# **Fuel**







#### **Fuel Uses**

- Fuel has several uses on a modern civil airliner:
  - Power the engines
  - Control of the aircraft centre of gravity and mass distribution
  - Heat sinking for other systems

The fuel system comprises of tanks, plumbing, pumps and control to facilitate these multiple uses.







#### What is it?

- Jet fuel is typically kerosene based (e.g. Jet A-1 the world wide standard fuel)
- Jet A-1 has a higher flash point and is less flammable than piston engine fuels (e.g. AVGAS AViation GASoline) making it safer to handle.
- Fuel used for international aviation generally attracts no tax (compare with 175% tax on petrol)







#### Some numbers

■ A380....

MTOW ~ 570 tons

Made up of;

Plane ~280 tons

Cargo ~ 90 tons

Fuel ~ 200 tons

(320,000 litres or 70,000 gallons)

~7m x7m x7m

Maximum range ~ 9,500 miles

Fuel economy ~1/8 mile per gallon.









## **Fuel storage**

■ The most efficient location to store the fuel is within the wing box (referred to as the "wet wing")

- All joint and access panels must be suitably sealed
- A natural tendency of the fuel to flow towards pumps located at the wing roots is essential.
- Intermediate ribs have to allow the fuel to flow and act as fuel baffles (to prevent sloshing during manoeuvring).
- The wet wings may extend to the fuselage centre line or end at the root ribs.
- Provisions for protection of the tank in the case of belly landing (skid) must be made.







## **Fuel storage**

- Generally 90% of wing volume between spars and skin outer surface is usable; although this may vary from only 75% near the tip to 95% near the root.
  - Loss is made up of
    - Structure displacement: 5%
    - Expansion space (mandatory): 2%
    - Trapped air: 1%
    - Unusable fuel: 0.5%
    - Equipment displacement: 1%
    - Sealant: 0.5%
- In the case where the fuel volume in the wing is not sufficient, fuel may be stored elsewhere
  - Cabin under floor
  - Aft fuselage or tail cone
  - Cabin tanks ahead of the fuselage pressure bulkhead
  - Wet empennage (trim tanks)





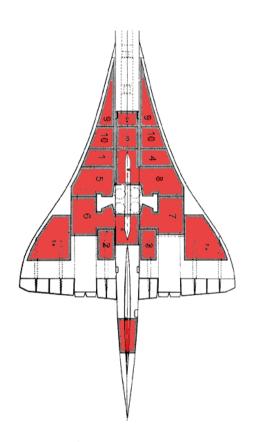


#### AVDASI 1 AENG 10001

### **Fuel tanks**



Generic civil airliner



Concorde







## Fuel gauging

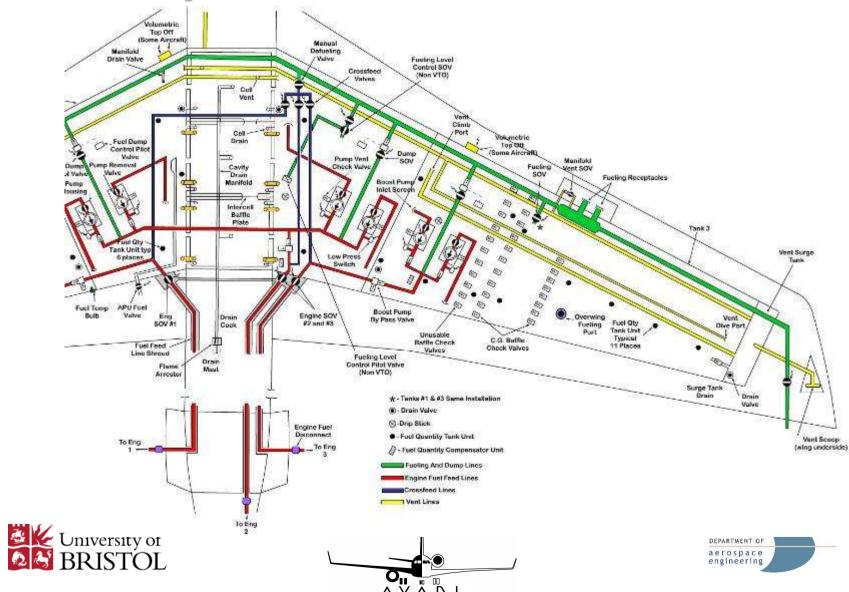
- Since running out of fuel has dire consequences it is important to put enough in and know how much is used.
- Capacitance systems Measure the variation of capacitance between the two concentric tubes immersed in the fuel.
- Ultra sonic systems Measure the time of an echo from the fuel surface.
- Fuel measuring systems are referred to by their precision;
  - Class I approx. 5% accurate in-flight
  - Class II approx. 2% accurate in-flight (business and regional aircraft)
  - Class III approx. 1% accurate in-flight (long-range business and commercial aircraft); best precision today is no better than 0.5% accurate







## 727 fuel system



## **Fuel delivery**

- The ideal primary delivery system is by gravity; It is simple, lightweight, cost effective and very reliable.
  - Not utilised as a primary delivery system for almost all modern transport aircraft.
- Most aircraft use powered systems;
  - Motive flow
    - It uses a small quantity of high pressure fuel provided by a mechanical pump at the engine to power a jet (ejector) pump at the wing tank.
    - Apart from the HMU at the engine, this system uses no mechanical moving parts, making it simple, low cost, and extremely reliable
  - Hydraulic/electric pumps







#### **Fuel Issues**

#### Fuel freezing

- Water in fuel may freeze out and cause a blockage.
- Fuel is monitored for water content and can be heated.

#### Venting/inerting of the fuel tanks

- Tanks must be vented to allow air in and out during use and re-fuelling
- This creates 'ullage' space above the liquid fuel filled with air and fuel vapour. At certain ratios the mixture of fuel-to-air is explosive.
- Inerting involves using a gas, e.g. nitrogen to fill the void rather than air.
  Other systems use a bladder to contain the fuel.
- The technology is costly and the general approach has been to limit ignition sources through regulation.







#### **Fuel Issues**

**A380....** 

MTOW ~ 570 tons

Made up of;

Plane ~ 280 tons

Cargo ~ 90 tons

Fuel ~ 200 tons

Max landing weight ~390 tons



On larger aircraft it is necessary to be significantly lighter to land than to take-off. Normally fuel burn in flight takes care of this but if an emergency landing is required it may be necessary to jettison fuel.









# Re-fuelling

- Gravity fuelling is done by pouring fuel through an opening on top of the wing at around 75% wing tank span
  - This location is chosen such that it is impossible to over-fuel the aircraft.
  - Fuel brought to this level should leave a large enough "air bubble" for the proper functioning of the fuel system
- Modern aircraft also have pressurized refuelling capability
  - A panel typically at a wing root or under a wing allows a connection to a standard fuel nozzle from a fuelling truck – an aircraft may have multiple nozzles
  - Pressure of 40 to 50 psi for fuelling and -8 to -3 psi for de-fuelling
  - Typical fuelling rates are 100's Gallons per minute per nozzle a large aircraft can take some time to refuel.







# **Energy in refuelling**

- We normally think of the power during flight as limiting the possibility of battery powered aircraft, but what about recharging batteries?
  - An A380 flies for ~17 hours on a full tank of fuel
  - It can be refuelled in less then an hour......
  - If we have 4 nozzles each feeding 32 l/sec and A1 jet fuel has an energy density of 53 MJ/l, the 'power' is:
  - $-4 \times 32 \times 53 = 6784 \text{ MW}!!!$
  - For reference Hinkley 'c' will be 3200MW......





