

Due Wednesday 04/14/2021 (midnight)

1. Consider the mass-spring-damper system described by

$$m \ddot{y} + \beta \dot{y} + k y = u$$

- (a) If  $y(t)$  and  $u(t)$  are available for measurement, design a gradient algorithm to estimate constant but unknown parameters  $m$ ,  $\beta$ , and  $k$ .
- (b) Simulate your algorithm in (a) assuming that true values are  $m = 20$ ,  $\beta = 0.1$ , and  $k = 5$ . Repeat your simulation for different choices of  $u(t)$  and observe the resulting parameter convergence properties.

2. Consider the reference model

$$\dot{y}_m = -a y_m + r(t), \quad a > 0$$

and the plant

$$\dot{y} = a^* y + b^* u, \quad b^* \neq 0.$$

- (a) Show that a controller of the form  $u = \theta_1 y + \theta_2 r(t)$  with an appropriate choice of gains  $\theta_1^*$  and  $\theta_2^*$ , drives the tracking error  $e := y - y_m$  asymptotically to zero.
- (b) Now suppose  $a^*$  and  $b^*$  are unknown parameters, but the sign of  $b^*$  is known. Show that the adaptive implementation of the controller above achieves tracking when the gains are updated according to the rule

$$\dot{\theta}_1 = -\text{sign}(b^*) \gamma_1 y e, \quad \dot{\theta}_2 = -\text{sign}(b^*) \gamma_2 r e$$

where  $\gamma_1 > 0$  and  $\gamma_2 > 0$ .

- (c) Provide a condition that also guarantees  $\theta_1(t) \rightarrow \theta_1^*$  and  $\theta_2(t) \rightarrow \theta_2^*$  as  $t \rightarrow \infty$ .

3. A simplified model of an axial compressor, used in jet engine control studies, is given by the following second order system

$$\begin{aligned} \dot{\phi} &= -\frac{3}{2}\phi^2 - \frac{1}{2}\phi^3 - \psi \\ \dot{\psi} &= \frac{1}{\beta^2}(\phi + 1 - u). \end{aligned}$$

This model captures the main surge instability between the mass flow and the pressure rise. Here,  $\phi$  and  $\psi$  are deviations of the mass flow and the pressure rise from their set points, the control input  $u$  is the flow through the throttle, and  $\beta$  is a positive constant.

- (a) Use backstepping to obtain a control law that stabilizes the origin  $(\phi, \psi) = 0$ .
- (b) Use Sontag's Formula and the Control Lyapunov Function obtained in part (a) to obtain an alternative control law.