

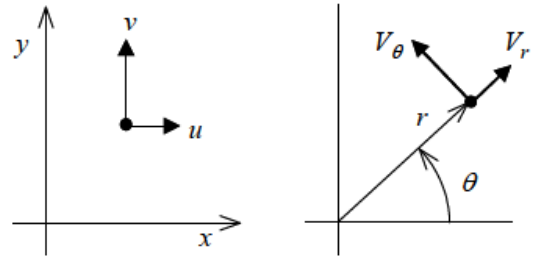
## Useful Equations

*Change between Polar and Cartesian coordinate systems*

$$r = \sqrt{x^2 + y^2}, \quad \theta = \tan^{-1}\left(\frac{y}{x}\right), \quad x = r \cos \theta, \quad y = r \sin \theta,$$

$$u = V_r \cos \theta - V_\theta \sin \theta, \quad v = V_r \sin \theta + V_\theta \cos \theta$$

$$V_r = u \cos \theta + v \sin \theta, \quad V_\theta = -u \sin \theta + v \cos \theta$$



### 2D Potential Flow

Velocity components in two-dimensional irrotational flow, in terms of the stream function and potential are given by

$V_r = \frac{1}{r} \frac{\partial \psi}{\partial \theta} \quad V_\theta = -\frac{\partial \psi}{\partial r}$	$u = \frac{\partial \psi}{\partial y} \quad v = -\frac{\partial \psi}{\partial x}$
$V_r = \frac{\partial \phi}{\partial r} \quad V_\theta = \frac{1}{r} \frac{\partial \phi}{\partial \theta}$	$u = \frac{\partial \phi}{\partial x} \quad v = \frac{\partial \phi}{\partial y}$
<p>⏟ polar coordinates</p>	<p>⏟ Cartesian coordinates</p>

The stream function & velocity potential in Polar coordinates and the velocity distribution for:

i) A uniform flow  $U_\infty$  parallel to the  $x$  axis:

$$\psi = U_\infty r \sin \theta, \quad \phi = U_\infty r \cos \theta, \quad V_r = U_\infty \cos \theta, \quad V_\theta = -U_\infty \sin \theta, \quad u = U_\infty, \quad v = 0$$

ii) A source, of strength  $\Lambda$ , at the origin:

$$\psi = \frac{+\Lambda \theta}{2\pi}, \quad \phi = \frac{+\Lambda}{2\pi} \ln r, \quad V_r = \frac{+\Lambda}{2\pi r}, \quad V_\theta = 0, \quad u = \frac{+\Lambda}{2\pi} \frac{x}{(x^2 + y^2)}, \quad v = \frac{+\Lambda}{2\pi} \frac{y}{(x^2 + y^2)}$$

iii) A doublet, of strength  $\kappa$ , at the origin:

$$\psi = \frac{-\kappa \sin \theta}{2\pi r}, \quad \phi = \frac{+\kappa \cos \theta}{2\pi r}, \quad V_r = \frac{-\kappa}{2\pi r^2} \cos \theta, \quad V_\theta = \frac{-\kappa}{2\pi r^2} \sin \theta,$$

$$u = \frac{-\kappa}{2\pi} \frac{(x^2 - y^2)}{(x^2 + y^2)^2}, \quad v = \frac{-\kappa}{2\pi} \frac{2xy}{(x^2 + y^2)^2}$$

iv) A vortex, of strength  $\Gamma$ , at the origin:

$$\psi = \frac{+\Gamma}{2\pi} \ln r, \quad \phi = \frac{-\Gamma}{2\pi} \theta, \quad V_r = 0, \quad V_\theta = \frac{-\Gamma}{2\pi r}, \quad u = \frac{+\Gamma}{2\pi} \frac{y}{(x^2 + y^2)}, \quad v = \frac{-\Gamma}{2\pi} \frac{x}{(x^2 + y^2)}$$

Using Bernoulli's equation between 1 and 2 for irrotational flow gives:  $p_1 - p_2 = \frac{1}{2} \rho (V_2^2 - V_1^2)$

If point 1 is in the freestream  $p_1 - p_\infty = \frac{1}{2} \rho (U_\infty^2 - V_1^2) \rightarrow c_p = \frac{p - p_\infty}{\frac{1}{2} \rho U_\infty^2} = 1 - \frac{V^2}{U_\infty^2}$

**Not given in exam: memorise**