

## Example 3.4

Calculate the shear stress distribution in the walls of the three-cell wing section shown in Figure 4, when it is subjected to an anticlockwise torque of 11.3 kN m.

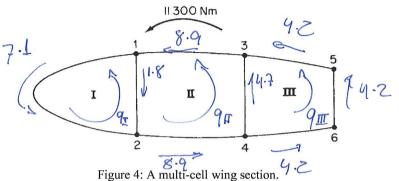


Table 4: Wall properties for the wing section.

Outer segment (leading edge) Inner segment (vertical web)

	Wall	Length (mm)	Hickness (m	m) (ti)eff	G (N/mm <sup>2</sup> )	Cell area (mm <sup>2</sup> )
2)	12°	1650	1.22	1.07	24 200	$A_{\rm I} = 258000$
)	12 <sup>i</sup>	508	2.03	2.03	27600 = Great	$A_{\rm II} = 355000$
	13, 24	775	1.22	( · O r	24 200	$A_{\rm III} = 161000$
	34	380	1.63	1.63	27600 = Gree	
	35, 46	508	0.92	0.69	20 700	
	56	254	0.92	0.69	20 700	

bi

Cell (I) 
$$d\Theta_{\pm} = \frac{9I\left(\frac{1650 + 508}{1.07} + \frac{508}{2.03}\right) - 9I\left(\frac{508}{2.03}\right)}{Z\left(258000 \text{ mm}^2\right)\left(27600 \text{ N}\right)}$$

$$(1.424 \times 10^{\circ}) \frac{d\Theta_{I}}{dz} = 17929_{I} - 2509_{I}$$

1547

250 45F

233 736

368

$$coll (II) = 9I \left[ \frac{508}{2.03} + \left( \frac{775}{1.07} \right) \times 2 + \frac{380}{1.63} \right]$$

$$Z \left( 335.000 \right) \left( 27600 \right)$$

$$\left(1.960 \times 10^{10}\right) \frac{d\theta}{dz} = -250 q_{\text{I}} + 1931 q_{\text{I}} - 233 q_{\text{II}}$$

Using tabulated values of 
$$\delta_i = \frac{bi}{(ti)_{eff}}$$

$$\frac{d\theta}{dz} = \frac{1}{736 + 736 + 368} - 9\pi(733)$$

$$\frac{d\theta}{dz} = \frac{1}{736 + 368} - 9\pi(733)$$

$$(0.889 \times 10^{\circ}) d\theta = -2339\pi + 20739\pi$$

$$T = \int q r ds = \sum_{j} Z A_{j} q_{j}$$

$$11-3 \times 10^{6} \ \text{N} \text{mm} = Z \left( 258000 \ q_{\text{I}} + 355000 \ q_{\text{I}} + 161000 \ q_{\text{II}} \right)$$

$$5.65 \times 10^{3} = 258 \ q_{\text{I}} + 335 \ q_{\text{II}} + 161 \ q_{\text{II}}$$

$$9^{\text{T}} = 7089 \text{ N/m}$$
 $9^{\text{T}} = 8880 \text{ N/m}$ 
 $9^{\text{T}} = 4154 \text{ N/m}$ 
 $4^{\text{O}} = 7.36 \times 10^{-4} \text{ rad/m}$