

Q1 Assuming the density to be constant

$$P = P_a + \rho gh = 101,300 + 1000 \times 9.81 \times 10 = 1.994 \times 10^5 \text{ Pa}$$

Note the approximate equivalence 1 atmosphere  $\approx 10$  m water.

$$\text{Q2 a) } p = \rho gh + p_v = p_m gh \Rightarrow h = \frac{101300}{13560 \times 9.81} = 0.762 \text{ m}$$

b) barometer reading remains unchanged, height unchanged by shape

$$\text{c) } \rho_o gh + p_v = P \Rightarrow h = \frac{101300 - 5.7 \times 10^3}{790 \times 9.81} = 12.336 \text{ m}$$

Q3 As the fluid interface in tube C rises, the heights of fluid in A and B also change, but not by the same amount as all 3 have different diameters (call these three heights  $\Delta h_A, \Delta h_B$  and  $\Delta h_C$ )

$$\Delta h_A = \Delta h_C \cdot \frac{A_C}{A_A} = 0.06 \times \frac{70}{500} = 8.4 \times 10^{-3} \text{ m}$$

$$\Delta h_B = \Delta h_C \cdot \frac{A_C}{A_B} = 0.06 \times \frac{70}{800} = 5.25 \times 10^{-3} \text{ m} \quad (\text{remember } \Delta h_B \text{ represents a decrease in height})$$

At the undisturbed interface (where  $P_a$  = atmospheric pressure) or some initial pressure applied to both A & B)

$$P_a + \rho_A gh_A = P_a + \rho_B gh_B \Rightarrow \rho_A gh_A = \rho_B gh_B$$

Once an additional pressure has been applied to B ( $\Delta P$  say) the pressure at interface becomes

$$P_a + \rho_A g(h_A + \Delta h_A - \Delta h_C) = P_a + \Delta P + \rho_B g(h_B - \Delta h_B - \Delta h_C)$$

using relation from undisturbed interface

$$\rho_A g(\Delta h_A - \Delta h_C) = \Delta P + \rho_B g(-\Delta h_B - \Delta h_C)$$

$$\Delta P = [800 \times (8.4 \times 10^{-3} - 0.06) - 900 \times (-5.25 \times 10^{-3} - 0.06)] \times 9.81$$

$$\Delta P = 171.13 \text{ N/m}^2$$

Q4 for the bubble  $\frac{P_0 V_0}{T_0} = \frac{P_s V_s}{T_s}$  (ideal gas equation)

where the subscripts 0 and s refer to its initial position at a depth of 9m and its final position at the surface.  
Temperature is constant (isothermal process) so

$$\frac{P_0}{P_s} = \frac{V_s}{V_0} = \frac{d_s^3}{d_0^3}$$

hence  $d_s^3 = \frac{P_0}{P_s} d_0^3$  and  $d_s = \sqrt[3]{\frac{P_0}{P_s}} d_0$

$$= \sqrt[3]{\frac{1.013 \times 10^5 + 9 \times 9.81 \times 1000}{1.013 \times 10^5}} \times 0.004$$

$$= 4.93 \times 10^{-3} \text{ m} = 4.93 \text{ mm}$$

Q6 Volume of the balloon =  $\frac{4}{3} \pi (0.4)^3 = 0.2681 \text{ m}^3$

The effective density of the balloon & equipment =  $\rho_{\text{eff}} = \frac{0.06 + 0.1 + M_H}{0.2681}$

where  $M_H$  is the mass of helium.

from Q5

$$P = 1.013 \times 10^5 \left(1 - \frac{0.0065 \times 6000}{288.15}\right)^{\frac{9.81}{287 \times 0.0065}} = 47150.405 \text{ N/m}^2$$

$$T = T_{s1} - \Delta T = 288.15 - 0.0065 \times 6000 = 249.15^\circ \text{ K}$$

$$\rho = \frac{P}{RT} = \frac{47150.405}{287 \times 249.15} = 0.6594 \text{ kg/m}^3$$

for equilibrium  $\rho = \rho_{\text{eff}}$   $\frac{0.06 + 0.1 + M_H}{0.2681} = 0.6594$

$$M_H = 0.0168 \text{ kg}$$

Q7 Gauge pressure =  $\rho g h$  where  $\rho = 910 \text{ kg/m}^3$  and  $g = 9.81 + 30 \text{ m/s}^2$

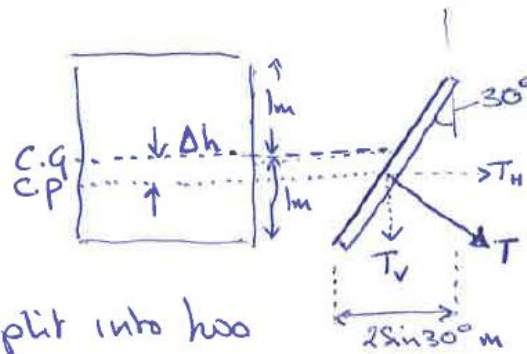
$$= 910 \times 39.81 \times 10 = 3.6227 \times 10^5 \text{ N/m}^2$$

Thrust = weight of fluid above =  $\pi r^2 \times h \times \rho \times g$

$$= 4.552 \times 10^6 \text{ N}$$

Q8

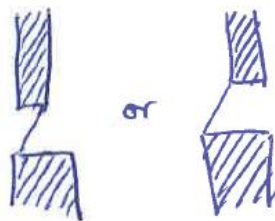
Case 1



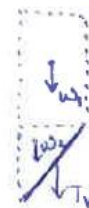
Consider the total thrust to be split into two components:  $T_H$  and  $T_V$

Consider  $T_V$

The vertical thrust equals the "weight of water above" the gate. As you can see this should be the equivalent weight of water above, so that even though there is little water above the gate in the drawing there is no difference to the pressure & hence thrust if we had



so take



$$T_V = W_1 + W_2$$

$$\text{So } T_V = 18 \times 2 \tan 30^\circ \times 2 \times 1000 \times 9.81 + \left(\frac{1}{2}\right) \times 2 \times 2 \tan 30^\circ \times 1000 \times 9.81$$

$$= 2 \times 2 \times \tan 30^\circ \times 1000 \times 9.81 \times (18 + \frac{1}{2}) = 430449.27 \text{ N}$$

Note we can leave out the weight of the air as the atmospheric pressure acts on the outside surface of the gate. Also we assume the atmospheric pressure does not change with the 20m change in height.

$T_H$  = "pressure at the centre of gravity"  $\times$  "vertical area"

$$= (P_a + \rho g h_{c.g.}) A_v - P_a A_v$$

$$= \rho g h_{c.g.} A_v$$

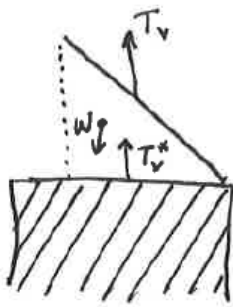
$$= 1000 \times 9.81 \times 19 \times 2 \times 2$$

$$= 745560 \text{ N}$$

note that we can neglect atmospheric pressure as it acts on ~~the~~ both sides of the gate

$$T = \sqrt{T_V^2 + T_H^2} = 8.609 \times 10^5 \text{ N}$$

Q8 Continued case 2



from the equality of vertical forces

$$T_v = T_v^* - W$$

where  $T_v^*$  is the resultant from the vertical thrust on the base given by

$$T_v^* = 20 \times 2 \tan 30^\circ \times 2 \times 1000 \times 9.81$$

and  $W$  is the weight of the fluid

$\frac{1}{2}$

$$W = \left(\frac{1}{2}\right) \times 2 \times 2 \tan 30^\circ \times 1000 \times 9.81$$

so

$$T_v = 2 \times 2 \times \tan 30^\circ \times 1000 \times 9.81 (20 - 1) = 430449 \quad (\text{as before})$$

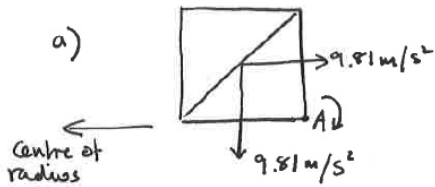
The horizontal thrust is unchanged hence  $T_h = 745560 \text{ N}$

$$T = 8.609 \times 10^5 \text{ N}$$

Height of Centre of pressure from the surface is (from notes)

$$h_{c.p} - h_{c.g} = \frac{I_{xx}}{h_{c.g} \times A} \Rightarrow h_{c.p} - 19 = \frac{4 \times 2.2^3}{12 \times 19 \times 2^2} \Rightarrow h_{c.p} = 19.018$$

Q9



Vertical thrust on base of fuel tank

$$T_v = 850 \times 9.81 \times 1 \times 1 \times 0.5 = 4169.25$$

acting downward (anticlockwise) at a distance  $\frac{1}{3}$  m

[Note Consider the C.G. of the triangle]

horizontal thrust given by  $T_H = \rho g h_{c.g.} \times A$

[This is exactly the same as the horizontal thrust of a full fuel tank]

$$T_H = 850 \times 9.81 \times 0.5 \times 1 \times 1 = 4169.25$$

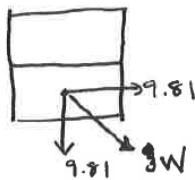
Line of action of the horizontal force ' ~~W~~ '  $h_{c.p.} - h_{c.g.} = \frac{I_{xx}}{h_{c.g.} \times A} = \frac{1 \times 1 \times 1^3}{12 \times 0.5 \times 1 \times 1}$

$$h_{c.p.} = \left( \frac{1}{2} + \frac{1}{6} \right) \text{ m}$$

So horizontal force acts at a height of  $(\frac{1}{3})$  m from A in a clockwise direction

$$\text{hence Moment}_A = \left\{ 4169.25 \times \frac{1}{3} - 4169.25 \times \frac{1}{3} \right\} = 0 \text{ Nm}$$

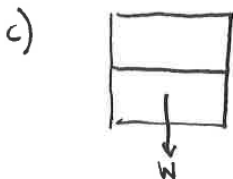
b) Constrained fluid acts like a solid



$$\text{hence moment} = (\text{fluid mass}) \times 9.81 \times \cancel{0.25} - (\text{fluid mass}) \times 9.81 \times 0.5$$

$$= -0.5 \times 1 \times 1 \times 850 \times 9.81 \times 0.25 \text{ Nm}$$

$$= -1042.13 \text{ Nm}$$



$$\text{moment} = (\text{fluid mass}) \times 9.81 \times 0.5$$

$$= -0.5 \times 1 \times 1 \times 850 \times 9.81 \times 0.5 \text{ Nm}$$

$$= -2084.6 \text{ Nm}$$

Note we see how the sloshing of the fluid means that the fluid is on the point of producing a "toppling" moment on the fuel tank.