

Flight Design Project 2 Wing dynamics

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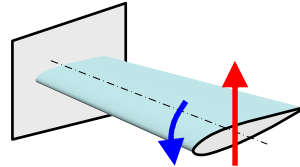
Introduction

- For details see PDF: **Flight Project 2 – Wing dynamics**
- Evaluate and assess properties of your design:
 - Stiffness,
 - (Fundamental) natural (or “resonant”) frequency,
 - Validate your predictions (using smart phone!)
- Prepare brief technical discussion on this topic:
 - Between 1 and 2 A4 sides,
 - Focus on using tables, graphs, etc.

The need for dynamic properties of wings



Highly elastic designs

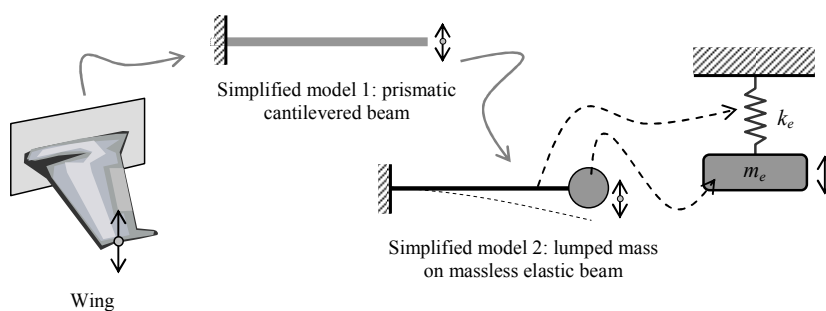


Destructive aero-structural interaction:

- flutter,
- divergence,
- excessive vibrations,
- ...

<http://www.youtube.com/watch?v=aE4pBmZrriU&feature=related>

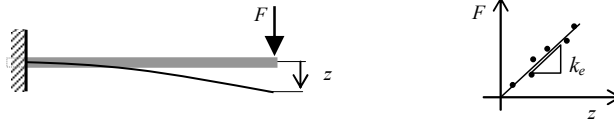
Natural frequency of a simple spring-mass oscillator



$$f_0 = \frac{1}{2\pi} \left(\frac{k_e}{m_e} \right)^{1/2} \quad [\text{Hz}]$$

Stiffness of your wing

... for your specific constraint and loading conditions!!!



Stiffness is defined for this configuration as follows: $F = k_e z$

... or differently: $\text{stiffness} = \frac{\text{applied load}}{\text{deformation}}$

Can you relate the above formulas to the following classical formula for the tip loaded cantilevered beam?

$$z = \frac{F L^3}{3EI}$$

Find your stiffness experimentally



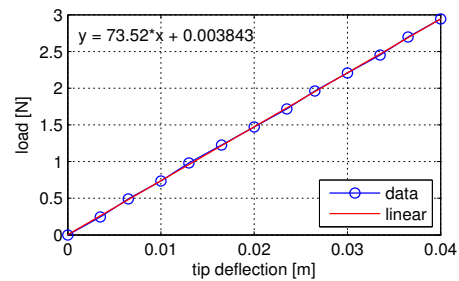
For example: 6 loads × 3 locations

Unit	z [?]	F [?]	Unit	z [?]	F [?]	Unit	z [?]	F [?]
1			1			1		
2			2			2		
3			3			3		
4			4			4		
...				

$k_{e,1}$

$k_{e,2}$

$k_{e,3}$



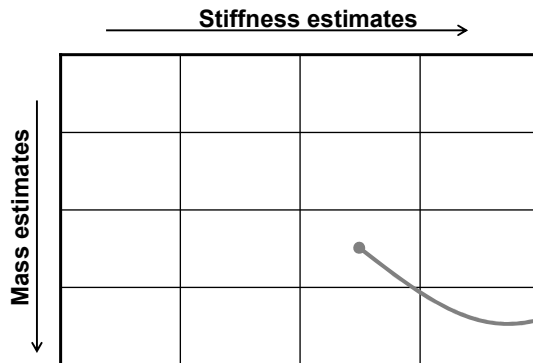
Calculate your natural frequency

Equivalent mass of a wing: $m_e = \gamma m_{wing}$

$$\gamma \approx 0.24$$

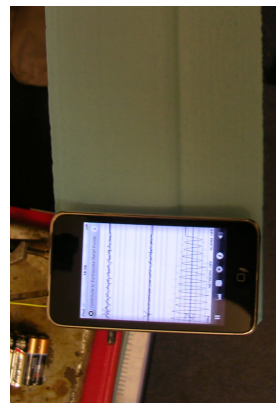
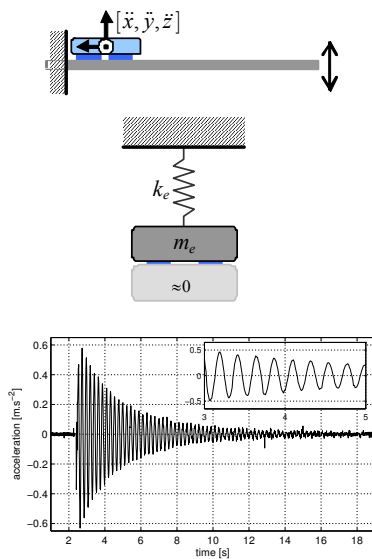
$$\gamma \in [0.23, 0.33]$$

For more details see
FDP – Dynamics note



$$f_0 = \frac{1}{2\pi} \left(\frac{k_e}{m_e} \right)^{1/2} [\text{Hz}]$$

Assess (and validate) your predictions



For example (my results):
 $f_{\text{exp}} = 18.6 \text{ Hz}$
 $f_{\text{model}} = 17.57 \text{ Hz } (\gamma = \dots)$
 Error = -5.5 % etc. etc.