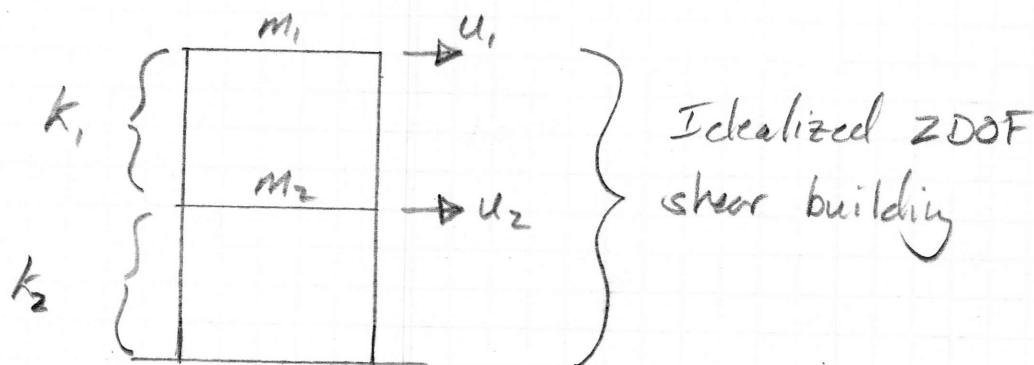


## LECTURE 4: Normal Mode Method & Free Vibration of MDOF Systems



$$k_1 = k_2 = 100 \text{ lb/in}$$

$$m_1 = m_2 = 0.5 \text{ lb}_m$$

$$[K] = \begin{bmatrix} 100 & -100 \\ -100 & 200 \end{bmatrix} \text{ lb/in} \quad [M] = \begin{bmatrix} 0.5 & 0 \\ 0 & 0.5 \end{bmatrix}$$

\$\Rightarrow\$ Recall from lecture 1

$$[M]\{\ddot{u}\} + [K]\{u\} = 0$$

\$\Rightarrow\$ Assume

$$u_i = \begin{Bmatrix} u_{1,i} \\ u_{2,i} \end{Bmatrix} \sin(\omega_i t)$$

↑ shape of vibration

@ frequency \$\omega\_i\$ (Mode Shape)

In general \$i = \# \text{DOF}\$, i.e. a 2DOF system will have two frequencies and two mode shapes.

\$\Rightarrow\$ Integrating twice..

$$\ddot{u}_i = -\omega_i^2 \begin{Bmatrix} u_{1,i} \\ u_{2,i} \end{Bmatrix} \sin(\omega_i t)$$

↑ same mode shape

$\Rightarrow$  Subst. into EOM.

$$-\omega_i^2 [M] \{\phi_i\} \sin(\omega_i t) + [K] \{\phi_i\} \sin(\omega_i t) = 0$$

$$\{[K] - \omega_i^2 [M]\} \{\phi_i\} = 0$$

$\hookrightarrow$  Eigen problem we solved previously

$\Rightarrow$  Let's look closer at this expression

$$\text{Mode } i \Rightarrow [K] \{\phi_i\} = \omega_i^2 [M] \{\phi_i\} \quad (1)$$

$$\text{Mode } j \Rightarrow [K] \{\phi_j\} = \omega_j^2 [M] \{\phi_j\} \quad (2)$$

$\Rightarrow$  Pre multiply Eq 1 by  $\{\phi_j\}^T$

$$\{\phi_j\}^T [K] \{\phi_i\} = \omega_i^2 \{\phi_i\}^T [M] \{\phi_i\} \quad (3)$$

$\Rightarrow$  Take the transpose of Eq 3

$$\{\phi_i\}^T [K] \{\phi_j\} = \omega_i^2 \{\phi_i\}^T [M] \{\phi_j\} \quad (4)$$

\*  $\Rightarrow$  Note:  $\underbrace{[K]}_{\text{Symm}} = \underbrace{[K]^T}_{\text{Symm}} \neq \underbrace{[M]}_{\text{Symm}} = \underbrace{[M]^T}_{\text{Symm}}$

$\Rightarrow$  Pre multiply Eq 2 by  $\{\phi_i\}^T$

$$\{\phi_i\}^T [K] \{\phi_j\} = \omega_j^2 \{\phi_i\}^T [M] \{\phi_j\} \quad (5)$$

$\Rightarrow$  Subtract Eq 5 from Eq 4.

$$(\omega_i^2 - \omega_j^2) \{\phi_i\}^T [M] \{\phi_j\} = 0 \quad (6)$$

for  $\omega_i^2 \neq \omega_j^2$  (i.e. no repeated roots)

$$\star \quad \{\phi_i\}^T [M] \{\phi_j\} = 0 \quad i \neq j \quad (7)$$

$\rightarrow$  The eigen vectors (mode shapes) are said to be orthogonal w.r.t. the mass matrix.

$\rightarrow$  Subst. Eq. 7 into Eq 4

$$\star \quad \{\phi_i\}^T [k] \{\phi_j\} = 0 \quad i \neq j \quad (8)$$

$\Rightarrow$  Let's check to see if this is true...

$\Rightarrow$  Back to example.

$$a = (0.5)(0.5) = 0.25$$

$$b = (100)(0.5) - (200)(0.5) = -150$$

$$c = (100)(200) - (100)(100) = 10,000$$

$$\omega_i^2 = \frac{150 \pm \sqrt{150^2 - 4(0.25)(10,000)}}{2(0.25)}$$

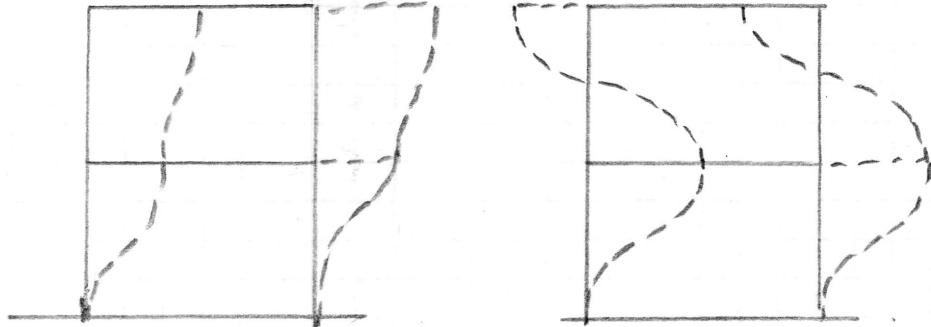
$$\omega_i^2 = 300 \pm 225.61$$

$$\omega = \begin{cases} 8.74 \\ 22.88 \end{cases} \text{ rad/s}$$

$$r_1 = \frac{100 + (76.39)(0)}{100 - (76.39)(0.5)} = 1.618$$

$$r_2 = \frac{100 + (523.5)(0)}{100 - (523.5)(0.5)} = -0.618$$

$$\phi_1 = \begin{Bmatrix} 1.0 \\ 0.618 \end{Bmatrix} \quad \phi_2 = \begin{Bmatrix} -0.618 \\ 1.0 \end{Bmatrix} \quad (\text{Unit normalized})$$



$$i=1, j=2$$

$$\phi^T [M]$$

$$\begin{Bmatrix} 1.0 & 0.618 \end{Bmatrix} \begin{bmatrix} 0.5 & 0 \\ 0 & 0.5 \end{bmatrix} = \begin{Bmatrix} 0.5 & 0.309 \end{Bmatrix}$$

$$\phi^T [M] \phi_2$$

$$\begin{Bmatrix} 0.5 & 0.309 \end{Bmatrix} \begin{Bmatrix} -0.618 \\ 1.0 \end{Bmatrix} = 0 \quad \checkmark$$

$$\phi^T [K]$$

$$\begin{Bmatrix} 1.0 & 0.618 \end{Bmatrix} \begin{bmatrix} 100 & -100 \\ -100 & 200 \end{bmatrix} = \begin{Bmatrix} 38.2 & 23.6 \end{Bmatrix}$$

$$\phi^T [K] \phi_2$$

$$\begin{Bmatrix} 38.2 & 23.6 \end{Bmatrix} \begin{Bmatrix} -0.618 \\ 1.0 \end{Bmatrix} = 0 \quad \checkmark$$

What does this mean??

- \*  $\Rightarrow$  Mode shapes can de-couple the stiffness & mass matrices
- \*  $\Rightarrow$  This essentially transforms a 2DOF system into two 1DOF systems.

$$\phi = \begin{bmatrix} \phi_1 & \phi_2 \end{bmatrix} = \begin{bmatrix} 1.0 & -0.618 \\ 0.618 & 1.0 \end{bmatrix}$$

$\phi^T [M]$

$$\begin{bmatrix} 1.0 & 0.618 \\ -0.618 & 1.0 \end{bmatrix} \begin{bmatrix} 0.5 & 0 \\ 0 & 0.5 \end{bmatrix} = \begin{bmatrix} 0.5 & 0.309 \\ -0.309 & 0.5 \end{bmatrix}$$

$\phi^T [M] \phi$

$$\begin{bmatrix} 0.5 & 0.309 \\ -0.309 & 0.5 \end{bmatrix} \begin{bmatrix} 1.0 & -0.618 \\ 0.618 & 1.0 \end{bmatrix} = \begin{bmatrix} 0.691 & 0 \\ 0 & 0.691 \end{bmatrix}$$

Note, these are rarely the same...

zero because of orthogonality.

$\phi^T [K]$

$$\begin{bmatrix} 1.0 & 0.618 \\ -0.618 & 1.0 \end{bmatrix} \begin{bmatrix} 100 & -100 \\ -100 & 200 \end{bmatrix} = \begin{bmatrix} 38.2 & 23.6 \\ -161.8 & 261.8 \end{bmatrix}$$

$\phi^T [K] \phi$

$$\begin{bmatrix} 38.2 & 23.6 \\ -161.8 & 261.8 \end{bmatrix} \begin{bmatrix} 1.0 & -0.618 \\ 0.618 & 1.0 \end{bmatrix} = \begin{bmatrix} 52.78 & 0 \\ 0 & 361.8 \end{bmatrix}$$

=> This transforms the problem into "Normal" coordinates (aka "Modal" coordinates)

=> Somewhat (kind of, sort of) analogous to transforming stresses into principal stresses

↳ this remains in physical coordinates though...

re-writing the EOM in Normal Coordinates.

$$\text{uncoupled} \quad \Rightarrow \text{2 SDOF system} \quad \left\{ \begin{bmatrix} M_1 & 0 \\ 0 & M_2 \end{bmatrix} \begin{bmatrix} \ddot{z}_1 \\ \ddot{z}_2 \end{bmatrix} + \begin{bmatrix} K_1 & 0 \\ 0 & K_2 \end{bmatrix} \begin{bmatrix} z_1 \\ z_2 \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \end{bmatrix} \right.$$

Where,

$M_1$  = mass associated w/ mode 1

$K_1$  = stiffness " " " "

$z_1$  = generalized coordinate associated w/ mode 1 (where  $\phi_1$  is the associated shape function).

Note:  $z_1, z_2$   
are referred to  
as "normal" or  
"modal" coordinates

$M_2, K_2, z_2$  are associated w/ mode 2.

↳  $\phi_2$  is the associated shape function.

Multiplying the EOM out...

$$M_1 \ddot{z}_1 + K_1 z_1 = 0 \quad \omega_1 = \sqrt{\frac{K_1}{m_1}} = \sqrt{\frac{52.18}{0.691}} = 8.69 \text{ rad/s}$$

$$M_2 \ddot{z}_2 + K_2 z_2 = 0 \quad \omega_2 = \sqrt{\frac{K_2}{M_2}} = \sqrt{\frac{361.8}{0.691}} = 22.88 \text{ rad/s}$$

### ⇒ Mode Normalization

→ Thus far, we've been using unit normalized modes, where the maximum amplitude is set to 1.0.

→ Since we can only know the shape of the mode, we can scale them in any way we wish.

→ Mass Normalized Modes.

$$\phi_{N_1} = \frac{\phi_1}{\sqrt{M_1}}$$

→ For the example problem.

$$\phi_{N_1} = \frac{1}{\sqrt{0.691}} \begin{Bmatrix} 1.0 \\ 0.618 \end{Bmatrix} = \begin{Bmatrix} 1.203 \\ 0.743 \end{Bmatrix}$$

$$\phi_{N_2} = \frac{1}{\sqrt{0.691}} \begin{Bmatrix} -0.618 \\ 1.0 \end{Bmatrix} = \begin{Bmatrix} -0.743 \\ 1.203 \end{Bmatrix}$$

Again, in most cases  $M_1 \neq M_2$

→ Using mass normalized modes gives.

$$\phi_N^T [M] \phi_N = \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix} = [I]$$

$$\phi_N^T [k] \phi_N = \underbrace{\begin{bmatrix} \omega_1^2 & 0 \\ 0 & \omega_2^2 \end{bmatrix}}_{\text{this is known as the spectral matrix}} = \Lambda$$

this is known as the spectral matrix

→ Normal Mode Method for Free Vibrations  
(Mode superposition)

→ General approach.

(1) Transform the problem from physical coordinates to modal or normal coordinates.

(2) Solve the resulting uncoupled eqns as a series of SDOF systems.

(3) Transform the SDOF responses back to physical coordinates and sum them (using superposition).

→ Recall from the derivation of element stiffness & mass matrices.

$$u(x, t) = \psi_1(x) u_1(t) + \psi_2(x) u_2(t) \dots$$

→ Similarly,

$$\{u(t)\} = \phi_1 z_1(t) + \phi_2 z_2(t) + \dots$$

↑ <sup>1st mode shape</sup>      ↑ <sup>response of mode 1</sup>      ↑ <sup>second mode shape</sup>  
 ↓ <sup>Node</sup>                  ↓ <sup>↑ response of each mode.</sup>                  ↑ <sup>response of mode 2</sup>

→ More explicitly...

$$\{u\} = [\phi_N] \{z\}$$

↑ <sup>response @ all elts.</sup>      ↑ <sup>shape matrix (Mass normalized)</sup>      ↑ <sup>↑ response of each mode.</sup>

$$[\phi_N]^T [M] [\phi_N] + [\phi_N]^T [K] [\phi_N] = 0$$

Initial disp  
in physical  
coord.

↓

$$\{u_0\} = [\phi_N] \{z_0\}$$

↳ Produces 'n' uncoupled equations or 'n' SDOF systems.

or      ↑ Initial disp. in modal coord.

$$\{z_0\} = [\phi_N]^T [M] \{u_0\}$$

2

$\uparrow$

Initial vel in  
modal coord.

Initial vel  
in physical  
coord.

$$\{\dot{u}_0\} = [\phi_N]^T \{\dot{z}_0\}$$

or

$$\{\dot{z}_0\} = [\phi_N]^T [M] \{\dot{u}_0\}$$

→ Back to example...

→ Solve for the free vibration response  
of the 2 DOF System subjected to  
the following initial cond.

$$u_0 = \begin{Bmatrix} 1.0 \\ 0 \end{Bmatrix} \text{ in } \dot{u}_0 = \begin{Bmatrix} 0 \\ 0 \end{Bmatrix}$$

→ Bring I.C. into Modal Coordinates.

$$z_0 \cdot [\phi_N]^T [M] \{u_0\} = \begin{bmatrix} 1.203 & 0.743 \\ 0.743 & 1.203 \end{bmatrix} \begin{bmatrix} 0.5 & 0 \\ 0 & 0.5 \end{bmatrix} \begin{Bmatrix} 1.0 \\ 0 \end{Bmatrix}$$

$$z_0 = \begin{bmatrix} 0.6015 & 0.3715 \\ -0.3715 & 0.6105 \end{bmatrix} \begin{Bmatrix} 1.0 \\ 0 \end{Bmatrix}$$

$$z_0 = \begin{Bmatrix} \sqrt{0.6015} \\ -0.3715 \end{Bmatrix} \text{ in mode 1}$$

↑ initial disp of  
mode 2

check (Ansle)

$$\text{Mode 1} + \text{Mode 2} = \text{Total}$$

$$0.6015 \begin{Bmatrix} 1.203 \\ 0.743 \end{Bmatrix} = \begin{Bmatrix} 0.724 \\ 0.447 \end{Bmatrix} \text{ in} + \begin{Bmatrix} -0.743 \\ 1.203 \end{Bmatrix} = \begin{Bmatrix} 0.276 \\ -0.447 \end{Bmatrix} = \begin{Bmatrix} 1.0 \\ 0 \end{Bmatrix}$$

$\Rightarrow$  Since  $\dot{z}_0 = \begin{cases} 0 \\ 0 \end{cases} \rightarrow$  No velocity in physical coord.

$\ddot{z}_0 = \begin{cases} 0 \\ 0 \end{cases} \rightarrow$  No velocity in Modal coord.

$\Rightarrow$  Recall from CIVE 801, Lecture, the solution for an unclamped SDOF system subjected to  $u_0$  &  $\dot{u}_0$  is.

$$u(t) = u_0 \cos(\omega t) + \frac{v_0}{\omega} \sin(\omega t)$$

$\Rightarrow$  Mode 1

$$z_1 = \underbrace{0.6015}_{\text{---}} \cos(8.69t) \quad \underbrace{\omega_1}_{\text{---}}$$

$\Rightarrow$  Mode 2

$$z_2 = \underbrace{-0.3715}_{\text{---}} \cos(22.88t) \quad \underbrace{\omega_2}_{\text{---}}$$

$\Rightarrow$  Using superposition.

$$u = \phi_{N_1} z_1 + \phi_{N_2} z_2$$

$$u = \begin{Bmatrix} 0.724 \\ 0.447 \end{Bmatrix} \cos(8.69t) + \begin{Bmatrix} 0.276 \\ -0.447 \end{Bmatrix} \cos(22.88t)$$