Problem 2

```
interr = 'latex';
% interr = 'none';
set(groot, 'defaulttextinterpreter', interr);
set(groot, 'defaultAxesTickLabelInterpreter',interr);
set(groot, 'defaultLegendInterpreter',interr);
R E = 6378.1363;
R J = 71492.0;
R P = 1162.0;
mu E = 398600.4415;
mu J = 126712767.8578;
mu P = 981.600887707;
mu S = 132712440017.99;
a E = 149597898.0;
a J = 778279959.0;
a P = 5907150229.0;
AU = 1.496e + 8;
```

Given

```
v_E_inf_mag = 12.8;
v_E_mag = sqrt(mu_S/a_E)
v_E_mag = 29.7847
v_J_mag = sqrt(mu_S/a_J)
v_J_mag = 13.0583
v_P_mag = sqrt(mu_S/a_P)
v_P_mag = 4.7399
```

Part a)

Transfer Orbit Before Jupiter Encounter (Heliocentric View)

```
rp_T_mag = a_E + 250, r1_mag = rp_T_mag;

rp_T_mag = 149598148

vp_T_mag = v_E_inf_mag + v_E_mag

vp_T_mag = 42.5847

a_T = -mu_S * (vp_T_mag^2 - 2*mu_S/rp_T_mag)^-1

a_T = -3.3852e+09

e_T = 1 - rp_T_mag/a_T

e_T = 1.0442
```

Earth Parking and Hyperbolic Orbits (Geocentric View)

```
rp_E = R_E + 250
```

 $rp_E = 6.6281e + 03$

 $vp_E_0_mag = 7.7548$

vp E n mag = 16.8557

dv1 mag = 9.1009

Jupiter Encounter Before Assist (Heliocentric View)

$$r_2_mag = a_J$$

 $r_2_mag = 778279959$

ths_2_0 =
$$acos(1/e_T * ((a_T*(1-e_T^2)/r_2 mag)-1))$$

 $ths_2_0 = 2.1912$

 $ths_2_0_deg = 125.5477$

ths
$$1 \ 0 = 0$$

ths_1_0 = 0

$$H 1 0 = conicanom(ths 1 0, e T)$$

H 1 0 = 0

$$H_2_0 = conicanom(ths_2_0, e_T)$$

 $H_2_0 = 0.5879$

TOF E J = 3.3500e+07

$$TOF_E_J_day = TOF_E_J/3600/24$$

 $TOF_E_J_day = 387.7281$

Part b)

startJD = 2453754.5000000

startJD = 2.4538e+06

endJD = 2454159.5000000

```
endJD = 2.4542e + 06
 TOF E J actual day = endJD - startJD
 TOF_E_J_actual_day = 405
 TOF E J actual sec = TOF E J actual day * 3600 * 24
 TOF E J actual sec = 34992000
Part c)
 H 2 0 = linspace(.4, .7, 2^14);
 for i = 1:length(H 2 0)
      TOF test(i) = sqrt(abs(a T)^3/mu S) * (e T*sinh(H 2 O(i)) - H 2 O(i));
 end
 diff TOF = abs(TOF test - TOF E J actual sec);
 ind = find(min(diff TOF) == diff TOF)
 ind = 10907
 H 2 0 = H 2 0 (ind), ths 2 0 = trueanom(H 2 0,e T), ths 2 0 deg = rad2deg(ths 2 0)
 H 2 0 = 0.5997
 ths 2 0 = 2.2064
 ths_2_0_deg = 126.4170
 r 2 0 mag = a T*(1-e T^2)/(1+e T * cos(ths 2 0))
 r 2 0 mag = 8.0453e + 08
 iCr 2 0 = [cos(ths 2 0) - sin(ths 2 0); sin(ths 2 0) cos(ths 2 0)]
 iCr_2_0 = 2 \times 2
   -0.5937 -0.8047
```

```
-0.5937 -0.8047
0.8047 -0.5937
```

```
r_2_0 = iCr_2_0 * [r_2_0_mag 0]'
```

```
r_2_0 = 2×1
10<sup>8</sup> ×
-4.7761
6.4742
```

```
p_T = a_T * (1-e_T^2)
```

 $p_T = 3.0581e + 08$

```
v_2_0_mag = sqrt(mu_S*(2/r_2_0_mag - 1/a_T))
```

v 2 0 mag = 19.2124

```
gamma_2_0 = acos(sqrt(mu_S*p_T)/r_2_0_mag/v_2_0_mag)
```

 $gamma_2_0 = 1.1460$

```
gamma_2_0_deg = 65.6600
 v_2 0 R = v_2 0 mag * [sin(gamma_2 0); cos(gamma_2 0)]
 v_2_0_R = 2 \times 1
    17.5047
     7.9184
 v 2 0 = iCr 2 0 * v 2 0 R
 v_2_0 = 2 \times 1
   -16.7639
     9.3855
 v J mag = sqrt((mu S + mu J)/r 2 0 mag)
 v_{J_mag} = 12.8497
 v J = iCr 2 0 * [0; v J mag]
 v_J = 2 \times 1
   -10.3404
    -7.6283
 v_{inf_J_0} = v_2_0 - v_J
 v_inf_J_0 = 2 \times 1
    -6.4235
    17.0139
 v inf J 0 mag = norm(v inf J 0)
 v_{inf_J_0_mag} = 18.1861
 v_inf_J_0 R = iCr_2_0.' * v_inf_J_0
 v_inf_J_0_R = 2 \times 1
    17.5047
    -4.9313
Part d)
i)
 rp J = 200*R J
 rp_J = 14298400
 a_hyp_J = -mu_J/v_inf_J_0_mag^2
 a_hyp_J = -3.8313e+05
 e_hyp_J = 1 - rp_J/a_hyp_J
 e_hyp_J = 38.3203
```

gamma 2 0 deg = rad2deg(gamma 2 0)

```
delta = 2*(asin(1/e hyp J)), delta deg = rad2deg(delta)
delta = 0.0522
delta deg = 2.9907
ths inf = acos(-1/e hyp J), ths inf deg = rad2deg(ths inf)
ths inf = 1.5969
ths_inf_deg = 91.4954
eta = a\sin(v \ 2 \ 0 \ mag/v \ inf \ J \ 0 \ mag * \sin(gamma \ 2 \ 0)), eta \ deg = rad2deg(eta)
eta = 1.2962
eta_deg = 74.2669
zeta = asin(v_J_mag/v_inf_J_0_mag * sin(gamma_2_0)), zeta_deg = rad2deg(zeta)
zeta = 0.6994
zeta deg = 40.0731
% eta = pi - (pi-zeta) - gamma 2 0
v inf J n mag = v inf J 0 mag
v_{inf_J_n_mag} = 18.1861
v_2 n mag = sqrt(v_J mag^2 + v_inf_J n mag^2 - 2*v_J mag*v_inf_J n mag*cos(eta+delta))
v_2_n = 19.8182
gamma 2 n = asin(v inf J n mag/v 2 n mag * sin(eta+delta)), gamma 2 n deg = rad2deg(gar)
gamma_2_n = 1.1085
gamma_2_n_deg = 63.5142
r 2 n = r 2 0
r 2 n = 2 \times 1
10<sup>8</sup> ×
  -4.7761
   6.4742
r 2 n mag = norm(r 2 n)
r_2_n = 8.0453e + 08
v 2 n R = v 2 n mag * [sin(gamma 2 n); cos(gamma 2 n)]
v_2_n_R = 2 \times 1
  17.7382
   8.8384
ths 2 n = atan((r 2 n mag * v 2 n mag^2 / mu S)*sin(gamma 2 n)*cos(gamma 2 n)...
    /((r_2 n_3 mag * v_2 n_3 mag^2 / mu_s)*cos(gamma_2 n)^2-1))+pi
ths 2 n = 2.0766
```

```
ths 2 \text{ n deg} = \text{rad2deg(ths } 2 \text{ n)}
ths_2_n_deg = 118.9821
v 2 n = iCr 2 0 * v 2 n R
v_2_n = 2 \times 1
  -17.6428
   9.0272
dv_eq = v_2_n - v_2_0
dv_eq = 2 \times 1
   -0.8789
   -0.3583
dv eq mag = norm(dv eq)
dv_eq_mag = 0.9492
dgamma = gamma 2 n - gamma 2 0 , dgamma deg = rad2deg(dgamma)
dgamma = -0.0375
dgamma_deg = -2.1457
beta = a\sin(v + 2 - n - mag/dv + eq - mag + sin(dgamma)); beta = [beta pi-beta]; beta(2) = -(2*pi-sin(dgamma))
beta = 1 \times 2
   -0.8975
           -2.2441
beta deg = rad2deg(beta)
beta_deg = 1 \times 2
  -51.4222 -128.5778
a T n = -mu S * (v_2 n mag^2 - 2*mu S/r_2 n mag)^-1
a_T_n = -2.1117e+09
h 2 n = cross([r 2 n; 0], [v 2 n; 0]), h 2 n mag = norm(h 2 n)
h \ 2 \ n = 3 \times 1
10<sup>9</sup> ×
        0
   7.1108
h_2_n = 7.1108e + 09
p_T n = h_2 n_mag^2/mu S
p_T_n = 3.8100e + 08
e T n = sqrt(1-p T n/a T n)
e_T_n = 1.0865
H 2 n = conicanom(ths 2 n, e T n), H 2 n deg = rad2deg(H 2 n)
```

```
H 2 n = 0.7206
 H 2 n deg = 41.2883
 N 2 n = e T n * sinh(H 2 n) - H 2 n, N 2 n deg = rad2deg(N 2 n)
 N 2 n = 0.1319
 N_2_n_{deg} = 7.5548
 ra = a T n*(1+e T n)/a E
 ra = -29.4520
Plotting
 plotorbit(a E,0,0,2*pi,0) % earth
 hold on
 plotorbit(a_J,0,0,2*pi,0) % jupiter
 plotorbit(a_P,0,0,2*pi,0) % pluto
 plot(0,0,'r*','MarkerSize',2)
 plotorbit(a_T,e_T,0,ths_2_0,0)
 maxlim = [-a P a P]*1.25;
 xlim(maxlim), ylim(maxlim)
 dw = ths 2 n-ths 2 0;
 plotorbit (a T n, e T n, ths 2 n, acos (-1/e T n), dw)
Part ii)
 rp J = 32*R J
 rp J = 2287744
 a hyp J = -mu J/v inf J 0 mag^2
 a_hyp_J = -3.8313e+05
```

rp_J = 32*R_J rp_J = 2287744 a_hyp_J = -mu_J/v_inf_J_0_mag^2 a_hyp_J = -3.8313e+05 e_hyp_J = 6.9712 delta = 2*asin(1/e_hyp_J), delta_deg = rad2deg(delta) delta = 0.2879 delta_deg = 16.4947 ths_inf = acos(-1/e_hyp_J), ths_inf_deg = rad2deg(ths_inf) ths_inf = 1.7147 ths_inf_deg = 98.2473 eta = asin(v_2_0_mag/v_inf_J_0_mag * sin(gamma_2_0)), eta_deg = rad2deg(eta) eta = 1.2962 eta_deg = 74.2669 zeta = asin(v_J_mag/v_inf_J_0_mag * sin(gamma_2_0)), zeta_deg = rad2deg(zeta)

zeta = 0.6994 zeta_deg = 40.0731

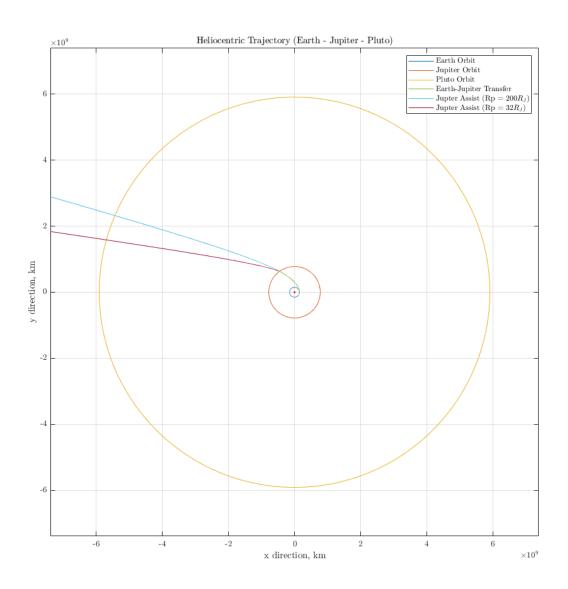
```
% eta = pi - (pi-zeta) - gamma 2 0
v inf J n mag = v inf J 0 mag
v_{inf_J_n_mag} = 18.1861
v 2 n mag = sqrt(v J mag^2 + v inf J n mag^2 - 2*v J mag*v inf J n mag*cos(eta+delta))
v_2_n = 22.4067
gamma 2 n = asin(v inf J n mag/v 2 n mag * sin(eta+delta)), gamma 2 n deg = rad2deg(gar
gamma_2_n = 0.9468
gamma_2_n_deg = 54.2490
r 2 n = r 2 0
r_2_n = 2 \times 1
10^8 \times
  -4.7761
   6.4742
r 2 n mag = norm(r 2 n)
r 2 n mag = 8.0453e+08
v 2 n R = v 2 n mag * [sin(gamma 2 n); cos(gamma 2 n)]
v 2 n R = 2 \times 1
  18.1845
  13.0914
ths 2 n = atan((r 2 n mag * v 2 n mag^2 / mu S)*sin(gamma 2 n)*cos(gamma 2 n)...
    /((r \ 2 \ n \ mag * v \ 2 \ n \ mag^2 \ / \ mu \ S) *cos(gamma \ 2 \ n)^2-1))
ths_2_n = 1.5438
ths 2 n deg = rad2deg(ths 2 n)
ths 2 n deg = 88.4533
v 2 n = iCr 2 0 * v 2 n R
v 2 n = 2 \times 1
  -21.3302
   6.8615
dv eq = v 2 n - v 2 0
dv eq = 2 \times 1
  -4.5663
  -2.5240
dv eq mag = norm(dv eq)
dv_eq_mag = 5.2175
```

```
dgamma = gamma 2 n - gamma 2 0 , dgamma deg = rad2deg(dgamma)
dgamma = -0.1992
dgamma_deg = -11.4109
beta = a\sin(v + 2 n mag/dv eq mag * sin(dgamma)); beta = [beta pi-beta]; beta(2) = -(2*psi)
beta = 1 \times 2
  -1.0153
          -2.1263
beta deg = rad2deg(beta)
beta deg = 1 \times 2
  -58.1742 -121.8258
a T_n = -mu_S * (v_2 n_mag^2 - 2*mu_S/r_2 n_mag)^-1
a_T_n = -7.7093e + 08
h 2 n = cross([r 2 n; 0], [v 2 n; 0]), h 2 n mag = norm(h 2 n)
h \ 2 \ n = 3 \times 1
10^{10} \times
        0
        0
   1.0532
h_2_n = 1.0532e + 10
p T n = h 2 n mag^2/mu S
p T n = 8.3588e + 08
e T n = sqrt(1-p T n/a T n)
e_T_n = 1.4437
H 2 n = conicanom(ths 2 n, e T n), H 2 n deg = rad2deg(H 2 n)
H_2_n = 0.8827
H_2_n_{deg} = 50.5740
N 2 n = e T n * sinh(H 2 n) - H 2 n, N 2 n deg = rad2deg(N 2 n)
N 2 n = 0.5637
N_2_n_{deg} = 32.2967
ra = a T n*(1+e T n)/a E
ra = -12.5932
```

Plotting

```
dw = ths_2_n - ths_2_0;
plotorbit(a_T_n,e_T_n,ths_2_n,acos(-1/e_T_n),dw)
maxlim = [-a_P a_P]*1.25;
xlim(maxlim), ylim(maxlim)
title('Heliocentric Trajectory (Earth - Jupiter - Pluto)')
xlabel('x direction, km')
ylabel('y direction, km')
```

```
legend('Earth Orbit', 'Jupiter Orbit', 'Pluto Orbit', '', 'Earth-Jupiter Transfer', 'Jupter
hold off
set(gcf, 'position', [0,0,1000,1000])
```



Function 1: Plotting Orbit

```
function plotorbit(ai,ei,ths1,ths2,rotate)
ths_plot = linspace(ths1,ths2,2^12)';
ri = (ai*(1-ei^2))./(1+ei*cos(ths_plot));
ri = ri .* [cos(ths_plot-rotate),sin(ths_plot-rotate)];
plot(ri(:,1),ri(:,2))
grid on
axis equal
end
```