

Advanced Bending and Torsion

Unsymmetric Bending – Definitions

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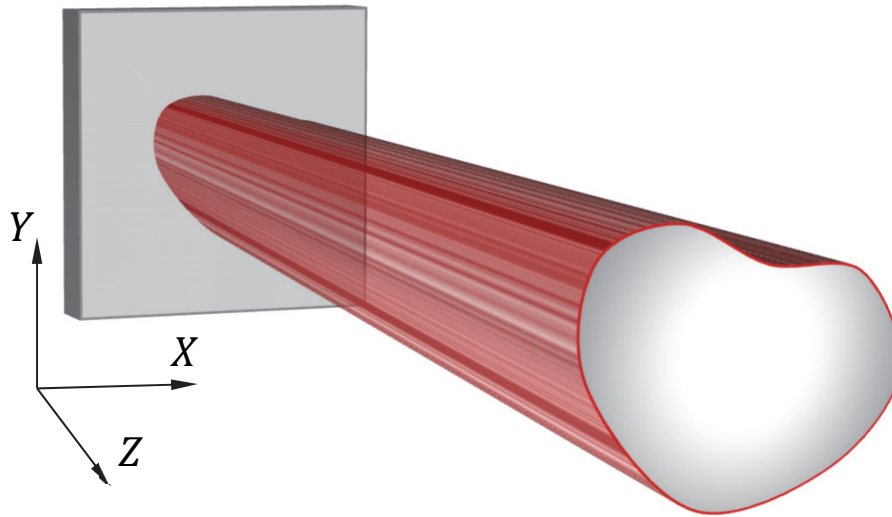
1. Axes

1. Global structural axes (X, Y, Z)
2. Local section axes (x, y, z)
3. Section principal axes (1, 2)
4. Loading axes
5. Neutral axis (NA)

2. Area properties

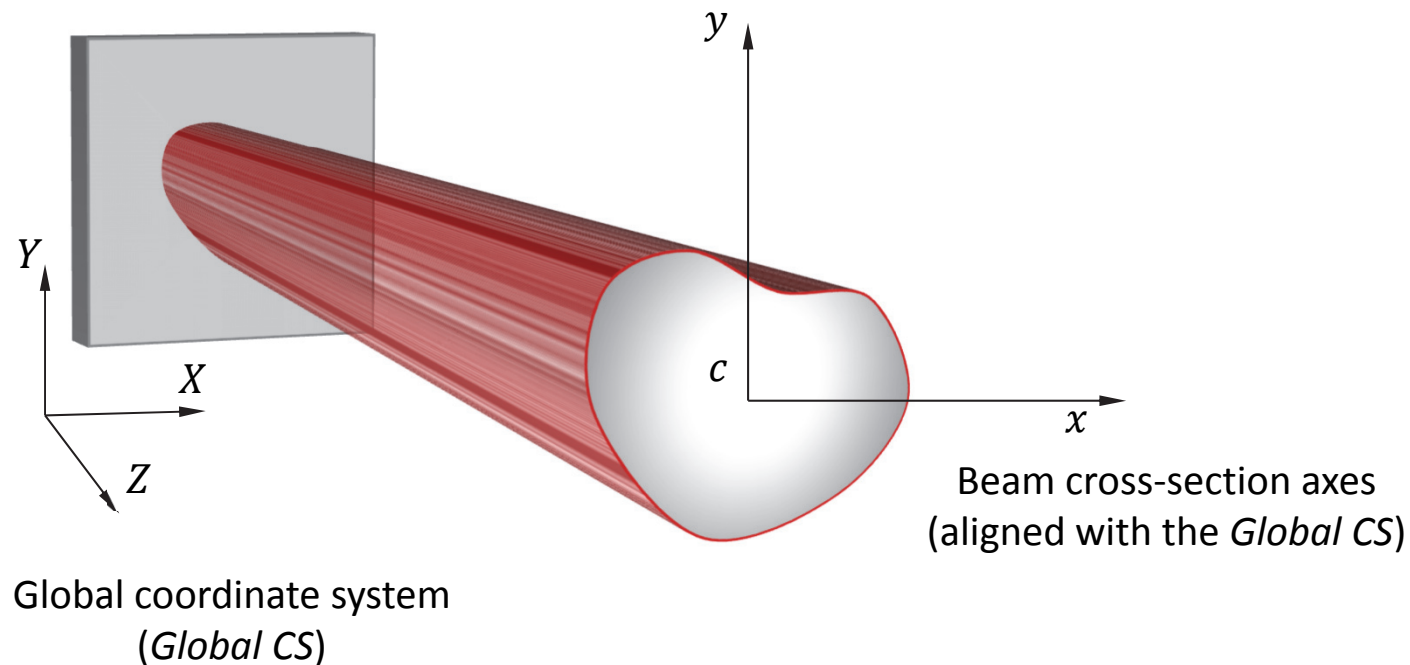
1. Area, centroid, 1st moments of area
2. 2nd moments of area (w.r.t. x, y axes)
3. Transformed 2nd moments of area (w.r.t. x', y' axes)
4. Principal 2nd moments of area (w.r.t. 1, 2 axes)
5. Product 2nd moments of area

- A global, structural coordinate system (*Global CS*) must be defined
- The structure is described in space based on the *Global CS*, which we name XYZ (uppercase roman typeface)
- The origin and orientation of the *Global CS* are somewhat arbitrary; here we will always align Z with the beam's length, so that all cross-section planes are parallel to the XY plane

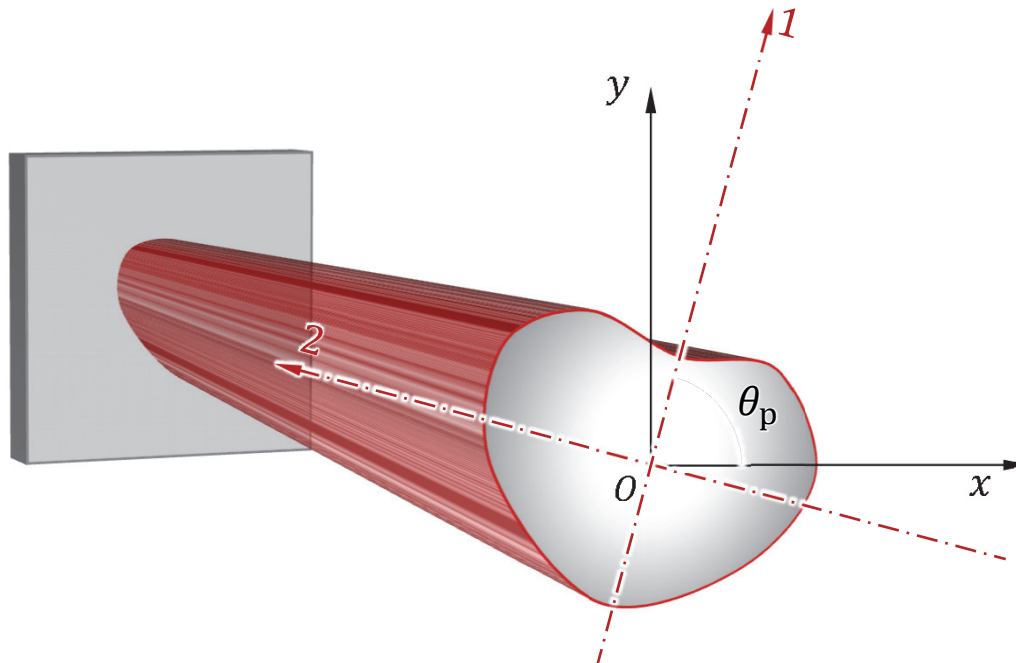


Global coordinate system
(*Global CS*)

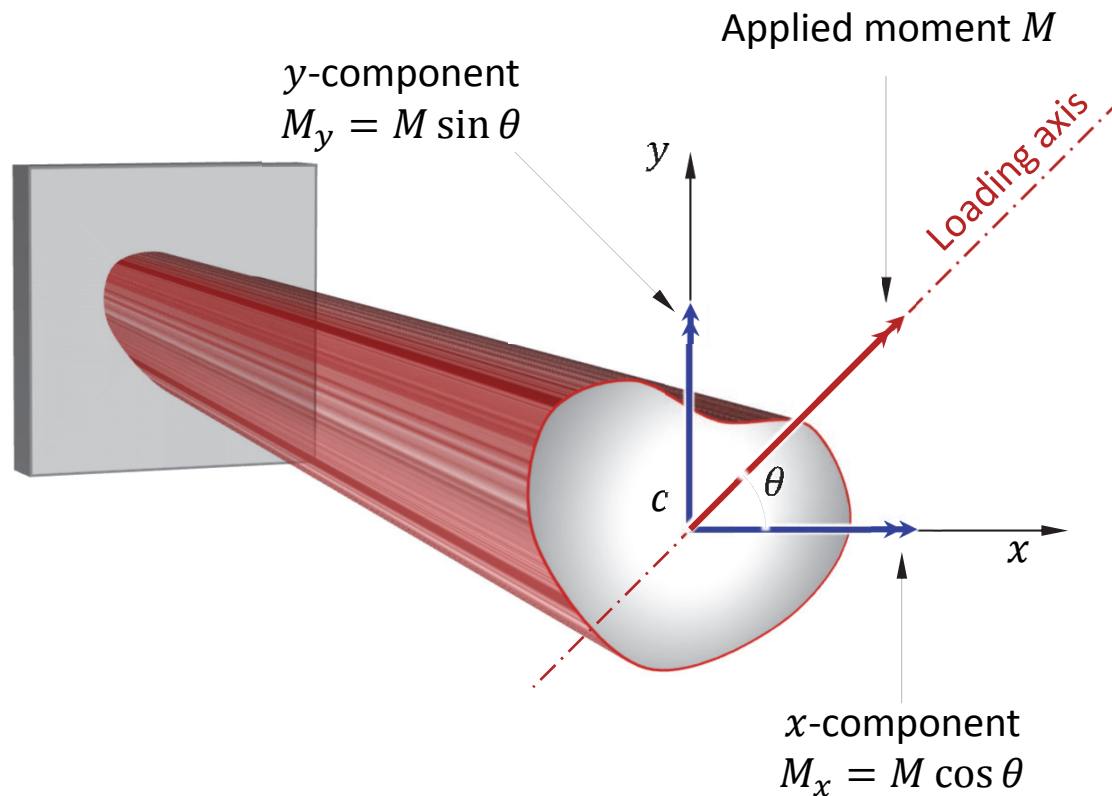
- A local section coordinate system (*Section CS*) is defined with origin at the centroid of the cross-section, but aligned with the *Global CS*
- The Section CS is defined by (x, y) (lowercase roman typeface)



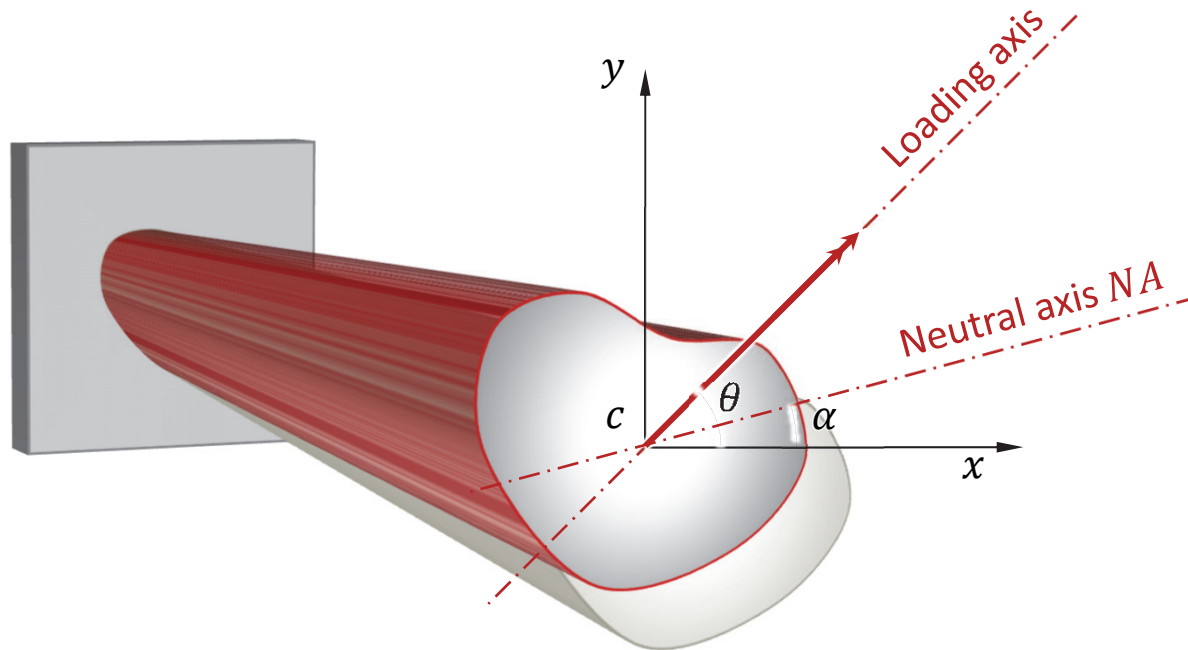
- Axes about which the **2nd moment of area is maximum or minimum**
 - pass through the centroid
 - coincident with symmetry axes (if applicable)
- Principal axis of minimum I -value is the axis of least bending resistance, *i.e.* the axis about which the beam will preferentially bend



- Axes of resultant applied bending moment
 - *i.e.* axis about which the beam is being forced to bend
- Moments are vectorial quantities and can be resolved onto chosen reference axis:



- Axis about which the beam **actually** bends once loaded
- Passes through the centroid
- Direct stress is zero on the neutral axis and maximum at furthest distance from it, varying linearly, as before
- Bending deflection is perpendicular to the neutral axis



- Area:

$$A = \int dA \text{ or } \sum A_i$$

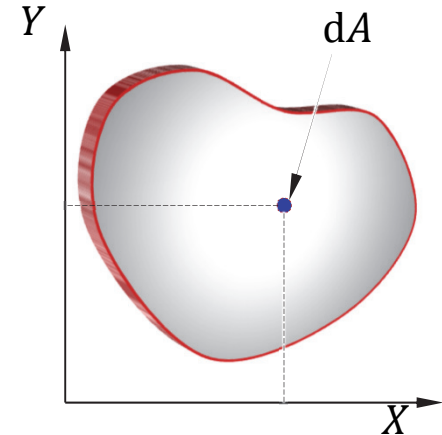
- 1st moment:

$$Q_{XX} = \int Y dA$$

$$(\text{= } \sum \bar{Y}_i A_i)$$

$$Q_{YY} = \int X dA$$

$$(\text{= } \sum \bar{X}_i A_i)$$



- Centroid:

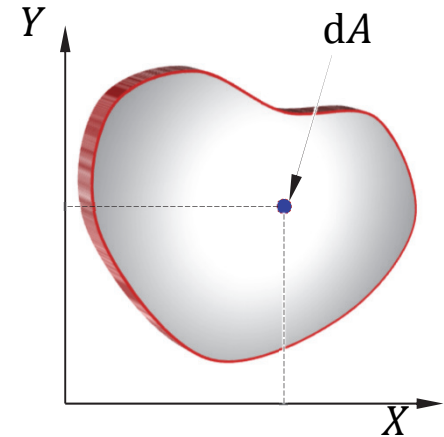
$$\bar{X} = \frac{Q_{YY}}{A} \text{ or } \bar{X} = \frac{\sum \bar{X}_i A_i}{\sum A_i}$$

$$\bar{Y} = \frac{Q_{XX}}{A} \text{ or } \bar{Y} = \frac{\sum \bar{Y}_i A_i}{\sum A_i}$$

- Again, same as before:

$$I_{XX} = \int Y^2 dA$$

$$I_{YY} = \int X^2 dA$$



- Now, the **product second moment of area** is:

$$I_{XY} = \int XY dA$$

- Named I_{XY}
- Can be positive, negative or zero, depending on which 'quadrant' of the XY plane the section is in
- Zero about principal axes (areas in opposite quadrants cancel out)
- Can be calculated by parallel axis theorem

$$I_{XX} = \int Y^2 \, dA = \int (\bar{Y} + y)^2 \, dA$$

$$I_{XX} = \bar{Y}^2 \int dA + 2 \bar{Y} \int y \, dA + \int y^2 \, dA$$

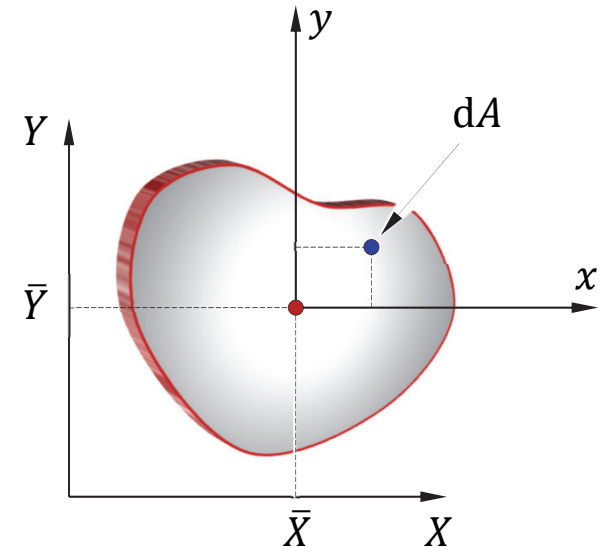
$$I_{XX} = \int y^2 \, dA + A \bar{Y}^2$$

- Similarly:

$$I_{YY} = \int x^2 \, dA + A \bar{X}^2$$

- Note that these are identical to the expressions seen in StM1 for compound sections:

$$I = \sum I_i + A_i d_i^2 \quad (\text{N.B. StM1 notation differs from StM2})$$



- The same principles apply to I_{XY} :

$$I_{XY} = \int XY \, dA = \int (\bar{X} + x)(\bar{Y} + y) \, dA$$

$$I_{XY} = \bar{X} \bar{Y} \int dA + \bar{X} \int y \, dA + \bar{Y} \int x \, dA + \int x y \, dA$$

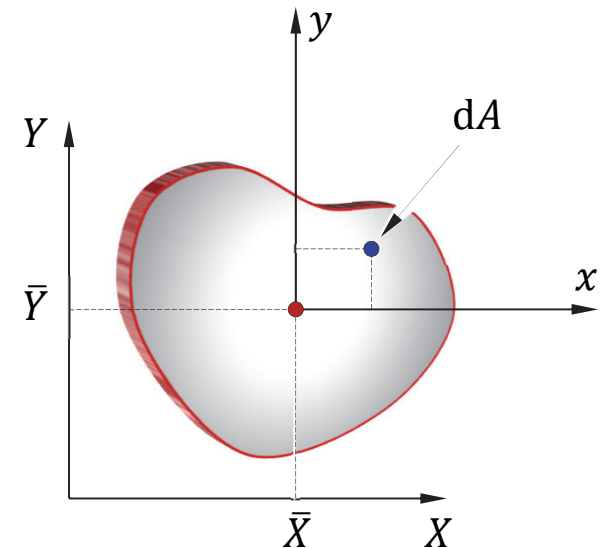
- But by definition:

$$\int x \, dA = 0 \text{ and } \int y \, dA = 0$$

- Therefore:

$$I_{XY} = \bar{X} \bar{Y} \int dA + \int x y \, dA$$

$$I_{XY} = I_{xy} + \bar{X} \bar{Y} \int dA$$



Parallel axes theorem

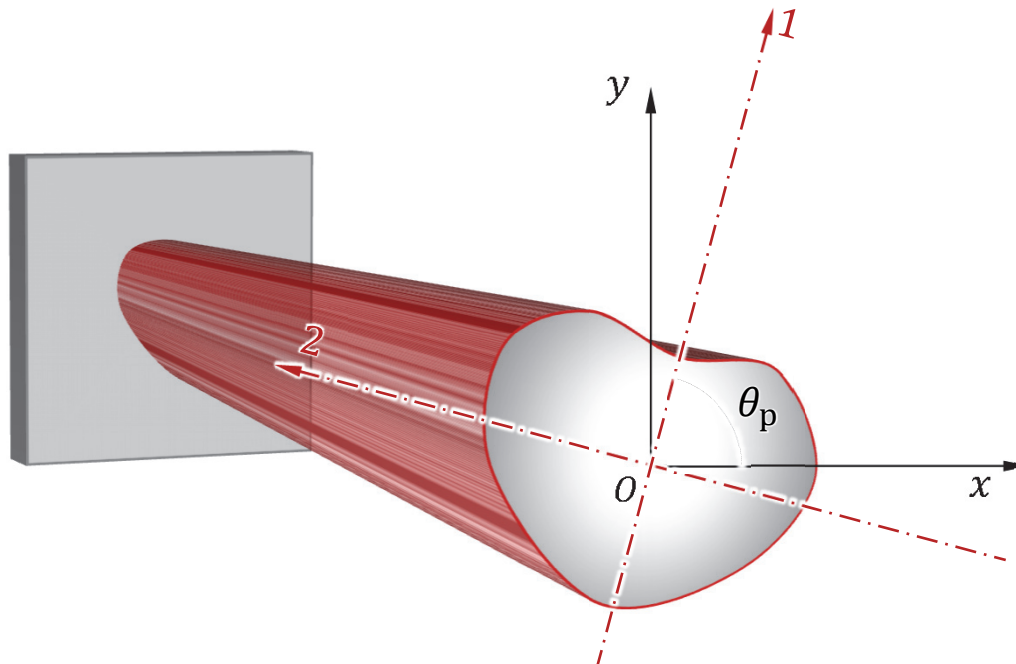
$$I_{XX} = \sum_{i=1}^n \left(I_{xx_i} + y_i^2 A_i \right)$$

$$I_{YY} = \sum_{i=1}^n \left(I_{yy_i} + x_i^2 A_i \right)$$

$$I_{XY} = \sum_{i=1}^n \left(I_{xy_i} + x_i y_i A_i \right)$$

Principal axes

$$\theta_p = \frac{1}{2} \tan^{-1} \left[\frac{2I_{XY}}{I_{YY} - I_{XX}} \right]$$



Parallel axes theorem

$$I_{XX} = \sum_{i=1}^n \left(I_{xx_i} + y_i^2 A_i \right)$$

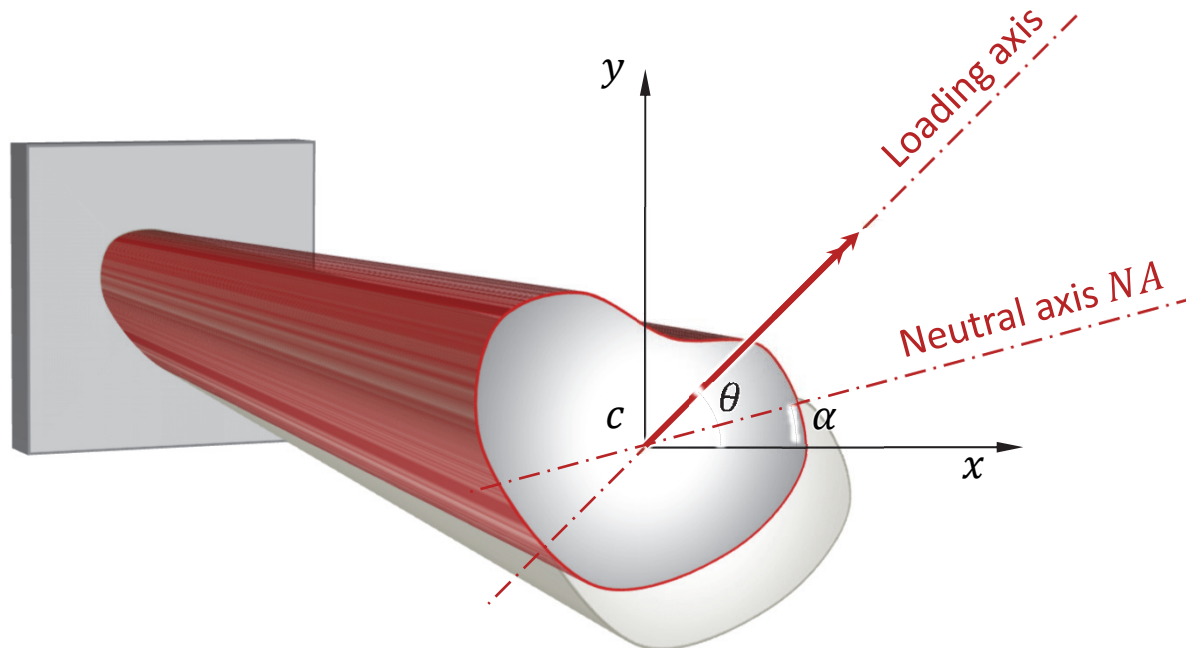
$$I_{YY} = \sum_{i=1}^n \left(I_{yy_i} + x_i^2 A_i \right)$$

$$I_{XY} = \sum_{i=1}^n \left(I_{xy_i} + x_i y_i A_i \right)$$

Neutral axes

$$\tan \alpha = \frac{Y_{NA}}{X_{NA}} = \left[\frac{M_X I_{XY} - M_Y I_{XX}}{M_X I_{YY} - M_Y I_{XY}} \right]$$

Applied moment M



Coordinates,

$$\begin{Bmatrix} X_p \\ Y_p \end{Bmatrix} = \begin{bmatrix} \cos \theta_p & \sin \theta_p \\ -\sin \theta_p & \cos \theta_p \end{bmatrix} \begin{Bmatrix} X \\ Y \end{Bmatrix}$$

Properties,

$$\begin{Bmatrix} I_{11} \\ I_{22} \\ I_{12} \end{Bmatrix} = \begin{bmatrix} \cos^2 \theta_p & \sin^2 \theta_p & -2 \sin \theta_p \cos \theta_p \\ \sin^2 \theta_p & \cos^2 \theta_p & 2 \sin \theta_p \cos \theta_p \\ \sin \theta_p \cos \theta_p & -\sin \theta_p \cos \theta_p & \cos^2 \theta_p - \sin^2 \theta_p \end{bmatrix} \begin{Bmatrix} I_{XX} \\ I_{YY} \\ I_{XY} \end{Bmatrix}$$

Alternatively,

$$I_{ii} = \frac{1}{2} (I_{xx} + I_{yy}) + \frac{1}{2} (I_{xx} - I_{yy}) \cos 2\theta - I_{xy} \sin 2\theta,$$

$$I_{jj} = \frac{1}{2} (I_{xx} + I_{yy}) - \frac{1}{2} (I_{xx} - I_{yy}) \cos 2\theta + I_{xy} \sin 2\theta,$$

$$I_{ij} = \frac{1}{2} (I_{xx} - I_{yy}) \sin 2\theta + I_{xy} \cos 2\theta.$$