

Satellites, instruments

1. Sentinel 1A, C-Band SAR (stripmap mode only)
2. Sentinel 1B, C-Band SAR (stripmap mode only)
3. ISS, GEDI

Mission Epoch: 1/28/2021 15:00:00 UTC Gregorian

Mission Duration: 2 days

Orbital Elements:: UTCGregorian, SMA, TA, RAAN, AOP, ECC, INC

Sentinel 1A

28 Jan 2021 15:00:00.000	7070.052000472739	135.7009006810358
38.39069816072309	196.5334447649118	0.0006318697105749462
98.1852224430984		

Sentinel 1B

28 Jan 2021 15:00:00.000	7070.688121440737	26.55162234490189
38.21545263412436	128.4350156317218	0.0005755569537709292
98.18526501753023		

ISS

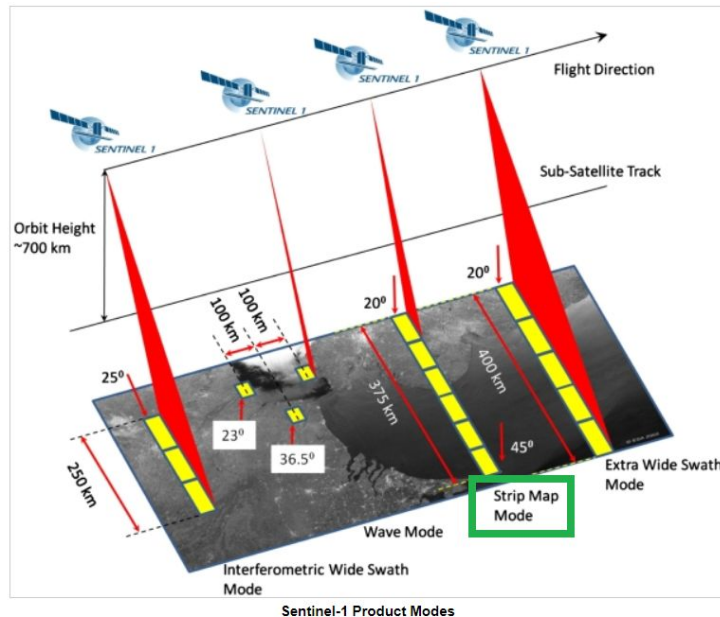
28 Jan 2021 15:00:00.000	6806.105884479364	21.93742673963972
313.1666332444946	308.8861243613201	0.0003303640879203458
51.67213619188961		

Sentinel 1 instrument

<https://sentinel.esa.int/web/sentinel/technical-guides/sentinel-1-sar/sar-instrument>

https://sentinel.esa.int/documents/247904/349449/S1_SP-1322_1.pdf

<https://dragon3.esa.int/web/sentinel/user-guides/sentinel-1-sar/acquisition-modes/stripmap>



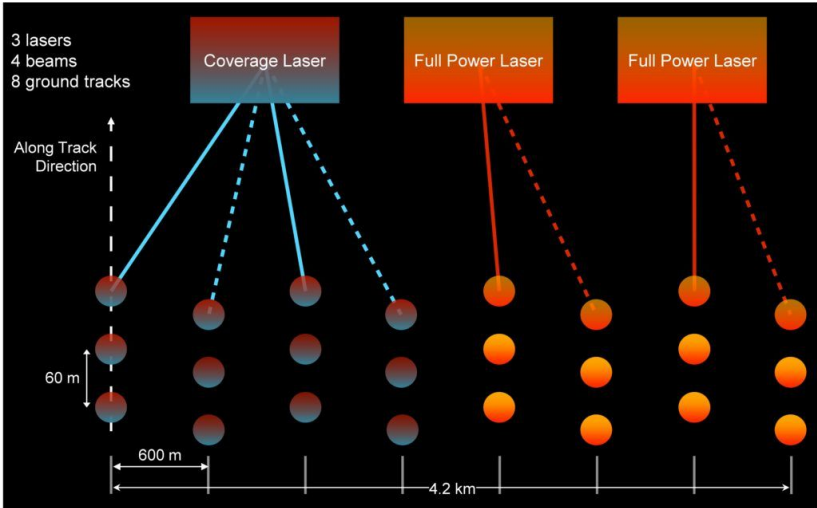
Only Stripmap mode is considered. In this any **one** of the 6 narrow-swath regions can be imaged **at a time**. The 6 narrow swaths correspond to 6 look angles (pointing-options). The extra-wide swath mode trades of image resolution for the wide swath.

Center freq	5.405 GHz
Bandwidth	50 MHz (selected from 0 to 100MHz)
Polarization	Dual
Antenna size	12.3m x 0.821m
Antenna efficiency	50% assumed
Pulse width	43us (selected from 5-100us)
Tx duty cycle	8.5% (Stripmap) (not used....)
PRF	To be appropriately chosen in the range (1000Hz to 3000Hz).
Receiver (System) noise figure	3.2 dB
2-way Atmos loss	2 dB assumed
Radar losses	3.5 dB assumed

sceneNoiseTemp	290K
Maneuverability (Look angles [deg])	<p>6 Look angles (pointing-options): 20.73, 23.665, 28.585, 33.01, 36.96, 39.27</p> <p>(Note taken at min altitude) 17.93-23.53 21.00-26.33 26.18-30.99 30.87-35.15 35.07-38.85 37.53-41.01</p>
Swath mode	Fixed swath of 80 km (i.e. swath width is fixed irrespective of the look-angle)
Peak power	4368 W

GEDI

<https://gedi.umd.edu/instrument/specifications/>



FOV	<p>Based on above diagram FOV is modelled (approximately) as rectangular shape with Cross-track = $\text{atan}(4.2\text{km}/408\text{km}) = 0.5898\text{deg}$</p> <p>Along-track FOV = $\text{atan}(60\text{m}/408\text{km}) = 8.4\text{milli-deg}$</p> <p>There shall be 8 spot beams within this FOV where the actual measurements occur.</p>
Maneuverability	<p>Can be rotated upto 6 deg either side of ISS</p> <p>Assume that rotation step size is 1 deg, this gives 13 look-angles</p>

	(pointing-options): -6, -5,0, 1, 2,, 6
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Notes:

- The coverage calculation involves finding the intersection of the instrument pointing-axis with the Earth. The FOV of the sensor is **not** utilized.