

Lecture 26

05/03/2021

Last time: input-output linearization

zero-dynamic

Today:

Normal form

Recap :

$$\dot{x} = f(x) + g(x) u$$

$$y = h(x)$$

$$\begin{aligned} y = h(x) \rightarrow \dot{y} &= \cancel{\frac{\partial h}{\partial x} \dot{x}} = \cancel{\frac{\partial h}{\partial x}} (f(x) + g(x) u) \\ &= \underbrace{\cancel{\frac{\partial h}{\partial x}} f(x)}_{L_f h(x)} + \underbrace{\cancel{\frac{\partial h}{\partial x}} g(x) u}_{L_g h(x)} \end{aligned}$$

$$Lgh(x) \neq 0 \Rightarrow r.d. = 1$$

if not continue to differentiate

→ example of a system that doesn't have globally defined relative degree :

$$\dot{x}_1 = x_2 + x_3^3$$

$$\dot{x}_2 = x_3$$

$$\dot{x}_3 = u$$

$$y = x_1$$

→

$$\dot{y} = \dot{x}_1 = x_2 + x_3^3$$

$$\ddot{y} = \dot{x}_2 + 2x_3^2 \dot{x}_3 = x_3 + 3x_3^2 u$$

Vanishes when $x_3 = 0$ ↵

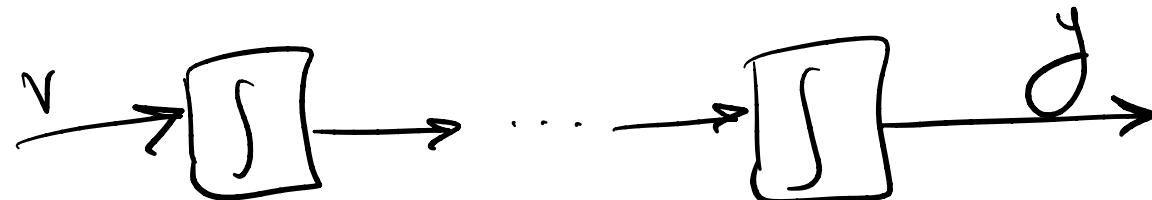
this system doesn't have a well-defined relative degree

when $x_3=0$ we would not be able to determine
the influence of u on \tilde{y} or the derivative of x_1 .

recall:

$$\tilde{y}^{(r)} = L_g^r h(x) + \underbrace{L_g L_f^{r-1} h(x) \cdot u}_{\neq 0}$$

$$u = \frac{1}{L_g L_f^{r-1} h(x)} \left\{ -L_f^r h(x) + v \right\}$$



One of the applications for this type of nonlinear control is output tracking, but it does show limitations in ideal conditions and there may be robustness issues when nonlinearities are not well characterized.

reminder

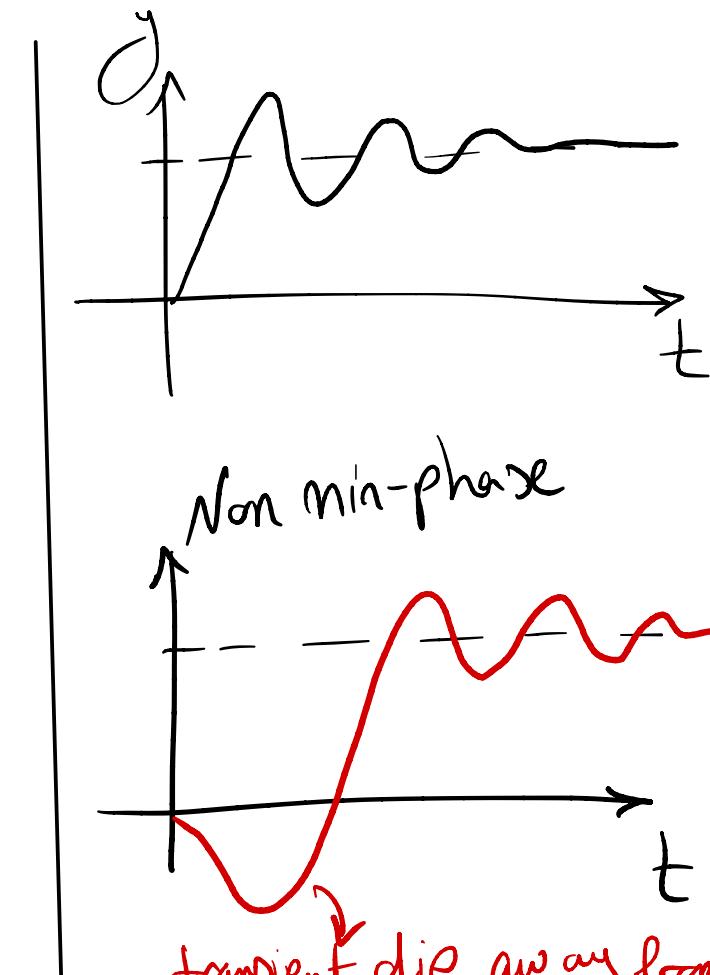
$$H_1(s) = s + 1 \quad \text{and} \quad H_2(s) = s - 1$$

the system with RHP zeros is non-minimum phase.

Ex.

$$G(s) = \frac{s-1}{s^2 + 3s + 6}$$

$$\text{then } \ddot{y} + 3\dot{y} + 6y = u - u$$



transient dip away from
desired output

Typically, a part of a non-minimum phase system
will not be controlled with reference tracking scheme
and would therefore require a form of feedback.

BIG PICTURE

zero-dynamics generalizes

RHP zeros to nonlinear dynamics

Normal Form : a set of coordinates that clearly displays the zero dynamics.

Consider the nonlinear system given by

$$\begin{cases} \dot{x} = f(x) + g(x) u \\ y = h(x) \end{cases}$$

if this system has relative degree r ($r < n$) then there are variables

$$\left\{ \begin{array}{l} z \in \mathbb{R}^{n-1} \\ \xi \in \mathbb{R}^r \end{array} \right.$$

such that there is mapping

$$T: \mathbb{R}^n \rightarrow \mathbb{R}^n \quad \text{st. } T(x) = \begin{bmatrix} z \\ \xi \end{bmatrix} \text{ where } T(0) = 0.$$

The mapping T is called a diffeomorphism

(i.e., its inverse T^{-1} exists where T and T^{-1} are continuous and differentiable)

The dynamics in the new coordinates are given by:

$$\dot{z} = f_0(z, \xi)$$

$$\begin{pmatrix} \dot{\xi}_1 \\ \dot{\xi}_2 \\ \vdots \\ \dot{\xi}_r \end{pmatrix} = \begin{pmatrix} \xi_2 \\ \xi_3 \\ \vdots \\ b(z, \xi) + a(z, \xi)u \end{pmatrix}$$

Note: for ξ -dynamics, there is a cascade connection
 therefore, we can find u st. we follow a
 a desired ξ trajectory

Note: ξ 's are the variables that represent the
 output and its derivatives.

$$b(z, \xi) = L_f^r h(x)$$

$$a(z, \xi) = L_g L_f^{r-1} h(x)$$

$$\xi = \begin{bmatrix} h(x) \\ L_f h(x) \\ \vdots \\ L_f^{r-1} h(x) \end{bmatrix}$$

Two requirements:

(1) ξ and z must be linearly independent

(2) Control u must not be in the equation for z .

As $r < n$, we could find an input/output linearizing controller (if $a(z, \xi)$ is welldefined & invertible!)

$$u = \frac{1}{a(z, \xi)} (-b(z, \xi) + v)$$

where $v = -K_1 \xi_1 - \dots - K_r \xi_r$. This will allow us to achieve the form:

$$\dot{\xi} = A_{\xi} \xi \quad \text{with} \quad A_{\xi} = \begin{bmatrix} 0 & 1 & 0 & \cdots & 0 \\ 0 & 0 & 1 & & \\ \vdots & \ddots & \ddots & \ddots & \\ \vdots & & & & 0 & 1 \\ -K_1 & -K_2 & \cdots & -K_r & & \end{bmatrix}$$

Stability properties of I/O feedback controller $\in \mathbb{R}^{r \times r}$

are determined completely by the choice of K_i 's.

Finally, we call the form

$$\dot{\xi} = A \xi \xi$$

$$\dot{z} = f_0(z, \xi)$$

the **normal form**.

The **zero-dynamics** are recovered by plugging
 $\xi = 0$ into the normal form.

Why $\zeta=0$? linearization gives local stability properties of

$$\begin{bmatrix} \dot{\zeta} \\ \dot{z} \end{bmatrix} = \begin{bmatrix} A_\zeta \\ \cancel{\frac{\partial f}{\partial z}} \Big|_{\zeta=0} \end{bmatrix} \begin{bmatrix} \zeta \\ z \end{bmatrix}$$

Cascade connection:

$\zeta=0$ means that stability properties are captured by Jacobian $\cancel{\frac{\partial f}{\partial z}} \Big|_{\zeta=0}$

Ex. Consider the system: $\dot{x}_1 = x_2$ with $y = x_1$,
 $\dot{x}_2 = \alpha x_3 + u$
 $\dot{x}_3 = \beta x_3 - u$

① Differentiate output to find relative degree.

$$y = x_1 \rightarrow \dot{y} = \dot{x}_1 = x_2 \rightarrow \ddot{y} = \ddot{x}_2 = \alpha x_3 + u$$

relative degree r.d. = r = 2 < 3 = n

part of the system will remain after I/O
linearization.

$$\textcircled{2} \quad \text{Therefore, } \dot{\xi}_1 = x_1 \Rightarrow \dot{\xi}_1 = \xi_2 \\ \xi_2 = x_2 \quad \dot{\xi}_2 = \alpha x_3 + u$$

~~Note!~~ Note! Cannot choose $x_3 = z$ otherwise we would have input in our equation for z !

Also, $T(x)$ would be trivially $x = I(\xi_z)$.

\textcircled{3} Instead, define $z = x_2 + x_3$ (to cancel input in z)

$$\dot{z} = \dot{x}_2 + \dot{x}_3 = (\alpha + \beta) x_3 = (\alpha + \beta) z - (\alpha + \beta) \xi_2$$

Therefore,

Normal form

$$\left\{ \begin{array}{l} \dot{\zeta}_1 = \zeta_2 \\ \dot{\zeta}_2 = \alpha(z - \zeta_2) + u \\ z = (\alpha + \beta)z - (\alpha + \beta)\zeta_2 \end{array} \right.$$

and the zero dynamics are given by $(\zeta_1, \zeta_2) = (0, 0)$

st. \Rightarrow $\dot{z} = (\alpha + \beta)z$

Stability of z will be completely determined by $\text{sgn}(\alpha + \beta)$.

Transfer function from u to y :

$$\frac{Y(s)}{U(s)} = \frac{s - (\alpha + \beta)}{s^2(s - \beta)}$$

↑ numerator determines zeros of
the T.F.

Therefore, for the purpose of reference tracking if we

have zeros of the TF. in the RHP then we will

encounter problems of stability in aerodynamics

and we cannot guarantee stability of nonlinear dynamics.

Summary :

Nonlinear system \rightarrow r. d.



linearizing
controller u \leftarrow Normal form

$$\begin{bmatrix} \dot{\xi} \\ \dot{z} \end{bmatrix} = \begin{bmatrix} A_\xi \\ \frac{\partial f}{\partial z} \end{bmatrix} \begin{bmatrix} \xi \\ z \end{bmatrix}$$

$$\begin{bmatrix} 0 \\ \frac{\partial f}{\partial z} \Big|_{\xi=0} \end{bmatrix} \begin{bmatrix} \xi \\ z \end{bmatrix}$$

the stability (LAS)
of the normal form

and original nonlinear dynamics is equivalent to $\frac{\partial f}{\partial z}$
and A_ξ being Hurwitz.

Obviously, A_g will be Hurwitz b/c of
our choice of linearizing controller u (and param. v)

So the LAS of nonlinear system essentially
amounts to the stability of the zero-dynamics

$$\frac{\partial f}{\partial z}|_0 \quad \downarrow$$

Note If you have input to state stability you will have
GAS as well.