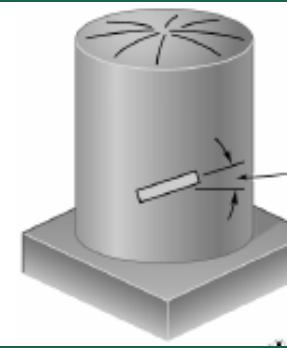


1. A strain gauge forming an angle $\beta = 18^\circ$ with a horizontal plane is used to determine the gauge pressure in the cylindrical steel tank shown (a kind of pressure vessel). The cylindrical wall of the tank is 6 mm thick, has a 600-mm inside diameter, and is made of a steel with $E = 200 \text{ GPa}$ and $\nu = 0.3$. Determine the pressure in the tank indicated by a strain gauge reading of 280×10^{-6} . [12 marks]



$$\sigma_{xx} = \frac{P\gamma}{t}$$

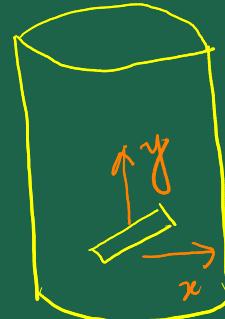
$$\sigma_{yy} = \frac{P\gamma}{2t}$$

$$\epsilon_{xx} = \frac{1}{E} (\sigma_{xx} - \nu \sigma_{yy})$$

$$= \frac{P\gamma}{Et} \left(1 - \frac{\nu}{2} \right)$$

$$\epsilon_{yy} = \frac{1}{E} (\sigma_{yy} - \nu \sigma_{xx})$$

$$= \frac{P\gamma}{Et} \left(\frac{1}{2} - \nu \right)$$



$$\epsilon_{xy} = 0 \quad (\because \tau_{xy} = 0)$$

$$\begin{aligned}\varepsilon_{x'x'} &= \frac{\varepsilon_{xx} + \varepsilon_{yy}}{2} + \frac{\varepsilon_{xx} - \varepsilon_{yy}}{2} \cos 2\beta + \varepsilon_{my}^0 \sin 2\beta \\ &= \frac{1}{2} \frac{P\gamma}{Et} \left(1 - \frac{\gamma}{2} + \frac{1}{2} - \gamma \right) + \frac{1}{2} \frac{P\gamma}{Et} \left(1 - \frac{\gamma}{2} - \frac{1}{2} + \gamma \right) \cos 2\beta \\ &= \frac{1}{2} \frac{P\gamma}{Et} \left[\frac{3}{2}(1-\gamma) + \frac{1}{2}(1+\gamma) \cos 2\beta \right]\end{aligned}$$

$$\therefore P = \frac{2Et\varepsilon_{x'x'}}{\gamma \left[\frac{3}{2}(1-\gamma) + \frac{1}{2}(1+\gamma) \cos 2\beta \right]}$$

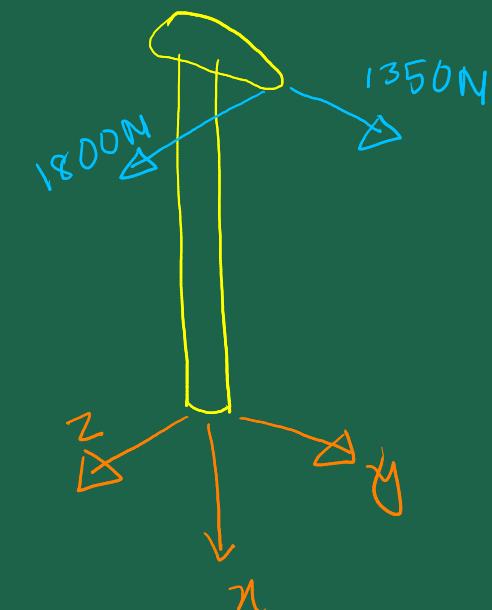
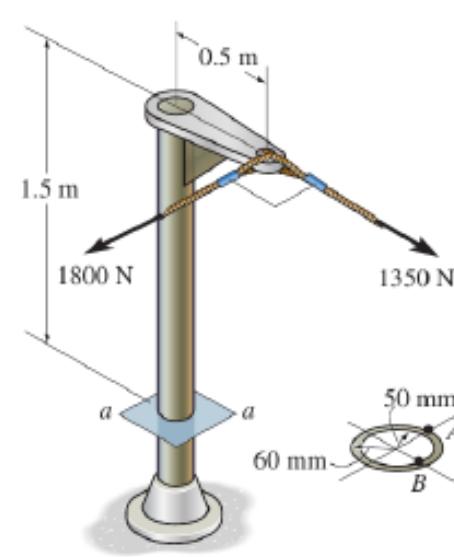
$$= 1.42 \text{ MPa}$$

This was a "gift".

Those who did the major step correctly received 10.

A few students received less because they did not even use stress/strain transformation.

2. Determine the normal stress and the shear stress at the point B on the cross-section of the post at section $a-a$. Note that B is a point on the inner boundary of the annulus. [16 marks]



$$V_y = -1350 \text{ N}$$

$$V_z = -1800 \text{ N}$$

$$T = -900 \text{ Nm}$$

$$M_y = -2700 \text{ Nm}$$

$$M_z = 2025 \text{ Nm}$$

No marks for these

$$I_y = I_z = \frac{\pi}{4} (60^4 - 50^4) = 5.27 \times 10^6 \text{ mm}^4$$

$$J = \frac{\pi}{2} (60^4 - 50^4) = 10.54 \times 10^6 \text{ mm}^4$$

$$(Q_1)_B = 0$$

$$(Q_2)_B = \frac{4}{3\pi} \sigma_o \left(\frac{\pi}{2} \sigma_o^2 \right) - \frac{4}{3\pi} \sigma_i \left(\frac{\pi}{2} \sigma_i^2 \right) = 6.067 \times 10^4 \text{ mm}^3 \quad (1 \text{ mark})$$

Normal Stress:

$$\sigma_B = - \frac{M_z y}{I_z} = -19.21 \text{ MPa} \quad \begin{array}{l} (1 \text{ mark for the -ve sign or writing compressive stress}) \\ (5 \text{ marks for } \sigma_B \text{ magnitude}) \end{array}$$

Shear Stress:

$$\text{Due to torsion: } \tau_{xz}|_T = \frac{T \sigma_i}{J} = 4.269 \text{ MPa} \quad (3 \text{ marks})$$

$$\text{Due to transverse load } \tau_{xz}|_V = \frac{V z Q_2}{I_y t} = 1.036 \text{ MPa} \quad (3 \text{ marks})$$

Both these shear stresses are in the same dir^n.

$$\therefore \tau_{xz|_{\text{total}}} = \tau_{xz}|_T + \tau_{xz}|_V = 5.305 \text{ MPa}$$

2 marks for the addition

4. A thin annular disk of inner radius r_i and outer radius r_o is fixed at its outer boundary to a rigid support. However axial displacement is not constrained. At the inner boundary, there is a force (per unit area) f acting radially inward. Starting from the governing differential equation for radial displacement in the form: $\frac{d^2u}{dr^2} + \frac{1}{r} \frac{du}{dr} - \frac{u}{r^2} = 0$, and using the appropriate boundary conditions, determine the radial stress at the outer boundary. [12 marks]

Let $u = \gamma^m$. Substituting in the eqn and solving gives us:

$$u = C_1 \gamma + C_2 \frac{1}{\gamma} \quad \text{--- (mark)}$$

$$\text{@ } r = r_o, u = 0 \Rightarrow C_1 r_o + C_2 \frac{1}{r_o} \Rightarrow C_2 = -C_1 r_o^2$$

Now, this is a plane stress case. So,

$$\begin{aligned}\sigma_{rr} &= \frac{E}{1-\nu^2} \left[\epsilon_{rr} + \nu \epsilon_{\theta\theta} \right] \\ &= \frac{E}{1-\nu^2} \left[\frac{du}{dr} + \nu \frac{u}{r} \right] \\ &= \frac{E}{1-\nu^2} \left[C_1 - \frac{C_2}{r^2} + \nu \left(C_1 + \frac{C_2}{r^2} \right) \right]\end{aligned}$$

(6 marks deducted for those who wrote the formula for plane strain. I had said on the last day of class that I would give D to those who make this mistake. But I changed my mind.)

$$\textcircled{a} \quad \delta = \delta_i, \quad \sigma_{\delta\delta} = f$$

$$\Rightarrow \frac{E}{1-\nu^2} \left[c_1(1+\nu) - \frac{c_2}{\delta_i^2} (1-\nu) \right] = f$$

$$\Rightarrow \frac{E}{1-\nu^2} \left[c_1(1+\nu) + c_1 \frac{\delta_0^2}{\delta_i^2} (1-\nu) \right] = f$$

$$\Rightarrow c_1 = \frac{f}{\frac{E}{1-\nu^2} \left[1+\nu + \frac{\delta_0^2}{\delta_i^2} (1-\nu) \right]}$$

$$\therefore \sigma_{\delta\delta} \Big|_{\delta=\delta_0} = \frac{E}{1-\nu^2} \left[c_1(1+\nu) - \frac{c_2}{\delta_0^2} (1-\nu) \right]$$

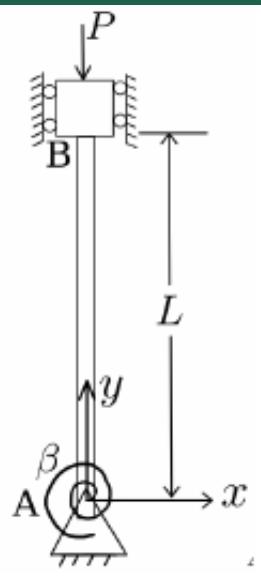
$$= \frac{E}{1-\nu^2} \left[c_1(1+\nu) + c_1 \cancel{\frac{\delta_0^2}{\delta_0^2}} (1-\nu) \right]$$

$$= \frac{E}{1-\nu^2} \left[2c_1 \right] = \frac{2f}{1+\nu + \frac{\delta_0^2}{\delta_i^2} (1-\nu)}$$

serious penalty for those
who wrote wrong B(S),
especially who wrote
something like

$$\sigma_{\delta\delta} = -P_0 \text{ @ } \delta = \delta_0 .$$

5. Consider a column with constant flexural rigidity (EI) as shown in the figure. The bottom end A ($x = 0$) is pinned and attached with a rotational spring of stiffness β . The top end B ($x = L$) is fixed to the vertical guide. Showing all the steps involving the solution of the 4th-order differential equation together with the use of the proper boundary conditions, determine the critical condition(s) for buckling. [12 marks]



[In the figure the labels of the x and y axes were interchanged by mistake.]

The general solution of the 4th order differential eqn, $\frac{d^4y}{dx^4} + K^2 \frac{d^2y}{dx^2} = 0$ is

$$y = A \sin Kx + B \cos Kx + Cx + D \quad (1 \text{ mark})$$

$$\frac{dy}{dx} = AK \cos Kx - BK \sin Kx + C$$

$$\frac{d^2y}{dx^2} = -AK^2 \sin Kx - BK^2 \cos Kx$$

(There are different ways of proceeding with the simplifications after writing the BCs.
But the BCs are most important.)

$$@ x=0, y=0 \quad - \text{ (2 marks)}$$

$$\Rightarrow B+D=0 \quad - \textcircled{1}$$

$$@ x=0, EI \frac{d^2y}{dx^2} = -\beta \frac{dy}{dx} \quad - \text{ (2 marks)}$$

$$\Rightarrow EI(-BK^2) = -\beta(AK + C) \Rightarrow C = \frac{EIK^2}{\beta}B - AK \quad - \textcircled{2}$$

$$@ x=L, y=0 \quad - \text{ (2 marks)}$$

$$\Rightarrow A \sin KL + B \cos KL + CL + D = 0 \quad - \textcircled{3}$$

$$@ x=L, \frac{dy}{dx}=0 \quad - \text{ (2 marks)}$$

$$\Rightarrow AK \cos KL - BK \sin KL + C = 0$$

$$\Rightarrow AK \cos KL - BK \sin KL + \frac{EIK^2}{\beta}B - AK = 0 \quad (\text{using } \textcircled{2})$$

$$\Rightarrow AK(\cos KL - 1) = BK \left(\sin KL - \frac{EIK}{\beta} \right) \quad - \textcircled{4}$$

Using ① and ② in ③, we have:

$$A \sin KL + B \cos KL + L \left(\frac{EI K^2}{\beta} B - AK \right) - B = 0$$

$$\Rightarrow A(\sin KL - KL) = B \left(1 - \cos KL - \frac{EI K^2 L}{\beta} \right) - \quad ⑤$$

Dividing ④ by ⑤, we obtain:

$$\frac{\cos KL - 1}{\sin KL - KL} = \frac{K \left(\sin KL - \frac{EI K}{\beta} \right)}{1 - \cos KL - \frac{EI K^2 L}{\beta}}$$

(Some students may have obtained other forms of the above critical condition. Those are fine as long as they have proceeded from the correct BCs.)

I kept 3 marks for the simplification. Not many students proceeded well. But those who tried to proceed somewhat, I gave $\frac{1}{2}$, 1 or 2 marks depending on their effort.