## **Girder Properties & Proportion Checks**

$E \equiv 290$	000 <b>k</b> .	si	
$F_{y} \equiv 50$	) ksi		
$t_w \equiv 0.8$			
$t_{topflan}^{-}$	$_{ge} \equiv 1.$	50 <b>i</b> i	n
$t_{bottom, eta}$	flange	$\equiv 1.75$	5 $in$
$b_f \equiv 16$	in		
$\vec{h} \equiv 55$			

$$I_x \equiv 64985.015 \ in^4$$

 $c_{xc} \equiv 13.772 \ in$ 

$$\begin{array}{c|c} b_f \\ \hline 2 \ t_{topflange} = 5.333 \\ 5.333 \leq 12 \quad \text{Ok!} \end{array}$$

$$\frac{o_f}{2 t_{cor}} = 5.333$$

$$=5.333$$
topflange  $5.333 < 12$ 

AASHTO Section 6.10 Checks

$$c_{xt} \equiv 17.522 \ in$$

$$=4.571$$

2 • 
$$^{t}$$
 bottomflange  $4.571 < 12$  Ok!

$$3.76 \cdot \sqrt{\left(\frac{E}{F_y}\right)} = 90.553$$

$$S_{xt} \equiv \frac{I_x}{c_{xt}} = 3708.767 \ in^3$$

 $S_{xc} \equiv \frac{I_x}{c_{res}} = 4718.633 \ in^3$ 

$$1.1 \cdot t_w = 0.55 \ in$$

$$1.50 \ge 0.55$$
 $1.75 \ge 0.55$ 

Ok!

$$3.76 \cdot \sqrt{\left(\frac{E}{F_y}\right)} = 90.553$$

$$\frac{S_{xt}}{G} = 0.786$$

$$\frac{h}{6}$$
 = 9.167 *in*  $16 \ge 9.167$  Ok!

$$S_{xt} = 0.786$$
 $S_{xc} = 137.274$ 
 $S_{xc} = 137.274$ 
 $I_{yc} = 21196.632 \ in^4$ 

$$I_{uc} \equiv 21196.632 \ in^4$$

$$\frac{I_{yc}}{I_{yt}} = 1.045$$

$$\frac{I_{yc}}{I_{yt}} = 1.045 \\ 0.1 \le 1.045 \le 10 \quad \text{Ok!}$$

$$I_{yt}\!\equiv\!20282.677~in^4$$

$$\frac{h}{t_w} = 110 \quad 110 \le 150 \quad \text{Ok!}$$

## **Moment Capacity Calculations**

 $M_U \equiv (1.25 \cdot (2525.39 \ ft \cdot kip + 488.28 \ ft \cdot kip)) + (1.50 \cdot 416.02 \ ft \cdot kip) + (1.75 \cdot 3328.29 \ ft \cdot kip) = 10215.625 \ ft \cdot kip$ 

-Plastic Moment Calculations (AASHTO Table D6.1-1)  $f_c \equiv 4 \text{ ksi}$   $t_s \equiv 8 \text{ in}$   $b_s \equiv 88 \text{ in}$ 

$$f' \equiv 4 \, ksi$$

$$t_{-}=8$$
 in

$$b_s \equiv 88 in$$

$$P_s \equiv 0.85 \; f'_c \cdot b_s \cdot t_s = 2393.6 \; kip$$

$$P_c \equiv F_y \cdot b_f \cdot t_{topflange} = 1200 \ kip$$

$$P_w \equiv F_y \cdot h \cdot t_w = 1375 \ kip$$

$$P_t \equiv F_y \cdot b_f \cdot t_{bottomflange} = 1400 \ kip$$

$$P_c + P_w + P_t = 3975 \ kip$$
  $P_s < (P_c + P_w + P_t)$ 

$$P_s < (P_c + P_w + P_t)$$

**AASHTO Case I:** 
$$(P_t + P_w) \ge (P_c + P_s)$$

$$(P_t + P_w) \ge (P_c + P_s)$$

$$P_t + P_w = 2775 \ kip$$

$$P_t + P_w = 2775 \ kip$$
  $P_c + P_s = 3593.6 \ kip$  2775  $kip < 3593.6 \ kip$ 

INVALID

**AASHTO Case II:**  $(P_t + P_w + P_c) \ge P_s$ 

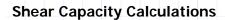
$$(P_t + P_w + P_c) \ge P$$

$$P_t + P_w + P_c = 3975 \ kip$$
  $P_s = 2393.6 \ kip$   $3975 \ kip > 2393.6 \ kip$ 

$$P_s = 2393.6 \ kip$$

$$3975 \; kip > 2393.6 \; kip$$

Therefore, the Plastic Neutral Axis (PNA) is located in the top flange of the beam.



$$V_{U} \coloneqq \left(1.25 \cdot 80.81 \ \textit{kip}\right) + \left(1.50 \cdot 13.31 \ \textit{kip}\right) + \left(1.75 \cdot 122.35 \ \textit{kip}\right) = 335.09 \ \textit{kip}$$

$$1.12 \cdot \sqrt{\frac{(29000 \cdot 5)}{50}} = 60.314 \qquad k = 5$$

$$\frac{85.3}{1.12} \cdot 1.4 = 106.625$$

$$C = \frac{1.57}{(110^2)} \cdot \left(\frac{29000 \cdot 5}{50}\right) = 0.376$$

$$V_p \equiv 0.58 (50 \text{ ksi} \cdot 55 \text{ in} \cdot 0.5 \text{ in}) = 797.5 \text{ kip}$$

$$V_p \cdot C = 300.084 \ kip$$

$$797.5 \cdot 0.376 = 299.86$$

$$V_n = 797.5 \ kip \left(C + \frac{0.87 \cdot (1 - C)}{1 + \sqrt{2}}\right) = 479.336 \ kip$$

$$\phi_V = 1.0$$

$$\phi_V \cdot V_n = 479.336 \ kip$$

$$\phi_V \! \cdot \! V_n \! \geq \! V_U$$

$$479.336 \; kip > 335.09 \; kip$$

VALID

Therefore, use one 1/2" thick shear stiffener located 55" away from abutments on each girder.