

MEMS1045

Automatic control

Lecture 3

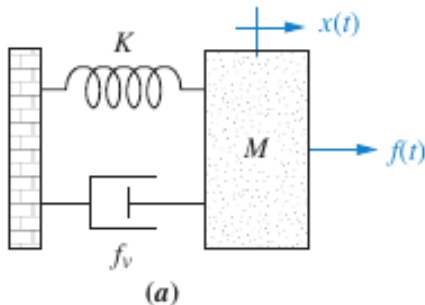
Modeling



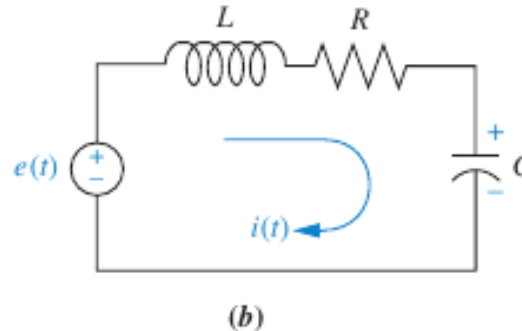
Objectives

- Describe analogous modeling between electrical and mechanical systems
- Derive the equations of motion for electromechanical systems
- Represent the equations of motion in matrix form using state variables
- Apply linearization to approximate the dynamics for small perturbations about operating conditions

Analogous systems



$$m \frac{d^2 x}{dt^2} + f_v \frac{dx}{dt} + kx = f(t)$$



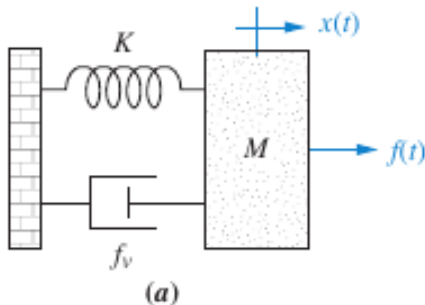
Charge = q
 $q = \frac{di}{dt}$

$$L \frac{d^2 q}{dt^2} + R \frac{dq}{dt} + \frac{1}{C} q = e(t)$$

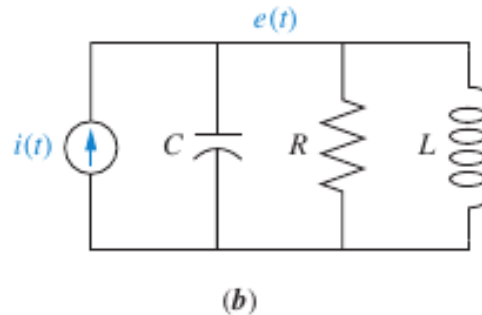
Force-Voltage Analogy

Mechanical Systems	Electrical Systems
Force p (torque T)	Voltage e
Mass m (moment of inertia J)	Inductance L
Viscous-friction coefficient b	Resistance R
Spring constant k	Reciprocal of capacitance, $1/C$
Displacement x (angular displacement θ)	Charge q
Velocity \dot{x} (angular velocity $\dot{\theta}$)	Current i

Analogous systems



$$m \frac{d^2 x}{dt^2} + f_v \frac{dx}{dt} + kx = f(t)$$



Magnetic flux = ψ

$$\frac{d\psi}{dt} = e$$

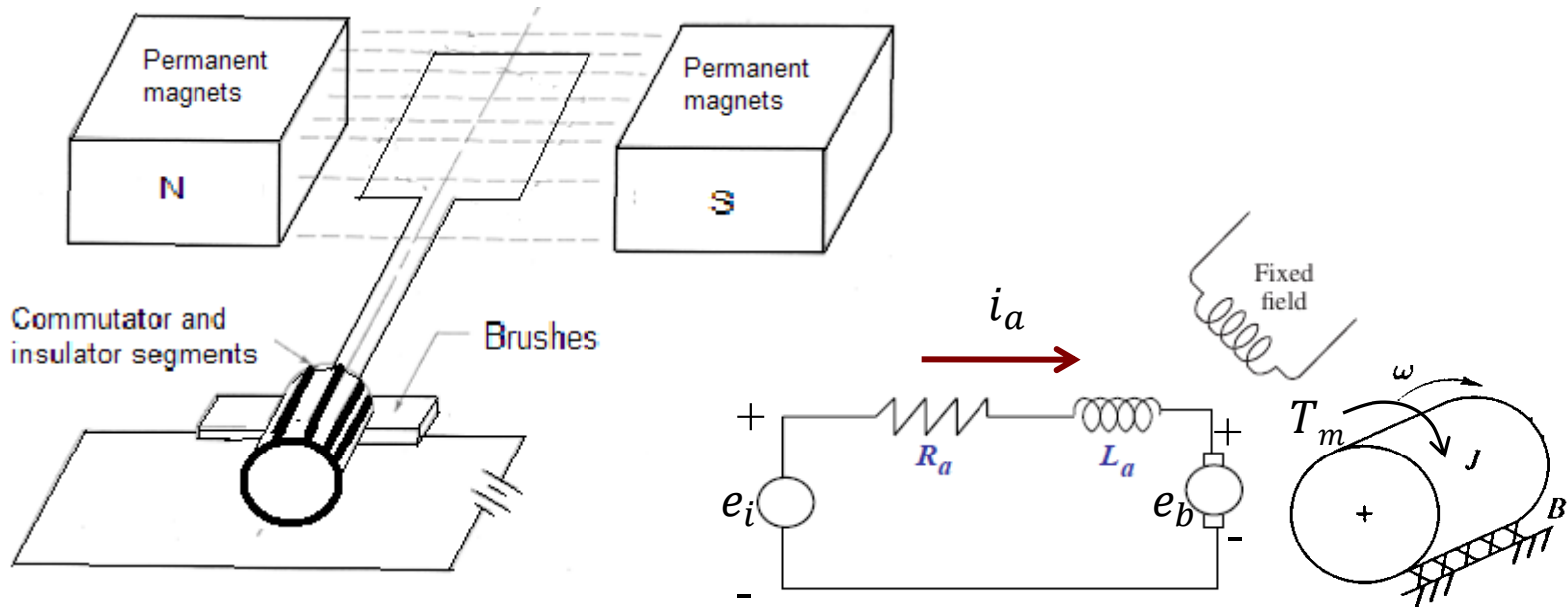
$$C \frac{d^2 \psi}{dt^2} + \frac{1}{R} \frac{d\psi}{dt} + \frac{1}{L} \psi = i(t)$$

Force-Current Analogy

Mechanical Systems	Electrical Systems
Force p (torque T)	Current i
Mass m (moment of inertia J)	Capacitance C
Viscous-friction coefficient b	Reciprocal of resistance, $1/R$
Spring constant k	Reciprocal of inductance, $1/L$
Displacement x (angular displacement θ)	Magnetic flux linkage ψ
Velocity \dot{x} (angular velocity $\dot{\theta}$)	Voltage e

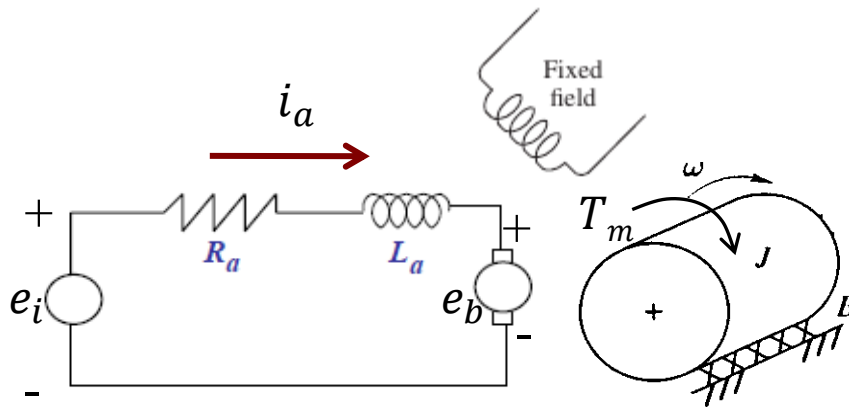
DC motor

Simplified diagram for permanent DC motor analysis:



- ❖ R_a = armature resistance
- ❖ L_a = armature inductance
- ❖ T_m = motor torque

DC motor



$$T_m = K_T i_a$$

$$T_m(s) = K_T I_a(s)$$

K_T = Torque constant

$$e_b = K_b \omega$$

$$E_b(s) = K_b s \theta_m(s)$$

K_b = Back EMF constant

$$e_i = i_a R_a + L_a \frac{di_a}{dt} + e_b$$

$$E_i(s) = R_a I_a(s) + L_a s I_a(s) + E_b(s)$$

$$E_i(s) = R_a I_a(s) + L_a s I_a(s) + K_b s \theta_m(s)$$

$$E_i(s) - K_b s \theta_m(s) = (R_a + L_a s) I_a(s)$$

$$I_a(s) = \frac{E_i(s)}{(R_a + L_a s)} - \frac{K_b s \theta_m(s)}{(R_a + L_a s)}$$

$$J \ddot{\theta}_m = -B \dot{\theta}_m + T_m$$

$$J s^2 \theta_m(s) = -B s \theta_m(s) + T_m(s)$$

$$J s^2 \theta_m(s) = -B s \theta_m(s) + K_T I_a(s)$$

DC motor

$$Js^2\theta_m(s) = -Bs\theta_m(s) + K_T I_a(s)$$

$$Js^2\theta_m(s) = -Bs\theta_m(s) + \frac{K_T E_i(s)}{(R_a + L_a s)} - \frac{K_b K_T s \theta_m(s)}{(R_a + L_a s)}$$

$$Js^2\theta_m(s) + Bs\theta_m(s) + \frac{K_b K_T s \theta_m(s)}{(R_a + L_a s)} = \frac{K_T E_i(s)}{(R_a + L_a s)}$$

For small L_a :

$$Js^2\theta_m(s) + Bs\theta_m(s) + \frac{K_b K_T s \theta_m(s)}{(R_a)} = \frac{K_T E_i(s)}{(R_a)}$$

$$s \left[s + \frac{1}{J} \left(B + \frac{K_b K_T}{(R_a)} \right) \right] \theta_m(s) = \frac{K_T}{J(R_a)} E_i(s)$$

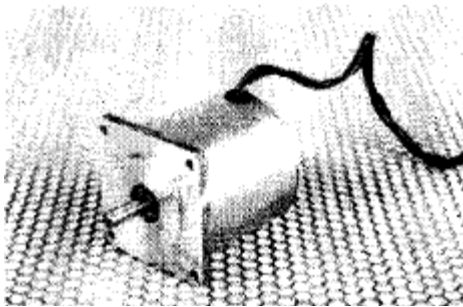
Motor transfer function

$$\frac{\theta_m(s)}{E_i(s)} = \frac{K_T / (R_a J)}{s \left[s + \frac{1}{J} \left(B + \frac{K_b K_T}{(R_a)} \right) \right]} = \frac{K_m}{s \left(s + \frac{1}{\tau_m} \right)}$$

K_m = motor
gain constant

τ_m = motor
time constant

Commercial DC motor

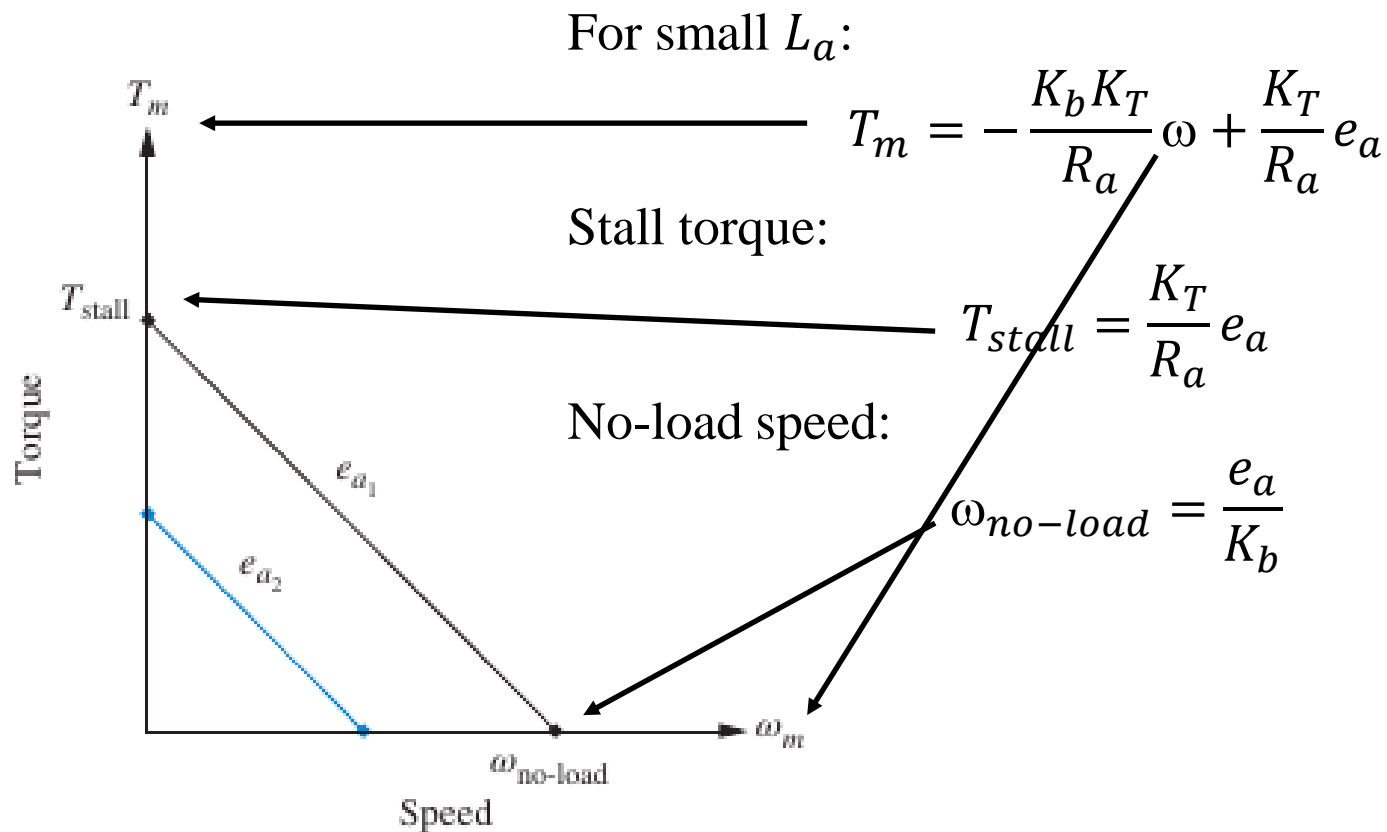


Brushless DC motor

SPECIFICATIONS

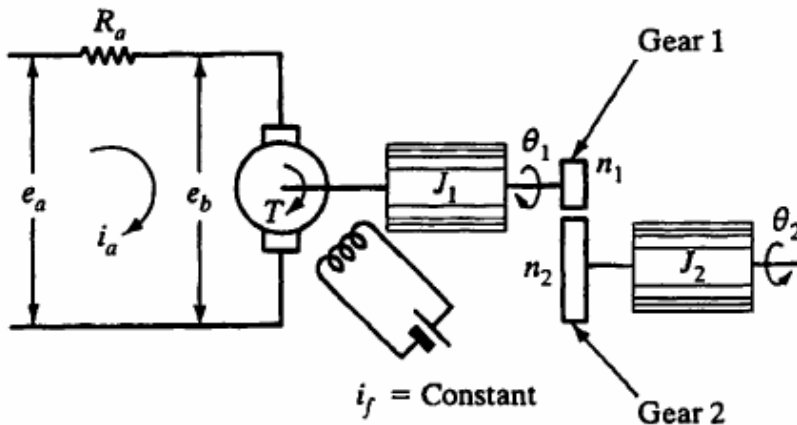
PARAMETER	UNITS	Value	
Torque Constant	oz-in/amp	1.65	← $K_T = 0.01164 \text{ Nm/A}$
Back EMF Constant	V/Krpm	1.22	← $K_b = 0.01164 \text{ Vs/rad}$
D.C. Resistance	ohms	0.12	
Inductance	mH	0.11	← Very small inductance
Max Speed	rpm	15,000	
Cont. Stall Torque	oz-in	25	
Motor Constant	oz-in/sq.rt.W	5.22	
Max. Winding Temp.	Deg.C	155	

Commercial DC motor



Example 1

Consider the dc servomotor system shown. The armature inductance is negligible and is not shown in the circuit. Obtain the transfer function between the output θ_2 and the input e_a .



n_1 = number of teeth on gear 1
 n_2 = number of teeth on gear 2

R_a = armature resistance,
 i_a = armature current,
 i_f = field current,
 e_a = applied armature voltage, e_b = back emf,
 θ_1 = angular displacement of the motor shaft,
 θ_2 = angular displacement of the load element,
 T = torque developed by the motor
 J_1 = moment of inertia of motor rotor,
 J_2 = moment of inertia of the load

Example 1

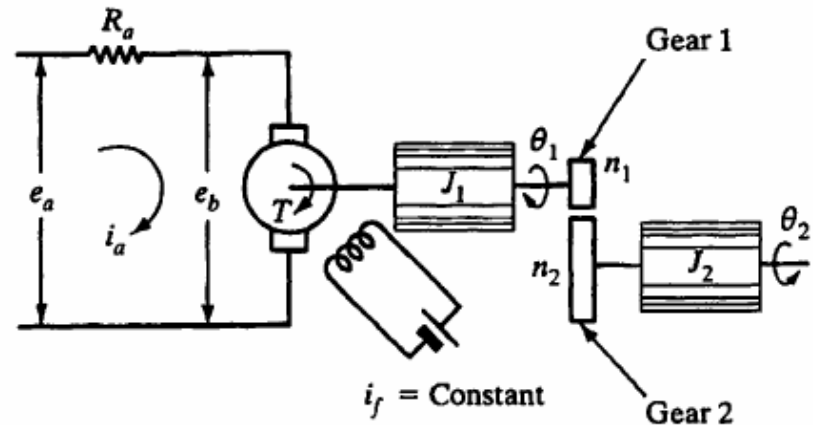
Torque developed by the motor is $T_m = K_T i_a$ or $T_m(s) = K_T I_a(s)$ where K_T = Torque constant

The induced voltage e_b is $e_b = K_b \omega$ or $E_b(s) = K_b s \theta_1(s)$ where K_b = Back EMF constant

The equation for the armature circuit is $e_a = i_a R_a + e_b$

$E_a(s) = R_a I_a(s) + E_b(s)$ or $E_a(s) = R_a I_a(s) + K_b s \theta_1(s)$

$$I_a(s) = \frac{E_a(s)}{R_a} - \frac{K_b s \theta_1(s)}{R_a}$$



Example 1

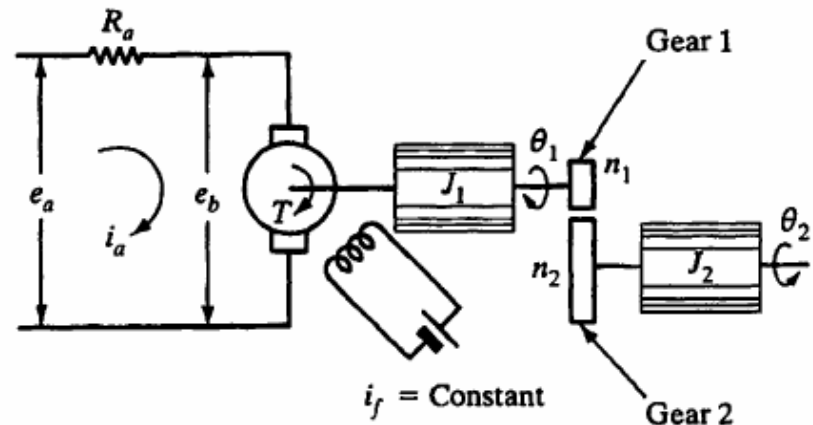
The total kinetic energy of the load inertias:

$$KE = \frac{1}{2}J_1\dot{\theta}_1^2 + \frac{1}{2}J_2\dot{\theta}_2^2 = \frac{1}{2}J_1\dot{\theta}_1^2 + \frac{1}{2}J_2\left(\frac{\dot{\theta}_2}{\dot{\theta}_1}\right)^2\dot{\theta}_1^2$$

$$KE = \frac{1}{2}J_1\dot{\theta}_1^2 + \frac{1}{2}J_2\left(\frac{n_1}{n_2}\right)^2\dot{\theta}_1^2 = \frac{1}{2}\left(J_1 + J_2\left(\frac{n_1}{n_2}\right)^2\right)\dot{\theta}_1^2$$

The equivalent moment of inertia of the motor rotor plus the load inertia referred to the motor shaft is

$$J_{1eq} = J_1 + N^2J_2 = J_1 + \left(\frac{n_1}{n_2}\right)^2 J_2$$



Example 1

$$J_{1eq} = J_1 + N^2 J_2 = J_1 + \left(\frac{n_1}{n_2}\right)^2 J_2$$

The armature current produces the torque that is applied to the equivalent moment of inertia J_{1eq} . Thus

$$J_{1eq} \ddot{\theta}_1 = T_m = K_T i_a$$

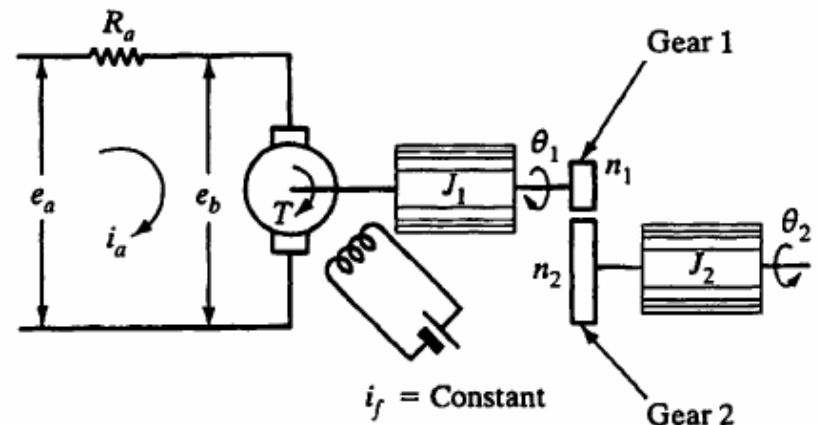
$$J_{1eq} s^2 \theta_1(s) = K_T I_a(s)$$

$$\text{But } I_a(s) = \frac{E_a(s)}{R_a} - \frac{K_b s \theta_1(s)}{R_a} \text{ or } J_{1eq} s^2 \theta_1(s) = K_T \left(\frac{E_a(s)}{R_a} - \frac{K_b s \theta_1(s)}{R_a} \right)$$

$$s \left(J_{1eq} s + \frac{K_T K_b}{R_a} \right) \theta_1(s) = \frac{K_T E_a(s)}{R_a}$$

$$s \left(J_{1eq} s + \frac{K_T K_b}{R_a} \right) \left(\frac{n_2}{n_1} \right) \theta_2(s) = \frac{K_T E_a(s)}{R_a}$$

$$\frac{\theta_2(s)}{E_a(s)} = \frac{\left(\frac{n_1}{n_2} \right) K_T}{s \left[R_a \left\{ J_1 + \left(\frac{n_1}{n_2} \right)^2 J_2 \right\} s + K_T K_b \right]}$$



State Variable Equations

- ❖ Transfer function relates the dynamic relationship between the input and output
- ❖ For a system with **many** inputs $u_1(t), u_2(t), \dots$ and **many** outputs y_1, y_2, \dots , we need to reduce a set of differential equations to a transfer function for each input and output pair
- ❖ An alternative is to represent the system using the state space approach
- ❖ A system of linear state equations (i.e. with n state variables, m inputs, and p outputs) can be expressed in a general matrix form:

$$\dot{\mathbf{x}} = \mathbf{A}\mathbf{x} + \mathbf{B}\mathbf{u}$$

$$\mathbf{y} = \mathbf{C}\mathbf{x} + \mathbf{D}\mathbf{u}$$

where $\mathbf{x} = (n \times 1)$ state vector; $\dot{\mathbf{x}}$ = derivative of the $(n \times 1)$ state vector w.r.t. time;

$\mathbf{u} = (m \times 1)$ input vector; $\mathbf{y} = (p \times 1)$ output vector;

$\mathbf{A} = (n \times n)$ system matrix; $\mathbf{B} = (n \times m)$ input matrix;

$\mathbf{C} = (p \times n)$ output matrix; $\mathbf{D} = (p \times m)$ feedforward matrix;

State Variable Equations

- ❖ State variables: The smallest set of linearly independent system variables such that the values of the members of the set at time t_0 along with known forcing functions completely determine the value of all system variables for all $t > t_0$
- ❖ Typically, the minimum number required equals the order of the differential equation describing the system

- ❖ Example: $m \frac{d^2 y}{dt^2} + b \frac{dy}{dt} + ky = u(t)$

To solve the ODE we need to know the values of the coefficients, the inputs and the 2 initial conditions/states at $y(t_0)$ and $\frac{dy}{dt}(t_0)$. Hence there are 2 state variables and they can be y and $\frac{dy}{dt}$

- ❖ For a n th-order differential equation, the state variables (i.e. x_1, x_2, \dots, x_n) reduce the n th-order differential equation into a set of first order ODEs

Example 2

Consider the ODE: $m \frac{d^2 y}{dt^2} + b \frac{dy}{dt} + ky = u(t)$. Obtain a state representation of the system with output y

Define state variables as $x_1 = y$ and $x_2 = \frac{dy}{dt}$

Derivatives of state variables $\dot{x}_1 = \frac{dy}{dt} = x_2$

$$\text{and } \dot{x}_2 = \frac{d^2 y}{dt^2} = \frac{1}{m} u - \frac{b}{m} \frac{dy}{dt} - \frac{k}{m} y = \frac{1}{m} u - \frac{b}{m} x_2 - \frac{k}{m} x_1$$

State equation:

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ -k/m & -b/m \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 \\ 1/m \end{bmatrix} [u]$$

Output equation:

$$[y] = [1 \quad 0] \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + [0][u]$$

Example 3

Find the state-space representation in phase-variable form for the transfer function

$$\frac{C(s)}{R(s)} = \frac{24}{s^3 + 9s^2 + 26s + 24}$$

Rewrite the transfer function as

$$(s^3 + 9s^2 + 26s + 24)C(s) = 24R(s)$$
$$\ddot{c} + 9\ddot{c} + 26\dot{c} + 24c = 24r(t)$$

Define state variables as $x_1 = c$; $x_2 = \dot{c}$; $x_3 = \ddot{c}$

Derivatives of state variables $\dot{x}_1 = \dot{c} = x_2$, $\dot{x}_2 = \ddot{c} = x_3$

and $\dot{x}_3 = \ddot{c} = 24r - 9\ddot{c} - 26\dot{c} - 24c = 24r - 9x_3 - 26x_2 - 24x_1$

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ -24 & -26 & -9 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ 24 \end{bmatrix} [r]$$

$$[y] = [1 \quad 0 \quad 0] \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix} + [0][u]$$

State space to transfer function

Given the state and output equations (assuming zero initial conditions):

$$\dot{\mathbf{x}} = \mathbf{A}\mathbf{x} + \mathbf{B}\mathbf{u} \quad \text{or} \quad \mathbf{sX}(s) = \mathbf{A}\mathbf{X}(s) + \mathbf{B}\mathbf{U}(s)$$

$$\mathbf{y} = \mathbf{C}\mathbf{x} + \mathbf{D}\mathbf{u} \quad \text{or} \quad \mathbf{Y}(s) = \mathbf{C}\mathbf{X}(s) + \mathbf{D}\mathbf{U}(s)$$

$$\mathbf{sX}(s) = \mathbf{A}\mathbf{X}(s) + \mathbf{B}\mathbf{U}(s)$$

$$[\mathbf{s} - \mathbf{A}]\mathbf{X}(s) = \mathbf{B}\mathbf{U}(s)$$

$$\mathbf{X}(s) = [\mathbf{s} - \mathbf{A}]^{-1}\mathbf{B}\mathbf{U}(s)$$

$$\mathbf{Y}(s) = \mathbf{C}\mathbf{X}(s) + \mathbf{D}\mathbf{U}(s) = \mathbf{C}[\mathbf{s} - \mathbf{A}]^{-1}\mathbf{B}\mathbf{U}(s) + \mathbf{D}\mathbf{U}(s)$$

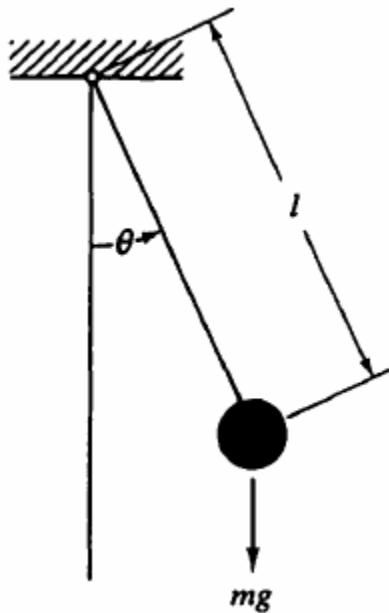
$$\mathbf{Y}(s) = \{\mathbf{C}[\mathbf{s} - \mathbf{A}]^{-1}\mathbf{B} + \mathbf{D}\}\mathbf{U}(s)$$

Hence

$$\frac{\mathbf{Y}(s)}{\mathbf{U}(s)} = \{\mathbf{C}[\mathbf{s} - \mathbf{A}]^{-1}\mathbf{B} + \mathbf{D}\}$$

Linearization

- ❖ All systems have some non-linearities
- ❖ We usually linearized the dynamics for small perturbations about the operating conditions



EOM:

$$J\ddot{\theta} = mgL \sin \theta$$

where $J = mL^2$

For small θ , $\sin \theta \approx \theta$ and the EOM simplifies to

$$J\ddot{\theta} = mgL\theta$$

System and models

Ordinary differential equations

$$\dot{\vec{x}} = f(\vec{x}, u)$$

$$y = h(\vec{x}, u)$$

Linearization
around $x_0=0$

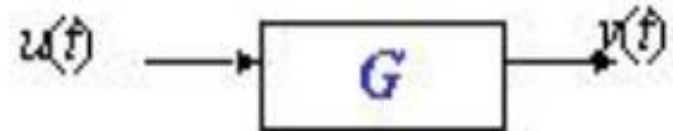
$$\dot{\vec{x}} = A\vec{x} + Bu$$

$$y = C\vec{x} + Du$$

Laplace transform

$$u(s)$$

Block Diagrams



Linearization
around $x_0=0$

