# Data-Driven System Identification etc

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# **Abstract**

## English

The English abstract.

### **Afrikaans**

Die Afrikaanse uittreksel.

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## **Nomenclature**

#### Variables and functions

p(x) Probability density function with respect to variable x.

P(A) Probability of event A occurring.

 $\varepsilon$  The Bayes error.

 $\varepsilon_u$  The Bhattacharyya bound.

B The Bhattacharyya distance.

S An HMM state. A subscript is used to refer to a particular state, e.g.  $s_i$ 

refers to the  $i^{\rm th}$  state of an HMM.

S A set of HMM states.

F A set of frames.

Observation (feature) vector associated with frame f.

 $\gamma_s(\mathbf{o}_f)$  A posteriori probability of the observation vector  $\mathbf{o}_f$  being generated by

HMM state s.

 $\mu$  Statistical mean vector.

 $\Sigma$  Statistical covariance matrix.

 $L(\mathbf{S})$  Log likelihood of the set of HMM states  $\mathbf{S}$  generating the training set

observation vectors assigned to the states in that set.

 $\mathcal{N}(\mathbf{x}|\mu,\Sigma)$  Multivariate Gaussian PDF with mean  $\mu$  and covariance matrix  $\Sigma$ .

The probability of a transition from HMM state  $s_i$  to state  $s_j$ .

N Total number of frames or number of tokens, depending on the context.

D Number of deletion errors.

I Number of insertion errors.

S Number of substitution errors.

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#### Acronyms and abbreviations

AE Afrikaans English

AID accent identification

ASR automatic speech recognition

AST African Speech Technology

CE Cape Flats English

DCD dialect-context-dependent

DNN deep neural network

G2P grapheme-to-phoneme

GMM Gaussian mixture model

HMM hidden Markov model

HTK Hidden Markov Model Toolkit

IE Indian South African English

IPA International Phonetic Alphabet

LM language model

LMS language model scaling factor

MFCC Mel-frequency cepstral coefficient

MLLR maximum likelihood linear regression

OOV out-of-vocabulary

PD pronunciation dictionary

PDF probability density function

SAE South African English

SAMPA Speech Assessment Methods Phonetic Alphabet

## Chapter 1

## Introduction

The last few years have seen great advances in speech recognition. Much of this progress is due to the resurgence of neural networks; most speech systems now rely on deep neural networks (DNNs) with millions of parameters [?,?]. However, as the complexity of these models has grown, so has their reliance on labelled training data. Currently, system development requires large corpora of transcribed speech audio data, texts for language modelling, and pronunciation dictionaries. Despite speech applications becoming available in more languages, it is hard to imagine that resource collection at the required scale would be possible for all 7000 languages spoken in the world today.

I really like apples.

## 1.1. Section heading

This is some section with two table in it: Table ?? and Table ??.

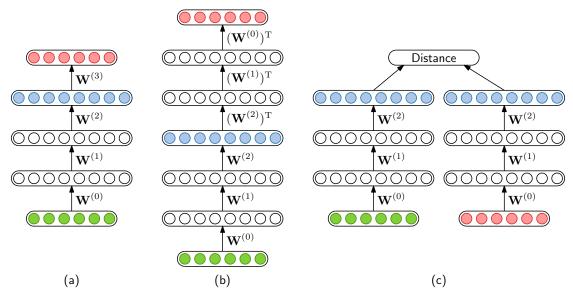
**Table 1.1:** Performance of the unconstrained segmental Bayesian model on TIDigits1 over iterations in which the reference set is refined.

Metric	1	2	3	4	5
WER (%)	35.4	23.5	21.5	21.2	22.9
Average cluster purity (%)	86.5	89.7	89.2	88.5	86.6
Word boundary $F$ -score (%)	70.6	72.2	71.8	70.9	69.4
Clusters covering $90\%$ of data	20	13	13	13	13

**Table 1.2:** A table with an example of using multiple columns.

	Accurac		
Model	Intermediate	Output	Bitrate
Baseline	27.5	26.4	116
VQ-VAE	26.0	22.1	190
CatVAE	28.7	24.3	215

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**Figure 1.1:** (a) The cAE as used in this chapter. The encoding layer (blue) is chosen based on performance on a development set. (b) The cAE with symmetrical tied weights. The encoding from the middle layer (blue) is always used. (c) The siamese DNN. The cosine distance between aligned frames (green and red) is either minimized or maximized depending on whether the frames belong to the same (discovered) word or not.

This is a new page, showing what the page headings looks like, and showing how to refer to a figure like Figure ??.

# Chapter 2

# Modelling

This chapter discusses the mathematical modelling of a quadrotor with a suspended payload which is based on a practical quadrotor UAV named Honeybee. The model is first derived as a 2D model. The system identification and control system techniques in later chapters will then be explained based on the 2D model to avoid unnecessary complexity. Finally, it will be described how this model and the techniques in later chapters are extended to the 3D case. This 3D mathematical model will be used in a nonlinear simulation of a quadrotor and suspended payload.

## 2.1. Coordinate frames

- 2.2. States
- 2.3. Forces and moments
- 2.4. Lagrangian mechanics
- 2.5. Linearised model
- 2.6. Discretised model
- 2.7. Model verification

## Chapter 3

# System identification

System identification is the process of creating mathematical models of a dynamical system by using input and output measurements of the system. Two major approaches are used to represent the dynamics of such a system, resulting in white-box or black-box models.

## 3.1. Parameter estimation

#### 3.1.1. White-box models

In white-box models the physics of a model are understood by the user. These models are therefore determined from first principles. This can be done by modelling physical processes with techniques like Lagrangian mechanics or Newton equations. In this case, the mathematical relations between physical properties are predefined in the modelling phase. System identification is then reduced to parameter estimation to determine values for the parameters used in die model.

This is the approach used by [?] and [?] for swing damping control of a quadrotor with an unknown suspended payload. The system was modelled as two rigid bodies connected by a link and the following assumptions were made regarding the suspended payload:

- The payload is a point mass.
- The link is massless.
- The link is rigid.
- The link is attached to the CoM of the quadrotor.

The only unknown parameters in the quadrotor and payload model is the payload mass and link length. These parameters are first estimated and then inserted into the predefined, linearised model. This model is used by a LQR controller to damp swing angles while also controlling the vehicle.

The approach works well for systems with predictable dynamics, but it is not very adaptable. The payload considered by [?] and [?] is limited to a small rigid mass suspended from the quadrotor by a non-stretching cable. In this use case it was shown that a LQR controller successfully controls a quadrotor while minimising payload swing angles. However, if a payload or cable is used that violates one of the modelling assumptions, the predefined model no longer accurately represent the system. Since the controller is dependent on this model, the mismatch between the model and actual dynamics may result in undesirable controller behaviour.

Parameter estimation

### 3.1.2. Payload mass estimation

RLS

### 3.1.3. Cable length estimation

The cable length is estimated from the measurement of natural frequency of the swinging payload. As described by [?], the natural frequency is given by:

$$\omega_n = \sqrt{\frac{g}{l} \cdot \frac{m_q + m_p}{m_q}} \tag{3.1}$$

The natural frequency is measured by performing a FFT on the payload swing angle response after a position step by the quadrotor. The dominant frequency identified by the FFT during free swing is the natural frequency of the payload.

?? shows the payload swing angle after the system is stimulated by a position step setpoint. As shown in ?? the first few seconds of the step response are not used in the FFT. This is to minimise the effect of the quadrotor controllers on the swing angle frequency by excluding the transient response in the FFT.

?? shows the resulting amplitude spectrum of the payload swing angle response.

Since  $m_q$  and g is known, and  $m_p$  and  $\omega_n$  has been estimated, l can now be determined from ??

## 3.2. Data-driven system identification

#### 3.2.1. Black-box models

In contrast to white-box models, black-box models do not require predefined mathematical relations between system parameters. No prior knowledge of the physics of the system are considered and no modelling assumptions are made. Black-box techniques determine the mathematical relationship between inputs and outputs of a system based only on measurement data. This is referred to as data-driven system identification.

Black-box models can be categorised as either non-linear or linear models. Non-linear models are often more accurate than linear models because complex, real-world dynamics are better approximated by non-linear systems. The dynamics of a quadrotor and suspended payload are also non-linear. Examples of black box models with quadrotors and payloads in literature???

However, non-linear models are inherently more complex than linear models. Controllers that use non-linear models are usually more computationally complex than those with linear models. Control archetictures for quadrotors used in practical applications are mostly implemented on onboard hardware. Therefore there is value in low-complexity, linear models since these may be simple enough to execute on low cost hardware. trade-off between accuracy and complexity. Non-linear models may require control implementations that are too computationally expensive and may not be practically realisable on the available hardware on a quadrotor.

#### 3.2.2. DMD

#### 3.2.3. HAVOK

# **Chapter 4 Summary and Conclusion**

# Appendix A<br/> Project Planning Schedule

This is an appendix.

# Appendix B Outcomes Compliance

This is another appendix.