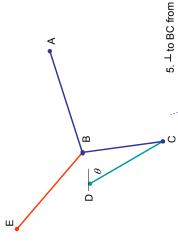
#### 3. Kinematics and Principle of Virtual Work

AENG21350 Design 2

Department of Aerospace Engineering University of Bristol



## 3.1 Kinematics of an undercarriage -2



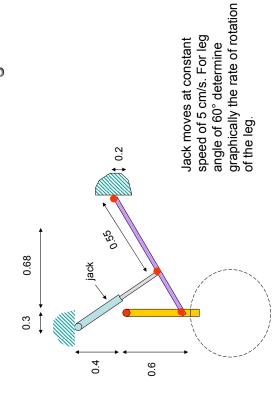
2. Absolute velocities on c and b lines ( $\bot$  to DC and AB respectively) 3. Jack velocity known in direction and magnitude

1. Fixed points: D, A, E.

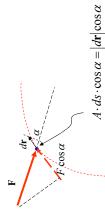
4, Projection on jack velocity on b line will identify point b

5.  $\pm$  to BC from point b will provide velocity (V<sub>B</sub>)<sub>c</sub> and intersection with c line  $\omega_{\log} = \frac{V_C}{CD} = \frac{19.5}{60} = 0.325 \quad rads^{-1}$ 6. Velocity point C identified. In scale:  $V_C = 19.5 \text{ cm/s}$ 

## 3.1 Kinematics of an undercarriage -1



#### 3.1 Work of a force and a couple

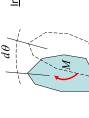


Infinitesimal work of a force:

$$dU = \mathbf{F} \cdot d\mathbf{r}$$

 $dU = F \cdot ds \cdot \cos \alpha$ 

 $dU = (\mathbf{i}F_x + \mathbf{j}F_y + \mathbf{k}F_z) \bullet (\mathbf{i}dx + \mathbf{j}dy + \mathbf{k}dz) = F_x dx + F_y dy + F_z dz$ 



Infinitesimal work of a couple:  $dU = M \cdot d\theta$ 

 $U = \int F \cos \alpha \cdot ds$  $U = (M \cdot d\theta)$ 

Integral forms:

### 3.2 Principle of virtual work (PVW)

The virtual work done by external active forces on a ideal mechanical system in equilibrium is zero for any and all virtual displacements consistent with the constraints

Constraint: restriction of motion by the supports.

 $\partial U = 0$ 

 $\delta U$  = total virtual work done on the system by all active forces during a virtual displacement

#### 2 advantages:

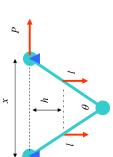
- Not necessary to dismember ideal systems to establish relations between the active
- We may determine the relations between the active forces directly without reference to reactive forces



cannot be used unless work done by internal friction forces is included forces do negligible work during any PVW requires that internal friction virtual displacement. If not, PVW

system under consideration only active forces (it is not a FBD!) diagram which isolates the When using PVW, draw a

#### 3.4 Example



mass  $\it m$  and length  $\it l.$  For a given force  $^{P}$  determine the angle  $\, heta$  for equilibrium Each of the 2 uniform hinged bars has

Applying PVW:

$$P \cdot \partial x + 2mg \cdot \partial h = 0$$

 $x = 2l\sin\frac{\theta}{2}$  $h = \frac{l}{2} \cos \frac{\theta}{2}$ 

Substituting:

$$Pl\cos\frac{\theta}{2}\cdot\delta\theta - 2mg\frac{l}{4}\sin\frac{\theta}{2}\cdot\delta\theta = 0$$

$$\frac{\theta}{2} \qquad \&x = l\cos\frac{\theta}{2} \cdot \delta\theta$$

$$\frac{\theta}{2} \qquad \&\theta = -\frac{l}{4}\sin\frac{\theta}{2} \cdot \delta\theta$$

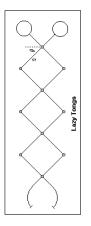
$$i\theta - 2mg\frac{l}{4}\sin\frac{\theta}{2}\cdot\delta\theta = 0$$

$$\theta = 2 \tan^{-1} \frac{2P}{mg}$$

### 3.3 Considerations for mechanisms

- 1. Identify number of degrees of freedom for your system (Grubnel formula)
- change at time while holding the others constant degrees of freedom. With each application, we 2. Apply PVW as many times as there are allow only one independent coordinate to

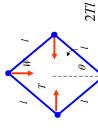
#### 3.5 Lazy tongs



Current collectors of electric locomotive If s is the length of a link in lazy tongs of distance of  $x = 2Ns \sin \theta$ , where  $\theta$  is the (pantograph), concertina mechanisms. find the velocity ratio  $dx/d\theta$ , and so the angle shown in the figure. It is easy to N sections, then they will extend a "leverage" of the tongs.

#### Example: car jack.

Load is applied by screw thread across horizontal diagonal. If the load on the jack is W, find the tension in the screw

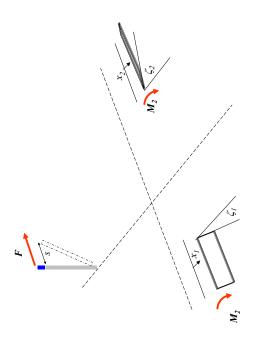


Work done by forces T:  $2T \cdot l \cdot \cos \theta \cdot \delta \theta$ **Ans**. Application of PVW. Only  $\theta$  is DOF

 $-W \cdot 2 \cdot l \cdot \sin \theta \cdot \delta \theta$ Work done by W:

 $2TI\cos\theta\cdot\delta\theta-2WL\sin\theta\cdot\delta\theta=0$   $|T=W\tan\theta|$ 

# 3.6 Applications to aircraft control circuit



#### List of concepts to know

- Velocity diagrams for undercarriage
- PVW
- Applications to lazy tong and aircraft control systems

# 3.6 Applications to aircraft control circuit

Applying PVW:

$$F \cdot \delta S - M_1 \delta \zeta_1 - M_2 \delta \zeta_2 = 0$$

Pilot must counteract the combined work due to the restoring moments on the left and right aileron.

$$F = M_1 \left( \frac{\partial \zeta_1}{\partial s} \right) + M_2 \left( \frac{\partial \zeta_2}{\partial s} \right) = M_2 \left( \frac{\partial \zeta_1}{\partial x_1} \frac{\partial x_1}{\partial s} \right) + M_2 \left( \frac{\partial \zeta_2}{\partial x_2} \frac{\partial x_2}{\partial s} \right)$$

If  $x_1=x_2=x o$  no differential applied to the system between control rod and point of application on each aileron:

$$F = M_2 \left( \frac{\partial \zeta_1}{\partial x} \frac{\partial x}{\partial s} \right) + M_2 \left( \frac{\partial \zeta_2}{\partial x} \frac{\partial x}{\partial s} \right)$$

Need to know control hinge moment at different speeds and information on relation between s