

a)

$$\omega_T = \frac{2\pi}{T} \text{ con } T = 86164 \text{ s (período sideral).}$$

$$\omega_T = \frac{2\pi}{86164 \text{ s}} = 7,2921 \cdot 10^{-5} \frac{\text{rad}}{\text{s}}$$

* Usamos la tercera ley de Kepler para órbita circular:

$$T^2 = \frac{4\pi^2 r^3}{\mu} \Rightarrow r = \left(\frac{\mu T^2}{4\pi^2} \right)^{1/3}$$

$$r = \left(\frac{398600,4418 \frac{\text{km}^3}{\text{s}^2}}{4\pi^2} \cdot (86164)^2 \text{ s}^2 \right)^{1/3}$$

$$r = 42164,14 \text{ km}$$

$$h = r - R_T = 42164,14 \text{ km} - 6371 \text{ km} = 35793,14 \text{ km}$$

$$v_0 = v_p = \sqrt{\mu} = 3,04 \frac{\text{km}}{\text{s}}$$