

**EE363 Automatic Control: Homework #7**

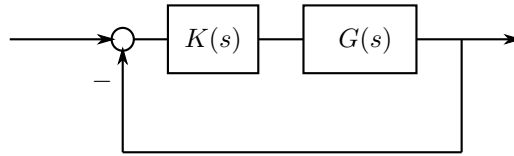
- 1) *Bode plots.* Sketch the Bode magnitude and phase plots, not by using computers, to the accuracy of the asymptotes.

a)  $G(s) = \frac{2000(s + 0.5)}{s(s + 10)(s + 50)}$

b)  $G(s) = \frac{1000(s + 1)}{s(s + 2)(s^2 + 8s + 64)}$

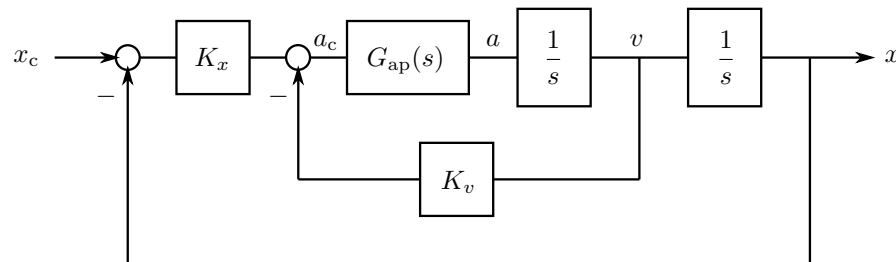
c)  $G(s) = \frac{4s(s + 10)}{(s + 50)(4s^2 + 5s + 4)}$

- 2) *Lead/lag compensation.* Consider the following system in the unity feedback configuration.



$$G(s) = \frac{s + 4}{s^2}$$

- a) Design a simple lead or lag compensator,  $K(s)$ , that will place the dominant poles of the closed-loop system at  $s = -2 \pm 2j$ . What is your  $K(s)$ ? Hint. your compensator may cancel the plant zero.
- b) Draw the Bode diagrams of  $K(s)G(s)$ . You may use MATLAB.
- c) What is the gain margin and the phase margin of your design?
- 3) *Runway approach problem.* Recall the runway approach problem for an aircraft, which we discussed several times in class. Your job is to design a controller that computes the acceleration command,  $a_c$ , from the lateral deviation,  $x$ , and the lateral velocity,  $v$ , that is, to choose  $K_v$  and  $K_x$ , and to check the robustness of your design. The block diagram describing the dynamics of the considered system is shown below.



Your controller computes the acceleration command,  $a_c$ , which is sent to the autopilot that somehow generates the actual acceleration response,  $a$ , through  $G_{ap}(s) = a(s)/a_c(s)$ . For now, assume that the autopilot is ideal,  $G_{ap}(s) = 1$ .

- a) Find  $K_v$  and  $K_x$  that place the closed loop pole at  $s = -1 \pm j$ , so that the closed loop bandwidth is 2 and the closed loop damping is  $1/\sqrt{2}$ .

Now fix  $K_v$  and  $K_x$  by the ones you found in a), and we assume that  $K_x$  is disturbed by a scale factor  $\xi > 0$ , so the new position gain can be  $\xi K_x$  while the velocity gain stays the same as  $K_v$ .

- b) For what range of  $\xi$ , is the closed loop system stable? You may use MATLAB to check the stability margin of your design.

A more realistic autopilot can be modelled by a third order system as

$$G_{\text{ap}}(s) = \frac{a(s)}{a_c(s)} = \frac{p\omega^2}{(s+p)(s^2 + 2\zeta\omega s + \omega^2)}$$

where we let  $\omega = 4$ ,  $\zeta = 0.7$ , and  $p = 6$ .

- c) Under presence of this autopilot model, for what range of  $\xi$ , is the closed loop system stable? You will need to use MATLAB to check the stability margin of your design.