In [1]:

```
using LinearAlgebra
using FFTW
using Plots
using BenchmarkTools
using OrdinaryDiffEq
include("../../code/TaylorFourier.jl");
```

The main problem of artificial satellite theory (in Stiefel-Scheifele VOP formulation)

Written as a system of ODEs for a fictitious time τ (with the physical time t as an additional state variable):

$$\frac{d}{d\tau}q = r v, \quad q(0) = q_0,$$

$$\frac{d}{d\tau}v = -\frac{\mu}{r^2}q - r \nabla V(q), \quad v(0) = v_0$$

$$\frac{d}{d\tau}t = r, \quad t(0) = t_0$$
(1)

where

$$q_0 = \begin{pmatrix} x_0 \\ y_0 \\ z_0 \end{pmatrix}, \quad \upsilon_0 = \begin{pmatrix} \dot{x}_0 \\ \dot{y}_0 \\ \dot{z}_0 \end{pmatrix}, \quad r = ||q||,$$

and

$$V(q) = \frac{C}{2r^3} (3 \sin^2 \theta - 1), \quad \sin \theta = \frac{z}{r}, \quad C = J_2 \mu R_e^2.$$

Constant parameters of the problem: $R_e=6378.135$ km, $\mu=398600$ km $^3/s^2$, $J_2=0.0010826157$.

The energy $\mathcal{E}(q,v):=\frac{1}{2}\left\langle v,v\right\rangle -\frac{\mu}{r}+V(q)$ is an invariant of the equations of motion. In addition, the z-component of the angular momentum

$$x\dot{v} - \dot{x}v$$

is also a first integral.

In [2]:

```
1
   function Energy(q,v,parms)
 2
        \mu = parms[1]
 3
        C = parms[2]
 4
        return 0.5*dot(v,v) - \mu/norm(q) + V(q,parms)
 5
   end
 6
 7
   function V(q,parms)
 8
        C = parms[2]
 9
        z = q[3]
10
        r = norm(q)
        sinth = z/r
11
        return C*(3*sinth^2-1)/(2*r^3)
12
13
   end
14
15
   function ZAngularMomentum(q,v)
        return q[1]*v[2] - q[2]*v[1]
16
17
   end
```

Out[2]:

ZAngularMomentum (generic function with 1 method)

Following~\cite{Stiefel-Scheifele1971}, $q(\tau)$ and $t(\tau)$ in (1) can be obtained as follows:

$$q(\tau) = L(u(\tau))u(\tau), \quad v(\tau) = \frac{2}{\|u(\tau)\|^2} L(u(\tau))u'(\tau),$$

$$u(\tau) = \cos(\omega \tau)\alpha(\tau) + \omega^{-1} \sin(\omega \tau)\beta(\tau), \quad u'(\tau) = -\omega \sin(\omega \tau)\alpha(\tau) + \cos(\omega \tau)\beta(\tau),$$

where

$$\omega = \sqrt{-\frac{1}{2} \mathcal{E}(q_0, v_0)}, \qquad L(\mathbf{u}) = \begin{pmatrix} u_1 & -u_2 & -u_3 & u_4 \\ u_2 & u_1 & -u_4 & -u_3 \\ u_3 & u_4 & u_1 & u_2 \end{pmatrix},$$

and $(\alpha(\tau), \beta(\tau), t(\tau))$ is the solution of the 8-dimensional ODE system

$$\frac{d}{d\tau}\alpha = \omega^{-1}\sin(\omega\tau)\nabla R(\cos(\omega\tau)\alpha + \omega^{-1}\sin(\omega\tau)\beta), \quad \alpha(0) = u_0,
\frac{d}{d\tau}\beta = -\cos(\omega\tau)\nabla R(\cos(\omega\tau)\alpha + \omega^{-1}\sin(\omega\tau)\beta), \quad \beta(0) = w_0,
\frac{d}{d\tau}t = \|\cos(\omega\tau)\alpha + \omega^{-1}\sin(\omega\tau)\beta\|^2, \quad t(0) = t_0.$$
(2)

Here, $R(u) = \frac{1}{4} \|u\|^2 V(L(u)u)$, that is,

$$R(u) = \frac{1}{8r^2} \left(3 \left(\sin(\theta) \right)^2 - 1 \right),$$

where $r = u_1^2 + u_2^2 + u_3^2 + u_4^2$ and $\sin(\theta) = 2 (u_1 u_3 + u_2 u_4)/r$. As for the vectors $u_0, w_0 \in \mathbb{R}^2$,

• $u_0 \in \mathbb{R}^4$ is chosen in such a way that

$$q_0 = L(u_0)u_0,$$

• $w_0 \in \mathbb{R}^4$ is determined as

$$w_0 = \frac{1}{2} L(u_0)^T \dot{q}_0.$$

Obviously, there are infinitely many vectors u_0 that satisfy $q_0 = L(u_0)u_0$. The function $\chi(q)$ implemented below computes one of them.

```
In [3]:
```

```
function \chi(q)
 1
 2
       x = q[1]
 3
        y = q[2]
 4
        z = q[3]
 5
        r = sqrt(x^2+y^2+z^2)
 6
        if x >= 0
 7
            aux = r + x
 8
            u1 = 0.5*sqrt(aux)
 9
            u4 = u1
10
            u2 = (y*u1 + z*u4)/aux
11
            u3 = (z*u1 - y*u4)/aux
       else
12
13
            aux = r - x
14
            u2 = 0.5*sqrt(aux)
15
            u3 = u2
16
            u1 = (y*u2 + z*u3)/aux
17
            u4 = (z*u2 - y*u3)/aux
18
        end
19
        return [u1, u2, u3, u4]
20
   end
21
22
   L(u) = [u[1] - u[2] - u[3] u[4]
23
         u[2] u[1] -u[4] -u[3]
24
         u[3]
              u[4] u[1] u[2] ];
```

Let us check it for a randomly chosen q_0 :

```
In [4]:
```

```
1 q0 = rand(3)

2 u0 = χ(q0)

3 norm(q0 - L(u0)*u0)
```

Out[4]:

6.206335383118183e-17

```
In [5]:
```

```
1 v0 = rand(3)

2 w0 = 0.5*(L(u0)')*v0

3 norm(2/dot(u0,u0)*L(u0)*w0 - v0)
```

Out[5]:

6.938893903907228e-17

Numerical solution with 9th order explicit RK method (Vern9)

We next implement in an efficient way the system of ODEs defined in (2).

In [6]:

```
1
 2
    function fODE!(dU, U, parms, \tau)
 3
        # Efficient implementation of the differential equations for (lpha,\ eta,\ t)
 4
        # \alpha = U[1:4], d\alpha/d\tau = dU[1:4]
 5
        \# \beta = U[5:8],
                         d\beta/d\tau = dU[5:8]
 6
        \# t = U[9],
                         dt/d\tau = dU[9]
 7
        C = parms[2]
 8
        \omega = parms[3]
 9
        \theta = \omega \star \tau
10
        c = cos(\theta)
        s = \sin(\theta)/\omega
11
        \#u = c * \alpha + s * \beta
12
        u1 = c * U[1] + s * U[5]
13
        u2 = c * U[2] + s * U[6]
14
        u3 = c * U[3] + s * U[7]
15
        u4 = c * U[4] + s * U[8]
16
17
        z = 2*(u1*u3 + u2*u4)
        r = u1^2 + u2^2 + u3^2 + u4^2
18
19
        w = 1/r^3
20
        sth = z/r
21
        A = 0.5*C*w*(1 - 6*sth^2)
22
        B = 1.5*C*w*sth
23
        gradR1 = A*u1+B*u3
24
        gradR2 = A*u2+B*u4
25
        gradR3 = A*u3+B*u1
26
        gradR4 = A*u4+B*u2
27
        dU[1] = s*gradR1
28
        dU[2] = s*gradR2
29
        dU[3] = s*gradR3
30
        dU[4] = s*gradR4
31
        dU[5] = -c*gradR1
32
        dU[6] = -c*gradR2
33
        dU[7] = -c*gradR3
        dU[8] = -c*gradR4
34
35
        dU[9] = r
36
        return nothing
37
   end
```

Out[6]:

fODE! (generic function with 1 method)

We now consider the initial state of a Geostationary satellite (Montenbruck 2000, pg. 116) and solve (2) numerically with an explicit RK method of order 9 due to Verner.

```
In [7]:
```

```
1 \mu = 398600.8
 2 R_e = 6378.135
 3 \in = 0.0010826157
 4 C = \mu * R_e^2 * \epsilon
 5 \# q0 = [0., 37947.73745727695, 0.]
 6 \#v0 = [3.297676220718193, 0., 0.8244190551795483]
 7
   # Geostationary satellite (Montenbruck pg. 116)
   q0 = [4.21491336e4, 0.0, 0.0]
                                                                   # km
 8
    v0 = [0.0, 3.075823259987749, 0.0010736649055318406] # km/s
 9
10
11
   \omega = \operatorname{sqrt}(-\operatorname{Energy}(q_0, v_0, [\mu, C])/2)
12
13
14 | u0 = \chi(q0)
15 | w0 = 0.5*(L(u0)') * v0
16
17
   \alpha 0 = u0
18 \mid \beta 0 = w0
19
   orbit_period = 2*\pi/\omega
20
21 t0 = 0.
22 n_orbits = 400
23
   tspan = (t0,t0+n_orbits*orbit_period)
24 U0 = [\alpha 0; \beta 0; t0]
25
26 parms = [\mu, C, \omega]
27
28 prob = ODEProblem(fODE!, U0, tspan, parms);
29
30
31
   times = range(tspan[1],tspan[2],length=Int64(n_orbits*2M)+1)
32
33
34
   tol = 1e-13
   @time sol = solve(prob, Vern9(), abstol=tol, reltol=tol, saveat=times);
```

4.712204 seconds (10.70 M allocations: 569.754 MiB, 3.92% gc time, 99.29% compilation time)

Let us check the errors in the conservarion of energy and the z component of the angular momentum.

In [8]:

```
function qfromU(U, \tau, parms)
 1
 2
          \omega = parms[3]
 3
          \theta = \omega \star \tau
 4
          c = cos(\theta)
 5
          s = \sin(\theta)
 6
          \alpha = U[1:4]
 7
          \beta = U[5:8]
          u = c * \alpha + (s/\omega) * \beta
 8
 9
          q = L(u)*u
10
          return q
11
    end
12
13
14 function vfromU(U, τ, parms)
15
          \omega = parms[3]
          \theta = \omega \star \tau
16
17
          c = cos(\theta)
         s = \sin(\theta)
18
          \alpha = U[1:4]
19
20
          \beta = U[5:8]
21
          \mathbf{u} = \mathbf{c} * \alpha + (\mathbf{s}/\omega) * \beta
22
          r = dot(u,u)
23
          w = -\omega * s * \alpha + c * \beta
24
          v = 2/r*L(u)*w
25
          return v
26 end
```

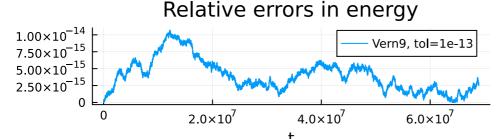
Out[8]:

vfromU (generic function with 1 method)

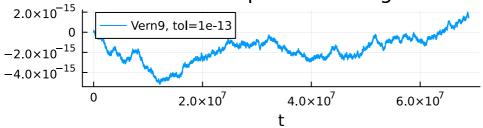
In [22]:

```
E0 = Energy(q0, v0, parms)
   qq = map((U,\tau) \rightarrow qfromU(U,BigFloat(\tau),parms), sol.u, sol.t)
 2
 3
   vv = map((U,\tau) \rightarrow vfromU(U,BigFloat(\tau),parms), sol.u, sol.t)
   tt = [U[9]  for U  in sol.u]
 5
    Eerrs = map((q,v) \rightarrow abs(Energy(q,v,parms) / E0 - 1), qq, vv)
 6
 7
 8
 9
   pl1 = plot(title="Relative errors in energy", xlabel="t")
10
   plot!(pl1, tt, Eerrs, label="Vern9, tol=1e-13")
11
   A0 = ZAngularMomentum(q0, v0)
12
   Aerrs = map((q,v) \rightarrow ZAngularMomentum(q,v) / A0 - 1, qq, vv)
13
14
15
   pl2 = plot(title="Relative errors in z component of angular momentum", xlabel=
   plot!(pl2, tt, Aerrs, label="Vern9, tol=1e-13")
16
17
   plot(pl1,pl2, layout=(2,1), size=(500,300))
18
```

Out[22]:



Relative errors in z component of angular mom



Global errors in positions (q) and physical time (t)

In order to estimate the global errors in q adn t of the numerical solution obtained above, we will compute a new numerical approximation with high precision: tol=1e-20, and higher precision arithmetic (BigFloat).

In [10]:

```
prob_BF = ODEProblem(fODE!, BigFloat.(U0), BigFloat.(tspan), BigFloat.(parms))

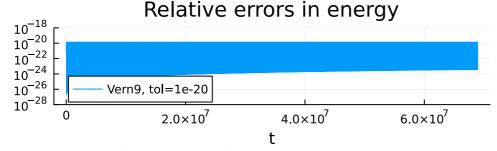
dtime sol_ex = solve(prob_BF, Vern9(), abstol=1e-20, reltol=1e-20, saveat=times
```

```
44.347835 seconds (472.26 M allocations: 21.264 GiB, 8.81% gc time, 15.47\% compilation time)
```

In [28]:

```
qq_ex = map((U,\tau) \rightarrow qfromU(U,\tau,parms), sol_ex.u, sol_ex.t)
 1
   vv_ex = map((U,\tau) \rightarrow vfromU(U,\tau,parms), sol_ex.u, sol_ex.t)
 2
 3
   tt_ex = [U[9] for U in sol_ex.u]
   EE = map((q,v) \rightarrow abs(Energy(q,v,parms)), qq ex, vv ex)
   Eerrs_ex = abs.(EE_ex ./ EE_ex[1] .- 1)
   AA = x = map((q,v) \rightarrow abs(ZAngularMomentum(q,v)), qq = x, vv = x)
 7
   Aerrs_ex = abs.(AA_ex ./ AA_ex[1] .- 1)
 8
 9
   pl1 = plot(title="Relative errors in energy", xlabel="t", yscale=:log10, ylims
10
   plot!(pl1, tt_ex, Eerrs_ex, label="Vern9, tol=1e-20", legend=:bottomleft)
11
12
13
   pl2 = plot(title="Relative errors in z-angular momentum", xlabel="t", yscale=:
14
   plot!(pl2, tt_ex,Aerrs_ex,label="Vern9, tol=1e-20")
15
16
17
   plot(pl1,pl2, layout=(2,1), size=(500,300))
```

Out[28]:



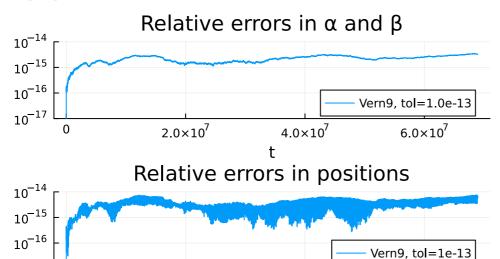


In [23]:

```
1 errors = map((u,u_ex) -> norm(u[1:8]-u_ex[1:8])/norm(u_ex[1:8]), sol.u, sol_ex
2
3 pl1 = plot(tt_ex, errors, title="Relative errors in α and β", label="Vern9, to
4    yscale=:log10, ylims=(le-17,le-14),xlabel="t", legend=:bottomright);
5 
6 qerrs = norm.(qq-qq_ex) ./ norm.(qq_ex) .+ eps(0.01)
7 pl2 = plot(title="Relative errors in positions", xlabel="t", yscale=:log10, yl
8 plot!(pl2, tt_ex, qerrs, label="Vern9, tol=1e-13", legend=:bottomright)
9 
10 plot(pl1, pl2, layout = (2,1), size=(500,300))
```

Out[23]:

10 -17



 4.0×10^{7}

t

 6.0×10^{7}

 2.0×10^{7}

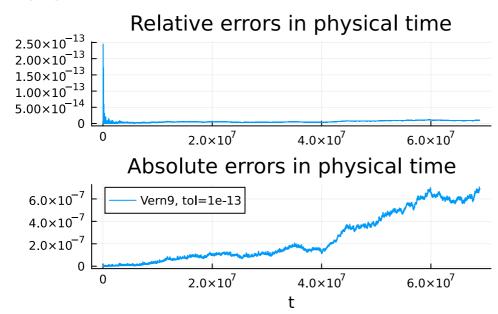
In [27]:

```
t_rel_errs = abs.(tt-tt_ex) ./ abs.(tt_ex)
pl1 = plot(tt_ex, t_rel_errs, legend=false, title="Relative errors in physical"

terrs = norm.(tt-tt_ex)
pl2 = plot(title="Absolute errors in physical time", xlabel="t")
plot!(pl2, tt_ex, terrs, label="Vern9, tol=le-13")

plot(pl1,pl2, layout=(2,1), size=(500,300))
```

Out[27]:



Taylor-Fourier integration

We next integrate the problem above with our Taylor-Fourier integrator

```
In [14]:
```

```
1 include("J2_VOP_ODE_TF.jl") # include the functions that define the ODE as re
2 # our TaylorFourier integrator
```

Out[14]:

J2_VOP_cache_init (generic function with 1 method)

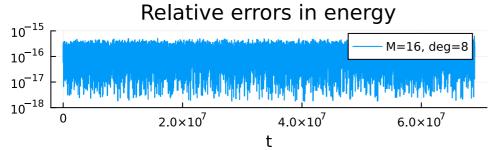
In [16]:

```
deg = 8
tf_cache = J2_VOP_cache_init(U0,parms,ω,deg,M)
prob_TF = PeriodicODEProblem(TF_ODE!, tf_cache, U0, ω)
sol_TF = TaylorFourierSolve(prob_TF,deg,M,n_orbits);
```

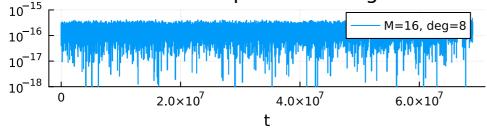
In [25]:

```
E0 = Energy(q0, v0, parms)
 1
   qq_TF = map((U,\tau) \rightarrow qfromU(U,BigFloat(\tau),parms), sol_TF.u, sol_TF.t)
 2
 3
   vv_TF = map((U,\tau) \rightarrow vfromU(U,BigFloat(\tau),parms), sol_TF.u, sol_TF.t)
   tt_TF = [U[9] for U in sol_TF.u]
   Eerrs = map((q,v) -> abs(Energy(q,v,parms) / E0 - 1), qq_TF, vv_TF) .+ eps(0.0
 5
 6
 7
 8
 9
   pl1 = plot(title="Relative errors in energy", xlabel="t", yscale=:log10, ylims
10
   plot!(pl1, tt_TF, Eerrs, label="M=$M, deg=$deg")
11
12
   A0 = ZAngularMomentum(q0, v0)
   Aerrs = map((q,v) -> ZAngularMomentum(q,v) / A0 - 1, qq_TF, vv_TF) .+ eps(0.01)
13
14
15
   pl2 = plot(title="Relative errors in z component of angular momentum", xlabel=
                      yscale=:log10, ylims=(1e-18,1e-15))
16
17
   plot!(pl2, tt_TF, Aerrs, label="M=$M, deg=$deg")
18
19
   plot(pl1,pl2, layout=(2,1), size=(500,300))
```

Out[25]:

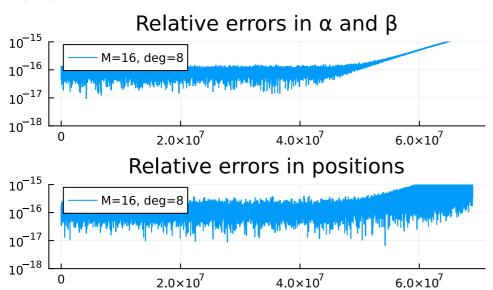


Relative errors in z component of angular momer



In [24]:

Out[24]:

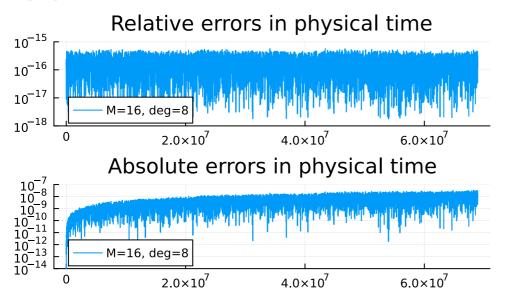


t

In [26]:

```
t rel errs =
                 abs.(tt_TF-tt_ex) ./ abs.(tt_ex) .+ eps(0.01)
2
   pl1 = plot(tt_ex, t_rel_errs, title="Relative errors in physical time",
                                                                              label
3
                 yscale=:log10, ylims=(1e-18,1e-15), legend=:bottomleft);
4
5
   terrs = norm.(tt TF-tt ex) .+ eps(0.01)
6
   pl2 = plot(title="Absolute errors in physical time", xlabel="t",
7
                  yscale=:log10, ylims=(1e-14,1e-7), legend=:bottomleft)
   plot!(pl2, tt_ex, terrs, label="M=$M, deg=$deg")
8
9
   plot(pl1,pl2, layout=(2,1), size=(500,300))
10
```

Out[26]:



t

In [20]:

```
1 @btime solve(prob, Vern9(), abstol=1e-13, reltol=1e-13,saveat=times)
2 @btime TaylorFourierSolve(prob_TF,deg,M,n_orbits);
```

```
30.123 ms (85970 allocations: 11.10 MiB) 2.413 ms (13260 allocations: 2.06 MiB)
```

In this example, our TaylorFourier integrator gives more precision with 15 times less CPU time.

In []: