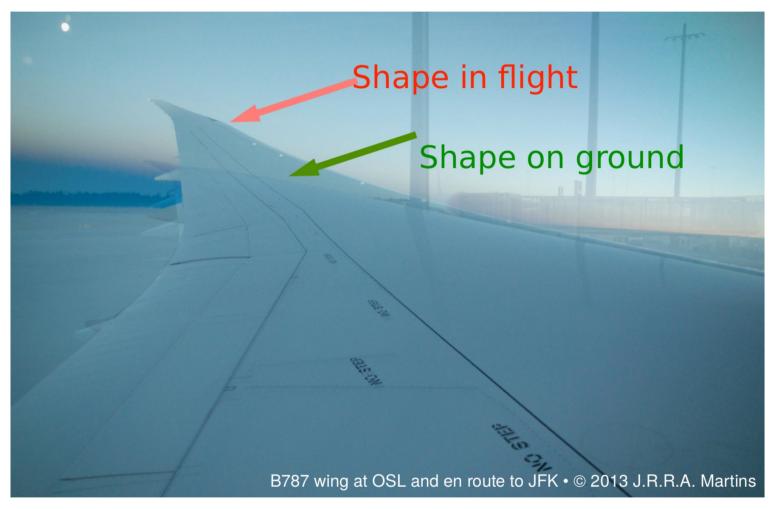
# Multidisciplinary Design Optimization

Coupled problem

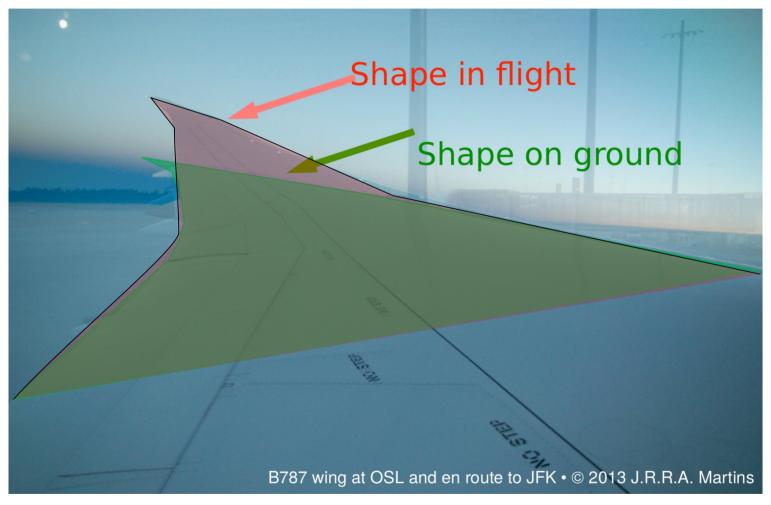
# Multidisciplinary Design Optimization

 Multidisciplinary Design Optimization (MDO) focuses on solving optimization problems spanning across multiple interacting disciplines

# Coupled problem

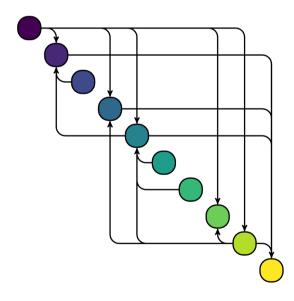


# Coupled problem



### Interdisciplinary Compatibility

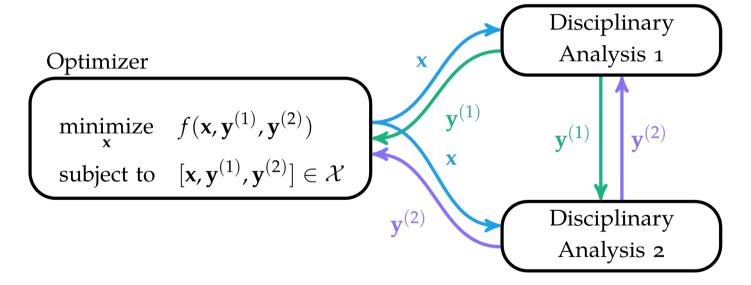
- If there are dependency cycles, then no topological ordering exists
- A general approach is iterative techniques such as the Gauss-Seidel method
- Depending on the nature of the problem, iterative methods can converge slowly



### Interdisciplinary Compatibility



• If there are multiple disciplines, then dependencies have to be considered



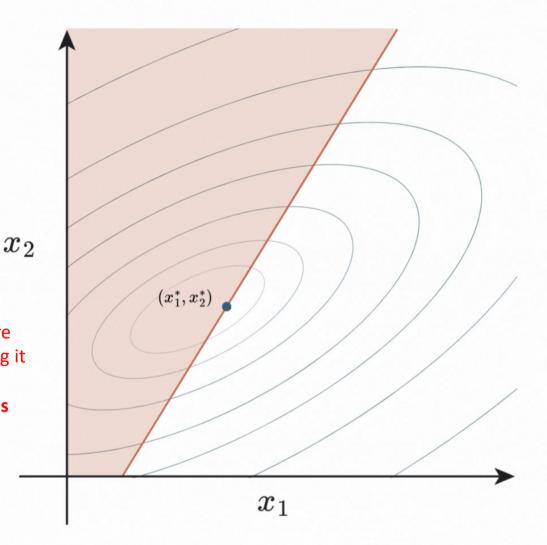
### Illustrative scenario

- •Optimal values  $(x_1^*, x_2^*)$  lie near the center of an objective function's contours, constrained by a red region.
- •In **sequential optimization**, only one variable is adjusted at a time, failing to follow a feasible descent path along the constraint.
- •In **simultaneous optimization**, both variables are adjusted together, enabling a feasible descent path toward the true optimum.

**Industry practice** often follows **sequential optimization**, where each discipline optimizes its part independently before passing it to the next.

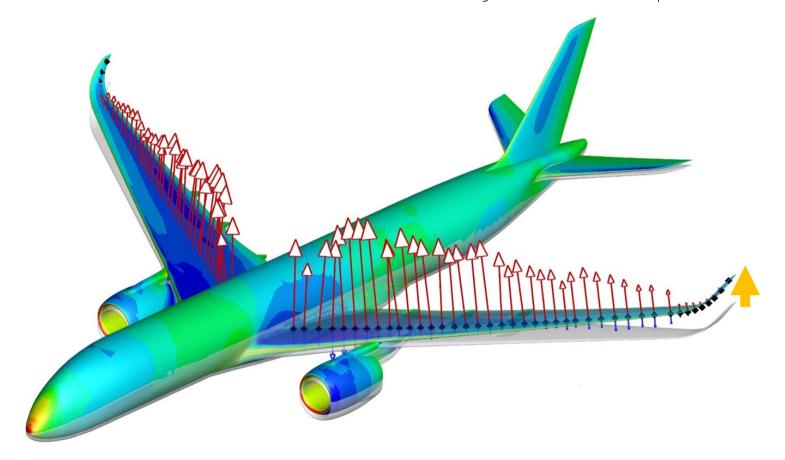
•Sequential optimization typically leads to suboptimal designs due to lack of coordination across disciplines. Two variables/disciplines:  $x_1$  and  $x_2$ .

https://medium.com/optimise-consultancy/what-is-multidisciplinary-design-optimisation-88e6e6b933b3



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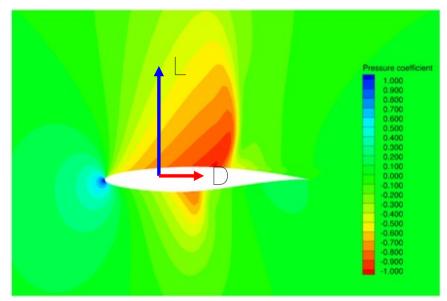
### What is an MDA? Static Aeroelasticity for example?



Source: DLR

### But first, what is Disciplinary Optimization?

Example: Aerodynamics (L/D max)

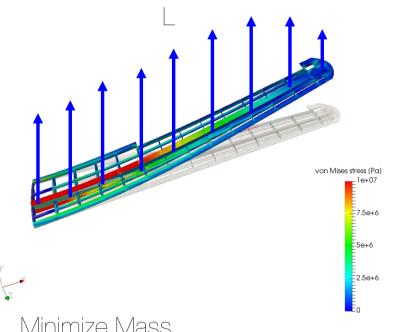


Source: NLR

Minimize D w.r.t. shape, a Subject to L = W

# What is Disciplinary Optimization (2)?

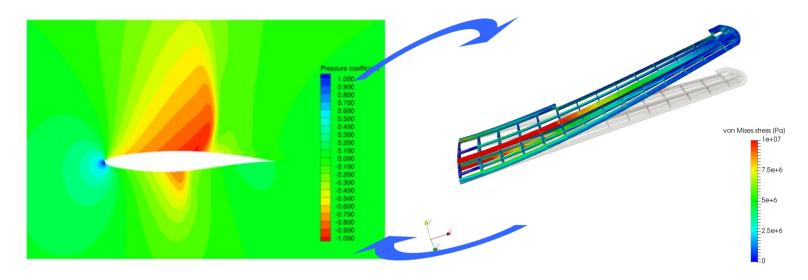
Another example: Structures



Source: simscale.com

Minimize Mass w.r.t. thicknesses Subject to  $\sigma \leq \sigma y$ 

### However... Disciplines are not isolated:



Structural deformation of wing

the changes in the shape exposed to airflow

Changes in the shape exposed to airflow → changes in the aerodynamic loads

# Then, how do we solve the complete system?

Nodal forces

Load Transfer

Rendall, T. C. S., & Allen, C. B. (2008). Unified fluid-structure interpolation and mesh motion using radial basis functions. International Journal for Numerical Methods in Engineering, 74(10), 1519-1559.

Forces on aero grid points

Structure Analysis

The solution of the complete system is the set of displacements and forces that "satisfy" this loop

Aerodynamic Analysis Nodal displacements

Radial Basis Function displacement interpolation (same for y and z)

$$u_x = \sum_{i=1}^{N_s} \alpha_i^x \phi(\|\mathbf{x} - \mathbf{x_i}\|) + \gamma_0^x + \gamma_x^x x + \gamma_y^x y + \gamma_z^x z$$

Displacement interpolation matrix

$$u_a = Hu_s$$



Loads using principle of virtual work

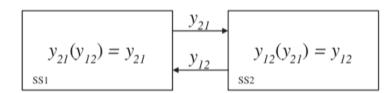
$$f_s = H^T f_a$$

Aero grid displacements

### Multi-Disciplinary Analysis

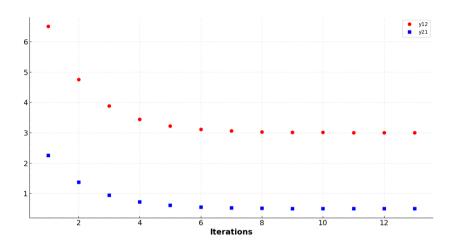
- Computation of the state variables at equilibrium for given x and z
  - Generally computed using a fixed-point algorithm (Jacobi or Gauss-Seidel)
  - Or a root-finding method (Newton-Raphson)

$$y21(y12)=0.25 \cdot y12-0.25$$
  
 $y12(y21)=2+2 \cdot y21$ 



Check the default tolerence

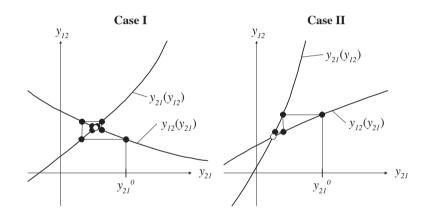
#### Introductio to FPI



- (Step 0) choose initial guess  $y_{12}^0$ , set i=0
- (Step 1) i = i + 1
- (Step 2)  $y_{21}^i = y_{21}(y_{12}^{i-1})$
- (Step 3)  $y_{12}^i = y_{12}(y_{21}^i)$
- (Step 4) if  $|y_{12}^i y_{12}^{i-1}| < \varepsilon$  stop, otherwise go to (Step 1)

### **Fixed Point Iteration Convergence**

Developed proof of new convergence condition form



$$\left| \frac{\partial y_{21}(y_{21})}{\partial y_{21}} \right| > \left| \frac{\partial y_{12}(y_{21})}{\partial y_{21}} \right| \Leftrightarrow \left| \frac{\partial y_{12}(y_{12})}{\partial y_{12}} \right| > \left| \frac{\partial y_{21}(y_{12})}{\partial y_{12}} \right|$$

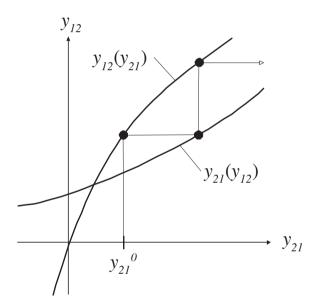
$$\mathbf{y_p} = \mathbf{y}(\mathbf{y_p})$$



University of Michigan Department of Mechanical Engineering

October 29, 2005

### **Fixed Point Iteration Divergence**



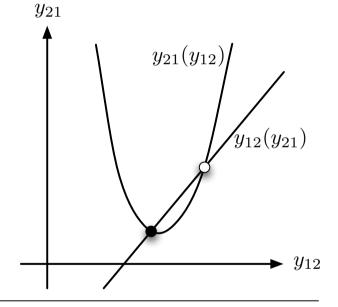


### **Multiple Fixed Points**

#### FPI:

Unknown if a repelling fixed point would have led to a better solution.

- Attractive fixed point
- Repelling fixed point





University of Michigan Department of Mechanical Engineering

October 29, 2005

### IF POSSIBLE

>> A=[-0.25 1; 1 -2]

Use inversion

 $y21 = 0.25 \cdot y12 - 0.25$ 

 $-0.25 \cdot y12 + y21 = -0.25$ 

 $y12 = 2 + 2 \cdot y21$ 

 $y12 - 2 \cdot y21 = 2$ 

or Gauss Seidel

-0.2500 1.0000 1.0000 -2.0000

b =

-0.2500 2.0000

>> y=A\b

y =

3.0000 0.5000

# Define the system of equations eq1 = y21 - (0.25 \* y12 - 0.25) eq2 = y12 - (2 + 2 \* y21)

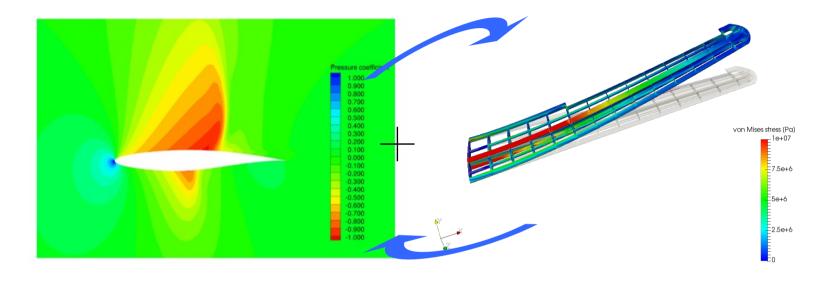
# Find the exact solution for comparison
solution = sp.solve((eq1, eq2), (y12, y21))
print("Exact Solution:")
print(f"y12 = {solution[y12]}")
print(f"y21 = {solution[y21]}")

A\*y=b

### So start...

https://colab.research.google.com/drive/1Spc5Sh1u0A91Ryk05ep1Xfhy7x7ieKbg#scrollTo=7abCXxl3Gmp

### So, we need to analyze BOTH disciplines at the SAME TIME



Minimize D, or Mass, {or a combination of D and Mass} w.r.t. shape, a, thicknesses
Subject to:

$$L = W$$
 $O \leq OV$ 

# In practice, how do we solve that problem?

One possible approach: MultiDisciplinary Feasible (MDF, probably the most intuitive one...)

### Steps:

- 1. Start from a set of particular design variables: shape, a, thicknesses
- 2. Solve the complete system (with all the interactions) for these values
- 3. Evaluate objective function and constraints
- 4. From these values, the optimizer proposes a new set of design variables.

These steps are repeated until the optimum is reached.

### Assembling MDO systems

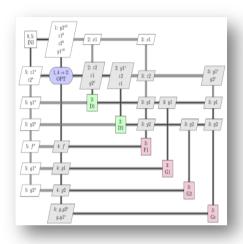


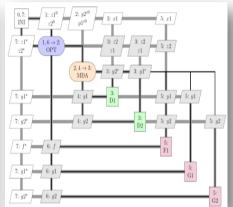
#### Monolithic

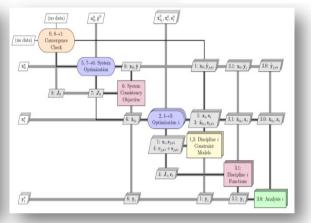
### All-at-Once (AAO) Simultaneous Analysis and Design (SAND) Individual Discipline Feasible (IDF) Multiple Discipline Feasible (MDF)

#### Distributed

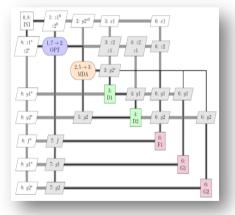
Concurrent Sub-Space Optimization (CSSO) Bi-Level System Synthesis (BLISS) Collaborative Optimization (CO) Analytical Target Cascading (ATC)

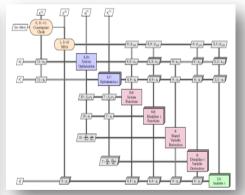


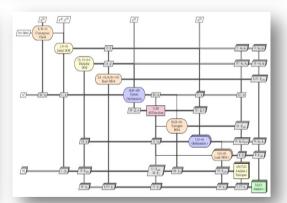




MDF Multidisciplinary Feasible approach—a complete analysis is performed at every optimization iteration. Also known as the All-in-One approach.

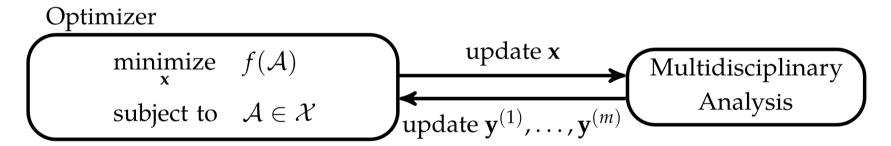






### Multidisciplinary Design Feasible

• The multidisciplinary design feasible architecture structures the MDO problem such that standard optimization algorithms can be directly applied to optimize the design variables



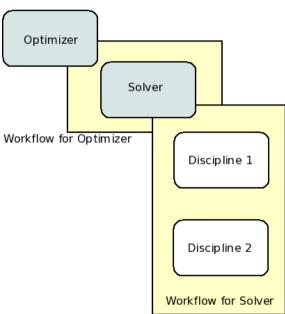
minimize 
$$f(\mathbf{x}, \mathbf{y}^{(1)}, \dots, \mathbf{y}^{(m)})$$
  $\longrightarrow$  minimize  $f(\mathrm{MDA}(\mathbf{x}))$  subject to  $\left[\mathbf{x}, \mathbf{y}^{(1)}, \dots, \mathbf{y}^{(m)}\right] \in \mathcal{X}$  subject to  $\mathrm{MDA}(\mathbf{x}) \in \mathcal{X}$ 

### Multidisciplinary Feasible (MDF)

■ The MDF architecture is the most intuitive for engineers

The optimization problem formulation is identical to the single discipline case, except the

disciplinary analysis is replaced by an MDA



# Illustrative example: the Sellar problem

2 disciplines involved Variables: x<sub>1</sub>, y<sub>1</sub>, y<sub>2</sub>, z<sub>1</sub>, z<sub>2</sub>

We'll see later what are the differences between these variables ...

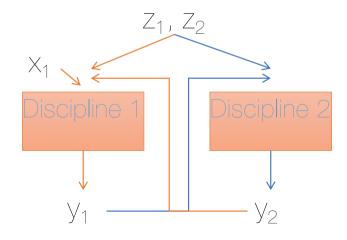
```
minimize x_1^2 + z_2 + y_1 + \exp(-y_2)
with respect to z, x or (z_1, z_2, x_1)
subject to :
3.16 - y_1 \le 0
y_2 - 24 \le 0
-10 \le z_1 \le 10
0 \le z_2 \le 10
0 \le x_1 \le 10
```

Discipline 1 : 
$$y_1(z_1, z_2, x_1, y_2) = z_1^2 + x_1 + z_2 - 0.2y_2$$
  
Discipline 2 :  $y_2(z_1, z_2, y_2) = \sqrt{y_1} + z_1 + z_2$ 

System Design", 34th Aerospace Sciences Meeting and Exhibit, Aerospace Sciences Meetings 1996

### Illustrative example: the Sellar problem

- Design variables: z<sub>1</sub>, z<sub>2</sub>, x<sub>1</sub> to minimize the objective
- Shared (or global) variables:  $Z_1, Z_2$
- Local variable: X<sub>1</sub>
- Coupling variables: y<sub>1</sub>, y<sub>2</sub>



minimize 
$$x_1^2 + z_2 + y1 + e^{-y_2}$$
  
with respect to  $z_1, z_2, x_1$   
subject to:  
 $\frac{y_1}{3.16} - 1 \ge 0$   
 $1 - \frac{y_2}{24} \ge 0$   
 $-10 \le z_1 \le 10$   
 $0 \le z_2 \le 10$   
 $0 \le x_1 \le 10$ 

Discipline 1: 
$$y_1(z_1, z_2, x_1, y_2) = z_1^2 + x_1 + z_2 - 0.2y_2$$
  
Discipline 2:  $y_2(z_1, z_2, y_1) = \sqrt{y_1} + z_1 + z_2$ 

Multidisciplinary analysis (MDA) consists in solution of the following equations

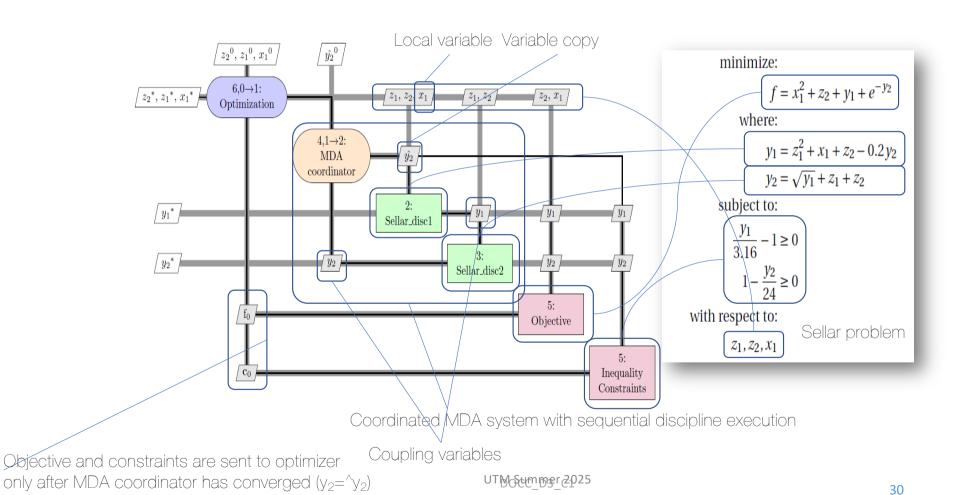
$$R_1 = C$$

 $R_1 = 0$   $\rightarrow$   $y_1$  solutions

$$R_2 = 0$$

 $R_2 = 0$   $\rightarrow$   $y_2$  solutions

# MDF illustration on the Sellar problem: MDF – Gauss-Seidel variant



# Sellar Problem using CasADi

https://colab.research.google.com/drive/1-qdgy0EKoe5qhFXzjSiDxLh39oWfbpyb#scrollTo=2LQITUPCibyA

#### Execute the scenario

Then, we execute the MDO scenario with the inputs of the MDO scenario as a dictionary. In this example, the gradient-based SLSQP optimizer is selected, with 10 iterations at maximum:

```
scenario.execute(input_data={"max_iter": 10, "algo": "SLSQP"})
                                Number of calls to the objective function by the optimizer: /
         INFO - 13:51:1/:
Out:
         INFO - 13:51:17:
         INFO - 13:51:17:
                                The solution is feasible.
         INFO - 13:51:17:
                                Objective: 3.1833939516400456
         INFO - 13:51:17:
                                Standardized constraints:
         INFO - 13:51:17:
                                   c_1 = 5.764277943853813e-13
         INFO - 13:51:17:
                                   c_2 = -20.244722233074114
         INFO - 13:51:17:
                                   y_1 = [2.06834549e-15]
                                   y_2 = [1.98063788e-15]
         INFO - 13:51:17:
         INFO - 13:51:17:
                                Design space:
         INFO - 13:51:17:
         INFO - 13:51:17:
                                | name | lower_bound
                                                             value
                                                                           upper_bound | type
         INFO - 13:51:17:
         INFO - 13:51:17:
                                                                                10
                                                                                         float
         INFO - 13:51:17:
                                 | z[0] |
                                             -10
                                                       1.977638883463326
                                                                                10
                                                                                         float
         INFO - 13:51:17:
                                              0
                                 | z[1]
                                                                                10
                                                                                         float
         INFO - 13:51:17:
                                 | y_1
                                             -100
                                                       1.777638883462956
                                                                               100
                                                                                          float
                                                                               100
         INFO - 13:51:17:
                                | y_2
                                             -100
                                                      3.755277766925886
                                                                                         float
         INFO - 13:51:17:
         INFO - 13:51:17: *** End MDOScenario execution (time: 0:00:00.128566) ***
     {'max_iter': 10, 'algo': 'SLSQP'}
```

https://gemseo.readthedocs.io/en/5.1.0/examples/mdo/plot\_se llar.html

### GO DEEPER?

- https://openmdao.org/newdocs/versions/latest/basic\_user\_guide/ multidisciplinary\_optimization/sellar.html
- https://gitlab.com/pablonorczyk/notes-on-mdao
- https://colab.research.google.com/github/OpenMDAO/OpenMDAO/ blob/master/openmdao/docs/openmdao book/basic user guide/sin gle disciplinary optimization/first optimization.ipynb

ITM Summer 2025

### Multidisciplinary Feasible (MDF)

#### Advantages:

- Intuitive procedure/no specialized knowledge required > Easy to incorporate existing models
- Always return a system design that satisfies the consistency constraints, even if the optimization process is terminated early good from a pratical engineering point of view

#### Disadvantages:

- Intermediate results do not necessarely satisfy the optimization constraints
- Cannot be parallelized
- Developing the MDA procedure with CSM/CFD might be time consuming\*, if not already available

\* Automatic mapping, postprocessing etc...

Gradients of the coupled

Gradients of the coupled

system more challenging to

compute

### Optimizer solver

- Requirements
- Problem to solve  $\begin{cases} & \min f(x) \\ & \text{wrt } x \in R^d \\ & \text{st } g_i(x) \leq 0 \text{ for } i = 1, ... \text{ m} \end{cases}$



- Evolutionnary Strategies (ES)
- Surrogate based Optimizer (SBO) or Bayesian Optimization (BO
- ...
- Gradient based Optimizer
- $\rightarrow$ Computation of the derivatives of f(x) and  $g_i(x)$  to iterate and satisfy the KKT optimality conditions
- → OpenMDAO focus on computation of sensitivities (adjoint vs direct)

$$\frac{\partial f}{\partial x_i}$$
,  $\frac{\partial g}{\partial x_i}$ ,  $\frac{\partial h}{\partial x_i}$ 

Optimizer

Search direction

Analysis

Gradient computation

Converged?

Need to be accurate

### When to use gradient-free optimizers???

- 1. Very cheap models
- 2. When you can't compute derivatives

### When to use gradient-free optimizers???

Noisy and discontinuous design space

# (YOU DON 'T KNOW a priori)

When you have a "noisy" design space, it means that the outputs change rapidly for a small change in the inputs. This noise might be caused by computational or physical reasons.

If your model is either CO or C1 discontinuous, gradient-free methods might make sense for you. CO discontinuities mean that there are jumps in the design space. These might be caused by if-then conditions or discrete variables in your model or something else. C1 discontinuities mean that the derivative space is not smooth and continuous.

For example, what are the derivatives for a wind turbine having two or three blades? 2.2 or 2.9 blades are not an option, so that's inherently introduces a discontinuity. The derivative doesn't exist for discrete variables.

### When to use gradient-free optimizers???

Multimodal problems

# (YOU DON 'T KNOW a priori)

Gradient-free algorithms don't automatically solve multimodal problems better than gradient-based ones.

<u>DIRECT</u>, <u>ISRES</u>, <u>particle swarm</u>, and <u>evolutionary</u> methods.

Nelder-Mead and COBYLA are local algorithms in that they are not made to explore the global design space. I suggest COBYLA as default optimizer

#### https://github.com/relf/cobyla

class COBYLA(maxiter=1000, disp=False, rhobeg=1.0, tol=None)

Constrained Optimization By Linear Approximation optimizer.

COBYLA is a numerical optimization method for constrained problems where the derivative of the objective function is not known.

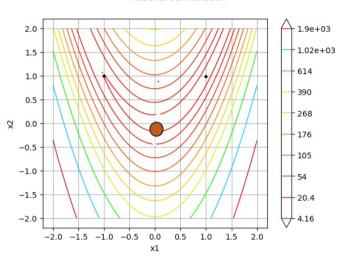
Uses scipy.optimize.minimize COBYLA. For further detail, please refer

to https://docs.scipy.org/doc/scipy/reference/generated/scipy.optimize.minimize.html

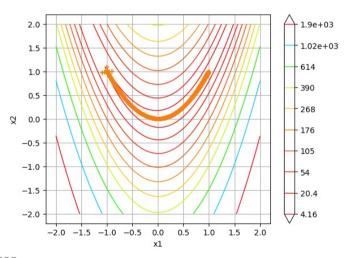
#### Parameters

- •maxiter (int) Maximum number of function evaluations.
- •disp (bool) Set to True to print convergence messages.
- •rhobeg (float) Reasonable initial changes to the variables.
- •tol (Optional[float]) Final accuracy in the optimization (not precisely guaranteed). This is a lower bound on the size of the trust region.

#### Rosenbrock function



#### Rosenbrock function solved with Cobyla



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38