

Kirsch equations

Analytical displacements and stresses for plane strain/stress infinite plate with circular hole under uniaxial tension

Table 1: Nomenclature

Symbol	Description	Unit
w	half plate width	mm
a	hole radius	mm
σ	uniaxial tension	MPa
E	Young's modulus	MPa
ν	Poisson's ratio	–
$\mathbf{T} = \boldsymbol{\sigma} \cdot \mathbf{N}$	traction	MPa
\mathbf{N}	unit normal vector	–
$\boxed{\varepsilon}$	plane strain state	–
$\boxed{\sigma}$	plane stress state	–
$R \equiv \frac{a^2}{r^2}$	quadratic ratio of hole radius a to r -coordinate	–

Polar and Cartesian coordinates: (only valid for quarter plate in first quadrant: $x \geq 0, y \geq 0$)

$$r = \sqrt{x^2 + y^2} \quad \theta = \arctan \frac{y}{x} \quad (1)$$

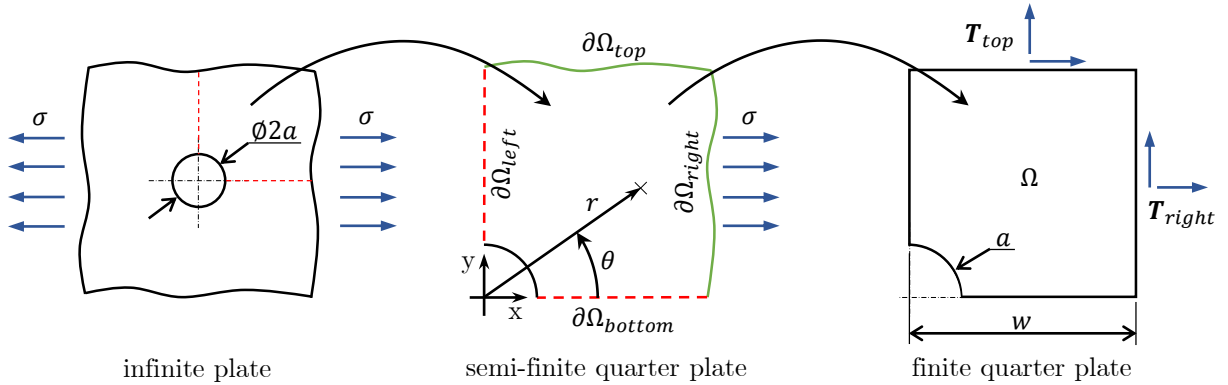


Figure 1: Benchmark problem with load (left) simplified to semi-finite quarter plate with additional boundary conditions (middle) and finite quarter plate with coupling tractions (right).

Stresses

Kirsch equations (polar stress components):

$$\sigma_{rr}^{\boxed{\varepsilon}} = \sigma_{rr}^{\boxed{\sigma}} = \frac{\sigma}{2} [1 - R + [1 + 3R^2 - 4R] \cdot \cos(2\theta)] \quad (2)$$

$$\sigma_{\theta\theta}^{\boxed{\varepsilon}} = \sigma_{\theta\theta}^{\boxed{\sigma}} = \frac{\sigma}{2} [1 + R - [1 + 3R^2] \cdot \cos(2\theta)] \quad (3)$$

$$\tau_{r\theta}^{\boxed{\varepsilon}} = \tau_{r\theta}^{\boxed{\sigma}} = -\frac{\sigma}{2} [1 - 3R^2 + 2R] \cdot \sin(2\theta) \quad (4)$$

Kirsch equations (Cartesian stress components):

$$\sigma_{xx}^{\underline{\varepsilon}} = \sigma_{xx}^{\underline{\sigma}} = \frac{\sigma}{2} \left[R^2 \left[-3 + 6 [\cos(2\theta)]^2 \right] - R [3 + 4 \cos(2\theta)] \cos(2\theta) + 2R + 2 \right] \quad (5)$$

$$\sigma_{yy}^{\underline{\varepsilon}} = \sigma_{yy}^{\underline{\sigma}} = -\frac{\sigma}{2} R \left[[\cos(2\theta)]^2 [6R - 4] + \cos(2\theta) - 3R + 2 \right] \quad (6)$$

$$\tau_{xy}^{\underline{\varepsilon}} = \tau_{xy}^{\underline{\sigma}} = \frac{\sigma}{2} R [6 \cos(2\theta) R - 4 \cos(2\theta) - 1] \sin(2\theta) \quad (7)$$

$$\sigma_{zz}^{\underline{\sigma}} = 0 \quad (8)$$

$$\sigma_{zz}^{\underline{\varepsilon}} = \nu \left[\sigma_{xx}^{\underline{\varepsilon}} + \sigma_{yy}^{\underline{\varepsilon}} \right] \quad (9)$$

von Mises stress (identically with polar stress components):

$$\sigma_{vM}^{\underline{\varepsilon}} = \sqrt{[\nu^2 - \nu + 1] [\sigma_{xx}^2 + \sigma_{yy}^2] + \sigma_{xx}\sigma_{yy} [2\nu^2 - 2\nu - 1] + 3\tau_{xy}^2} \quad (10)$$

$$\sigma_{vM}^{\underline{\sigma}} = \sqrt{\sigma_{xx}^2 + \sigma_{yy}^2 - \sigma_{xx}\sigma_{yy} + 3\tau_{xy}^2} \quad (11)$$

Displacements

Plane strain polar displacement components:

$$u_r^{\underline{\varepsilon}} = -\frac{\sigma r}{2E} \left[[-1 + R^2 + [4\nu - 4]R] \cos(2\theta) - R + 2\nu - 1 \right] [1 + \nu] \quad (12)$$

$$u_\theta^{\underline{\varepsilon}} = -\frac{\sigma r}{2E} \sin(2\theta) [1 + \nu] [R^2 + [-4\nu + 2]R + 1] \leq 0 \quad (13)$$

Plane stress polar displacement components:

$$u_r^{\underline{\sigma}} = -\frac{\sigma r}{2E} \left[[[\nu + 1]R^2 - 4R - \nu - 1] \cos(2\theta) - [\nu + 1]R + \nu - 1 \right] \quad (14)$$

$$u_\theta^{\underline{\sigma}} = -\frac{\sigma r}{2E} \left[[\nu + 1]R^2 + 2[-\nu + 1]R + \nu + 1 \right] \sin(2\theta) \leq 0 \quad (15)$$

Transformation from polar displacements to Cartesian components (identically for plane strain/stress):

$$u_x = \sqrt{u_r^2 + u_\theta^2} \cdot \cos(\theta + \phi) \quad u_y = \sqrt{u_r^2 + u_\theta^2} \cdot \sin(\theta + \phi) \quad (16)$$

$$\text{with } \phi = \begin{cases} \arctan\left(\frac{u_\theta}{u_r}\right) & \text{if } u_r > 0 \\ \arctan\left(\frac{u_\theta}{u_r}\right) + \pi & \text{if } u_r < 0 \\ \frac{3\pi}{2} & \text{if } u_r = 0 \end{cases} \quad (17)$$

Coupling tractions (on green lines, identically for plane strain/stress)

$$[(T_{right})_i] = \sigma_{ij} \cdot (N_1)_j = \begin{bmatrix} \sigma_{xx} & \tau_{xy} \\ \tau_{xy} & \sigma_{yy} \end{bmatrix} \cdot \begin{bmatrix} 1 \\ 0 \end{bmatrix} = \begin{bmatrix} \sigma_{xx} \\ \tau_{xy} \end{bmatrix} \quad \text{on } \partial\Omega_{right} \quad (18)$$

$$[(T_{top})_i] = \sigma_{ij} \cdot (N_2)_j = \begin{bmatrix} \tau_{xy} \\ \sigma_{yy} \end{bmatrix} \quad \text{on } \partial\Omega_{top} \quad (19)$$

Symmetry constraints (on red dashed lines)

$$\text{Polar constraints: } u_\theta = 0 \quad \text{on } \partial\Omega_{left} \Leftrightarrow (\theta = \pi/2) \quad u_\theta = 0 \quad \text{on } \partial\Omega_{bottom} \Leftrightarrow (\theta = 0) \quad (20)$$

$$\text{Cartesian constraints: } u_x = 0 \quad \text{on } \partial\Omega_{left} \Leftrightarrow (x = 0) \quad u_y = 0 \quad \text{on } \partial\Omega_{bottom} \Leftrightarrow (y = 0) \quad (21)$$