

The Three Body Problem

Arenstorf orbit

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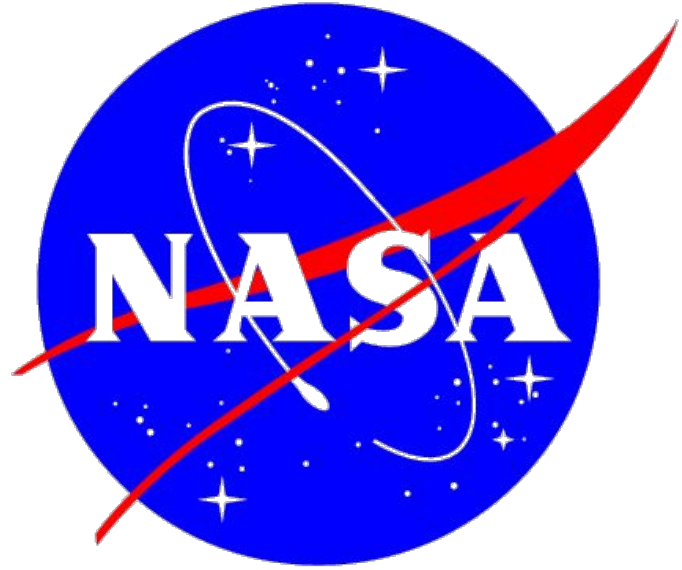


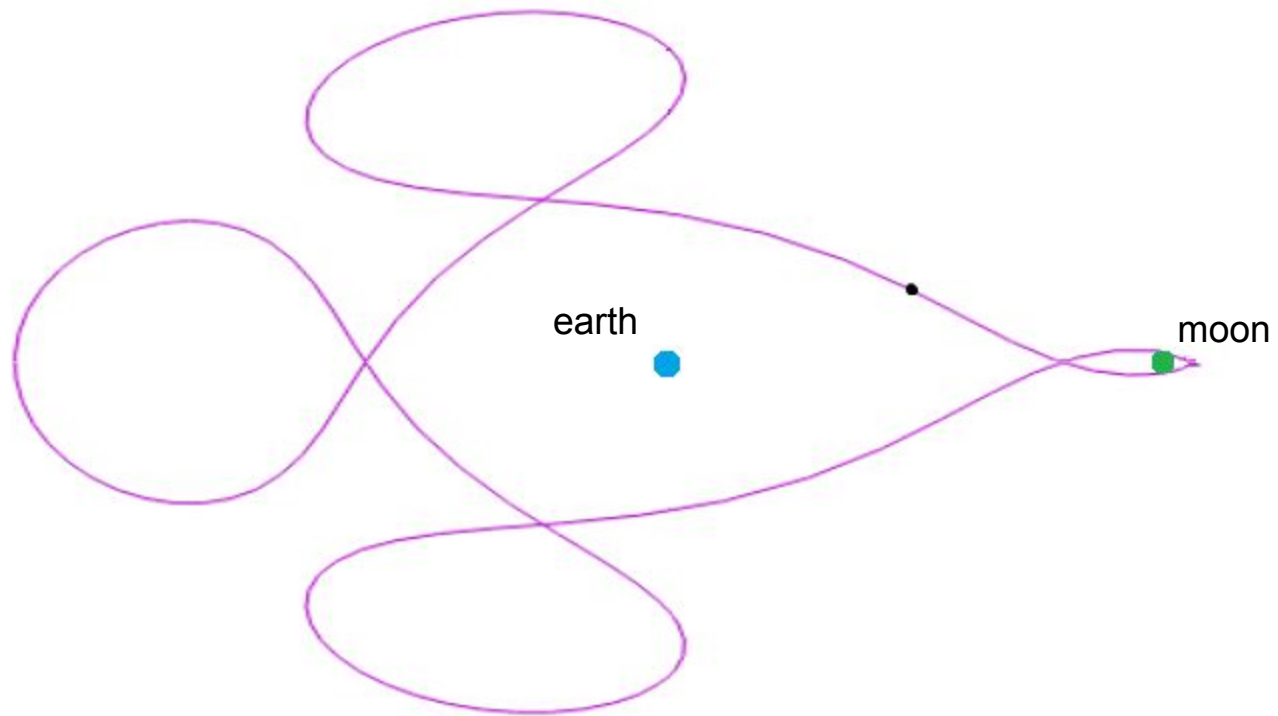
Richard Arenstorf

cuerpo de masa despreciable

M_1 M_2

$\frac{M_1}{M_2} = \mu \quad \mu: (1 - \mu)$





Ecuaciones de movimiento

$$\ddot{x} = x + 2\dot{y} - \frac{(1 - \mu)(x + \mu)}{r^3} - \frac{\mu(x - 1 + \mu)}{s^3},$$

$$\ddot{y} = y - 2\dot{x} - \frac{(1 - \mu)y}{r^3} - \frac{\mu y}{s^3},$$

$$\ddot{z} = -\frac{(1 - \mu)z}{r^3} - \frac{\mu z}{s^3},$$

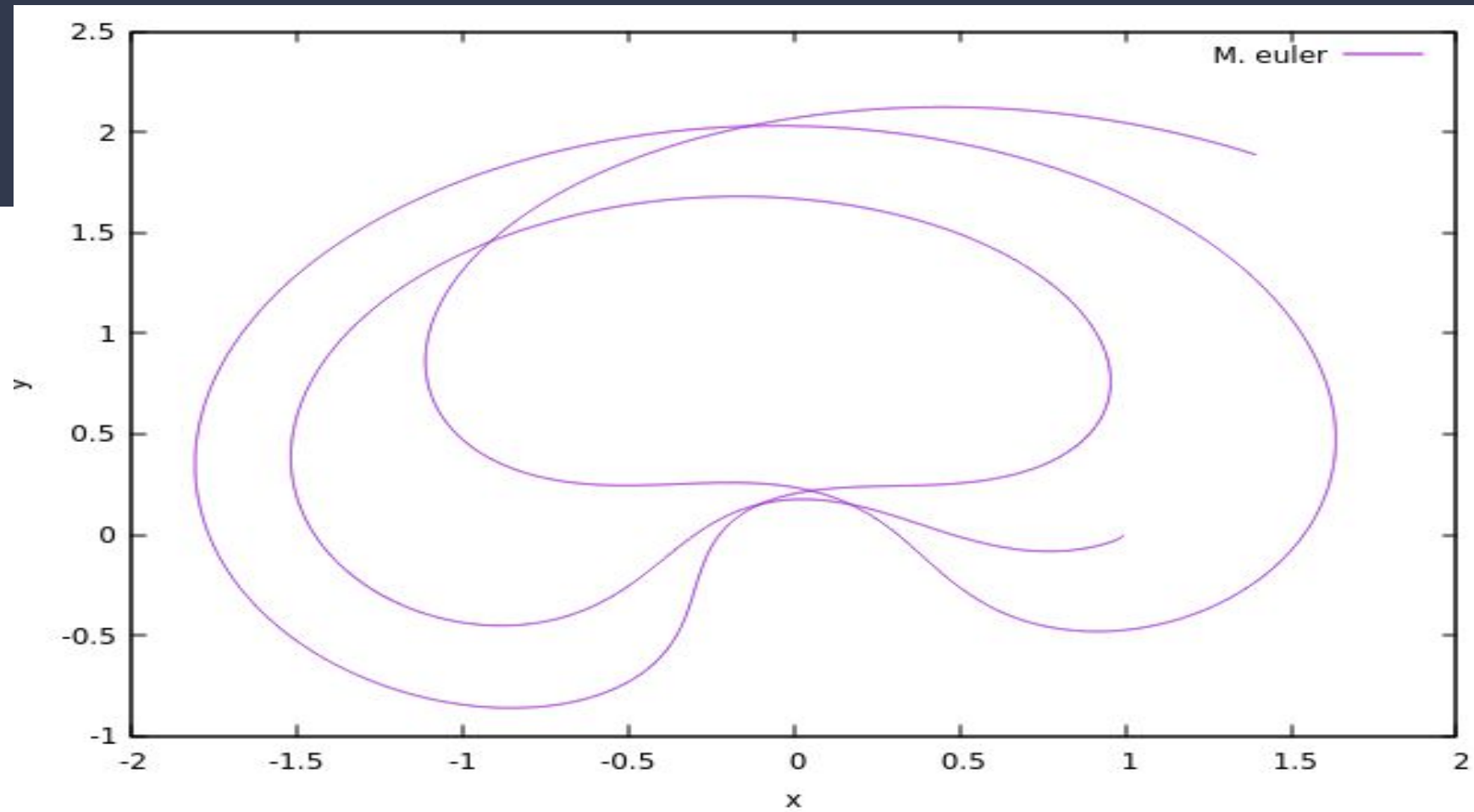
$$r = \sqrt{(x + \mu)^2 + y^2 + z^2}, \quad s = \sqrt{(x - 1 + \mu)^2 + y^2 + z^2}.$$

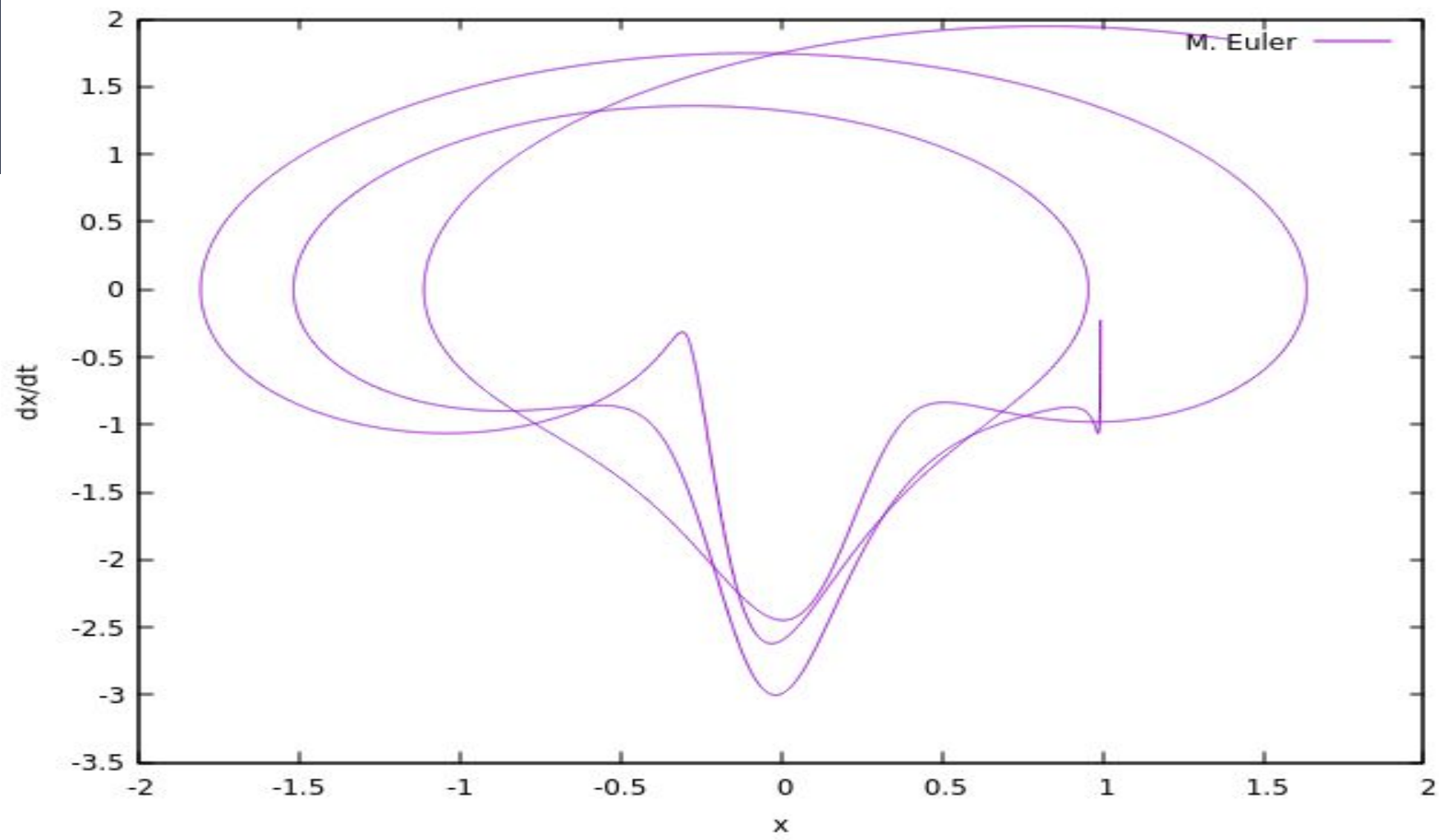
Método de Euler



$$\frac{d\mathbf{y}(t)}{dt} \simeq \frac{\mathbf{y}(t_{n+1}) - \mathbf{y}(t_n)}{h} = \mathbf{f}(t, \mathbf{y}),$$

$$\mathbf{y}_{n+1} \simeq \mathbf{y}_n + h\mathbf{f}(t_n, \mathbf{y}_n),$$







Runge-Kutta de 4 orden



$$\mathbf{y}_{n+1} = \mathbf{y}_n + \frac{1}{6}(\mathbf{k}_1 + 2\mathbf{k}_2 + 2\mathbf{k}_3 + \mathbf{k}_4),$$

$$\mathbf{k}_1 = h\mathbf{f}(t_n, \mathbf{y}_n),$$

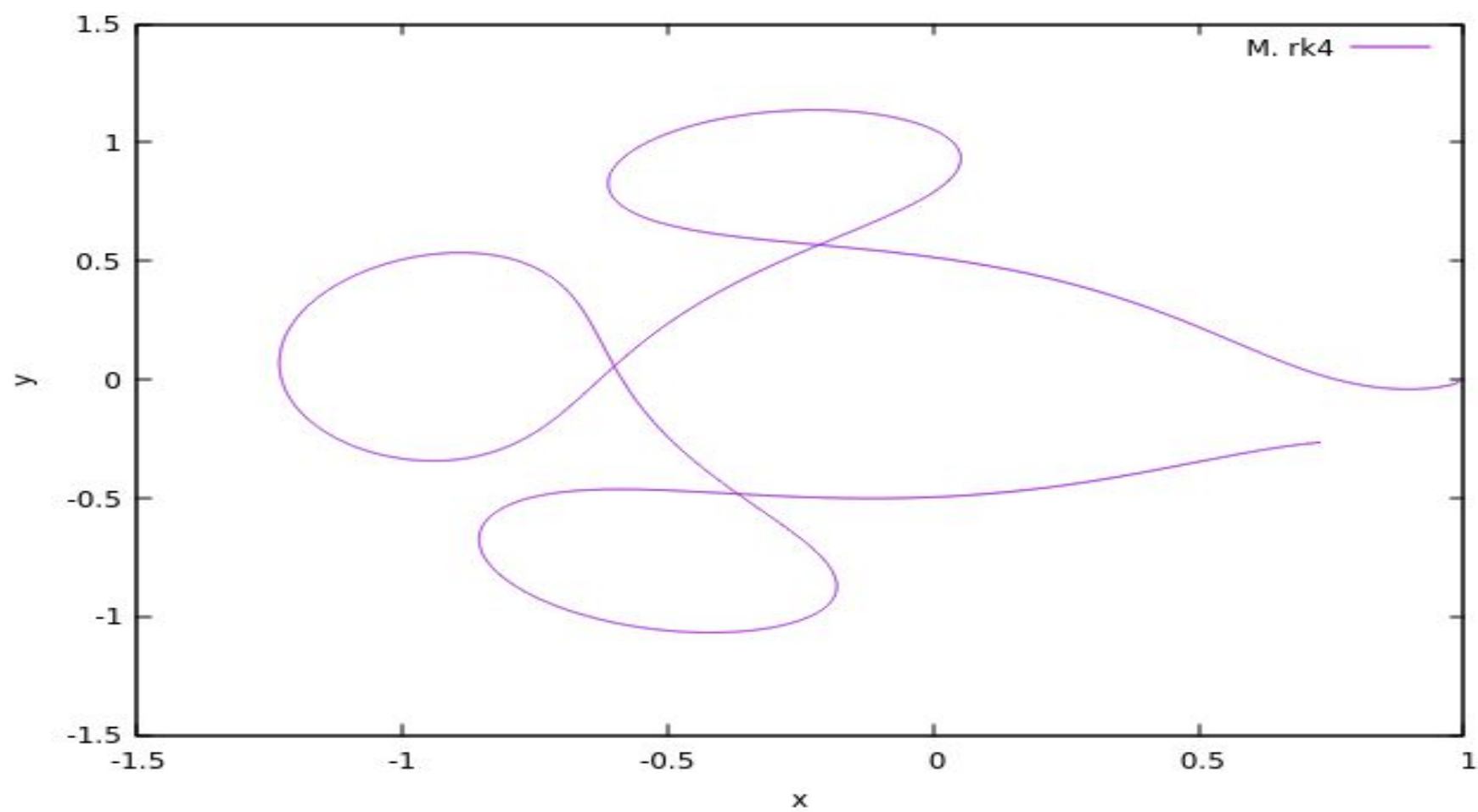
$$\mathbf{k}_2 = h\mathbf{f}\left(t_n + \frac{h}{2}, \mathbf{y}_n + \frac{\mathbf{k}_1}{2}\right),$$

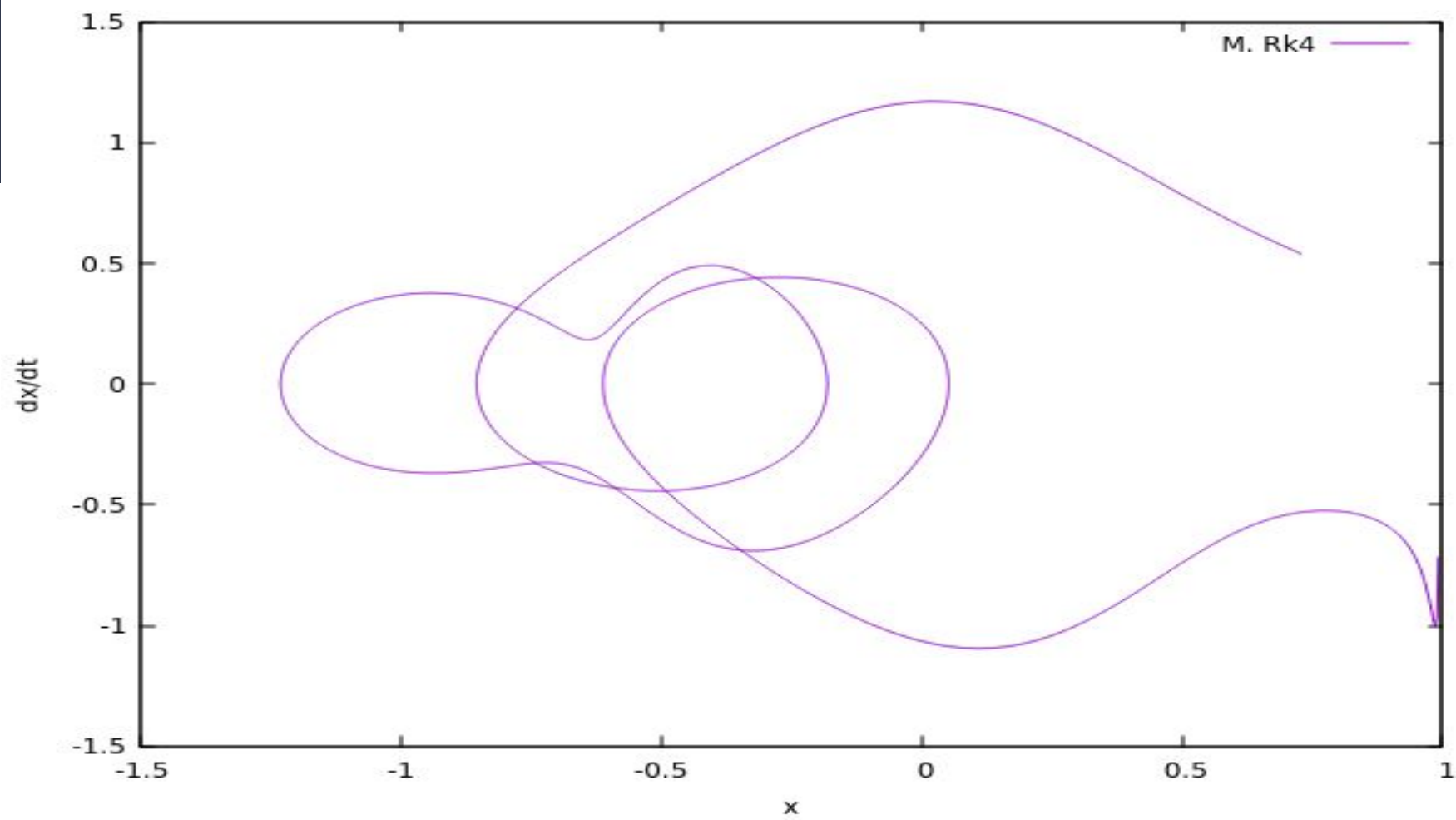
$$\mathbf{k}_3 = h\mathbf{f}\left(t_n + \frac{h}{2}, \mathbf{y}_n + \frac{\mathbf{k}_2}{2}\right),$$

$$\mathbf{k}_4 = h\mathbf{f}(t_n + h, \mathbf{y}_n + \mathbf{k}_3).$$

Matriz de Butcher

0				
1/2	1/2			
1/2	0	1/2		
1	0	0	1	
<hr/>				
	1/6	2/6	2/6	1/6





Método Dormand-Prince 5(4)

$$\begin{aligned}k_1 &= hf(t_k, y_k), \\k_2 &= hf(t_k + \frac{1}{5}h, y_k + \frac{1}{5}k_1), \\k_3 &= hf(t_k + \frac{3}{10}h, y_k + \frac{3}{40}k_1 + \frac{9}{40}k_2), \\k_4 &= hf(t_k + \frac{4}{5}h, y_k + \frac{44}{45}k_1 - \frac{56}{15}k_2 + \frac{32}{9}k_3), \\k_5 &= hf(t_k + \frac{8}{9}h, y_k + \frac{19372}{6561}k_1 - \frac{25360}{2187}k_2 + \frac{64448}{6561}k_3 - \frac{212}{729}k_4), \\k_6 &= hf(t_k + h, y_k + \frac{9017}{3168}k_1 - \frac{355}{33}k_2 - \frac{46732}{5247}k_3 + \frac{49}{176}k_4 - \frac{5103}{18656}k_5), \\k_7 &= hf(t_k + h, y_k + \frac{35}{384}k_1 + \frac{500}{1113}k_3 + \frac{125}{192}k_4 - \frac{2187}{6784}k_5 + \frac{11}{84}k_6).\end{aligned}$$

Runge-Kutta 4

$$y_{k+1} = y_k + \frac{35}{384}k_1 + \frac{500}{1113}k_3 + \frac{125}{192}k_4 - \frac{2187}{6784}k_5 + \frac{11}{84}k_6.$$

Runge-Kutta 5

$$z_{k+1} = y_k + \frac{5179}{57600}k_1 + \frac{7571}{16695}k_3 + \frac{393}{640}k_4 - \frac{92097}{339200}k_5 + \frac{187}{2100}k_6 + \frac{1}{40}k_7.$$

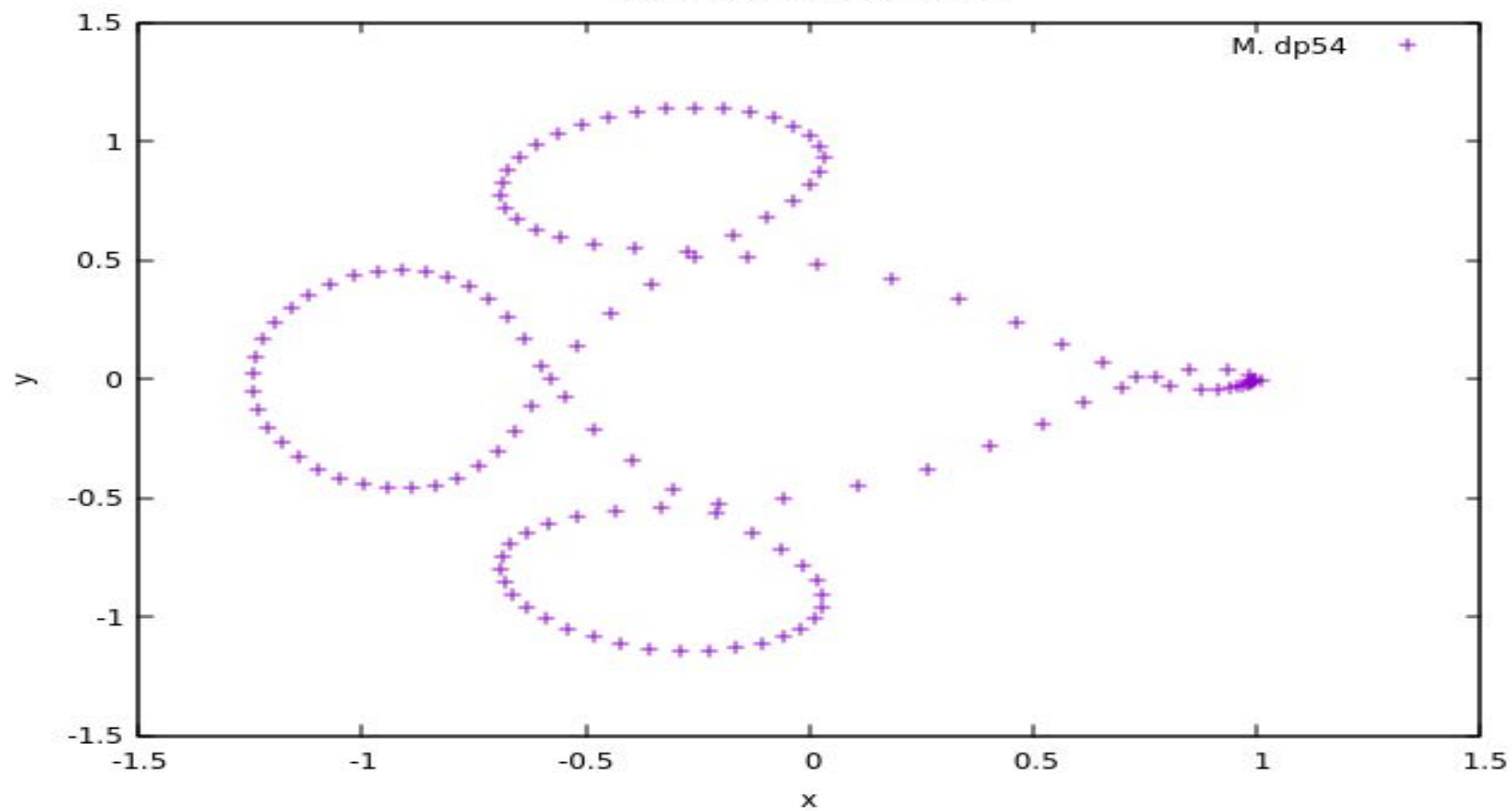
Paso adaptativo

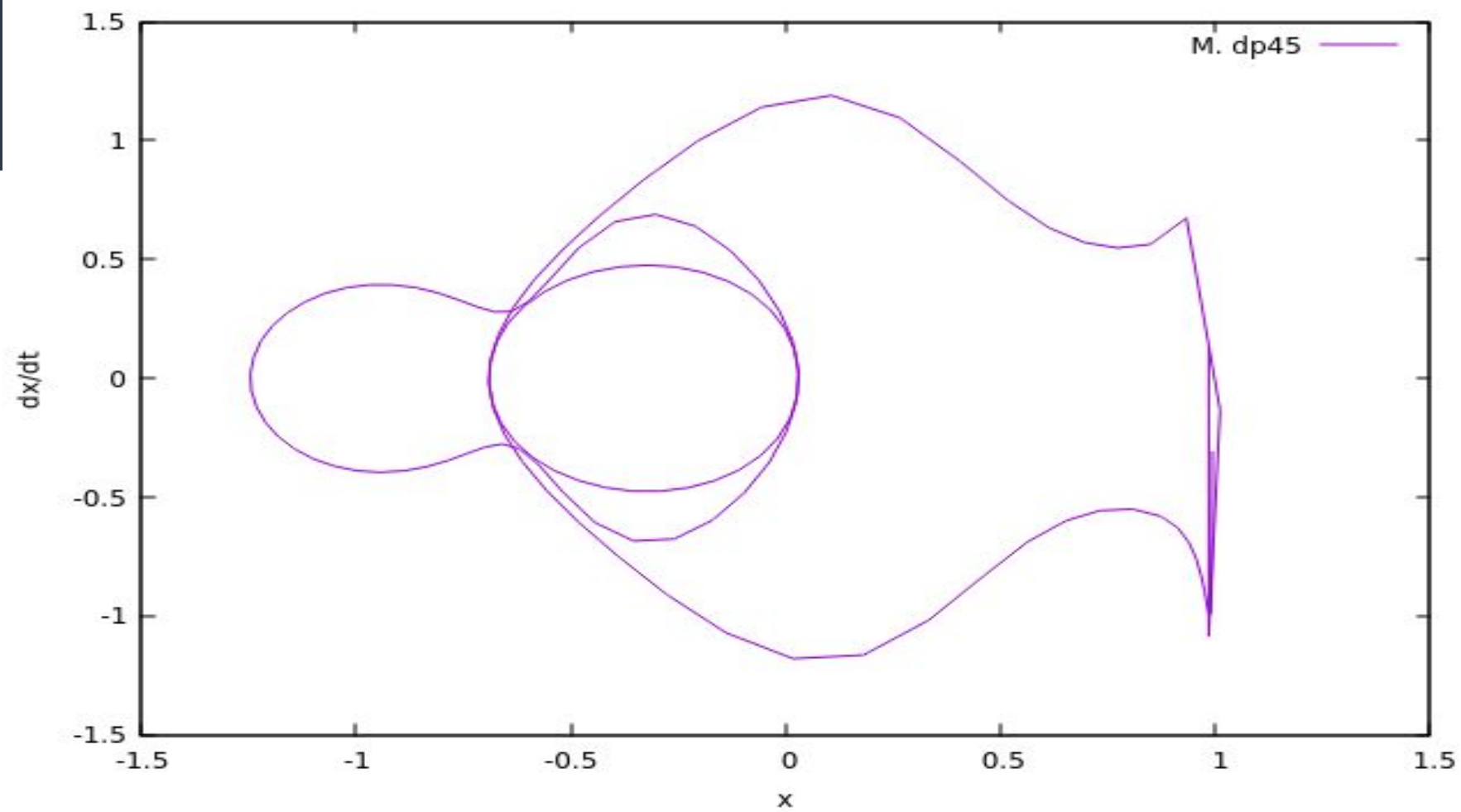
$$|z_{k+1} - y_{k+1}| = \left| \frac{71}{57600}k_1 - \frac{71}{16695}k_3 + \frac{71}{1920}k_4 - \frac{17253}{339200}k_5 + \frac{22}{525}k_6 - \frac{1}{40}k_7 \right|.$$

```
double dtnew(const double p, const double v, const double dt)
{
    double hmin=0.0001;
    double hmax=0.137;
    double sprom = (saux(p,dt) + saux(v,dt))*0.5;
    if(dt*sprom <= hmin){
        return hmin;
    }
    if(dt*sprom >= hmax){
        return hmax;
    }
    else{
        return dt*sprom;
    }
}
```

$$s = \left(\frac{\epsilon dt}{2|z_{k+1} - y_{k+1}|} \right)^{\frac{1}{5}}$$
$$dt_{opt} = s dt$$

Dp54 con paso adaptativo



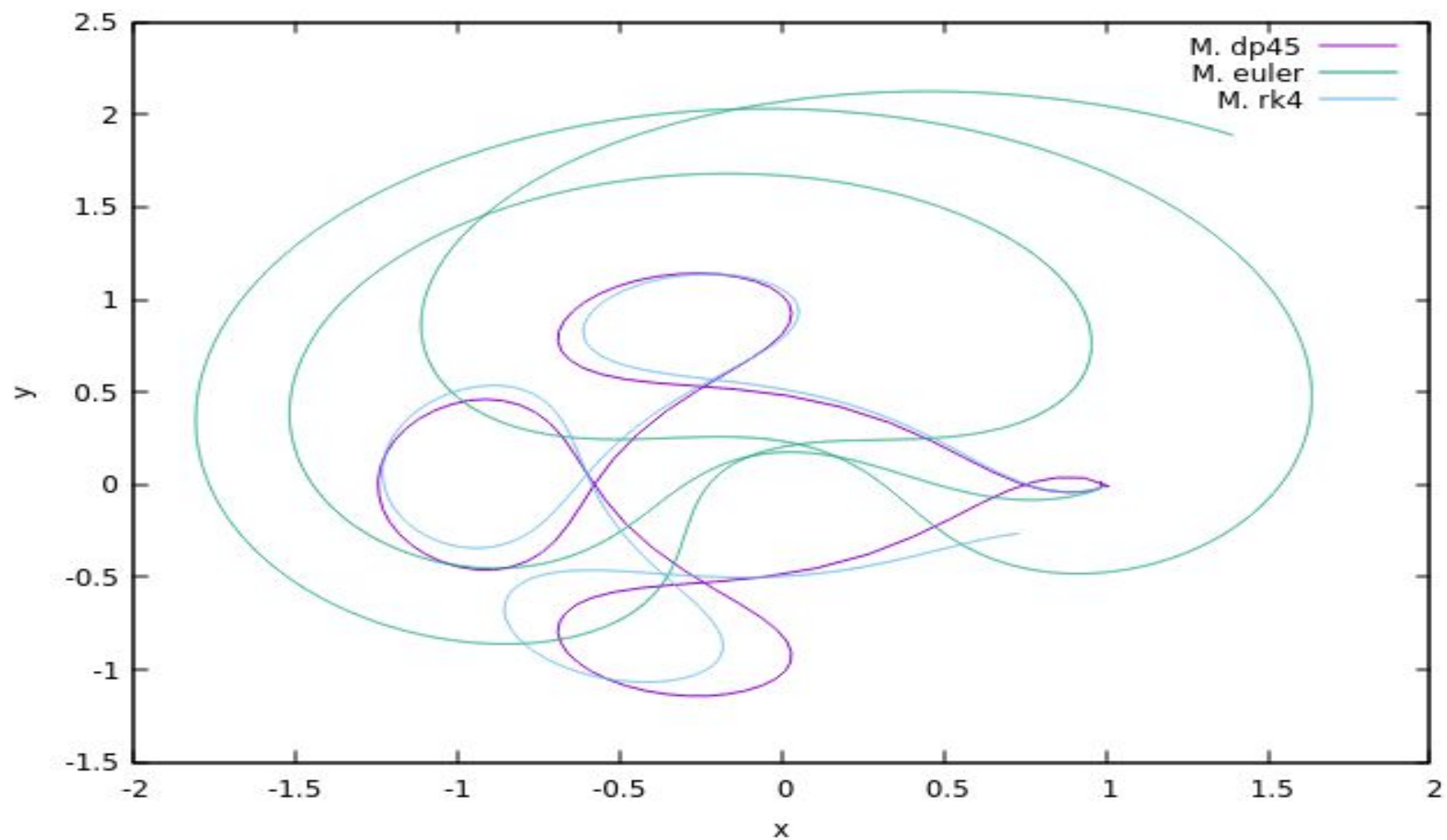


Cantidad de iteraciones con cada método para obtener una órbita completa.

M. Euler , 24.000 iteraciones

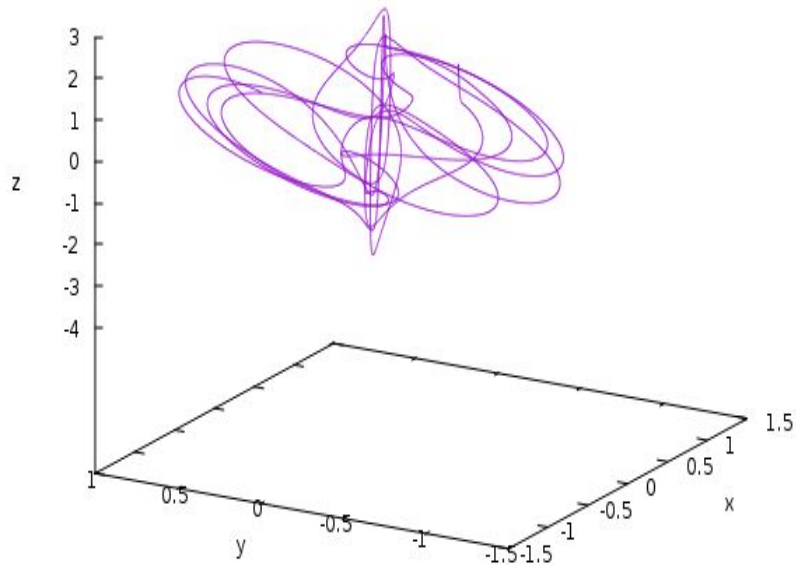
M. Rk4 , 5.977 iteraciones

M. Dp5(4), 134 iteraciones

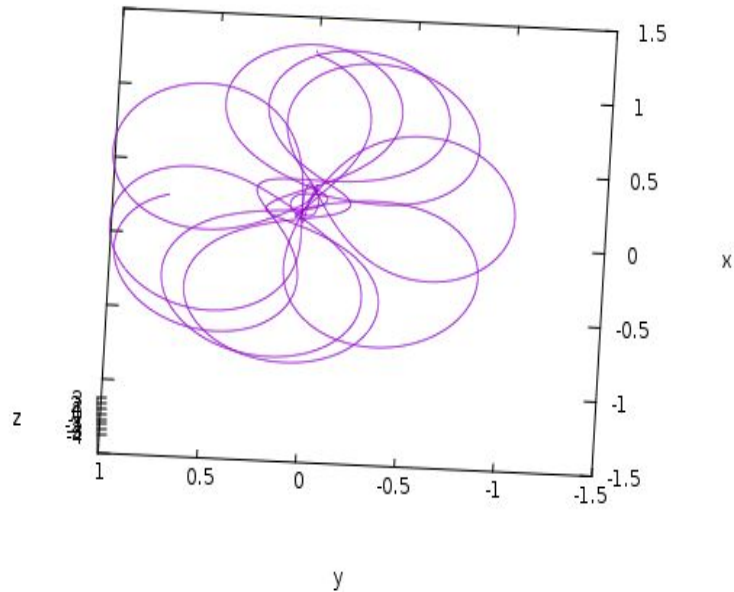


Órbitas más generales

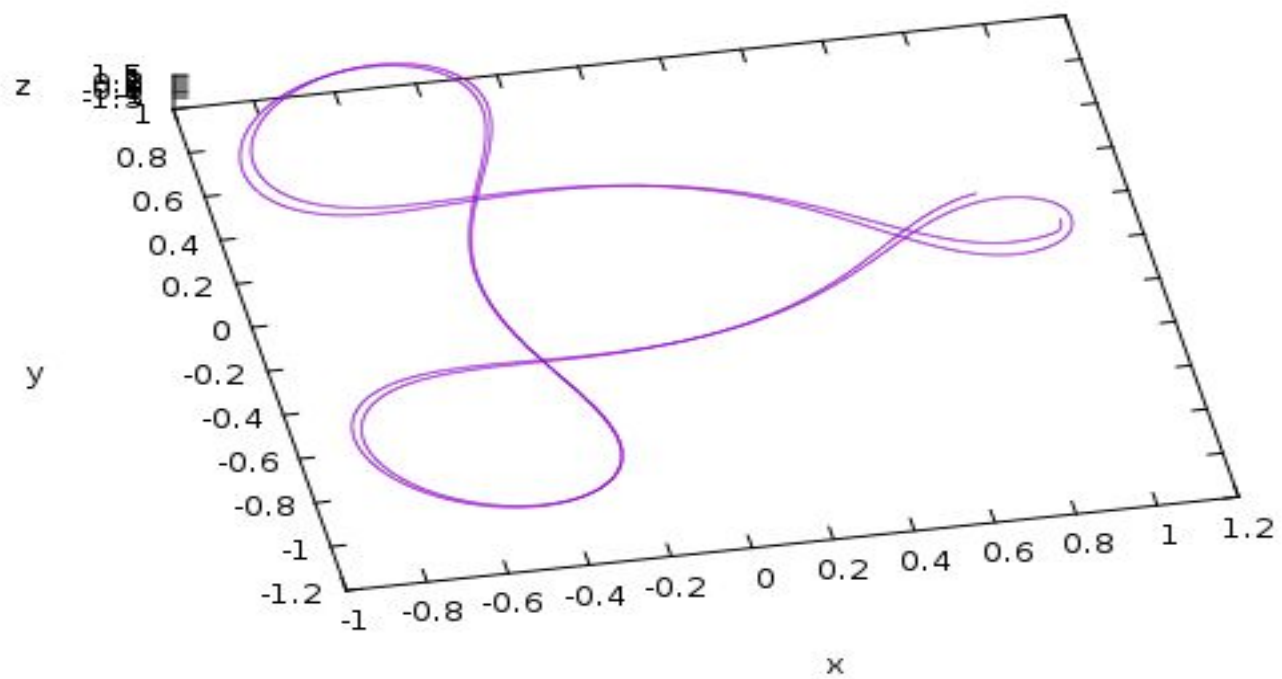
Dormand-prince $z \neq 0$ —



Dormand-prince $z \neq 0$ —



Dormand-prince $z!=0$ —



Bibliografía

- Simon Sirca and Martin Horvat. Computational Methods for Physicists. Springer Berlin Heidelberg, 2012.
- Toshinori Kimura, On Dormand-Prince Method, 2009.
- logotipo NASA. https://commons.wikimedia.org/wiki/File:NASA_logo.svg;