1 Flow Chart

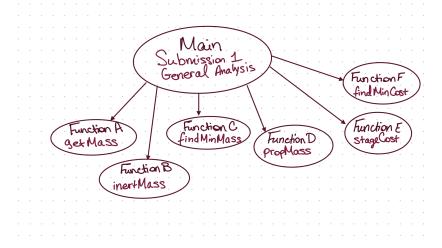


Figure 1: Flow Chart of Function Organization

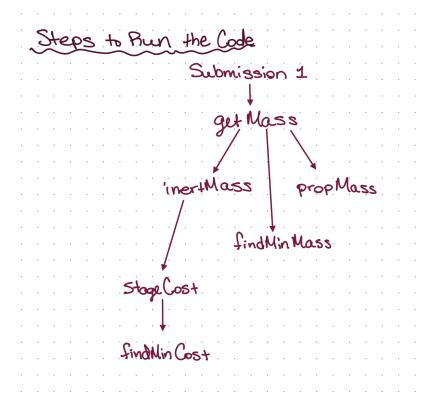


Figure 2: Flow Chart of Function Call Order

2 Code

```
% Submission 1 Template, Group 1
  close all;
  clear;
  clc;
  % Givens
  delta_v = 12.3; \% \text{ km/s}
  m_{-}pl = 26000; \% kg
  delta = 0.08; % Inert mass fraction for both stages
  chi = 0.2:0.01:0.8; \% array
  Isp1 = 327; % s, 1st stage, LOX/CH4
           LOX/CH4 LOX/LH2 LOX/RP1 Solid
                                                Storables
12
  Isp2 = [327,
                                                285]; % s, 2nd stage
                              311,
                                       269,
13
                     366,
  stage1Prop = "LOX/CH_4";
14
   stage2Prop = ["LOX/CH<sub>-4</sub>", "LOX/LH<sub>-2</sub>", "LOX/RP1", "Solid", "Storables"];
15
16
17
   for k=1:length(Isp2)
18
       [M01_array, M02_array, chi_array] = getMass(delta_v, m_pl, delta, chi, Isp1,
19
           Isp2(k));
        [m_pr1, m_pr2] = propMass(delta, M01_array, M02_array, m_pl);
20
       [m_in1, m_in2] = inertMass(delta, M01_array, M02_array);
21
       Mo = M01\_array + M02\_array;
22
23
       M01_{array} = M01_{array}./1000;
24
       M02\_array = M02\_array./1000;
25
       Mo = Mo./1000;
26
       [\min Mass, \min Index] = findMinMass(Mo);
28
       minMass
       costForMinMassS1 = stageCost(m_in1(mIndex));
30
       costForMinMassS2 = stageCost(m_in2(mIndex));
31
       costForMinMass = costForMinMassS1 + costForMinMassS2
32
       costS1 = stageCost(m_in1);
33
       costS2 = stageCost(m_in2);
34
       costTotal = costS1 + costS2;
35
       [minCost, cIndex] = findMinCost(costTotal);
36
       minCost
37
       massForMinCost = Mo(cIndex)
39
       1 = \text{round}(\text{length}(\text{chi}_{\text{array}})/4);
       figure (k)
41
       hold on;
42
       grid on;
43
       plot (chi_array (mIndex-1:mIndex+1), M02_array (mIndex-1:mIndex+1));
       plot (chi_array (mIndex-l:mIndex+l), M01_array (mIndex-l:mIndex+l));
45
       plot(chi_array(mIndex-l:mIndex+l), Mo(mIndex-l:mIndex+l));
46
       plot(chi_array(mIndex), minMass, "o");
47
       legend ('M_{02}', 'M_{01}', 'M_0', "M_{min}");
48
       titleMass = sprintf("Total Masses vs. Chi (S1: %s, S2: %s)", stage1Prop,
49
           stage2Prop(k));
       title (title Mass);
50
```

Submission 1

```
ylabel ('Mass (tonnes)');
51
       xlabel('Chi');
52
53
       figure(length(Isp2) + k)
54
       hold on;
55
       grid on;
       plot(chi_array(cIndex-l:cIndex+l), costS2(cIndex-l:cIndex+l));
57
       plot(chi_array(cIndex-l:cIndex+l), costS1(cIndex-l:cIndex+l));
       plot(chi_array(cIndex-l:cIndex+l), costTotal(cIndex-l:cIndex+l));
59
       plot(chi_array(cIndex), minCost, "o");
       \textbf{legend ('Cost_2', 'Cost_1', 'Cost_0', "Cost_{\{\min\}''\};}
61
       titleMass = sprintf("Cost vs. Chi (S1: %s, S2: %s)", stage1Prop,
62
           stage2Prop(k));
       title(titleMass);
63
       ylabel('Cost ($M)');
64
       xlabel('Chi');
65
66
  end
67
   function [M01_array, M02_array, chi_array] = getMass(delta_v,m_pl,delta,chi,
      Isp1, Isp2)
  % GETMASS Get total stage masses, M01 and M02.
  %
  %
       [M01_array, M02_array, chi_array] = GETMASS(delta_v, m_pl, delta, chi,
4
      Isp1, Isp2)
  %
       Use the rocket equation to determine total stage masses given:
  %
       - delta_v: the required change in velocity
  %
       - m_pl: the required payload mass
  %
       - delta: the estimated inert mass fraction
       - chi: the target delta V split between the first and second stage
  %
       - Isp1: the specific impulse of the first stage engines
  %
       - Isp2: the specific impulse of the second stage engines
11
  %
12
  %
       See also PROPMASS, INERTMASS, FINDMINMASS.
13
       syms M02 M01
14
       %Initialize mass arrays
15
       M02_array = [];
16
       M01_array = [];
17
       chi_array = [];
18
19
       for k = 1: length(chi)
20
           g = 9.81/1000;\% \text{km/s}^2
           %Rocket equation for stage two
22
           eqn2 = -Isp2*g*log((m_pl+delta*M02) / M02) - (1-chi(k))*delta_v == 0;
           soln2 = vpasolve(eqn2, M02); %Use vpa solver to find M02
24
           The below loop excludes mass values that are negative and
           %populates the matrices for second stage
26
           if soln2 > 0 \%
27
                try
28
                    M02\_array(end+1) = double(soln2); %Store mass values
29
                    chi_array(end+1) = [chi(k)]; %Store corresponding chi values
30
                catch
31
                end
32
           end
33
```

Submission 1

```
end
34
35
       %Repeat same steps but for first stage
36
       for k = 1:length(chi_array)
37
           g = 9.81/1000; %km/s<sup>2</sup>
38
           %Rocket equation for stage one
           eqn1 = -Isp1*g*log((M02\_array(k)+delta*M01) / (M02\_array(k)+M01)) -
40
               chi_array(k)*delta_v = 0;
           soln1 = vpasolve(eqn1, M01);
41
           if soln1 > 0
                try
43
                    M01\_array(end+1) = double(soln1); %Store mass values
                catch
45
                end
46
           end
47
       end
48
       The below loop scales the chi array to the mass array with lower length
49
       if length (M02_array) > length (M01_array)
50
           chi_array = chi_array(1:length(M01_array)); %Scale chi to M01
51
           M02\_array = M02\_array(1:length(M01\_array));%Scale M02 to M01
52
       elseif length (M01_array) > length (M02_array)
53
           chi_array = chi_array(1:length(M02_array)); %Scale chi to M02
54
           M01\_array = M01\_array(1:length(M02\_array));%Scale M01 to M02
       end
56
57
  end
58
  function [m_in1, m_in2] = inertMass(delta, M01_array, M02_array)
  % INERTMASS Find the inert mass given the inert mass fraction and the stage
      masses.
  %
  %
       [m_in1, m_in2] = INERTMASS(delta, M01_array, M02_array) Finds the
4
  %
       inert mass of stage 1 (from M01_array) and the inert mass of stage 2
  %
       (from M02_array) using inert mass fraction delta.
  %
  %
       See also GETMASS, PROPMASS, FINDMINMASS.
       m_in1 = delta.*M01_array;
       m_in2 = delta.*M02_array;
10
  end
11
  function [m_pr1, m_pr2] = propMass(delta, M01, M02, m_pl)
  % PROPMASS Find the propellant masses from the total stage masses.
  %
3
  %
       [m_pr1, m_pr2] = PROPMASS(delta, M01, M02, m_pl) Find propellant masses
4
  %
       for stage 1 and 2 (m_pr1 and m_pr2), from total stage masses for stage 1
5
      and 2 (M01 and
  %
       M02), using inert mass fraction (delta) and payload mass (m_pl).
6
  %
       See also GETMASS, INERTMASS, FINDMINMASS.
       m_{pr1} = [];
       m_{pr2} = [];
10
       for k = 1: length (M01)
11
           m_in1 = delta*M01(k);
12
           m_{pr1}(k) = M01(k) - m_{in1};
13
```

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Submission 1

```
m_in2 = delta*M02(k);
14
           m_{pr}^{2}(k) = M02(k) - m_{pl} - m_{in}^{2};
15
       end
16
  end
17
  function [mass, index] = findMinMass(Mos)
  % FINDMINMASS Find the minimum mass given a list of total vehicle masses.
  %
  %
       [mass, index] = FINDMINMASS(Mos) Finds the minimum mass in Mos.
  %
  %
       See also FINDMINCOST.
      % call min function to find the min total mass, return the mass value
      % and index
10
       [\max, index] = \min(Mos);
11
  end
12
  function cost = stageCost(stageInertMass)
  % STAGECOST Find the cost of a stage based on inert mass.
  %
  %
       [cost] = STAGECOST(stageInertMass) Find the cost of a stage in millions
  %
       of 2025 dollars based on the stageInertMass
  %
  %
       See also FINDMINCOST, INERTMASS.
       cost = 13.52.*stageInertMass.^0.55;
  end
  function [cost, index] = findMinCost(costs)
  % FINDMINCOST Find the minimum cost given a list of total vehicle costs.
  %
  %
       [cost, index] = FINDMINCOST(costs) Finds the minimum cost in costs.
  %
  %
       See also FINDMINMASS, STAGECOST.
       [\cos t, index] = \min(\cos ts);
  end
```