Ideal Nozzle Simulation Inputs:

a: 0.05 meter ** 2 / kilogram

n: 0.65 m: -0.2

Oxidiser:

Initial Volume: 0.41 liter Initial Mass: 0.68 lbs

Injector Mass Flow Rate: 0.035 kilogram / second

Number of Injectors: 1 Ideal O/F Ratio: 4.83 External Temp: 70 degF Time Step: 0.01 second

Simulation Results:

Total Burn Time: 8.73 second

Impulse: 1160.78 newton * second Average Thrust: 132.96 newton

Motor: J133

Nozzle Results:

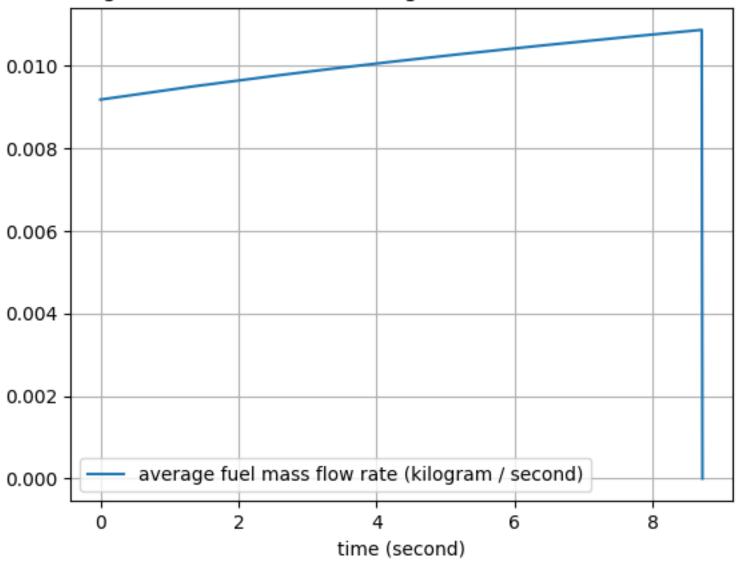
Suggested Throat Diameter: 0.193 inch Suggested Exit Diameter: 0.438 inch Suggested Diffuser Length: 0.457 inch

Fuel Grain

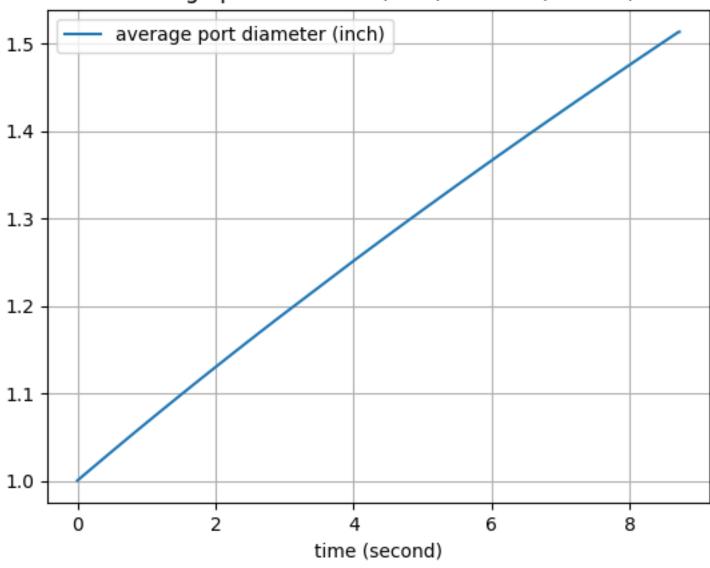
Port Length: 13.4 inch

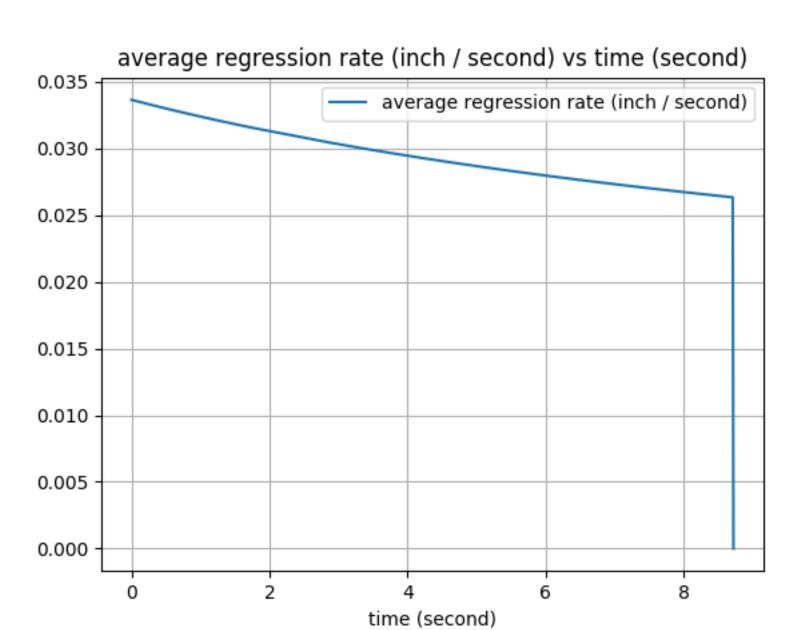
Fuel Density: 3.96 kilogram / meter ** 3

Grain Diameter: 1.75 inch Initial Port Diameter: 1.0 inch Final Port Diameter: 1.513 inch average fuel mass flow rate (kilogram / second) vs time (second)

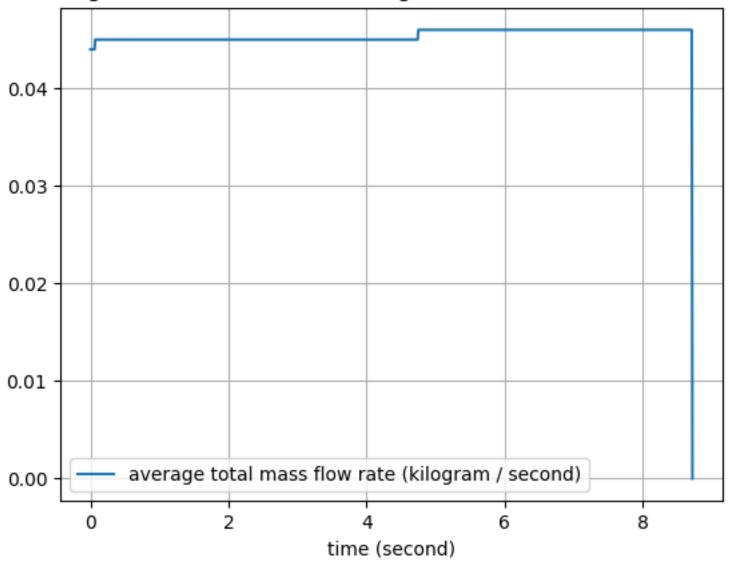


average port diameter (inch) vs time (second)

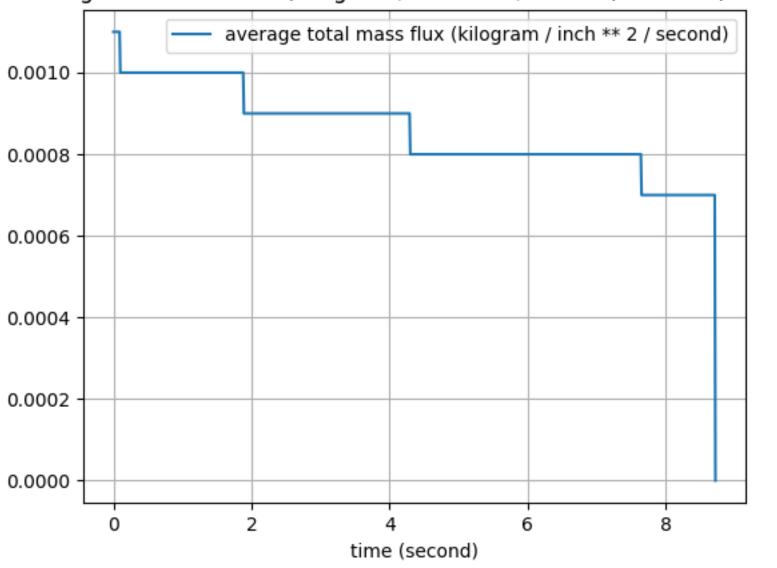




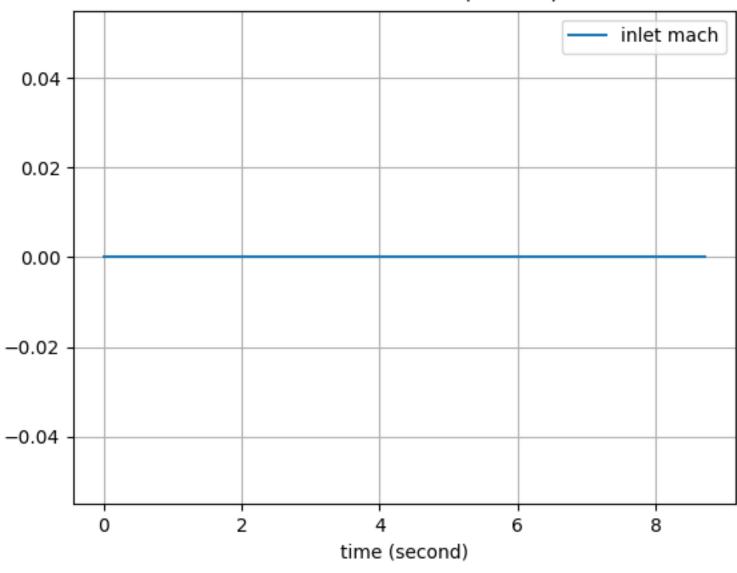
average total mass flow rate (kilogram / second) vs time (second)



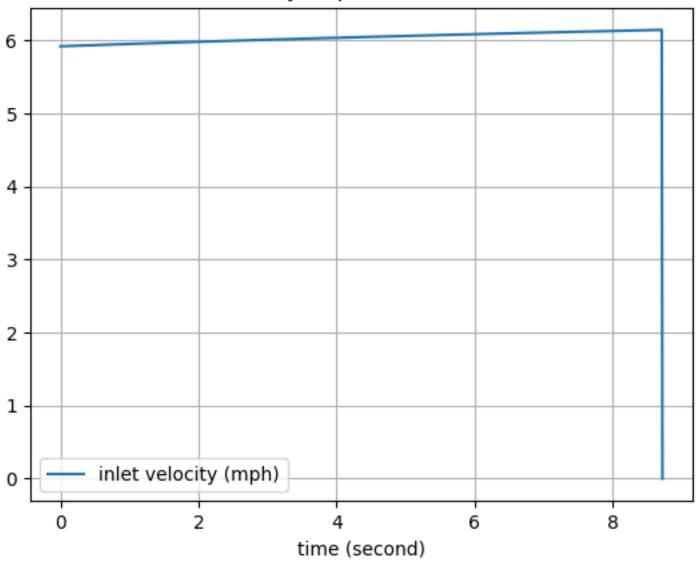
average total mass flux (kilogram / inch ** 2 / second) vs time (second)

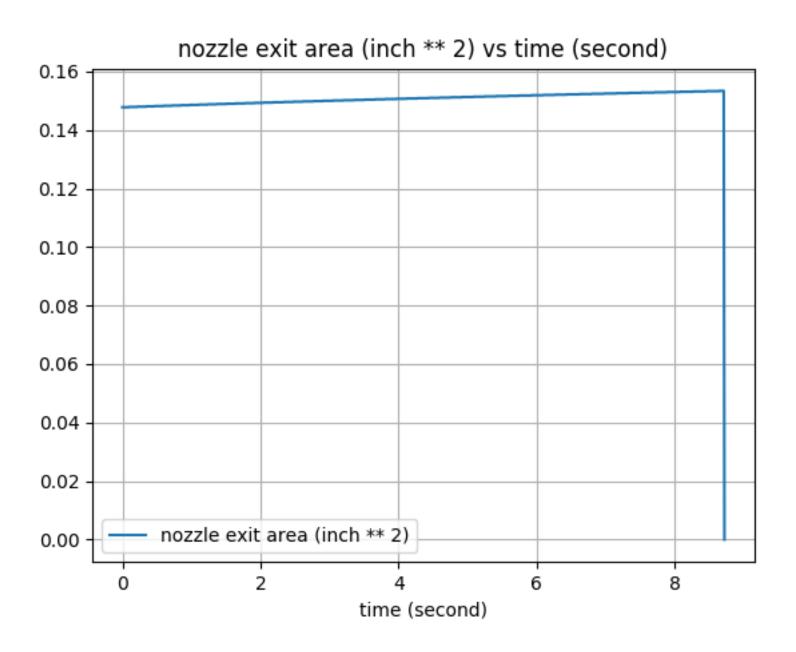


inlet mach vs time (second)

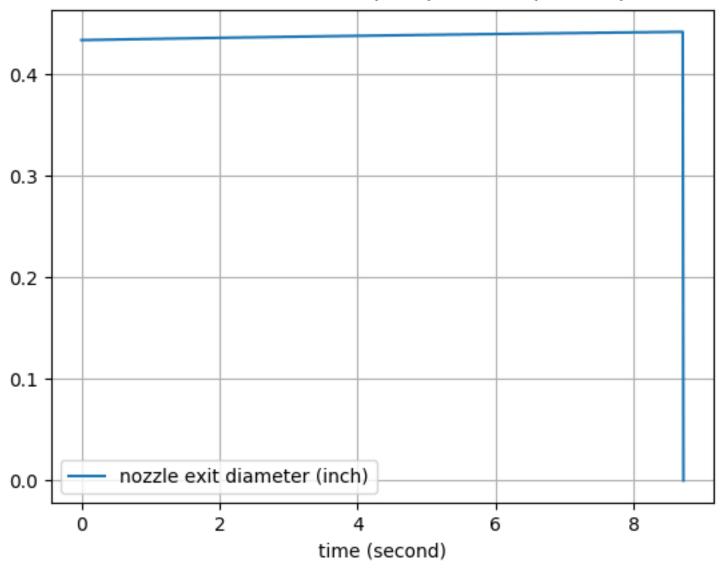


inlet velocity (mph) vs time (second)

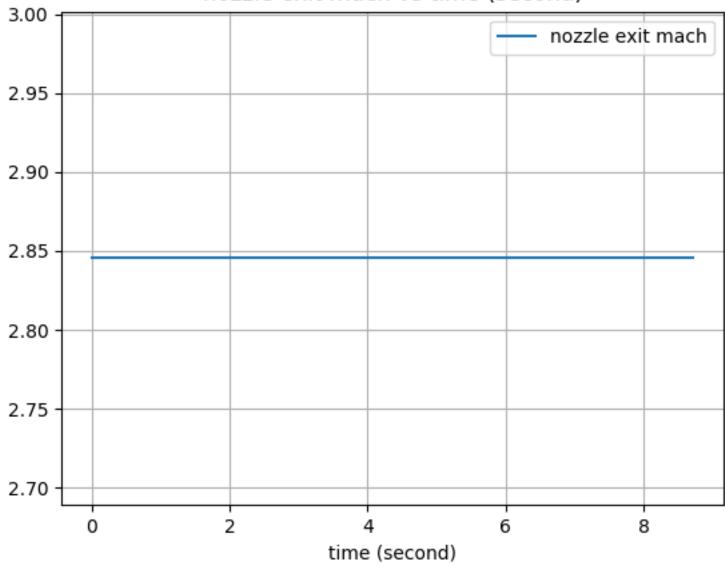




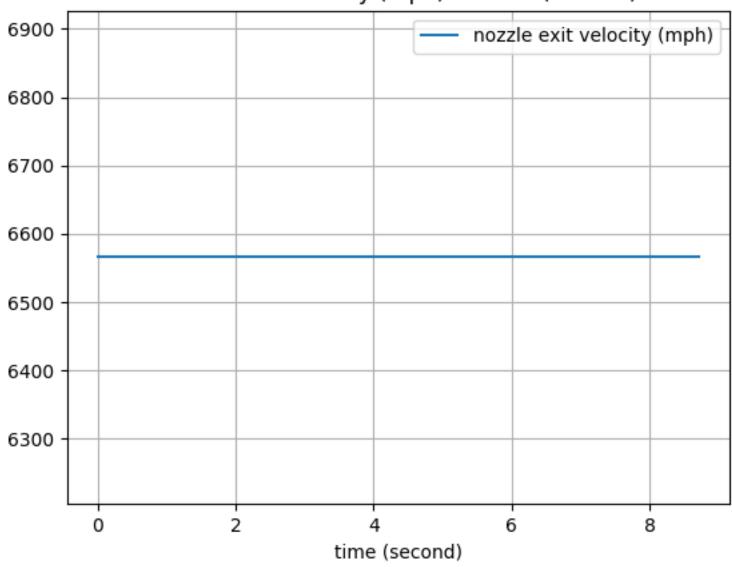
nozzle exit diameter (inch) vs time (second)

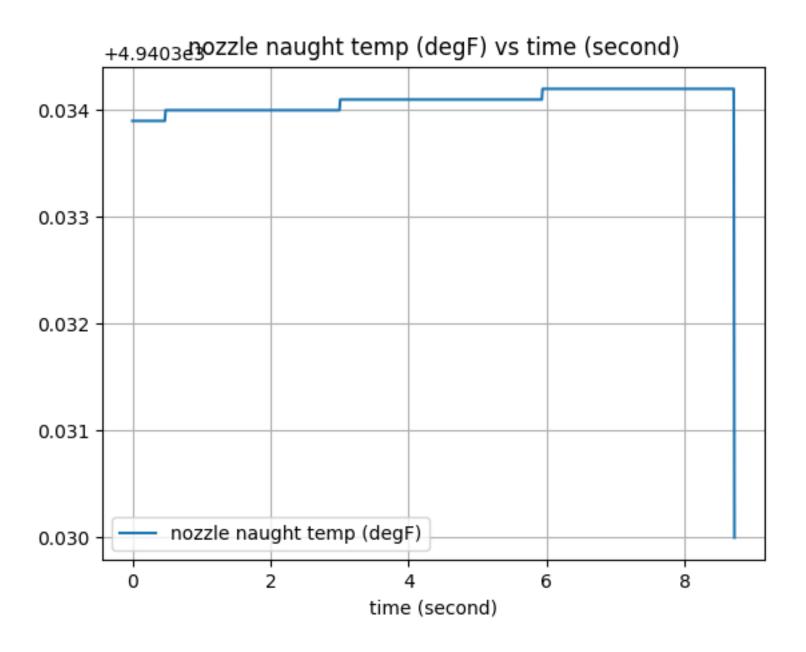


nozzle exit mach vs time (second)

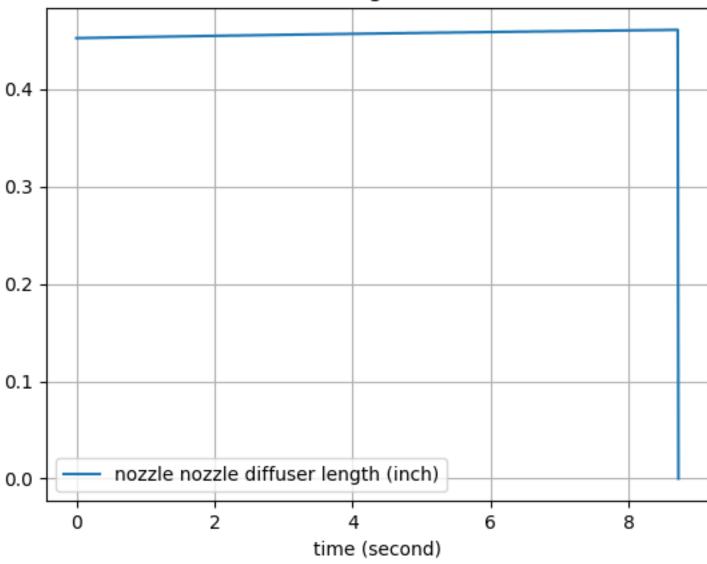


nozzle exit velocity (mph) vs time (second)

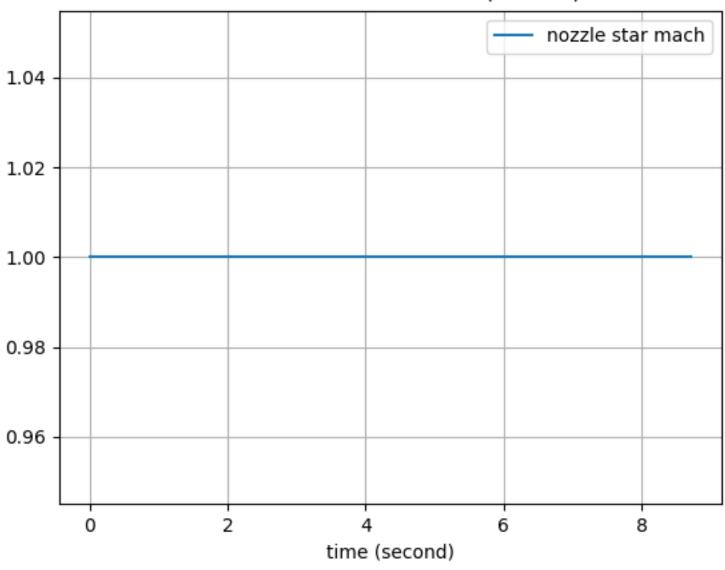




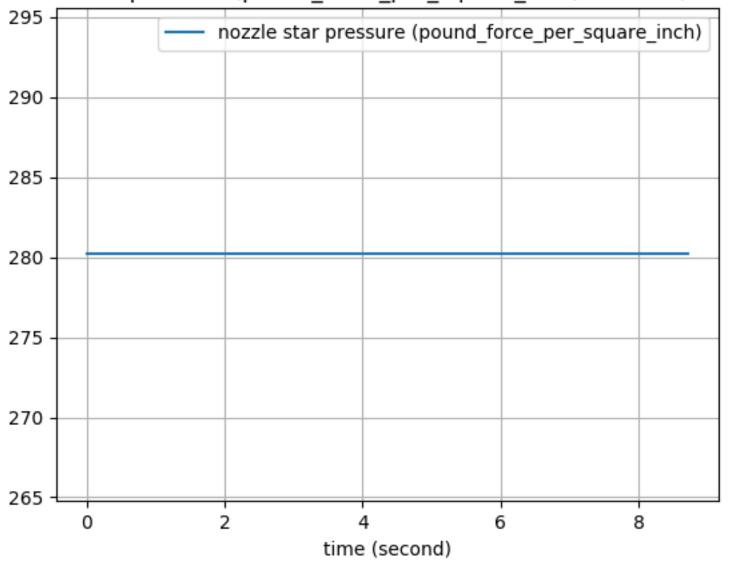
nozzle nozzle diffuser length (inch) vs time (second)

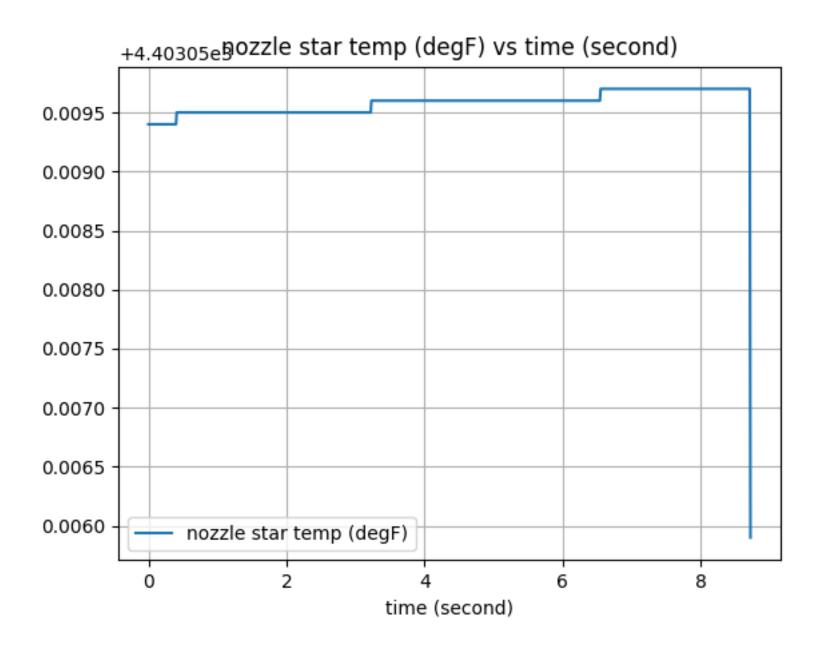


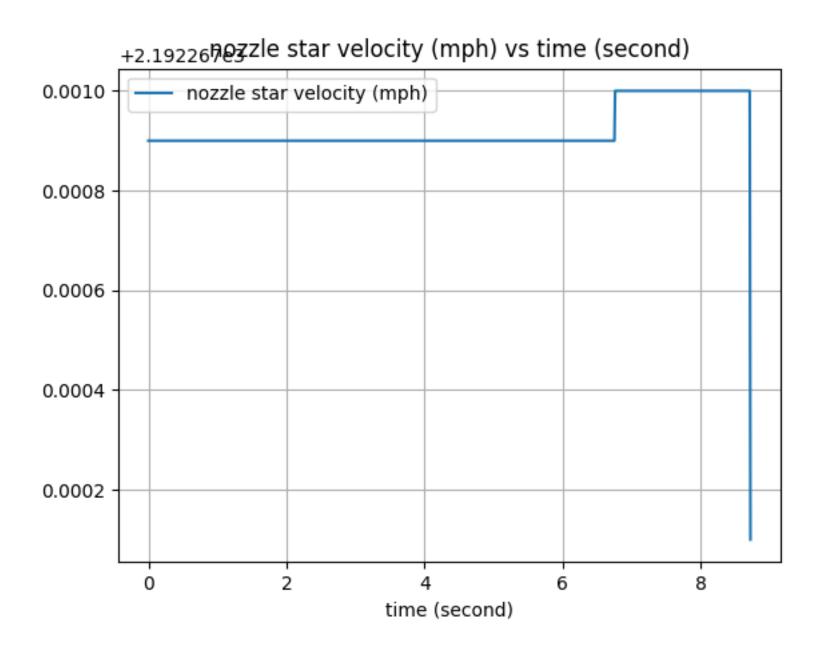
nozzle star mach vs time (second)



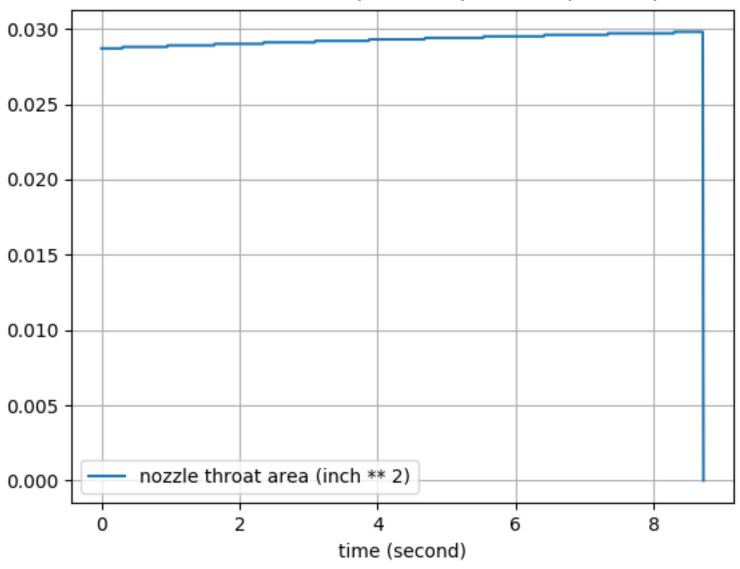
nozzle star pressure (pound_force_per_square_inch) vs time (second)



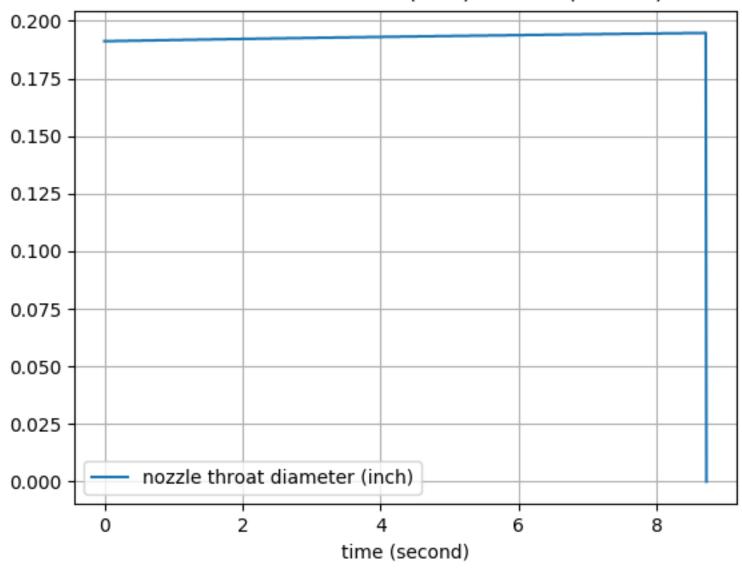




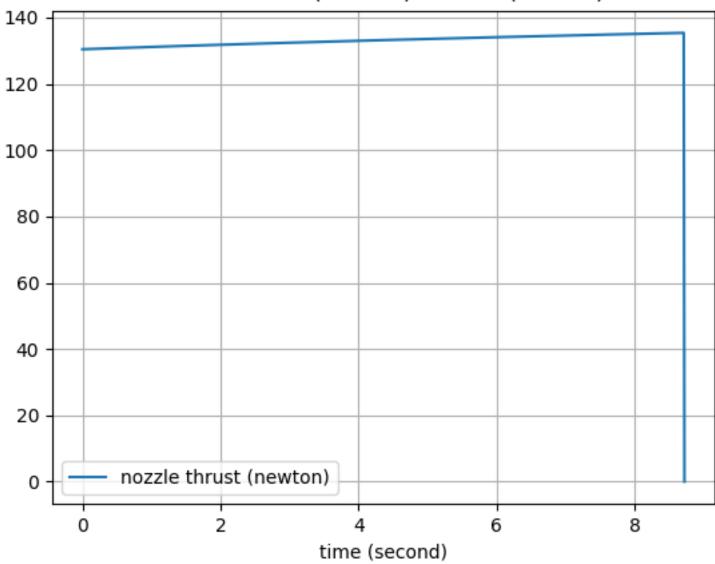
nozzle throat area (inch ** 2) vs time (second)



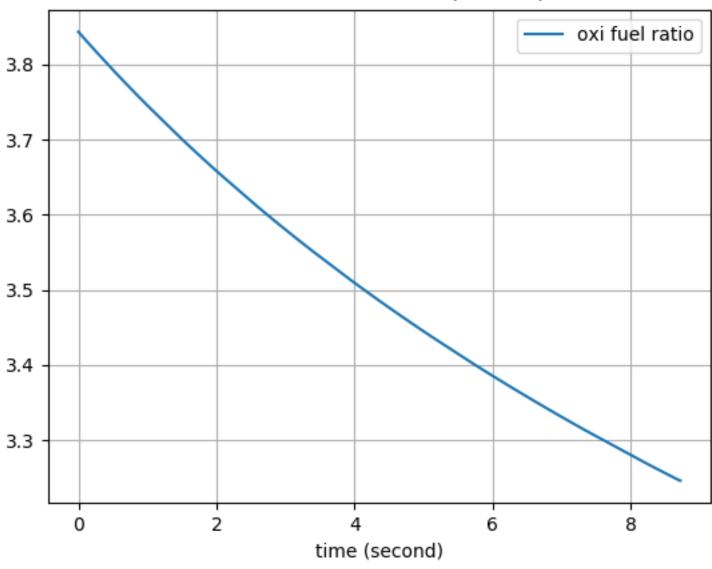
nozzle throat diameter (inch) vs time (second)



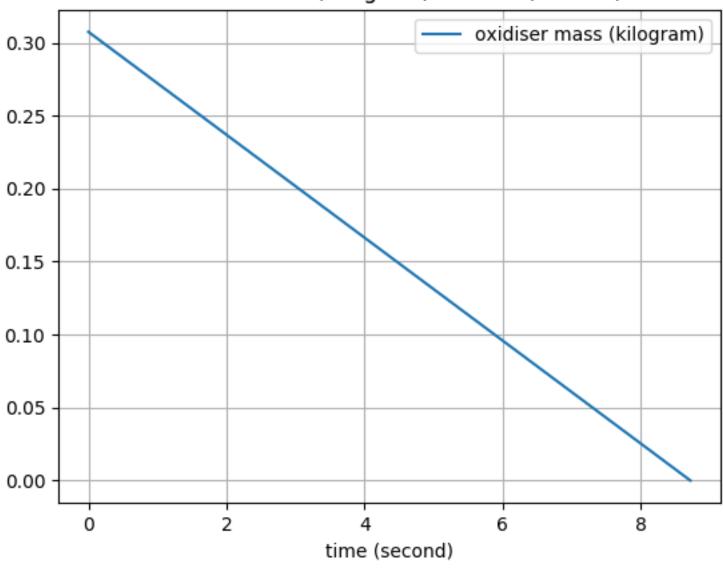
nozzle thrust (newton) vs time (second)



oxi fuel ratio vs time (second)



oxidiser mass (kilogram) vs time (second)



oxidiser mass flow rate (kilogram / second) vs time (second)

