

# Unmanned Aerial Vehicles

## Modern Control Design

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February 2018

# Linear state space models

## State space models

# Linear state space models

## Linearization

# Linear state space models

## Linear state space model

$$\begin{cases} \dot{\mathbf{x}}(t) = \mathbf{A}(t)\mathbf{x}(t) + \mathbf{B}(t)\mathbf{u}(t) \\ \mathbf{y}(t) = \mathbf{C}(t)\mathbf{x}(t) + \mathbf{D}(t)\mathbf{u}(t) \end{cases}$$

# Linear state space models

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The term  $\mathbf{D}(t)\mathbf{u}(t)$  corresponds to a direct effect of the input on the output. That is not possible in general, hence we will always consider  $\mathbf{D}(t) = \mathbf{0}$ .

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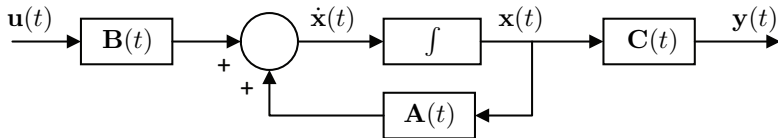
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A special case occurs when the system matrices are constant, yielding the so-called linear time invariant (LTI) system:

$$\begin{cases} \dot{\mathbf{x}}(t) = \mathbf{A}\mathbf{x}(t) + \mathbf{B}\mathbf{u}(t) \\ \mathbf{y}(t) = \mathbf{C}\mathbf{x}(t) \end{cases}$$

# Linear state space models

State space vector block diagram



# Linear state space models

Conversion from state-space to transfer function matrix for LTI systems

- 1 Take the Laplace transform of both sides considering zero initial conditions:

$$\begin{cases} s\mathbf{X}(s) = \mathbf{A}\mathbf{X}(s) + \mathbf{B}\mathbf{U}(s) \\ \mathbf{Y}(s) = \mathbf{C}\mathbf{X}(s) \end{cases}$$



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- 2 Determine  $\mathbf{Y}(s)$  as a function of  $\mathbf{U}(s)$ :

$$\mathbf{Y}(s) = \mathbf{C}\mathbf{X}(s) = \mathbf{C} (s\mathbf{I} - \mathbf{A})^{-1} \mathbf{B}\mathbf{U}(s)$$

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$$\begin{cases} \dot{\mathbf{x}}(t) = \mathbf{A}\mathbf{x}(t) + \mathbf{B}\mathbf{u}(t) \\ \mathbf{y}(t) = \mathbf{C}\mathbf{x}(t) \end{cases} \quad \Leftrightarrow \quad \mathbf{G}(s) = \mathbf{C} (s\mathbf{I} - \mathbf{A})^{-1} \mathbf{B}$$

# Linear state space models

Solution of the unforced system and transition matrix

## Theorem

For any  $t_0$  and  $\mathbf{x}_0$ , with  $\mathbf{A}(t)$  continuous, the linear state equation

$$\begin{cases} \dot{\mathbf{x}}(t) = \mathbf{A}(t)\mathbf{x}(t) \\ \mathbf{x}(t_0) = \mathbf{x}_0 \end{cases}$$

has a unique solution given by

$$\mathbf{x}(t) = \Phi(t, t_0) \mathbf{x}_0,$$

where  $\Phi(t, t_0)$  is the transition matrix associated with  $\mathbf{A}(t)$ , given by the Peano-Baker series

$$\Phi(t, t_0) := \mathbf{I} + \int_{t_0}^t \mathbf{A}(\sigma_1) d\sigma_1 + \int_{t_0}^t \mathbf{A}(\sigma_1) \int_{t_0}^{\sigma_1} \mathbf{A}(\sigma_2) d\sigma_2 d\sigma_1 + \dots$$

# Linear state space models

Complete solution

$$\begin{cases} \dot{\mathbf{x}}(t) = \mathbf{A}(t)\mathbf{x}(t) + \mathbf{B}(t)\mathbf{u}(t) \\ \mathbf{y}(t) = \mathbf{C}(t)\mathbf{x}(t) \\ \mathbf{x}(t_0) = \mathbf{x}_0 \end{cases}$$

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$$\begin{cases} \mathbf{x}(t) = \Phi(t, t_0) \mathbf{x}_0 + \int_{t_0}^t \Phi(t, \sigma) \mathbf{B}(\sigma) \mathbf{u}(\sigma) d\sigma \\ \mathbf{y}(t) = \mathbf{C}(t) \Phi(t, t_0) \mathbf{x}_0 + \int_{t_0}^t \mathbf{C}(t) \Phi(t, \sigma) \mathbf{B}(\sigma) \mathbf{u}(\sigma) d\sigma \end{cases}$$

# Linear state space models

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$$\begin{cases} \mathbf{x}(t) = \Phi(t, t_0) \mathbf{x}_0 + \int_{t_0}^t \Phi(t, \sigma) \mathbf{B}(\sigma) \mathbf{u}(\sigma) d\sigma \\ \mathbf{y}(t) = \mathbf{C}(t) \Phi(t, t_0) \mathbf{x}_0 + \int_{t_0}^t \mathbf{C}(t) \Phi(t, \sigma) \mathbf{B}(\sigma) \mathbf{u}(\sigma) d\sigma \end{cases}$$

Check the solution of  $\mathbf{x}(t)$  by direct differentiation, noting that

$$\frac{d}{dt} \Phi(t, t_0) = \mathbf{A}(t) \Phi(t, t_0).$$

It is also possible to check this by direct computation.

# Linear state space models

## Verification of the solution (1)

$$\textcircled{1} \quad \frac{d}{dt} \Phi(t, t_0) = \mathbf{A}(t) \Phi(t, t_0)$$

$$\begin{aligned} \frac{d}{dt} \Phi(t, t_0) &= \frac{d}{dt} (\mathbf{I}) + \frac{d}{dt} \left( \int_{t_0}^t \mathbf{A}(\sigma_1) d\sigma_1 \right) \\ &\quad + \frac{d}{dt} \left( \int_{t_0}^t \mathbf{A}(\sigma_1) \int_{t_0}^{\sigma_1} \mathbf{A}(\sigma_2) d\sigma_2 d\sigma_1 \right) \\ &\quad + \frac{d}{dt} \left( \int_{t_0}^t \mathbf{A}(\sigma_1) \int_{t_0}^{\sigma_1} \mathbf{A}(\sigma_2) \int_{t_0}^{\sigma_2} \mathbf{A}(\sigma_3) d\sigma_3 d\sigma_2 d\sigma_1 \right) \\ &\quad + \dots \end{aligned}$$

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# Linear state space models

## Verification of the solution (2)

2 Complete solution:

$$\dot{\mathbf{x}}(t) = \frac{d}{dt} \left( \Phi(t, t_0) \mathbf{x}_0 + \int_{t_0}^t \Phi(t, \sigma) \mathbf{B}(\sigma) \mathbf{u}(\sigma) d\sigma \right)$$

# Linear state space models

## Verification of the solution (2)

### 2 Complete solution:

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# Linear state space models

## Verification of the solution (2)

### 2 Complete solution:

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### 2 Complete solution:

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# Linear state space models

## Transition matrix for LTI systems

For LTI systems, the transition matrix can be easily computed...

$$\begin{aligned}\Phi(t, t_0) &:= \mathbf{I} + \int_{t_0}^t \mathbf{A} d\sigma_1 + \int_{t_0}^t \mathbf{A} \int_{t_0}^{\sigma_1} \mathbf{A} d\sigma_2 d\sigma_1 \\ &\quad + \int_{t_0}^t \mathbf{A} \int_{t_0}^{\sigma_1} \mathbf{A} \int_{t_0}^{\sigma_2} \mathbf{A} d\sigma_3 d\sigma_2 d\sigma_1 + \dots \\ &= \mathbf{I} + (t - t_0) \mathbf{A} + \frac{(t - t_0)^2}{2} \mathbf{A}^2 + \frac{(t - t_0)^3}{3} \mathbf{A}^3 + \dots \\ &= \mathbf{e}^{(t-t_0)\mathbf{A}}\end{aligned}$$

# Linear state space models

Solution for LTI systems

$$\begin{cases} \dot{\mathbf{x}}(t) = \mathbf{A}\mathbf{x}(t) + \mathbf{B}\mathbf{u}(t) \\ \mathbf{y}(t) = \mathbf{C}\mathbf{x}(t) \\ \mathbf{x}(t_0) = \mathbf{x}_0 \end{cases}$$



# Linear state space models

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$$\begin{cases} \mathbf{x}(t) = \mathbf{e}^{\mathbf{A}(t-t_0)}\mathbf{x}_0 + \int_{t_0}^t \mathbf{e}^{\mathbf{A}(t-\sigma)}\mathbf{B}\mathbf{u}(\sigma) d\sigma \\ \mathbf{y}(t) = \mathbf{C}\mathbf{e}^{\mathbf{A}(t-t_0)}\mathbf{x}_0 + \int_{t_0}^t \mathbf{C}\mathbf{e}^{\mathbf{A}(t-\sigma)}\mathbf{B}\mathbf{u}(\sigma) d\sigma \end{cases}$$

# Linear state space models

Exponential matrix (distinct eigenvalues)

**Eigenstructure of  $\mathbf{A} \in \mathbb{R}^{n \times n}$ :**

$$\mathbf{A}_i \mathbf{v}_i = \lambda_i \mathbf{v}_i$$

$$\mathbf{w}_i^T \mathbf{A} = \lambda_i \mathbf{w}_i^T$$

- $i$ -th eigenvalue (all distinct):  $\lambda_i \in \mathbb{C}$
- Right eigenvector:  $\mathbf{v}_i$
- Left eigenvector:  $\mathbf{w}_i^T$

**Diadic formula:**

$$\mathbf{A} = \sum_{i=1}^n \lambda_i \mathbf{v}_i \mathbf{w}_i^T$$

**Orthonormal vectors:**

$$\mathbf{w}_i^T \mathbf{v}_j = \begin{cases} 1, & i = j \\ 0, & i \neq j \end{cases}$$

**Exponential matrix:**

$$e^{\mathbf{A}t} = \sum_{i=1}^n e^{\lambda_i t} \mathbf{v}_i \mathbf{w}_i^T$$

### Unforced LTI dynamics modal response:

$$\begin{cases} \dot{\mathbf{x}}(t) = \mathbf{A}\mathbf{x}(t) \\ \mathbf{x}(t_0) = \mathbf{x}_0 \end{cases}$$

$$\mathbf{x}(t) = e^{(t-t_0)\mathbf{A}}\mathbf{x}_0 = \sum_{i=1}^n e^{\lambda_i(t-t_0)} \mathbf{v}_i (\mathbf{w}_i^T \mathbf{x}_0)$$

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#### Definition

Consider the unforced system  $\dot{\mathbf{x}}(t) = \mathbf{A}\mathbf{x}(t)$ , with  $\mathbf{x}(t_0) = \mathbf{x}_0$ . The  $i$ -th mode is  $e^{\lambda_i(t-t_0)}\mathbf{v}_i$ , defined by the right eigenvector  $\mathbf{v}_i$  and the associated eigenvalue  $\lambda_i$ .

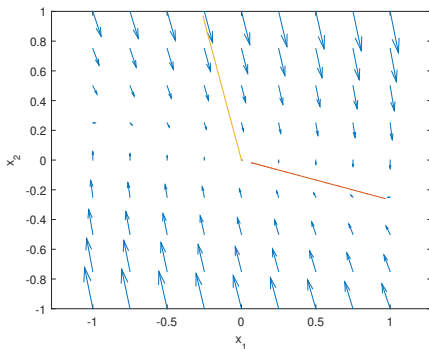
# Linear state space models

## Modes and phase portrait

### Unforced LTI dynamics modal response:

$$\mathbf{A} = \begin{bmatrix} 0 & 1 \\ -1 & -4 \end{bmatrix}$$

$$\mathbf{A}\mathbf{V} = \mathbf{V}\mathbf{\Lambda}, \quad \mathbf{\Lambda} = \begin{bmatrix} -0.2679 & 0 \\ 0 & -3.7321 \end{bmatrix}, \quad \mathbf{V} = \begin{bmatrix} 0.9659 & -0.2588 \\ -0.2588 & 0.9659 \end{bmatrix}$$



# Controllability and observability

## Motivation (1)

**Can I fly, or park, an airplane laterally?**



# Controllability and observability

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# Controllability and observability

## Motivation (2)

**Given the linear state equation**

$$\dot{\mathbf{x}}(t) = \mathbf{A}(t)\mathbf{x}(t) + \mathbf{B}(t)\mathbf{u}(t)$$

**with initial condition  $\mathbf{x}(0) = \mathbf{0}$  and given an arbitrary state  $\mathbf{x}_f$ , is it possible to make**

$$\mathbf{x}(t_f) = \mathbf{x}_f$$

**by appropriate choice of  $\mathbf{u}(t)$ ?**



# Controllability and observability

## Motivation (2)

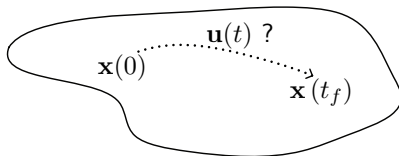
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# Controllability and observability

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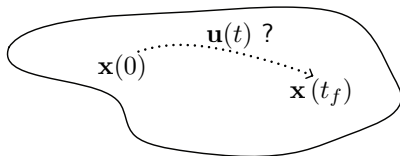
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**by appropriate choice of  $\mathbf{u}(t)$ ?**



**The answer depends on the pair of matrices  $\mathbf{A}(t)$  and  $\mathbf{B}(t)$ .**

# Controllability and observability

## Definition of controllability

### Definition of controllability

The linear state equation

$$\dot{\mathbf{x}}(t) = \mathbf{A}(t)\mathbf{x}(t) + \mathbf{B}(t)\mathbf{u}(t)$$

is called controllable on  $[t_0, t_f]$  if, given any initial state  $\mathbf{x}(0) = \mathbf{x}_0$ , there exists a continuous input signal  $\mathbf{u}(t)$  such that the corresponding solution satisfies  $\mathbf{x}(t_f) = \mathbf{0}$ .

# Controllability and observability

## Controllability for LTI system

For LTI system there exists a simple controllability test!

### Theorem

The LTI system

$$\dot{\mathbf{x}}(t) = \mathbf{A}\mathbf{x}(t) + \mathbf{B}\mathbf{u}(t)$$

is controllable if and only if the controllability matrix

$$\mathbf{C} := [\mathbf{B} \quad \mathbf{A}\mathbf{B} \quad \mathbf{A}^2\mathbf{B} \quad \dots \quad \mathbf{A}^{n-1}\mathbf{B}]$$

has rank equal to the number of states of the system, i.e.,

$$\text{rank } \mathbf{C} = n, \mathbf{x} \in \mathbb{R}^n.$$

# Controllability and observability

## Examples of controllability analysis for LTI systems

$$\dot{\mathbf{x}}(t) = \mathbf{A}\mathbf{x}(t) + \mathbf{B}\mathbf{u}(t)$$

- Controllable system:

$$\mathbf{A} = \begin{bmatrix} 0 & 1 \\ 0 & 0 \end{bmatrix} \quad \mathbf{B} = \begin{bmatrix} 0 \\ 1 \end{bmatrix}$$

$$\mathbf{C} := [\mathbf{B} \quad \mathbf{AB}] = \begin{bmatrix} 0 & 1 \\ 1 & 0 \end{bmatrix}$$

- Uncontrollable system:

$$\mathbf{A} = \begin{bmatrix} 0 & 1 \\ 0 & 0 \end{bmatrix} \quad \mathbf{B} = \begin{bmatrix} 1 \\ 0 \end{bmatrix}$$

$$\mathbf{C} := [\mathbf{B} \quad \mathbf{AB}] = \begin{bmatrix} 0 & 0 \\ 1 & 0 \end{bmatrix}$$

# Controllability and observability

## Definition of observability

### Definition of observability

The linear state equation

$$\begin{cases} \dot{\mathbf{x}}(t) = \mathbf{A}(t)\mathbf{x}(t) \\ \mathbf{y}(t) = \mathbf{C}(t)\mathbf{x}(t) \end{cases}$$

is called observable on  $[t_0, t_f]$  if any initial state  $\mathbf{x}(0) = \mathbf{x}_0$  is uniquely determined by the corresponding response  $\mathbf{y}(t)$  for  $t \in [t_0, t_f]$ .

# Controllability and observability

## Observability and non-zero input case

**Question:** Does anything change if we consider

$$\begin{cases} \dot{\mathbf{x}}(t) = \mathbf{A}(t)\mathbf{x}(t) + \mathbf{B}(t)\mathbf{u}(t) \\ \mathbf{y}(t) = \mathbf{C}(t)\mathbf{x}(t) \end{cases}$$

with a known input  $\mathbf{u}(t)$ ,  $t \in [t_0, t_f]$ ?

# Controllability and observability

## Observability and non-zero input case

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$$\begin{cases} \dot{\mathbf{x}}(t) = \mathbf{A}(t)\mathbf{x}(t) + \mathbf{B}(t)\mathbf{u}(t) \\ \mathbf{y}(t) = \mathbf{C}(t)\mathbf{x}(t) \end{cases}$$

with a known input  $\mathbf{u}(t)$ ,  $t \in [t_0, t_f]$ ?

**No!**

When the input is known, we can always subtract from the output the component that depends on the input, and thus compute the output corresponding to the zero-input case.

**Conclusion:** For LTV systems, the observability is independent of the input. In general, that is not the case for nonlinear systems, which makes the analysis much more intricate...



# Controllability and observability

## Observability for LTI system

For LTI system there exists a simple observability test!

### Theorem

The LTI system

$$\dot{\mathbf{x}}(t) = \mathbf{A}\mathbf{x}(t) + \mathbf{B}\mathbf{u}(t)$$

is observable if and only if the observability matrix

$$\mathcal{O} := \begin{bmatrix} \mathbf{C} \\ \mathbf{CA} \\ \vdots \\ \mathbf{CA}^{n-1} \end{bmatrix}$$

has rank equal to the number of states of the system, i.e.,

$$\text{rank } \mathcal{O} = n, \mathbf{x} \in \mathbb{R}^n.$$

# Controllability and observability

## Examples of observability analysis for LTI systems

$$\begin{cases} \dot{\mathbf{x}}(t) = \mathbf{A}\mathbf{x}(t) \\ \mathbf{y}(t) = \mathbf{C}\mathbf{x}(t) \end{cases}$$

- Observable system:

$$\mathbf{A} = \begin{bmatrix} 0 & 1 \\ 0 & 0 \end{bmatrix} \quad \mathbf{C} = \begin{bmatrix} 1 & 0 \end{bmatrix}$$

$$\mathcal{O} := \begin{bmatrix} \mathbf{C} \\ \mathbf{CA} \end{bmatrix} = \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix}$$

- Unobservable system:

$$\mathbf{A} = \begin{bmatrix} 0 & 1 \\ 0 & 0 \end{bmatrix} \quad \mathbf{C} = \begin{bmatrix} 0 & 1 \end{bmatrix}$$

$$\mathcal{O} := \begin{bmatrix} \mathbf{C} \\ \mathbf{CA} \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ 0 & 0 \end{bmatrix}$$

# Controllability and observability

## Matlab examples

### Controllability and observability matrices

```
>> A = [0 1; 0 0]; B = [0; 1]; CM = ctrb(A, B)
```

```
CM =
```

```
0 1
```

```
1 0
```

```
>> A = [0 1; 0 0]; C = [1, 0]; OM = obsv(A, C)
```

```
OM =
```

```
1 0
```

```
0 1
```

```
>> rank(CM), rank(OM)
```

```
ans =
```

```
2
```

```
ans =
```

```
2
```

# Controllability and observability

## Controllability and observability Gramians

For LTV systems there also exist powerful tools for controllability and observability analysis.

### Definition of controllability Gramian

The controllability Gramian associated with the pair  $(\mathbf{A}(t), \mathbf{B}(t))$  on  $[t_0, t_f]$  is given by

$$\mathbf{W}_C(t_0, t_f) := \int_{t_0}^{t_f} \phi(t_0, t) \mathbf{B}(t) \mathbf{B}^T(t) \phi^T(t_0, t) dt.$$

### Definition of observability Gramian

The observability Gramian associated with the pair  $(\mathbf{A}(t), \mathbf{C}(t))$  on  $[t_0, t_f]$  is given by

$$\mathbf{W}_O(t_0, t_f) := \int_{t_0}^{t_f} \phi^T(t, t_0) \mathbf{C}^T(t) \mathbf{C}(t) \phi(t, t_0) dt.$$

# Controllability and observability

Controllability and observability for LTV systems and nonlinear systems

## Theorem

The LTV system

$$\begin{cases} \dot{\mathbf{x}}(t) = \mathbf{A}(t)\mathbf{x}(t) + \mathbf{B}(t)\mathbf{u}(t) \\ \mathbf{y}(t) = \mathbf{C}(t)\mathbf{x}(t) \end{cases}$$

is controllable (observable) if and only if the controllability Gramian  $\mathcal{W}_C(t_0, t_f)$  (observability Gramian  $\mathcal{W}_O(t_0, t_f)$ ) is invertible.

# Controllability and observability

Controllability and observability for LTV systems and nonlinear systems

## Theorem

The LTV system

$$\begin{cases} \dot{\mathbf{x}}(t) = \mathbf{A}(t)\mathbf{x}(t) + \mathbf{B}(t)\mathbf{u}(t) \\ \mathbf{y}(t) = \mathbf{C}(t)\mathbf{x}(t) \end{cases}$$

is controllable (observable) if and only if the controllability Gramian  $\mathcal{W}_C(t_0, t_f)$  (observability Gramian  $\mathcal{W}_O(t_0, t_f)$ ) is invertible.

For nonlinear systems there are some local results but no general results...



**Is it possible to control the “behavior” of a system using linear state feedback?**

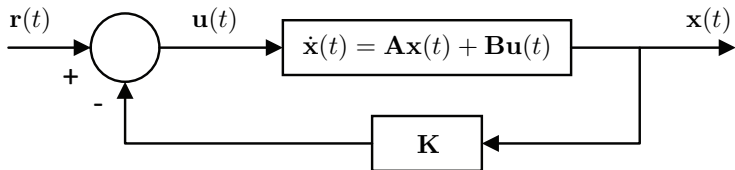
**Is it possible to control the “behavior” of a system using linear state feedback?**

Let us focus on LTI systems only, for now... What is linear state feedback?



**Is it possible to control the “behavior” of a system using linear state feedback?**

Let us focus on LTI systems only, for now... What is linear state feedback?



**Linear state feedback:**

$$\mathbf{u}(t) = \mathbf{r}(t) - \mathbf{K}\mathbf{x}(t)$$

**Is it possible to assign the poles of a LTI system using linear state feedback?**

**Is it possible to assign the poles of a LTI system using linear state feedback?**

**Theorem: Pole placement for LTI systems**

Suppose that the LTI system

$$\dot{\mathbf{x}}(t) = \mathbf{A}\mathbf{x}(t) + \mathbf{B}\mathbf{u}(t)$$

is controllable. Given any real, monic polynomial  $p(\lambda)$  of degree- $n$ , there exists a constant state feedback gain  $\mathbf{K}$  such that

$$\det(\lambda\mathbf{I} - \mathbf{A} - \mathbf{BK}) = p(\lambda).$$

# Feedback control of linear state variables

## Tools for pole placement

- 1 By direct computation (comparison of polinomials)
- 2 Bass-Gura formula
- 3 Ackermann's formula

### Matlab example

```
>> A = [0 1; 0 0]; B = [0; 1]; p = [-1+i -1-i];  
>> place(A, B, p)  
ans =  
    2.0000    2.0000
```

# Feedback control of linear state variables

Example by direct computation

$$\mathbf{A} = \begin{bmatrix} 0 & 1 \\ 0 & 0 \end{bmatrix} \quad \mathbf{B} = \begin{bmatrix} 0 \\ 1 \end{bmatrix}$$

Desired poles:

$$-1 + i, -1 - i$$

# Feedback control of linear state variables

Example by direct computation

$$\mathbf{A} = \begin{bmatrix} 0 & 1 \\ 0 & 0 \end{bmatrix} \quad \mathbf{B} = \begin{bmatrix} 0 \\ 1 \end{bmatrix}$$

Desired poles:  
 $-1 + i, -1 - i$

1 Desired characteristic polynomial:

$$p(\lambda) = [\lambda - (-1 + i)] [\lambda - (-1 - i)] = \lambda^2 + 2\lambda + 2$$

# Feedback control of linear state variables

Example by direct computation

$$\mathbf{A} = \begin{bmatrix} 0 & 1 \\ 0 & 0 \end{bmatrix} \quad \mathbf{B} = \begin{bmatrix} 0 \\ 1 \end{bmatrix} \quad \text{Desired poles: } -1 + i, -1 - i$$

- 1 Desired characteristic polynomial:

$$p(\lambda) = [\lambda - (-1 + i)] [\lambda - (-1 - i)] = \lambda^2 + 2\lambda + 2$$

- 2 Closed-loop system matrix:

$$\mathbf{A} - \mathbf{BK} = \begin{bmatrix} 0 & 1 \\ 0 & 0 \end{bmatrix} - \begin{bmatrix} 0 \\ 1 \end{bmatrix} \begin{bmatrix} k_1 & k_2 \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ -k_1 & -k_2 \end{bmatrix}$$

# Feedback control of linear state variables

Example by direct computation

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- 3 Closed-loop polynomial:

$$\det(\lambda \mathbf{I} - (\mathbf{A} - \mathbf{BK})) = \lambda^2 + k_2\lambda + k_1$$



# Feedback control of linear state variables

Example by direct computation

$$\mathbf{A} = \begin{bmatrix} 0 & 1 \\ 0 & 0 \end{bmatrix} \quad \mathbf{B} = \begin{bmatrix} 0 \\ 1 \end{bmatrix} \quad \text{Desired poles: } -1 + i, -1 - i$$

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- 3 Closed-loop polynomial:

$$\det(\lambda \mathbf{I} - (\mathbf{A} - \mathbf{BK})) = \lambda^2 + k_2\lambda + k_1$$

- 4 By comparison,  $k_1 = k_2 = 2$ .

**How do we implement linear state feedback when we don't have access to the entire state of the system?**

How do we implement linear state feedback when we don't have access to the entire state of the system?

# State observers

How do we implement linear state feedback when we don't have access to the entire state of the system?

# State observers

**Problem:** Given the dynamic system

$$\begin{cases} \dot{\mathbf{x}}(t) = \mathbf{A}(t)\mathbf{x}(t) + \mathbf{B}(t)\mathbf{u}(t) \\ \mathbf{y}(t) = \mathbf{C}(t)\mathbf{x}(t) \end{cases},$$

design a state observer that yields a “good” estimate of  $\mathbf{x}(t)$ .

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design a state observer that yields a “good” estimate of  $\mathbf{x}(t)$ .

What is a “good” estimate of  $\mathbf{x}(t)$ ?

**Why not just buy sensors to measure the state?**

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**Why not just buy sensors to measure the state?**





**Idea:** Replica of the system with feedback of the output estimation error

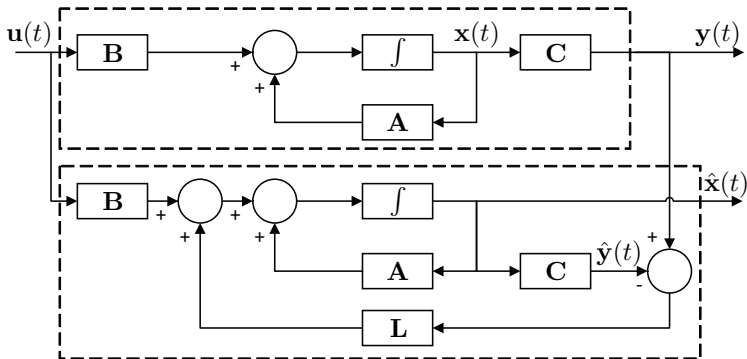
$$\dot{\hat{\mathbf{x}}}(t) = \mathbf{A}(t)\hat{\mathbf{x}}(t) + \mathbf{B}(t)\mathbf{u}(t) + \mathbf{L}(t) [\mathbf{y}(t) - \mathbf{C}(t)\hat{\mathbf{x}}(t)]$$

# Linear observers

## Luenberger observer

**Idea:** Replica of the system with feedback of the output estimation error

$$\dot{\hat{\mathbf{x}}}(t) = \mathbf{A}(t)\hat{\mathbf{x}}(t) + \mathbf{B}(t)\mathbf{u}(t) + \mathbf{L}(t)[\mathbf{y}(t) - \mathbf{C}(t)\hat{\mathbf{x}}(t)]$$



### Observer estimation:

$$\tilde{\mathbf{x}}(t) := \mathbf{x}(t) - \hat{\mathbf{x}}(t)$$

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### Observer estimation error dynamics:

$$\begin{aligned}\dot{\tilde{\mathbf{x}}}(t) &= \dot{\mathbf{x}}(t) - \dot{\hat{\mathbf{x}}}(t) \\ &= \mathbf{A}(t)\mathbf{x}(t) - \mathbf{A}(t)\hat{\mathbf{x}}(t) - \mathbf{L}(t) [\mathbf{y}(t) - \mathbf{C}(t)\hat{\mathbf{x}}(t)] \\ &= \mathbf{A}(t)\tilde{\mathbf{x}}(t) - \mathbf{L}(t) [\mathbf{C}(t)\mathbf{x}(t) - \mathbf{C}(t)\hat{\mathbf{x}}(t)] \\ &= [\mathbf{A}(t) - \mathbf{L}(t)\mathbf{C}(t)] \tilde{\mathbf{x}}(t)\end{aligned}$$

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$$\tilde{\mathbf{x}}(t) := \mathbf{x}(t) - \hat{\mathbf{x}}(t)$$

### Observer estimation error dynamics:

$$\begin{aligned}\dot{\tilde{\mathbf{x}}}(t) &= \dot{\mathbf{x}}(t) - \dot{\hat{\mathbf{x}}}(t) \\ &= \mathbf{A}(t)\mathbf{x}(t) - \mathbf{A}(t)\hat{\mathbf{x}}(t) - \mathbf{L}(t) [\mathbf{y}(t) - \mathbf{C}(t)\hat{\mathbf{x}}(t)] \\ &= \mathbf{A}(t)\tilde{\mathbf{x}}(t) - \mathbf{L}(t) [\mathbf{C}(t)\mathbf{x}(t) - \mathbf{C}(t)\hat{\mathbf{x}}(t)] \\ &= [\mathbf{A}(t) - \mathbf{L}(t)\mathbf{C}(t)] \tilde{\mathbf{x}}(t)\end{aligned}$$

**Question:** How does one design the observer gain  $\mathbf{L}(t)$ ?

For linear systems, the control and estimation problems are **dual!**

This is very clear in the LTI case...

The eigenvalues of

$$\mathbf{A} - \mathbf{LC}$$

are the same of

$$(\mathbf{A} - \mathbf{LC})^T = \mathbf{A}^T - \mathbf{C}^T \mathbf{L}^T.$$

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The eigenvalues of

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are the same of

$$(\mathbf{A} - \mathbf{L}\mathbf{C})^T = \mathbf{A}^T - \mathbf{C}^T\mathbf{L}^T.$$

## Duality

$$\mathbf{A} \longleftrightarrow \mathbf{A}^T$$

$$\mathbf{B} \longleftrightarrow \mathbf{C}^T$$

$$\mathbf{K} \longleftrightarrow \mathbf{L}^T$$

**Linear state feedback**

+

**State observers**



**Linear state feedback**

+

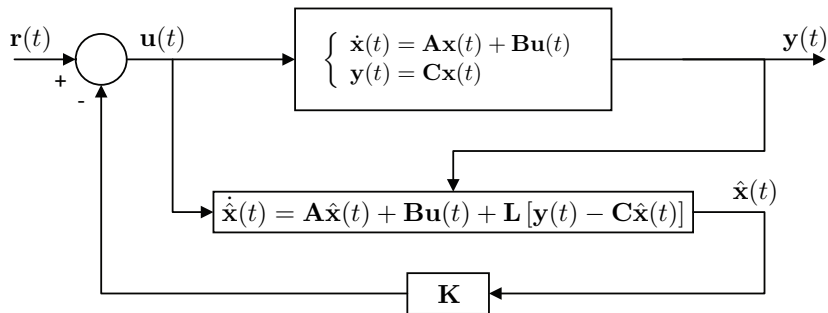
**State observers**

**Does it work?**

**Can they be designed independently?**

# Separation theorem

Control scheme with state estimation for LTI systems



# Separation theorem

## Closed-loop LTI dynamics

$$\begin{cases} \dot{\mathbf{x}}(t) = \mathbf{A}\mathbf{x}(t) - \mathbf{BK}\hat{\mathbf{x}}(t) + \mathbf{B}\mathbf{r}(t) \\ \mathbf{y}(t) = \mathbf{C}\mathbf{x}(t) \\ \dot{\hat{\mathbf{x}}}(t) = \mathbf{A}\hat{\mathbf{x}}(t) - \mathbf{BK}\hat{\mathbf{x}}(t) + \mathbf{B}\mathbf{r}(t) + \mathbf{L}[\mathbf{y}(t) - \mathbf{C}\hat{\mathbf{x}}(t)] \end{cases}$$

Recall that, for the observer, we shape the dynamics of the estimation error  $\tilde{\mathbf{x}}(t) = \mathbf{x}(t) - \hat{\mathbf{x}}(t)$ .

$$\begin{cases} \dot{\mathbf{x}}(t) = (\mathbf{A} - \mathbf{BK})\mathbf{x}(t) + \mathbf{BK}\tilde{\mathbf{x}}(t) + \mathbf{B}\mathbf{r}(t) \\ \dot{\tilde{\mathbf{x}}}(t) = (\mathbf{A} - \mathbf{LC})\tilde{\mathbf{x}}(t) \\ \mathbf{y}(t) = \mathbf{C}\mathbf{x}(t) \end{cases}$$

# Separation theorem

Closed-loop LTI dynamics: Separation principle

$$\begin{cases} \begin{bmatrix} \dot{\mathbf{x}}(t) \\ \dot{\tilde{\mathbf{x}}}(t) \end{bmatrix} = \begin{bmatrix} \mathbf{A} - \mathbf{BK} & \mathbf{BK} \\ \mathbf{0} & \mathbf{A} - \mathbf{LC} \end{bmatrix} \begin{bmatrix} \mathbf{x}(t) \\ \tilde{\mathbf{x}}(t) \end{bmatrix} + \begin{bmatrix} \mathbf{B} \\ \mathbf{0} \end{bmatrix} \mathbf{r}(t) \\ \mathbf{y}(t) = \begin{bmatrix} \mathbf{C} & \mathbf{0} \end{bmatrix} \begin{bmatrix} \mathbf{x}(t) \\ \tilde{\mathbf{x}}(t) \end{bmatrix} \end{cases}$$

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## Theorem

The eigenvalues of

$$\begin{bmatrix} \mathbf{A} - \mathbf{BK} & \mathbf{BK} \\ \mathbf{0} & \mathbf{A} - \mathbf{LC} \end{bmatrix}$$

are the union of the eigenvalues of  $\mathbf{A} - \mathbf{BK}$  and  $\mathbf{A} - \mathbf{LC}$ .

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are the union of the eigenvalues of  $\mathbf{A} - \mathbf{BK}$  and  $\mathbf{A} - \mathbf{LC}$ .

**Conclusion:** The eigenvalues of the controller and the observer may be assigned independently.

- The separation principle concerns the design of controllers and observers for LTI systems by pole assignment and basically states that these processes can be carried out independently.
- Separation theorem: for linear systems, a much stronger property can be shown, which concerns the design of optimal controllers and observers, which can be carried out, again, independently! More on this later...
- For nonlinear systems there are no general results.

Consider the LTI system

$$\begin{cases} \dot{\mathbf{x}}(t) = \mathbf{A}\mathbf{x}(t) + \mathbf{B}\mathbf{u}(t) \\ \mathbf{x}(0) = \mathbf{x}_0 \end{cases} .$$

- We know how to “*shape*” the closed-loop dynamics by pole placement.



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- Is it possible to do better, to optimize some kind of cost functional?

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- Is it possible to do better, to optimize some kind of cost functional?



# Optimal linear quadratic control

Infinite-horizon Linear Quadratic Regulation (LQR) problem formulation for LTI systems

## LQR problem

Given the LTI state equation

$$\dot{\mathbf{x}}(t) = \mathbf{A}\mathbf{x}(t) + \mathbf{B}\mathbf{u}(t),$$

consider the LQR optimization criterion

$$J := \int_0^{+\infty} \mathbf{x}^T(t)\mathbf{Q}\mathbf{x}(t) + \mathbf{u}^T(t)\mathbf{R}\mathbf{u}(t),$$

where  $\mathbf{Q}$  is positive semi-definite and  $\mathbf{R}$  is positive definite.

**Problem:** What is the control law that minimizes the LQR criterion?

### Theorem

Suppose that the pair  $(\mathbf{A}, \mathbf{B})$  is stabilizable and the pair  $(\mathbf{A}, \mathbf{G})$  is detectable, with  $\mathbf{Q} = \mathbf{G}^T \mathbf{G}$ . Then:

- there exists a unique, positive definite solution  $\mathbf{P}$  of the algebraic Riccati equation

$$\mathbf{A}^T \mathbf{P} + \mathbf{P} \mathbf{A} + \mathbf{Q} - \mathbf{P} \mathbf{B} \mathbf{R}^{-1} \mathbf{B}^T \mathbf{P} = \mathbf{0};$$

- The feedback law

$$\mathbf{u}(t) := -\mathbf{K} \mathbf{x}(t), \quad \mathbf{K} := \mathbf{R}^{-1} \mathbf{B}^T \mathbf{P}$$

minimizes the LQR criterion and  $J = \mathbf{x}^T(0) \mathbf{P} \mathbf{x}(0)$ ; and

- The eigenvalues of  $\mathbf{A} - \mathbf{B} \mathbf{K}$  all have negative real part.

# Optimal linear quadratic control

## Relaxation of the controllability and observability conditions

It is sufficient that the pair  $(\mathbf{A}, \mathbf{B})$  is stabilizable and the pair  $(\mathbf{A}, \mathbf{G})$  is detectable.

- Stabilizability: A system is controllable if all the uncontrollable modes are stable.
- Detectability: A system is detectable if all the unobservable modes are stable.

# Optimal linear quadratic control

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**Do these conditions make sense?**

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It is sufficient that the pair  $(\mathbf{A}, \mathbf{B})$  is stabilizable and the pair  $(\mathbf{A}, \mathbf{G})$  is detectable.

- Stabilizability: A system is controllable if all the uncontrollable modes are stable.
- Detectability: A system is detectable if all the unobservable modes are stable.

### Do these conditions make sense?

- If the system is not stabilizable there are modes that we cannot stabilize!
- If the pair  $(\mathbf{A}, \mathbf{G})$  is not detectable, it could be possible to make  $J$  very small, yet the state would be exploding!

### Trade-off between performance and control cost:

- If  $\mathbf{Q} \gg \mathbf{R}$ , the easiest way to minimize  $J$  is to use a lot of control resulting in a small controlled state.
- If  $\mathbf{Q} \ll \mathbf{R}$ , the easiest way to minimize  $J$  is to use little control, resulting in a large controlled state.



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### Bryson's rule:

$$\mathbf{Q}_{ii} = \frac{1}{\text{maximum acceptable value of } x_i^2(t)}$$
$$\mathbf{R}_{ii} = \frac{1}{\text{maximum acceptable value of } u_i^2(t)}$$

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$$\mathbf{R}_{ii} = \frac{1}{\text{maximum acceptable value of } u_i^2(t)}$$

Trial and error...

# Optimal linear quadratic control

Matlab example

## LQR design

```
>> A = [0 1; 0 0]; B = [0; 1]; Q = eye(2); R = 0.1; K  
= lqr(A, B, Q, R)  
K =  
    3.1623    4.0404
```

# Optimal linear quadratic control

## Matlab example

### LQR design

```
>> A = [0 1; 0 0]; B = [0; 1]; Q = eye(2); R = 0.1; K  
= lqr(A, B, Q, R)
```

```
K =
```

```
3.1623 4.0404
```

```
>> A = [0 1; 0 0]; B = [0; 1]; Q = eye(2); R = 10; K  
= lqr(A, B, Q, R)
```

```
K =
```

```
0.3162 0.8558
```

- W. Rugh, *Linear System Theory*, 2<sup>nd</sup> edition, Prentice Hall, 1995.
- C.-T. Chen, *Linear System Theory and Design*, 4<sup>th</sup> edition, Oxford University Press, 2012.
- J. Hespanha, *Linear Systems Theory*, Princeton University Press, 2009.