

# MEMS1045

## Automatic control

Lecture 10

Root Locus 2



# Objectives

- Describe the common industrial controllers and their transfer functions
- Explain controller design process using root locus
- Design controller using root locus to meet given time response specifications

# Design specifications

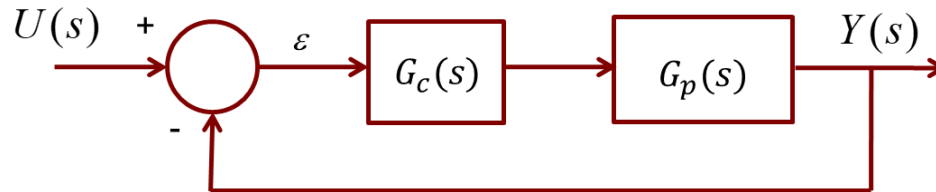
For practical applications, the system time response needs to meet certain design specifications, which can be classified into:

- ❖ Steady state response specifications such as steady state errors
- ❖ Transient response specifications such as percentage overshoot (or damping ratio), settling time, peak time, rise time, etc.

If the response does not meet these specifications, possible solutions include:

- ❖ modification of the plant dynamics, which may not be possible in many practical situations because the plant may be fixed and not modifiable
- ❖ Use a proportional controller and adjust the gain to change the location of the closed-loop poles using the root locus techniques. In many cases, the root-locus plot of a system may indicate that the desired performance cannot be achieved just by the adjustment of gain. In some cases, the system may not be stable for all values of gain
- ❖ Use other controllers that modify the root locus by adding poles and zeros

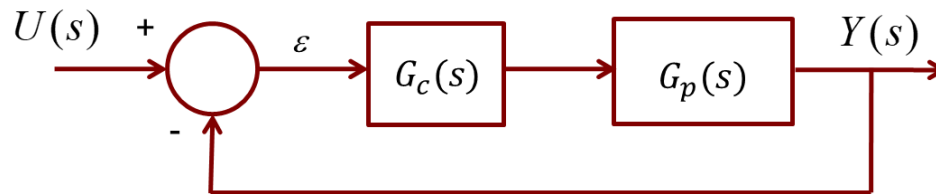
# Common industrial controllers



Commonly used industrial controllers include the following:

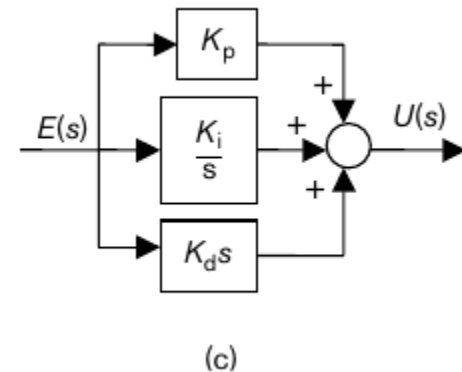
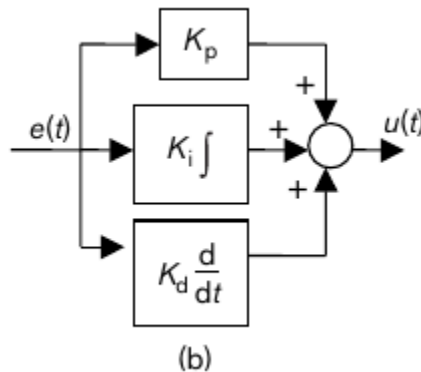
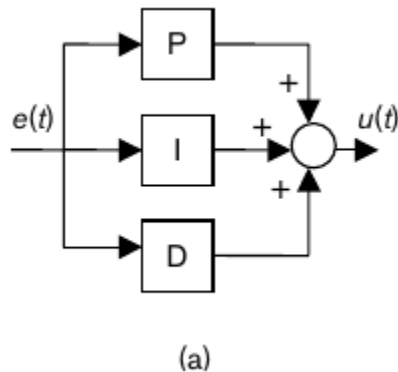
- a) On-off controller; (not considered in this course)
- b) Proportional controller (P); (this was discussed in last lecture)
- c) Proportional and integral controller (PI);
- d) Proportional and derivative controller (PD);
- e) Proportional, integral, and derivative controller (PID);
- f) Lead compensator;
- g) Lag compensator;
- h) Lead-lag compensator

# Design by root locus



- ❖ The design by the root-locus method is based on reshaping the root locus of the system by adding poles and zeros to the system's open-loop transfer function and forcing the root loci to pass through desired closed-loop poles in the  $s$  plane
- ❖ The concept is based on the using the dominant closed-loop poles to approximate the response characteristics
- ❖ P controller transfer function:  $G_c(s) = k_p$  does not change the open-loop poles or zeros and confined to the path of the root locus
- ❖ The controllers we will discussed add open-loop poles and zeros to reshape the root locus

# Controller transfer functions



- ❖ P controller transfer function:  $G_c(s) = k_p$
- ❖ PI controller transfer function:  $G_c(s) = k_p + \frac{k_I}{s} = \frac{k_p s + k_I}{s}$
- ❖ PD controller transfer function:  $G_c(s) = k_p + k_D s$
- ❖ PID controller transfer function:  $G_c(s) = k_p + \frac{k_I}{s} + k_D s = \frac{k_D s^2 + k_p s + k_I}{s}$
- ❖ Lead compensator:  $G_c(s) = \frac{(s+z)}{(s+p)}$  where  $|z| < |p|$
- ❖ Lag compensator:  $G_c(s) = \frac{(s+z)}{(s+p)}$  where  $|z| > |p|$

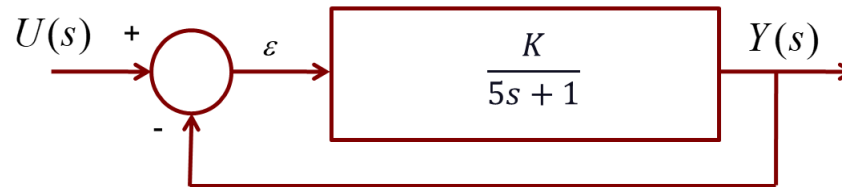
# Typical controller characteristics

- ❖ Proportional control: speeds up response but results in steady state error
- ❖ Integral control: eliminates steady state error but produces oscillating response; It increases the order of the closed-loop system and can introduce stability concerns
- ❖ Derivative control: adds damping and improves the transient response
- ❖ Lead compensator: It tends to shift the root locus toward the left half plane. This results in an improvement in the system's stability and an increase in the response speed
- ❖ Lag compensator: It tends to shift the root locus to the right, which is undesirable. For this reason, the pole and zero of a lag compensator must be placed close together (usually near the origin) so they do not appreciably change the transient response or stability characteristics of the system
- ❖ Lead-lag compensator: first design the lead compensator to achieve the desired transient response and stability, and then add on a lag compensator to improve the steady-state response

# Example 1

Given the system shown, design a controller with the following specifications:

- ❖ Zero steady-state error to a unit step input
- ❖ Settling time  $< 4\text{sec.}$ ; percent overshoot  $< 25\%$



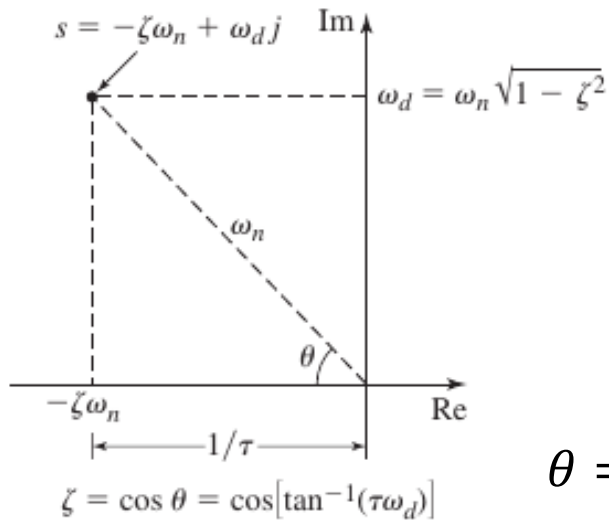
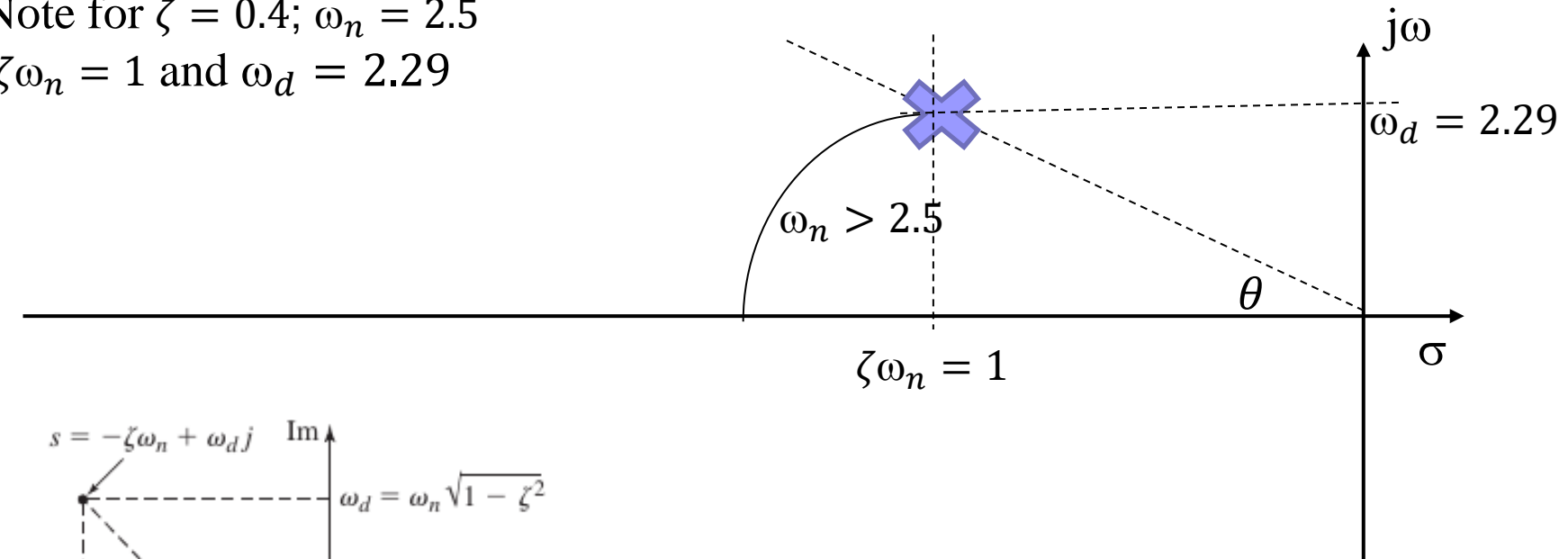
Note: system is type 0 and will not meet steady error specs (must change to type 1)

- ❖ For percent overshoot  $< 25\%$ ,  $\zeta = \frac{-\ln(\%OS/100)}{\sqrt{\pi^2 + \ln^2(\%OS/100)}} > 0.4$
- ❖ For settling time  $< 4\text{ sec.}$ ,  $T_s = \frac{4}{\zeta\omega_n}$  or  $\omega_n > 2.5$
- ❖ The original closed loop system is 1<sup>st</sup> order and would meet the OS% specs
- ❖ The closed-loop pole is located at  $s = \frac{-1-k}{5}$ ;  $K > 4$  to meet  $T_s$  specs



# Example 1

Note for  $\zeta = 0.4$ ;  $\omega_n = 2.5$   
 $\zeta\omega_n = 1$  and  $\omega_d = 2.29$



$$\theta = \cos^{-1} \zeta = 66.4^\circ$$

# Example 1

Plot root locus of original system using open loop transfer function

$$G(s) = \frac{K}{5s + 1}$$

Step 1: Locate the open-loop poles and zeros in s-plane

- ❖ Note:  $n = 1$  branch; starts from  $K = 0$  at open-loop pole  $s = -0.2$
- ❖ Note:  $m = 0$  (there are  $(n - m = 1)$  asymptote as  $K \rightarrow \infty$  on real axis

Step 2: Determine the root loci occupying the real axis

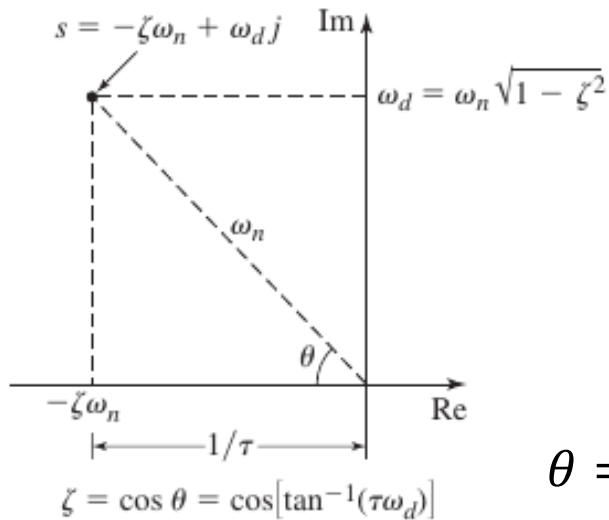
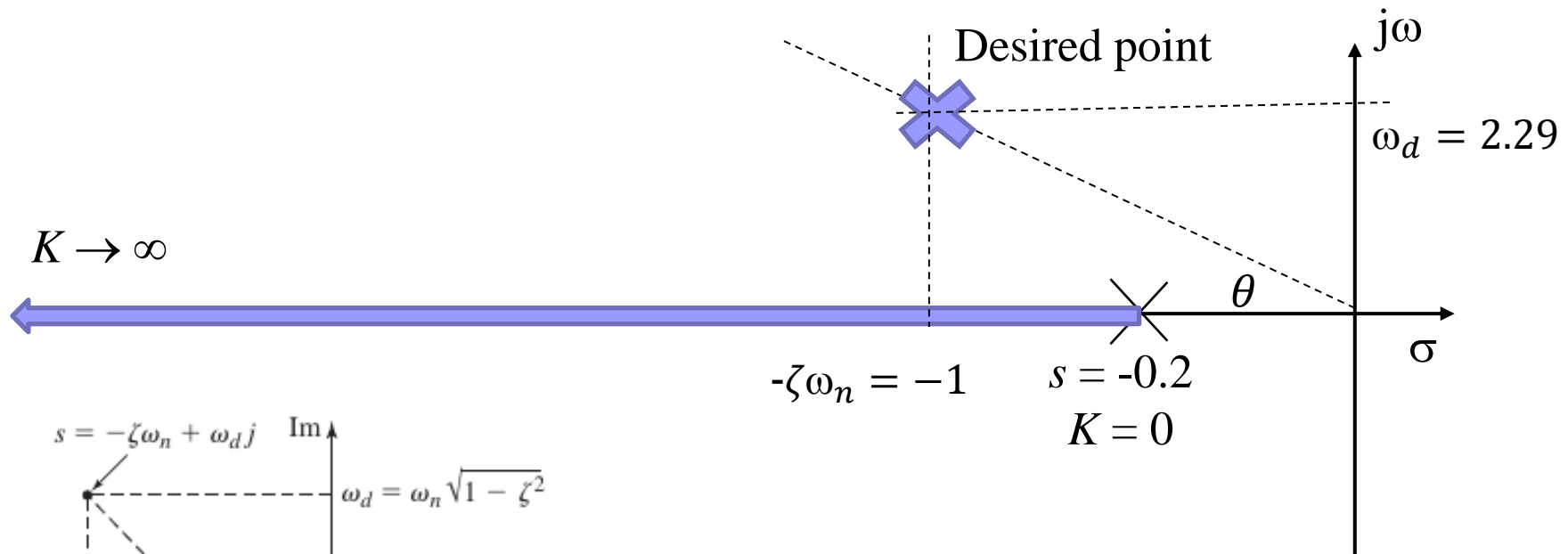
- ❖ It will occupy the real axis from  $-0.2$  to  $-\infty$

Note: characteristics equation is  $5s + 1 + K = 0$

Step 3: Determine the asymptotes of the root loci (if any) – no need

Step 4: Locate the breakaway or break-in points (if any) – no need

# Example 1



How to reshape the root locus to pass through the desired point?

$$\theta = \cos^{-1} \zeta = 66.4^\circ$$

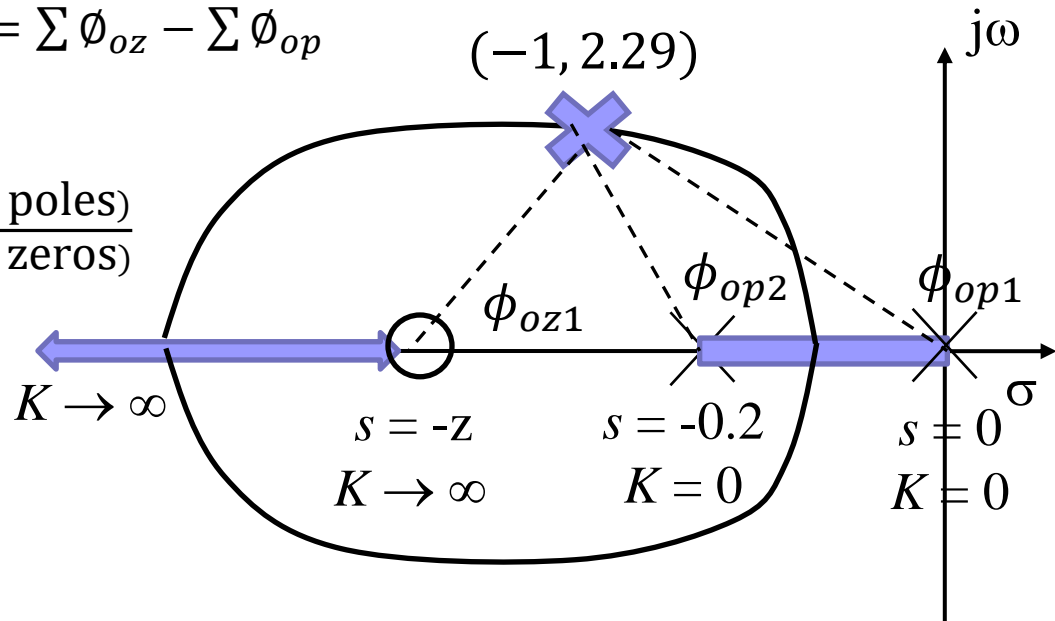
# Example 1

Angle criterion:  $180^\circ(2k + 1) = \sum \phi_{oz} - \sum \phi_{op}$

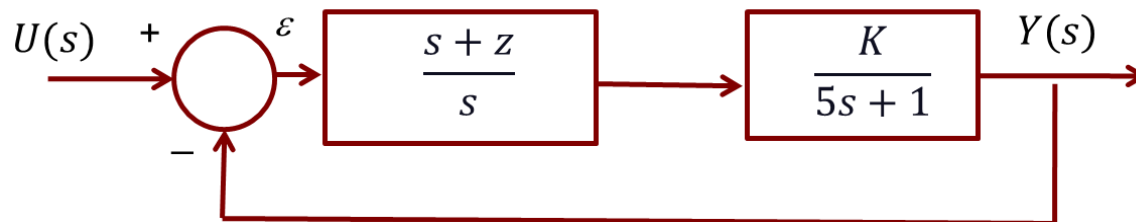
Where  $k = 0, \pm 1, \pm 2, \pm 3, \dots$

Magnitude criterion:

$$K = \frac{\prod(\text{lengths from } s_0 \text{ to OLTF poles})}{\prod(\text{lengths from } s_0 \text{ to OLTF zeros})}$$



Add a PI (one pole at origin and another zero and then adjust the gain  $K$ )



# Example 1

- ❖ Use angle criterion to find zero for branch to pass  $s = -1 + j2.29$   
 $180^\circ(2k + 1) = \sum \phi_{oz} - \sum \phi_{op} = (\phi_{oz1} + \dots + \phi_{ozm}) - (\phi_{op1} + \dots + \phi_{opn})$  where  $k = 0, \pm 1, \pm 2, \pm 3, \dots$
- ❖  $\phi_{op1} = 180^\circ - \tan^{-1}(2.29/1) = 113.6^\circ$
- ❖  $\phi_{op2} = \tan^{-1}(2.29/(1 - 0.2)) = 109.26^\circ$
- ❖ Hence  $\phi_{oz1} - (\phi_{op1} + \phi_{op2}) = 180^\circ(2k + 1)$
- ❖  $\phi_{oz1} - 222.86^\circ = 180^\circ$ ; hence  $\phi_{oz1} = 42.86^\circ$
- ❖ Hence  $\tan(42.86^\circ) = 2.29/(z - 1)$ , or  $z = 3.47$

The controller is  $(s + 3.47)$

To find the gain  $K$ , we use the magnitude criterion:

$$K = \frac{\prod(\text{lengths from } s_0 \text{ to OLTF poles})}{\prod(\text{lengths from } s_0 \text{ to OLTF zeros})}$$

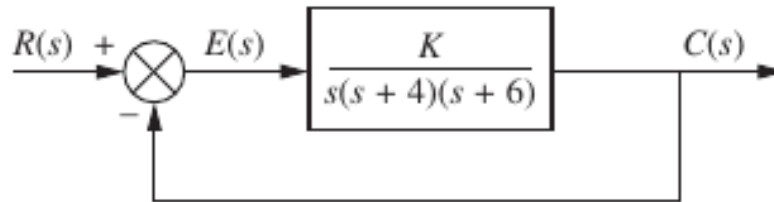
# Example 1

- ❖ The point is at  $s_0 = -1 + j2.29$
- ❖  $L_{op1} = \sqrt{1^2 + 2.29^2} = 2.5$
- ❖  $L_{op2} = \sqrt{0.8^2 + 2.29^2} = 2.43$
- ❖  $L_{oz1} = \sqrt{2.47^2 + 2.29^2} = 3.37$
- ❖  $K = \frac{\prod(\text{lengths from } s_0 \text{ to OLTf poles})}{\prod(\text{lengths from } s_0 \text{ to OLTf zeros})} = 1.8$
- ❖ The response of the compensated system should be checked and refined if needed for the compensated system



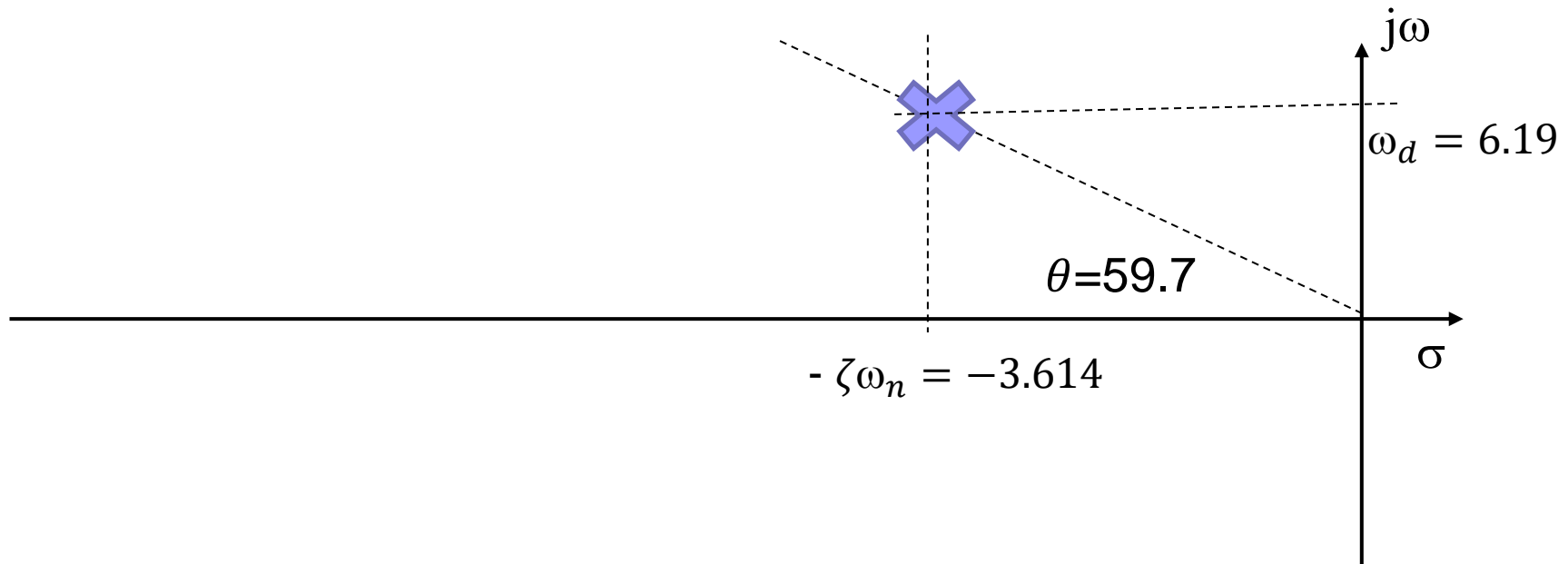
# Example 2

Given the system shown, the gain  $K$  is tuned to 43.35 so that the settling time is 3.32 sec. with 16% overshoot. Design a controller so that settling time is reduced by threefold with percent overshoot remaining at 16%



- ❖ For percent overshoot = 16%,  $\zeta = \frac{-\ln(\%OS/100)}{\sqrt{\pi^2 + \ln^2(\%OS/100)}} = 0.504$
- ❖  $\theta = \cos^{-1} \zeta = 59.7^\circ$
- ❖ For settling time = 3.32 sec.,  $T_s = \frac{4}{\zeta\omega_n}$  or  $\zeta\omega_n = 1.205$
- ❖ To reduce  $T_s$  to 1.1067 sec., new  $\zeta\omega_n = 3.614$

# Example 2



To meet the specs, the new root locus must pass through the point:

- ❖ New  $\zeta\omega_n = 3.614 =$  real part with  $\zeta = 0.504$  and  $\omega_n = 7.17$
- ❖ New  $\omega_d = \omega_n\sqrt{1 - \zeta^2} = 6.19$



# Example 2

Plot root locus of original system using open loop transfer function

$$G(s) = \frac{K}{s(s+4)(s+6)}$$

Step 1: Locate the open-loop poles and zeros in s-plane

- ❖ Note:  $n = 3$  branches; start from  $K = 0$  at open-loop poles  $s = 0$ ,  $s = -2$  and  $s = -6$
- ❖ Note:  $m = 0$  (there are  $(n - m = 3)$  asymptotes as  $K \rightarrow \infty$  with 1 on real axis (need to determine 2 more asymptotes))

Step 2: Determine the root loci occupying the real axis

- ❖ It will occupy the real axis from 0 to -2 and from -6 to  $-\infty$

Note: characteristics equation is  $s^3 + 10s^2 + 24s + K = 0$

Step 3: Determine the asymptotes of the root loci

- ❖  $\sum P_O = -10$ ,  $\sum Z_O = 0$ ,  $\bar{x} = \frac{\sum P_O - \sum Z_O}{n-m} = -3.33$ ,  $\theta = \frac{(2b+1)180^\circ}{n-m} = \pm 60^\circ$

# Example 2

Step 4: Locate the breakaway or break-in points

- ❖ breakaway will occur between 0 and -4
- ❖ Get characteristic equation, differentiate, equate to zero to solve for  $s$ :

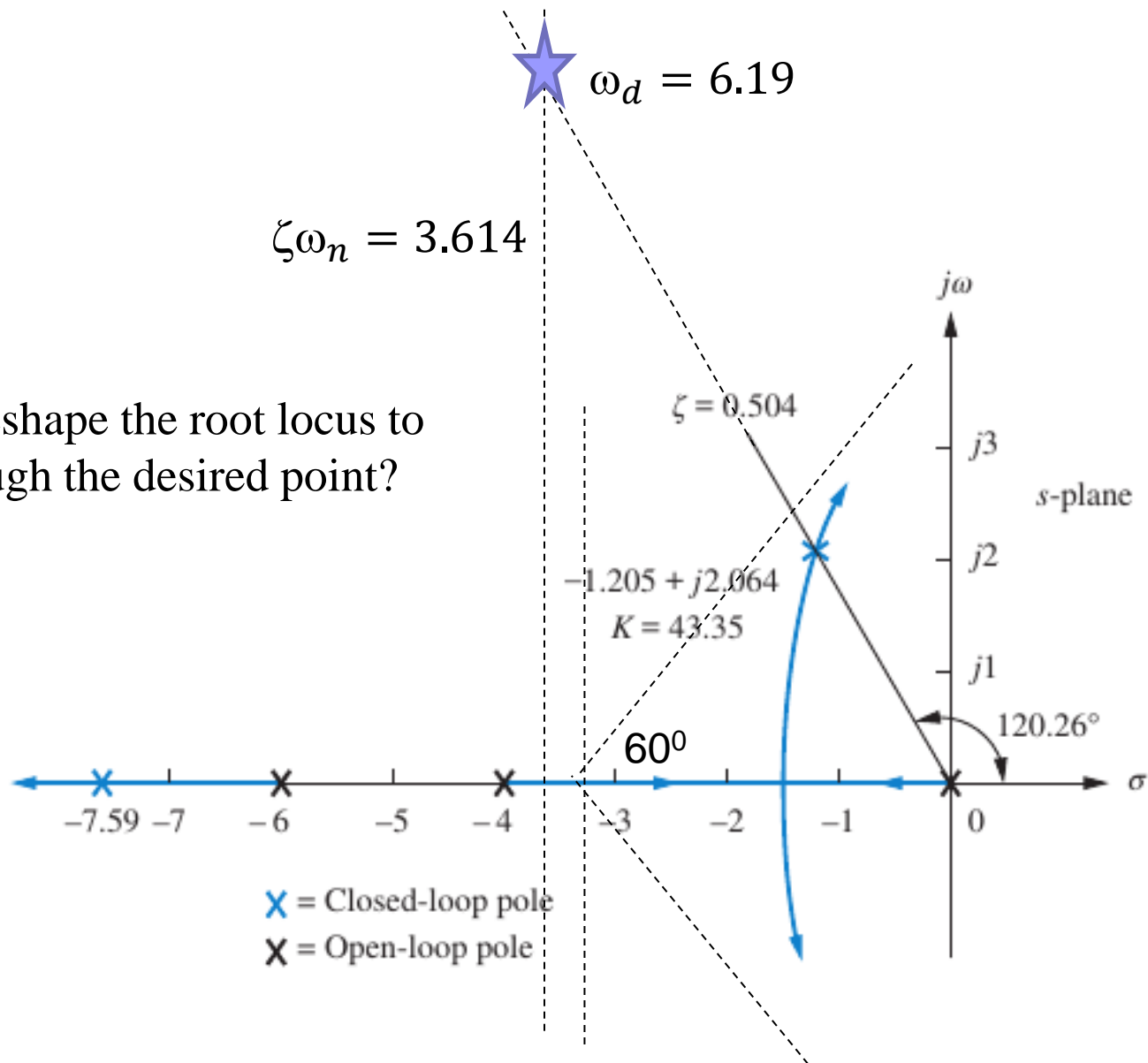
$$\frac{\partial K}{\partial s} = \frac{\partial}{\partial s} f(s) = \frac{\partial}{\partial s} (-s^3 - 10s^2 - 24s) = -2s^2 - 20s - 24 = 0 \text{ or at}$$

$s = -1.39$ ; Note no break-n or breakaway at  $s = -8.605$

Note: when gain is tuned to  $K=43.35$ , the settling time = 3.32 sec with

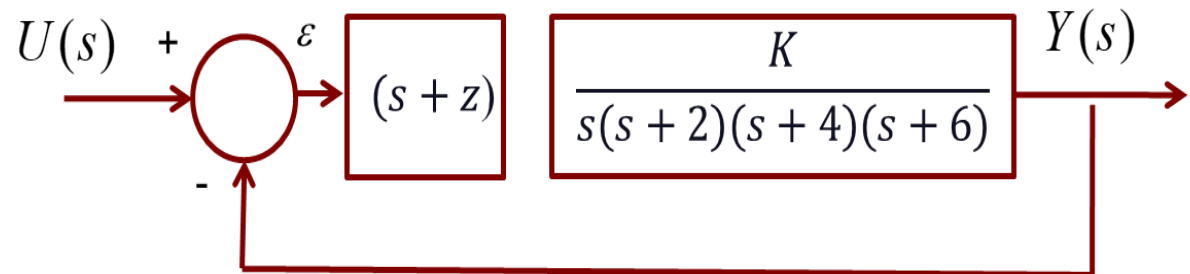
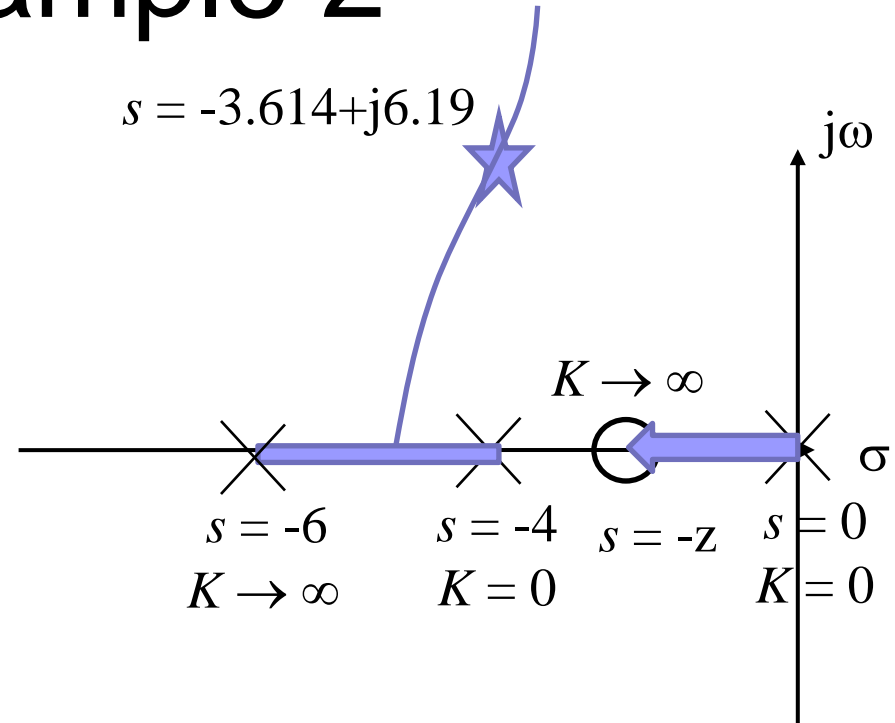
- ❖  $\zeta\omega_n = 1.205 = \text{real part with } \zeta = 0.504 \text{ and } \omega_n = 2.39$
- ❖  $\omega_d = \omega_n\sqrt{1 - \zeta^2} = 2.064$

How to reshape the root locus to pass through the desired point?



# Example 2

Add a PD. How to determine zero location so that the root locus to pass through the desired point?



# Example 2

❖ Use angle criterion to find zero for branch to pass  $s = -3.614 + j6.19$   
 $180^\circ(2k + 1) = \sum \phi_{oz} - \sum \phi_{op} = (\phi_{oz1} + \dots + \phi_{ozm}) - (\phi_{op1} + \dots + \phi_{opn})$  where  $k = 0, \pm 1, \pm 2, \pm 3, \dots$

❖  $\phi_{op1} = 180^\circ - \tan^{-1}(6.19/3.614) = 120.28^\circ$

❖  $\phi_{op2} = \tan^{-1}(6.19/(4 - 3.614)) = 86.43^\circ$

❖  $\phi_{op3} = \tan^{-1}(6.19/(6 - 3.614)) = 68.92^\circ$

❖ Hence  $\phi_{oz1} - (\phi_{op1} + \phi_{op2} + \phi_{op3}) = 180^\circ(2k + 1)$

❖  $\phi_{oz1} - 275.63^\circ = 180^\circ$ ; hence  $\phi_{oz1} = 95.63^\circ$

❖ Hence  $\tan(180 - 95.63) = 6.19/(3.614 - z)$ , or  $z = -3$

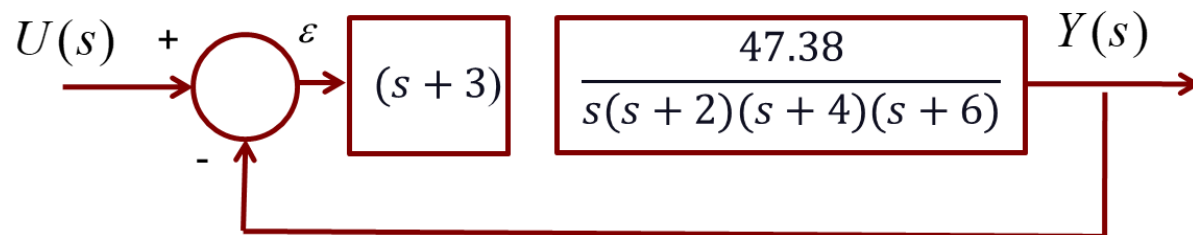
Hence the controller is  $(s+3)$

To find the gain  $K$ , we use the magnitude criterion:

$$K = \frac{\prod(\text{lengths from } s_0 \text{ to OLTF poles})}{\prod(\text{lengths from } s_0 \text{ to OLTF zeros})}$$

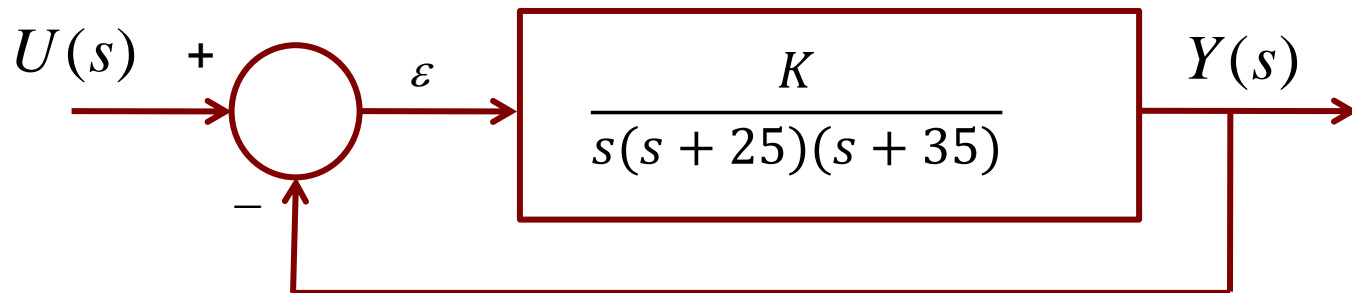
# Example 2

- ❖ The point is at  $s_0 = -3.614 + j6.19$
- ❖  $L_{op1} = \sqrt{3.614^2 + 6.19^2} = 7.17$
- ❖  $L_{op2} = \sqrt{0.386^2 + 6.19^2} = 6.20$
- ❖  $L_{op3} = \sqrt{2.386^2 + 6.19^2} = 6.63$
- ❖  $L_{oz1} = \sqrt{0.614^2 + 6.19^2} = 6.22$
- ❖  $K = \frac{\prod(\text{lengths from } s_0 \text{ to OLTf poles})}{\prod(\text{lengths from } s_0 \text{ to OLTf zeros})} = 47.38$
- ❖ The response of the compensated system should be checked and refined if needed



# Example 3

Given the system shown. Sketch the root locus of the system. Describe without calculation how you would reshape the root locus with a PID controller so that the output response has zero steady state error to the ramp input and damping of 0.7071 with settling time of less than 1 sec.



Specifications:

Zero steady state error to ramp: need type 2 (need to add one pole at  $s = 0$ )

$$T_s = \frac{4}{\zeta\omega_n} \text{ or } \zeta\omega_n = 4$$

$$\zeta = 0.707 \text{ or } \theta = \cos^{-1} \zeta = 45^\circ$$

