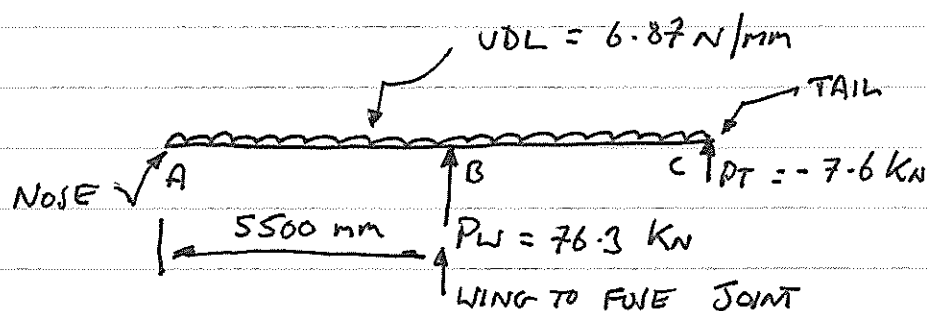


CALCULATION OF SHEARING FORCE & BENDING MOMENT MAXIMUMS FOR G280 BUSINESS JET EXAMPLE.

CONSIDER FUSELAGE AS A BEAM WITH A UNIFORMLY DISTRIBUTED LOAD OF 6.87 N/mm

WING TO FUSELAGE JOIN IS AT POINT 5500 mm AFT FROM THE NOSE



SHEAR FORCE AT POINT B = SUMMATION OF ALL FORCES FORWARD OF POINT B

$$\begin{aligned}
 &= \text{UDL} \times \text{LENGTH} \\
 &= 6.87 \times 5500 \quad \left(\frac{\text{KN}}{\text{mm}} \times \text{mm} \right) \\
 &= \underline{\underline{37.8 \text{ KN}}}
 \end{aligned}$$

BENDING MOMENT AT POINT B = SUMMATION OF ALL MOMENTS ACTING FORWARD OF POINT B

$$\begin{aligned}
 &= \text{MOMENT DUE TO UDL} \\
 &= (\text{TOTAL FORCE}) \times (\text{DISTANCE TO CENTRE OF ACTION OF FORCE}) \\
 &= (6.87 \times 5500) \times (5500/2) \\
 &= 37.785 \text{ KN} \times 2750 \text{ mm} \\
 &= 103.9 \times 10^6 \text{ KNmm} \\
 &= \underline{\underline{103.9 \text{ KNm}}}
 \end{aligned}$$