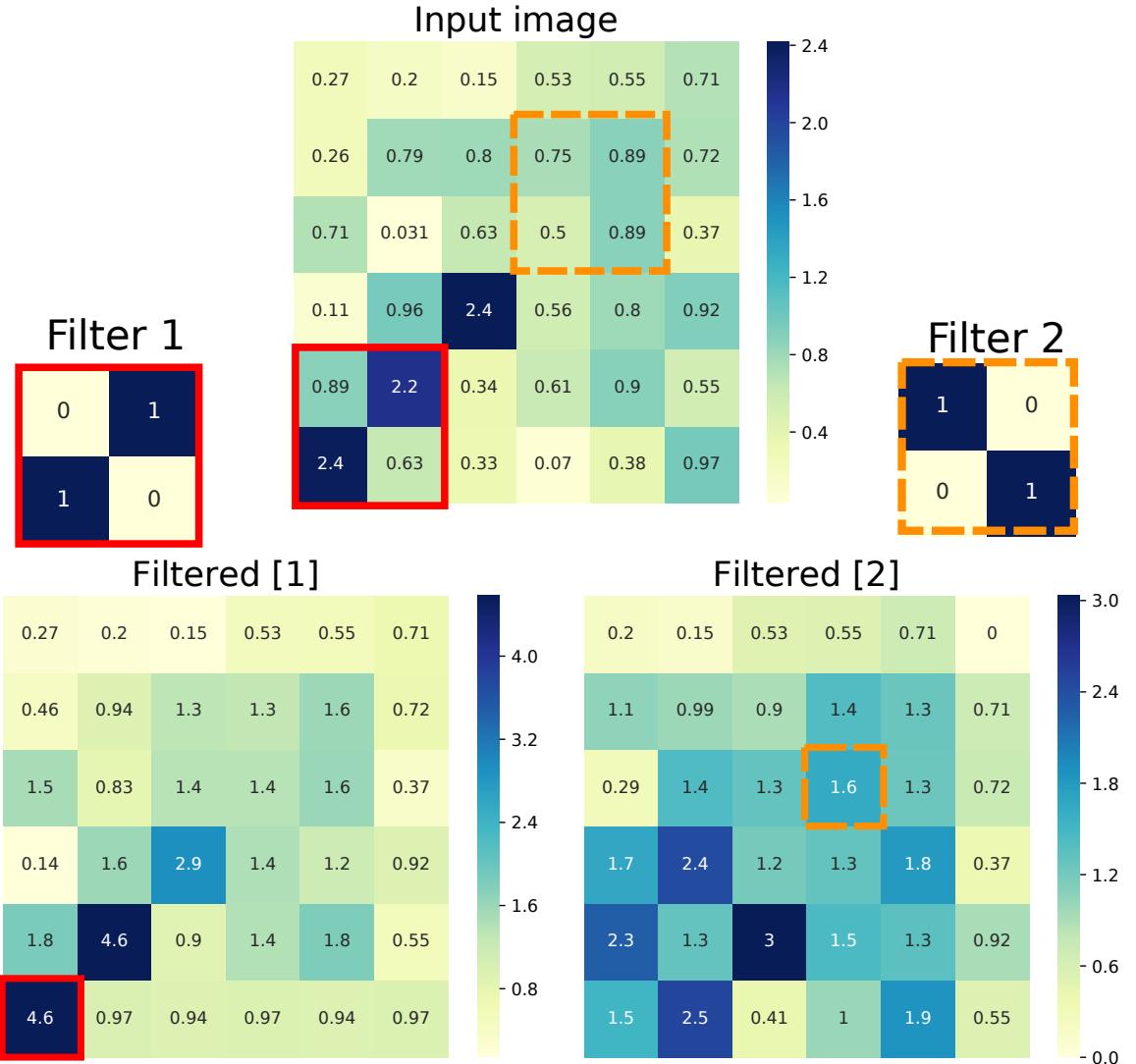


Figure 4.5: Convolutional filters can be designed to ‘pick out’ certain features within an image. In this simple example above, the first filter (filter 1) matches the diagonal line in the bottom left of the input better than filter 2. The output filtered image exaggerates this filter. The coefficients of the filter i.e.  $F_{m,n}$  in Eq. 4.2, in this are set to ones and zeros. These are the weights which are trained by the network. In this case, to get the same size image in the output as the input, the image is padded with zeros. In this case, it was necessary to pad above and to the right of the image. The output of a convolutional layer is then the filtered images above after a bias and activation function have been applied.

by the same factor of stride. i.e. if you skip one pixel (a stride of 2) then the image will be half the size on output. This has a similar affect to max-pooling which we describe in Sec. 4.4.2 and use for the rest of this work.

The convolutional layers will reduce the number of updatable parameters used in each network or model. However, the output of a convolutional layer is a number of images which are potentially the same size as the input. This has potentially increased the size of the parameter space for the next layer. To decrease this a type of layer known as max-pooling is used.

#### 4.4.2 Max pooling layers

Max pooling layers are designed to reduce the size of the problem whilst holding onto as much important information as possible. These do not contain any trainable parameters. The idea of this layer is relatively simple, it reduces the image size by taking the maximum value in a region of a given size. Fig. 4.6 shows the output of the first filtered image in Fig. 4.5. The image is then reduced by a  $2 \times 2$  max pooling layer. The output of max-pooling then shows a large value in the bottom left, this is where the input image matched the filter in Fig. 4.5. This demonstrates how the max-pooling layer can hold on to important information whilst reducing the image size.

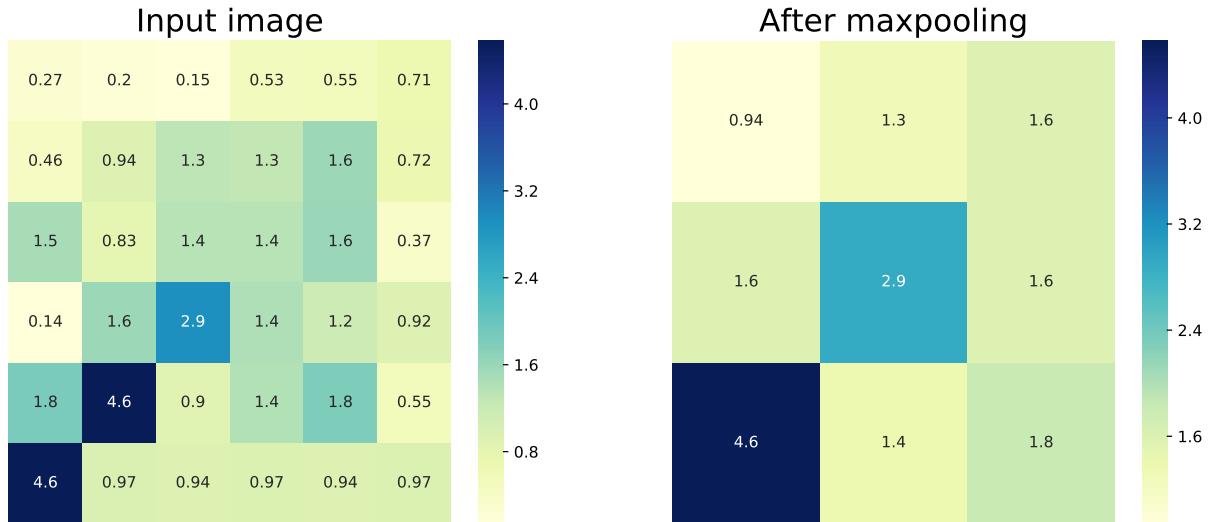


Figure 4.6: Max pooling layers aim to reduce the size of an image whilst retaining important information within the original image. Above shows an example where a  $2 \times 2$  max-pooling layer is used on the output of Filter 1 in Fig. 4.5. This retains the information that the input image matches the filter in the bottom left.

### 4.4.3 CNN structure

CNNs are usually structured such that they can extract larger features from an input image, then the outputs from this are passed on to be classified. The ‘feature extraction’ part of the network consists of the convolutional layers and the max-pooling described in Sec. 4.4. The outputs of the final max-pooling layer are then flattened and used as the input to a fully connected network. This fully connected network classifies these outputs into a number of classes. Fig. 4.7 shows an example of the layout. Here an input image which is the same as in previous examples is passed onto a single convolutional layer with two different filters. The output of two filtered images is passed to a max-pooling layer. The two max-pooled images are flattened into 18 input neurons, this then passes through a fully connected network to a single output neuron. This shows a simple example, however, there are many hyper-parameters of the network which can be changed. These include: the number of filters in a convolutional layer, the number of convolutional layers and max-pooling layers, the number of hidden layers in the fully connected section and the number of neurons in the hidden layers. This example also shows the network being classified to a single output as this is how we use CNNs for the following work.

## 4.5 Training

Once the structure of the network is decided, the network needs to be trained. This means that the weights and bias’ for every neuron and filter need to be updated such that the neural network gives a useful output. For this work we will classify input time-frequency spectrograms into a signal or noise class using a single output neuron. This neuron outputs a value in the range  $[0,1]$  by using a sigmoid activation function. In our case the CNN is trained using a process called supervised learning. In supervised learning, the class of each input example is known. For example, we assign a label of 1 when the input is a time-frequency spectrogram which includes a simulated CW signal. Similarly a time-frequency spectrogram with no simulated signal is assigned a label of 0. In general when training neural networks this way, the performance of the network can be improved by increasing the number of input examples which are shown to the network. This stops the network from over-fitting to specific examples. Instead it should generalise to the full input and learn the underlying features within the data.

### 4.5.1 Loss function

Initially each of the training examples is propagated through the network to its single output value which lies between 0 and 1. Using a loss function, this output is then compared to the label of the input data which is either 0 or 1. There are many types of

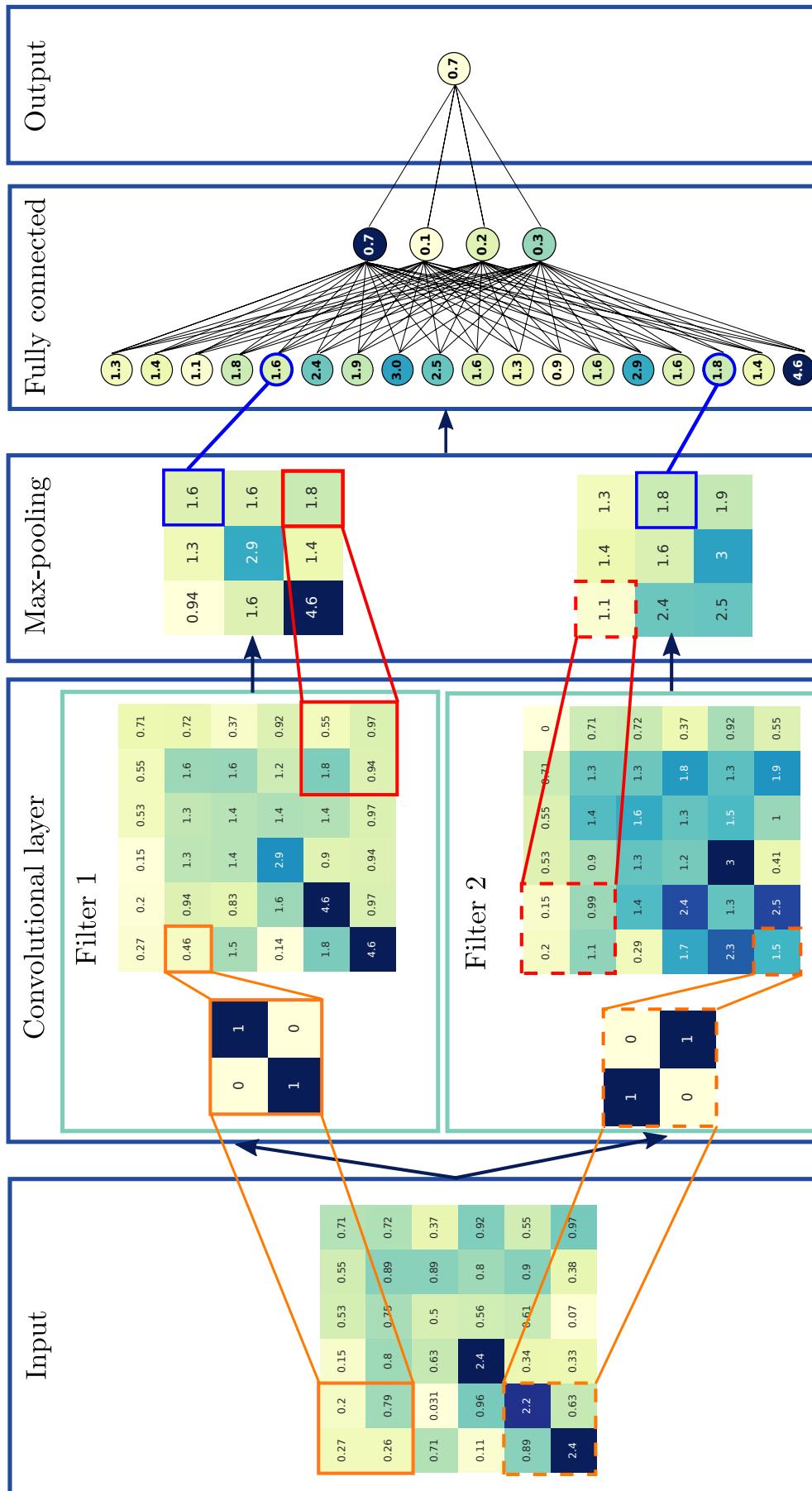


Figure 4.7: Convolutional neural networks consist of two broad sections, the ‘feature extraction’ part which is the convolutional and max-pooling layers, and the classification part which is the fully connected part of the network. This diagram shows a simple example of an image passing through a single convolutional layer with two filters, a single max-pooling layer and a simple fully connected network with a single hidden layer consisting of 4 neurons. The values in this diagram omit the use of any activation function such that the values are easier to follow. In a real network the activation functions are important.

loss function which can be used, this depends on the type of problem which one wants to solve. As we are classifying between two classes in our networks, the loss function,  $L$ , is the binary crossentropy defined as,

$$L = -\frac{1}{N} \sum_i^N y_i \log(p_i) + (1 - y_i) \log(1 - p_i), \quad (4.3)$$

where  $p$  is the networks predicted output which has any value in the range  $[0, 1]$  and  $y$  is the true output which has binary labels 0 or 1. This is calculated as the sum over all training examples. The loss function is minimised when the output matches the ‘truth’. This essentially tells the neural network how close to the truth it is. The weights and bias’ of the neural network can be updated based on the value of this loss function. The process of updating the weights and other parameters is called back-propagation, and typically uses a form of gradient descent [92]. Back-propagation uses the derivative of the loss function with respect to a weight to update that weight. If changing that weight in a particular direction decreases the loss function, then the weight will be updated in that direction. The size of the change of the weight value is related to the change in the size loss function. This means that the weights can be updated to minimise the loss function and therefore improve the performance of the network.

#### 4.5.2 Training procedure

The training procedure entails passing a set of training examples through the network a number of times. Once the entire training data set has been passed through the network (forward pass) and the weights have been updated accordingly (back propagation), the training has completed one epoch. If the data was passed and the weights were updated a single time, the loss may decrease by is likely not at a minimum. Passing the data through again may move the weights to a lower loss. This process is repeated a number of times to try and find the minimum loss. When training there the value of the loss at each epoch is monitored, the trend of the loss of the training set should always decrease. In general subset of the training data is set aside and not used in the training procedure, this is known as validation data. After each epoch the value of the loss for this validation set can be measured, i.e. all the validation data is passed though the network. This can be used to monitor the training of the network. If the validation loss begins to increase then this is a sign that the network is over-fitting to the training data-set.

## 4.6 Application to CW search

The aim for this work is to use a **CNN** to classify **LIGO** data into one of two classes: signal or noise. Here the signal class refers to a **CW** signal from an isolated neutron star as described in Sec. 2.1. Noise then refers to anything else which appear in the data, from Gaussian noise to instrumental artefacts. In Sec. 3.10 to reduce the effect of instrumental artefacts, each of the search sub-bands was analysed by eye to determine if a sub-band was contaminated. Sub-bands which contained an artefact were then removed from the search. This is a time consuming process. The main goal of the **CNN** approach is to automate this part of the search. This section will describe how we design the network to extract features and distinguish signals from instrumental artefacts. We will then present results from searches in a range of **LIGO** observing runs which include: S6, O1 and O2.

### 4.6.1 Network structure

In this section the structure of the networks which are used in this analysis are described. There are three main inputs of data for each **CNN**: spectrograms, Viterbi maps and the Viterbi statistic. Each of these are different representations of the raw detector data. In this analysis we train a separate **CNN** for each of these inputs and then a further three which use these combinations of inputs: Viterbi map + spectrogram, Viterbi map + Viterbi statistic and Viterbi map + Viterbi statistic + spectrogram. In all of the layers excluding the output layer of each **CNN**, the activation functions in Eq. 4.2 and 4.1 are defined by a function titled ‘leakyRELU’ [87]. For our output neuron a sigmoid function is used as an activation function such that the output is limited between 0 or 1. For a given input a **CNN** can then output a value between 0 and 1. When the output value is closer to 1, the input is more likely to contain a signal. The structure of the network is shown in Fig. 4.8 and is explained below.

**Viterbi statistic** This is the simplest of the networks and will give the exact same result as the Viterbi statistic on its own. This is a single neuron which takes in the Viterbi statistic applies a weight and bias and then passes through a sigmoid function.

**Viterbi map** The Viterbi map **CNN** takes in a down-sampled Viterbi map of size (156,89), this is described more in Sec. 4.7.3. This **CNN** consists of two convolutional layers and 3 fully connected layers. The first layer has 8 filters which have a size of  $5 \times 5$  pixels, the second layer has 8 filters with a size of  $3 \times 3$  pixels. After each of these layers we use a max-pooling layer with a size of  $8 \times 8$  pixels. This then passed into three fully connected layers which all have 8 neurons and used leakyRELU activation functions. Finally these lead to an output neuron which uses a sigmoid function.

**Spectrogram** The spectrogram CNN takes in a down-sampled spectrograms of size (156,89), this is described more in Sec. 4.7.3. This CNN has an identical structure as the Viterbi map CNN, however, takes two channels as input. The two channels are the spectrograms of two different detectors.

The next three networks are constructed from combinations of the previous described CNNs.

**Viterbi map and spectrogram** To combine the spectrogram and Viterbi map network, we remove the final output neuron and its 8 weights from each of the networks. The outputs from each network is then 8 neurons. These can be combined to a single sigmoid neuron which has 16 new weights.

**Viterbi map and Viterbi statistic** In this network we combine the Viterbi statistic with the Viterbi map. As before, this uses the pre-trained Viterbi map and Viterbi statistic CNNs. The output sigmoid neuron and corresponding weights are removed from each network. The 8 neurons from the Viterbi map network and the single neuron from the Viterbi statistic network are then combined to a single neuron with 9 new weights.

**Viterbi map, Viterbi statistic and spectrogram** This combination takes all component CNNs from above. As before the final sigmoid output and the corresponding weights from each network are removed. The 8 neurons from the Viterbi map and spectrograms CNNs and the single neuron from the Viterbi statistic are then joined into a single output neuron with 17 new weights.

When combining CNNs we use a process called transfer learning [93]. This uses the pre-trained weights of the networks as a starting point to continue training. In our examples we found that we could fix the weights inside the pre-trained networks and just train the final 16 output weights from the neurons as in Fig. 4.8. These combinations of networks were chosen as the different representations of the data should contain slightly different information on the input. For example, the Viterbi statistic contains no information on the structure of the track in the data and the Viterbi maps lost some information about lines in the band. The addition of the spectrograms aimed to include even more information about this piece of data. Where when each of these are combined, the CNN should be able to pick to important information from each of these representations.

## 4.7 Data generation

To train the CNNs we need to generate many examples of data, this include the three data products above: Time-frequency spectrograms, Viterbi maps and the Viterbi statistic.

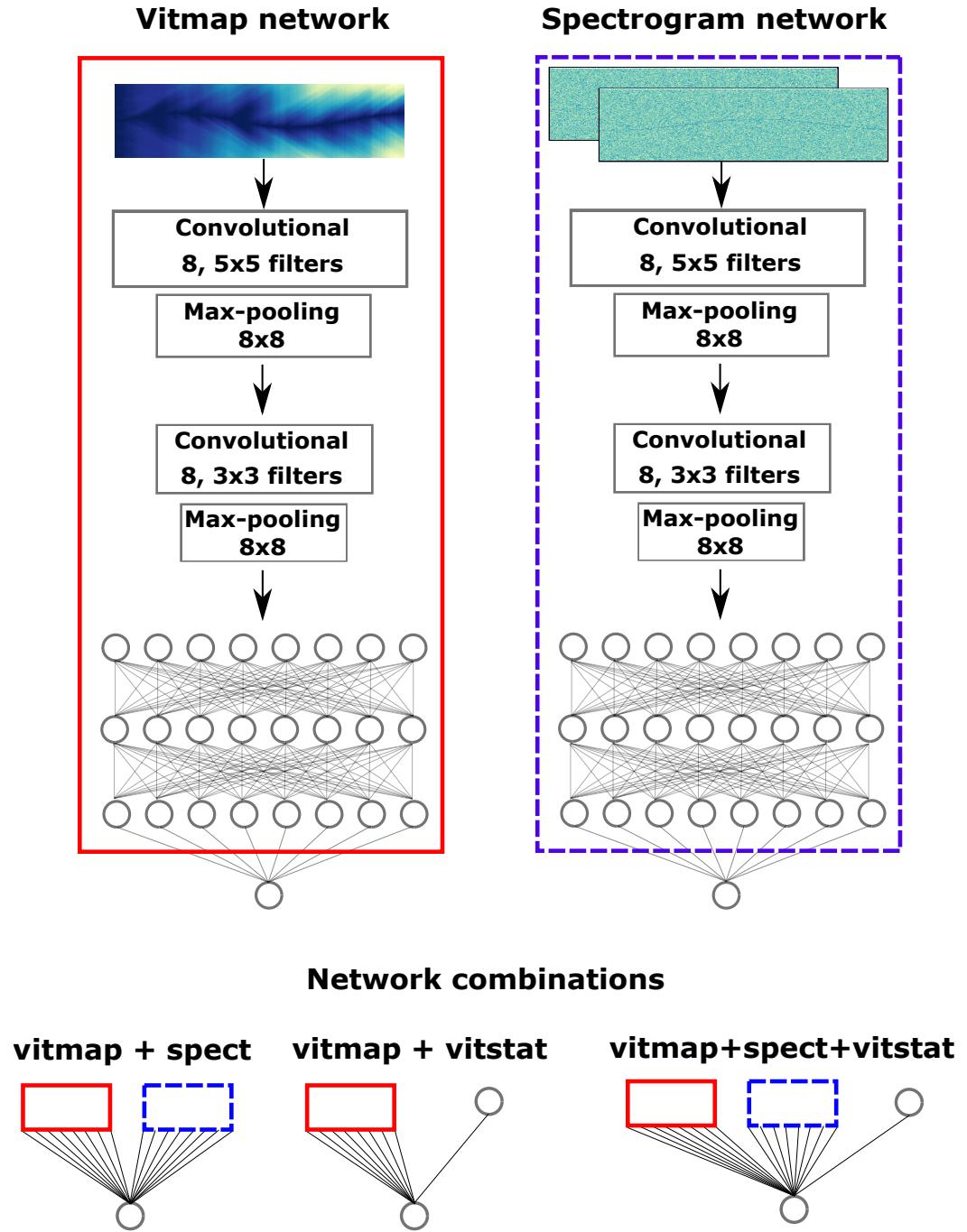


Figure 4.8: The structure of the Viterbi map and spectrogram CNNs used in this analysis are the same, with the difference that the spectrogram takes two images as input. They each use two convolutional layers and 3 fully connected layers before they're output to a single neuron which represents the probability of belonging to the signal class. The Viterbi statistic network is a single neuron that transforms the statistic into a number between 0 and 1 representing the probability of belonging to the signal class. For the combinations of networks, we remove the final output neuron and its 8 weights, i.e. we take the part inside the red or blue box. The 8 outputs from each network are then combined to a single neuron with 16 new weights.

When a **CNN** is trained it needs to see examples of all possible features which could appear in the data. This include, Gaussian noise, non-Gaussian artefacts and **CW** signals. As non-Gaussian artefacts are difficult to simulate, it is possible to use the non-Gaussian artefacts in real data as part of the training set. Therefore, for the majority of the analysis that follows, the time-frequency spectrograms which are used to generate the Viterbi data are from real detector data. The exact observing runs used will be explained in Sec. ??.

For the analysis that follows there are three main sets of data: training data, test data and search data. Training data uses a set of augmented (see Sec. ??) time-frequency spectrograms containing simulated signals and is used to train each of the networks. Test data is a separate set of simulations in time-frequency spectrograms which are not augmented. These are used to generate efficiency curves and test the network. Search data does not contain any simulated signal injections and is used to search for real signals within the data.

When training and testing a network it is important that the networks are not trained and tested on the same data. Otherwise the **CNNs** can learn specific features of the training data and not the underlying distribution of features. To avoid this, the spectrograms are split into 0.1 Hz wide sub-bands where alternating bands are designated as ‘odd’ or ‘even’. This means that bands starting with 100.1,100.3 are odd and 100.2,100.4 are even etc. The networks can then be trained on the odd bands and tested on the even bands and vice versa. This then means that each time we want to search over data, we will have two final networks. One which will be run on odd bands and a separately trained network which is run on even bands.

### 4.7.1 Signal simulations

To inject the simulated signals into real data we generate a random set of signal parameters which are drawn from prior distributions defined in Table ???. The **SNR** of each simulation is then uniformly distributed between 50 and 150. Where the **SNR** is the integrated ‘recovered’ **SNR**. This is calculated for each time segment using the definition of optimal **SNR** in [52], the total **SNR** is then the sum of the squares of these. The **GW** amplitude  $h_0$  is scaled based on the noise **PSD** to achieve this **SNR**. The power spectrum of the signal can then be simulated in each time segment of a time-frequency spectrogram. This is done by assuming that the spectrogram is  $\chi^2$  distributed. The antenna pattern functions are taken into account for the given source parameters and detector such that the **SNR** for each time segment is calculated. This **SNR** is spread over neighboring frequency bins dependent on its location in frequency. The power spectrum values can then be drawn from a non-central  $\chi^2$  distribution with the non centrality parameter equal to the square of the **SNR**. Each signal is simulated in two detectors: **LIGO**s H1 and L1. The **SNRs** reported below are then the sum of the squares of the **SNRs** from each detector.

Table 4.1: Table shows the upper and lower limits over which each signal parameter was randomized. The parameters  $\alpha$ ,  $\sin(\delta)$ ,  $f$ ,  $\log(\dot{f})$ ,  $\cos(\iota)$ ,  $\phi_0$ ,  $\psi$  were sampled uniformly in the ranges specified in the table. The frequencies  $f_l$  and  $f_u$  refer to the lower and upper frequency of the band that each signal is injected into. Excluding the distribution of frequencies  $f$ , all the injections parameters are sampled from the same distributions as the S6 MDC [53].

	$\alpha$ [rad]	$\sin(\delta)$ [rad]	$f$ [Hz]	$\log_{10}(\dot{f}[\text{Hz/s}])$	$\cos \iota$ [rad]	$\phi$ [rad]	$\psi$ [rad]
lower bound	0	-1	$f_l + 0.25$	-9	-1	0	0
upper bound	$2\pi$	1	$f_u - 0.25$	-16	1	$2\pi$	$\pi/2$

### 4.7.2 Augmentation

To train a neural network, many examples of data from each class are needed to avoid over-fitting. In our case when we use data between 40-500 Hz, splitting the data into 0.1 Hz wide sub-bands does not give enough data for the networks to be trained effectively. Therefore, using a technique called data augmentation [94, 95] we can artificially increase the number of training examples. Augmentation is when data is transformed such that, to the network, it appears to be ‘new’ data. For example, by shifting a time-frequency band up and down in frequency, this appears to be a new realisation of noise which we can then inject a simulated signal into. This would double the size of the training data-set and reduce the likelihood of over-fitting to the training data.

The augmentations are applied to the spectrograms from each of the detectors. The augmentations that are used on each sub-band are: reversing the data in time, flipping the data in frequency, rolling the data in time by a small number of segments and shifting the data in frequency by a small number of bins. As we use real data, there are gaps in time where the detectors were not operating. We preserve the location of these gaps when augmenting the data. When shifting the data in frequency, we shift each band up and down by 30 frequency bins (0.016 Hz) and up and down by 60 frequency bins (0.032 Hz). When rolling the data in time, we roll each sub-band by 100 time segments (100 days). Fig. 4.9 shows examples of the original data, a flip in frequency, a roll in time and a flip in time. For each frequency shift, we flip the sub-band in time and frequency and roll the sub-band in time. This then gives us 3 transformations for each of the 4 frequency shifts, which including the original data gives 20 times the number of training examples.

### 4.7.3 Downsampling

One further issue for our data sets are their size. The spectrograms we use have a large number of pixels within them. This means that as the spectrograms are passed through the network, there are a large number of computations. Both this number of computations and

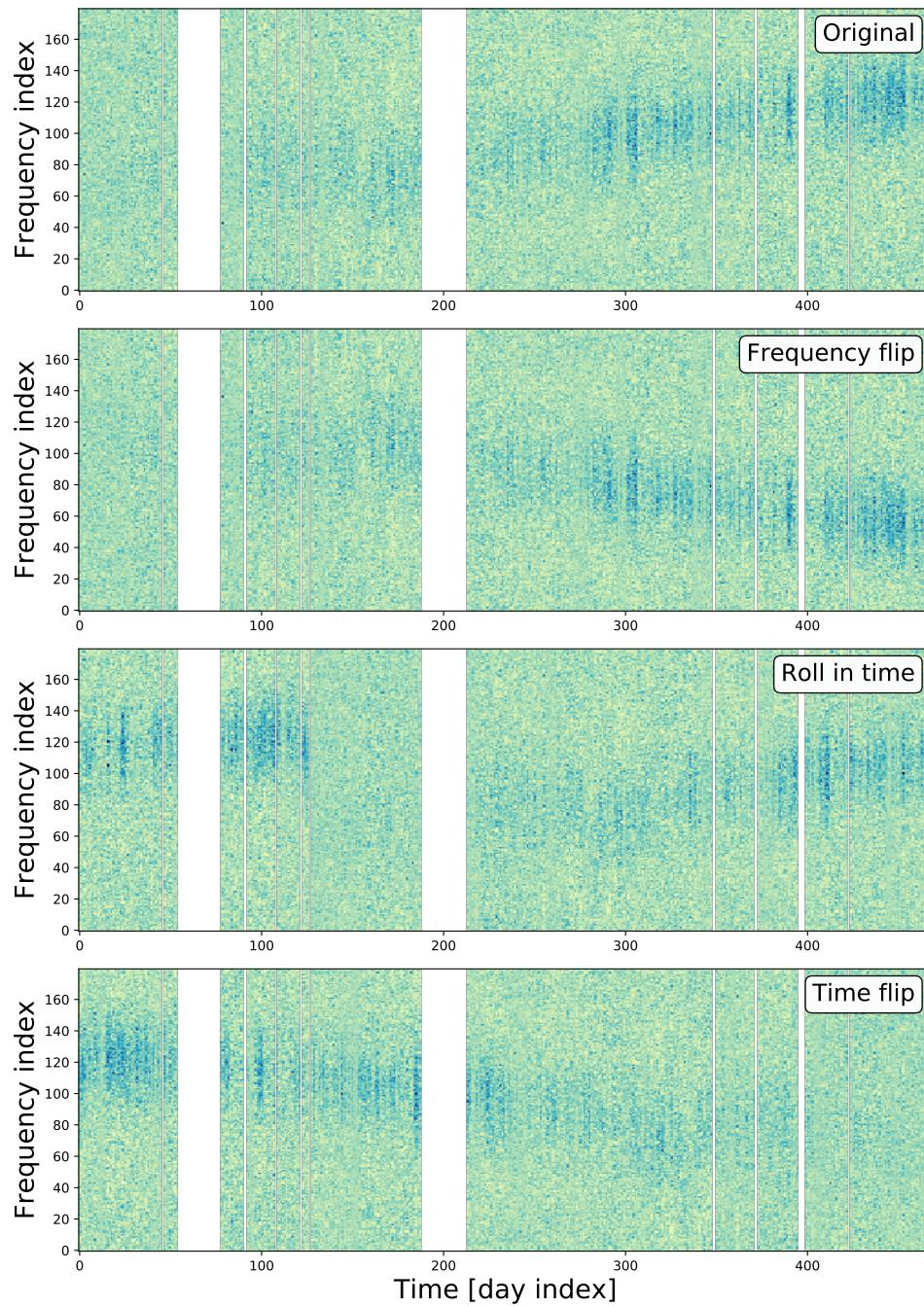


Figure 4.9: The data is transformed by flipping the data in frequency (panel 2), rolling the data in time by 100 bins (panel 3) and flipping the data in time (panel 4). The original summed spectrogram is show in panel 1. Simulated signals can then be injected using this data as noise. The plots above show a broad wandering line to demonstrate the changes to the data when it is augmented, however, the majority of sub-bands contain almost Gaussian noise.

the memory requirements of the GPU mean that training a network with a large number of data points takes longer. We implement a few methods to reduce the size of the data: summing time segments of spectrograms and down-sampling these summed spectrograms.

The spectrograms are summed over one day, i.e., every 48 time segments, as in [82]. This should increase the **SNR** for a given signal within a given time-frequency bin assuming that the signal remains within the frequency bin for the majority of the time segment. To reduce the size of the data further, the package ‘resize’ from scikit-image [96] is used, this uses interpolation to resize the summed spectrograms to a size of (156,89) [time segments,frequency bins]. This size was defined based on the summed spectrograms of the S6 data-set. This is 1/3 the number of summed segments in time, 1/2 the number of segments in frequency. The down-sampling is applied to the spectrograms and vitmap. In [82] we demonstrated that summing spectrograms can increase the speed and sensitivity of our search. When down-sampling the image, we found that reducing the amount of data had a small affect on the sensitivity of the **CNNs** used.

## 4.8 Search pipeline

In previous sections each component of the search pipeline has been described, however, described below is how each component fits together. Figure 4.10 shows a flow diagram of the pipeline. The pipeline is run in three different ways: training the **CNN**, testing the search and running a search on real data.

- 1. SFTs** Generate 1800s long **SFTs** from detector time-series data. **SFTs** of this length are a standard set for **CW** searches which are continuously generated during observing runs by members of the **LIGO** collaboration.
- 2. Normalising** The **SFTs** are then divided by their running median with a window width of 100 frequency bins. If we assume the resulting **SFTs** to be  $\chi^2$  distributed, we can apply a correction factor using LALSuite code `XLALSFTtoRngmed` [78] such that their power spectrum has a mean of  $\sim 1$ . By then multiplying this by 2, the noise like parts of the spectrum are  $\chi^2$  distribution with two degrees of freedom.
- 3. Narrowbanding** The computational efficiency can be improved if the data is split into narrow bands. This is because the analysis can be completed on each band in parallel on separate CPU nodes. In this search the spectrograms are split into 2.1 Hz wide bands every 2 Hz, i.e. 100.0-102.1, 102.0-104.1 etc. The bands are 2.1 Hz wide as the analysis on each node will further split the data into 0.1 Hz wide sub-bands. The overlap then allows the sub-band from 1.95-2.05 to be calculated on a node. This band size was chosen based on the available computational memory at the time.

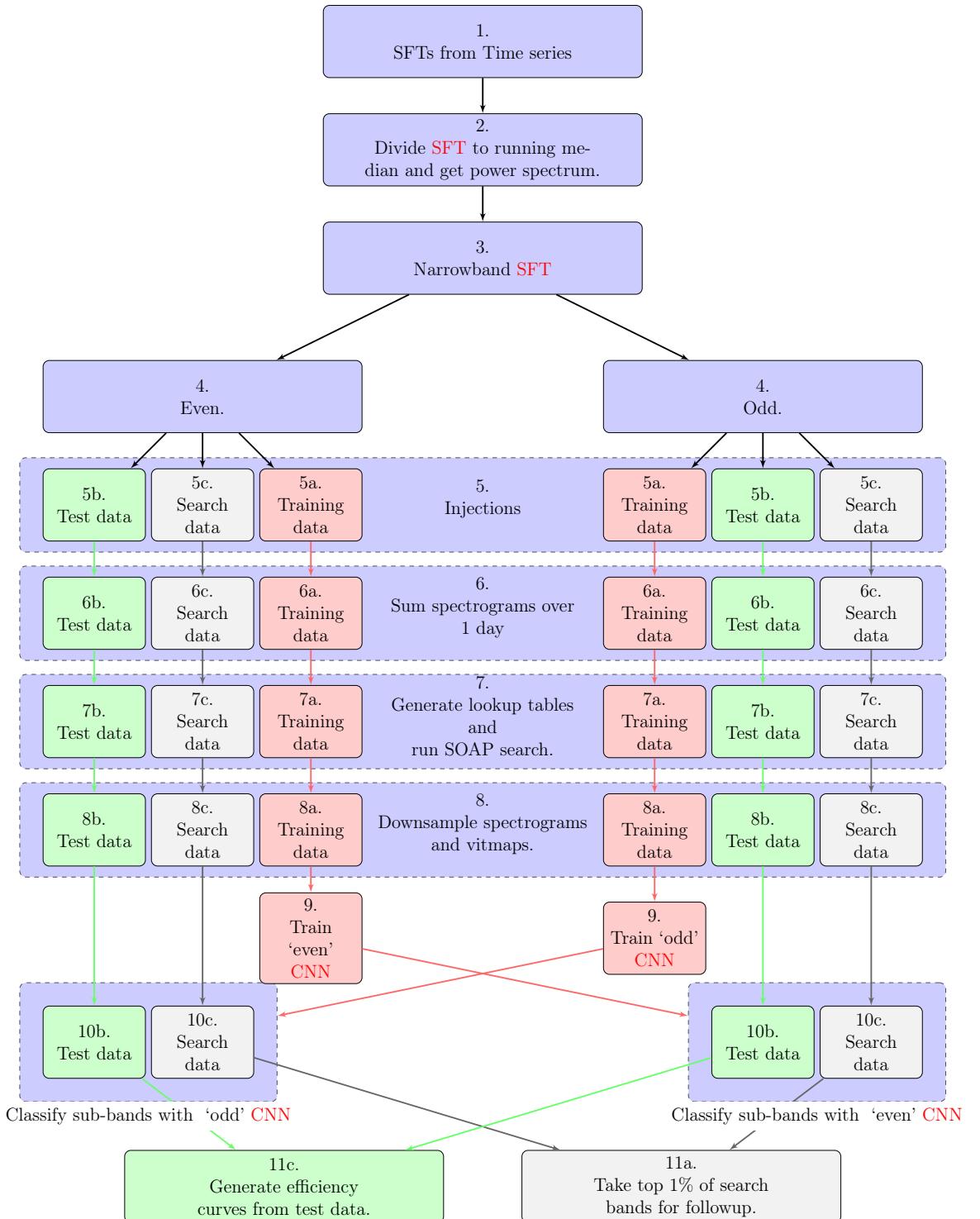


Figure 4.10: This diagram shows the SOAP pipeline from start to finish. There are three main sections: Training (red), Testing (green) and Searching (grey) for both the odd and even bands. The blue sections mean that the same operations is done in all cases.

4. **Band splitting** A CNN should not be trained on the same data that it will be tested on. For this reason, each of the 0.1 Hz wide sub-bands are split into ‘odd’ or ‘even’ bands. A CNN can then be trained on even bands and tested on odd bands and vice versa.
- 5a. **Training data generation** To generate training data the process is the same as described in Sec. ???. Each of the 0.1 Hz sub-bands is ‘augmented’ as in Sec. 4.7.2. For each of the augmented bands, the data is duplicated such that there is a second copy of every augmented band. In the copied set of bands, signals are injected into them with SNRs in the range 50-150. This gives us an example for a noise class and a signal class. There are two of these sets, one for ‘even’ bands and one for ‘odd’.
- 5b. **Test data generation** For test data, signals following the parameters in Tab. 4.1 are injected into 50% of the 0.1 Hz sub-bands. These signals have an SNR in the range 20-200. The SNR range here is wider than the training set as a method to test how the trained networks perform on a wider range of SNRs. Here we again have a set for ‘odd’ and a set for ‘even’.
- 5c. **Search data** This data is generated such that we can search for a real signal. The sub-bands described in part 4 are now overlapping by 0.05 Hz. This means that if there is an astrophysical signal it should be fully contained within at least one sub-band. We do assume that a signal’s frequency does not drift by more than 0.1 Hz, which is assumed to be true for isolated neutron stars < 500 Hz. There are both ‘odd’ and ‘even’ versions of this search data.
6. **Summing spectrogram** As in [82] the spectrograms are summed over one day, i.e., every 48 time segments (1 day) of the spectrogram are summed. This is done separately for each of the 6 data-sets (3 for ‘odd’, 3 for ‘even’).
7. **Generate lookup tables and run SOAP search** Before the SOAP search is run, the line-aware statistic lookup tables need to be generated as in [82]. Then for each of the 6 data-sets (3 for ‘odd’, 3 for ‘even’) the SOAP search is run separately.
8. **Down-sample data** At this stage there are four elements which are saved for each of the 6 data-sets. The two spectrograms, the Viterbi maps and the Viterbi statistic. The spectrograms and the Viterbi maps are down-sampled to a size of  $(156 \times 89)$  using interpolation from scikit-image’s resize [96]. This size was chosen based on the S6 MDC data-set, where this is 1/3 the length in time and 1/2 the width in frequency of the summed spectrograms. This was chosen such that the CNNs trained efficiently and still achieved a reasonable sensitivity.

**9. Train Networks** The down-sampled training data is then used to train a **CNNs**. One **CNN** is trained on ‘odd’ bands and a different **CNN** with the same structure is trained on ‘even’ bands.

**10b. Run search on test data** The trained **CNNs** from part 9 are then used to classify each sub-band in the test data with injections, this returns a statistic on the range  $[0, 1]$ . The closer the value is to 1 the more likely it is from an astrophysical signal, therefore, the statistic can be interpreted as an estimate of the probability of a signal being present. Here the **CNN** trained on the ‘odd’ bands is tested using the ‘even’ bands and vice versa. The algorithms are run on this test data to assess the sensitivity of the analysis.

**10c. Run search on real data** The trained **CNNs** from part 9 are then used to classify each sub-band in the search data, this returns a statistic in  $[0, 1]$ . This statistic is the same as in part 10b. Once again the **CNN** trained on the ‘odd’ bands is tested using the ‘even’ bands and vice versa.

**11a. Signal candidates** The signals which have a statistic in the top 1% can be taken as potential candidates. This can then potentially be followed up with other **CW** search methods.

**11c. Efficiency curves** The output statistics from the test data-set (11b.) can be plotted against **SNR** to see how the network classified signals with the **SNR** of the injection. This can potentially be extended to other signal parameters also. Then the efficiency curves can be generated, this is described in further detail in Sec. 4.9.1.

**CHRIS:** One thing I am still confused about is search, test and training data. Maybe I missed it but can you point me to the place (or add text to explain) that the search data is just the data, the training data is a mix of augmented data and injections, and so is the test data. In all cases the data is real. The injections in the test data are different to the ones in the training data and the augmentations in the test data is different to the training data. **JOE:** this is said at the start of the data section ??

## 4.9 Results

The networks described in Sec. 4.6.1 were trained and tested on four different data-sets: the S6 **MDC** as in [82, 53], our own injections into O2 data, Gaussian noise which had the same gaps and noise floor as the S6 data-set, and our own injections into real S6 data. Each of the searches use training and test data in the frequency range of 100-400 Hz, except the S6 **MDC** which uses data in the range 40-500 Hz for testing and training.

### 4.9.1 Sensitivity

To investigate the sensitivity of the pipeline we use two measures: the sensitivity depth  $\mathcal{D}$  [52] and optimal **SNR**  $\rho$  [79] which are both defined in [82] as,

$$\mathcal{D}(f) = \frac{\sqrt{S_h(f)}}{h_0}, \quad (4.4)$$

where  $S_h(f)$  is the single-sided noise **PSD** and  $h_0$  is the **GW** amplitude. The optimal **SNR** is defined as,

$$\rho^2 = \sum_X 4\Re \int_0^\infty \frac{\tilde{h}^X(f)\tilde{h}^{X*}(f)}{S^X(f)} df, \quad (4.5)$$

where  $X$  indexes the detectors and  $\tilde{h}(f)$  is the Fourier transform of the time series of the signal  $h(t)$ . This expression is defined in [52] for a double-sided **PSD** and we have defined it for the more common single-sided case.

The sensitivity curves shown in Fig. 4.12, 4.13 and 4.14 were generated using a 1% false alarm rate, where the false alarm threshold is the value of our statistic where 1% of sub-bands which do not contain an injection exceed that value. This is then used as a detection threshold. The efficiency is defined as the fraction of events which exceed the false alarm rate for any given **SNR**. As the **SNR** is sampled uniformly between the range 20-200 as in Sec. 4.8, the efficiency cannot be calculated for any individual **SNR**. Instead, one can define some window around a point in **SNR** and count the fraction of statistics which exceed the false alarm threshold within that window. We define the window as a Gaussian with a standard deviation of 2, this is wide enough to contain enough injections per **SNR** to achieve a reliable value. The Gaussian window was chosen to down weights statistics which are further away from the given point in **SNR**. This can then slide along the **SNR** axes and estimate the efficiency as a function of **SNR**. The efficiency curves  $y$  are then be calculated using,

$$y(x) = \frac{\sum_i H(O_i - O^{1\%})\mathcal{G}(x_i - x, 2)}{\sum_i \mathcal{G}(x_i - x, 2)}, \quad (4.6)$$

where  $O_i$  is the output statistic from the **CNN**,  $O^{1\%}$  is the statistic value at a 1% false alarm rate,  $H$  is the Heaviside step function which has a value of 1 for positive values and 0 for negative values.  $x_i$  is the **SNR** of a simulation with output  $O_i$ ,  $x$  is the current location in **SNR** and  $\mathcal{G}(x_i - x, 2)$  is a Gaussian with a mean of the current **SNR** and a standard deviation of 2. The sensitivity curves for each of the described data-sets are shown in Figs. 4.12, 4.11, 4.13 and 4.14.

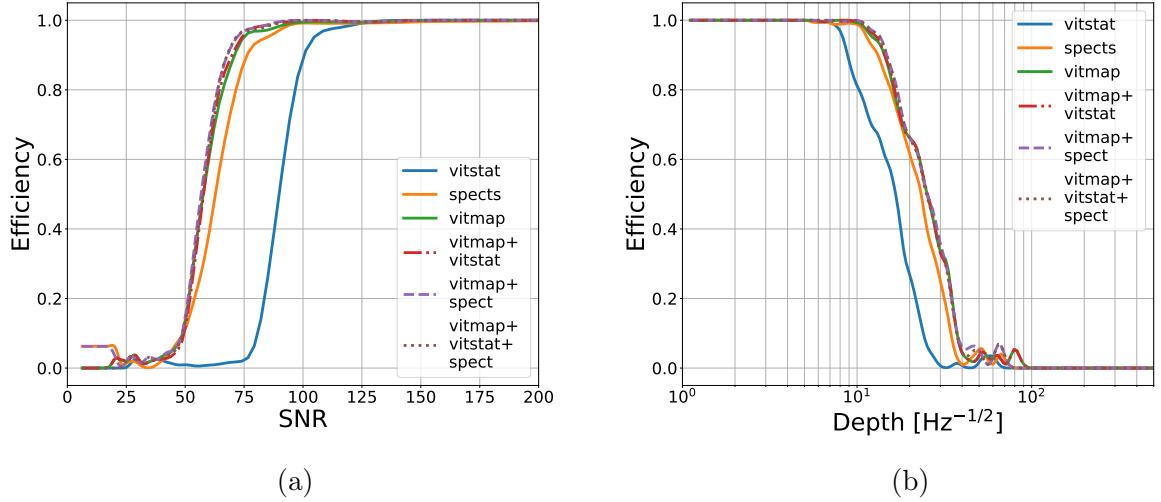


Figure 4.11: In the O1 data-set, each of the six **CNNs** were tested. The efficiency plots above are for a 1% false alarm rate.

## O1

For the first test, injections were made into the O1 data-set as in Sec. 4.7 between 100 Hz and 400 Hz. Then each of the 6 networks described in Sec. ?? were trained and tested on this data. Fig. 4.11 shows the sensitivity curves for this test for both **SNR** and sensitivity depth for each of the 6 networks. Focusing on Fig. 4.11a, the least sensitive, i.e. furthest to the right, of the **CNNs** is the Viterbi statistic (vitstat), this is expected as we know that the Viterbi statistic is sensitive to instrumental lines. The spectrogram **CNN** has an improved sensitivity over the Viterbi statistic, this importantly does not involve the SOAP search but is run entirely on down-sampled and summed spectrograms. Whilst this network is approaching the most sensitive of the examples in Fig. 4.11, and with further efforts may reach it, this network takes  $\sim 10$  times the amount of training time. This will be explained in more detail in Sec. 4.9.2. The remaining networks achieved almost the same sensitivity. The vitmap network however, is the fastest of these to train and is used as an input for all of these remaining networks. For the O1 data-set we show that with a false alarm of 1% the Viterbi map **CNN** achieves a sensitivity of SNR 73 and sensitivity depth of  $12 \text{ Hz}^{-1/2}$  with 95% efficiency. The **SNR** here should not be compared between different runs as this is the integrated ‘Recovered’ **SNR**. Therefore, observing runs, such as O1, which were shorted will appear to have a greater sensitivity when they in fact do not.

## O2

For the next test, injections were made into the O2 data-set in the same way as in the last section. Then each of the 6 networks described in Sec. 4.6.1 were trained and tested on

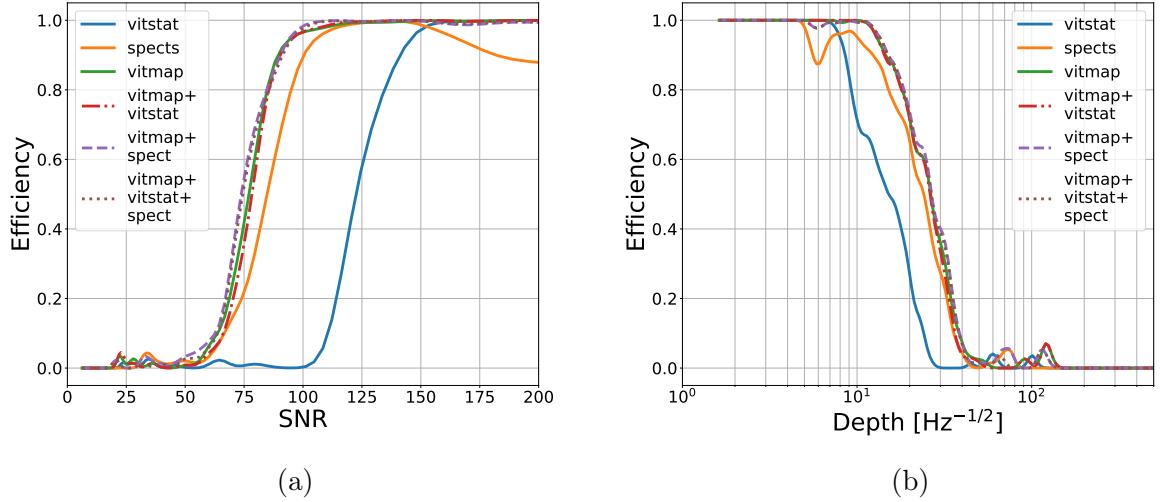


Figure 4.12: In the O2 data-set, each of the six **CNNs** were tested. The efficiency plots above are for a 1% false alarm rate. These show that the use of spectrograms and the Viterbi maps with **CNNs** greatly improve the sensitivity when compared to the Viterbi statistic. For each **CNN** which uses a combination of inputs it achieves a similar sensitivity to the **CNN** which uses just the Viterbi map. This implies that the Viterbi map contains the most information out of the tested **CNNs**.

this data. Fig. 4.12 shows the sensitivity curves for this test for both **SNR** and sensitivity depth for each of the 6 networks. The results here are very similar to the results from the O1 simulations. The Viterbi statistic is the least sensitive ‘network’, followed by the spectrogram **CNN**. The remaining four networks all achieve a similar sensitivity, each of these remaining networks contain the Viterbi map as one of their inputs. Therefore, it is assumed that the dominating effect on the sensitivity originated from the Viterbi maps. In following tests the focus will be with the Viterbi map **CNN** as in all cases this is among the most sensitive. For the O2 data-set we show that with a false alarm of 1% the Viterbi map **CNN** achieves a sensitivity of **SNR** 95 and sensitivity depth of  $12 \text{ Hz}^{-1/2}$  with 95% efficiency. This result in the sensitivity depth in Fig. 4.12b does not vary much from the results from O1, Fig. 4.11b. Whilst one might expect the sensitivity to increase due to the increase in sensitivity and longer observing run. However, this is not the case **JOE: why?**.

### Gaussian noise

The next test involves using injections into Gaussian noise. For this test we tried to replicate the S6 data-set without including instrumental artefacts such as lines. We included the same gaps in data as S6 and the noise floor of S6 was replicated by scaling the **SNR** of any injection in any given **SFT**. Fig. 4.13 shows the sensitivity curves for the Viterbi statistic and Viterbi map **CNN** for both the Gaussian noise run with S6 gaps and for injections

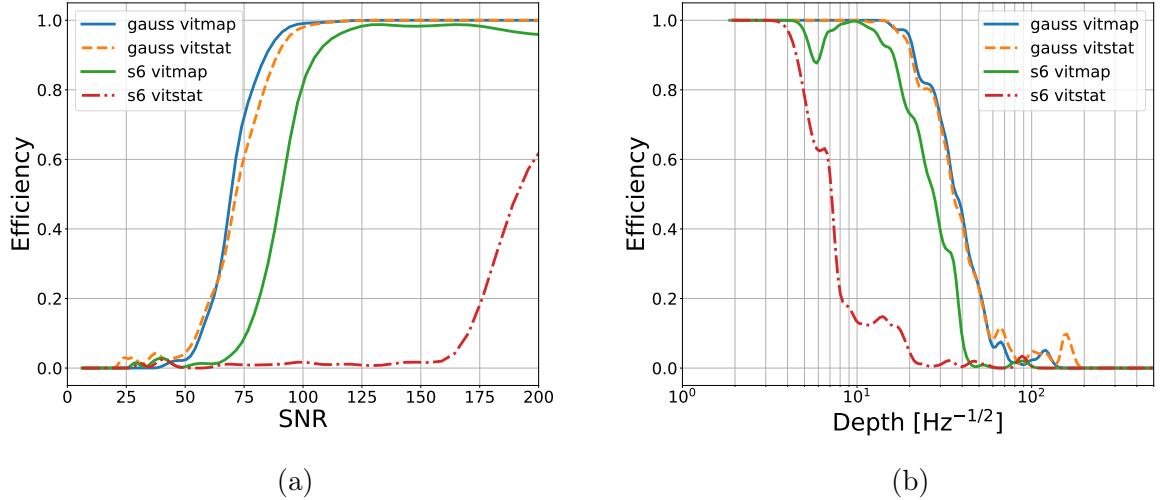


Figure 4.13: We aimed to compare the sensitivity of this search on simulations on real data (s6) to simulations in Gaussian noise (gauss). The Gaussian noise injections included the same gaps in data as the S6 data set. The **SNR** of the simulated signal in Gaussian noise was adjusted based on the noise floor of S6. The sensitivity curves show that the Viterbi map **CNN** (vitmap) achieves a similar sensitivity to the Viterbi statistic (vitstat) in Gaussian noise. This is because the main factor which effects the sensitivity of the Viterbi statistic is instrumental lines. As expected in real data, the Viterbi statistic is far less sensitive. The power of the **CNN** becomes clear in tests on real data. The sensitivity of the Viterbi map **CNN** is improved by over a factor of 2 in **SNR** when tested on S6 data.

into the S6 data-set. In the Gaussian noise data-set the curves for both statistics, Viterbi map and the Viterbi statistic, show very similar results, this is to be expected as the main use of the **CNN** was to reduce the effect of instrumental lines, for which there is none in this data set. The advantage of using the Viterbi maps in a **CNN** becomes clear which it is tested on injections into real data with many instrumental lines. The remaining curves in Fig. 4.13 show these tests, and it becomes clear that the Viterbi map **CNN** reduces the effect of instrumental lines and therefore increases the searches sensitivity. These tests in S6 also show that the effect of instrumental lines was far greater in this run than in O2. This is shown in Fig. 4.12 where the separation between the Viterbi statistic curves and the Viterbi map curves is much smaller than the S6 curves in Fig. 4.13. For injections into Gaussian noise following S6 gaps we show that with a false alarm of 1% the Viterbi map **CNN** achieves a sensitivity of SNR 85 and sensitivity depth of  $20 \text{ Hz}^{-1/2}$  with 95% efficiency. For injections into real S6 data the search achieves a sensitivity of SNR 115 and sensitivity depth of  $11 \text{ Hz}^{-1/2}$  with 95% efficiency and 1% false alarm. We can also see from Fig. 4.13 that the sensitivity in Gaussian noise with S6 gaps is better than in real S6 data, so there are still some elements of the search which reduces the sensitivity.

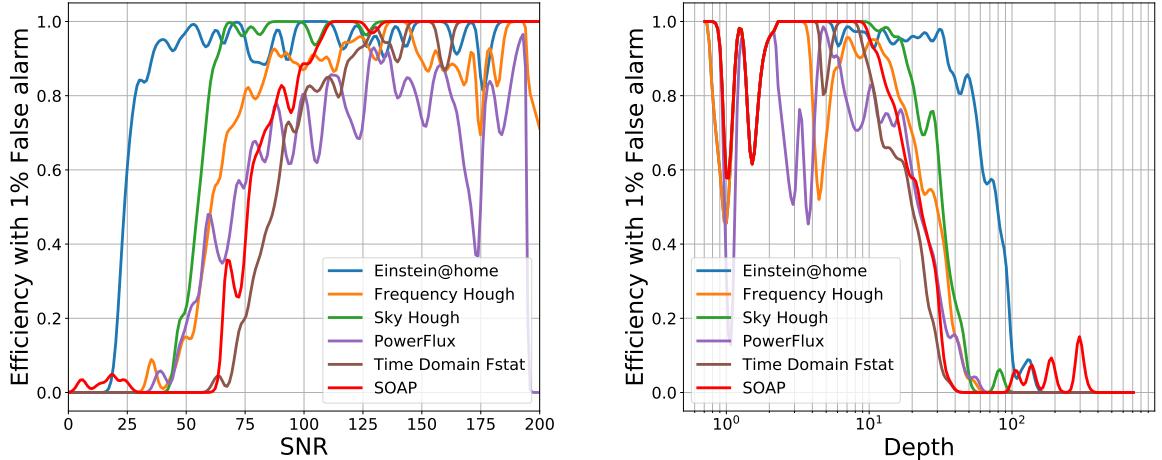


Figure 4.14: In the S6 MDC [53], sensitivity curves were made for a set of CW searches. We have taken the list of detected pulsars for each search from this paper [53] and replotted using the method in Sec. 4.9.1 to compare the sensitivities. This is results for all pulsar injections in the 40-500 Hz band in the S6 MDC. This shows how the SOAP search with a CNN achieves a similar sensitivity some other CW searches.

## S6

The final test was set up to again use the S6 data-set, however, we use a standard set of injections in the S6 MDC [53] to compare directly to other CW search pipelines. In Fig. 4.14 we show the results of the sensitivity curves from these injections. Fig. ?? shows the direct comparison in depth of the results in [53] with the results from the SOAP search with the Viterbi map CNN. This shows that we achieve a sensitivity similar to many other semi-coherent searches with the exception of the Einstein@home search [97]. For tests in the S6 MDC we show that with a false alarm of 1% the Viterbi map CNN achieves a sensitivity of SNR 90 and sensitivity depth of  $16 \text{ Hz}^{-1/2}$  with 95% efficiency.

### 4.9.2 Computational time

A key part of any CW search is the computational time taken, the majority of the time for this search is used generating the appropriate data. For each section of the pipeline we have take the average total time for each job to complete, where each job runs on 2.0 Hz and it runs between 100 and 400 Hz. These are approximate timings taken for all the jobs to finish when run on CIT and can vary.

This gives the entire search including testing a total run time of  $\sim 386$  hours, however, the majority of the time is taken generating data which can be easily parallelised. Rather than taking hundreds of hours, the generating data sections takes  $\mathcal{O}(1)$  hours in real time.

Table 4.2: Table shows the timings for each part of the search. These are approximate timings and vary when different amounts of data are input. This was run on data between 100-400 Hz.

Generating data on CPU		
Time [hrs]		
Narrow-banding	9	
Training data	239	
Test data	75	
Search data	40	
Training CNNs on GPU		
Training time [hrs]		Loading time [hrs]
Viterbi statistic	0.03	0.2
Viterbi map	0.8	0.7
spectrogram	9	1
Viterbi map		
+ Viterbi statistic	1	0.7
Viterbi map		
+ spectrogram	1.4	1.6
Viterbi map		
+ Viterbi statistic		
+ spectrogram	1.5	2
Testing CNNs on real data on GPU		
Testing [s]		Loading [s]
5		60 – 160

## 4.10 Sensitivity with the size of dataset

When training a network, the general rule is that the more data the better. More data can limit effects such as over-training mentioned in Sec. 4.5. To investigate how the sensitivity of the search changes with the number of training examples, the Viterbi map (vitmap) network in Sec. 4.9 was trained using a range of sizes of the training set. These networks are then tested on the same dataset to see how they perform. This was repeated for two data-sets: CW simulations in Gaussian noise and simulations in LIGO's O1 data-set. For both of these cases six different networks were trained, these used: 100, 500, 1000, 5000, 10000 and 15000 Viterbi maps as their training datasets. Each of these different sizes of training data is a randomly selected subset of the training data used in Sec. 4.9.1. The test data for each was the same entire tests sets as in Sec. 4.9.1.

In the Gaussian noise case, the majority of the networks performed the same. Fig. 4.15a

shows that in the Gaussian noise case, the majority of the networks performed with approximately the same sensitivity. This is with the exception of the network which was trained with 100 input Viterbi maps. Fig. 4.15b shows that the sensitivity does appear to decrease, however, this is only by a few values in **SNR**. The implication of this is that the information in the Viterbi maps is relatively easy to extract when simulations are in Gaussian noise. In this case the network is trying to distinguish Gaussian noise from a simulated **CW** signal. Therefore, one would expect this to be an easier problem for the network to solve than when trying to distinguish an **CW** signal from instrumental lines and Gaussian noise.

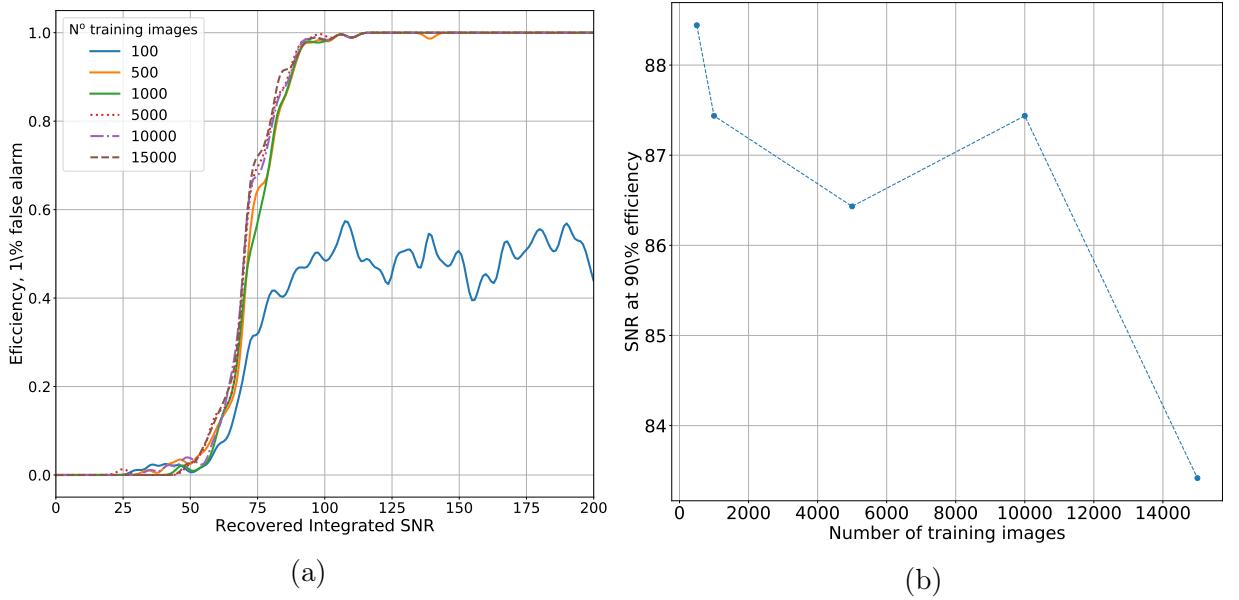


Figure 4.15: The amount of training data needed for an **CNN** to perform well depends on the problem. Here I show how the sensitivity changes as a function of the number of training examples for simulations in Gaussian noise. Fig. 4.15a shows the efficiency curves for each of the different data-set sizes and Fig. 4.15b shows the values of **SNR** at 90% efficiency as the data size increases. This shows that the increase in data size causes a slight increase in the sensitivity of the search. The efficiency curve for 100 training examples is not shown in Fig. 4.15b as it does not reach the 90% efficiency mark. **JOE: would be nice to get some errorbars of some sort on the right hand side plot**

When simulating signals in real O1 data, many of the sub-bands will contain instrumental lines. The noise class for the networks then contains many variations compared to the Gaussian noise case. This a harder challenge to the neural network by essentially increasing the size of the parameter space. Because of this, one would expect the network to need many more training examples to be able to achieve a similar sensitivity to Gaussian noise. In Fig. 4.16a, one can see that using 100 training examples is not enough and the network does not achieve any sensitivity at any **SNR**. This means that the **CNN** cannot classify any injection as detected using this number of training examples. Fig. 4.16b seems to agree that real data poses a harder problem as the sensitivity drastically increases as the

number of training examples is increased. Therefore, more training examples are needed for the network to perform well on real data.

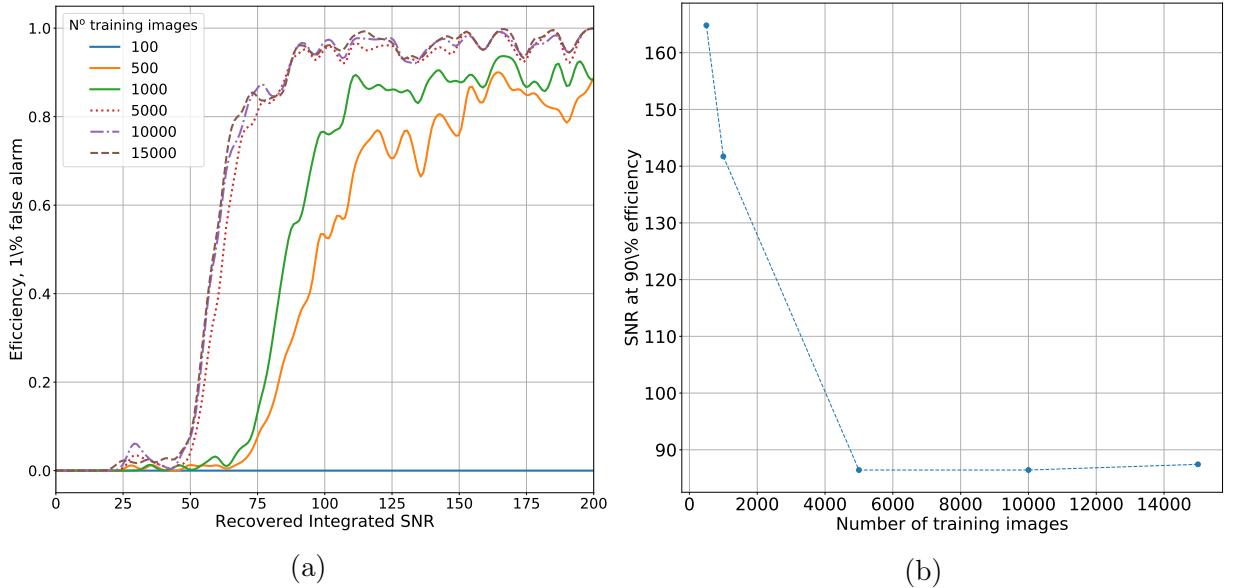


Figure 4.16: Here I show how the sensitivity changes as a function of the number of training examples for simulations in O1 data. Fig. 4.16a shows the efficiency curves for each of the different data-set sizes and Fig. 4.16b shows the values of **SNR** at 90% efficiency as the data size increases. This shows that the increase in data size causes a large increase in the sensitivity of the **CNN**. The efficiency curve for 100 training examples is not shown in Fig. 4.16b as it does not reach the 90% efficiency mark.

## 4.11 Network Visualisation

Neural network are generally hard to visualise due to the large number or parameters in the network that have to be varied. However, there are methods which can be used to see how input data is affected by the network. This can be useful to see how the network performs when given certain types of data and gives some insight into how the networks work

In the examples above we use **CNNs**, the first few layers of this are build using convolutional filters. The filters weights should ideally correspond to the shape of the feature which one wants to extract from the image. Generally this is only useful to picture at the first layer as the network can make subsequent layer and representations quite abstract.

## 4.12 Summary

In this paper we summarise an extension of the SOAP algorithm which makes use of a **CNN** to limit the effect of instrumental lines in a search for sources of continuous gravitational

waves. The SOAP search has a number of outputs for a given input spectrogram, where the main focus here is the Viterbi statistic and the Viterbi map. The Viterbi statistic has previously been used as a measure of whether there is a signal in a given frequency band, and the Viterbi maps are images which have the same shape as the input spectrogram, but gives a likelihood that there is a signal in a given location. The aim of the **CNN** was to use the Viterbi maps and spectrograms as input images such that each frequency band can be classified to either having a signal or not. This would then remove the need to manually look through frequency bands and remove ones which are contaminated with non astrophysical features.

We tested 6 separate **CNNs** which all take in different input data or different combinations of data as input. The three input data types are: the Viterbi statistic, the Viterbi map and the spectrograms which are summed and divided by a running median. The aim of using different input data types is that each would provide a different piece of information than the others, this had the hope that the combinations of these should then increase our sensitivity. The tests found that the **CNN** which uses the Viterbi map alone as input was more sensitive than any other which used a single data type as input. Each of the **CNNs** which used a combination of input data types had a similar sensitivity to the Viterbi map **CNN**, therefore, it is assumed that the Viterbi map provides the most useful information when detecting a signal. Given that the main aim of this paper was to reduce the effect of instrumental lines on the SOAP search, in Gaussian noise data, the **CNN** search should achieve a similar sensitivity to the Viterbi statistic alone. The tests in Gaussian noise with S6 gaps showed that at a 95 % efficiency and a 1% false alarm rate the Viterbi statistic and Viterbi map achieved a sensitivity of SNR 95 and 90 respectively. When the same test was run in real S6 data at a 95 % efficiency and a 1% false alarm rate the Viterbi statistic and Viterbi map achieved a sensitivity of SNR 300 and 120 respectively. This demonstrates that the Viterbi map has a much larger effect when used on real data due to the presence of many instrumental lines within real data.

These tests were once again repeated using a standard set of injections into S6 data such that a direct comparison can be made with other **CW** search pipelines. At a 95 % efficiency and a 1% false alarm rate the Viterbi map **CNN** achieved a sensitivity of **SNR**  $\sim 90$  and sensitivity depth  $\sim 14 \text{ Hz}^{-1/2}$ . We have shown that the SOAP + **CNN** approach can achieve a similar sensitivity to other semi-coherent **CW** search algorithms but with a greatly reduced computational cost.

This search also offers a lot of flexibility in the signal type which can be searched, in the above examples the focus is on isolated neutron stars such that a comparison can be made to other **CW** searches, however, this search is un-modelled. By changing the input parameters of the search, different signal types can be searched over, and in the future we aim to test its ability to identify more exotic sources of **GW**. Further to this, we aim to

make minor modifications to this pipeline such that some of the source parameters can be approximated. This should then return enough information to pass onto a more sensitive search for the signal.

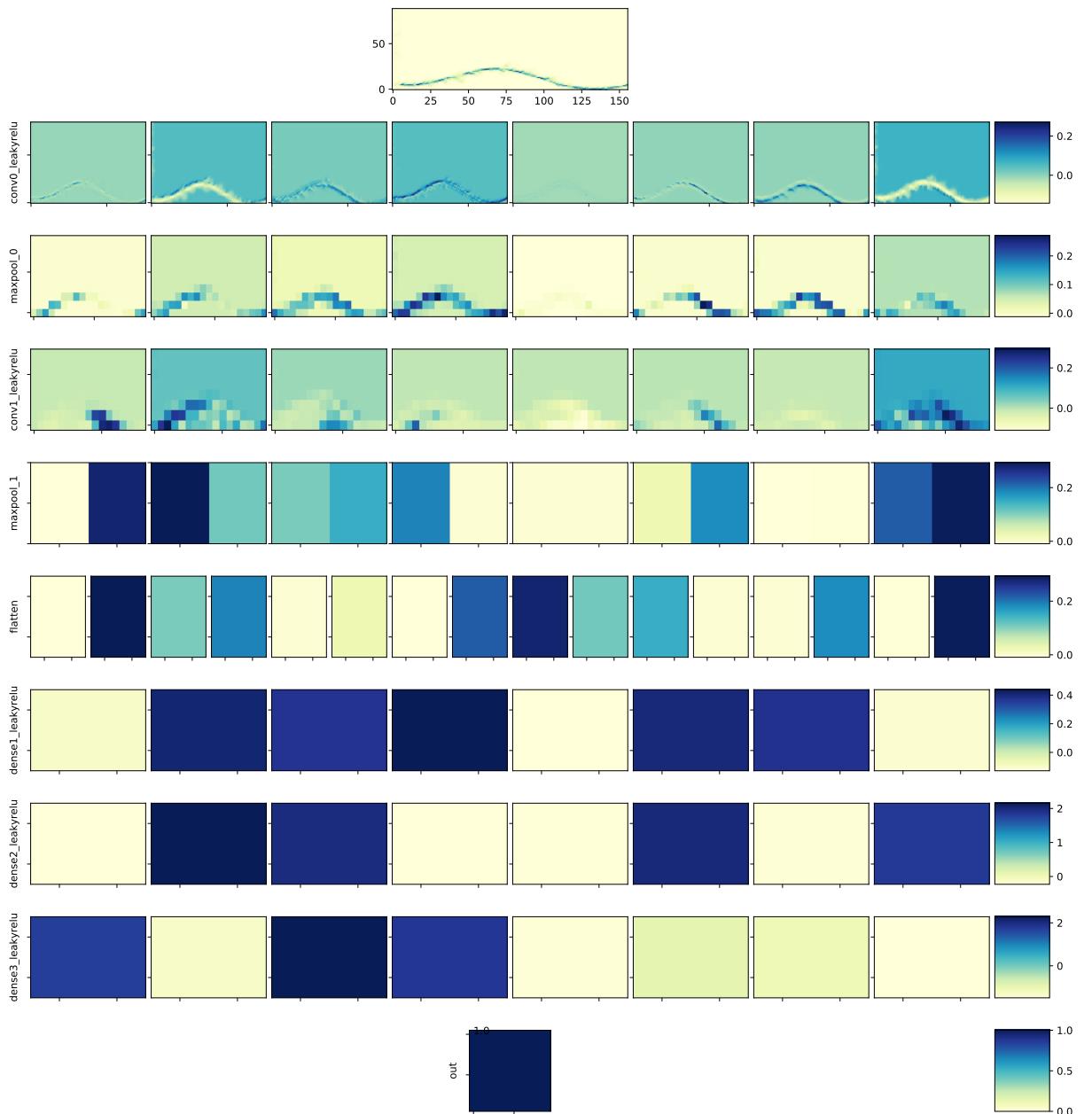


Figure 4.17: This shows a visualisation of the convolution neural network used for Viterbi maps above

# Chapter 5

## Detector Characterisation with SOAP

When searching for **GW** signals, it is important to understand the origins of noise artefacts in the detectors data which does not originate from an astrophysical source. A large fraction of **GW** search algorithms, including SOAP above, assume that the detectors noise follows a Gaussian distribution. However, the detectors contain artefacts which are not distributed as a Gaussian. These artefacts can negatively affect many searches for **GW** as they can be easily mistaken for a real **GW** signal. Some of the potential sources of these artefacts have been mentioned in Sec. 1.3.1. There are many different classes of artefact, some key types include: glitches, these are short duration broad band bursts in power and instrumental lines which are long duration narrow-band signals. To conduct a reliable search there are two main tasks which are necessary for detector characterisation. The first is identifying the artefact such that any search knows which areas in frequency and time are contaminated. The search can then address that section of data, this could mean removing that section of data or use more sophisticated techniques to deal with the artefact []. The second task is to find the source of the artefact. If the source of the artefact is found, it can potentially be removed or limited for future data runs.

The focus of this chapter is on a particular class of artefact called instrumental lines and how the affect **CW** searches. Sec. 5.1 will introduce different classes of instrumental line and how it affects a **CW** search. Sec. 5.2 will outline how these artefacts are detected and monitored, and describe current tools used for this task. Sec. 5.3 will describe how the **CW** search algorithm introduced in Sec. 3 can be used to search for instrumental lines. Finally Sec. 5.4 will show the outputs of the search and this is displayed for ease of use.

### 5.1 Instrumental lines

Instrumental lines have the general structure that they are persistent noise artefacts. There are many classes of instrumental line including: narrow, fixed frequency spectral artefacts or broad features which have a time varying frequency known as wandering lines.

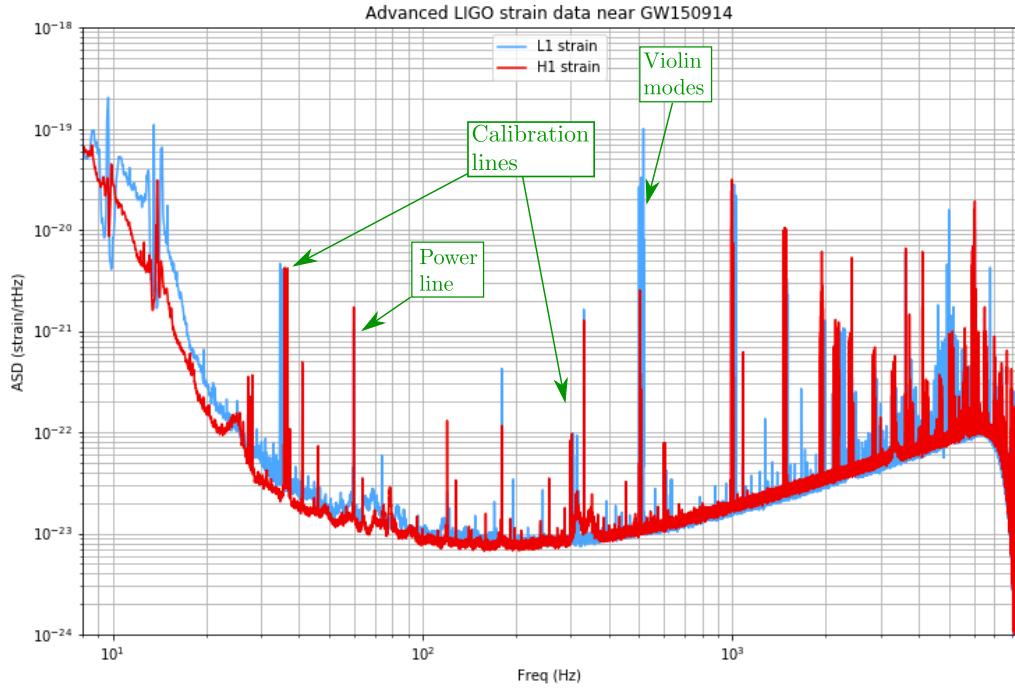


Figure 5.1: There are many features in the **LIGO** detectors averaged **ASD**. This image was taken from [98] where I have annotated some of the stronger lines. The power line from the mains in the USA is at 60 Hz. Some of the calibration lines are around 30 Hz, 331 Hz and 1083 Hz. The various violin modes of the suspensions are at 300, 500, 600 and 900 Hz where I have only marked the 500 Hz mirror suspension modes [98].

For many of these lines, it is difficult to distinguish them from an astrophysical signal. This affects search methods in two main ways. They can cause the search to produce outliers which are then considered as **GW** candidates. Extra efforts then have to be made to analyse these outliers further. If the line is close to the **GW** signal in frequency, then it can conceal the power of the **GW**, or if it is overlapping, the search can be almost impossible. It is therefore crucial to understand the structure and origin of these lines when performing a search for **GW**, specifically **CW** and stochastic searches.

Some instrumental lines are clearly visible when looking at one of the **LIGO** detectors frequency spectrum. Fig. 5.1 shows the **amplitude spectral density (ASD)** for **LIGO**'s Hanford and Livingston detectors during their first observing run (O1) [98]. This clearly shows some strong lines, some of which have been labelled. There are however, many more weaker lines which become visible when spectra are averaged over longer times. The frequency spectrum in Fig. 5.1 shows the time averaged spectra the **GW** channel of the **LIGO** detectors. The **GW** channel is essentially the error signal on the mirrors whilst they are being held on a resonance, i.e. how much the mirrors have to be moved to keep

the same output of the interferometer. The lines seen in the spectrum are not from any **GW** and are usually from ground based sources. To see the lines in the **GW** channel, they must be coupled in via some mechanism. There are a number of ways in which this happens which are outlined in [99]. This includes coupling via shared power and grounds. When different components share the same power supplies, if a component draws power with a given period, then the voltage will decrease repeatedly at this frequency. Another component which shares this same power supply can then also see this drop in voltage and this can potentially affect a stored output. Another mechanism is coupling through magnetic fields, this is common when cables are close to each other, the magnetic field in one can affect the other, therefore, coupling noise between different systems. Coupling can also occur through a physical connection, known as mechanical coupling, for example the resonances of the suspension fibres which couple directly into the mirrors.

Many of the spectral lines seen in the frequency spectrum, Fig. 5.1 are fundamental to the design of the detector. These cannot be eliminated, therefore need to be understood such that their effect on searches is minimised. Some of the strongest of these lines are listed below.

**Power line** The power line harmonics are fundamental to the detector and originate from the mains supply in the USA. These lines exist at 60 Hz which is the frequency of the mains alternating current oscillates at [80].

**Violin modes** The violin modes are associated with the suspensions of the mirrors and the beam splitter in the detector. These are designed to have a narrow frequency spectrum such that they contaminate as smaller part of the spectrum as possible. These are the lines around 500 Hz for the mirrors and 300, 600 and 900 Hz for the beam-splitter [98] in Fig. 5.1.

**Calibration lines** The mirrors of the **LIGO** detectors, are held at a resonance of the cavity in the arms of the interferometer. This then requires a feedback loop to hold them at resonance as a **GW** passes the detector. Calibration lines are used to calibrate this feedback loop such that the arm length changes are accurate [100, 81].

In [101], techniques are used to counter the affect of power lines and calibration lines on searches.

Along with the fundamental lines of the detector which cannot be removed at the source, there are a large number of other lines whose source has been found and can be removed. Many of these are from mechanisms described earlier such as shared power supplies or grounds. These can be removed by, for example, using a different power supply for different systems.

These lines have large effect on all searches for **GWs** both if the astrophysical signals frequencies overlap with the frequency of the line and can cause outliers. Long duration

searches for **CWs** are particularly sensitive to this type of artefact. As described in Sec. 2, **CW** are long duration signals with a slowly varying frequency. In the case of an isolated neutron star, the signal which is searched for is narrow-band and a fixed frequency which is Doppler modulated by the earths rotation and orbit. For certain areas of parameter space, the astrophysical signal of an isolated neutron star can appear very similar to an instrumental line. Many of these lines can be mitigated by using multiple detectors data. If a signal appears in a one detector and not the others, then it is likely that the signal is from an instrumental lines. These contaminated frequency bands can be removed of a statistic similar to that described in Sec. ?? or [102] can be used to limit their effect. However, there are many examples of instrumental line which appear at the same or similar frequencies in multiple detectors. These pose a real challenge to some **CW** searches, and require a lot of investigation to limit their affect.

## 5.2 Identifying and monitoring instrumental lines

When a detector is running, it is very important to identify instrumental lines and monitor them. This can then lead to either locating the origin of the line such that it can be removed, or allowing it to be flagged for other search algorithms. The lines affect searches in the **GW** channel, this is the output of the detector which **GW** are observed and is the data using is previous chapters. This is the channel where the affect of lines is intended to be minimised.

As well and the **GW** channel the detector records many different channels known as auxiliary channels. These channels monitor many components of the detector, and importantly are not sensitive to **GW**. Many of the channels useful for line searches are the **physical environment monitors (PEMs)**. These include sensors such as seismometers measuring the ground motion, temperature sensors etc. Magnetometers are a useful sensor in this case, they can be located around the electronics on the site to measure how these may be affecting the detector. These channels can be very useful in identifying the source of an instrumental line. The main goal is to reduce the number of artefacts in the **GW** such that it is as close to Gaussian noise as possible. If an artefact shows up in the **GW** channel in coincidence with one of the **PEM** or other channels, then this is an indicator that the artefact originates from something related to that **PEM**. For example, say a magnetometer located near some electronics has a long duration line in its spectrum at the same frequency as the main **GW** channel. This indicates that the magnetic field of that piece of electronics is somehow coupling into the detector. One can then investigate that piece of electronics further to see how it couples in.

There are a number of tools which are used to monitor these spectral lines, along with a team of people which regularly look though the results, a summary of these for the first

two observing runs of LIGO can be found in [99]. Some of the tools used are described below.

**Fscan** Fscan [81] takes FFTs of the raw detector data, typically these are 1800s long.

This is done for all of the auxiliary channels as well as the GW channel. The FFTs are then averaged over a day and time-frequency spectrograms are generated. After known lines such as Violin modes and power lines are subtracted, noise lines can be identified. A threshold can be set and anything in the spectrograms which exceed this threshold are flagged as a line. These can then be compared across multiple different channels. More detail on how the lines are identified can be found in [81]

**Coherence** Coherence searches for the coherence between different channels and different detectors. This is similar to searches for stochastic gravitational waves. More detail of how this works can be found in [99]

**Finetooth** Many of the instrumental lines found are part of combs. These are repeating structures where the harmonics (teeth) of a line make up the comb which is defined by the start frequency and tooth spacing. Finetooth is a tool which identifies and monitors these combs [].

**NoEMi** NoEMi uses FFTs and identifies peaks within the spectrum. It can then find coincidences with auxiliary channels and label is there are overlapping peaks with the GW channel. This can then tracks the lines with time. All of the information is stored in a database where more detail on its operation can be found in [103].

These tools offer many ways for the detector characterisation team to identify instrumental lines and hunt for their source. A summary of these efforts for the advanced LIGO data can be found in [99]. The following sections describe how the SOAP search described in Sec. 3 can be used as an extra tool to aid in the identification and monitoring of instrumental lines.

### 5.3 Identifying and cleaning lines with SOAP

The SOAP search has been run on a number of observing runs to search for CW. One of the major factors which limited the sensitivity of the search is instrumental lines. Through many of the searches performed on real data, many instrumental lines were identified. An example of this can be seen in Fig. ???. This demonstrates a broad and wandering line in a single detector and how the SOAP search is affected by it. Whilst for the majority of this thesis I have worked on developing techniques to limit the affect instrumental lines, the search also has the ability to identify these lines well. In this section, I will explain the setup of the search to identify instrumental lines.

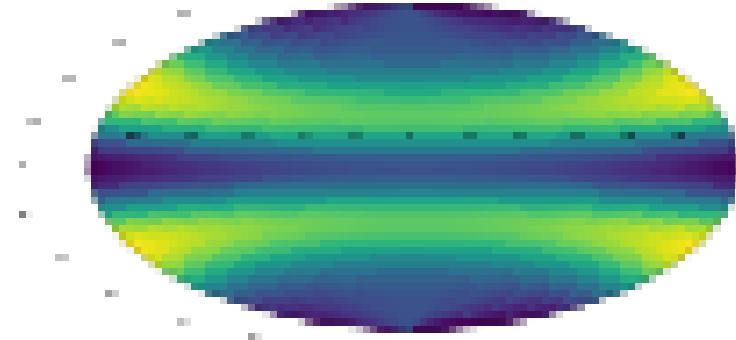


Figure 5.2: image of broad wandering line found by soap astrophysical search

Whilst it is useful to run searches on auxiliary channels when trying to identify the lines source, the aim of this search was to flag potential lines which can be then be investigated further. This can be done by eye or by using other tools. In the future this could be modified to search multiple channels, however, in this work searches are only run on the **GW** channel. Sec. 3 described how multiple detectors can be used to increase the sensitivity of a search for **CWs**. However, when searching for instrumental lines, the aim is to have the reverse effect, therefore, a simple search is run separately on each detector. This then removed any of the statistics developed in Sec. ?? and uses the ‘normalised’ **FFT** power as the statistic in the SOAP search. The single detector search then has one parameter to vary, the transition matrix parameter. This governs how probable the frequency track is to transition up straight or down a frequency bin. In this search we are aiming to find any non Gaussian artefacts, therefore, we allow an equal probability for the track to jump in any direction. The Fscan search described above generates **FFTs** every day of varying lengths, this include 1800 s long **FFTs**. We use this to our advantage as the search is already set up to search through 1800 s long time-frequency spectrograms. For this line search, we split the 1800 s long Fscan **FFTs** into 0.2 Hz wide sub-bands and run the single detector search on each sub-band. This then returns the same outputs as described in Sec. 3 and Sec. ??: the frequency track (Viterbi track), a Viterbi map and a Viterbi statistic. Here the Viterbi statistic is just the sum of the **FFT** power along the frequency track. This outputs a plot as shown in Fig. 5.3, 5.4 and 5.5. These plots and the equivalent plots for other sub-bands can then allow the sub-bands to be easily classified into containing an instrumental artefact or not.

Fig. 5.3 shows an example of a sub-band which does not contain any signal or spectral artefacts but is mostly Gaussian distributed noise. The track in the top panel of this plots shows the power spectrum including detector gaps from 1800 s **SFTs** in **LIGO** Hanfords data from its second observing run (O2). The Viterbi track is overlaid and the structure of the track indicates that there is no signal present, this is because the track appears

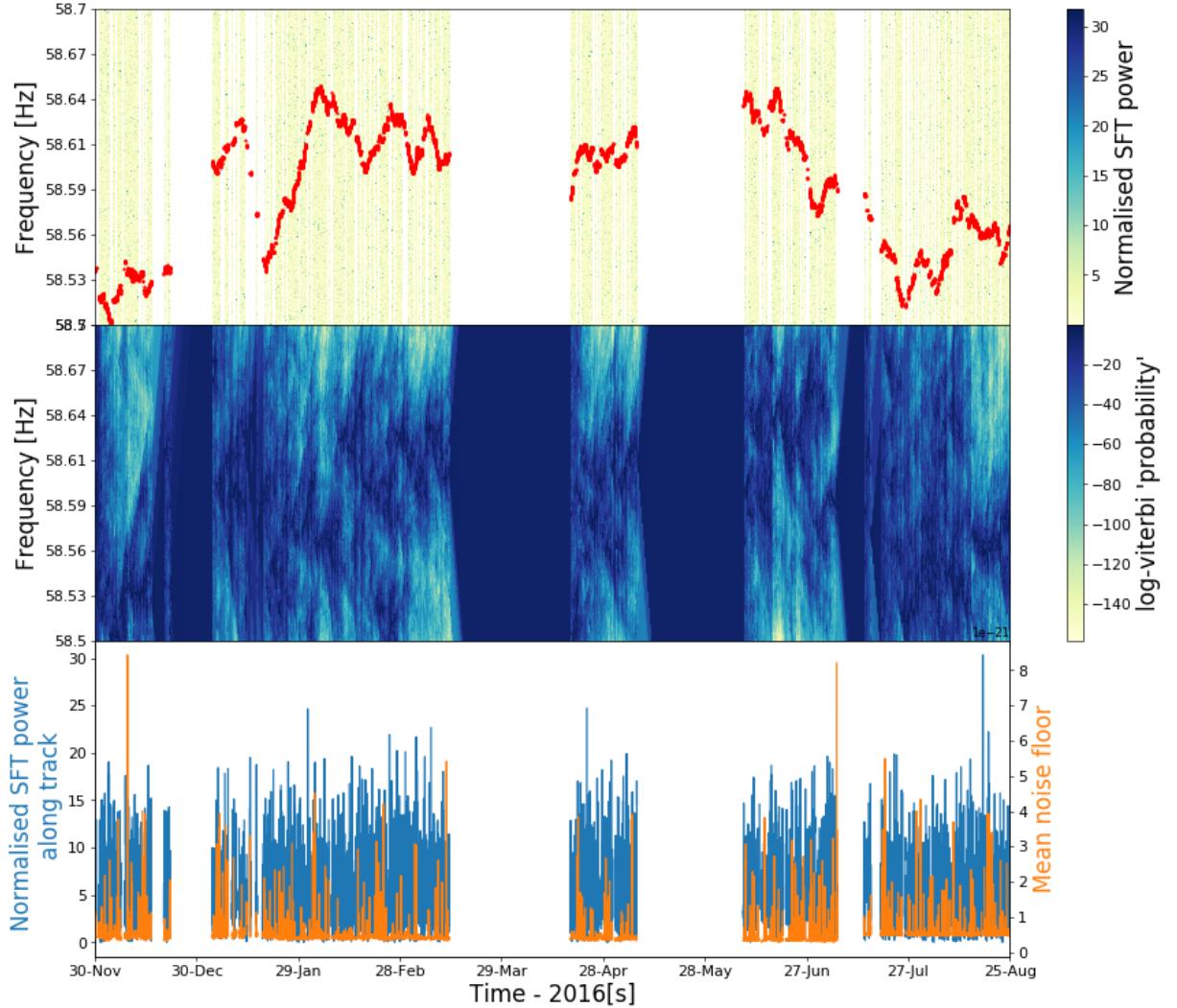


Figure 5.3: The SOAP search outputs three main quantities, the Viterbi maps, the Viterbi track and the Viterbi statistic. The Viterbi track is shown above overlaid onto the 1800s SFT power spectrum including the detector gaps for LIGO's Hanford detector (H1) in its second observing run (O2) [1]. This track is an indicator as to what type of signal the track is following. The above track indicated that this is just noise. The returned Viterbi statistic is also consistent with that of noise. The Viterbi map is another visualisation of the sub-band, how to interpret this has been explained in previous sections. However, here there does not appear to be a clear signal. The final panel is a way to visualise how the SFT power changes along the Viterbi track. Also on this plot is an estimate of the mean noise floor for this band to visualise how the sensitivity of the detector changed over the course of the run.

to be randomly wandering and has a large spread over the entire band. The Viterbi map plot does not show much structure and contains areas of low log-probability. Using this information along with the Viterbi statistic result, this particular sub-band can be considered to not contain an instrumental artefact. Fig. 5.4 demonstrates the same figure as before but now with a strong and narrow instrumental line in the sub-band. The Viterbi track now clearly indicates that there is a narrow spectral artefact present in the sub-band. The track does not have much spread over the band and stays at an approximately fixed frequency. The Viterbi map plot then shows that there are clear areas around the track which are of high log-probability. The areas of white in the Viterbi maps area areas where the probability of a signal falls to zero. These pieces of information along with the large value of the Viterbi statistic indicate that there is a narrow spectral artefact within the sub-band. Fig. 5.5 once again shows the equivalent plots to Fig. 5.3 and 5.4 but now contains some wandering spectral artefact. This is some line which wanders in frequency as it moves though the band. This can be seen in the frequency track, which here does not have much spread, however, the frequency of the potential signal changes with time. There are also areas where the signal could have potentially switched to a separate spectral artefact within the same band. Fig. 5.5 shows this discrete jump around January. The combination of this with the Viterbi statistic and Viterbi map would flag this sub-band as containing some sort of instrumental artefact. The example plots here show some specific examples, however, there is a lot of variation in the types of lines and features which appear in all of the SOAP outputs. **JOE: need to mention why this is good as extra tool, i.e. searching for wandering lines, identifies track of very weak lines etc**

This line search then works by initially ranking all of the Viterbi statistics, the largest values will be most likely from an instrumental artefact. The higher values of the statistic can then be viewed for further investigation. The high values are defined rather arbitrarily, this could either be something like the top 10% of sub-bands or could involve looking through the ranked list until signals begin to look like noise. The top results which are deemed to come from an instrumental line can be compared to the known line lists from the other line tools mentioned in Sec. 5.2. Any new lines can be investigated further by methods described in [99] using the tools in Sec. 5.2.

When running this search SOAP identified lines which did not appear in other line lists, therefore, offers a method to search for weaker lines. **JOE: actually check this**

## 5.4 Summary pages

Summary pages are an important tool when searching for instrumental lines. There is such a large amount of data both in frequency and time space and channels to search

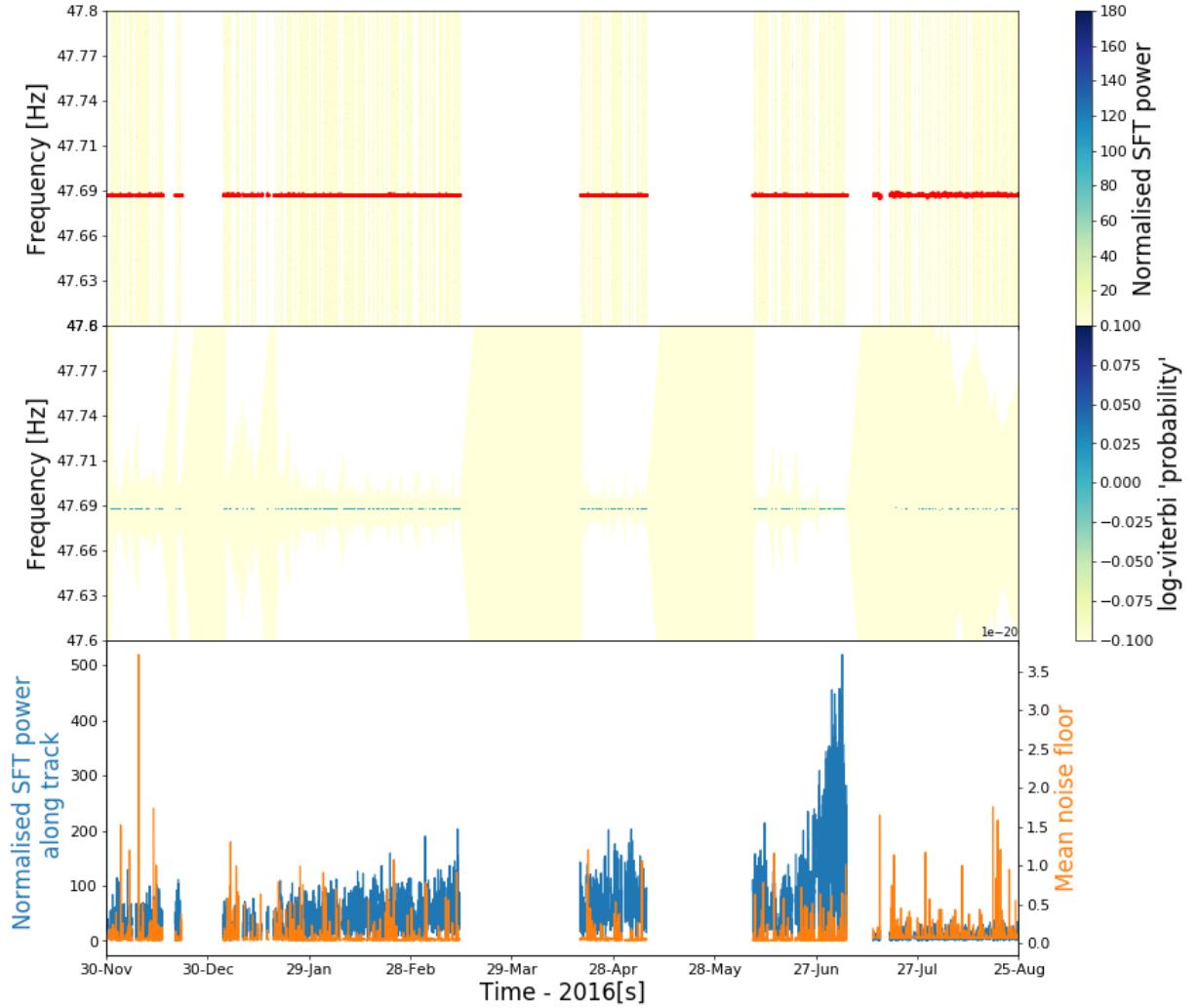


Figure 5.4: The equivalent plot as in Fig. 5.3 can be made when there is a narrow spectral artefact in the band. The above is again results from LIGO's Hanford detector (H1) in its second observing run (O2) using 1800s SFT power spectrum. In this there is a narrow spectral line at  $\sim 47.69$  Hz. The Viterbi track then follows this line of high power. The Viterbi map has much higher values for the log-probability in this line case compared to the noise case, this is an indicator some real signal. The probability in the Viterbi maps drops to zero in some areas due to the strength of the instrumental line.

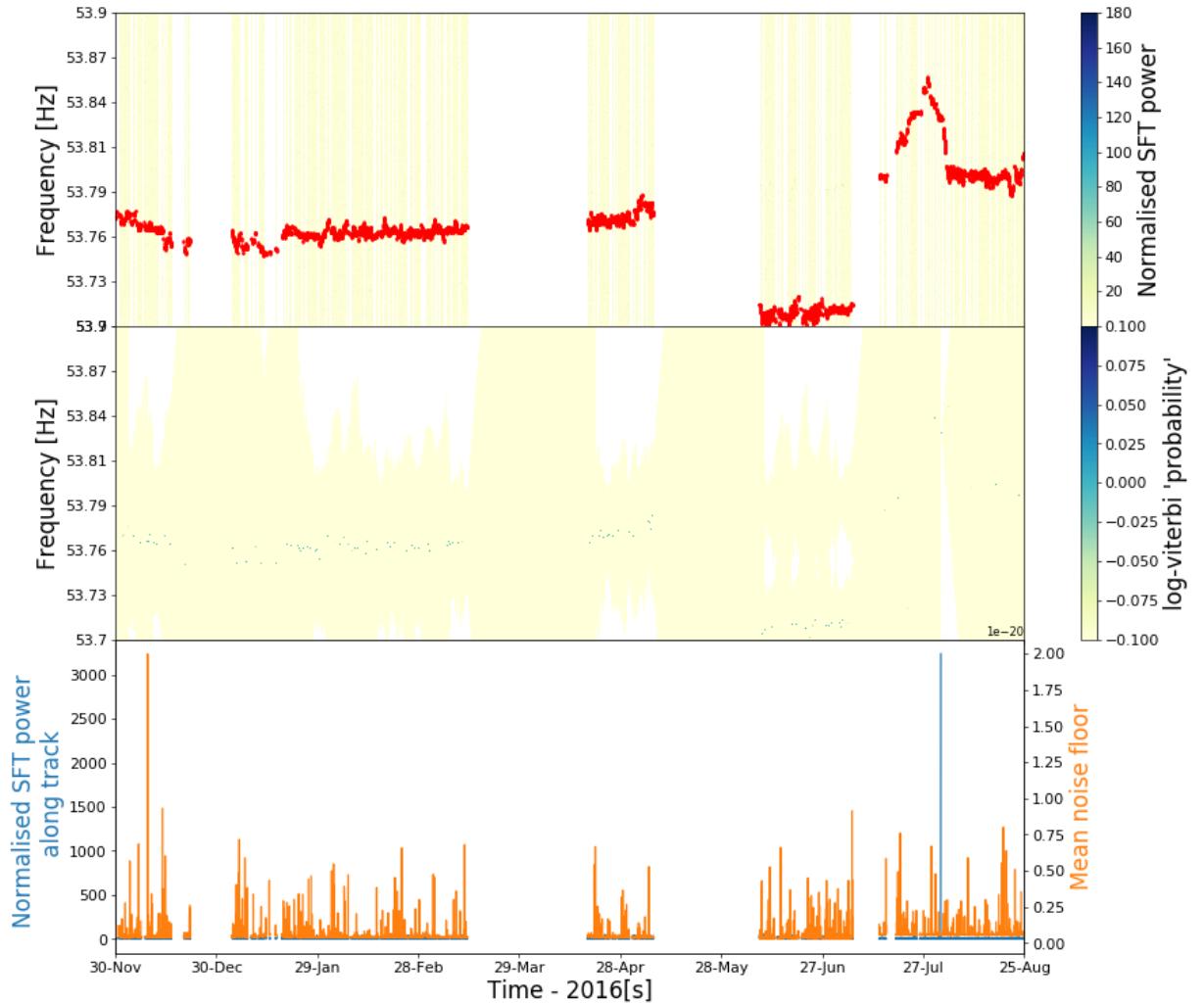


Figure 5.5: The equivalent plot as in Fig. 5.3 can be made when there is a wandering spectral line. The above is again results from LIGO's Hanford detector (H1) in its second observing run (O2) using 1800s SFT power spectrum. This shows how some spectral lines do not have a fixed frequency bin can wander through the band. These are especially hard to track and monitor. The Viterbi track here shows is clearly different from the noise case in Fig. 5.3 as the track is more tightly concentrated around some areas of power.

though when looking for instrumental lines. Summary pages should provide an easy way to view the important information to identifying the line as well as be easy to navigate to find a particular area of the spectrum which is of interest. These summary pages exist for the above searches in [104] where this is only accessible by LIGO members.

For the SOAP search summary pages were generated for each observing run and for the two LIGO detectors. This was done for various timescales: for the entire observing run and separately for each month. This allows the variation of a line to be observed for the entire length and also artefacts on shorter timescales to be observed. Once the detector, observing run and timescale is set, the band is split into 0.2 Hz wide sub-bands. The Viterbi search with a flat transition matrix and using the summed SFT power as the statistic. A flow diagram of how the SOAP search works for instrumental line searches can be found in Fig. 5.6. These stages are as follows:

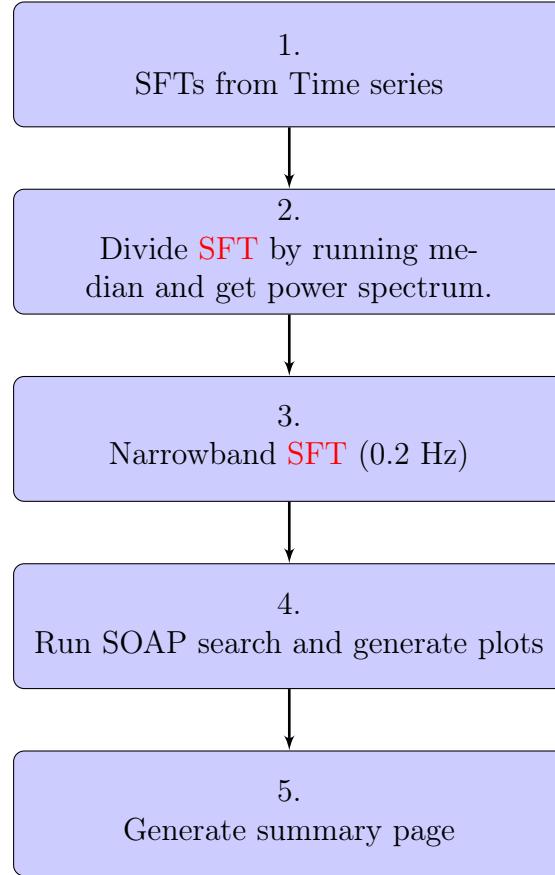


Figure 5.6: The SOAP search for instrumental lines is simpler than other searches. A simple version of the search is run separately for each detector, where the raw SFTs are divided by their running median, narrow-banded and then the search is run.

- 1. SFTs from timeseries** The SFTs are generated for the GW output channel. This is done by the Fscan search, therefore we do not repeat this process. Currently the search only runs on the GW channel, however, in the future could be made to run on others.

2. **Divide SFT by running median** In this stage each SFT is divided by its running median which is 100 bins wide. The running median takes each SFT and applies a window of 100 bins, where the median of these 100 frequency bins is taken. This window then slides over the SFT producing an ‘filtered’ SFT which should exclude outliers.
3. **Narrow-band SFT** The SFT is then split into 0.2 Hz wide sub-bands for the SOAP search to run on. These smaller bands are chosen as the SOAP search pull information on the most likely track, therefore, smaller bands are not contaminated by areas of high power in neighbouring frequency bands.
4. **Run SOAP and generate plots** This stage runs the SOAP search with a flat transition matrix probability and generates plots as shown in Fig. 5.3.
5. **Generate summary page** Finally the summary pages are built which take all of the bands and puts them in a table. This table can be ordered by the value of the Viterbi statistic, or can be searched for particular frequency bands.

An example of a summary page is shown in Fig. 5.7. This has been annotated showing how to navigate the page. There are generally two separate parts to the page: selecting the observing run and frequencies, and viewing the outputs.

The summary pages are then hoped to be useful alongside the other tools for searching for instrumental lines. **JOE: more**

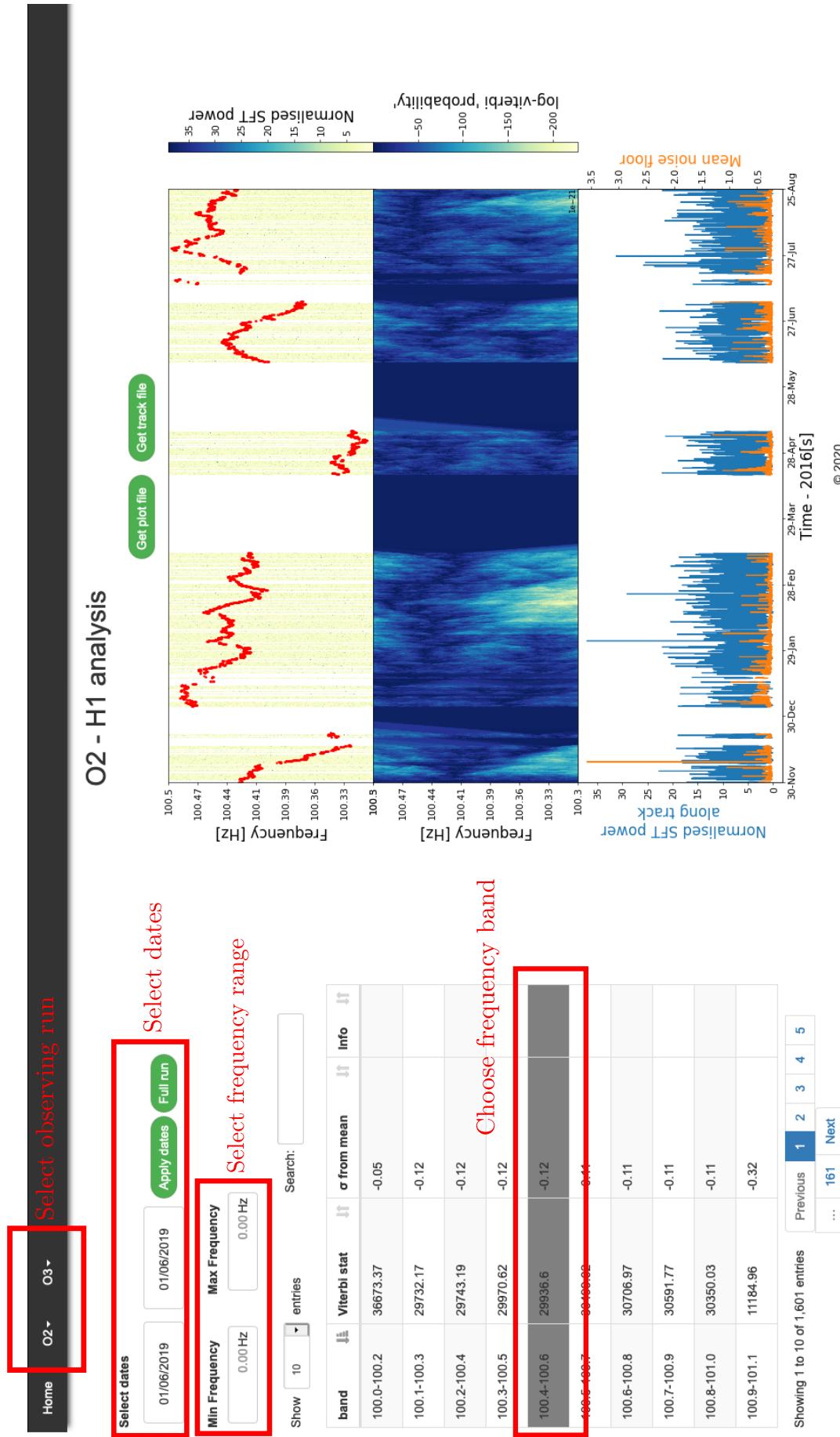


Figure 5.7: The summary pages are made for each observing run (in this case just O2 and O3). The range of times can then be selected from a set of start and end times. This is in general the entire observing run and monthly runs of this search. These pages can be found at [104].

# Chapter 6

## Summary

# Appendix A

## Continuous gravitational wave injections

In this section I outline how we inject a **CW** signal into data. This can generally be done in two different ways: simulating a signal in the time domain and injecting into time domain noise or simulating the signal in the frequency domain. The searches described in Sec. 3 and Sec. ?? only use output power spectra] Generating the time series and performing a Fourier transform or generating the signal in the frequency domain is time consuming. In this section I will outline how I simulate the power spectrum of **CW** signals and inject them into a **PSD**. This should greatly improve the speed of data generation.

### A.1 Signal SNR

To inject into a spectrogram the power spectrum of the signal will need to be simulated. In our injection we do not have access to a time-series, therefore, we do not simulate the signal in the time series or complex frequency domain. Instead, the **SNR** of the signal can be estimated in given frequency bins and injected straight into the power spectrum.

It can be shown that the **PSD** of Gaussian noise with zero mean and unit variance is a  $\chi^2$  distribution with 2 degrees of freedom. Therefore, if we want to generate a spectrogram for Gaussian noise, we just generate a two dimensional array of values distributed as  $\chi^2$  with two degrees of freedom. Assuming that there is some sinusoidal signal with a given **SNR** within a Gaussian noise time-series with zero mean and unit variance, the **FFT** power in a particular frequency bin can be estimated using a non-central  $\chi^2$  distribution with 2 degrees of freedom, where the non centrality parameter is the square of the **SNR**. To calculate the **SNR** in a given frequency bin the equation in [52] for optimal **SNR** was used,

$$\rho(0)^2 = \frac{1}{2} h_0^2 T S^{-1} [\alpha_1 A + \alpha_2 B + \alpha_3 C], \quad (\text{A.1})$$

where  $h_0$  is the **GW** amplitude,  $T$  is the total observing time in seconds,  $S^{-1}$  is the mean **PSD** noise floor. The values of  $\alpha$  are then defined in [52] by,

$$\begin{aligned}\alpha_1 &= \frac{1}{4} (1 + \cos^2(\iota))^2 \cos^2(2\psi) + \cos^2(\iota) \sin^2(2\psi) \\ \alpha_2 &= \frac{1}{4} (1 + \cos^2(\iota))^2 \sin^2(2\psi) + \cos^2(\iota) \cos^2(2\psi) \\ \alpha_3 &= \frac{1}{4} (1 - \cos^2(\iota))^2 \sin(2\psi) \sin^2(2\psi)\end{aligned}\quad (\text{A.2})$$

where  $\psi$  is the gravitational wave phase and  $\iota$  is the inclination angle of the source. The values of  $A$ ,  $B$  and  $C$  in Eq. A.1 are functions which represent the time averages antenna patterns, they are defined by,

$$\begin{aligned}A &\equiv \langle a^2 \rangle \\ B &\equiv \langle b^2 \rangle \\ C &\equiv \langle ab \rangle,\end{aligned}\quad (\text{A.3})$$

where  $a$  and  $b$  are the antenna pattern functions defined in [45] as,

$$\begin{aligned}a(t) &= \frac{1}{16} \sin 2\gamma (3 - \cos 2\lambda) (3 - \cos 2\delta) \cos[2(\alpha - \phi_r - \Omega t)] \\ &\quad - \frac{1}{4} \cos 2\gamma \sin \lambda (3 - \cos 2\delta) \sin[2(\alpha - \phi_r - \Omega t)] \\ &\quad + \frac{1}{4} \sin 2\gamma \sin 2\lambda \sin 2\delta \cos[\alpha - \phi_r - \Omega t] \\ &\quad - \frac{1}{2} \cos 2\gamma \cos \lambda \sin 2\delta \sin[\alpha - \phi_r - \Omega t] \\ &\quad + \frac{3}{4} \sin 2\gamma \cos^2 \lambda \cos^2 \delta,\end{aligned}\quad (\text{A.4})$$

$$\begin{aligned}b(t) &= \cos 2\gamma \sin \lambda \sin \delta \cos[2(\alpha - \phi_r - \Omega t)] \\ &\quad + \frac{1}{4} \sin 2\gamma (3 - \cos 2\lambda) \sin \delta \sin[\alpha - \phi_r - \Omega t] \\ &\quad + \cos 2\gamma \cos \lambda \cos \delta \cos[\alpha - \phi_r - \Omega t] \\ &\quad + \frac{1}{4} \sin 2\gamma \sin 2\lambda \cos \delta \sin[\alpha - \phi_r - \Omega t].\end{aligned}$$

Here  $\gamma$  is the orientation of the detectors arms,  $\lambda$  is the latitude of the detectors site,  $\alpha$  and  $\delta$  are the right ascension and declination of the **GW**,  $\phi_r$  is a deterministic phase defining the position of the earth and  $\Omega$  is the rotational angular velocity of the earth. This takes into account the antenna pattern modulation of the signal as the earth rotates the sun and orbits the earth. We then have a description of the **SNR** of a signal with a set of parameters for any given time and duration. This however does not describe how the

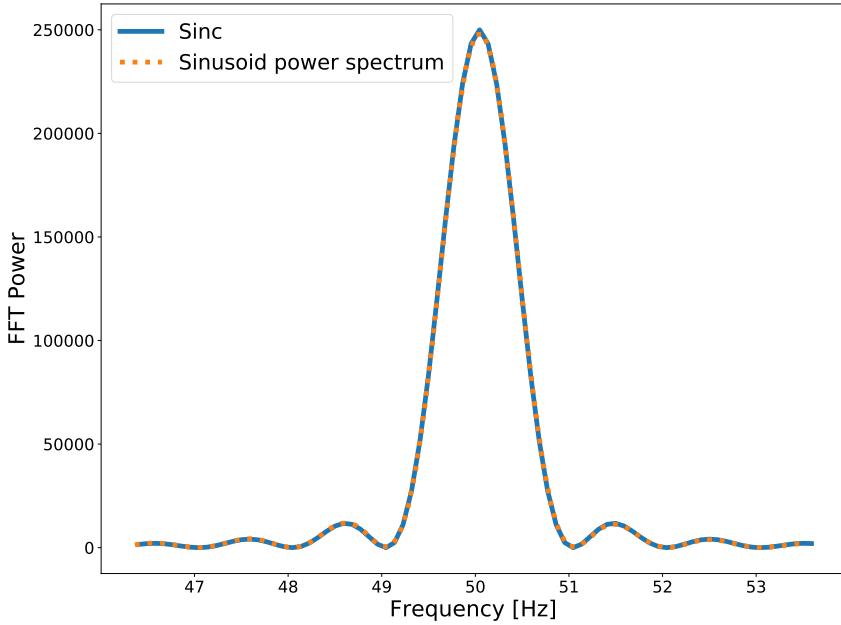


Figure A.1: If one takes the power spectrum of a sinusoid with finite length, this returns the same distribution as the square of a sinc function.

**SNR** of a signal varies with frequency which we need for spectrogram injections.

## A.2 SNR with frequency

If one takes a sinusoidal signal and the takes the Fourier transform of that, then this should be a delta function at the frequency of the sinusoid. However, in the real world this sinusoid has a limited length and one will instead take the **FFT** of that signal, the signal will then be broken into discrete frequency bins. If the sinusoids frequency falls at the center of a frequency bin then the entire power of the signal will be contained within that frequency bin. However, if it is not perfectly centered on a frequency bin, then the power of the signal will begin to be spread over surrounding frequency bins. The aim is then to simulate how the signal is spread over surrounding frequency bins. If one has a sinusoid with a finite length, this is equivalent to taking an infinitely long sinusoid and convolving it square window (box). The frequency response is then the Fourier transform of the sinusoid convolved with the Fourier transform of the box window. The Fourier transform of a box window is a sinc and of an infinitely long sinusoid is a delta function. The resulting Fourier transform is then a sinc function. One can show this by taking the power spectrum of a finite length sinusoid, and plotting the square of a sinc function on top as shown in Fig. A.1.

For a sinusoid which has some frequency derivative the Fourier transform changes slightly. One can start with the definition of a signal with a constant frequency derivative,

$$n(t) = a \cos [2\pi(f_0 t + 1/2\dot{f}t^2) + \phi], \quad (\text{A.5})$$

where  $f$  is its frequency,  $\dot{f}$  is its frequent derivative and  $t$  is time. In cases that follow we will have a finite frequency length, therefore when computing the Fourier transform, this is equivalent to applying a box window the signal. If we define the length of our signal as  $T$  we can redefine some terms as,

$$\begin{aligned} u &= \frac{t}{T} \\ r &= fT \\ \dot{r} &= \dot{f}T^2. \end{aligned} \quad (\text{A.6})$$

The signal can now be written in terms of  $u$  and  $r$  as,

$$\begin{aligned} n(u) &= a \cos [2\pi(r_0 u + 1/2\dot{r}u^2) + \phi] \\ &= \frac{1}{2}a \left[ e^{i2\pi(r_0 u + 1/2\dot{r}u^2)} e^{i\phi} + e^{-i2\pi(r_0 u + 1/2\dot{r}u^2)} e^{-i\phi} \right] \end{aligned} \quad (\text{A.7})$$

As the signal is zero outside of  $0 \leq u \leq 1$ , as in [105] we can write the Fourier transform at some frequency  $r_c$  as,

$$\begin{aligned} \tilde{n}(r_c) &= \frac{aN}{2} \int_0^1 n(u) \exp \{-i2\pi u r_c\} du \\ &= \frac{aN}{2} \left[ e^{i\phi} \int_0^1 e^{i2\pi(\frac{\dot{r}}{2}u^2 + r_{0c}u - r_cu)} du + e^{-i\phi} \int_0^1 e^{-i2\pi(\frac{\dot{r}}{2}u^2 + r_{0c}u + r_cu)} du \right] \\ &= \frac{aN}{2} e^{i\phi} [I_1 + I_2], \end{aligned} \quad (\text{A.8})$$

where  $r_c = r + \dot{r}/2$  and the signal frequency at bin center is  $r_{0c} = r_0 + \dot{r}/2$ . The first integral can then be written as,

$$\begin{aligned} I_1 &= \int_0^1 \exp \left\{ i2\pi \left( \frac{\dot{r}}{2}u^2 + r_{0c}u - r_cu \right) \right\} du \\ &= \int_0^1 \exp \left\{ i2\pi \left( \frac{\dot{r}}{2}u^2 + (r_{0c} - r_c)u \right) \right\} du \\ &= \int_0^1 \exp \left\{ i\frac{\pi}{2}2\dot{r} \left( u^2 + 2\frac{(r_{0c} - r_c)}{\dot{r}}u \right) \right\} du \\ &= \int_0^1 \exp \left\{ i\frac{\pi}{2}2\dot{r} \left[ \left( u + \frac{(r_{0c} - r_c)}{\dot{r}} \right)^2 - \left( \frac{(r_{0c} - r_c)}{\dot{r}} \right)^2 \right] \right\} du \\ &= \exp \left\{ -i\frac{\pi}{2}2\dot{r} \left( \frac{(r_{0c} - r_c)}{\dot{r}} \right)^2 \right\} \int_0^1 \exp \left\{ i\frac{\pi}{2}2\dot{r} \left[ \left( u + \frac{(r_{0c} - r_c)}{\dot{r}} \right)^2 \right] \right\} du \end{aligned} \quad (\text{A.9})$$

We can then substitute using,

$$v = \sqrt{2\dot{r}} \left( u + \frac{(r_{0c} - r_c)}{\dot{r}} \right) \quad (\text{A.10})$$

where,

$$dv = \sqrt{2\dot{r}} du \quad (\text{A.11})$$

and when  $u = 1$  we can define  $Y$  as,

$$Y = \sqrt{2\dot{r}} \left( 1 + \frac{(r_{0c} - r_c)}{\dot{r}} \right) \quad (\text{A.12})$$

The integral then becomes,

$$\begin{aligned} I_1 &= \exp \left\{ -i\frac{\pi}{2} 2\dot{r} \left( \frac{(r_{0c} - r_c)}{\dot{r}} \right)^2 \right\} \sqrt{2\dot{r}} \int_0^Y \exp \left\{ i\frac{\pi}{2} v^2 \right\} dv \\ &= \exp \left\{ -i\frac{\pi}{2} 2\dot{r} \left( \frac{(r_{0c} - r_c)}{\dot{r}} \right)^2 \right\} \sqrt{2\dot{r}} \int_0^Y \cos \left( \frac{\pi}{2} v^2 \right) + i \sin \left( \frac{\pi}{2} v^2 \right) dv \end{aligned} \quad (\text{A.13})$$

Similarly, using,

$$Z = \sqrt{2\dot{r}} \left( 1 - \frac{(r_{0c} - r_c)}{\dot{r}} \right) \quad (\text{A.14})$$

one can write,

$$\begin{aligned} I_2 &= \exp \left\{ -i\frac{\pi}{2} 2\dot{r} \left( \frac{(r_{0c} - r_c)}{\dot{r}} \right)^2 \right\} \sqrt{2\dot{r}} \int_0^Z \exp \left\{ i\frac{\pi}{2} v^2 \right\} dv \\ &= \exp \left\{ -i\frac{\pi}{2} 2\dot{r} \left( \frac{(r_{0c} - r_c)}{\dot{r}} \right)^2 \right\} \sqrt{2\dot{r}} \int_0^Z \cos \left( \frac{\pi}{2} v^2 \right) + i \sin \left( \frac{\pi}{2} v^2 \right) dv \end{aligned} \quad (\text{A.15})$$

One can then use the Fresnel  $S(x)$  and  $C(x)$  defined by

$$\begin{aligned} C(x) &= \int_0^x \cos \frac{\pi}{2} t^2 dt \\ S(x) &= \int_0^x \sin \frac{\pi}{2} t^2 dt \end{aligned} \quad (\text{A.16})$$

The Fourier transform is then,

$$A_{r_c} = \frac{aN}{2\sqrt{2\dot{r}}} e^{i(\phi - \pi q^2 / \dot{r})} [S(Z) - S(Y) + i(C(Y) - C(Z))], \quad (\text{A.17})$$

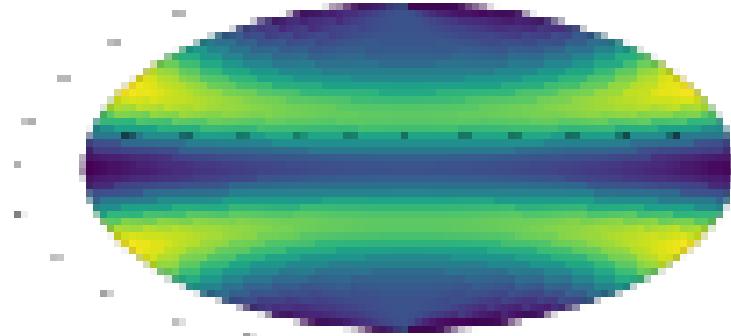


Figure A.2: comparing sin fft with sinc function

where, The power spectrum is then

$$|A_{r_c}|^2 = \frac{a^2 N^2}{4\dot{r}} [(S(Z) - S(Y))^2 + (C(Y) - C(Z))^2] \quad (\text{A.18})$$

This can then be converted back into frequencies using the relations in Eq. ??,

$$|A_{r_c}|^2 = \frac{a^2 N^2}{4\dot{f}T^2} [(S(Z) - S(Y))^2 + (C(Y) - C(Z))^2] \quad (\text{A.19})$$

and,

$$\begin{aligned} Z &= \sqrt{2\dot{f}T} \left( 1 - \frac{(f_{0c} - f_c)}{\dot{f}T} \right) \\ Y &= \sqrt{2\dot{f}T} \left( 1 + \frac{(f_{0c} - f_c)}{\dot{f}T} \right) \end{aligned} \quad (\text{A.20})$$

**JOE: there is a factor of 2 missing and its annoying**  $2\frac{(f_{0c} - f_c)}{\dot{f}T}$

By taking a simple signal, we can verify that this is correct. Fig. A.2 demonstrates the FFT of a simple signal and the power spectrum estimation using Eq. A.19.

The values of this for the center location of each frequency bin surrounding the signal frequency can then be calculated for each time segment. A full spectrogram of a loud signal ( $\sim 1000$  SNR) can be seen in Fig. A.3, A.4, A.5 and A.6 to demonstrate the signal simulations. The signal frequency for each time segment can be found using the model described in Sec. 2.1. Fig. A.3 shows an example of a signal of fixed frequency which is simulated at the center of a frequency bin. When the signals frequency is then moved to the edge of a bin, the power can be seen to be distributed evenly between the two surrounding frequency bins. This can be seen in Fig. A.4. The doppler shift of a signal can then be added such that the frequency changes with time. This is shown in Fig. A.5. Finally Fig. A.6 shows the Doppler modulation and the antenna pattern modulation of a

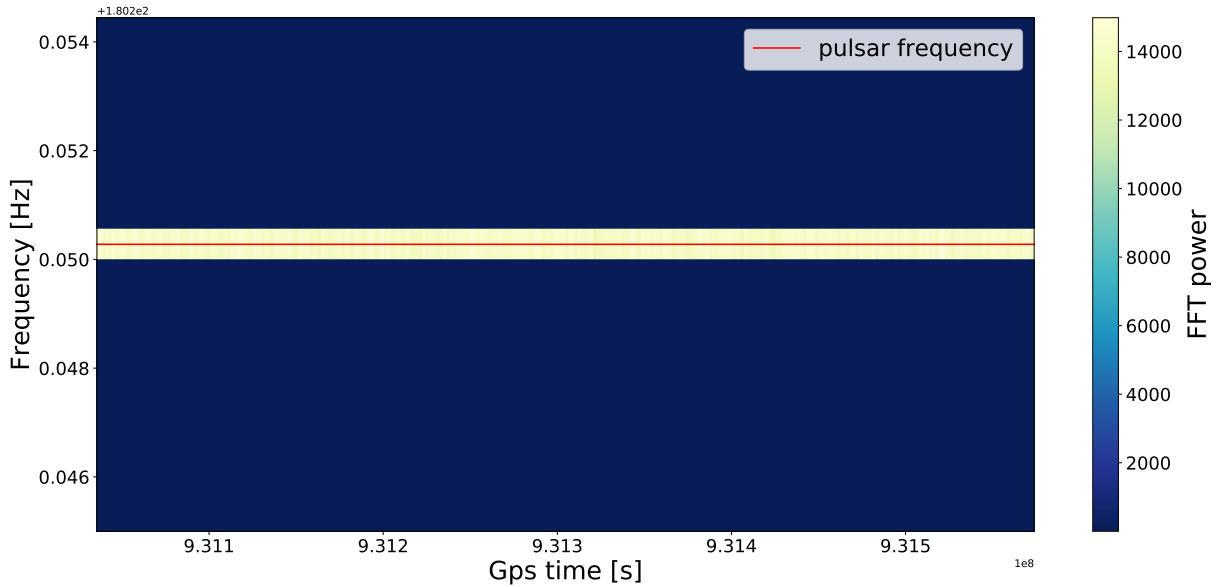


Figure A.3: By simulating a signal with a frequency in the center of a band, all of the signals power is contained within a single frequency bin. This shows an example of this kind of simulation in a spectrogram. The red line is the signals frequency evolution.

CW signal.

These simulations in the power spectrum greatly increased the speed of data generation when compared to simulating the signal in the time-series and taking their Fourier transform.

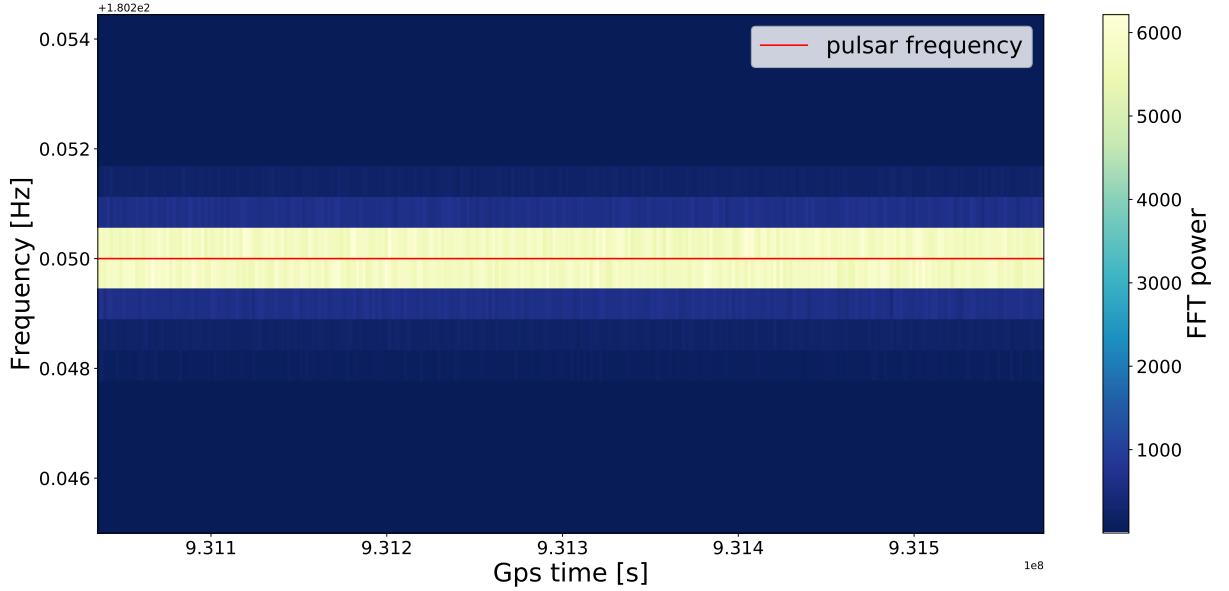


Figure A.4: By placing the signal at the edge of a frequency bin, the power is distributed around surrounding frequency bin, this can be seen above. The red line is the signals frequency evolution.

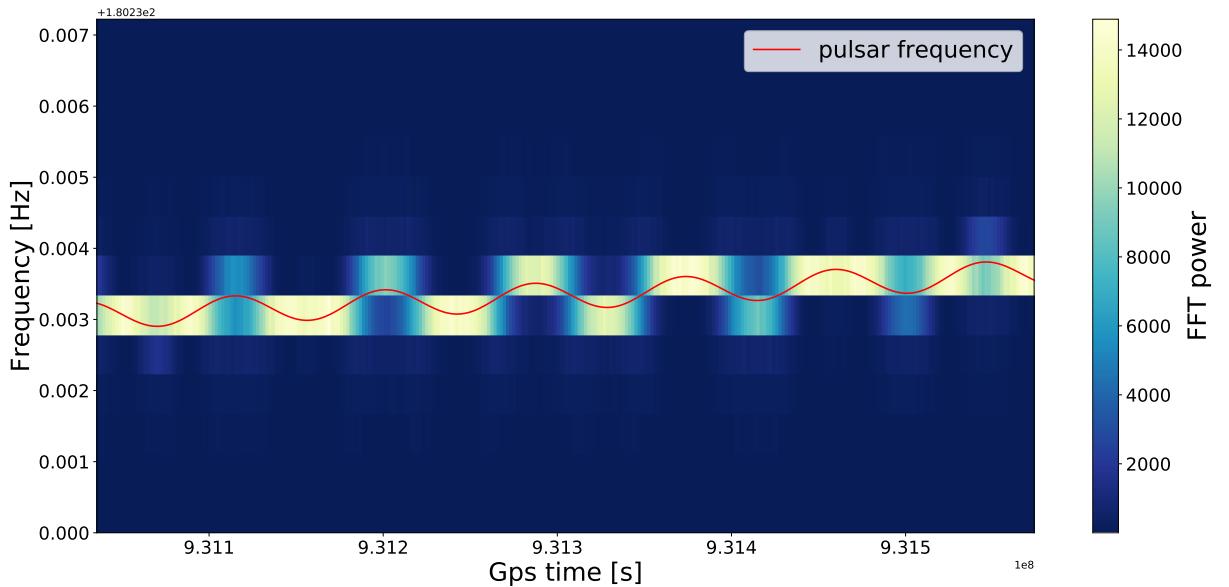


Figure A.5: Including the Doppler shift of a signal due to the earths rotation and orbit cause the signal to be modulated in frequency. This also causes a modulation in the **SNR** of the signal in a single frequency bin as it moves between bins. The red line is the signals frequency evolution.

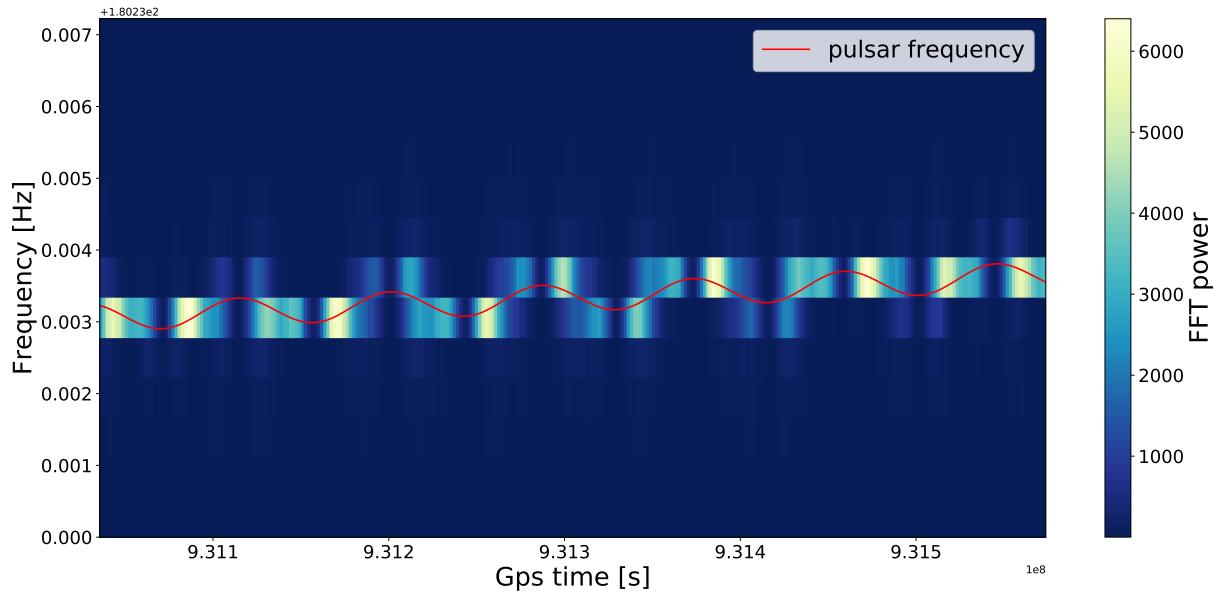


Figure A.6: The antenna pattern modulation can also be included to completely simulate a potential **CW** signal in a spectrogram. The red line is the signals frequency evolution.

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