

# spanwise\_distributions for aoa:10.0 [deg]

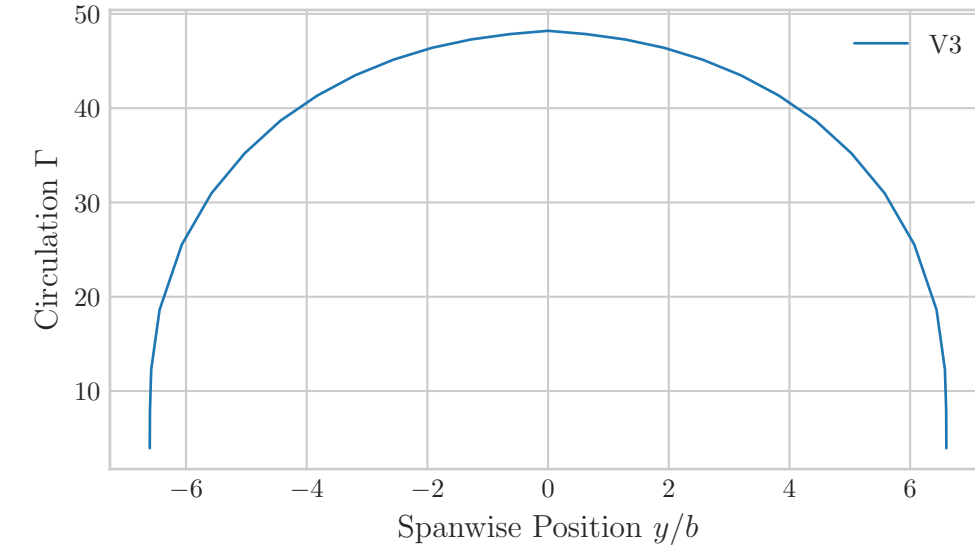
## $C_L$ Distribution



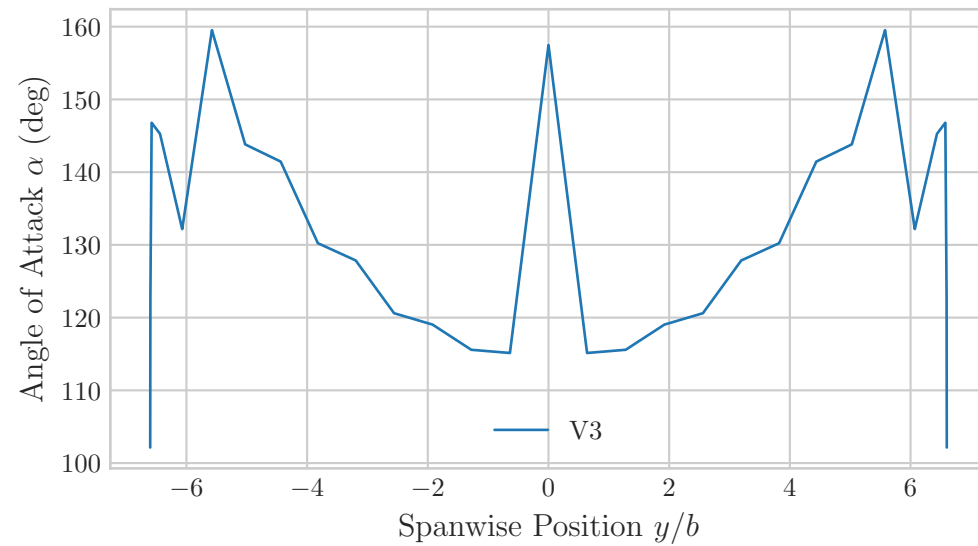
## $C_D$ Distribution



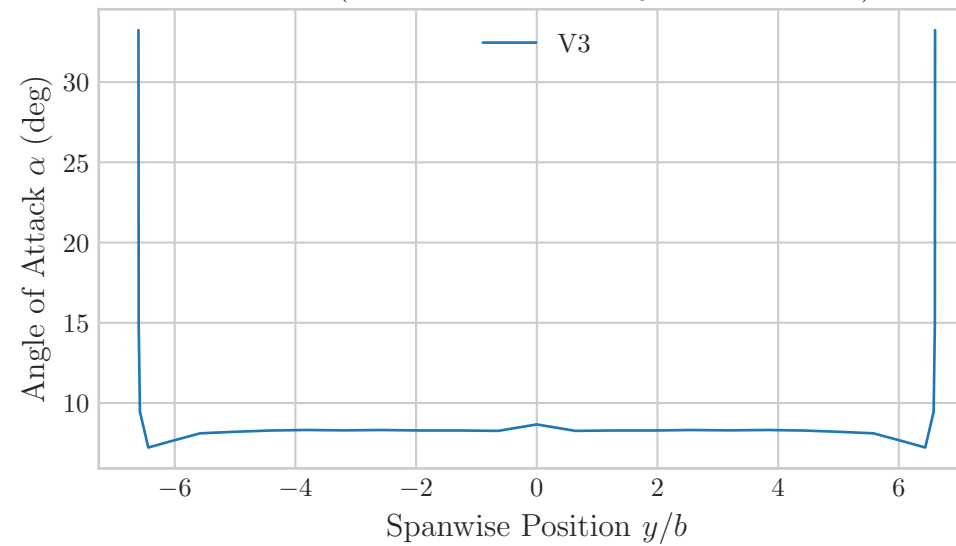
## $\Gamma$ Distribution



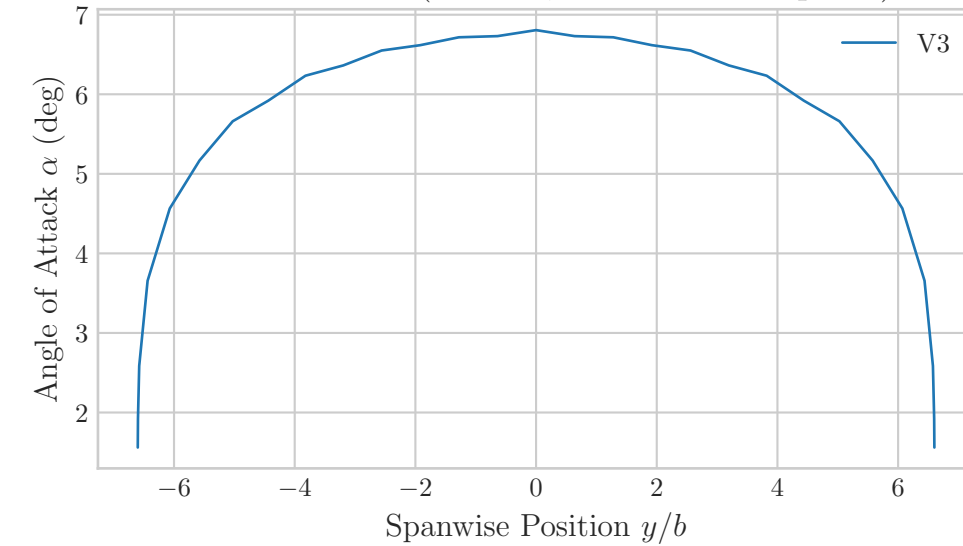
## $\alpha$ Geometric



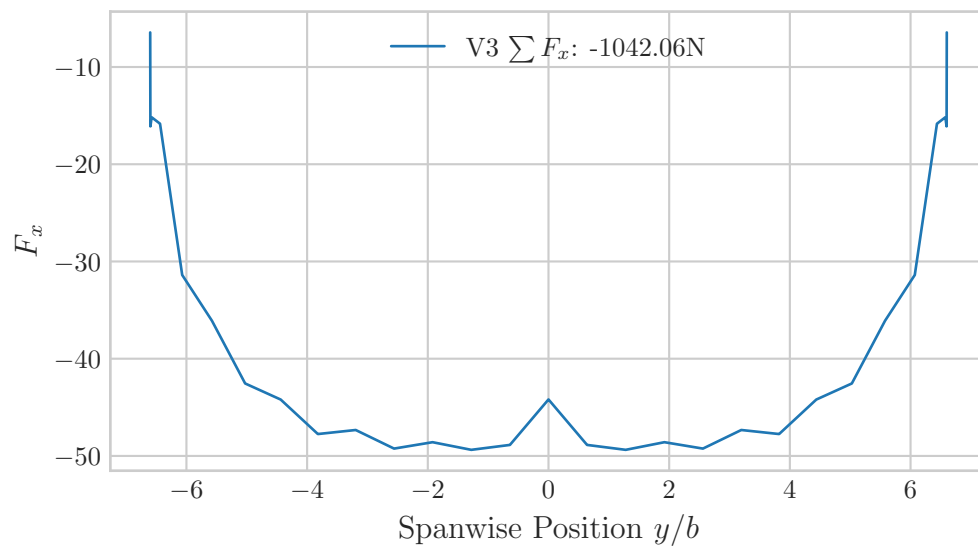
## $\alpha$ result (corrected to aerodynamic center)



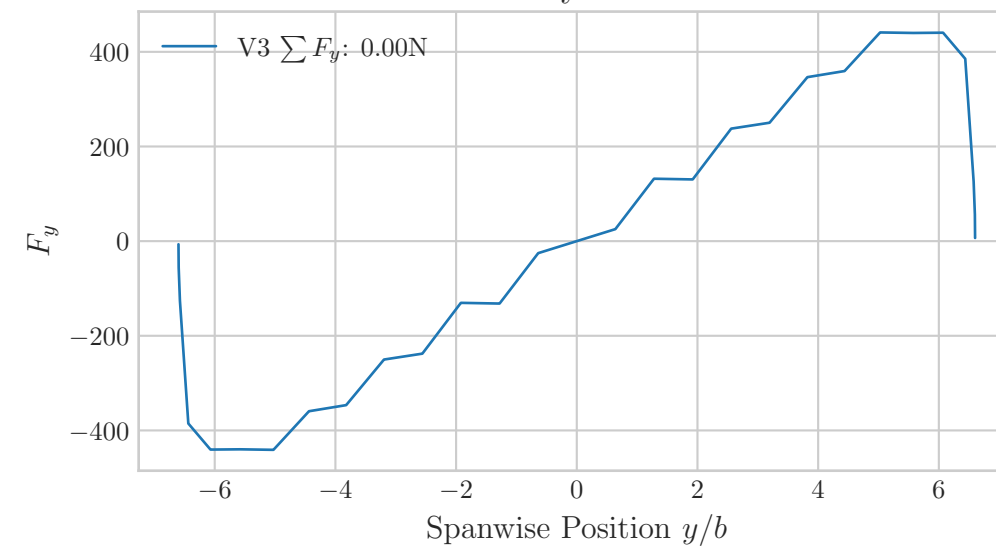
## $\alpha$ Uncorrected (if VSM, at the control point)



## Force in x direction



## Force in y direction



## Force in z direction

