

NCERT Solutions for Class 11 Physics Chapter 7 – Gravitation

1. Answer the following:

A. You Can Shield a Charge from Electrical Forces by Putting it Inside a Hollow Conductor. Can You Shield a Body from the Gravitational Influence of Nearby Matter by Putting it Inside a Hollow Sphere or by Some Other Means?

Ans: No.

The gravitational force of matter on nearby objects cannot be eliminated by any means. This is because gravitational force is independent of the nature of the medium's material.

B. An Astronaut Inside a Small Spaceship Orbiting Around the Earth Cannot Detect Gravity. If the Space Station Orbiting Around the Earth Has a Large Size, Can He Hope to Detect Gravity?

Ans: Yes.

If the space station's size is large enough, then the astronaut will identify the change in Earth's gravity.

C. If You Compare the Gravitational Force on the Earth Due to the Sun to That Due to the Moon, You Would Find That the Sun's Pull Is Greater Than the Moon's Pull. (You Can Check This Yourself Using the Data Available in the Succeeding Exercises). However, the Tidal Effect of the Moon's Pull Is Greater Than the Tidal Effect of the Sun. Why?

Ans: The tidal effect depends on the inverse of distance's cube while gravitational force depends on the inverse of distance's square.

Since the distance between the Moon and the Earth is shorter than the distance between the Sun and the Earth, the tidal influence of the Moon's pull is higher than the tidal effect of the Sun's influence.



2. Choose the Correct Alternative:

A. Acceleration Due to Gravity Increases/decreases With Increasing Altitude.

Ans: Acceleration due to gravity decreases with increasing altitude.

Acceleration due to gravity at height h is given by:

$$g_h = \left(1 - \frac{2h}{R_o}\right)g$$

Where.

R = Radius of the Earth

g = Acceleration due to gravity on the surface of the Earth

It is clear from the relation that acceleration due to gravity lowers with a height increment.

 B. Acceleration Due to Gravity Increases/decreases With Increasing Depth. (assume the Earth to Be a Sphere of Uniform Density).

Ans: Acceleration due to gravity decreases with increasing depth.

Acceleration due to gravity at depth d is given by:

$$g_d = \left(1 - \frac{d}{R_s}\right)g$$

It is clear from the relation that acceleration due to gravity lowers with a depth increment.

C. Acceleration due to gravity is independent of mass of the earth/mass of the body.

Ans: Acceleration due to gravity is independent of the mass of the body.

Acceleration due to gravity of body having mass m

is given by:

$$g = \frac{GM}{R^2}$$



Where.

G = Universal gravitational constant

M = Mass of the Earth

R = Radius of the Earth

Hence, acceleration due to gravity is not dependent on the body's mass.

D. The Formula $-GMm\left(\frac{1}{r_2} - \frac{1}{r_1}\right)$ is more/less accurate than the formula $mg(r_2 - r_1)$ for the difference of Potential Energy Between Two Points and Distance Away from the Centre of the Earth.

Ans: Gravitational potential energy of two points at r, and r,

distance away from the Earth centre is given by:

$$V(\mathbf{r}_1) = -\frac{GmM}{\mathbf{r}_1}$$
 and $V(\mathbf{r}_2) = -\frac{GmM}{\mathbf{r}_2}$

.. Difference in potential energy,

$$V = V(r_2) - V(r_1)$$

$$\Rightarrow$$
 V = -GMm $\left(\frac{1}{r_2} - \frac{1}{r_1}\right)$

Hence, this formula is more reliable than the formula $mg(r_2 - r_1)$.

3. Suppose there Existed a Planet That Went Around the Sun Twice as Fast as the Earth. What would Its Orbital Size be Compared to that of the Earth?

Ans: Time taken by the Earth to complete one revolution, T, =1 year.

Orbital radius of the Earth in its orbit, R, =1AU.

Time taken by the planet to complete one revolution,

$$T_{_{0}}=\frac{1}{2}T_{_{1}}=\frac{1}{2}$$
 year .



Orbital radius of the planet = R_a.

From Kepler's third law of planetary motion, we can write:

$$\left(\frac{R_p}{R_p}\right)^3 = \left(\frac{T_p}{T_o}\right)^2$$

$$\Rightarrow \frac{R_p}{R_n} = \left(\frac{T_p}{T_n}\right)^{\frac{2}{3}}$$

$$\Rightarrow \frac{R_{\rho}}{R_{s}} = \left(\frac{\frac{1}{2}}{1}\right)^{\frac{2}{3}}$$

$$\Rightarrow \frac{R_p}{R_s} = (0.5)^{\frac{2}{3}}$$

$$\Rightarrow \frac{R_p}{R} = 0.63$$

Hence, the orbital radius of the planet will be 0.63 times smaller than that of the Earth.

I_o, One of the Satellites of Jupiter, Has an Orbital Period of 1.769 days and the radius
of the orbit is 4.22×10⁸m. Show That the Mass of Jupiter is About One-Thousandth That
of the Sun.

Ans: Given that,

Orbital period of $I_0 = T_{I_0} = 1.769$ days = $1.769 \times 24 \times 60 \times 60$ s

Orbital radius of Io = Rto = 4.22×108m

Satellite Io is revolving around Jupiter.

Mass of the Jupiter is given by:



$$M_{J} = \frac{4\pi^{2}R_{i_{u}}^{1}}{GT_{i_{u}}^{2}}.....(i)$$

Where.

M, = Mass of Jupiter

G is the Universal gravitational constant

Orbital period of the Earth, T, = 365.25days = 365.25 x 25 x 60 x 60s

Orbital radius of the Earth, R, =1AU =1.496×1011 m

Mass of sun is given as:

$$\begin{split} &M_{s} = \frac{4\pi^{2}R_{s}^{3}}{GT_{s}^{2}}(ii) \\ &\Rightarrow \frac{M_{s}}{M_{1}} = \frac{4\pi^{2}R_{s}^{3}}{GT_{s}^{2}} \times \frac{GT_{to}^{2}}{4\pi^{2}R_{to}^{3}} = \frac{R_{s}^{3}}{R_{to}^{3}} \times \frac{T_{to}^{2}}{T_{s}^{2}} \\ &\Rightarrow \frac{M_{s}}{M_{1}} = \left(\frac{1.769 \times 24 \times 60 \times 60}{365.25 \times 24 \times 60 \times 60}\right)^{2} \times \left(\frac{1.496 \times 10^{11}}{4.22 \times 10^{8}}\right)^{3} \\ &\Rightarrow \frac{M_{s}}{M_{1}} = 1045.04 \\ &\therefore \frac{M_{s}}{M_{1}} \sim 1000 \end{split}$$

We get,

Hence, it can be concluded that the mass of Jupiter is about one-thousandth that of the Sun.

5. Let Us Assume That Our Galaxy Consists of 2.5×10¹¹ stars Each of One Solar Mass. How Long Will a Star be at a Distance of 50,000ly From the Galactic Centre to Complete One Revolution? Take the Diameter of the Milky Way to be 10⁵ly.

Ans: Mass of galaxy Milky way,

 $M = 2.5 \times 10^{11}$ solar mass Solar mass = Mass of sun = 2.0×10^{36} kg



Mass of our galaxy, $M = 2.5 \times 10^{11} \times 2 \times 10^{20}$

$$M = 5 \times 10^{41} \text{kg}$$

Diameter of Milky Way, d = 105 ly

Radius of Milky Way,

$$r = 5 \times 10^4 \text{ly}$$

$$1 \text{ ly} = 9.46 \times 10^{15} \text{ m}$$

$$r = 5 \times 10^4 \times 9.46 \times 10^{15}$$

$$\Rightarrow r = 4.73 \times 10^{20} \text{ m}$$

Since a star rotates around the galactic center of the Milky Way, its time period is given by:

$$T = \left(\frac{4\pi^2 r^3}{GM}\right)^{\frac{1}{2}} = \left(\frac{4\times (3.14)^2 \times (4.73)^3 \times 10^{60}}{6.67\times 10^{-11}\times 5\times 10^{41}}\right)$$

$$\Rightarrow$$
 T = $(125.27 \times 10^{90})^{\frac{1}{2}}$

$$\Rightarrow$$
 T = 1.12×10¹⁶ s

As we know, 1 year = $365 \times 24 \times 60 \times 60s$

We get,
$$1s = \frac{1}{365 \times 24 \times 60 \times 60}$$
 years

$$\Rightarrow 1.12 \times 10^{16} \text{ s} = \frac{1.12 \times 10^{16}}{365 \times 24 \times 60 \times 60}$$

$$1.12 \times 10^{16} \text{ s} = 3.55 \times 10^{8} \text{ years}$$

Star will take 3.55×10 years to complete one revolution.

Choose the Correct Alternative:

A. If the Zero of Potential Energy is at Infinity, the Total Energy of an Orbiting Satellite Is Negative of Its Kinetic/Potential Energy.



Ans: Kinetic energy.

The sum of kinetic energy and potential energy is the total mechanical energy. The satellite's gravitational potential energy is zero at infinity. As we know, the Earth-satellite system is a bound system; then, the satellite's total energy is negative.

Thus, at infinity, the orbiting satellite's total energy is equal to the negative of its kinetic energy.

B. the Energy Required to Launch an Orbiting Satellite Out of Earth's Gravitational Influence is More/less Than the Energy Required to Project a Stationary Object at the Same Height (as the Satellite) Out of Earth's Influence.

Ans: Less.

An orbiting satellite gets a certain amount of energy that permits it to rotate around the Earth. Its orbit gives this energy. It requires comparatively lesser energy to go out of the influence of the Earth's gravitational field than a fixed object on the Earth's surface that initially holds no energy.

Does the Escape Speed of a Body from the Earth Depend on

A. The Mass of the Body?

Ans: No.

Escape velocity of a body from the Earth is given by the relation:

$$v = \sqrt{2gR}$$

The escape speed of a body does not depend on the mass of the body.

B. The Location from Where it is Projected?

Ans: No.

Escape velocity of a body from the Earth is given by the relation:

$$v = \sqrt{2gR}$$



The escape speed of a body does not depend on the location from where it is projected.

C. The Direction of Projection?

Ans: No.

Escape velocity of a body from the Earth is given by the relation:

$$V = \sqrt{2gR}$$

The escape speed of a body does not depend on the direction of projection.

D. The Height of the Location from Where the Body is Launched?

Ans: Yes.

Escape velocity of a body from the Earth is given by the relation:

$$v = \sqrt{2gR}$$

The escape speed of a body does slightly depend on the height of the location from where the body is launched. This is because escape velocity is slightly dependent on the gravitational potential at a certain height.

8. A Comet Orbits the Sun in a Highly Elliptical Orbit. Does the Comet have a Constant

A. Linear Speed?

Ans: No.

The expression for angular momentum is:

$$L = \frac{mv^2}{r}$$

As angular momentum is constant, linear speed varies as r varies.

B. Angular Speed?

Ans: No.



The expression for angular momentum is:

 $L = mw^2r$

As angular momentum is constant, angular speed varies as r varies.

C. Angular Momentum?

Ans: Yes.

There is no external torque and thus, the angular momentum is constant.

D. Kinetic Energy?

Ans: No.

The expression for kinetic energy is:

$$K = \frac{1}{2}mv^2$$

Kinetic energy is not constant because linear speed varies.

E. Potential Energy?

Ans: No.

By law of conservation of energy, total energy is constant, and it is the sum of potential energy and kinetic energy.

As kinetic energy is not constant, potential energy is also not constant.

F. Total Energy Throughout Its Orbit? Neglect Any Mass Loss of the Comet When it Comes Very Close to the Sun.

Ans: Yes.

By law of conservation of energy, the total energy is constant.



Which of the Following Symptoms is Likely to Afflict an Astronaut in Space

A. Swollen Feet

Ans: Legs hold the whole mass in a standing place due to gravitational force.

In space, an astronaut appears weightless because of the absence of gravity. Therefore, swollen feet of a spaceman do not affect him/her in space.

B. Swollen Face

Ans: A swollen face is usually caused because of seeming weightlessness in space.

Sense organs such as the nose, eyes, ears, and mouth establish a person's face. This symptom can affect a spacewalker in space.

C. Headache

Ans: Headaches are because of mental stress.

It can influence the working of an astronaut in space.

D. Orientational Problem

Ans: Space has diverse orientations.

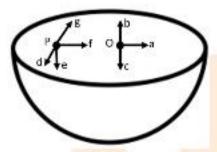
Therefore, orientational difficulty can affect an astronaut in space.

10. Choose the Correct Answer from Among the Given Ones:

The gravitational intensity at the center of a hemispherical shell of uniform mass density has the direction indicated by the arrow (see Fig 8.12).

- i. a.
- ii. b,
- iii. c,
- iv. O



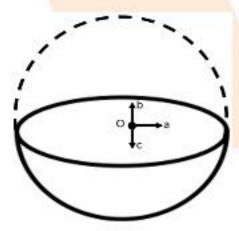


Ans: Option (iii) is correct.

Gravitational potential (V) is fixed at all points in a spherical shell. Hence, the potential gravitational gradient $\left(\frac{dV}{dr}\right)$ is zero everywhere inside the spherical surface.

The potential gravitational gradient is equivalent to the negative gravitational intensity. Hence, intensity is also zero at all locations inside the spherical shell. This shows that gravitational forces operating at a point in a spherical shell are symmetric.

If the top half of a spherical shell is cut out, then the net gravitational force working on a particle located at center O will be downward.





Since gravitational intensity is described as the gravitational force per unit mass at that location, it will also act downward. Thus, the gravitational intensity at the center of the given hemispherical shell has the direction indicated by the arrow c.

11. Choose the Correct Answer from Among the Given Ones:

For the problem 8.10, the direction of the gravitational intensity at an arbitrary point P is indicated by the arrow

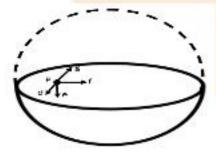
- i d.
- ii e.
- iii f.
- iv q.

Ans: Option (ii) is correct.

Gravitational potential (V) is fixed at all points in a spherical shell. Hence, the potential gravitational gradient $\left(\frac{dV}{dr}\right)$ is zero everywhere inside the spherical surface.

The potential gravitational gradient is equivalent to the negative gravitational intensity. Hence, intensity is also zero at all locations inside the spherical shell. This shows that gravitational forces operating at a point in a spherical shell are symmetric.

If the top half of a spherical shell is cut out, then the net gravitational force working on a particle located at center P will be downward.



Since gravitational intensity is described as the gravitational force per unit mass at that location, it will also act downward.



Thus, the gravitational intensity at the arbitrary point P of the given hemispherical shell has the direction indicated by the arrow e.

12. A Rocket is Fired from the Earth Towards the Sun. At What Distance from the Earth's Centre is the Gravitational Force on Rocket Zero?

Mass of the sun= 2×10^{10} kg, mass of the earth= 6×10^{24} kg. Neglect the effect of other planets etc. (orbital radius= 1.5×10^{11} m).

Ans: Mass of the Sun, M, = 2×1030 kg

Mass of the Earth, M, = 6×1024 kg

Orbital radius, r=1.5×1011m

Mass of the rocket= m



Let x be the distance from the Earth center where the gravitational force working on the satellite P becomes zero.

From Newton's law of gravitation, we can equalize gravitational forces acting on satellite P under the effect of the Sun and the Earth as:

$$\frac{GmM_s}{(r-x)^2} = Gm \frac{M_s}{x^2}$$

We get,



$$\begin{split} &\left(\frac{r-x}{x}\right)^2 = \frac{M_x}{M_x} \\ &\Rightarrow \frac{r-x}{x} = \left(\frac{2 \times 10^{30}}{60 \times 10^{24}}\right)^{\frac{1}{2}} = 577.35 \\ &\Rightarrow 1.5 \times 10^{11} - x = 577.35 x \\ &\Rightarrow 578.35 x = 1.5 \times 10^{11} \\ &\Rightarrow x = \frac{1.5 \times 10^{11}}{578.35} = 2.59 \times 10^8 \, \text{m} \end{split}$$

Therefore, 2.59×108 m is the distance from the earth's center at which the gravitational force on the rocket is zero.

 How will you 'weigh the sun', that is, estimate its mass? The mean orbital radius of the earth around the sun is 1.5×10⁸ km.

Ans: Orbital radius of the Earth around the Sun, r=1.5×10thm.

Time taken by the Earth to cover one revolution around the Sun,

T =1 year
$$\Rightarrow$$
 T=365.25 days
We get.

Universal gravitational constant, G = 6.67 × 10-11 Nm2 kg-2.

Thus, we can use the relation given below for calculating the mass of the Sun,

$$M = \frac{4\pi^{2}r^{3}}{GT^{2}}$$

$$\Rightarrow M = \frac{4\times(3.14)^{2}\times(1.5\times10^{11})^{3}}{6.67\times10^{-11}\times(365.25\times24\times60\times60)^{2}}$$

$$\Rightarrow M = \frac{133.24\times10^{13}}{6.64\times10^{4}}$$

$$\Rightarrow M = 2.0\times10^{30} \text{kg}$$



Hence, the mass of the Sun is 2×1030 kg.

14. A Saturn year is 29.5 times the earth year. How far is Saturn from the sun if the earth is 1.50×10¹ km away from the sun?

Ans: Distance of the Earth from the Sun,

$$r_a = 1.5 \times 10^8 \text{ km} = 1.5 \times 10^{11} \text{ m}$$

Time period of the Earth = T,

Time period of Saturn, T, = 29.5T,

Distance of Saturn from the Sun - r,

From Kepler's third law of planetary motion, we have,

$$T = \left(\frac{4\pi^2 r^3}{GM}\right)^{\frac{1}{2}}$$

For Saturn and Sun, we can write,

$$\frac{{r_s}^3}{{r_s}^3} = \frac{{T_s}^2}{{T_s}^2}$$

$$\Rightarrow r_{_{S}} = r_{_{S}} \left(\frac{T_{_{A}}}{T_{_{A}}} \right)^{\frac{2}{3}}$$

$$\Rightarrow r_{s} = 1.5 \times 10^{11} \left(\frac{29.5T_{e}}{T_{e}} \right)^{\frac{2}{3}}$$

$$\Rightarrow$$
 $r_a = 1.5 \times 10^{11} (29.5)^{\frac{2}{3}}$

$$\Rightarrow$$
 r_s = 1.5×10¹¹×9.55

Hence, the distance between the Sun and Saturn is 1.43×1012 m

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15. A Body weighs 63N on the surface of the earth. What is the gravitational force on it due to the earth at a height equal to half the radius of the Earth?

Ans: Weight of the body, W = 63N

Acceleration due to gravity at height h from the Earth's surface

R. - Radius of the Earth

For $h = \frac{R_b}{2}$, gravity at h is given by:

$$g_h = \frac{g}{\left(1 + \frac{h}{R_s}\right)^2}$$

$$\Rightarrow g_h = \frac{g}{\left(1 + \frac{1}{2}\right)^2}$$

$$\Rightarrow$$
 $g_h = \frac{4}{9}g$

Weight of a body of mass mat height h is given as: W' = mgh

$$\Rightarrow$$
 W' = m $\times \frac{4}{9}$ g

$$\Rightarrow$$
 W' = $\frac{4}{9}$ × mg

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Thus, the gravitational force on the body due to the earth at a height equal to half the radius of the earth is 28N.



16. Assuming the Earth to Be a Sphere of Uniform Mass Density, How Much Would a Body Weigh Halfway Down to the Centre of the Earth If it Weighed 250N on the Surface?

Ans: Weight of a body of mass m

at the Earth's surface is given by, W = mg = 250N

Body of mass m is located at depth,

$$d = \frac{1}{2}R_e$$

Where.

R, is the radius of the Earth.

Acceleration due to gravity at depth g, is given by the relation:

$$g_{d} = \left(1 - \frac{d}{R_{s}}\right)g$$

$$\Rightarrow \left(1 - \frac{R_{s}}{2 \times R_{s}}\right)g = \frac{1}{2}g$$

Weight of the body at depth d,

W' = mg

$$\Rightarrow W' = m \times \frac{1}{2}g = \frac{1}{2}mg = \frac{1}{2}W$$

$$\Rightarrow W' = \frac{1}{2} \times 250 = 125N$$

The weight of the body halfway down to the center of the earth is 125N.

17. A Rocket is fired vertically with a speed of 5kms⁻¹ from the earth's surface. How far from the earth does the rocket go before returning to the earth? Mass of the earth = 6.0×10²⁴ kg; mean radius of the earth = 6.4×10⁶ m; G = 6.67×10⁻¹¹ Nm² kg⁻².

Ans: Distance from the center of the Earth = 8×108 m



Velocity of the rocket, v = 5kms-1 = 5×103 ms-1

Mass of the Earth, M, = 6.0×1024 kg

Radius of the Earth, $R_c = 6.4 \times 10^6 \text{ m}$

Height reached by rocket mass m, is h.

At the surface of the Earth,

Total energy of the rocket = Kinetic energy + Potential energy

$$= \frac{1}{2}mv^2 + \left(\frac{-GM_*m}{R_s}\right)$$

At highest point h, v = 0 And,

Potential energy = $-\frac{GM_em}{R_e + h}$

Total energy of the rocket = $0 + \left(\frac{GM_em}{R_e + h}\right) = -\frac{GM_em}{R_e + h}$

From the law of energy conservation, we have

Total energy of the rocket at the Earth's surface = Total energy of rocket at height h.

We have,

$$\begin{split} &\frac{1}{2}mv^2 + \left(-\frac{GM_em}{R_e}\right) = -\frac{GM_em}{R_e + h} \\ &\Rightarrow \frac{1}{2}v^2 = GM_e\left(\frac{1}{R_e} - \frac{1}{R_e + h}\right) \\ &\Rightarrow \frac{1}{2}v^2 = GM_e\left(\frac{R_e + h - R_e}{R_e(R_e + h)}\right) \end{split}$$

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$$\Rightarrow \frac{1}{2}v^2 = \frac{GM_eh}{R_e(R_e + h)} \times \frac{R_e}{R_e}$$
$$\Rightarrow \frac{1}{2}v^2 = \frac{gR_eh}{R_e + h}$$

Where, $g = \frac{GM}{R^2} = 9.8 ms^{-2}$, is the acceleration due to gravity on the Earth's surface.

Clearly.

$$v^{2}(R_{e} + h) = 2gR_{e}h$$

 $\Rightarrow v^{2}R_{e} = h(2gR_{e} - v^{2})$
 $\Rightarrow h = \frac{R_{e}v^{2}}{2gR_{e} - v^{2}}$
 $\Rightarrow h = \frac{6.4 \times 25 \times 10^{12}}{100.44 \times 10^{6}}$
 $\Rightarrow h = 1.6 \times 10^{6} m$

Height achieved by the rocket with respect to the center of the Earth, H is given by:

$$H = R_a + h$$

 $\Rightarrow H = 6.4 \times 10^6 + 1.6 \times 10^6$
We get, $H = 8.0 \times 10^6 \text{ m}$

The distance from the earth is 8.0×106 m where the rocket goes before returning to the earth.

18. The escape speed of a projectile on the earth's surface is 11.2kms⁻¹. A body is projected out with thrice this speed. What is the speed of the body far away from the earth? Ignore the presence of the sun and other planets.

Ans: Projectile's escape velocity from the Earth, v = 11.2kms-1

Projection velocity of the projectile, $v_p = 3_{k_p}$

Mass of the projectile = m



Projectile's velocity far away from the Earth = v,

Projectile's total energy on the Earth $=\frac{1}{2}mv_p^2 - \frac{1}{2}mv_{eq}^2$

The projectile's gravitational potential energy far away from the Earth is zero.

Total energy of the projectile far away from the Earth $=\frac{1}{2}mv_r^2$

From the law of energy conservation, we have

$$\frac{1}{2} m v_{\mu}^{2} - \frac{1}{2} m v_{esc}^{2} = \frac{1}{2} m v_{1}^{2}$$

We get,

$$v_{t} = \sqrt{v_{\mu}^{2} - v_{esc}^{2}}$$

$$\Rightarrow v_{t} = \sqrt{(3v_{esc})^{2} - (v_{esc})^{2}}$$

$$\Rightarrow v_{t} = \sqrt{8}v_{esc}$$

$$\Rightarrow v_{t} = \sqrt{8} \times 11.2$$

$$\Rightarrow v_{t} = 31.68kms^{-1}$$

Thus, the speed of the body far away from the earth is 31.68kms 1.

19. A Satellite Orbits the Earth at a height of 400km above the surface. How much Energy must be expended to Rocket the Satellite out of the earth's gravitational influence?

Mass of the satellite = 200 kg; mass of the earth = $6.0 \times 10^{24} \text{kg}$; radius of the earth = 6.4×10^6 ; $G = 6.67 \times 10^{-11} \text{Nm}^2 \text{kg}^{-2}$.

Ans: Given that.

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Mass of the Earth, M, = 6.0×1028 kg



Mass of the satellite, m = 200kg

Radius of the Earth, R = 6.4×10° m

Universal gravitational constant, G = 6.67 × 10-11 Nm2 kg-2

Height of the satellite, $h = 400 \text{km} = 4 \times 10^5 \text{m} = 0.4 \times 10^6 \text{m}$

Total energy of the satellite at height h,

$$TE = \frac{1}{2}mv^2 + \left(\frac{-GM_em}{R_e + h}\right)$$

Orbital velocity of the satellite,

$$v = \sqrt{\frac{GM_{\alpha}}{R_{\alpha} + h}}$$

Total energy at height h,

$$TE = \frac{1}{2}m\left(\frac{GM_a}{R_a + h}\right) - \frac{GM_am}{R_a + h} = -\frac{1}{2}\left(\frac{GM_am}{R_a + h}\right)$$

The negative sign explains that the satellite is attached to the Earth. This is called the bound energy of the satellite.

The energy needed to send the satellite out of its orbit is equal to the negative of Bound energy.

Bound energy is,
$$BE = \frac{1}{2} \left(\frac{GM_*m}{R_e + h} \right)$$

Where, M, is the mass of the Earth.

R, is the radius of the Earth.

h is the height.

m is the mass of the satellite.



⇒ BE =
$$\frac{1}{2} \times \frac{6.67 \times 10^{-11} \times 6.0 \times 10^{24} \times 200}{(6.4 \times 10^6 + 0.4 \times 10^6)}$$

⇒ BE = $\frac{1}{2} \times \frac{6.67 \times 6 \times 2 \times 10}{6.4 \times 10^6}$
⇒ BE = 5.9×10^9 J

Clearly, 5.9×10°J energy must be expended to rocket the satellite out of the earth's gravitational influence.

20. Two stars each of one solar mass (= 2×10³⁰ kg) are approaching each other for a head on collision. When they are a distance 10³⁰ km, their speeds are negligible. What is the speed with which they collide? The radius of each star is 10⁴ km. Assume the stars remain undistorted until they collide. (Use the known value of G).

Ans: Mass of each star, M = 2×1030 kg

Radius of each star, R=104km=107m

Distance between the stars, r = 109 km = 1012 m

For negligible speeds, v = 0

Total energy of two stars separated at distance r is given by

$$TE = \frac{-GMM}{r} + \frac{1}{2}mv^{2}$$

$$\Rightarrow TE = \frac{GMM}{r} + 0.....(i)$$

Now, consider the case when the stars are about to collide:

Velocity of the stars = v

Distance between the centers of the stars = 2R

Total kinetic energy of both stars



$$=\frac{1}{2}Mv^2+\frac{1}{2}Mv^2=Mv^2$$

Total potential energy of both stars

$$=\frac{-GMM}{2R}$$

Total energy of the two stars

Using the law of conservation of energy, we can write:

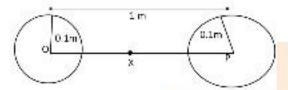
$$\begin{split} \text{Mv}^2 &- \frac{\text{GMM}}{2\text{R}} = \frac{-\text{GMM}}{\text{R}} \\ \Rightarrow \text{V}^2 &= \frac{-\text{GM}}{\text{r}} + \frac{\text{GM}}{2\text{r}} = \text{GM} \left(-\frac{1}{\text{r}} + \frac{1}{2\text{R}} \right) \\ \Rightarrow \text{V}^2 &= 6.67 \times 10^{-30} \times 2 \times 10^{20} \left[-\frac{1}{10^{12}} + \frac{1}{2 \times 10^7} \right] \\ \Rightarrow \text{V}^2 &= 13.34 \times 10^{19} \left[-10^{-12} + 5 \times 10^{-8} \right] \\ \Rightarrow \text{V}^2 &\sim 13.34 \times 10^{19} \times 5 \times 10^{-9} \\ \Rightarrow \text{V}^2 &\sim 6.67 \times 10^{12} \\ \Rightarrow \text{V} &= \sqrt{6.67 \times 10^{12}} \\ \Rightarrow \text{V} &= 2.58 \times 10^6 \text{m/s} \end{split}$$

The speed with which the two stars collide is 2.58×106 m/s.

21. Two heavy spheres each of mass 100kg and radius 0.10m are placed 1.0m apart on a horizontal table. What is the gravitational force and potential at the midpoint of the line joining the centers of the spheres? Is an object placed at that point in equilibrium? If so, is the equilibrium stable or unstable?

Ans: The situation is represented in the given figure:





Mass of each sphere, M = 100kg

Separation between the spheres, r=1m

X is the midpoint between the spheres. Gravitational force at point

X will be zero. This is because the gravitational force applied by each sphere will act in opposite directions.

Gravitational potential at point X

$$= \frac{-GM}{\left(\frac{r}{2}\right)} - \frac{GM}{\left(\frac{r}{2}\right)}$$

$$\Rightarrow PE = -4\frac{GM}{r}$$

$$\Rightarrow PE = \frac{4 \times 6.67 \times 10^{-11} \times 100}{r}$$

$$\Rightarrow PE = -2.67 \times 10^{-8} \text{ J/kg}$$

Any object placed at point X will be in equilibrium state, but the equilibrium is unstable. This is because any change in the position of the object will vary the effective force in that direction.