

ME 1071: Applied Fluids

Lecture 1 Internal Incompressible Viscous Flow

Spring 2021





- > Introduction
- Internal Flow Characteristics
- > Fully Developed Internal Flows
 - Moving Plates without Pressure Gradient (Couette Flow)
 - Parallel Stationary Plates with Pressure Gradient (Poiseuille Flow)
 - Moving Plates with Pressure Gradient
 - Pipe Flow





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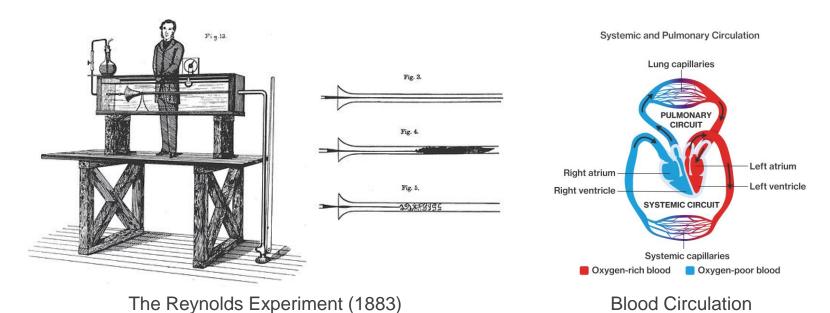
Introduction



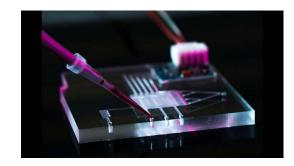


Internal flow

- Flows completely bounded by solid surfaces.
- Laminar or turbulent
- Analytical solutions are possible for some laminar case.



Water Supply Control



Lab-on-a-chip





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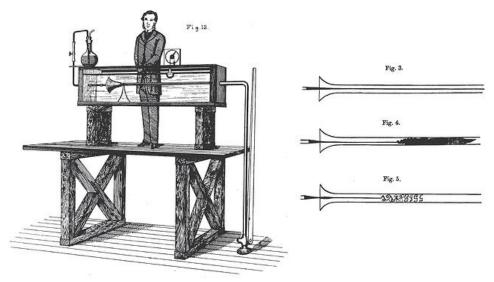
Introduction





Internal flow Characteristics

- Laminar versus turbulent
- Transition to Turbulence at $Re \approx 2300$
- The entrance region



The Reynolds Experiment (1883)

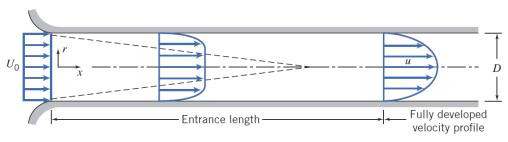


Fig. 8.1 Flow in the entrance region of a pipe.

Fully developed flow

- When the velocity or temperature profile shapes no longer change with increasing distance x
- For laminar flow, the entrance length is a function of Reynolds number

$$\frac{L}{D} \simeq 0.06 \,\mathrm{Re} = 0.06 \,\frac{\rho \bar{V} D}{\mu}, \qquad L < 138 D$$

For turbulent flow,

$$\frac{L}{D} \simeq 25-40$$
 for velocity, $\frac{L}{D} > 80$ for detailed motion





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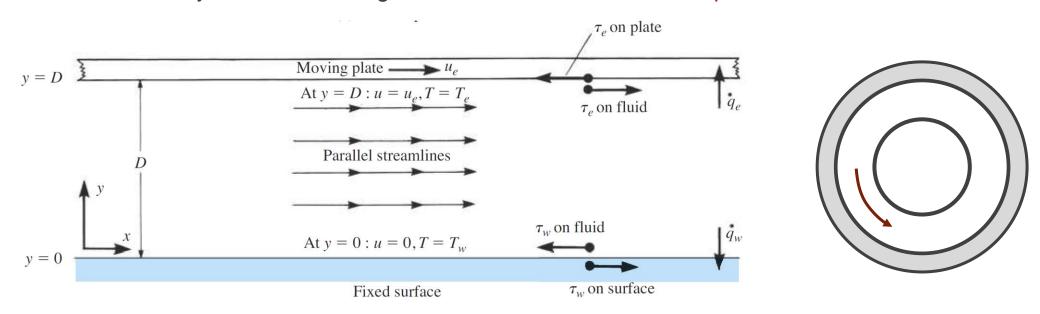
Introduction to Couette Flow





Couette Flow

- The flow of a viscous fluid in the space between two (infinite) surfaces, one of which is moving tangentially relative to the other. No pressure gradient is involved.
- The flow is driven by the viscous drag force and the streamlines are parallel.



Model for Couette flow between two plate or two concentric cylinder (同心圆柱).

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Introduction to Couette Flow





General Governing Equations (Review)

Continuity

$$\frac{\partial \rho}{\partial t} + \frac{\partial (\rho u)}{\partial x} + \frac{\partial (\rho v)}{\partial y} + \frac{\partial (\rho w)}{\partial z} = 0$$

Momentum

$$\rho \frac{Du}{Dt} = -\frac{\partial p}{\partial x} + \frac{\partial \tau_{xx}}{\partial x} + \frac{\partial \tau_{yx}}{\partial y} + \frac{\partial \tau_{zx}}{\partial z}$$

$$\rho \frac{Dv}{Dt} = -\frac{\partial p}{\partial y} + \frac{\partial \tau_{xy}}{\partial x} + \frac{\partial \tau_{yy}}{\partial y} + \frac{\partial \tau_{zy}}{\partial z}$$

$$\rho \frac{Dw}{Dt} = -\frac{\partial p}{\partial z} + \frac{\partial \tau_{xz}}{\partial x} + \frac{\partial \tau_{yz}}{\partial y} + \frac{\partial \tau_{zz}}{\partial z}$$

$$\rho \frac{Du}{Dt} = -\frac{\partial p}{\partial x} + \frac{\partial \tau_{xx}}{\partial x} + \frac{\partial \tau_{yx}}{\partial y} + \frac{\partial \tau_{zx}}{\partial z} \qquad \tau_{xy} = \tau_{yx} = \mu \left(\frac{\partial v}{\partial x} + \frac{\partial u}{\partial y} \right) \quad \tau_{xx} = \lambda (\nabla \cdot \mathbf{V}) + 2\mu \frac{\partial u}{\partial x}$$

$$\rho \frac{Dv}{Dt} = -\frac{\partial p}{\partial y} + \frac{\partial \tau_{xy}}{\partial x} + \frac{\partial \tau_{yy}}{\partial y} + \frac{\partial \tau_{zy}}{\partial z} \qquad \tau_{yz} = \tau_{zy} = \mu \left(\frac{\partial w}{\partial y} + \frac{\partial v}{\partial z} \right) \quad \tau_{yy} = \lambda (\nabla \cdot \mathbf{V}) + 2\mu \frac{\partial v}{\partial y}$$

$$\frac{\partial Dw}{\partial t} = -\frac{\partial p}{\partial z} + \frac{\partial \tau_{xz}}{\partial x} + \frac{\partial \tau_{yz}}{\partial y} + \frac{\partial \tau_{zz}}{\partial z} \qquad \tau_{zx} = \tau_{xz} = \mu \left(\frac{\partial u}{\partial z} + \frac{\partial w}{\partial x} \right) \quad \tau_{zz} = \lambda (\nabla \cdot \mathbf{V}) + 2\mu \frac{\partial w}{\partial z}$$

Energy

$$\rho \frac{D(e+V^{2}/2)}{Dt} = \rho \dot{q} + \frac{\partial}{\partial x} \left(k \frac{\partial T}{\partial x} \right) + \frac{\partial}{\partial y} \left(k \frac{\partial T}{\partial y} \right) + \frac{\partial}{\partial z} \left(k \frac{\partial T}{\partial z} \right)$$

$$-\nabla \cdot p \mathbf{V} + \frac{\partial (u\tau_{xx})}{\partial x} + \frac{\partial (u\tau_{yx})}{\partial y} + \frac{\partial (u\tau_{zx})}{\partial z} + \frac{\partial (v\tau_{xy})}{\partial x} + \frac{\partial (v\tau_{yy})}{\partial y} + \frac{\partial (v\tau_{zy})}{\partial z}$$

$$+ \frac{\partial (w\tau_{xz})}{\partial x} + \frac{\partial (w\tau_{yz})}{\partial y} + \frac{\partial (w\tau_{zz})}{\partial z}$$

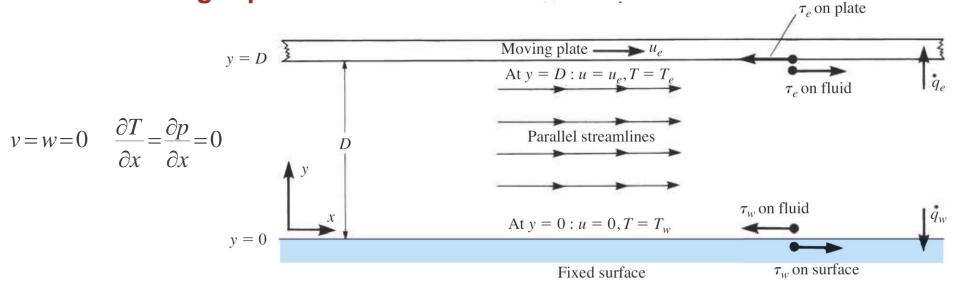
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Introduction to Couette Flow





The Governing Equations for Couette Flow



Continuity

$$\frac{\partial(\rho u)}{\partial x} = 0$$

Momentum

$$\frac{\partial}{\partial y} \left(\mu \frac{\partial u}{\partial y} \right) = 0$$

$$\frac{\partial}{\partial y} \left(k \frac{\partial T}{\partial y} \right) + \frac{\partial}{\partial y} \left(\mu u \frac{\partial u}{\partial y} \right) = 0$$

$$\frac{\partial p}{\partial y} = 0$$
 (pressure is constant throughout the entire flow field) 海納省川 有容乃大

Incompressible Couette Flow





Assumptions

- ρ, μ, k are treated as constants.
- T is also constant.

$$\frac{\partial}{\partial y} \left(\mu \frac{\partial u}{\partial y} \right) = 0$$

$$\downarrow \quad \mu = \text{const.}$$

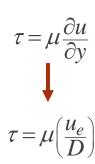
$$\frac{\partial^2 u}{\partial y^2} = 0$$

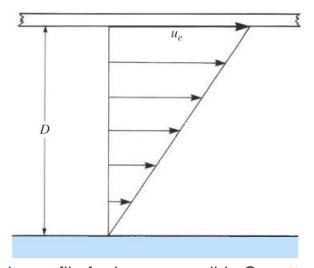
$$\downarrow \quad u = ay + b$$

$$\downarrow \quad y = 0, \ u = 0 \rightarrow b = 0;$$

$$\downarrow \quad y = D, \ u = u_e \rightarrow a = u_e/D.$$

$$u = u_e(y/D)$$





Velocity profile for incompressible Couette flow

For incompressible Couette flow

- Velocity varies linearly across the flow.
- τ increases linearly with u_e .
- τ is inversely proportional to D.





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Parallel Stationary Plates with Pressure Gradient (Poiseuille Flow)

- Both plates are stationary.
- The fluid is driven by pressure gradient.
- Flow is laminar and fully developed, body force is neglected.
- Boundary condition at y=0 u=0 at y=a u=0

$$dF_L = \left(p - \frac{\partial p}{\partial x} \frac{dx}{2}\right) dy dz + dF_T = \left(\tau_{yx} + \frac{d\tau_{yx}}{dy} \frac{dy}{2}\right) dx dz$$

Ш

$$dF_R = -\left(p + \frac{\partial p}{\partial x} \frac{dx}{2}\right) dy dz + dF_B = -\left(\tau_{yx} - \frac{d\tau_{yx}}{dy} \frac{dy}{2}\right) dx dz$$

$$\frac{d\tau_{yx}}{dy} = \frac{\partial p}{\partial x} = \text{constant}$$

Taylor Series Expansion

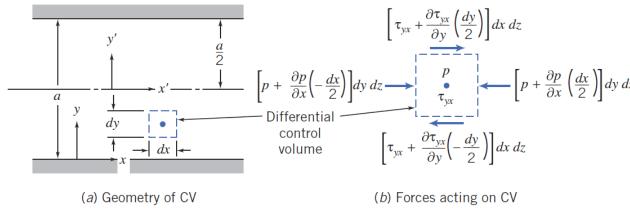


Fig. 8.3 Control volume for analysis of laminar flow between stationary infinite parallel plates.

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Parallel Stationary Plates with Pressure Gradient (Poiseuille Flow)

- Both plates are stationary.
- The fluid is driven by pressure gradient.
- Flow is laminar and fully developed, body force is neglected.
- Boundary conditions at y=0 u=0at y = a u = 0

$$\frac{d\tau_{yx}}{dy} = \frac{\partial p}{\partial x} = \text{constant}$$

$$\tau_{yx} = \mu \, \frac{du}{dy}$$

apply the boundary conditions

$$u=rac{a^2}{2\mu}\Big(rac{\partial p}{\partial x}\Big)\Big[\Big(rac{y}{a}\Big)^2-\Big(rac{y}{a}\Big)\Big]$$

Velocity Profile

control $\left[\tau_{yx} + \frac{\partial \tau_{yx}}{\partial y} \left(-\frac{dy}{2} \right) \right] dx dz$

volume

(a) Geometry of CV

(b) Forces acting on CV

Taylor Series Expansion

Fig. 8.3 Control volume for analysis of laminar flow between stationary infinite parallel plates.





Parallel Stationary Plates with Pressure Gradient (Poiseuille Flow)

- Both plates are stationary.
- The fluid is driven by pressure gradient.
- Flow is laminar and fully developed, body force is neglected.

• Boundary conditions at
$$y=0$$
 $u=0$ at $y=a$ $u=0$

$$\tau_{yx} = \mu \frac{\partial u}{\partial y} = a \Big(\frac{\partial p}{\partial x} \Big) \Big[\frac{y}{a} - \frac{1}{2} \Big] \quad \text{Shear Stress Distribution}$$

$$Q = \int_0^a u l dy \text{ or } \frac{Q}{l} = \int_0^a \frac{1}{2\mu} \left(\frac{\partial p}{\partial x} \right) (y^2 - ay) dy$$

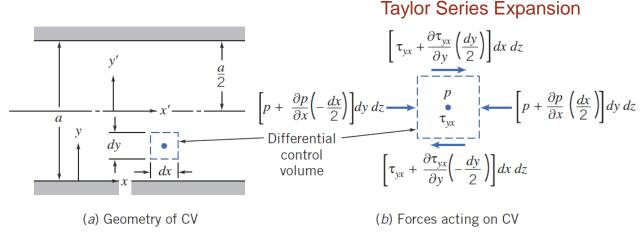


Fig. 8.3 Control volume for analysis of laminar flow between stationary infinite parallel plates.

$$\frac{Q}{l} = -\frac{1}{12\mu} \left(\frac{\partial p}{\partial x} \right) a^3 = \frac{a^3 \Delta p}{12\mu L}$$
(constant pressure gradient)

Volume Flow Rate





Parallel Stationary Plates with Pressure Gradient (Poiseuille Flow)

- Both plates are stationary.
- The fluid is driven by pressure gradient.
- Flow is laminar and fully developed, body force is neglected.
- Boundary conditions at y=0 u=0at y = a u = 0

$$ar{V} = rac{Q}{A} = rac{a^2 \Delta p}{12 \mu L} ext{(constant pressure gradient)}$$

Average Velocity

$$u = rac{a^2}{2\mu} \left(rac{\partial p}{\partial x}
ight) \left[\left(rac{y}{a}
ight)^2 - \left(rac{y}{a}
ight)
ight] rac{y = a/2}{2} u_{
m max} = -rac{1}{8\mu} \left(rac{\partial p}{\partial x}
ight) a^2 = rac{3}{2} ar{V}$$

Maximum Velocity

NOTE: ALL this results are valid for laminar flow only!





Example 8.1

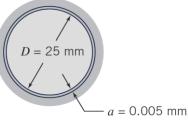
• A hydraulic system operates at a gage pressure of 20 MPa and 55°C with a hydraulic fluid of SAE 10W oil. A control valve consists of a piston 25 mm in diameter, fitted to a cylinder with a mean radial clearance of 0.005 mm. Determine the leakage flow rate if the gage pressure on the low pressure side of the piston is 1.0 MPa and the piston is 15 mm long.

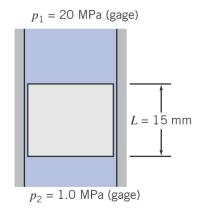
$$\frac{Q}{l} = \frac{a^3 \Delta p}{12\mu L}$$

$$Q = \frac{a^3 l \Delta p}{12 \mu L} = \frac{a^3 \pi D \Delta p}{12 \mu L}$$

$$Q = \frac{\pi}{12} \times 0.005^3 \times 25 \times (20 - 1) \times 10^6 \times \frac{1}{0.018} \times \frac{1}{15 \times 10^{-3}} = 57.6 \text{ mm}^3/\text{s}$$

NOTE: Remember to double check the Reynolds number.





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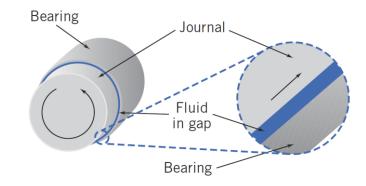




Upper Moving Plates with Pressure Gradient

- The fluid is driven by pressure gradient and the moving plate.
- Flow is laminar and fully developed, body force is neglected.

• Boundary conditions
$$u = 0$$
 at $y = 0$
 $u = U$ at $v = a$



$$u=rac{Uy}{a}+rac{a^{\,2}}{2\mu}\Big(rac{\partial p}{\partial x}\Big)\Big[\Big(rac{y}{a}\Big)^{\,2}-\Big(rac{y}{a}\Big)\Big]$$

Velocity Profile

$$au_{yx}\!=\!\murac{\partial u}{\partial y}\!=\!\murac{U}{a}+a\!\left(\!rac{\partial p}{\partial x}\!
ight)\!\left[rac{y}{a}-rac{1}{2}
ight]$$

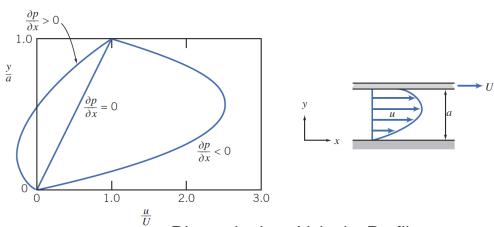
Shear Stress Distribution

$$\frac{Q}{l} = \frac{Ua}{2} - \frac{1}{12\mu} \left(\frac{\partial p}{\partial x} \right) a^3$$

Volume Flow Rate

$$\frac{du}{dy} = 0$$
 at $y = \frac{a}{2} - \frac{U/a}{(1/\mu)(\partial p/\partial x)}$

Point of Maximum Velocity



Dimensionless Velocity Profile





Pipe Flow

- The fluid is driven by pressure gradient in a pipe.
- Flow is laminar and fully developed, body force is neglected.
- Boundary conditions u = 0 at r = R (no slip condition)

$$dF_L = p2\pi r \, dr + dF_O = \left(\tau_{rx} + \frac{d\tau_{rx}}{dr} \, dr\right) 2\pi (r + dr) dx$$

$$dF_R = -\left(p + \frac{\partial p}{\partial x} \, dx\right) 2\pi r \, dr + dF_I = -\tau_{rx} 2\pi r \, dx$$

$$\downarrow \qquad \qquad \downarrow \qquad \qquad$$

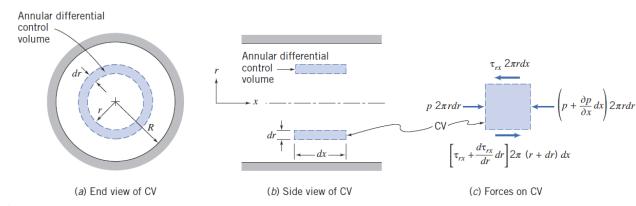


Fig. 8.7 Differential control volume for analysis of fully developed laminar flow in a pipe.

$$u = -\frac{R^2}{4\mu} \left(\frac{\partial p}{\partial x}\right) \left[1 - \left(\frac{r}{R}\right)^2\right]$$

Velocity Profile





Pipe Flow

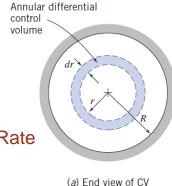
- The fluid is driven by pressure gradient in a pipe.
- Flow is laminar and fully developed, body force is neglected.
- Boundary conditions u = 0 at r = R (no slip condition)

$$au_{xy}\!=\!\murac{du}{dr}\!=\!rac{r}{2}\!\left(\!rac{\partial p}{\partial x}\!
ight)$$

Shear Stress Distribution

$$Q = -\frac{\pi R^4}{8\mu} \left(\frac{\partial p}{\partial x} \right) = \frac{\pi \Delta p R^4}{8\mu L} = \frac{\pi \Delta p D^4}{128\mu L}$$

Volume Flow Rate



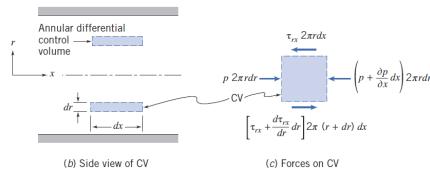


Fig. 8.7 Differential control volume for analysis of fully developed laminar flow in a pipe.

$$ar{V} = rac{Q}{A} = - \left. rac{R^2}{8 \mu} \left(rac{\partial p}{\partial x}
ight)
ight|$$
 Averaged Velocity

 $u_{
m max}\!=\!-\,rac{R^2}{4\mu}\!\left(\!rac{\partial p}{\partial x}\!
ight)\!=\!2ar{V}$

Maximum Velocity

Friction lose and other loses caused by valves and elbows are neglected.

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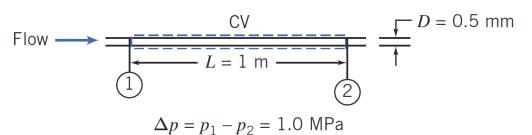
Example 8.4

A test of a certain liquid in a capillary viscometer gave the following data:

$$Q = 880 \frac{mm^3}{s}$$
, $L = 1 m$, $D = 0.50 mm$, $\Delta p = 1.0 MPa$

Determine the viscosity of the fluid.

$$Q = \frac{\pi \Delta p D^4}{128\mu L}$$



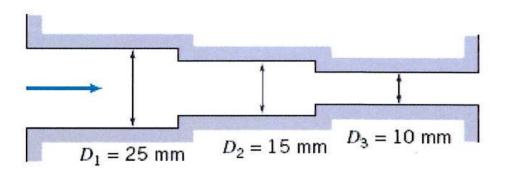
$$\mu = \frac{\pi \Delta p D^4}{128 LQ} = \frac{\pi \times 1 \times 10^6 \times (0.5 \times 10^{-3})^4}{128 \times 1 \times 880 \times 10^{-9}} = 1.74 \times 10^{-3} \,\text{N} \cdot \text{s/m}^2$$





Problem 8.3

8.3 Air at 40°C flows in a pipe system in which diameter is decreased in two stages from 25 mm to 15 mm to 10 mm. Each section is 2 m long. As the flow rate is increased, which section will become turbulent first? Determine the flow rates at which one, two, and then all three sections first become turbulent. At each of these flow rates, determine which sections, if any, attain fully developed flow.



$$Q_{1} = \frac{Re_{crit}\pi v D_{1}}{4}$$

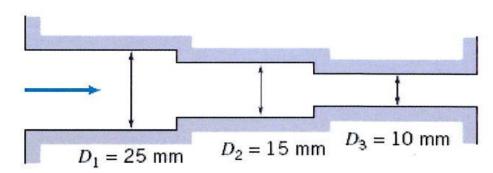
$$Q_{2} = \frac{Re_{crit}\pi v D_{2}}{4}$$

$$Q_{3} = \frac{Re_{crit}\pi v D_{3}}{4}$$





Problem 8.3



$$L_{laminar} = 0.06 ReD$$
, $Turbulent: L_{min} = 25D$ $L_{max} = 40D$

Q_3 :

$$L_{min,3} = 25D_3$$
 $L_{max,3} = 40D_3$

$$L_{laminar,3} = 0.06 \left(\frac{4Q_3}{\pi v D_3}\right) D_3,$$

$$L_{laminar,2} = 0.06 \left(\frac{4Q_3}{\pi v D_2} \right) D_2$$

$$L_{laminar,1} = 0.06 \left(\frac{4Q_3}{\pi v D_1}\right) D_1$$

$$Q_2$$
:
 $L_{min,3} = 25D_3$ $L_{max,3} = 40D_3$
 $L_{min,2} = 25D_2$ $L_{max,2} = 40D_2$
 $L_{laminar,2} = 0.06 \left(\frac{4Q_2}{\pi v D_2}\right) D_2$
 $L_{laminar,1} = 0.06 \left(\frac{4Q_2}{\pi v D_1}\right) D_1$

$$Q_1$$
:
$$L_{min,3} = 25D_3 \qquad L_{max,3} = 40D_3$$

$$L_{min,2} = 25D_2 \qquad L_{max,2} = 40D_2$$

$$L_{min,1} = 25D_1 \qquad L_{max,1} = 40D_1$$

$$L_{laminar,1} = 0.06 \left(\frac{4Q_1}{\pi v D_1}\right) D_1$$





Problem 8.12

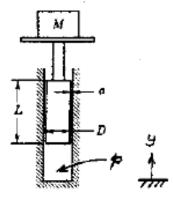
A large mass is supported by a piston of diameter D = 100 mm and length L = 100 mm. The piston sits in a cylinder closed at the bottom, and the gap a = 0.025 mm between the cylinder wall and piston is filled with SAE 10 oil at 20° C. The piston slowly sinks due to the mass, and oil is forced out at a rate of $6 \times 10^{-6} \ m^3/s$. What is the mass (kg)?

Flow between stationary plates:

$$V_{ave} = \frac{Q}{a\pi D} \rightarrow Re = \frac{aV_{ave}}{v}$$

$$\frac{Q}{l} = \frac{a^3 \Delta p}{12\mu L}$$

$$\Delta p \to F \to m$$







Problem 8.18

Consider fully developed laminar flow between infinite parallel plates separated by gap width d = 10 mm. The upper plate moves to the right with speed $U_2 = 0.5$ m/s; the lower plate moves to the left with speed $U_1 = 0.25$ m/s. The pressure gradient in the direction of flow is zero. Develop an expression for the velocity distribution in the gap. Find the volume flow rate per unit depth (m³/sec/m) passing a given cross section.

$$\frac{dP}{dx} = 0 \quad \rightarrow \quad \frac{d\tau}{dy} = \mu \frac{d^2u}{dy^2} = 0$$

$$\frac{d^2u}{dy^2} = 0 \quad \to \quad u(y), \qquad u(0) = -U_1, \qquad u(d) = U_2$$

$$Q = \int u \, dA = b \int_0^d u \, dy \qquad \rightarrow \qquad \frac{Q}{b} = ???$$

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Problem 8.39

Consider first water and then SAE 10W lubricating oil flowing at 40°C in a 6 mm diameter tube. Determine the maximum flow rate and corresponding pressure gradient for each fluid assuming laminar flow.

$$Re = \frac{\rho \overline{V}D}{\mu} = \frac{4Q}{\pi Dv}$$

$$Q = Q_{max} = \frac{Re_{max}\pi Dv}{4} = \left(\frac{Re_{max}\pi D}{4}\right)v$$

$$\frac{Q_{max}}{A} = -\frac{R^2}{8\mu} \frac{\delta p}{\delta x}$$

$$\frac{\delta p}{\delta x} = -\frac{128\mu Q_{max}}{\pi D^4} = \left(-\frac{128}{\pi D^4}\right) Q_{max} \mu$$

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Problem 8.40

For fully developed laminar flow in a pipe, determine the radial distance from the pipe axis at which the velocity equals the average velocity.

$$\bar{V} = \frac{Q}{A} = -\frac{R^2}{8\mu} \frac{\delta p}{\delta x}$$

$$u = -\frac{R^2}{4\mu} \frac{\delta p}{\delta x} \left[1 - \left(\frac{r}{R}\right)^2 \right]$$





This assignment is due by 6pm on March 18th.

- Upload your solution to BB.
- You can either type your solution out using a word editor like Microsoft Word or clearly hand write the information and then copy/scan your solution to a digital file.