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DESIGN AND NUMERICAL STUDY OF HYDROFOILS

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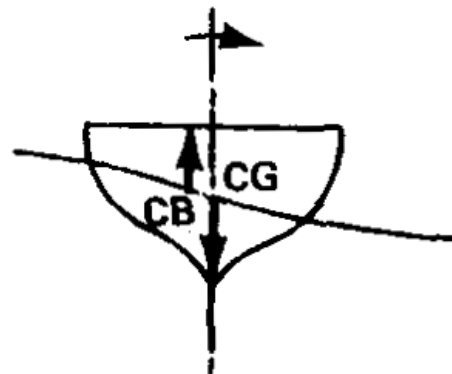
- Wing- like structures that lift a ship's hull out of water.
- Supported by hydrodynamic lift rather than by hydrostatic buoyancy.
- Reduces drag by reducing wetted area.
- Eliminates the instability effects of waves on the hull.
- First patented in the late 1880s; first test run in 1894- Meacham Brothers.



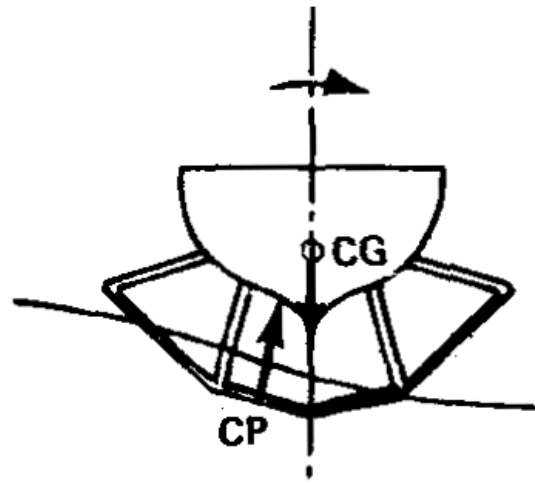
Hydrofoil Types^[1]

- Two general classifications:

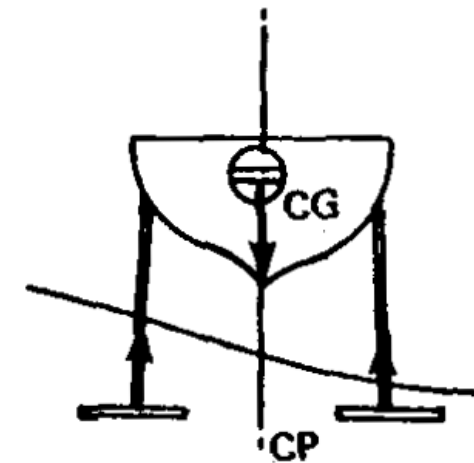
1. Surface Piercing
2. Submerged



DISPLACEMENT



SURFACE PIERCING



SUBMERGED FOIL

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Hydrofoil Types

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Surface Piercing

- Portions of the foils are designed to extend through the air/ sea interface.
- Encountering a wave- ship pitches or heaves and brings its weight and lift to balance.
- Self stabilization where water acts as the damper.
- For high speed operations an active control system is used.
- Rougher ride.



Hydrofoil Types

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Submerged

- Operate underwater at all times.
- Not self stabilizing- hence control is necessary.
- Uncoupled from effects of waves.
- Attains high speeds in any sea conditions, and still be comfortable for the crew.



Hydrofoil Configurations

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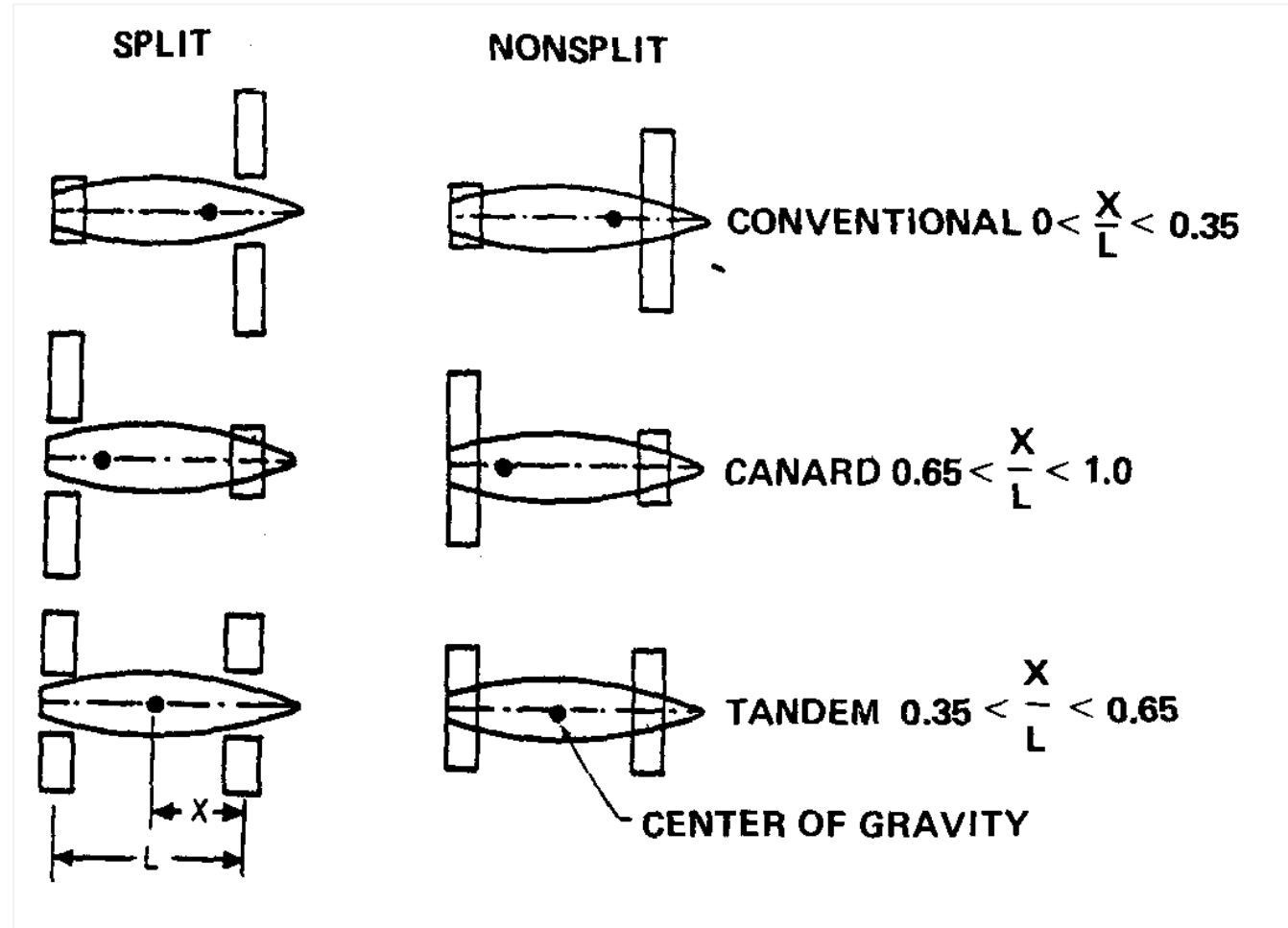
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- Canard- 65% or more of weight is supported on the fore or aft foil.
- Tandem- relatively equal weight distribution on both foils.





Hydrofoil Configurations

- Canard- 65% or more of weight is supported on the front or aft foil.
- Tandem- relatively equal weight distribution on both foils.

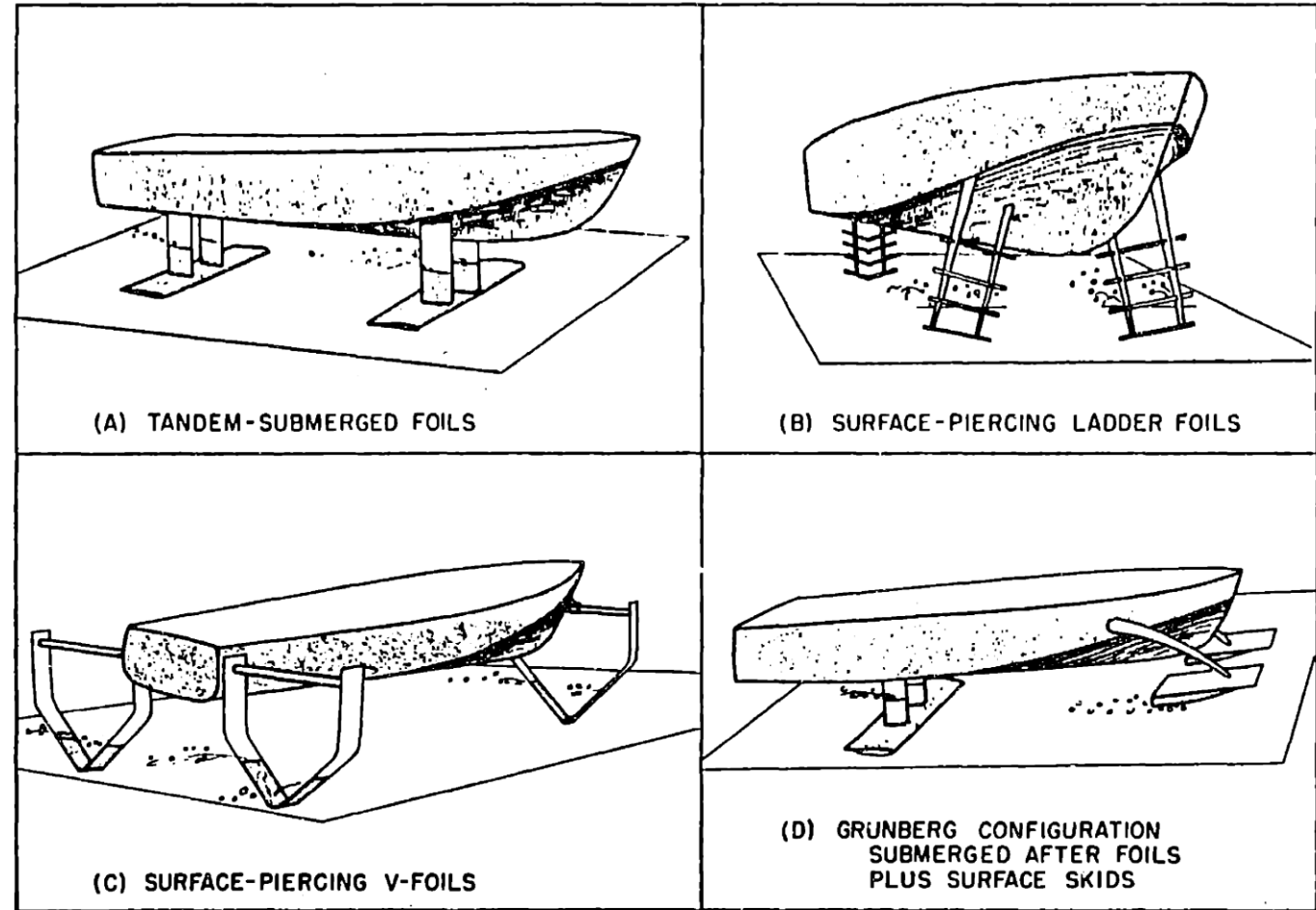


FIG. 1.—TYPICAL HYDROFOIL CONFIGURATIONS

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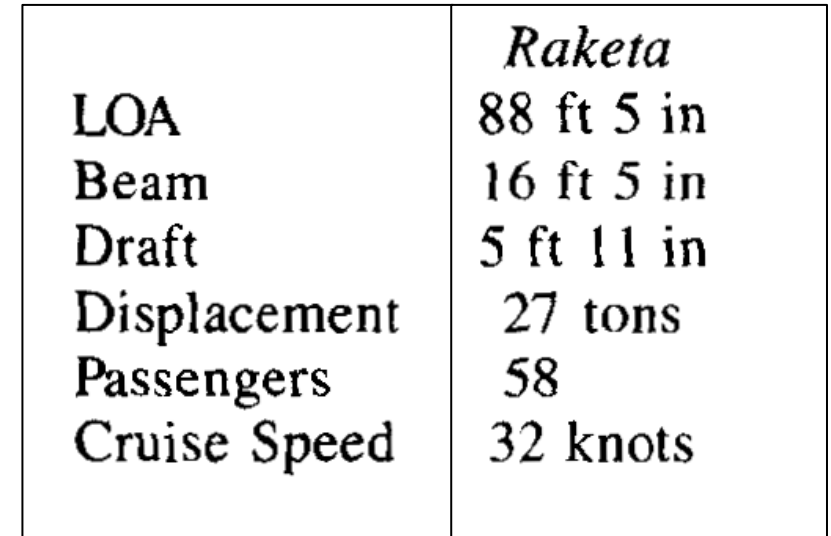
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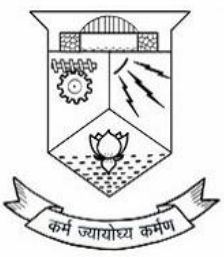
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Rakheta, German passenger hydrofoil boat; 1961



Design Parameters^[1]

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1. Geometry

- Chord
- Camber
- Thickness
- Span



- Aspect Ratio, $AR = \frac{Span^2}{Area}$

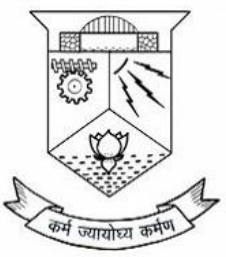
2. Angle of Attack (AoA)

3. Flow Speed -

- Reynolds Number, $Re = \frac{\rho \cdot v \cdot D}{\mu}$

4. C_D , C_L

- Froude Number, $Fr = \frac{U}{\sqrt{gh}}$



Forces on a hydrofoil

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- **Lift:-** calculating the lift coefficient ^[1]

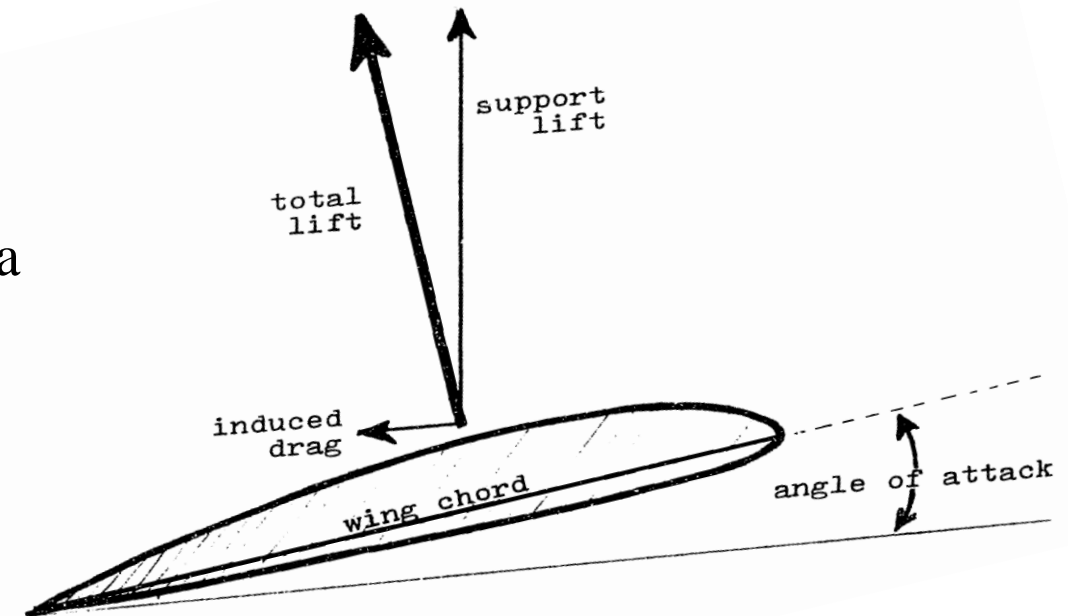
$$\rho \cdot v(y) \cdot \Gamma(y) = C(y) \cdot \frac{\rho}{2} \cdot U^2 \cdot c$$

- Polhamus' Lifting Line Theory:

A lifting surface can be thought of as a bound vortex.

- $$C_L = \frac{L}{\frac{1}{2} \rho U^2 c} = \frac{\rho U \Gamma}{\frac{1}{2} \rho U^2 c} = \frac{2 \Gamma}{U c}$$

- U = Velocity of foil; c = Chord; Γ = Circulation of bound vortex.





Geometry- Cross Section

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According to Parkin *et al* [2], aero foil theory can be used to design hydrofoils by similitude.

- The dimensionless numbers characteristic of the flow has to be the same.
- Effect of aspect ratio (AR), angle of attack (AoA), Reynolds Number (Re) can be readily understood.
- Hence existing, reliable codes solving the governing equations for a given set of boundary conditions can be used here as in aerodynamics.



Solutions

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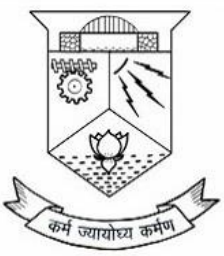
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References

1. Inverse Conformal Mapping method – by Lighthill, M.J. [3]
 2. XFOIL – Panel Method- by M.Drela et al [4]
 3. Genetic and Evolutionary Algorithms [5]
- Design is limited by the effects of the free surface on the hydrofoil.
 - These can be accounted for by [2]
 - Double body translation due to Dawson et al.
 - First order linearised approximations by Wu et al.
 - Laitone's theory based on Ausman's experimental results.



Numerical Optimization

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- The wings designed by numerical solutions have to be optimized using CFD [6].
- Advantages of Knowledge of Flow Pattern- Parkin et al [2]:
 - Flow visualization reveals the effect of variables not accounted for and the effect of approximations used.
 - Data for more real- life flow conditions can be obtained by simulation.
- Solver used
 - Fluent[7]
 - OpenFOAM[8]



Basic Control Parameters^[1]

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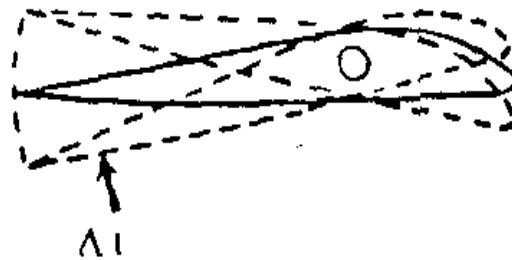
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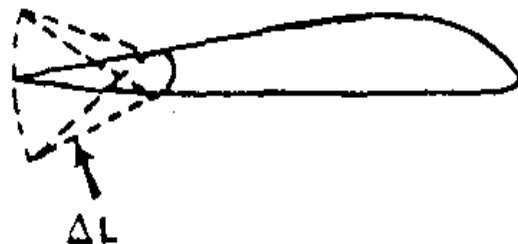
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- Changing the Angle of Attack (variable incidence).
- Change of Aspect Ratio (Trailing edge flaps)



VARIABLE INCIDENCE CONTROL



TRAILING - EDGE FLAP CONTROL



Project Objectives

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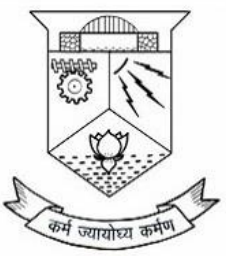
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Objectives

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- Find the shortest span of split hydrofoils for steady operation of a light boat.
- Study the surface effects on shallow foils.
- Investigate the possibility of using hydrophobic material coating on the foils.
- Acquire data about variation of geometry (AR), and incidence (AoA) for minimal drag and stable lift in different flow conditions.
- Investigate the application of swept foil configuration.



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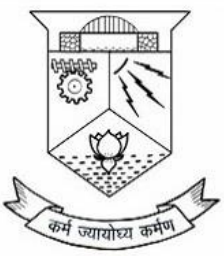
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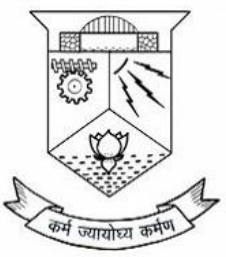
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Thanks to

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References

1. Dr Ranjith S Kumar, Project Supervisor
2. Dr Abhilash Suriyan, CET
3. Dr Mark Drela (Originator of XFOIL code, AeroAstro, MIT)
4. You