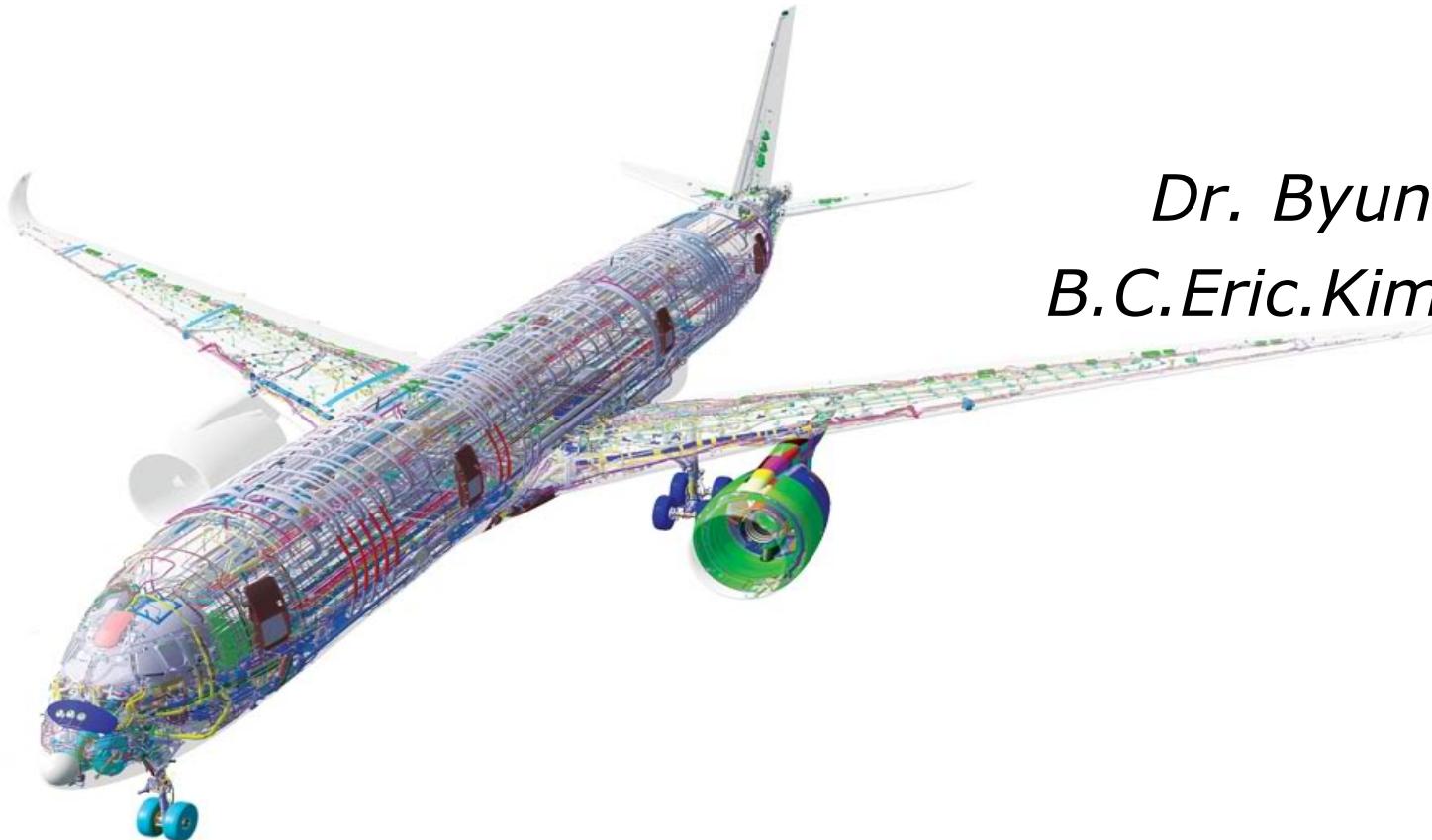


# StM3. Aircraft manufacture

## Part 4



*Dr. ByungChul Eric Kim*  
*B.C.Eric.Kim@bristol.ac.uk*

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# *Joining Part 2*

## *Adhesive joints*

# Joining method 2: Adhesive joining

- Is it really strong enough?



*Is this really amazing?*

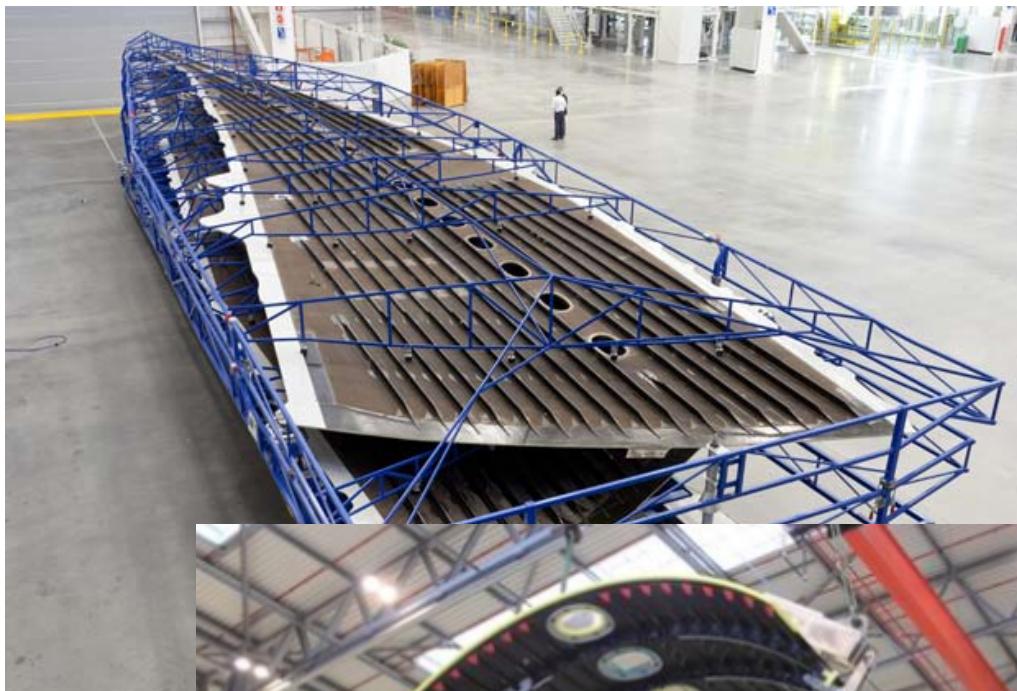
*Tensile stress applied to the adhesive layer:*

$$\begin{aligned}1,200\text{kgf} \times 9.8\text{N/m}^3 / 900\text{mm}^2 \\= 13 \text{ MPa}\end{aligned}$$

*The tensile strength of this epoxy adhesive is about 20 MPa.*

*Not amazing at all in this case,  
Strangely they couldn't use the  
full strength of the epoxy  
adhesive. Why?*

# Adhesive joining in aircraft manufacturing



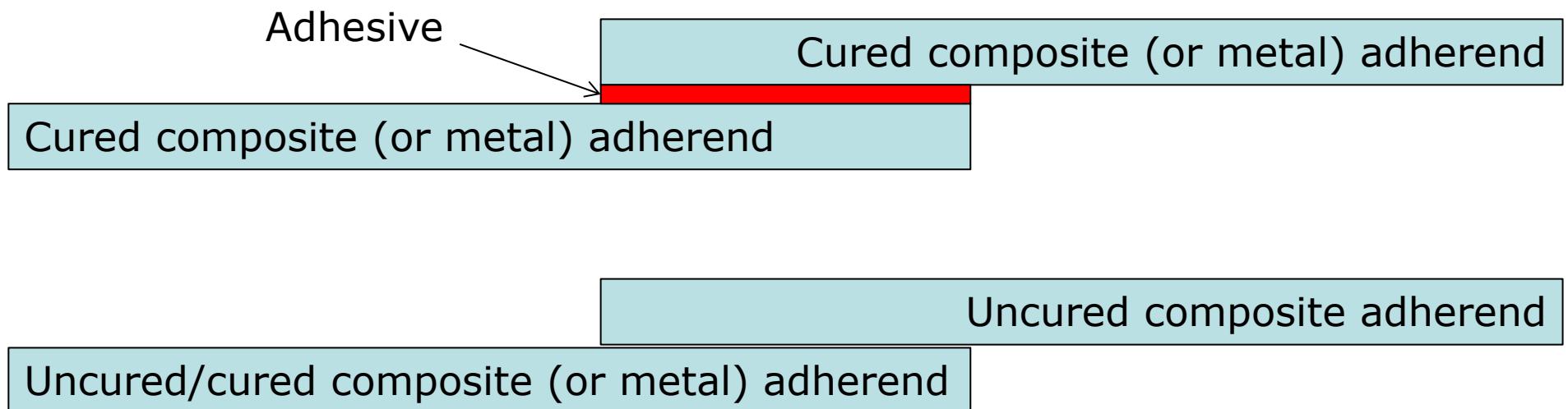
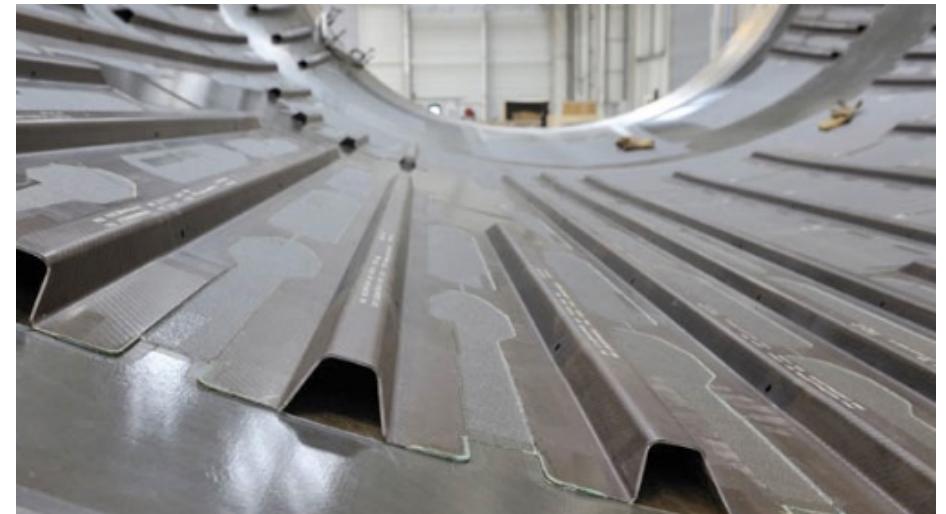
Airbus A350XWB  
Composite wing skin & fuselage

Total bonding length:  
~ 5 km

Significant cost reduction  
by eliminating riveting  
process  
But, more strict quality  
control and inspection  
processes are required.

# Joining method 2: Adhesive joining

- Adhesive bonding
  - Applying polymeric adhesive on cured composites or metals and curing it at high (or room) temperature
  - Co-curing: Uncured composite plies are cured and bonded simultaneously during curing.



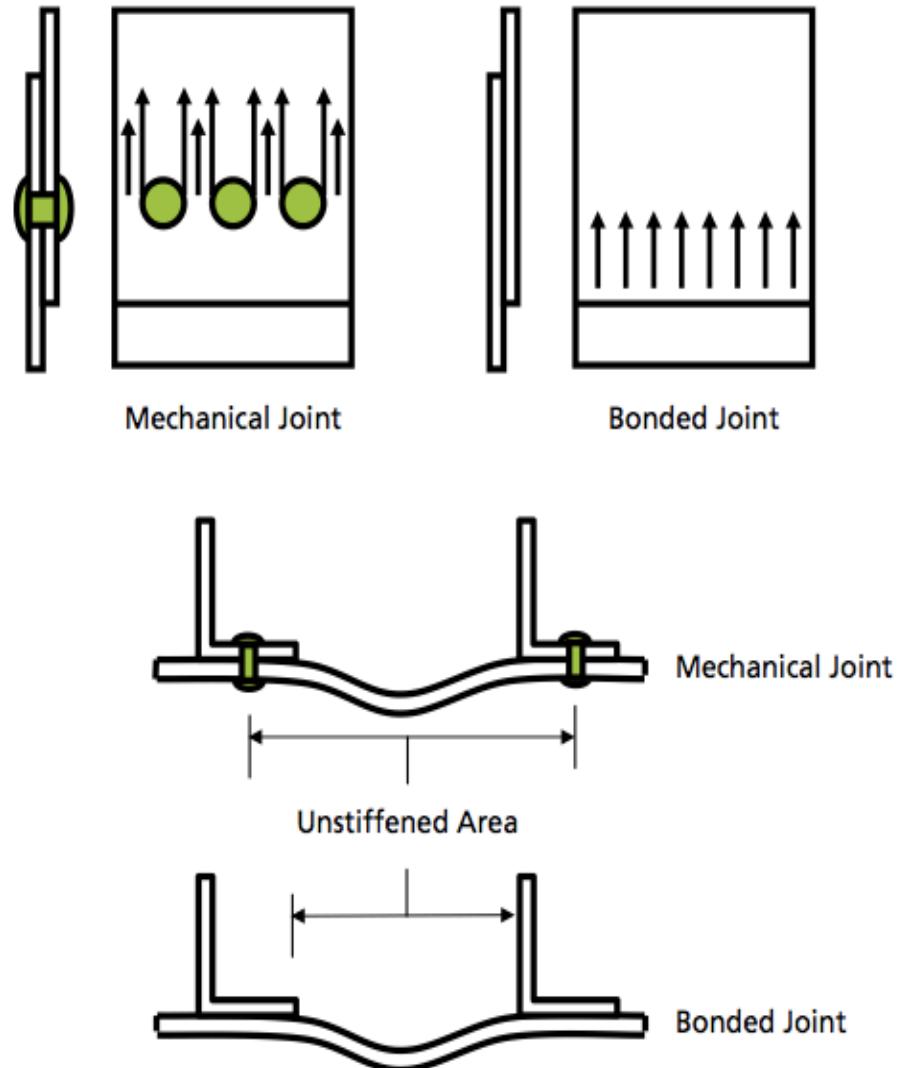
# Joining method 2: Adhesive joining

- Advantages

- More uniform stress distribution
- Reducing unstiffened areas
- Joining a large area at once
- Weight saving
- Smooth aircraft surface
- Electrical insulation (No galvanic corrosion)

- Disadvantages

- Difficult to inspect and repair
- Unable to disassemble
- Low environmental resistance
- Surface pre-treatment required
- Sensitive to peel stress
- Weak in fatigue loading



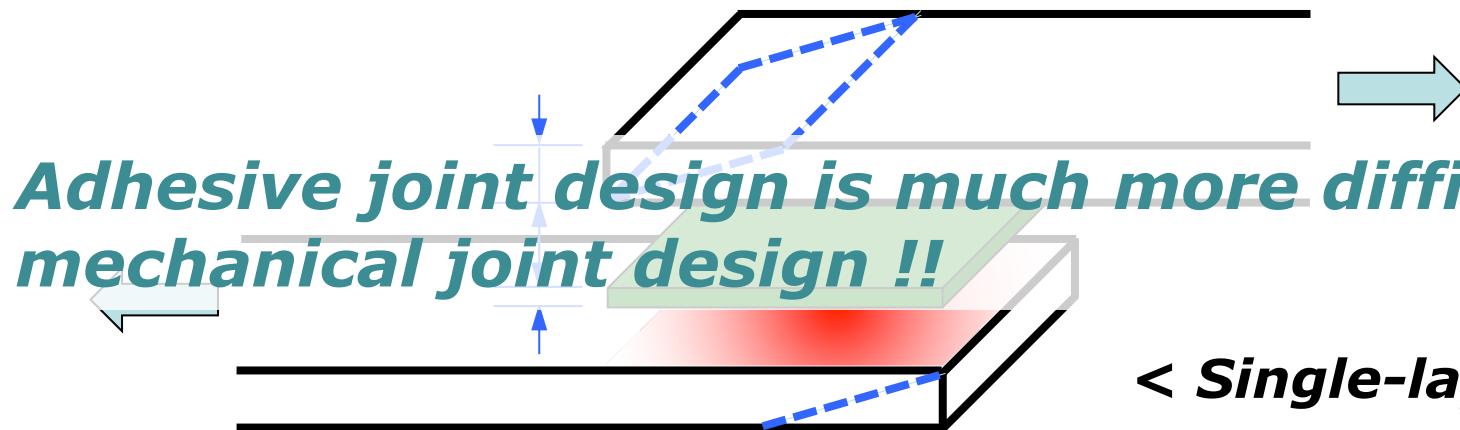
# Adhesive joint design

Given:

- Adherend materials
- Loading conditions

## Joint configuration

- Joint type (single or double lap, scarf, step)
- Overlap length
- Bonding thickness
- Tensile/flexural stiffness of adherends
- Reduction of stress concentration



## Surface treatment

- Surface cleaning
- Suitable method for adherend materials
- Damage of adherend (surface damage, degradation)
- Cost & speed
- Environmental effect, Health & safety

## Type of adhesive

- Curing temp. → Residual stress & joint strength
- Glass transition temp. → Service temp.
- Fracture toughness → Fatigue & Impact
- Viscosity → Application method
- Environmental effect (Humidity, UV)
- Pot life & Cost

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# 1. Adhesive Materials

# Adhesive materials

- Non-linear shear behaviour & temperature dependency

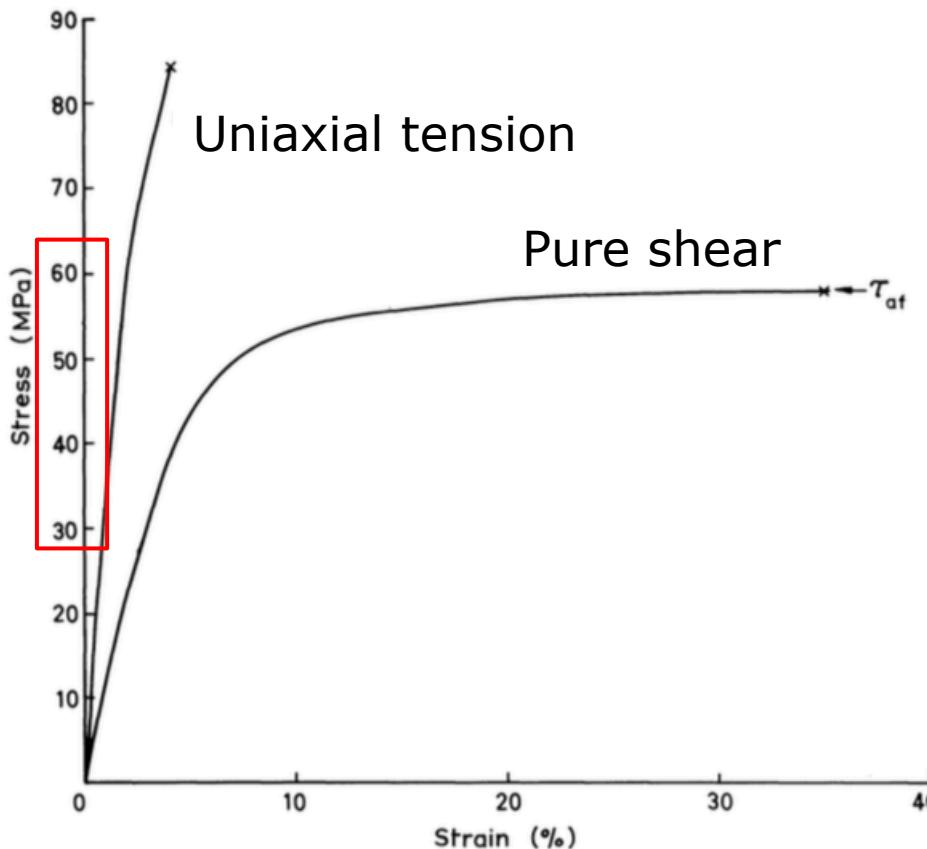


Figure 6.3 Stress versus strain curves for a hot-cured rubber-toughened epoxy adhesive (at 23 °C and an average strain rate of  $7.5 \times 10^{-4} \text{ s}^{-1}$ ) in uniaxial tension and pure shear [8].

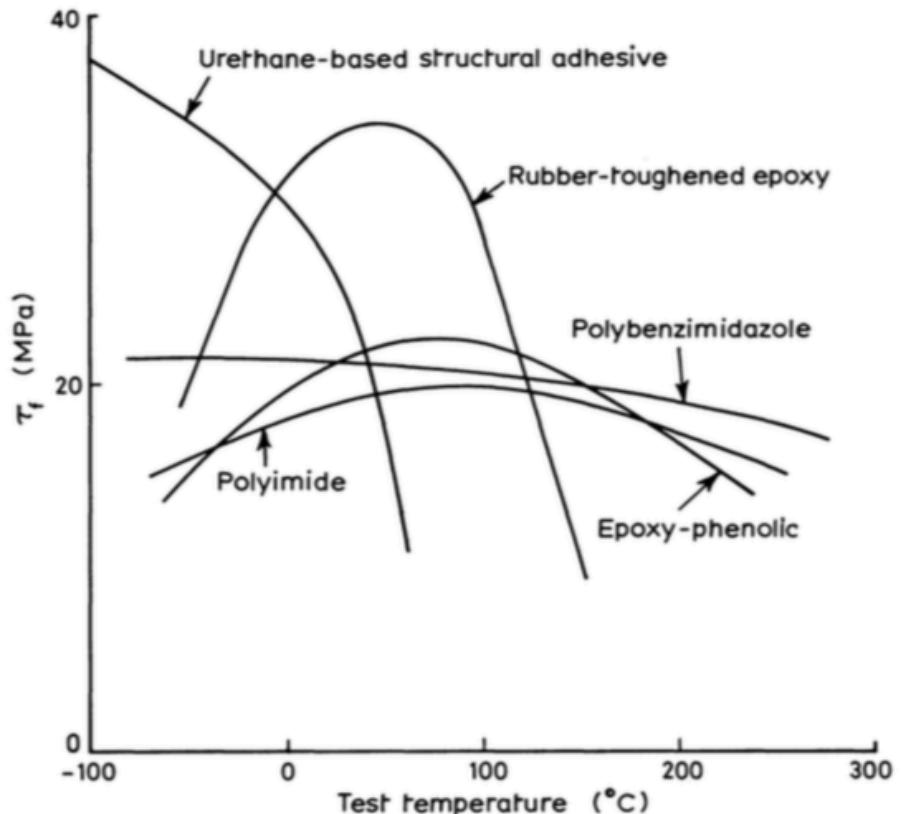


Figure 6.20 Typical strengths of single lap joints loaded in tension using various structural adhesives as a function of test temperature [76,77].

# Adhesive materials

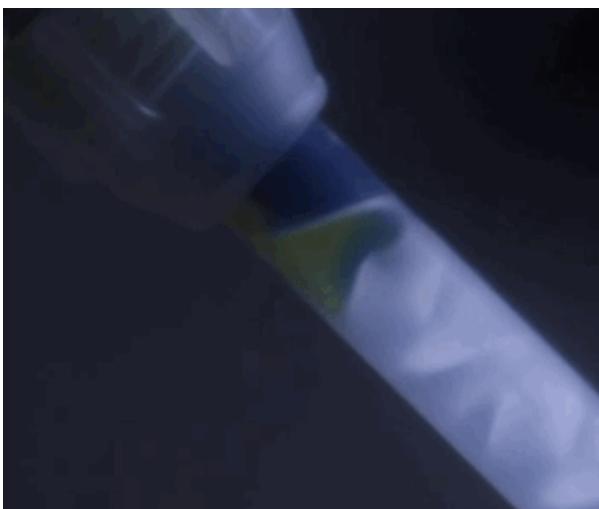
**Table 9.2 Advantages and limitations of the five most widely used chemically reactive structural adhesives**

<b>Advantages</b>				
Epoxy	Polyurethane	Modified acrylic	Cyanoacrylate	Anaerobic
High strength	Varying cure times	Good flexibility	Rapid room-temperature cure	Rapid room-temperature cure
Good solvent resistance	Tough	Good peel and shear strengths	One component	Good solvent resistance
Good gap-filling capabilities	Excellent flexibility even at low temperatures	No mixing required	High tensile strengths	Good elevated-temperature resistance
Good elevated-temperature resistance	One or two components, room- or elevated-temperature cure	Will bond dirty (oily) surfaces	Long pot life	No mixing
Wide range of formulations	Moderate cost	Room-temperature cure	Good adhesion to metal	Indefinite pot life
Relatively low cost		Moderate cost	Dispenses easily from package	Nontoxic
				High strength on some substrates
				Moderate cost
<b>Limitations</b>				
Epoxy	Polyurethane	Modified acrylic	Cyanoacrylate	Anaerobic
Exothermic reaction	Both uncured and cured are moisture sensitive	Low hot-temperature strength	High cost	Not recommended for permeable surfaces
Exact proportions needed for optimum properties	Poor elevated-temperature resistance	Slower cure than with anaerobics or cyanoacrylates	Poor durability on some surfaces	Will not cure in air as a wet fillet
Two-component formulations require exact measuring and mixing	May revert with heat and moisture	Toxic	Limited solvent resistance	Limited gap cure
One-component formulations often require refrigerated storage and an elevated-temperature cure	Short pot life	Flammable	Limited elevated-temperature resistance	
Short pot life (more waste)	Special mixing and dispensing equipment required	Odor	Bonds skin	
		Limited open time		
		Dispensing equipment required		

Ref. Adhesives, Engineered Materials Handbook, ASM international, 1995.

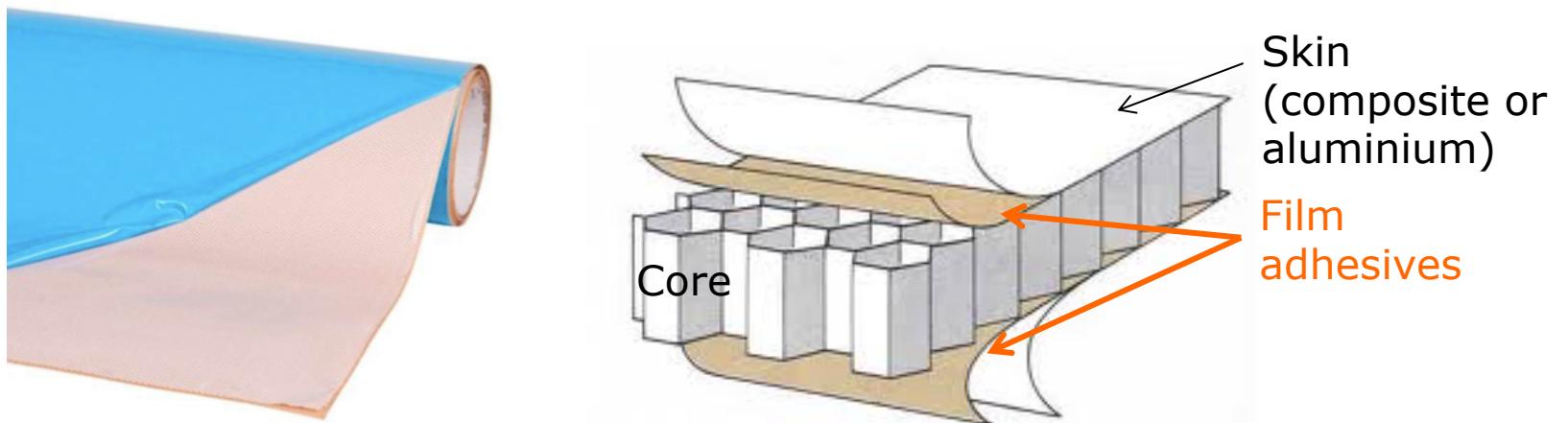
# Epoxy adhesives

- Liquid type adhesives
  - Toughened epoxy is the most popular.
  - Application: Resin + Hardener > Mixing > Degassing > Pasting > Pressing with adherends > Curing
  - Good penetration, but difficult to control the bonding thickness
  - Difficult to apply
  - Wide curing temperature range, various grades
  - Considerations: Viscosity, pot-life, service temperature.

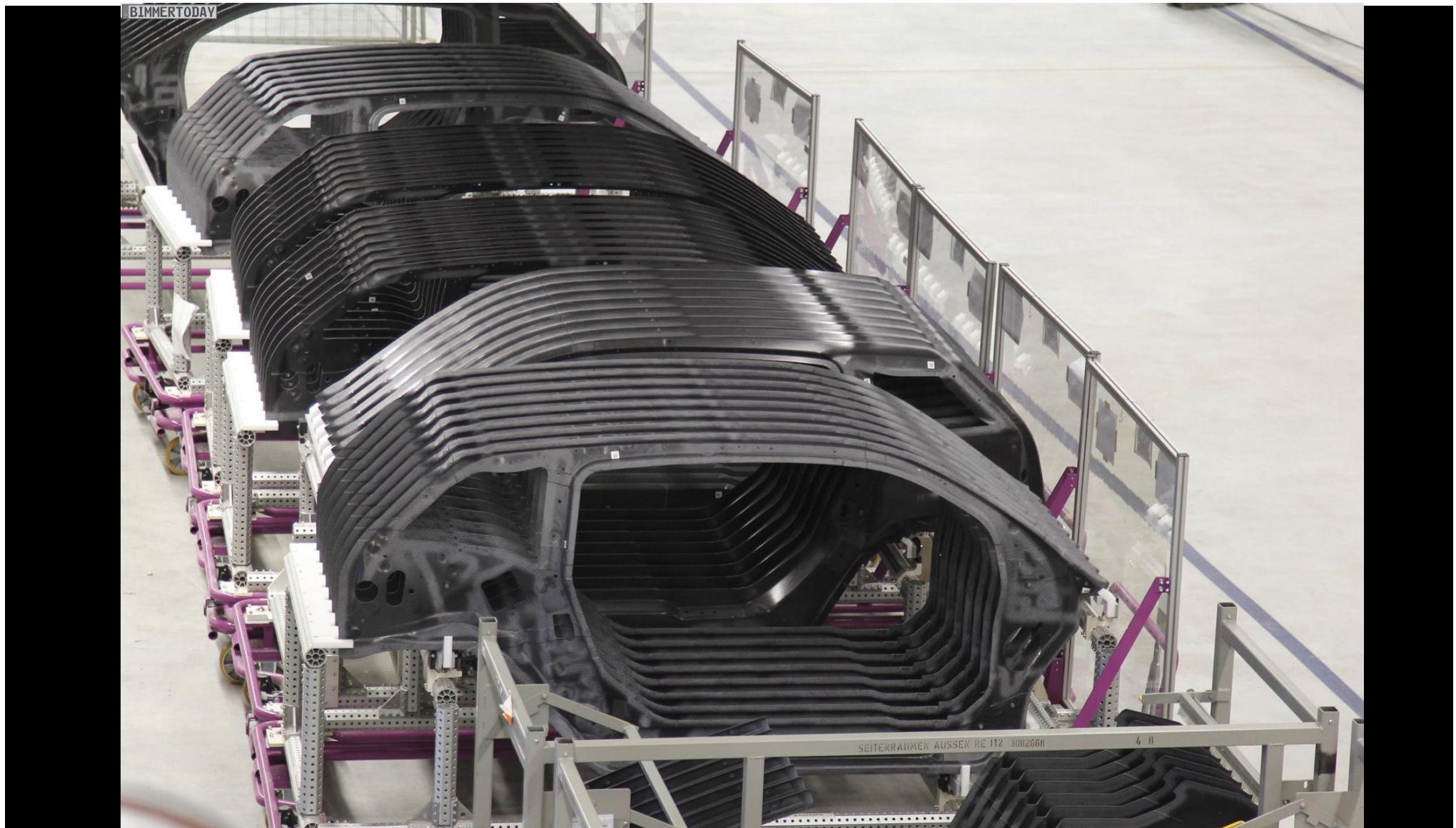


# Epoxy adhesives

- Film (Hot-melt) adhesives
  - B-stage epoxy resin film (partially cured, so sticky)
  - Application: Cutting > Attaching to adherends > Pressing > Curing at high-temperature
  - Easy to handle and clean
  - Easy to control the bonding thickness
  - But, expensive. High-temperature curing only. Short shelf-life
  - Limited adhesive types available, low temperature storage
  - Considerations: Tackiness, areal weight, shelf-life



# Adhesive bonding in automotive industry



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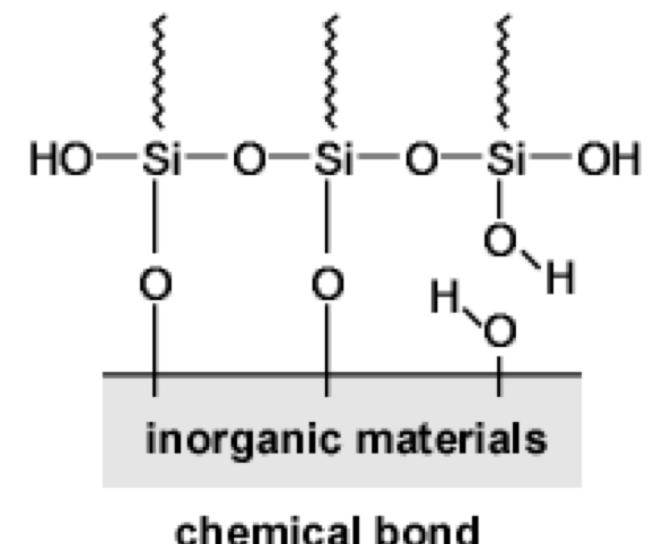
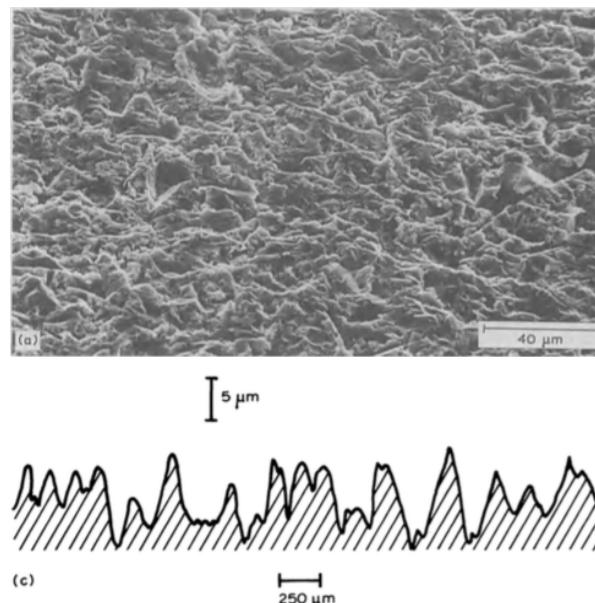
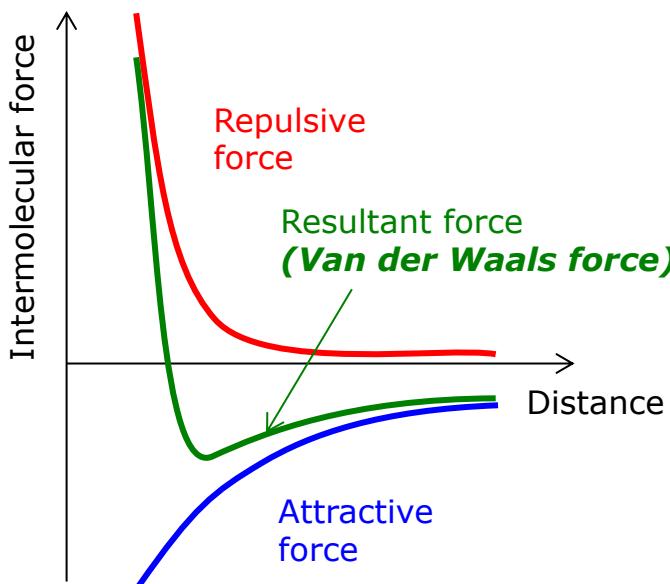
# 2. *Surface Treatment*

# Mechanism of adhesion

Adhesion by

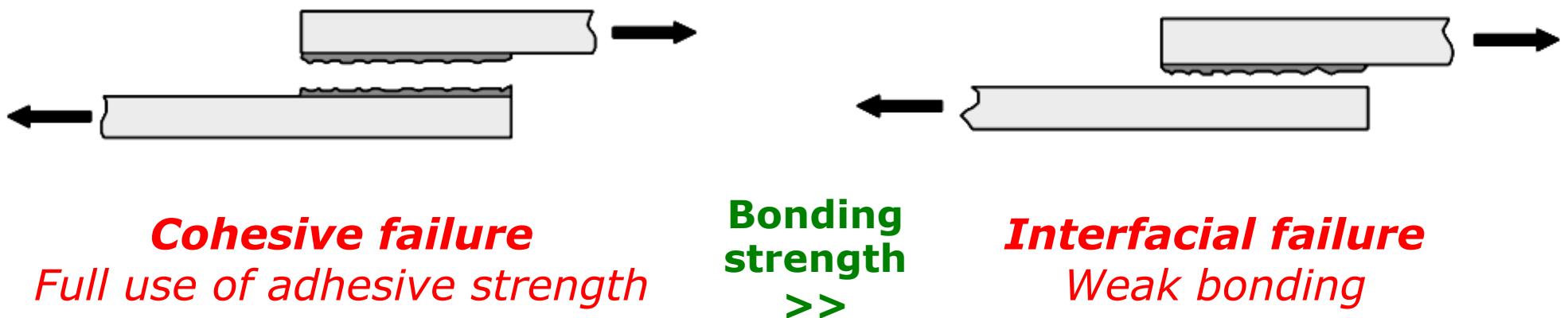
- Physical absorption
- Mechanical interlocking
- Chemical bonding
- Diffusion
- Electrostatic

***Surface treatment is crucial to make the adhesion mechanism work properly.***



# Surface treatment

- Purpose of surface treatment
  - Removing surface contamination (e.g. oil contamination during machining or rolling)
  - Surface activation (increasing surface free energy & wettability)
  - Creating surface roughness
  - Surface chemistry modification
- Impact on joint failure mode



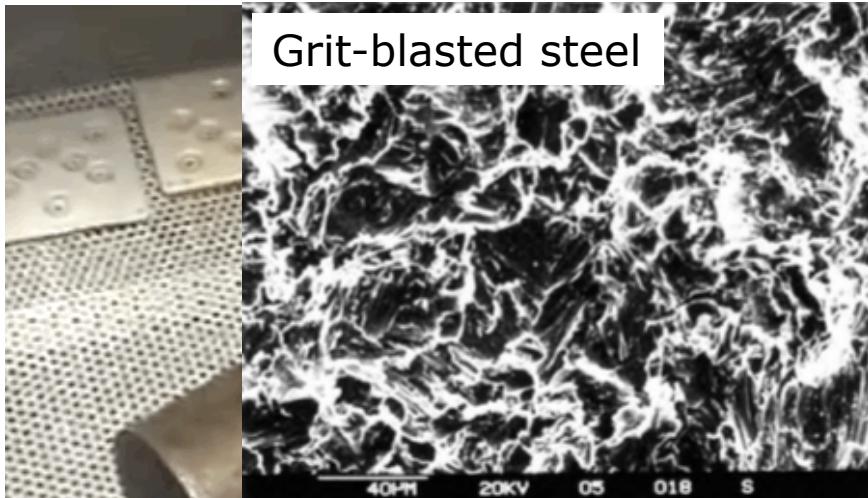
# Surface treatment methods

- Types
  - Mechanical abrading (e.g. grid-blasting, grinding, manual abrading)
  - Chemical etching (e.g. phosphoric acid anodising)
  - Plasma treatment (Corona)
  - UV treatment
  - Flame treatment
  - Primer coating (e.g. silane coupling agent)
- *Thermoplastic adherends cannot be bonded with thermoset adhesives without changing surface chemistry (surface functionalisation) due to their low surface energy. Even if bonded, the interfacial failure causes a very low joint strength.*
  - e.g. Teflon (PTFE), PEEK: Very low surface energy

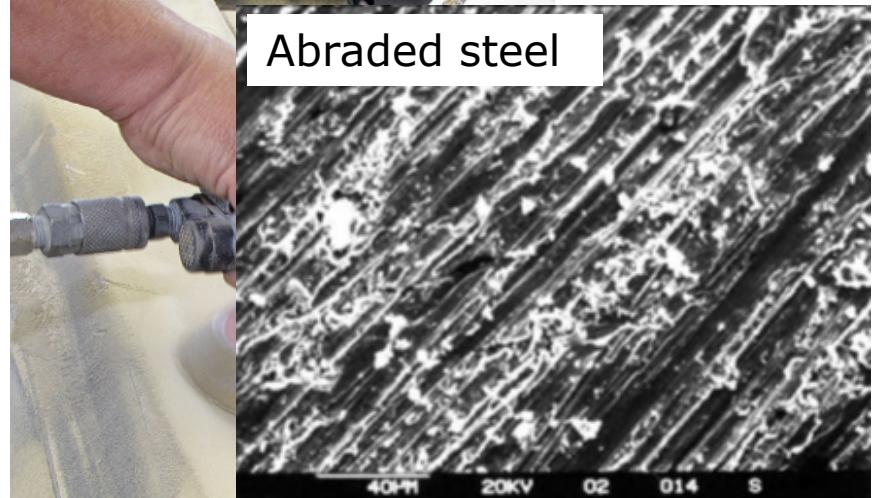


# Surface treatment methods

- Grit-blasting



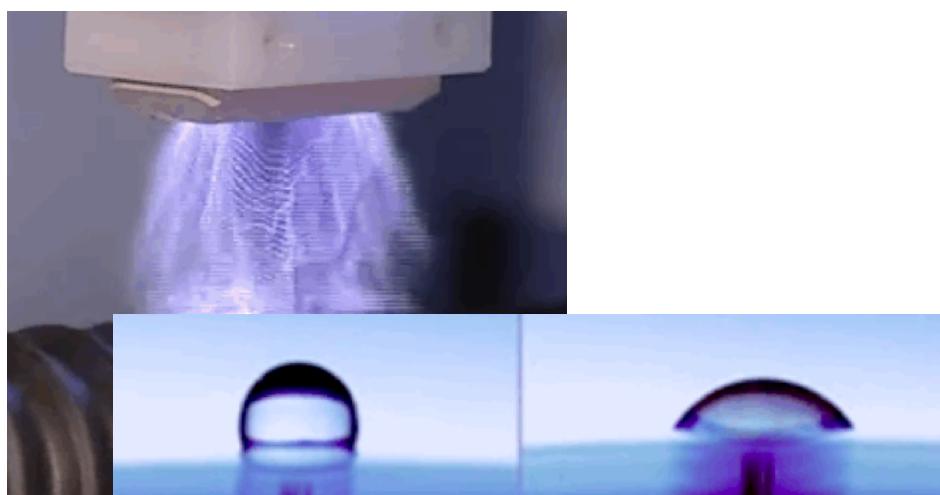
- Manual abrading



- Chemical etching



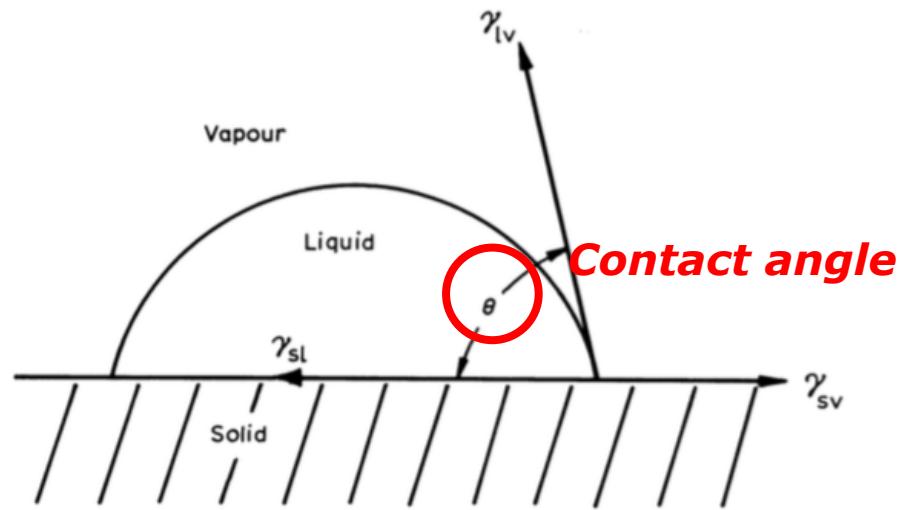
- Plasma treatment



# Surface free energy (Physical Absorption)

- **Wettability**

- Wetting: intimate molecular contact between two substances



$$\gamma_{sv} = \gamma_{sl} + \gamma_{lv} \cos\theta \quad (\text{Young's equation})$$

$\gamma_{sv}$  Surface free energy of the solid substrate

$\gamma_{sl}$  Interfacial free energy b/w solid and liquid

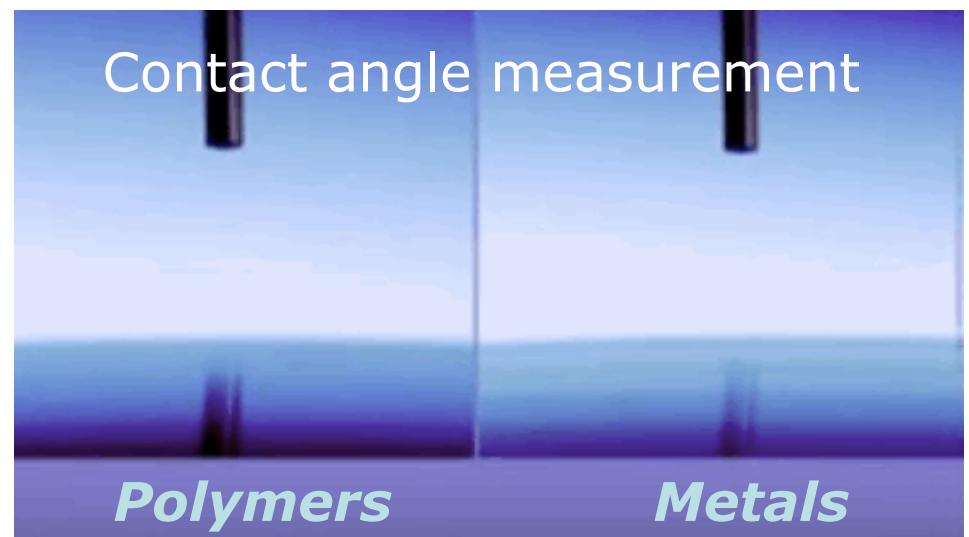
$\gamma_{lv}$  Surface free energy of the liquid  
(= Surface tension)

**Spontaneous wetting condition**

$$\gamma_{sv} \geq \gamma_{sl} + \gamma_{lv}$$

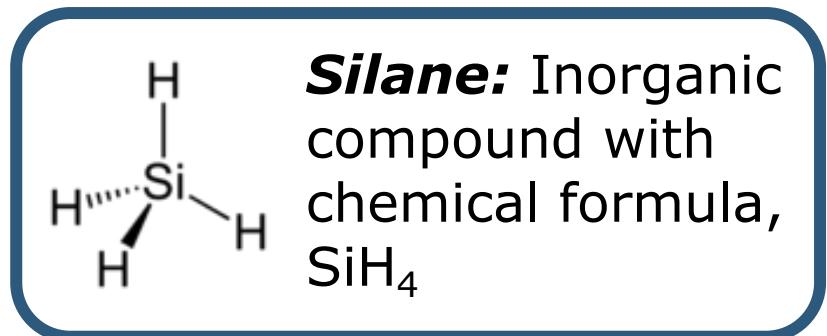
**Equilibrium spreading coefficient**

$$S = \gamma_{sv} - \gamma_{sl} - \gamma_{lv}$$

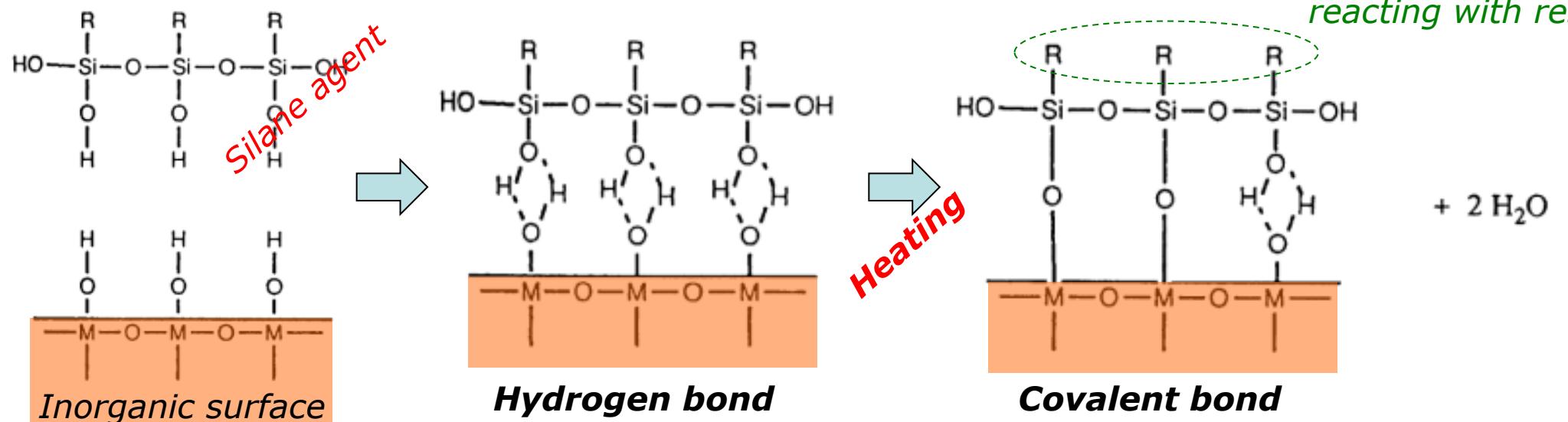


# Silane coupling agent (Chemical bonding)

- Silane coupling agents have the ability to form a durable bond between organic and inorganic materials.



- The most popular application is the surface treatment of glass fibres for glass composites. (Organic polymer resin & inorganic oxide fibre)



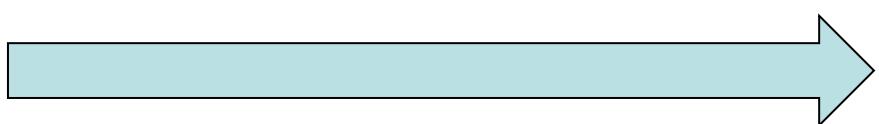
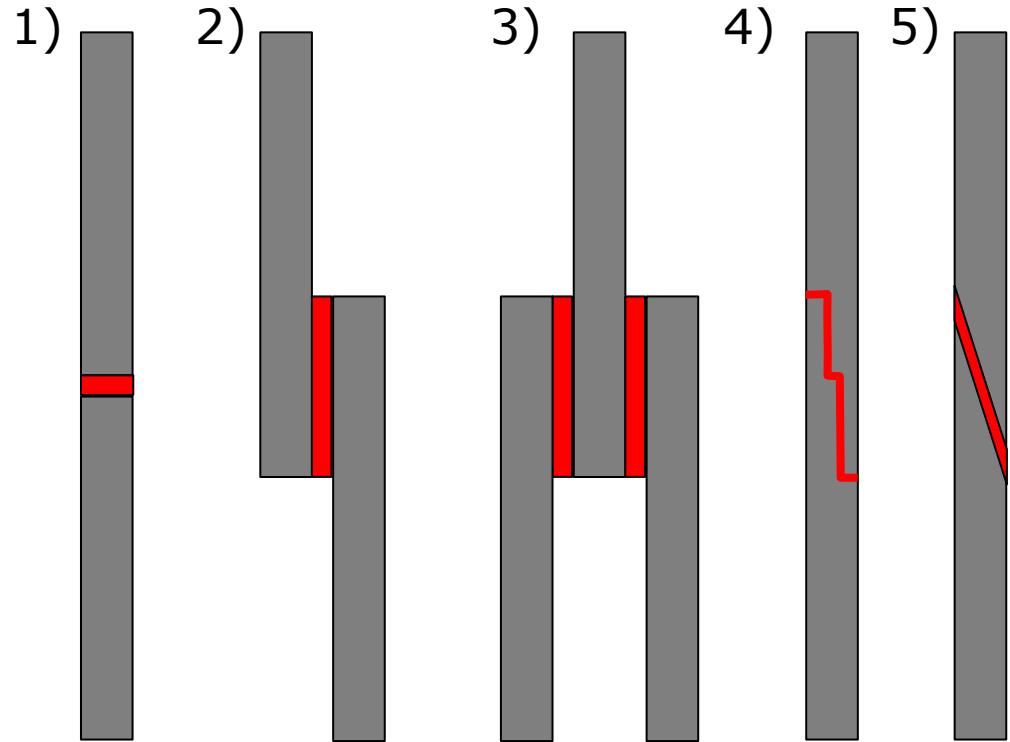
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# 3. Joint Configuration

# Adhesive joint design

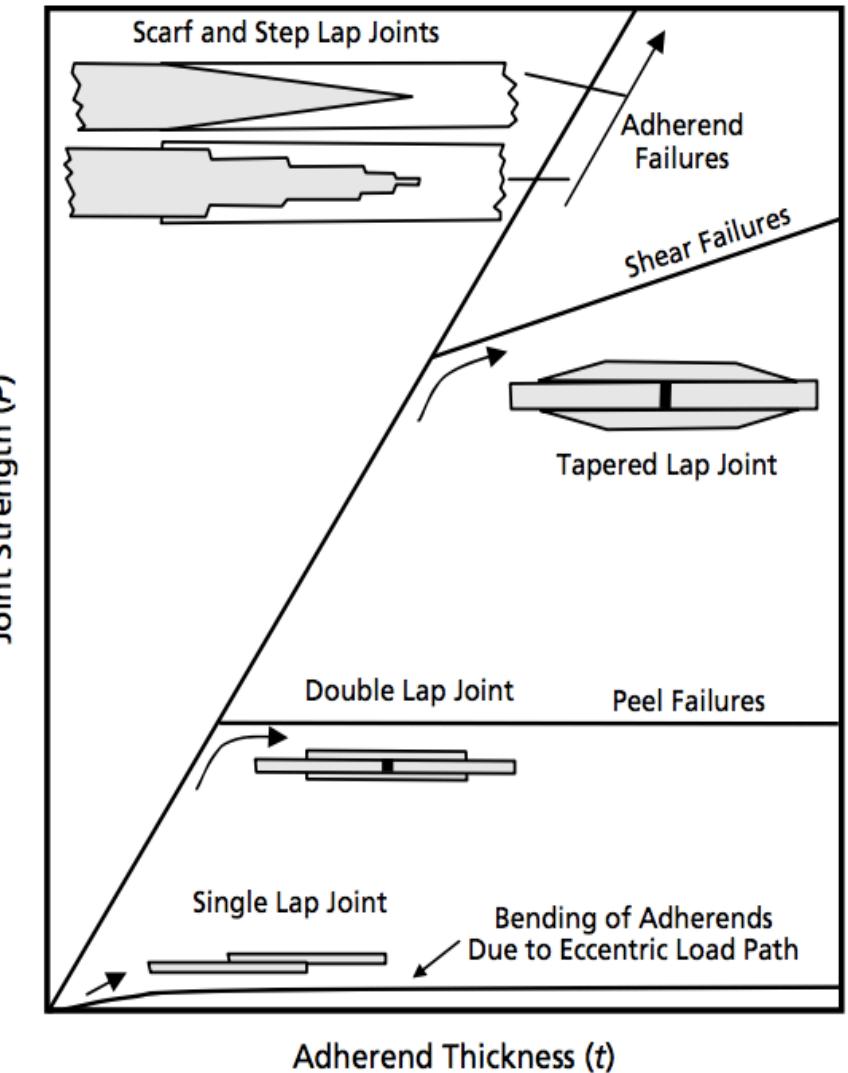
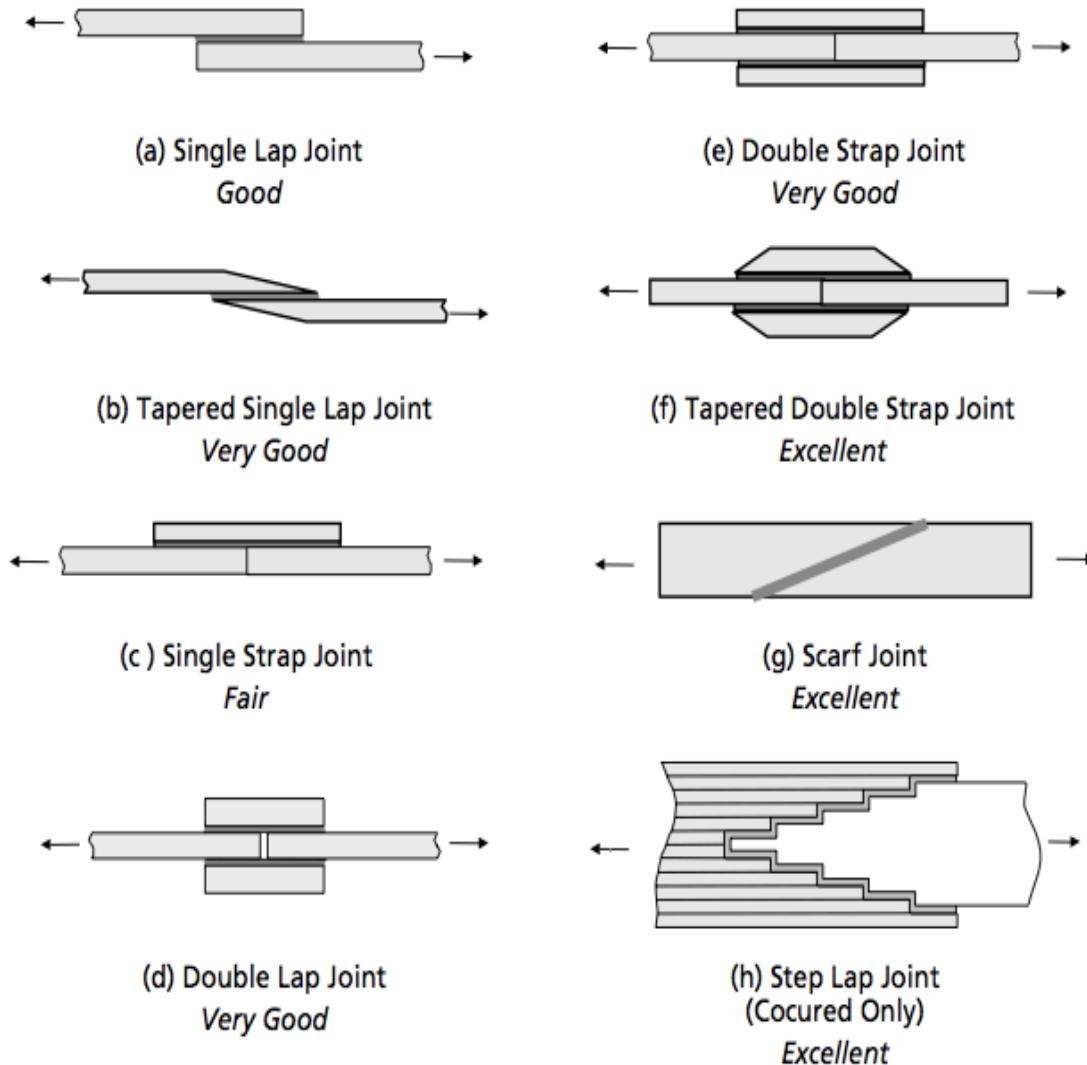
## ***Joint configuration***

- 1) Butt joint
- 2) Single lap joint
- 3) Double lap joint
- 4) Step joint
- 5) Scarf joint
- 6) Tubular joint

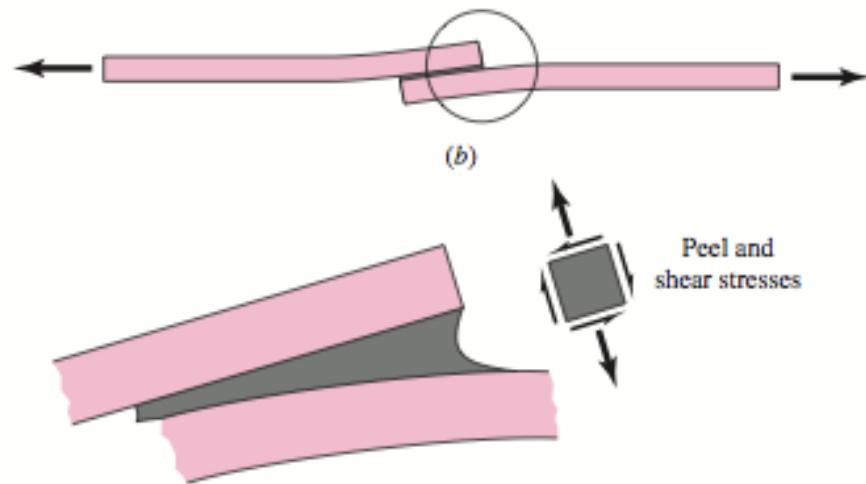


Higher strength, but  
more expensive to make

# Adhesive joint design

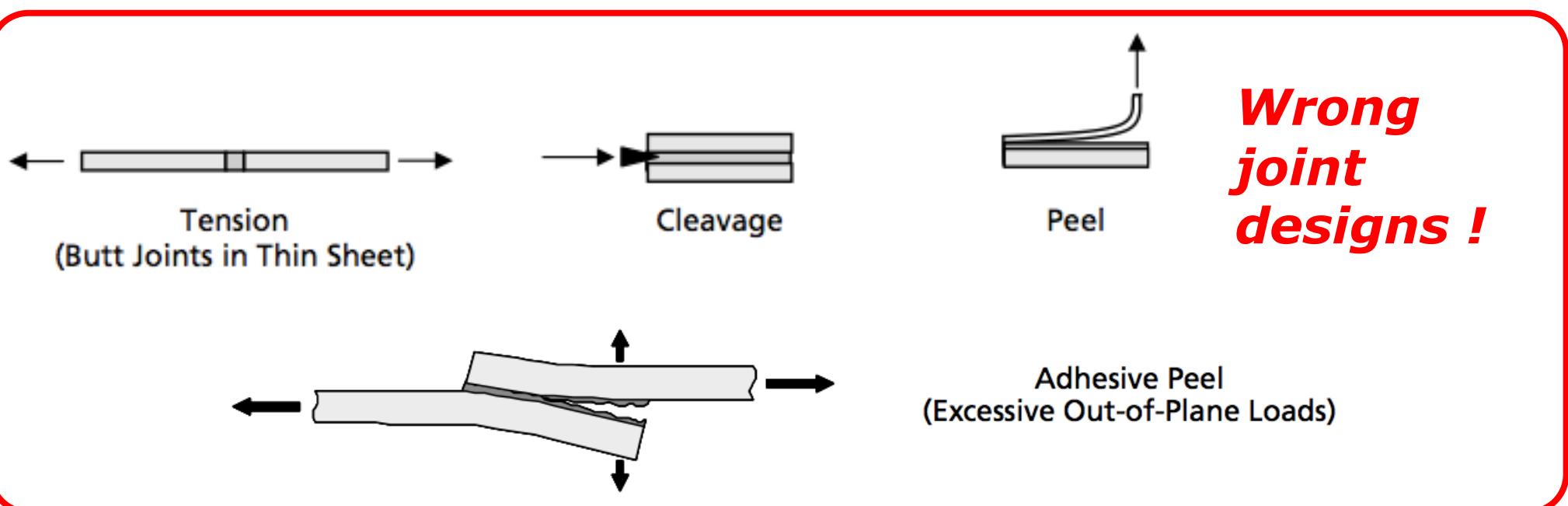


# Adhesive joint design



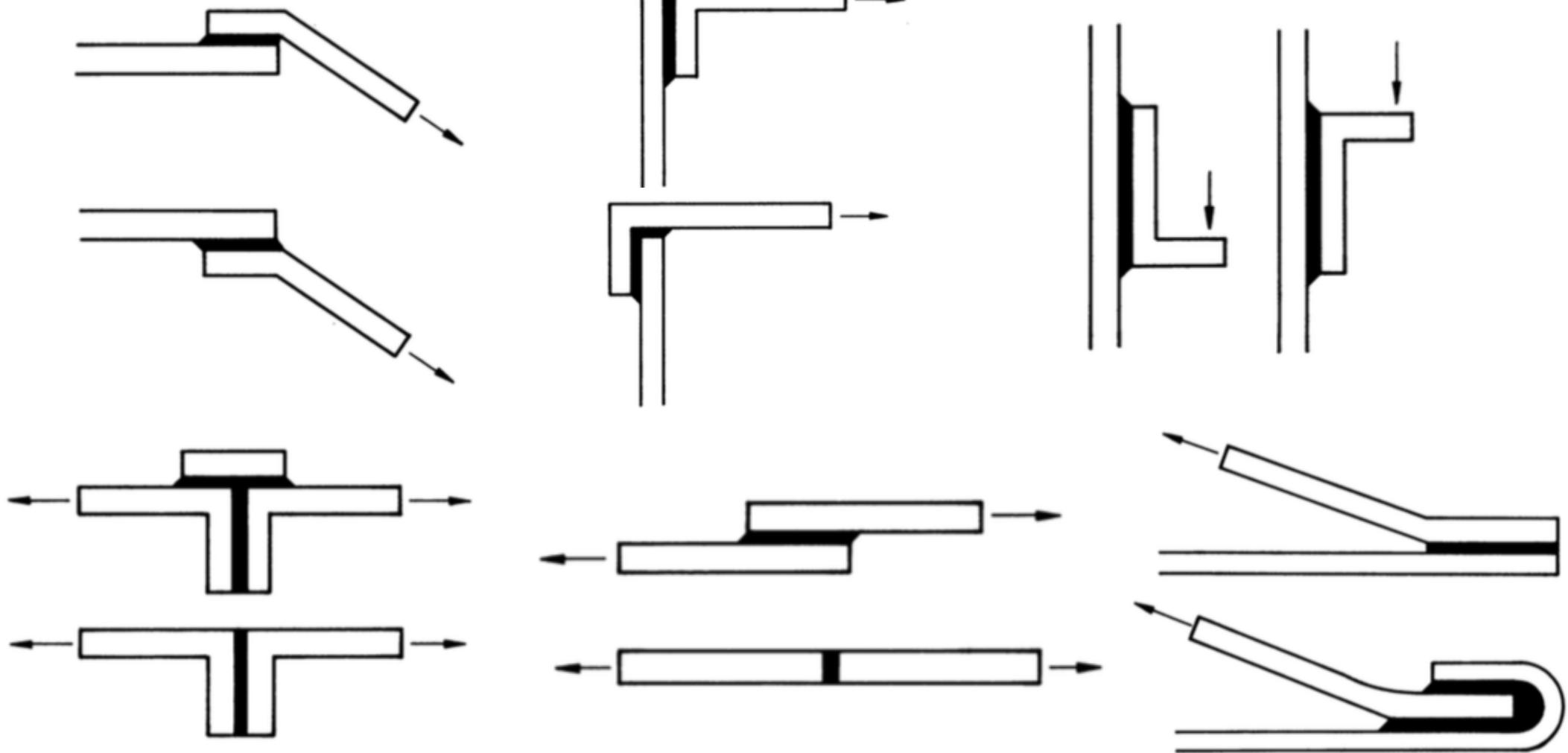
**Adhesive is very strong to shear loadings, but very weak to peel stress.**

How to minimise peel stress on the edge is critical.



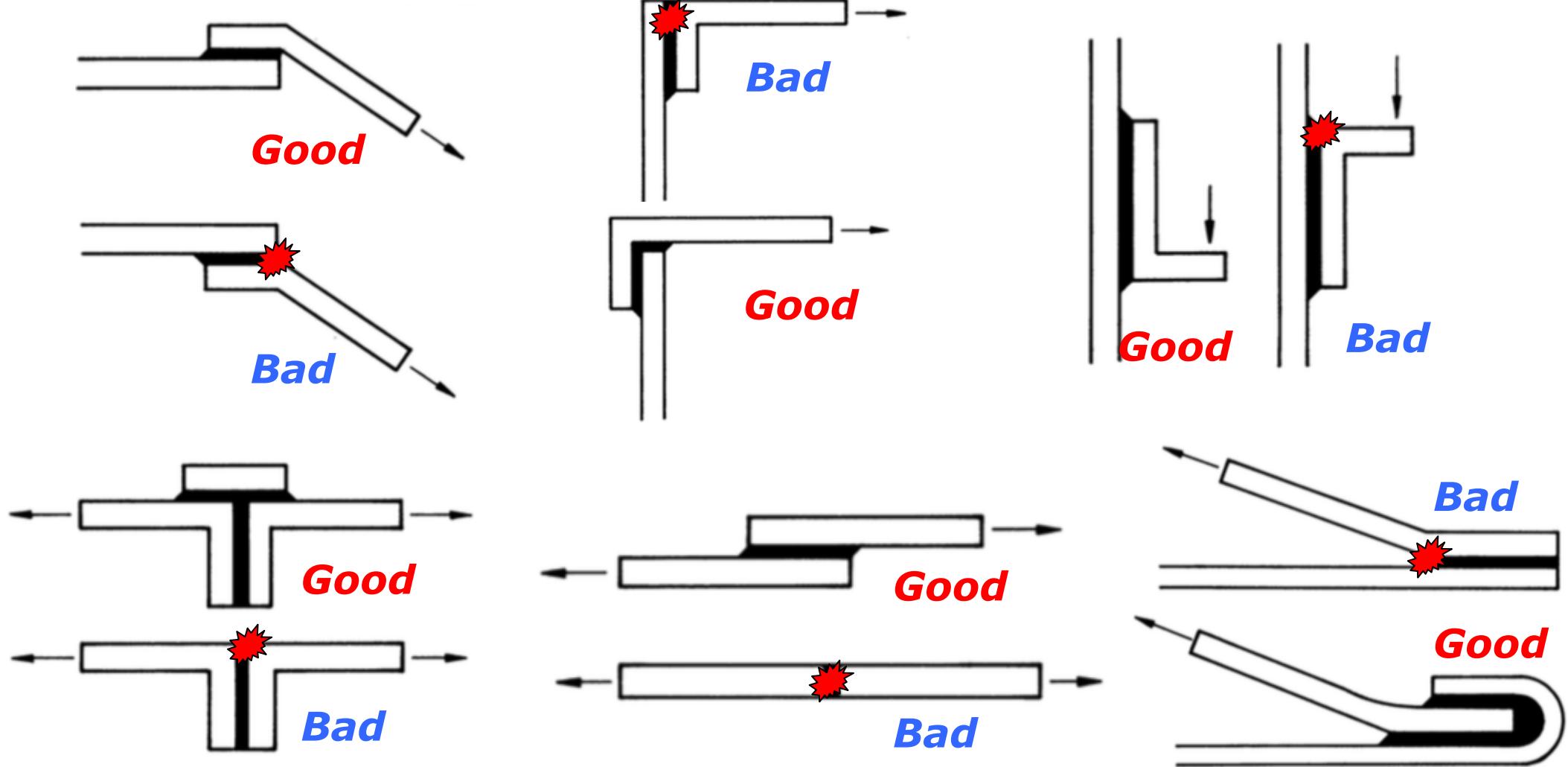
# Quizzes

- Which one is a good adhesive joint design?



# Quizzes

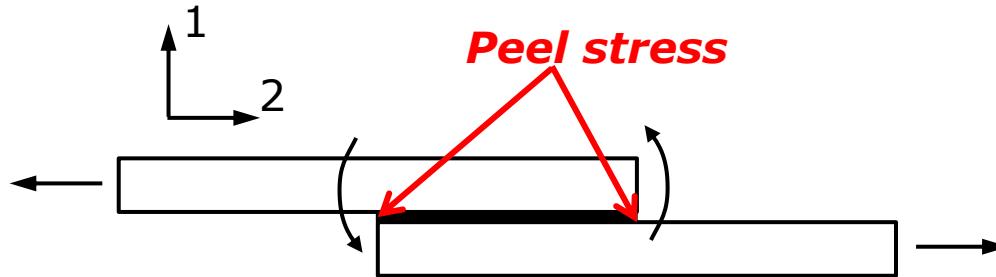
- Which one is a good adhesive joint design?



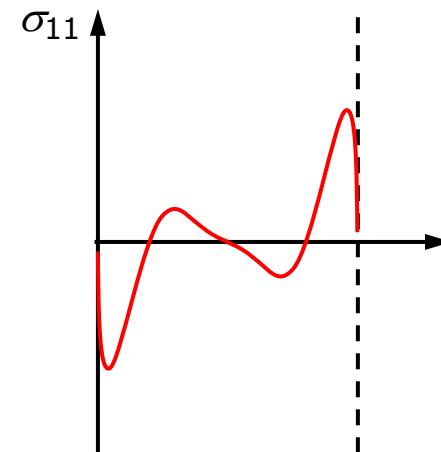
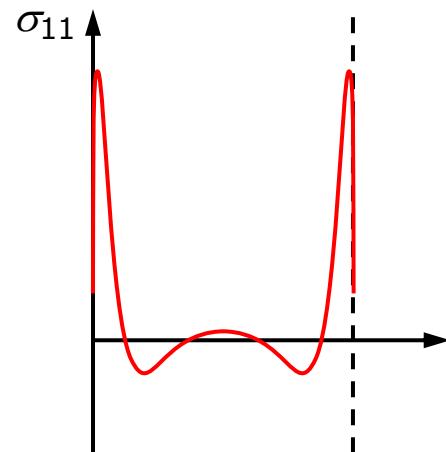
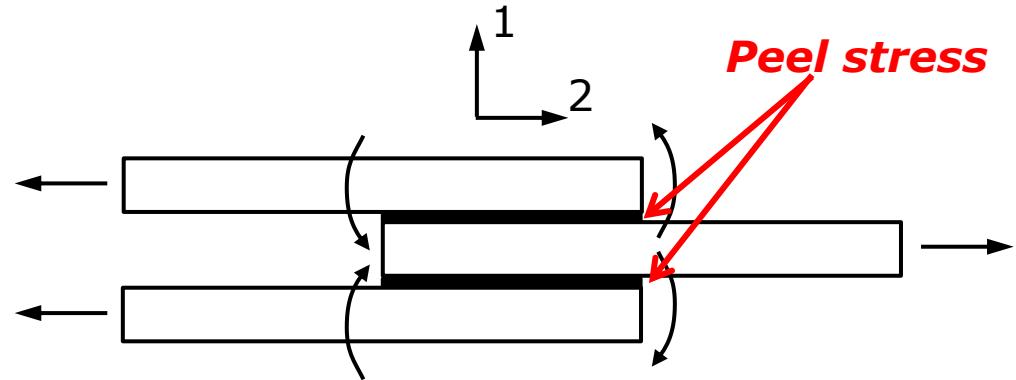
# Adhesive joint design

- **Peel stress** distribution

< Single-lap joint >

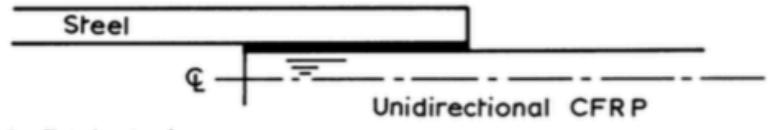
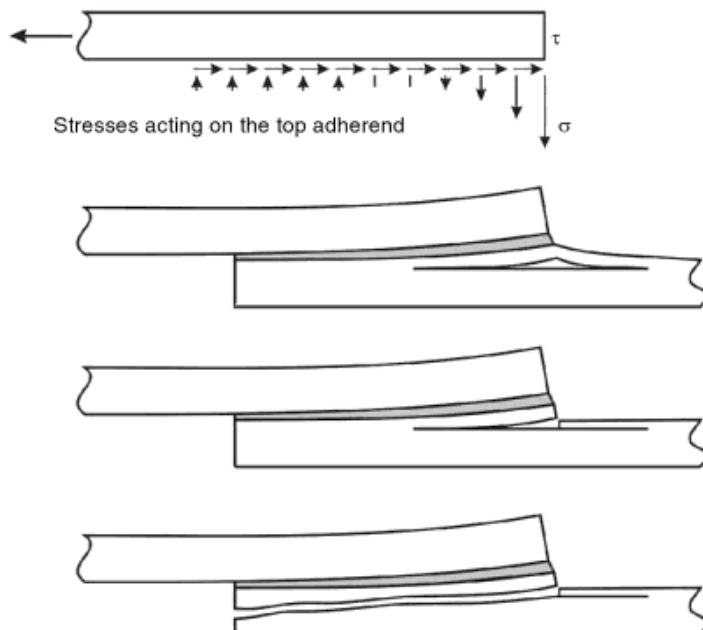
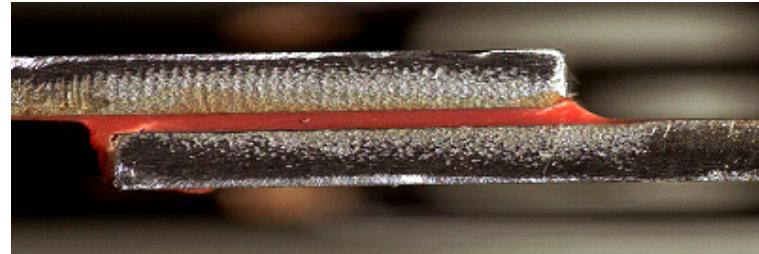


< Double-lap joint >



# Adhesive joint design

- Reduction of peel stress concentration



1. Basic design



2. Outside taper



3. Inside taper



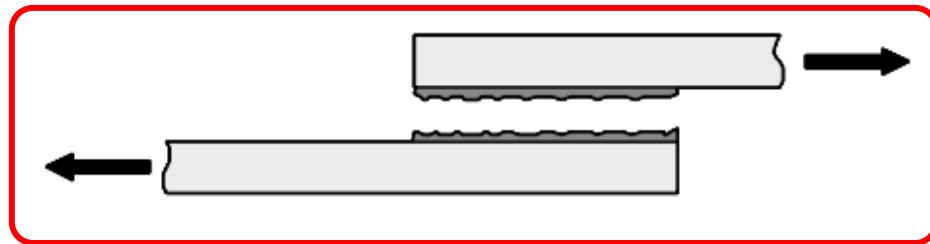
5. Inside taper and adhesive fillet

< Interlaminar failure of composite adherend  
due to the peel stress concentration >

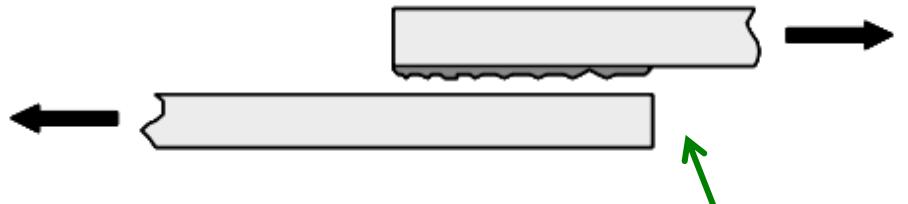
< Design modification reducing stress concentration >

# Failure modes

- Cohesive failure



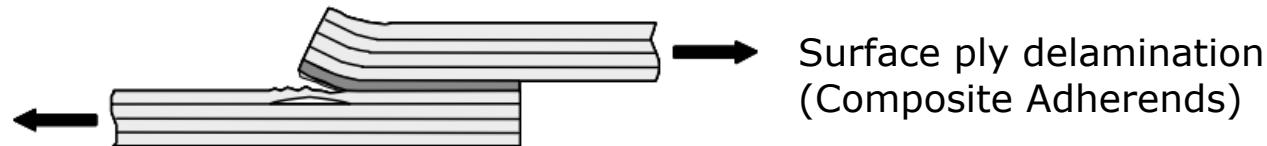
- Interfacial failure



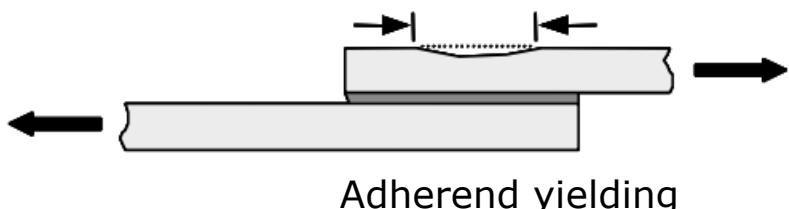
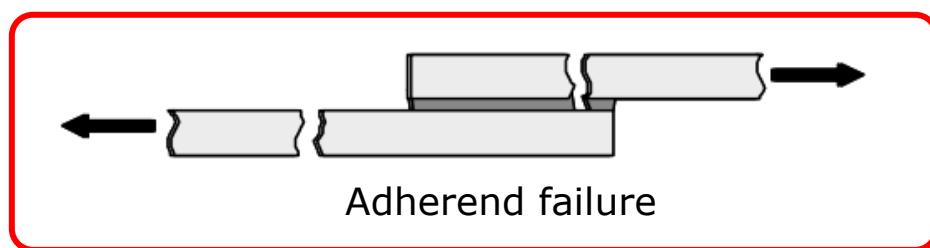
- Mixed failure



- Adherend failure



*It matters how to make full use of adhesive strength minimising the weight!*



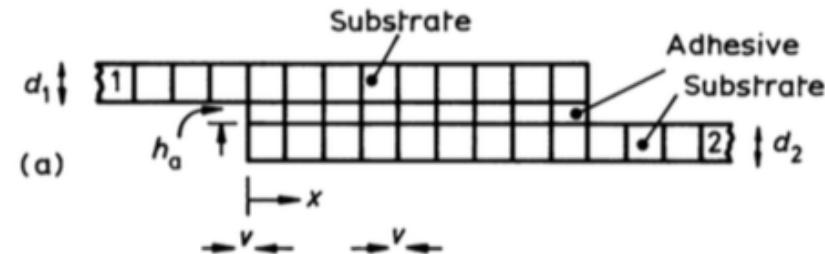
Adherend failure

Adherend yielding

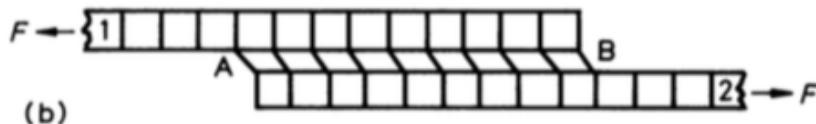
# Adhesive joint design

- Shear stress distribution**

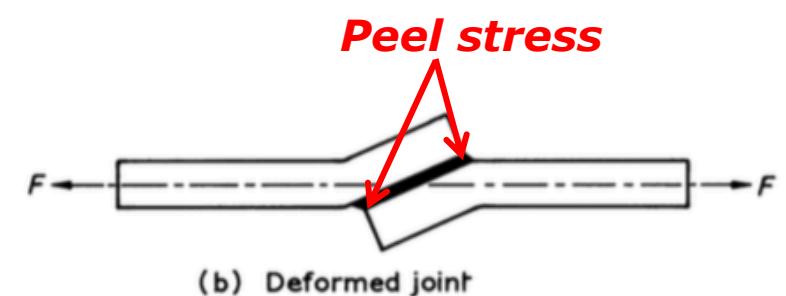
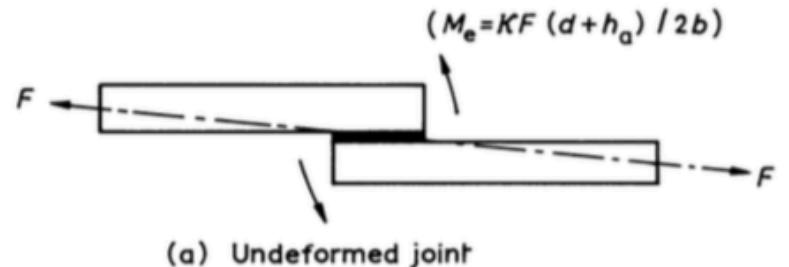
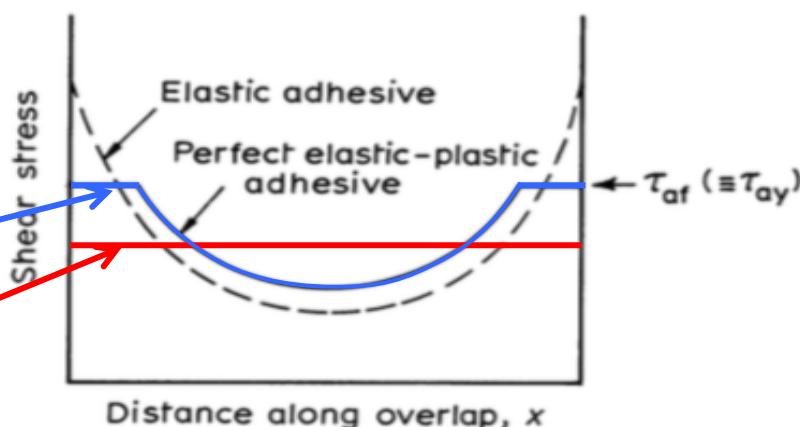
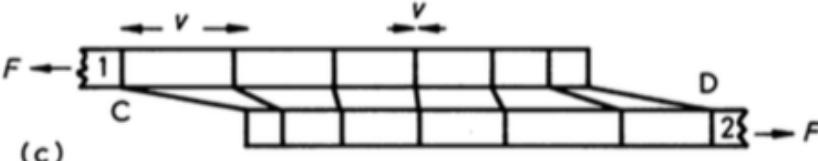
Unloaded



Inextensible  
adherends



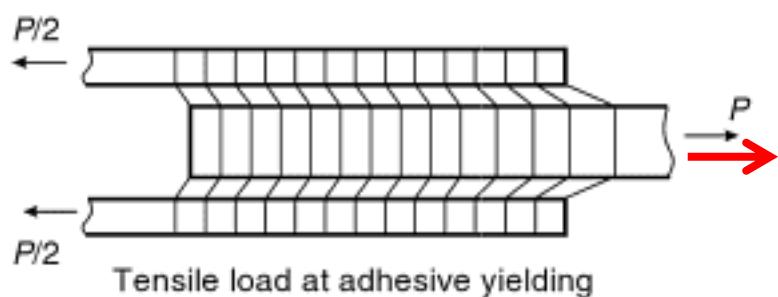
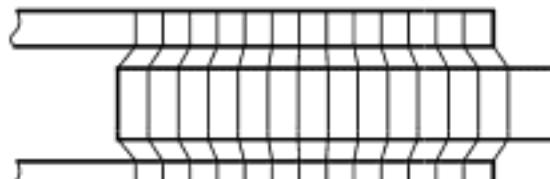
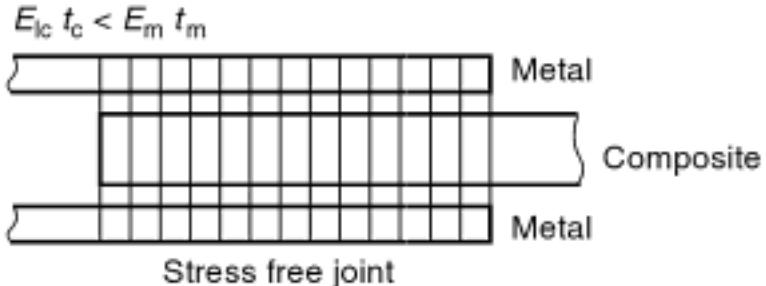
Extensible  
adherends



We often use the average shear stress for the joint strength, but should bear in mind that the actual shear stress is not uniform.

# Adhesive joint design

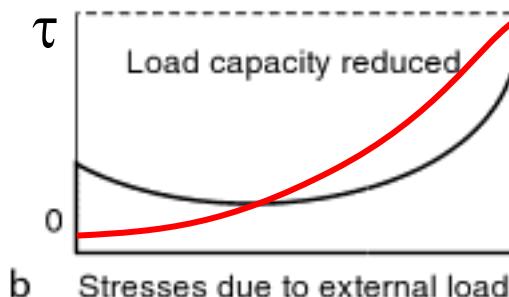
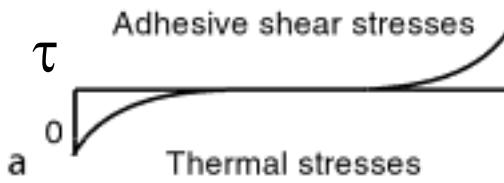
- Thermal residual stress



**CTE (Coefficient of Thermal Expansion)**

- Adhesive (Polymer):  $30-60 \times 10^{-6}$  m/m°C
- Steel:  $\sim 11 \times 10^{-6}$  m/m°C
- Aluminium:  $\sim 24 \times 10^{-6}$  m/m°C
- Carbon composite:  $\sim 0$  m/m°C

*Adhesive curing temp.: 80~180°C*



## Factors affecting the thermal residual stress

- Adherends stiffness
- Curing temperature
- CTEs of adherends & adhesive
- **Bonding thickness**

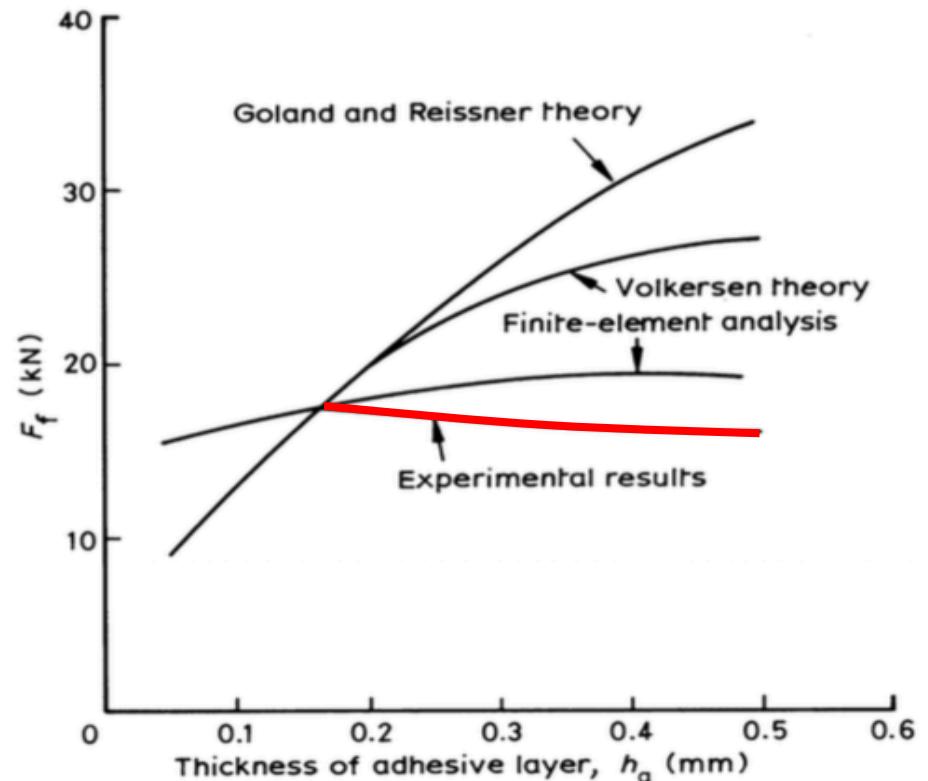
# Adhesive joint design

- Importance of the bonding thickness
  - The bonding thickness significantly affects the joint strength.
  - For brittle adhesive, a thinner adhesive layer causes less stress concentration as well as less thermal residual stress.
  - Generally, 0.1~0.2 mm thickness is recommended.

However, analytical models predict the opposite trend. (Thermal residual stress is not taken into account.)

For ductile adhesive cured at room temperature (e.g. polyurethane), a thicker adhesive layer may exhibit higher strength.

Many conflicting experimental results exist → Generalisation is very difficult.



# Adhesive joint design

***Correct prediction of the stress distribution within the adhesive layer is important to predict the joint failure. When you choose a analytical model, you should check its limitation.***

## Analytical models

- Elastic analysis
  - Volkersen (1938)
  - Goland & Reissner (1944)
  - Ojalvo & Eidinoff (1978)
  - ...
- Elastic-plastic analysis
  - Hart Smith (1973)
  - Adams (1997)
  - ...

- Can it consider the material non-linearity of the adhesive and adherends?
- Can it consider dissimilar adherends?
- Can it predict both peel and shear stresses?
- Is it a closed-form or a numerical solution?

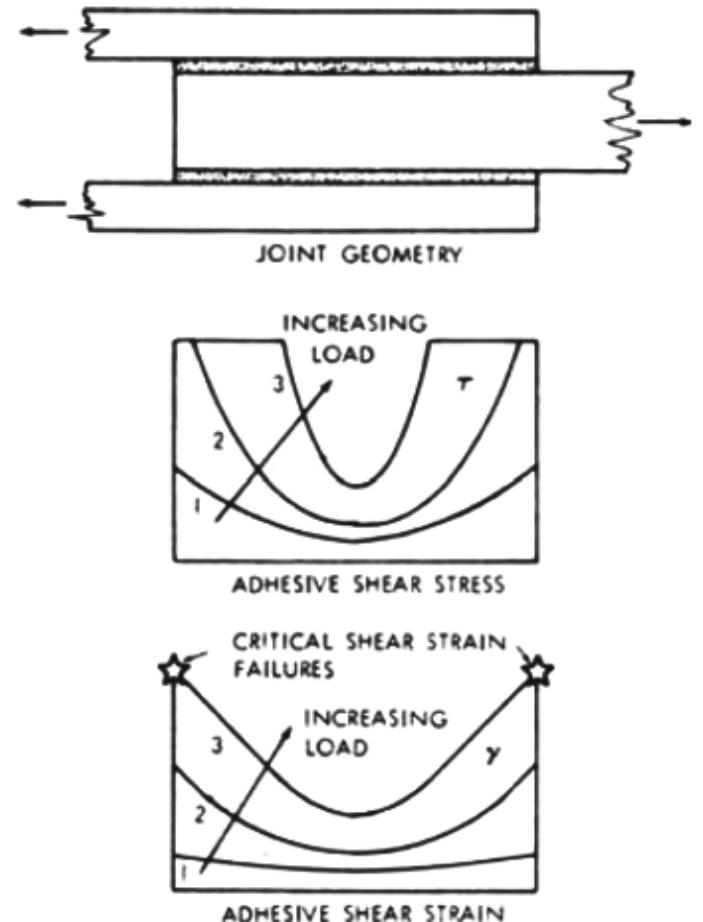
***For complex geometry and loading conditions (e.g. thermal), FEA is used. But, singularity issues exist.***

# Adhesive joint design

***One should choose an appropriate failure criterion.***

- Based on Continuum Mechanics
  - Maximum stress/strain criterion
  - Von Mises stress/strain criterion
- Based on Fracture Mechanics
  - Stress intensity factor
  - Energy release rate
  - Damage model
- For Fatigue loading
- For Impact loading

\* *Unfortunately, there is no golden rule due to the complexity, and people still highly rely on experimental methods.*

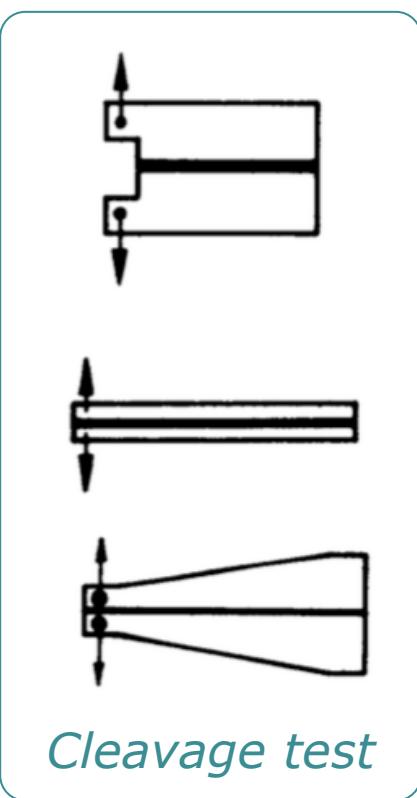
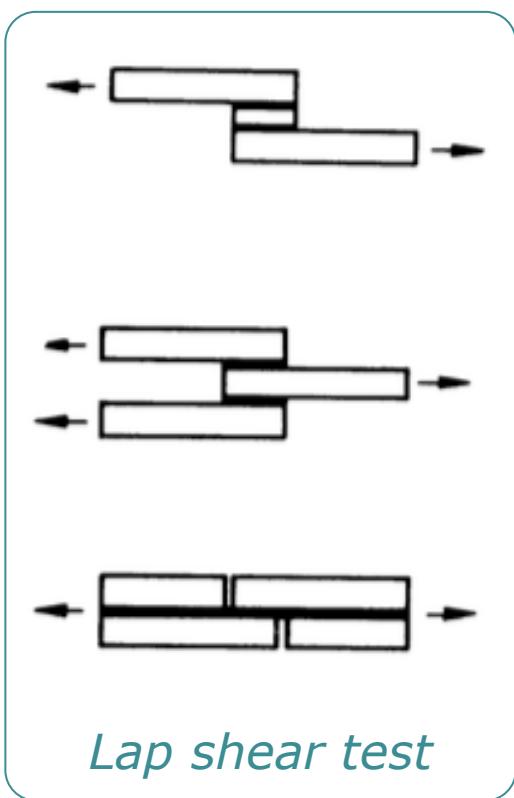


Failure prediction based on max. shear strain (Hart-Smith model)

# Adhesion strength tests

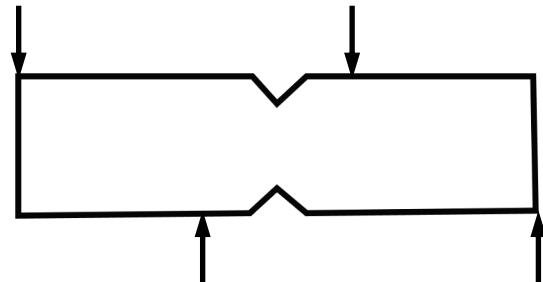
*Testing is crucial since the analytical models and failure predictions are not perfect !!*

- Structural testing

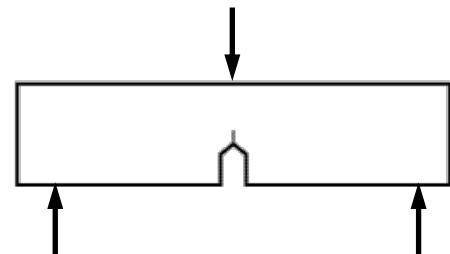
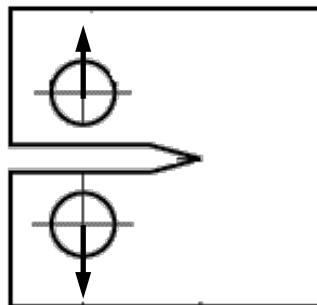


- Material testing

- Tensile and shear test



- Fracture toughness test



# References

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- FC Campbell, Manufacturing Technology for Aerospace Structural Materials, 2006, Elsevier.
- John Cutler, Understanding Aircraft Structures, 2005, Blackwell Publishing.
- A Skorupa and M Skorupa, Riveted Lap Joints in Aircraft Fuselage, Springer.
- MCY Niu, Airframe Structural Design, 1995, Comilit press.
- MCY Niu, Composite Airframe Structures, 1992, Comilit press.
- Shigley's Mechanical Engineering design, McGraw-Hill.
- AJ Kinloch, Adhesion and Adhesives: Science and Technology, 1987, Springer.
- LFM da Silva, A Öchsner, RD Adams, Handbook of Adhesion Technology, 2011, Springer.
- JYS Ahmad, Machining of Polymer Composites, 2009, Springer.
- RS Shoberg, Mechanical Testing of Threaded Fasteners and Bolted Joints, Vol. 8, ASM Handbook, ASM International, 2000.