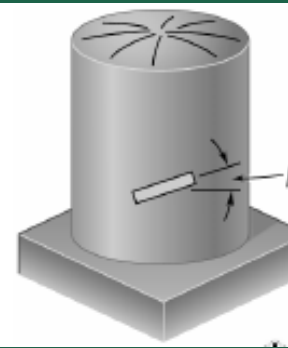


1. A strain gauge forming an angle $\beta = 18^\circ$ with a horizontal plane is used to determine the gauge pressure in the cylindrical steel tank shown (a kind of pressure vessel). The cylindrical wall of the tank is 6 mm thick, has a 600-mm inside diameter, and is made of a steel with $E = 200$ GPa and $\nu = 0.3$. Determine the pressure in the tank indicated by a strain gauge reading of 280×10^{-6} . [12 marks]



$$\sigma_{xx} = \frac{pr}{t}$$

$$\sigma_{yy} = \frac{pr}{2t}$$

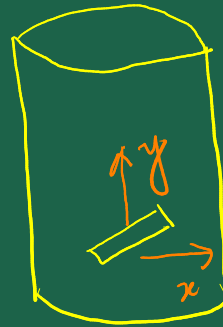
$$\epsilon_{xx} = \frac{1}{E} (\sigma_{xx} - \nu \sigma_{yy})$$

$$= \frac{pr}{Et} \left(1 - \frac{\nu}{2}\right)$$

$$\epsilon_{yy} = \frac{1}{E} (\sigma_{yy} - \nu \sigma_{xx})$$

$$= \frac{pr}{Et} \left(\frac{1}{2} - \nu\right)$$

$$\epsilon_{xy} = 0 \quad (\because \tau_{xy} = 0)$$



$$\begin{aligned}
 \epsilon_{x'x'} &= \frac{\epsilon_{xx} + \epsilon_{yy}}{2} + \frac{\epsilon_{xx} - \epsilon_{yy}}{2} \cos 2\beta + \cancel{\epsilon_{xy}}^0 \sin 2\beta \\
 &= \frac{1}{2} \frac{P\gamma}{Et} \left(1 - \frac{\nu}{2} + \frac{1}{2} - \nu \right) + \frac{1}{2} \frac{P\gamma}{Et} \left(1 - \frac{\nu}{2} - \frac{1}{2} + \nu \right) \cos 2\beta \\
 &= \frac{1}{2} \frac{P\gamma}{Et} \left[\frac{3}{2} (1 - \nu) + \frac{1}{2} (1 + \nu) \cos 2\beta \right]
 \end{aligned}$$

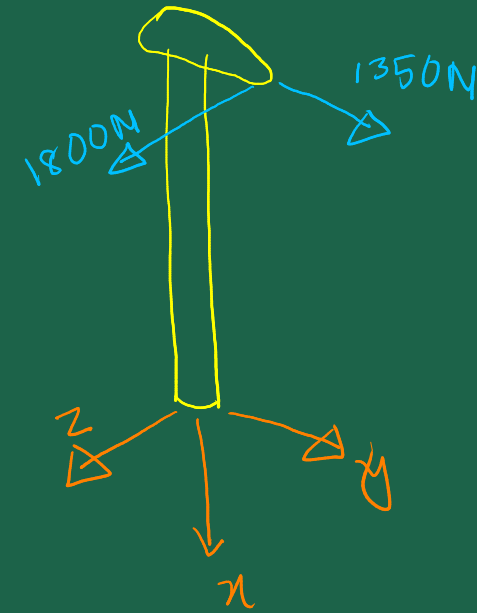
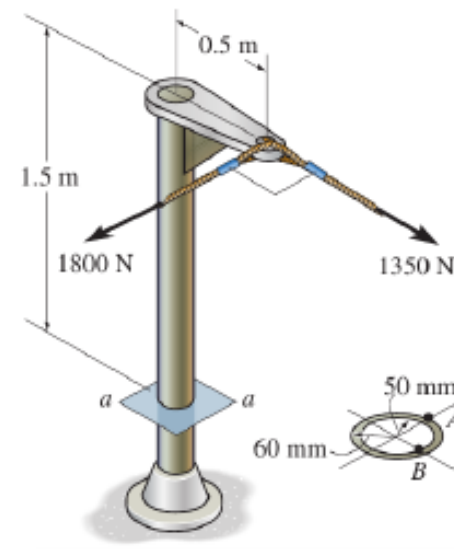
$$\begin{aligned}
 \therefore P &= \frac{2 Et \epsilon_{x'x'}}{\gamma \left[\frac{3}{2} (1 - \nu) + \frac{1}{2} (1 + \nu) \cos 2\beta \right]} \\
 &= 1.42 \text{ MPa}
 \end{aligned}$$

This was a "gift".

Those who did the major step correctly received 10.

A few students received less because they did not even use stress/strain transformation.

2. Determine the normal stress and the shear stress at the point B on the cross-section of the post at section $a - a$. Note that B is a point on the inner boundary of the annulus. [16 marks]



$$V_y = -1350 \text{ N}$$

$$V_z = -1800 \text{ N}$$

$$T = -900 \text{ Nm}$$

$$M_y = -2700 \text{ Nm}$$

$$M_z = 2025 \text{ Nm}$$

No marks for these

$$I_y = I_z = \frac{\pi}{4} (60^4 - 50^4) = 5.27 \times 10^6 \text{ mm}^4$$

$$J = \frac{\pi}{2} (60^4 - 50^4) = 10.54 \times 10^6 \text{ mm}^4$$

1 mark

$$(Q_1)_B = 0$$

$$(Q_2)_B = \frac{4}{3\pi} \sigma_o \left(\frac{\pi}{2} \sigma_o^2 \right) - \frac{4}{3\pi} \sigma_i \left(\frac{\pi}{2} \sigma_i^2 \right) = 6.067 \times 10^4 \text{ mm}^3 \quad (1 \text{ mark})$$

Normal Stress:

$$\sigma_B = - \frac{M_z y}{I_z} = -19.21 \text{ MPa} \quad \left(1 \text{ mark for the -ve sign or writing compressive stress} \right)$$

(5 marks for σ_B magnitude)

Shear Stress:

$$\text{Due to torsion: } \tau_{xz}|_T = \frac{T \sigma_i}{J} = 4.269 \text{ MPa} \quad (3 \text{ marks})$$

$$\text{Due to transverse load } \tau_{xz}|_V = \frac{V_z Q_2}{I_y t} = 1.036 \text{ MPa} \quad (3 \text{ marks})$$

Both these shear stresses are in the same dirⁿ.

$$\therefore \tau_{xz}|_{\text{total}} = \tau_{xz}|_T + \tau_{xz}|_V = 5.305 \text{ MPa}$$

2 marks for the addition

4. A thin annular disk of inner radius r_i and outer radius r_o is fixed at its outer boundary to a rigid support. However axial displacement is not constrained. At the inner boundary, there is a force (per unit area) f acting radially inward. Starting from the governing differential equation for radial displacement in the form: $\frac{d^2 u}{dr^2} + \frac{1}{r} \frac{du}{dr} - \frac{u}{r^2} = 0$, and using the appropriate boundary conditions, determine the radial stress at the outer boundary. [12 marks]

Let $u = r^m$. Substituting in the eqn and solving gives us:

$$u = C_1 r + C_2 \frac{1}{r} \quad \text{--- (1 mark)}$$

$$\text{@ } r = r_o, u = 0 \Rightarrow C_1 r_o + C_2 \frac{1}{r_o} \Rightarrow C_2 = -C_1 r_o^2$$

Now, this is a plane stress case. So,

$$\begin{aligned} \sigma_{rr} &= \frac{E}{1-\nu^2} \left[\epsilon_{rr} + \nu \epsilon_{\theta\theta} \right] \\ &= \frac{E}{1-\nu^2} \left[\frac{du}{dr} + \nu \frac{u}{r} \right] \\ &= \frac{E}{1-\nu^2} \left[C_1 - \frac{C_2}{r^2} + \nu \left(C_1 + \frac{C_2}{r^2} \right) \right] \end{aligned}$$

(6 marks deducted for those who wrote the formula for plane strain. I had said on the last day of class that I would give 0 to those who make this mistake. But I changed my mind.)

$$@ \sigma = \sigma_i, \sigma_{\sigma\sigma} = f$$

$$\Rightarrow \frac{E}{1-\nu^2} \left[c_1(1+\nu) - \frac{c_2}{\sigma_i^2} (1-\nu) \right] = f$$

$$\Rightarrow \frac{E}{1-\nu^2} \left[c_1(1+\nu) + \frac{c_1 \sigma_0^2}{\sigma_i^2} (1-\nu) \right] = f$$

$$\Rightarrow c_1 = \frac{f}{\frac{E}{1-\nu^2} \left[1+\nu + \frac{\sigma_0^2}{\sigma_i^2} (1-\nu) \right]}$$

$$\sigma_{\sigma\sigma} \Big|_{\sigma=\sigma_0} = \frac{E}{1-\nu^2} \left[c_1(1+\nu) - \frac{c_2}{\sigma_0^2} (1-\nu) \right]$$

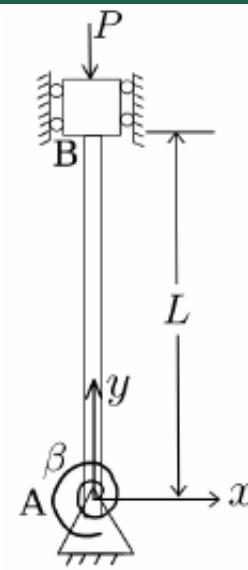
$$= \frac{E}{1-\nu^2} \left[c_1(1+\nu) + \frac{c_1 \cancel{\sigma_0^2}}{\cancel{\sigma_0^2}} (1-\nu) \right]$$

$$= \frac{E}{1-\nu^2} [2c_1] = \frac{2f}{1+\nu + \frac{\sigma_0^2}{\sigma_i^2} (1-\nu)}$$

Serious penalty for those who wrote wrong BLS, especially who wrote something like

$$\sigma_{\sigma\sigma} = -P_0 @ \sigma = \sigma_0.$$

5. Consider a column with constant flexural rigidity (EI) as shown in the figure. The bottom end A ($x = 0$) is pinned and attached with a rotational spring of stiffness β . The top end B ($x = L$) is fixed to the vertical guide. Showing all the steps involving the solution of the 4th-order differential equation together with the use of the proper boundary conditions, determine the critical condition(s) for buckling. [12 marks]



[In the figure the labels of the x and y axes were interchanged by mistake.]

The general solution of the 4th order differential eqn, $\frac{d^4 y}{dx^4} + k^2 \frac{d^2 y}{dx^2} = 0$ is

$$y = A \sin Kx + B \cos Kx + Cx + D \quad - (1 \text{ mark})$$

$$\frac{dy}{dx} = AK \cos Kx - BK \sin Kx + C$$

$$\frac{d^2 y}{dx^2} = -AK^2 \sin Kx - BK^2 \cos Kx$$

(There are different ways of proceeding with the simplifications after writing the BCs. But the BCs are most important.)

$$@ x=0, y=0 \quad \text{--- (2 marks)}$$

$$\Rightarrow B+D=0 \quad \text{--- (1)}$$

$$@ x=0, EI \frac{d^2 y}{dx^2} = -\beta \frac{dy}{dx} \quad \text{--- (2 marks)}$$

$$\Rightarrow EI(-BK^2) = -\beta(AK+C) \Rightarrow C = \frac{EI K^2}{\beta} B - AK \quad \text{--- (2)}$$

$$@ x=L, y=0 \quad \text{--- (2 marks)}$$

$$\Rightarrow A \sin KL + B \cos KL + CL + D = 0 \quad \text{--- (3)}$$

$$@ x=L, \frac{dy}{dx} = 0 \quad \text{--- (2 marks)}$$

$$\Rightarrow AK \cos KL - BK \sin KL + C = 0$$

$$\Rightarrow AK \cos KL - BK \sin KL + \frac{EI K^2}{\beta} B - AK = 0 \quad (\text{Using (2)})$$

$$\Rightarrow AK(\cos KL - 1) = BK \left(\sin KL - \frac{EI K}{\beta} \right) \quad \text{--- (4)}$$

Using ① and ② in ③, we have:

$$A \sin KL + B \cos KL + L \left(\frac{EI k^2}{\beta} B - AK \right) - B = 0$$

$$\Rightarrow A(\sin KL - KL) = B \left(1 - \cos KL - \frac{EI k^2 L}{\beta} \right) \quad \text{--- ⑤}$$

Dividing ④ by ⑤, we obtain:

$$\frac{\cos KL - 1}{\sin KL - KL} = \frac{k \left(\sin KL - \frac{EI k}{\beta} \right)}{1 - \cos KL - \frac{EI k^2 L}{\beta}}$$

(Some students may have obtained other forms of the above critical condition. These are fine as long as they have proceeded from the correct BCs.)

I kept 3 marks for the simplification. Not many students proceeded well. But those who tried to proceed somewhat, I gave $\frac{1}{2}$, 1 or 2 marks depending on their effort.