#### Ideal Nozzle Simulation Inputs:

a: 0.05 meter \*\* 2 / kilogram

n: 0.65 m: -0.2

#### Oxidiser:

Initial Volume: 0.41 liter Initial Mass: 0.68 lbs

Injector Mass Flow Rate: 0.026 kilogram / second

Number of Injectors: 1 Ideal O/F Ratio: 4.83 External Temp: 70 degF Time Step: 0.01 second

#### Simulation Results:

Total Burn Time: 11.76 second

Impulse: 1197.85 newton \* second Average Thrust: 101.86 newton

Motor: J102

#### Nozzle Results:

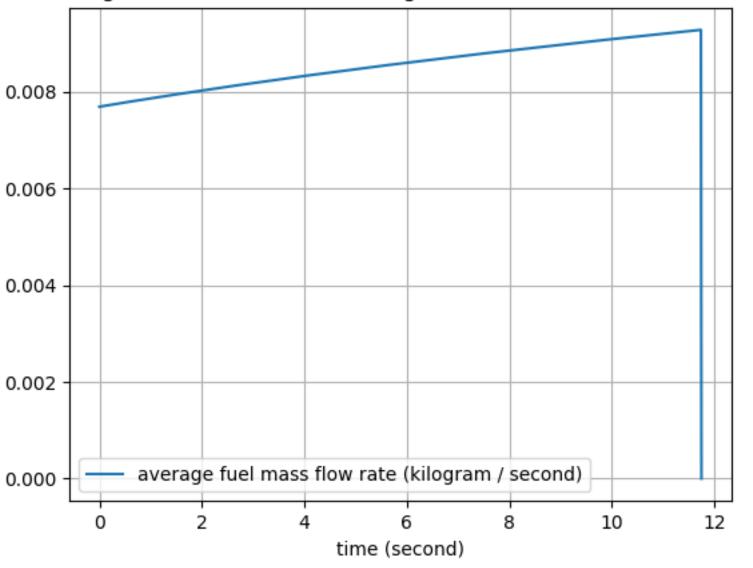
Suggested Throat Diameter: 0.169 inch Suggested Exit Diameter: 0.383 inch Suggested Diffuser Length: 0.4 inch

#### Fuel Grain

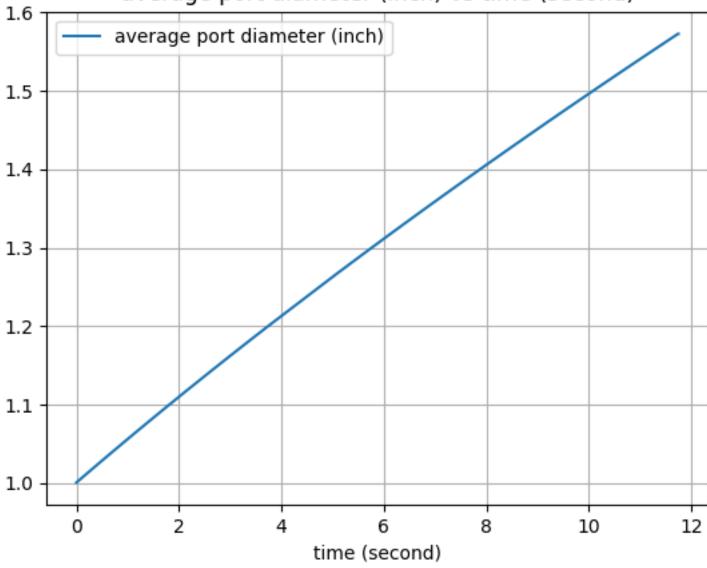
Port Length: 13.4 inch

Fuel Density: 3.96 kilogram / meter \*\* 3

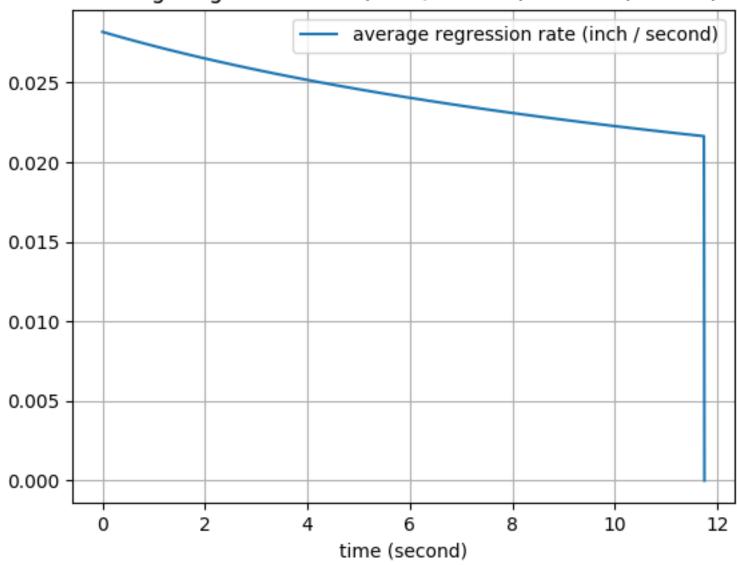
Grain Diameter: 1.75 inch Initial Port Diameter: 1.0 inch Final Port Diameter: 1.573 inch average fuel mass flow rate (kilogram / second) vs time (second)



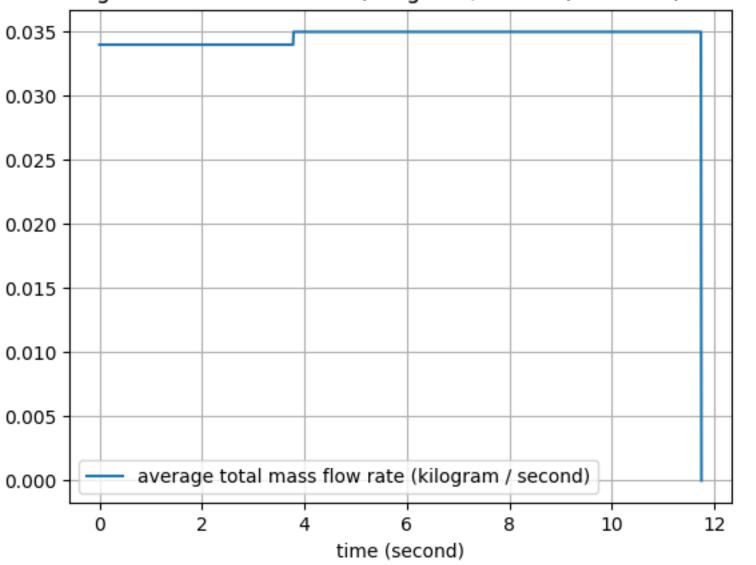
# average port diameter (inch) vs time (second)



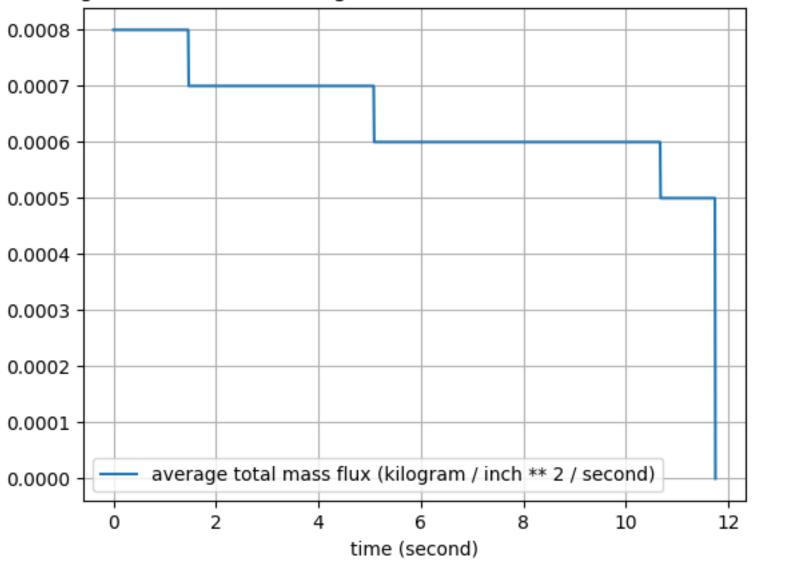
### average regression rate (inch / second) vs time (second)



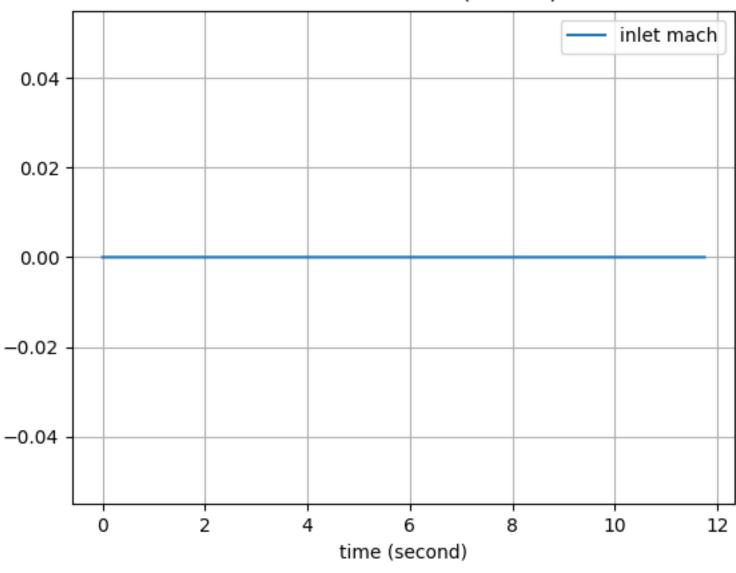
average total mass flow rate (kilogram / second) vs time (second)



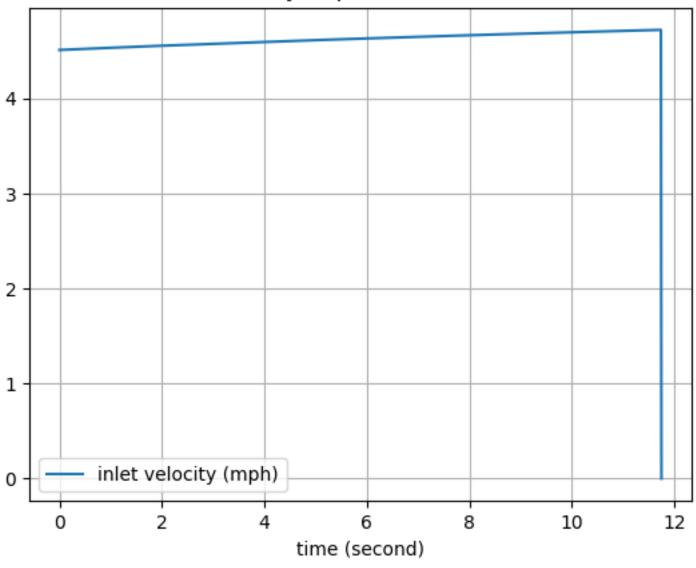
average total mass flux (kilogram / inch \*\* 2 / second) vs time (second)



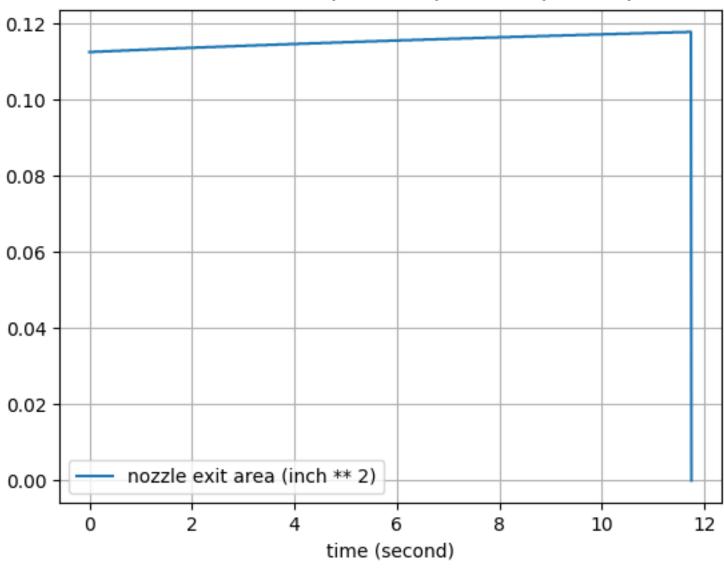
# inlet mach vs time (second)



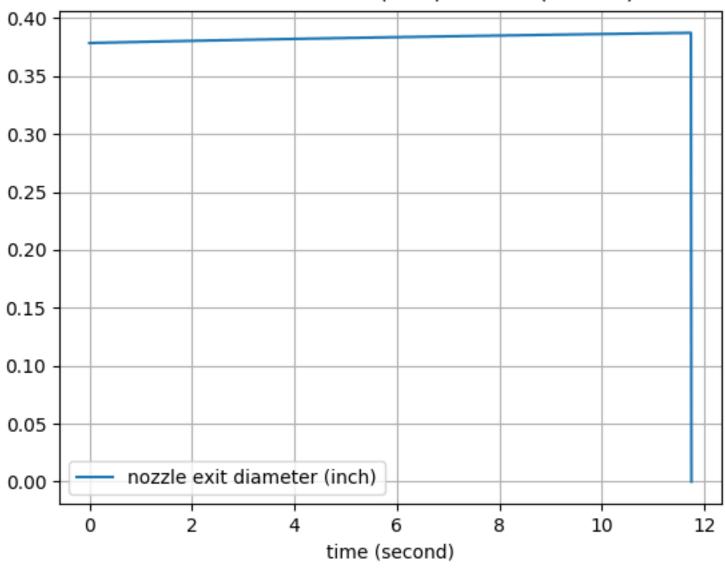
# inlet velocity (mph) vs time (second)



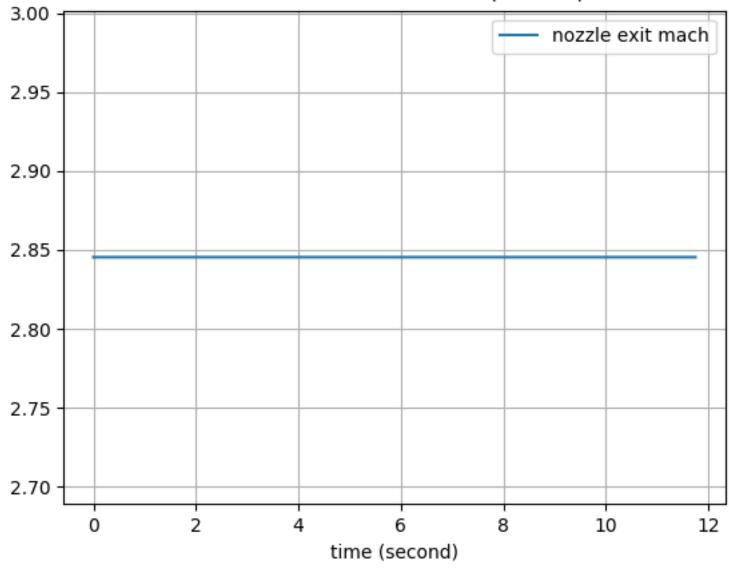
### nozzle exit area (inch \*\* 2) vs time (second)



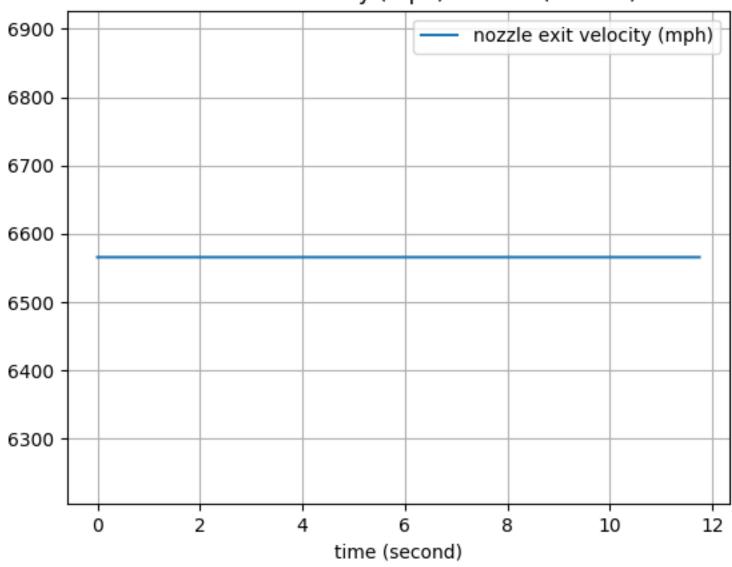
# nozzle exit diameter (inch) vs time (second)

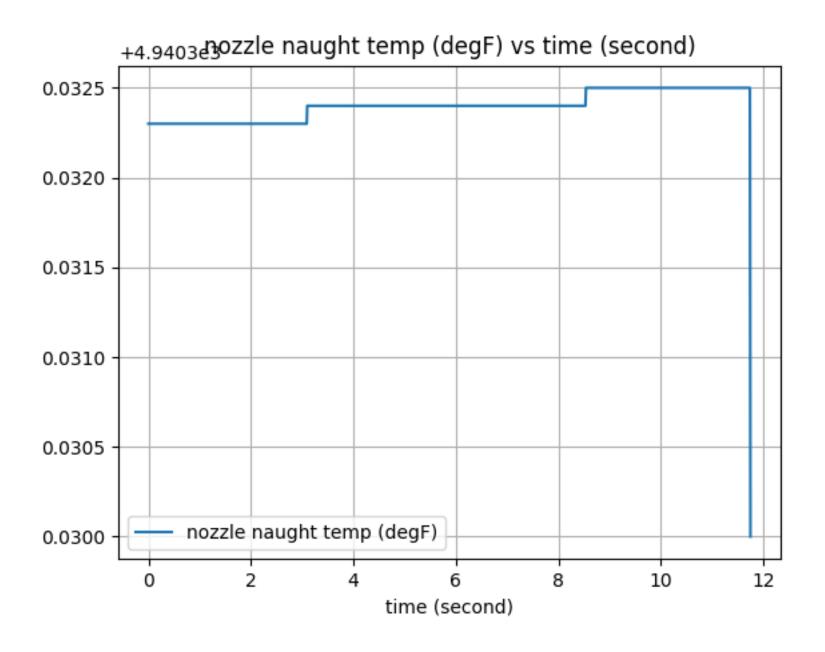


# nozzle exit mach vs time (second)

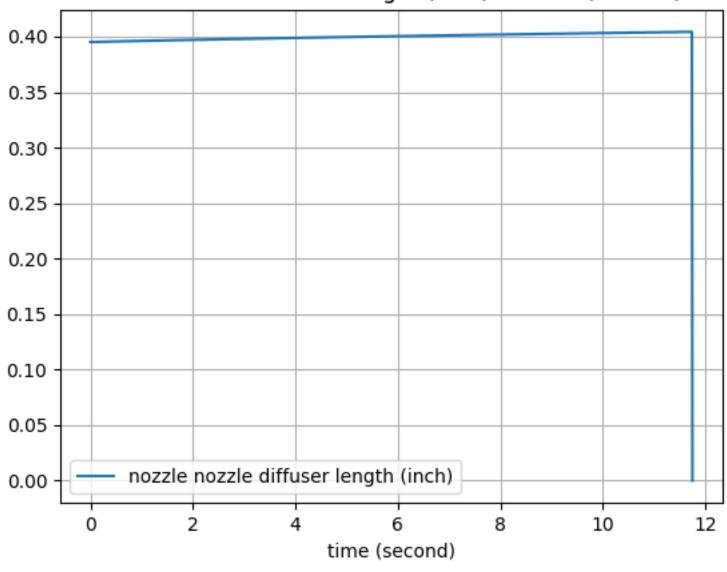


# nozzle exit velocity (mph) vs time (second)

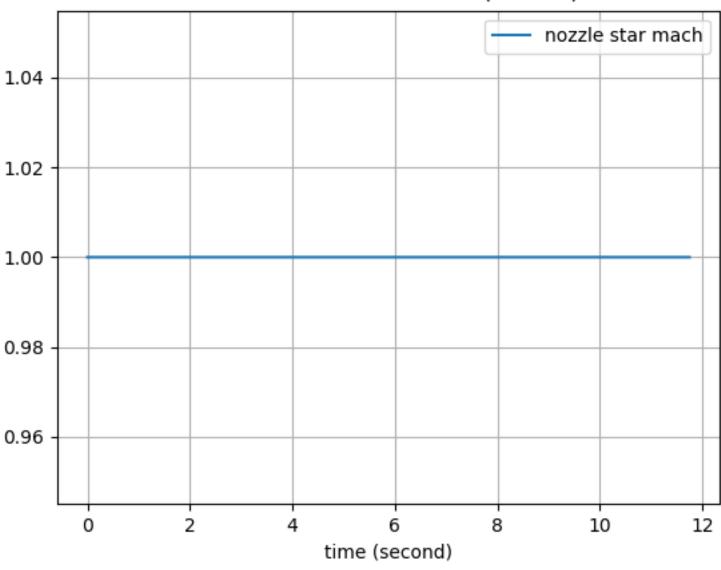




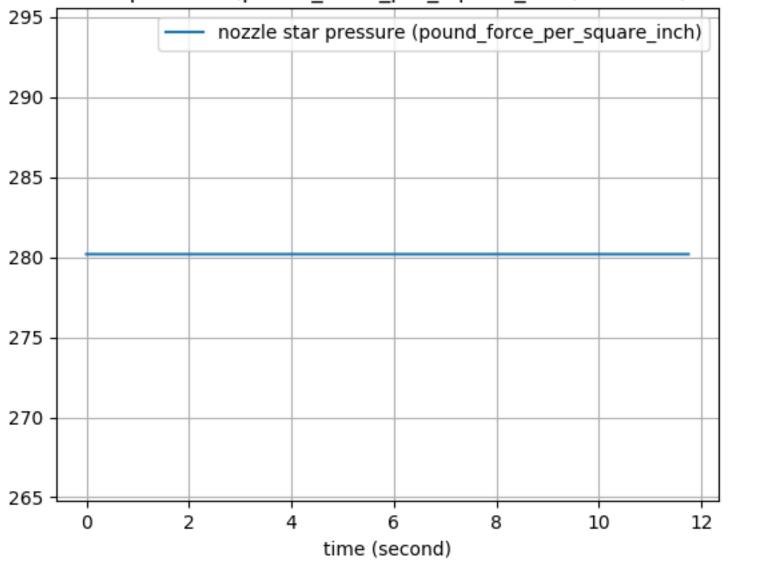
# nozzle nozzle diffuser length (inch) vs time (second)



# nozzle star mach vs time (second)

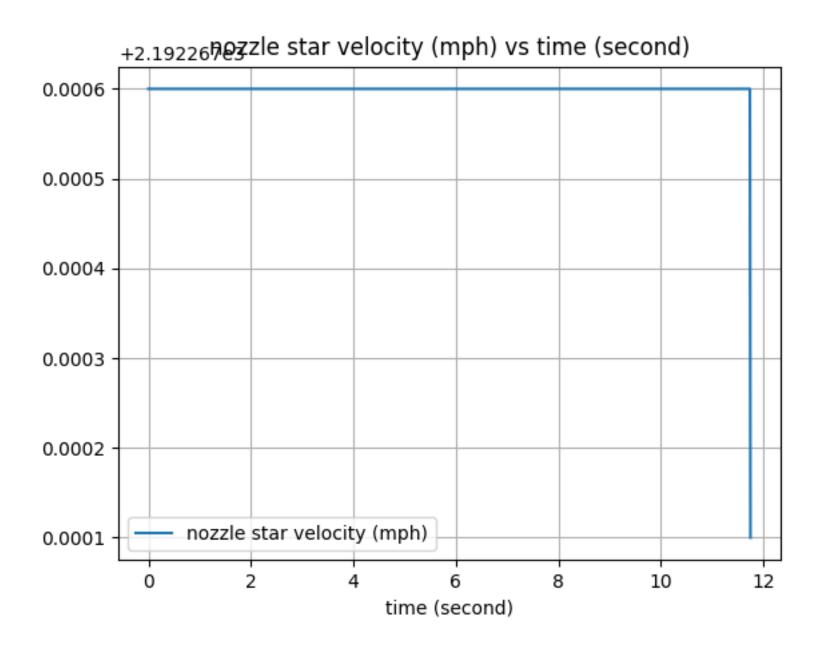


nozzle star pressure (pound\_force\_per\_square\_inch) vs time (second)

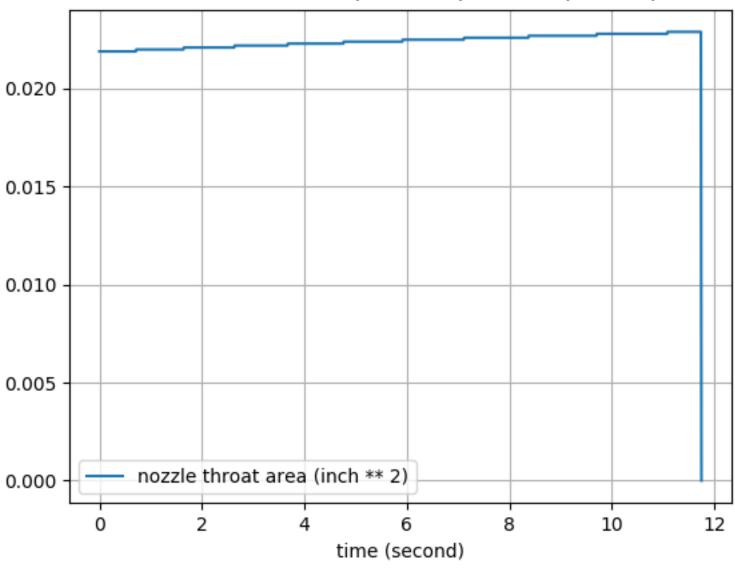


 $_{+4.40305e}$ gozzle star temp (degF) vs time (second) nozzle star temp (degF) 0.0080 -0.0075 0.0070 -0.0065 0.0060 -10 0 12

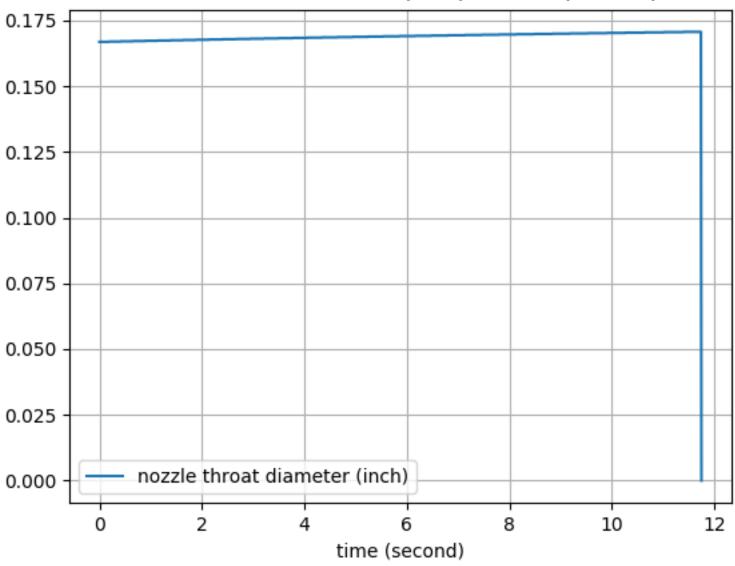
time (second)



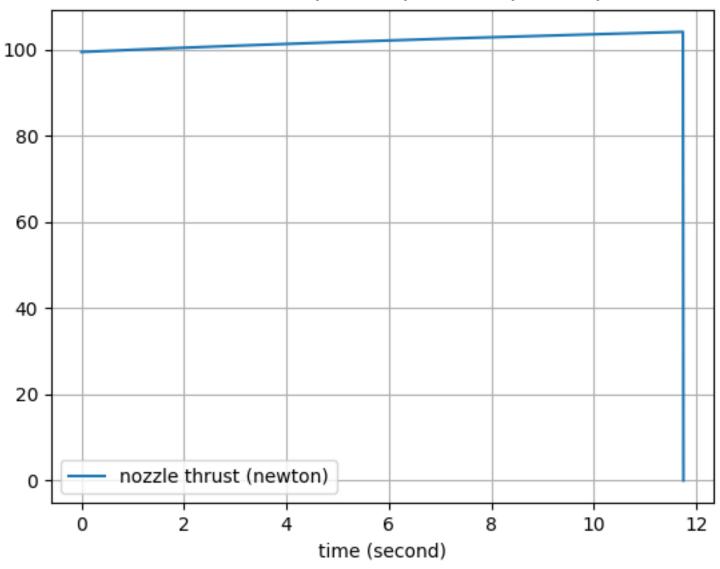
### nozzle throat area (inch \*\* 2) vs time (second)



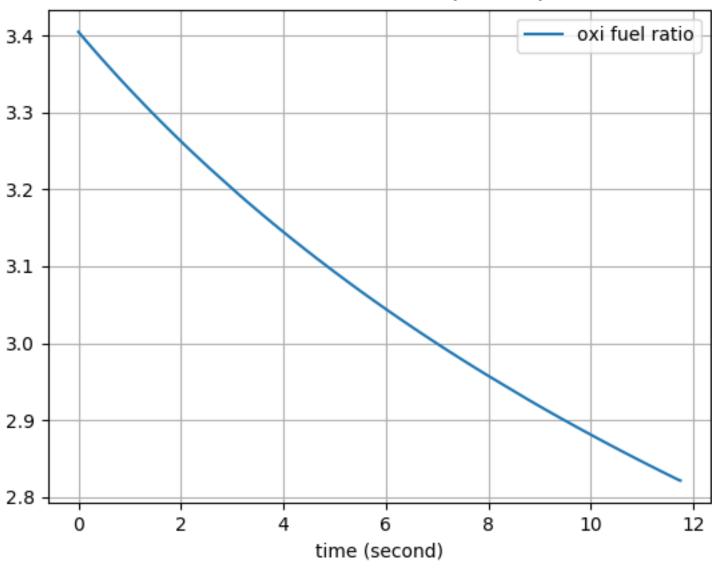
### nozzle throat diameter (inch) vs time (second)



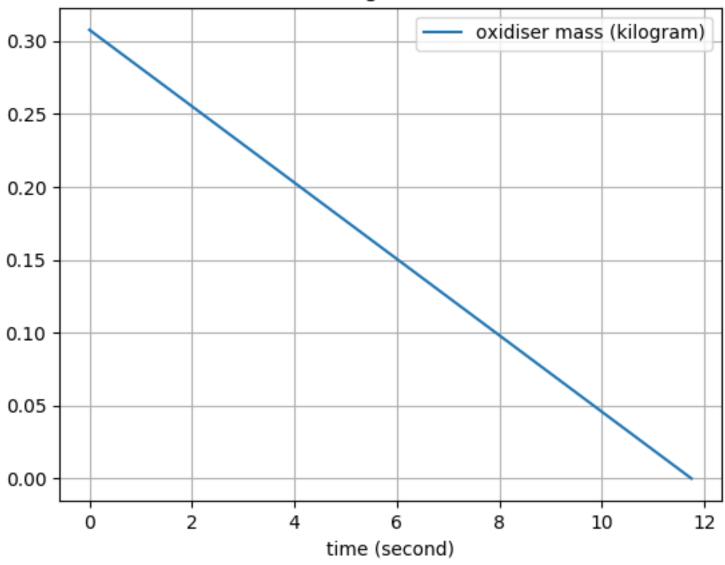
# nozzle thrust (newton) vs time (second)



# oxi fuel ratio vs time (second)



# oxidiser mass (kilogram) vs time (second)



oxidiser mass flow rate (kilogram / second) vs time (second)

