Problem 2

```
Re = 6378.1;
 mu = 398600.4415;
  r1 = Re * [-5.025, -0.087, -2.06]'
 r1 = 3 \times 1
 10<sup>4</sup> ×
    -3.2050
    -0.0555
    -1.3139
  v1 = [-0.2186, -2.904, -1.483]'
 v1 = 3 \times 1
    -0.2186
    -2.9040
    -1.4830
Part a)
  r1mag = norm(r1)
  r1mag = 3.4643e+04
 v1mag = norm(v1)
 v1mag = 3.2681
 a1 = -mu/(2*(v1mag^2/2 - mu/r1mag))
  a1 = 3.2324e + 04
 h1 = cross(r1, v1)
 h1 = 3 \times 1
  10<sup>4</sup> ×
    -3.7332
    -4.4658
    9.2952
  h1mag = norm(h1)
 h1mag = 1.0967e+05
 p1 = h1mag^2/mu
  p1 = 3.0176e + 04
  e1 = sqrt(1-p1/a1)
  e1 = 0.2578
  ths1 = acos(1/e1*(p1/r1mag - 1))
  ths1 = 2.0947
  ths1 deg = rad2deg(ths1)
```

```
ths1_deg = 120.0169
h1hat = h1/h1mag
h1hat = 3 \times 1
  -0.3404
  -0.4072
   0.8475
r1hat = r1/r1mag
r1hat = 3 \times 1
  -0.9251
  -0.0160
  -0.3793
ths1hat = cross(h1hat, r1hat)
ths1hat = 3 \times 1
   0.1680
  -0.9132
  -0.3713
% 3D Angles
iCr1 = [r1hat, ths1hat, h1hat]
iCr1 = 3 \times 3
  -0.9251
            0.1680
                    -0.3404
  -0.0160
            -0.9132
                     -0.4072
            -0.3713
                      0.8475
  -0.3793
i1 = acos(h1hat(3)), i1 deg = rad2deg(i1)
i1 = 0.5595
i1_{deg} = 32.0550
om1 = asin(h1hat(1)/sin(i1)); om1 = [om1 pi-om1], om1 deg = rad2deg(om1)
om1 = 1 \times 2
  -0.6963
             3.8379
om1_deg = 1 \times 2
 -39.8944 219.8944
om1check = acos(-h1hat(2)/sin(i1)); om1check = om1check*[1 -1], om1check_deg = rad2deg
om1check = 1 \times 2
   0.6963
           -0.6963
om1check_deg = 1 \times 2
  39.8944 -39.8944
om1 = om1(1), om1 deg = rad2deg(om1)
om1 = -0.6963
om1_deg = -39.8944
th1 = asin(r1hat(3)/sin(i1)); th1 = [th1 pi-th1], th1_deg = rad2deg(th1)
th1 = 1 \times 2
  -0.7961
            3.9377
th1_deg = 1 \times 2
 -45.6109 225.6109
```

```
th1check = 1\times2
    2.3455 3.9377
 th1check\_deg = 1 \times 2
  134.3891 225.6109
 th1 = th1(2), th1_deg = rad2deg(th1)
 th1 = 3.9377
 th1_deg = 225.6109
 v1r = iCr1.' * v1
 v1r = 3 \times 1
    0.8112
    3.1658
    0.0000
 r1r = iCr1.' * r1
 r1r = 3 \times 1
 10<sup>4</sup> ×
    3.4643
    0.0000
 % Orbit Characteristics
 E1 = eccenAnom(ths1,e1), E1 deg = rad2deg(E1)
 E1 = 1.8529
 E1_{deg} = 106.1637
 M1 = E1-e1*sin(E1), M1 deg = rad2deg(M1)
 M1 = 1.6053
 M1_{deg} = 91.9784
 w1 = th1 - ths1, w1 deg = rad2deg(w1)
 w1 = 1.8430
 w1_deg = 105.5940
Part b)
 ths2 = pi-w1, ths2 deg = rad2deg(ths2)
 ths2 = 1.2986
 ths2_deg = 74.4060
 n1 = sqrt(mu/a1^3)
 n1 = 1.0864e-04
 E2 = eccenAnom(ths2,e1), E2 deg = rad2deg(E2)
 E2 = 1.0559
 E2_{deg} = 60.4973
```

```
dt1 = orbittime(e1,E1-2*pi,n1)
dt1 = -4.3058e + 04
dt2 = orbittime(e1, E2, n1)
dt2 = 7.6540e + 03
period1 = 2*pi/n1
period1 = 5.7835e+04
dt = dt2-dt1, dt hr = dt/3600
dt = 5.0712e+04
dt_hr = 14.0867
p1 = a1*(1-e1^2)
p1 = 3.0176e + 04
r2mag = p1/(1+e1*cos(ths2))
r2mag = 2.8220e+04
th2 = w1+ths2, th2 deg = rad2deg(th2)
th2 = 3.1416
th2_deg = 180
iCr2 = findDCM(om1, i1, th2)
iCr2 = 3 \times 3
  -0.7672
           -0.5436 -0.3404
           -0.6503
                    -0.4072
   0.6414
   0.0000
           -0.5307
                    0.8475
r2r = r2mag * [1 0 0]'
r2r = 3 \times 1
10<sup>4</sup> ×
   2.8220
r2 = iCr2 * r2r
r2 = 3 \times 1
10^4 \times
  -2.1651
   1.8100
   0.0000
v2mag = sqrt(mu*(2/r2mag - 1/a1))
v2mag = 3.9897
gamma2 = acos(h1mag/r2mag/v2mag), gamma2_deg = rad2deg(gamma2)
```

```
gamma2 = 0.2281
gamma2_deg = 13.0720
v2r = v2mag*[sin(gamma2), cos(gamma2), 0]
v2r = 3 \times 1
   0.9024
   3.8863
        0
v2 = iCr2 * v2r
v2 = 3 \times 1
  -2.8049
   -1.9483
   -2.0626
dv vnb = [0.2, 0.7, 0.3]' % delta v in vnb frame using v2hat as vhat
dv_vnb = 3 \times 1
   0.2000
   0.7000
   0.3000
dvmag = norm(dv vnb)
dvmag = 0.7874
vhat = v2/v2mag
vhat = 3 \times 1
   -0.7030
   -0.4883
   -0.5170
nhat = h1hat
nhat = 3 \times 1
   -0.3404
   -0.4072
   0.8475
bhat = cross(vhat, nhat)
bhat = 3 \times 1
   -0.6244
   0.7718
   0.1200
iCv = [vhat, nhat, bhat]
iCv = 3 \times 3
   -0.7030
           -0.3404 -0.6244
           -0.4072
                    0.7718
   -0.4883
   -0.5170
           0.8475
                    0.1200
dv = iCv * dv_vnb % in xyz
```

```
dv = 3 \times 1
    -0.5662
    -0.1512
     0.5259
 beta = asin(dv_vnb(2)/dvmag); beta = [beta pi-beta], beta_deg = rad2deg(beta)
  beta = 1 \times 2
     1.0952
             2.0464
  beta_deg = 1 \times 2
    62.7480 117.2520
  dv vb = [dv vnb(1), 0, dv vnb(3)]' % in vnb
  dv vb = 3 \times 1
     0.2000
     0.3000
  dv_vb_inertial = iCv * dv_vb % in xyz
  dv vb inertial = 3 \times 1
    -0.3279
     0.1339
    -0.0674
 phi = acos(dot(dv vb inertial,iCr2(:,2))/norm(dv vb))
  phi = 1.2109
  alpha = phi - gamma2, alpha_deg = rad2deg(alpha)
  alpha = 0.9828
  alpha_deg = 56.3099
 beta_check = acos(0.2/(dvmag*cos(alpha)))
 beta_check = 1.0952
  v3 = v2 + dv
  v3 = 3 \times 1
    -3.3711
    -2.0995
    -1.5367
  r3 = r2
 r3 = 3 \times 1
  10<sup>4</sup> ×
    -2.1651
     1.8100
     0.0000
Part c)
  rN = r3, vN = v3
```

```
rN = 3 \times 1
10<sup>4</sup> ×
   -2.1651
   1.8100
   0.0000
vN = 3 \times 1
   -3.3711
   -2.0995
  -1.5367
rNmag = norm(rN)
rNmag = 2.8220e+04
vNmag = norm(vN)
vNmag = 4.2583
aN = -mu/(2*(vNmag^2/2 - mu/rNmag))
aN = 3.9404e + 04
hN = cross(rN, vN)
hN = 3 \times 1
10<sup>5</sup> ×
   -0.2781
   -0.3327
   1.0647
hNmag = norm(hN)
hNmag = 1.1497e + 05
pN = hNmag^2/mu
pN = 3.3159e+04
eN = sqrt(1-pN/aN)
eN = 0.3981
thsN = acos(1/eN*(pN/rNmag -1)), thsN_deg = rad2deg(thsN)
thsN = 1.1157
thsN_deg = 63.9247
rNhat = rN/rNmag
rNhat = 3 \times 1
   -0.7672
   0.6414
   0.0000
hNhat = hN/hNmag
hNhat = 3 \times 1
```

```
-0.2419
  -0.2894
   0.9261
thNhat = cross(hNhat,rNhat)
thNhat = 3 \times 1
  -0.5940
  -0.7106
  -0.3772
iCrN = [rNhat,thNhat,hNhat]
iCrN = 3 \times 3
  -0.7672
          -0.5940 -0.2419
   0.6414
          -0.7106 -0.2894
   0.0000
          -0.3772
                   0.9261
rNr = iCrN.' * rN
rNr = 3 \times 1
10<sup>4</sup> ×
   2.8220
  -0.0000
vNr = iCrN.' * vN
vNr = 3 \times 1
   1.2398
   4.0739
   0.0000
iN = acos(hNhat(3)), iN deg = rad2deg(iN)
iN = 0.3868
iN_deg = 22.1609
omN = asin(hNhat(1)/sin(iN)); omN = [omN pi-omN]
omN = 1 \times 2
  -0.6963
           3.8379
omNcheck = acos(-hNhat(2)/sin(iN)), omN = omN(1), omN deg = rad2deg(omN)
omNcheck = 0.6963
omN = -0.6963
omN deg = -39.8944
thN = asin(rNhat(3)/sin(iN)); thN = [thN pi-thN]
thN = 1 \times 2
   0.0000
           3.1416
thNcheck = acos(thNhat(3)/sin(iN))
thNcheck = 3.1416 - 0.0000i
thN = thN(2), thN deg = rad2deg(thN)
```

```
thN = 3.1416
thN_deg = 180
wN = thN - thsN, wN deg = rad2deg(wN)
wN = 2.0259
wN deg = 116.0753
iCrNcheck = findDCM(omN, iN, thN) *iCrN.'
iCrNcheck = 3 \times 3
   1.0000
          0.0000
                    0.0000
  -0.0000
          1.0000
                    0.0000
  -0.0000
          -0.0000
                    1.0000
% % Orbit Characteristics
% E1 = eccenAnom(ths1,e1), E1 deg = rad2deg(E1)
% M1 = E1-e1*sin(E1), M1 deg = rad2deg(M1)
```

Function 1: Eccentric Anomaly

```
function E = eccenAnom(th,e)
   E = 2*atan(sqrt((1-e)/(1+e))*tan(th/2));
end
```

Function 2: Time Since Periapsis

```
function dt = orbittime(e,E,n)
  dt = (E - e*sin(E))/n;
end
```

Function 3: Find Direction Cosine Matrix

```
function iCr = findDCM(om, inc, th)
    col1 = [cos(om)*cos(th) - sin(om)*cos(inc)*sin(th);
        sin(om)*cos(th)+cos(om)*cos(inc)*sin(th);
        sin(inc)*sin(th)];
응
     col2 = [-cos(om)*sin(th)-sin(om)*cos(inc)*cos(th);
          -\sin(om)*\sin(th)+\cos(om)*\cos(inc)*\cos(th);
응
응
          sin(inc)*cos(th)];
    col3 = [sin(om)*sin(inc);
        -cos(om)*sin(inc);
        cos(inc)];
    col2 = cross(col3, col1);
    iCr = [col1 col2 col3];
end
```