

LECTURE 1: Formulation of EOM for MDOF Sys.

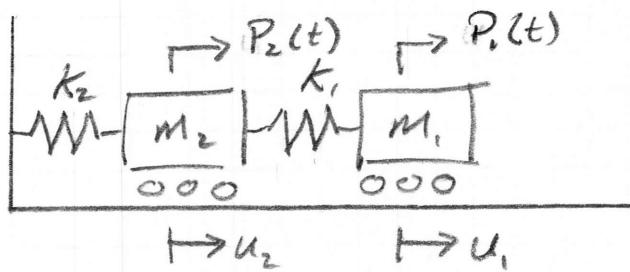
↑
Multiple Degree of Freedom.

→ While SDOF systems are quite useful to illustrate Str. Dyn., in many cases structures cannot be sufficiently simulated as SDOF

↳ E.g. multiple masses that can move independently of one another.

⇒ Review Distributed & Discrete Parameter Models ⇒ pp 2 of CIVE 801 Lecture 2.

→ Discrete, "Mathematical" System.



$$\begin{bmatrix} m_1 & 0 \\ 0 & m_2 \end{bmatrix} \begin{bmatrix} \ddot{u}_1 \\ \ddot{u}_2 \end{bmatrix} + \begin{bmatrix} k_1 & -k_1 \\ -k_1 & k_1+k_2 \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \end{bmatrix} = \begin{bmatrix} P_1 \\ P_2 \end{bmatrix}$$

↑ ↑ ↑
 Mass matrix Stiffness matrix Load or
 Force vector

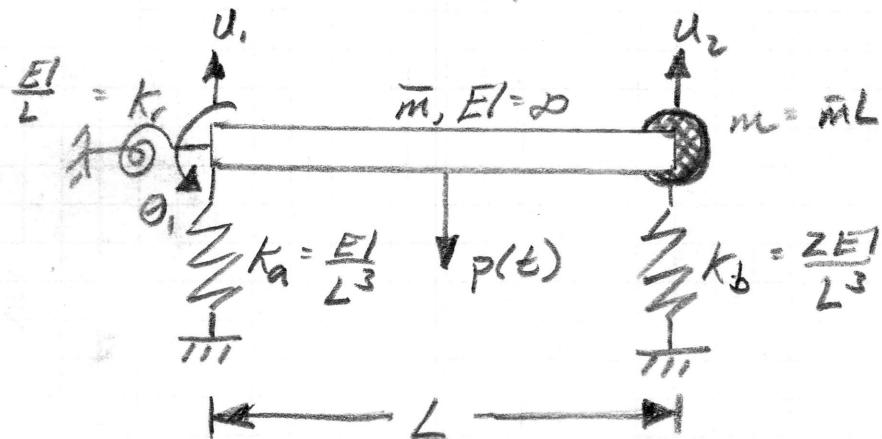
→ similarly,

$$c\ddot{u} \Rightarrow \begin{bmatrix} c_{11} & c_{12} \\ c_{21} & c_{22} \end{bmatrix} \begin{bmatrix} \ddot{u}_1 \\ \ddot{u}_2 \end{bmatrix}$$

→ However due to our inability to estimate damping we will formulate the damping matrix differently... stay tuned.

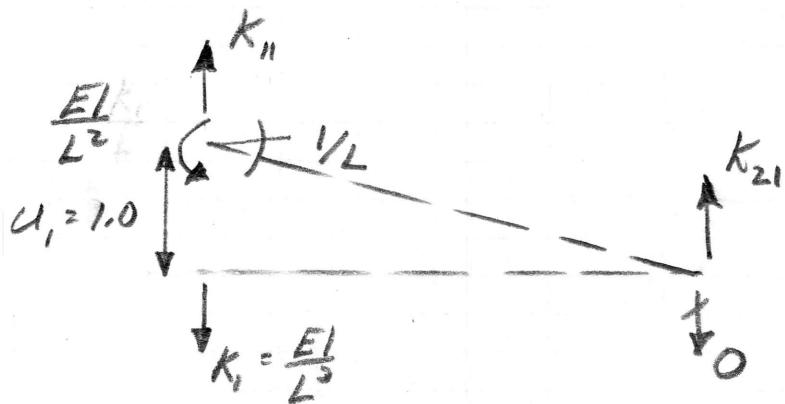
⇒ 3 Approaches to formulating EOM.

(1) "Direct" ⇒ only applicable to simple systems.



→ Start w/ DOFs u_1 & u_2

→ Solve for Stiffness Matrix.



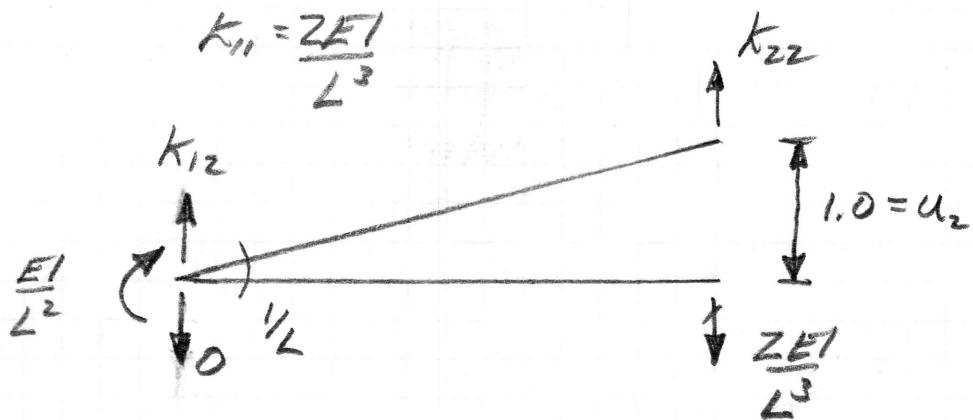
$$\text{G} \sum M_a = 0$$

$$k_{21}(L) + \frac{EI}{L^2} = 0$$

$$k_{21} = -\frac{EI}{L^3}$$

$$\sum F_y = 0$$

$$k_{11} = \frac{2EI}{L^3}$$



$$\therefore \sum M_a = 0$$

$$-\frac{EI}{L^2} - \frac{2EI(L)}{L^3} + k_{22}(L) = 0$$

$$k_{22} = \frac{3EI}{L^3}$$

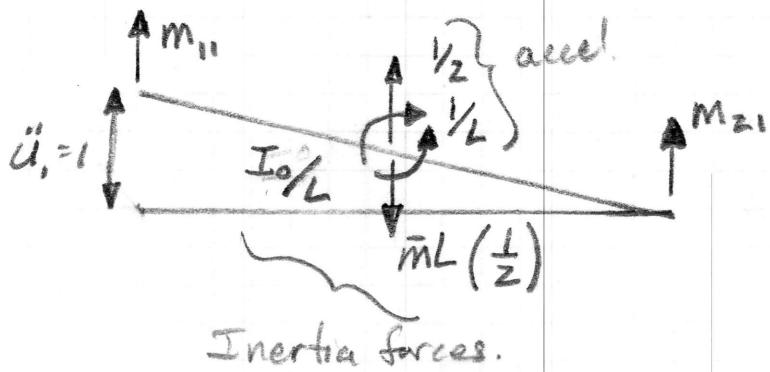
$$\sum F_y = 0$$

$$k_{12} = -\frac{EI}{L^3}$$

$$K = \begin{bmatrix} \frac{2EI}{L^3} & -\frac{EI}{L^3} \\ -\frac{EI}{L^3} & \frac{3EI}{L^3} \end{bmatrix} \quad \left. \right\}$$

Symm & positive definite.

→ Solve for Mass Matrix.



$$\therefore \sum M_a = 0$$

$$\frac{mL^3}{12} - \frac{\bar{m}L}{2} \left(\frac{L}{2}\right) + m_{21}L = 0$$

$$m_{21} = -\frac{\bar{m}L}{12} + \frac{\bar{m}L}{4} = \frac{\bar{m}L}{6}$$

$$\sum F_y = 0$$

$$m_{11} - \frac{\bar{m}L}{2} + \frac{\bar{m}L}{6} = 0$$

$$m_{11} = \frac{\bar{m}L}{3}$$

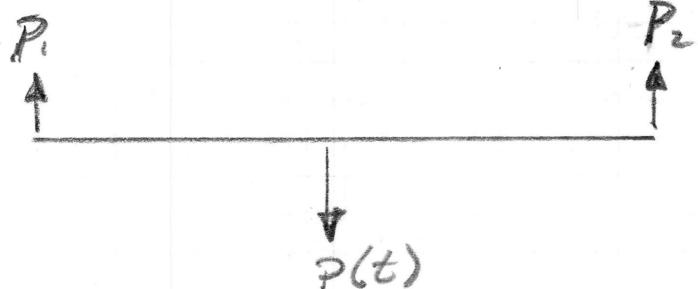
\rightarrow By inspection, $m_{22} = \frac{9\bar{m}L}{8}$ $m_{12} = \frac{\bar{m}L}{6}$

$$M = \begin{bmatrix} \frac{\bar{m}L}{3} & \frac{\bar{m}L}{6} \\ \frac{\bar{m}L}{6} & \frac{9\bar{m}L}{8} \end{bmatrix} \quad \left. \begin{array}{l} \text{symm \& positive} \\ \text{definite.} \end{array} \right\}$$

\rightarrow Alternate approach \Rightarrow lumped mass.

$$M = \begin{bmatrix} \frac{\bar{m}L}{2} & 0 \\ 0 & \frac{3\bar{m}L}{2} \end{bmatrix} \quad \left. \begin{array}{l} \text{lumped mass matrices} \\ \text{are diagonal and more} \\ \text{efficient for computations} \\ \text{but less accurate.} \end{array} \right\}$$

\rightarrow Solve for Load Vector.



\Rightarrow "Statically" Equivalent Forces.

$$G \sum M_a = -P\left(\frac{L}{2}\right)$$

$$P_2 L = -P\left(\frac{L}{2}\right)$$

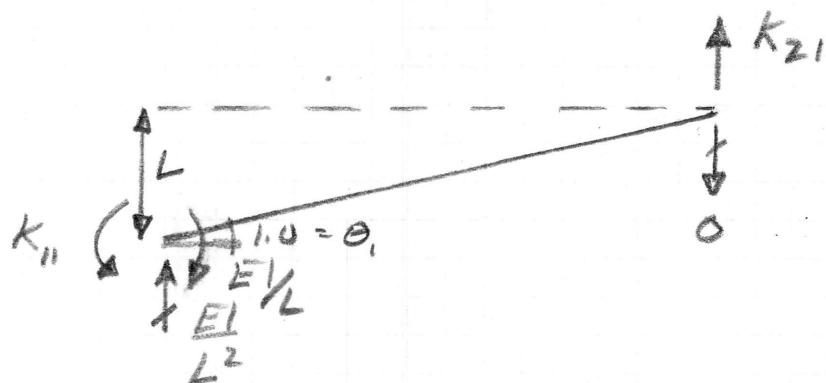
$$P_2 = -\frac{P}{2}$$

$$\sum F_y = -P \Rightarrow P_1 = -\frac{P}{2}$$

$$\bar{m}L \begin{bmatrix} \frac{1}{3} & \frac{1}{6} \\ \frac{1}{6} & \frac{4}{3} \end{bmatrix} \begin{bmatrix} \ddot{u}_1 \\ \ddot{u}_2 \end{bmatrix} + \frac{EI}{L^3} \begin{bmatrix} 2 & -1 \\ -1 & 3 \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \end{bmatrix} = \begin{bmatrix} -\frac{P}{2} \\ -\frac{P}{2} \end{bmatrix}$$

\rightarrow Solve using DOFs θ_1 & u_2

\rightarrow Stiffness Matrix.

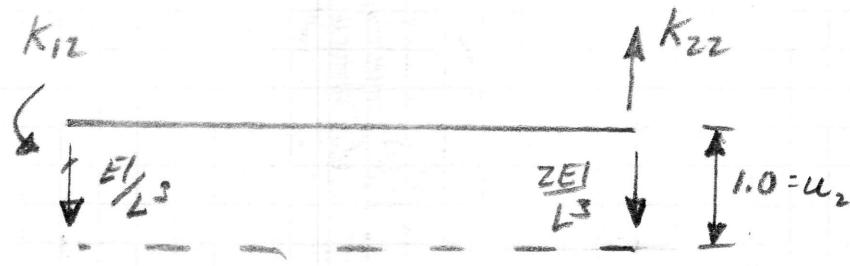


$$G \sum M_b = 0$$

$$-\frac{EI}{L^2}(L) - \frac{EI}{L} + k_{11} = 0 \quad k_{11} = \frac{2EI}{L}$$

$$\sum F_y = 0$$

$$k_{21} = -\frac{EI}{L^2}$$



$$\sum F_y = 0$$

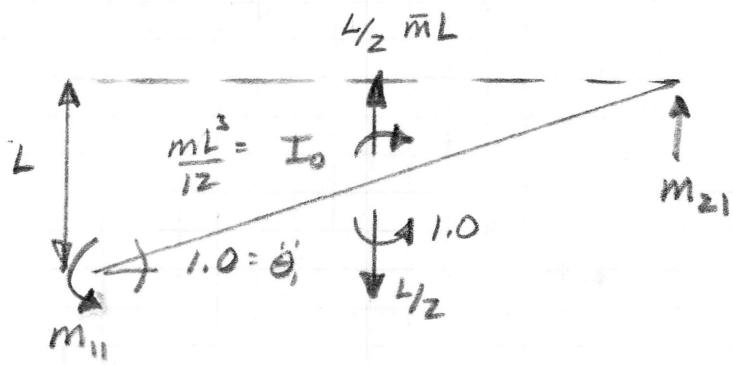
$$K_{22} = \frac{3EI}{L^3}$$

$$\text{G} \sum M_a = 0$$

$$K_{12} + \frac{3EI}{L^2}(L) - \frac{2EI}{L^3}(L) = 0$$

$$K_{12} = -\frac{EI}{L^3} \quad \checkmark$$

→ Solve for Mass Matrix



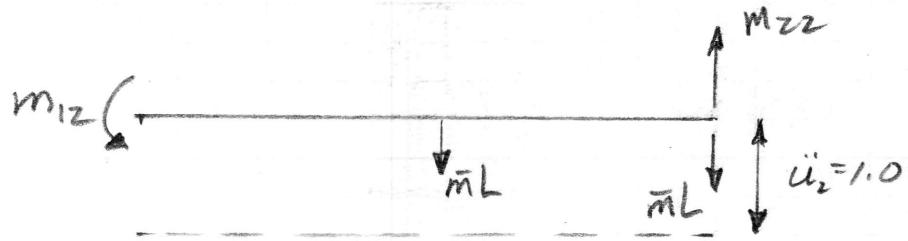
$$\text{G} \sum M_b = 0$$

$$m_{11} - \frac{mL^3}{12} - \frac{\bar{m}L^2}{2}\left(\frac{L}{2}\right) = 0$$

$$m_{11} = \frac{mL^3}{3}$$

$$\sum F_y = 0$$

$$m_{21} = -\frac{\bar{m}L^2}{2}$$



$$\sum F_y = 0$$

$$m_{22} = 2\bar{m}L$$

$$\text{G} \sum M_b = 0$$

$$\bar{m}L\left(\frac{L}{2}\right) + m_{12} = 0$$

$$m_{12} = -\frac{\bar{m}L^2}{2} \quad \checkmark$$

→ Solve for Load Vector.



$$\sum F_y = -P$$

$$P_2 = -P$$

$$\text{G} \sum M_c = -PL/2$$

$$-PL + P_1 = -\frac{PL}{2}$$

$$P_1 = \frac{PL}{2}$$

$$\bar{m}L \begin{bmatrix} \frac{L^2}{3} & -\frac{L}{2} \\ -\frac{L}{2} & 2 \end{bmatrix} \begin{Bmatrix} \ddot{\theta}_1 \\ \ddot{u}_2 \end{Bmatrix} + \frac{EI}{L^2} \begin{bmatrix} 2L^2 & -L \\ -L & 3 \end{bmatrix} \begin{Bmatrix} \theta_1 \\ u_2 \end{Bmatrix} = \begin{Bmatrix} PL/2 \\ -P \end{Bmatrix}$$

Are the two representations (i.e. EOMs)
equivalent?

→ Recall for SDOF

$$m\ddot{u} + ku = 0$$

$$u = A \sin(\omega t) \quad (\text{assume harmonic motion})$$

$$\ddot{u} = -A\omega^2 \sin(\omega t)$$

$$-m A\omega^2 \sin(\omega t) + k A \sin(\omega t) = 0$$

$$k - m\omega^2 = 0$$

$$\omega = \sqrt{\frac{k}{m}}$$

→ Use a similar approach for MDOF (sqrt w/
2DOF)

$$[M]\{\ddot{u}\} + [k]\{u\} = 0$$

$$u_i = \begin{cases} u_{1,i} \\ u_{2,i} \end{cases} \sin(\omega_i t)$$

↑ can have multiple frequencies.
can have multiple shape functions
↳ "mode" shapes.

$$\ddot{u}_i = -\omega_i^2 \begin{cases} u_{1,i} \\ u_{2,i} \end{cases} \sin(\omega_i t)$$

↑ shape of disp is the same as the shape of acceleration.

\Rightarrow Subst $m_{1,1}, \dots$

$$-w_i^2 \begin{bmatrix} m_{11} & m_{12} \\ m_{12} & m_{22} \end{bmatrix} \begin{Bmatrix} u_{1,i} \\ u_{2,i} \end{Bmatrix} \xrightarrow{\text{sin } \omega_i t} + \begin{bmatrix} k_{11} & k_{12} \\ k_{12} & k_{22} \end{bmatrix} \begin{Bmatrix} u_{1,i} \\ u_{2,i} \end{Bmatrix} \xrightarrow{\text{sin } \omega_i t} = 0$$

$$\left[\begin{bmatrix} k_{11} & k_{12} \\ k_{12} & k_{22} \end{bmatrix} - w_i^2 \begin{bmatrix} m_{11} & m_{12} \\ m_{12} & m_{22} \end{bmatrix} \right] \begin{Bmatrix} u_{1,i} \\ u_{2,i} \end{Bmatrix} = 0$$

$$\{[K] - w_i^2 [M]\} \{ \phi_i \} = 0$$

\uparrow natural frequency i \uparrow mode shape i

\Rightarrow Eigen problem

w_i^2 = eigen values (square of freq.)

ϕ_i = eigen vectors (mode shapes)

\Rightarrow Trivial Solutions.

$\phi_i = 0 \Rightarrow$ no vibration.

\Rightarrow Non-trivial Solution

$$\det \left| [K] - w_i^2 [M] \right| = 0$$

$$\det \begin{vmatrix} k_{11} - w_i^2 m_{11} & k_{12} - w_i^2 m_{12} \\ k_{12} - w_i^2 m_{21} & k_{22} - w_i^2 m_{22} \end{vmatrix} = 0$$

$$k_{11}k_{22} + w_i^4 m_{11}m_{22} - w_i^2 (k_{11}m_{22} + k_{22}m_{11})$$

$$-k_{12}^2 + w_i^4 m_{12}^2 - 2w_i^2 k_{12} m_{12} = 0$$

$$w_i^4 (m_{11}m_{22} + m_{12}^2) + w_i^2 (k_{11}m_{22} + k_{22}m_{11} - 2k_{12}m_{12})$$

$$+ k_{11}k_{22} - k_{12}^2 = 0$$

\Rightarrow Quadratic in terms of w_i^2

$$w_i^2 = \frac{-b \pm \sqrt{b^2 - 4ac}}{2a}$$

$$a = \det[M]$$

$$b = -(k_{11}m_{22} + k_{22}m_{11} - 2k_{12}m_{12})$$

$$c = \det[K]$$

\Rightarrow Once you solve for w_i^2 , you can sub back in & solve for ϕ

$$(k_{11} - w_i^2 m_{11})(\phi_{1,i}) + (k_{12} - w_i^2 m_{12})(\phi_{2,i}) = 0$$

$$r_i = \frac{\phi_{1,i}}{\phi_{2,i}} = \frac{-k_{12} + w_i^2 m_{12}}{k_{11} - w_i^2 m_{11}} \quad \left. \begin{array}{l} r_i < 1.0 \\ r_i > 1 \end{array} \right\} \text{can only solve for mode shapes in a relative sense...}$$

↑
modal amplitude ratio.

$$\phi_i = \begin{cases} \{r_i\} & \text{or} \\ \{1.0\} & \{Y_{r_i}\} \end{cases}$$

This provides "unit normalized" mode shapes..

→ Solve for ω_i & ϕ_i for dof u_1 & u_2

$$a = \bar{m}^2 L^2 \left(\frac{4}{9} - \frac{1}{36} \right) = \frac{15}{36} \bar{m}^2 L^2$$

$$b = -\frac{EI}{L^2} \left(\frac{8}{3} + 1 + \frac{1}{3} \right) = -4 \frac{EI}{L^2}$$

$$c = \left(\frac{EI}{L^3} \right)^2 (6-1) = 5 \frac{E^2 I^2}{L^6}$$

$$\omega_i^2 = \frac{4 \frac{EI}{L^2} \pm \sqrt{16 \frac{E^2 \bar{m}^2}{L^4} - 4 \left(\frac{15}{36} \right) (5) \frac{E^2 I^2 \bar{m}^2}{L^6}}}{2 \left(\frac{15}{36} \right) \bar{m}^2 L^2}$$

$$\omega_i^2 = (4.8 \pm 3.3) \frac{EI}{\bar{m} L^4}$$

$$* \quad \omega_1 = 1.22 \sqrt{\frac{EI}{\bar{m} L^4}}$$

$$* \quad \omega_2 = 2.85 \sqrt{\frac{EI}{\bar{m} L^4}}$$

$$r_1 = \frac{\frac{EI}{L^2} + 1.488 \frac{EI}{\bar{m} L^4} \frac{\bar{m} L}{6}}{2 \frac{EI}{L^2} - 1.488 \frac{EI}{\bar{m} L^4} \frac{\bar{m} L}{3}} = \frac{1.248}{1.504} = 0.83$$

$$* \quad \phi_1 = \begin{cases} 0.83 \\ 1.0 \end{cases}$$

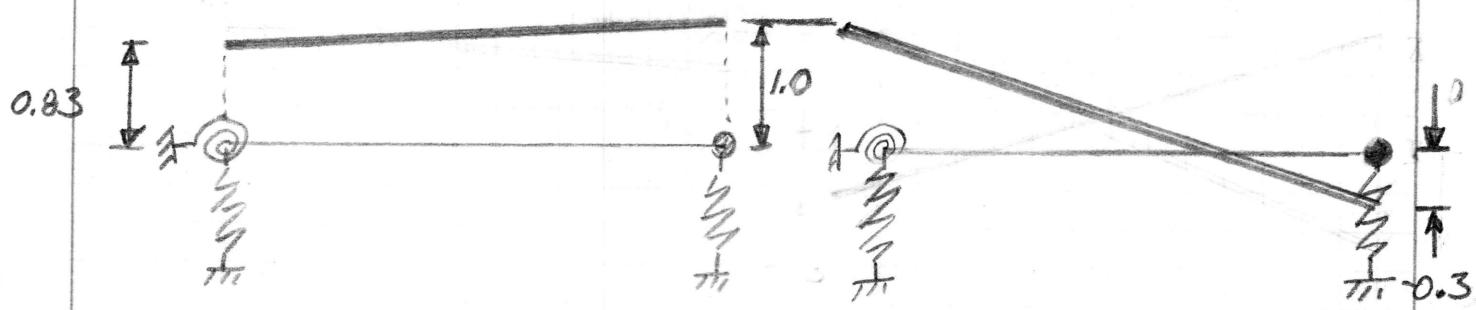
$$r_2 = \frac{1 + 8.123 \left(\frac{1}{6} \right)}{2 - 8.123 \left(\frac{1}{3} \right)} = \frac{2.35}{-0.707} = -3.32$$

$$* \quad \phi_2 = \begin{cases} 1.0 \\ -0.30 \end{cases}$$

→ Plot Mode Shapes.

$$\underline{\text{Mode 1}} \quad \omega_1 = 1.22 \sqrt{\frac{EI}{\bar{m}L^4}}$$

$$\underline{\text{Mode 2}} \quad \omega_2 = 2.85 \sqrt{\frac{EI}{\bar{m}L^4}}$$



AMPAD

→ Solve for $\omega_i \notin \phi_i$ for $\theta_i \notin u_2$

$$a = \bar{m}^2 L^2 \left(\frac{2L^2}{3} - \frac{L^2}{4} \right) = \frac{5\bar{m}^2 L^4}{12}$$

$$b = -\frac{EI\bar{m}}{L^2} (4L^2 + L^2 - L^2) = -4EI\bar{m}$$

$$c = \frac{E^2 I^2}{L^6} (6L^2 - L^2) = 5 \frac{E^2 I^2}{L^4}$$

$$\omega_i^2 = \frac{-4EI\bar{m} \pm \sqrt{16E^2 I^2 \bar{m}^2 - 4(\frac{5}{12})(5)E^2 I^2 \bar{m}^2}}{(2)\frac{5\bar{m}^2 L^4}{12}}$$

$$\omega_i^2 = (4.8 \pm 3.3) \frac{EI}{\bar{m}L^4}$$

$$* \quad \omega_1 = 1.22 \sqrt{\frac{EI}{\bar{m}L^4}}$$

$$* \quad \omega_2 = 2.85 \sqrt{\frac{EI}{\bar{m}L^4}}$$

$$\gamma_1 = \frac{\frac{EI}{L^2} + 1.488 \frac{EI}{mL^4} \left(-\frac{mL^2}{2}\right)}{\frac{2EI}{L} - 1.488 \frac{EI}{mL^4} \frac{mL^3}{3}} = \frac{0.1256/L}{1.504} = \frac{0.17}{L}$$

* $\phi_1 = \begin{cases} 0.17 \\ 1.0 \end{cases} \quad \leftarrow \text{Note, when you mix dofs (rot & trans), the mode shapes are in terms of } L.$

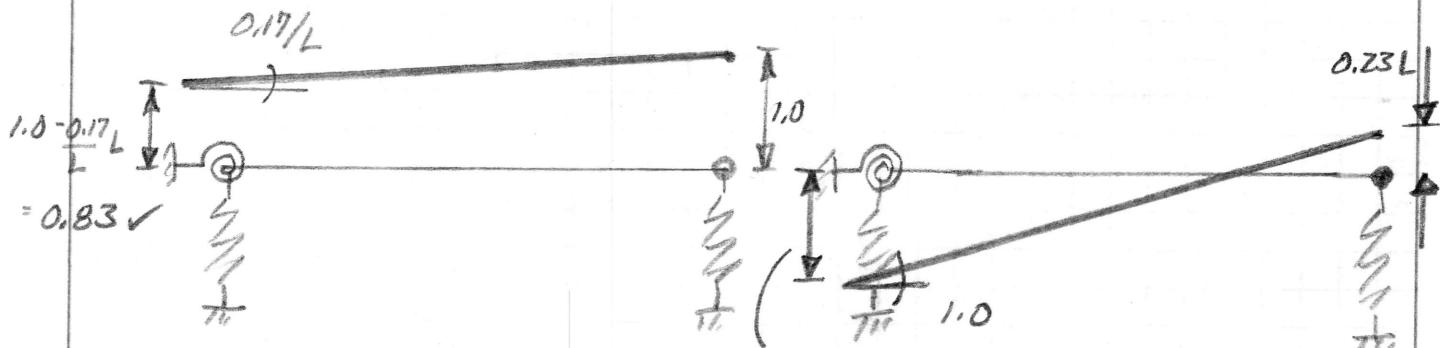
$$\gamma_2 = \frac{1 + 8.122 \left(-\frac{1}{2}\right)}{L \left(2 - 8.122 \left(\frac{1}{3}\right)\right)} = \frac{-3.06/L}{-0.707} = \frac{4.33}{L}$$

* $\phi_2 = \begin{cases} 1.0 \\ 0.23L \end{cases}$

→ Plot Mode Shapes

$$\text{Mode 1 } w_1 = 1.22 \sqrt{\frac{EI}{mL^4}}$$

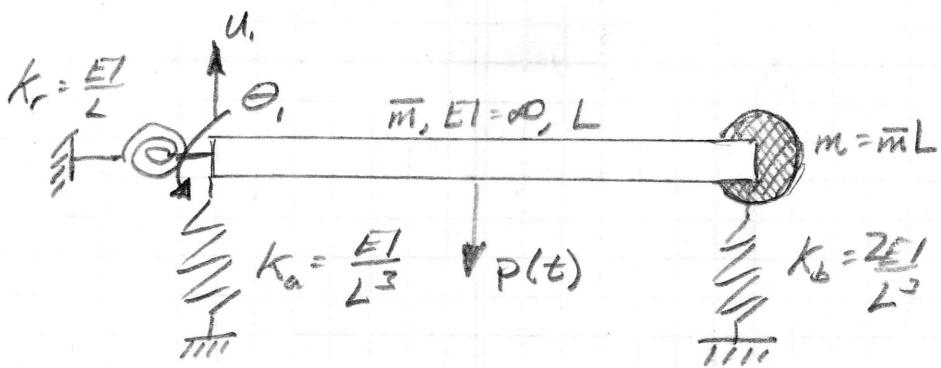
$$\text{Mode 2 } w_2 = 2.85 \sqrt{\frac{EI}{mL^4}}$$



$$\phi_2 = \begin{cases} -0.77L \\ 0.23L \end{cases} \rightarrow$$

or divide by -0.77L
 $\checkmark \phi_2 = \begin{cases} 1.0 \\ -0.3 \end{cases} \leftarrow$

Homework #1



- (1) Solve for $[k]$ & $[M]$ using u & θ ,
- (2) Solve for w_i & ϕ_i and show these are equivalent to the ones obtain in class
- (3) Write the equation of motion.